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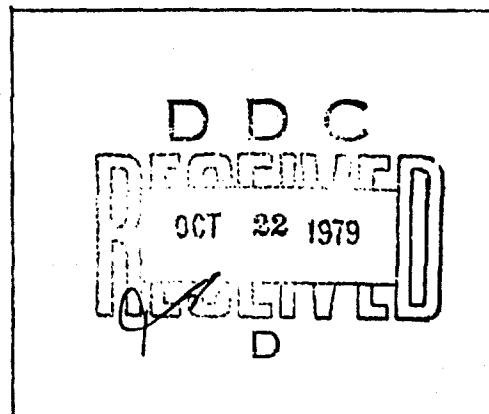
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REPORT DOCUMENTATION PAGE			READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER UCRL-50108-Vol. 1-Rev. 1	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER	
4. TITLE (and Subtitle) Compendium of Shock Wave Data. Introduction. Section Al. Elements.		5. TYPE OF REPORT & PERIOD COVERED	
		6. PERFORMING ORG. REPORT NUMBER	
7. AUTHOR(s) M. van Thiel J. Shaner E. Salinas (editors)		8. CONTRACT OR GRANT NUMBER(S) W-7405-ENG-48	
9. PERFORMING ORGANIZATION NAME AND ADDRESS Lawrence Livermore Laboratory P.O. Box 808 Livermore, CA 94550		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS	
11. CONTROLLING OFFICE NAME AND ADDRESS Lawrence Livermore Laboratory P.O. Box 808 Livermore, CA 94550		12. REPORT DATE June 1977	13. NUMBER OF PAGES 663 pages
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		15. SECURITY CLASS. (of this report) Unclassified	
		16a. DECLASSIFICATION/DOWNGRADING SCHEDULE	
16. DISTRIBUTION STATEMENT (of this Report) Distribution limited to U.S. Government Agencies only. All other requests, both foreign and domestic, must be referred to [REDACTED], Lawrence Livermore Laboratory, P.O. Box 808, Livermore, CA 94550.		TUE 22 OCT 1979	
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)			
18. SUPPLEMENTARY NOTES			
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) shock waves very high pressures handbook elements equations of state thermodynamic properties			
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) This three volume set lists in a concise manner the thermodynamic data in condensed media obtained by shock wave techniques. Both dynamic variables and volumetric quantities associated with the shock wave are given. The format was selected to make the volume useful in engineering as well as scientific research activities. Volume 1-Rev. 1 consists of a general introduction and Section Al which lists shock wave data for Elements. The entire set should be useful to people working either in the shock wave field or in the area of thermodynamic properties at high pressure.			

COMPENDIUM OF SHOCK WAVE DATA

INTRODUCTION AND SECTION A1

M. van Thiel

June 1977

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UCRL-50108 Vol. 1 Rev. 1

COMPENDIUM OF SHOCK WAVE DATA

Introduction
Section Al-Elements

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MS. date: June 1977

Acknowledgments

For their past support, we thank V. Davis, who served as programmer and editor of the first edition; E. Collins, who updated the programs; K. Johnson and G. Wilcox, who provided programming support; A. A. Kusubov and A. C. Mitchell, who served as editors of the first edition; and R. E. Duff, who suggested and encouraged this work.

Notice

The completeness of this compilation depends upon its users. To assure its continued usefulness, users are urged to send any missing or new shock wave data and corrections to the Editor, M. van Thiel, Lawrence Livermore Laboratory, P.O. Box 808, Livermore, California, U.S.A., 94550.

New and revised pages will be distributed as necessary.

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COMPENDIUM OF SHOCK WAVE DATA

Introduction

Abstract

The purpose of this volume is to list in a concise manner the thermodynamic data in condensed media obtained by shock wave techniques. The volume should be useful both to people working in the shock wave field and to others interested primarily in thermodynamic properties at high pressure. Therefore, both dynamic variables and volumetric quantities associated with the shock wave are given. The format has been selected to make the volume useful in engineering as well as scientific research activities.

Content

Tabular Data — The information tabulated is generally derived from shock compression of a sample of known density. Any two of the measurable quantities tabulated (pressure, volume or density, free-surface and particle velocity behind the shock front, as well as shock velocity) may be used to derive the others. Shock and free-surface velocities were measured in most of the experiments performed to date. A description of the basic assumptions and techniques has been given by Rice et al.¹ Some useful texts also exist for those interested in the experimental techniques² and for those more theoretically inclined.²⁻⁴

Pressure (P), volume (V), and energy (E) are usually obtained from the shock (U_{s_1}) and particle velocity (U_p) by application of the conservation conditions of mass momentum and energy across a well-defined shock wave. It is, therefore, assumed that steady state exists in front and behind this wave. The conditions are schematically presented in Fig. 1. The distance between the two (cross-hatched) steady-state regions is usually small compared to the dimensions of the system. Conservation of mass and momentum then yields:

$$P_1 - P_0 = \left(U_{s_1} - U_{p_0} \right) \left(U_{p_1} - U_{p_0} \right) / V_0$$

and

$$V_1/V_0 = 1 - \frac{U_{p_1} - U_{p_0}}{U_{s_1} - U_{p_0}},$$

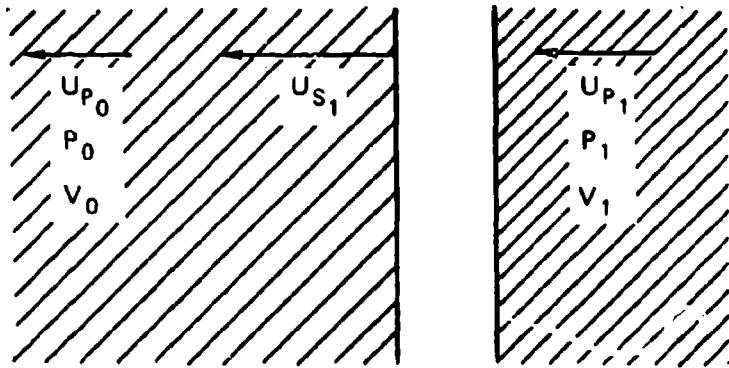


Fig. 1. Shock wave conditions.¹

while the energy conservation condition is:

$$E_1 - E_0 = (1/2) (P_0 + P_1) (V_0 - V_1).$$

In the usual single wave experiment, the initial particle velocity, U_{p0} , is zero. Frequently multiple waves are considered in this compilation. The more general form of the above equations for multiple wave systems is obtained by replacing the subscript 0 by i and 1 by i + 1.

Whenever they are obtainable from shock wave measurements, other directly measured derivatives, such as heat capacity and expansion coefficient, are also listed. However, there is no developed technology for these measurements at the time of this writing.

A convenient summary of the tabulated data may be given in terms of a functional relation between shock (U_s) and particle velocity (U_p) since a valuable feature of dynamic (shock wave) measurements is the slow, and in many cases, vanishing curvature of plots of U_s vs U_p . Least-square fits have been given in this plane whenever the data warrant it:

$$U_s = C + \sum_{i=1}^n s_i U_p^i.$$

The sum is limited to $n_{\max} = 1$ for most solids. This Hugoniot equation may be transformed to the pressure-volume plane by use of conservation of mass and momentum equations.

With each set of data from a particular source, the properties that define the initial condition of the sample before shock compression are listed if available. These are:

- V_0 : Initial volume.
- V_{0i} : Volume from crystallographic or accurate volumetric techniques (useful for porous materials).
- T_0 : Initial temperature.
- Composition.
- C_L , C_s , and C_b or C_0 : Respective longitudinal sound velocity, transverse (or shear) wave velocity, and "bulk sound velocity" $[\equiv (\partial P / \partial \rho)_S^{1/2} \text{ km/sec}]$ derived from static compression or for an isotropic sample $(C_L^2 - 4/3 C_s^2)^{1/2}$.

The quantity C_0 is helpful in allowing interpolations of the bulk compressibility between the lower limit of the dynamic range and the compressibility at one atmosphere.

It may be noted that no subscripts have been printed in the body of the compendium. This is due to the limited character list of the automatic printer that produced these pages. Therefore, V_0 is written as V0, C_L' as CL, etc.

While the above characterization is adequate for most purposes, some structural transformations are comparable to or slower than the measuring time in shock experiments. Under these conditions, other sample specifications are needed to adequately describe the sample. This problem has come up in detailed investigations of dynamic yield points and in the study of phase transitions and chemical reactions. In the former case, for example, impurity concentration and thermal history become highly important. Mixtures of solids and liquids exhibit relaxation times that depend in a complex fashion on microscopic structure and strength and, for high pressure shocks, on the temperature distribution behind the front.

Comments — A comments section is provided for further characterizations of the material and the experiment. The first comment is reserved for a source statement. This may be the name of a person and an address or journal reference. The second comment contains a characterization of the experimental method and the technique of data analysis. To avoid lengthy descriptions on each data page, short descriptions of each method are given in the section on techniques. These descriptions, of course, do not dwell on details of experimental setup or data handling but are primarily an attempt to give a short and unambiguous classification of a series of data recording and analysis methods. For more detail, the user is referred to papers and books on the subject. A review has been given by Duval and Fowles.²

Any number of comments may be listed by the compiler or suggested by the investigator. The extent and format of these comments are flexible and depend on relatively arbitrary decisions of the compiler. It is clear that these comments can never replace the full publication.

Filing Order

The filing order chosen for this volume is independent of arbitrary nomenclature and trade names and, when understood, is equally useful to any language group. However, for those users preferring the English language as a basis for searching this volume, an index for each section is provided. The author index for each section is a convenient guide to the contributors.

The filing order is based on the periodic table. The basic system is described in detail by the Office of Critical Tables,⁵ and some modifications have been made in accordance with the latest modification of a numerical order of the elements proposed by Rossini.⁶ (See Table 1 and Fig. 4 at the end of the techniques listings.) With some small exceptions, the categories, too, are essentially those given by Rossini. They are:

- A₁. Elements: Filed by order number. For example: O - #1 is filed before H - #2, before He - #3, etc.
- A₂. Compounds, including intermetallic compounds: These are divided into groups which are determined by the element in the compound with the largest order number. Subsets within this group are then determined by the element with the next to highest number, etc. To this end, each compound is characterized by the set of order numbers (called index or finding numbers) of its elements written in order of decreasing order number.

For example:

Name	Formula	Index
Sodium oxide	Na ₂ O*	99-1
Sodium hydroxide	NaOH	99-2-1
Sodium fluoride	NaF	99-9
Sodium chloride	NaCl	99-10
Sodium sulfate	Na ₂ SO ₄	99-14-1
Sodium carbonate	Na ₂ CO ₃	99-23-1

* Note that the automatic printer used for the data pages is restricted to capital letters, so that the molecular formulas must be written with their elements separated by dashes or brackets: Na₂O becomes NA2-O, etc.

These are all members of the group filed under sodium in the above order. All potassium compounds (#100) containing elements with numbers not larger than 100 follow this group, while all lithium compounds containing elements with numbers not larger than 98 precede the sodium group, etc.

The only compounds excluded from this category are those with two or more carbon atoms (#23) and no elements above #23. These are filed separately under categories B and C.

B. Hydrocarbons: Organic compounds containing only C (#23) and H (#2).

For example:

Name	Formula	Index
Methane	CH ₄	23-2 (1-4)
Benzene	C ₆ H ₆	23-2 (6-6)
Hexane	C ₆ H ₈	23-2 (6-8)
Octane	C ₈ H ₁₀	23-2 (6-10)

are listed in the above order. Therefore, in order to find a hydrocarbon, one must first find the set containing the correct number of carbon atoms and, second, the compound with the proper number of hydrogen atoms of that set.

Polymers are filed under the monomer.

- C. All other organic compounds: Compounds with more than two carbon atoms (#23) and no element with order number larger than 23.
- D. Alloys, solutions, and mixtures: Materials that are not simple compounds and that contain no component (element or compound) over 90 mole percent. Those that contain a component over 90 mole percent will be listed under A₁, A₂, B, or C to avoid scattering almost pure materials under different headings.

The finding number may be a compound number derived from the constituent compounds. The leading index will be that of the major constituent - not necessarily the one containing the largest order number.

For example:

41-24-1--93-24-1

Olivine

(Name)	(Formula)	(Composition)
Fayalite	Fe ₂ SiO ₄	55.0%
Forsterite	Mg ₂ SiO ₄	45.0%

In this case, the compound index consists of the indexes of two minerals (Fayalite and Forsterite). These indexes are of necessity complex, but they allow for a minimum amount of conflict in filing order. No distinction has, so far, been made between solutions and granular mixtures in generating the index numbers.

The above example forms the heading for the data page of the particular rock.

A large number of the materials studied have not been characterized sufficiently to allow assignment of a finding number. These materials are, however, useful for many engineering applications. They are synthetic composites and geological materials and are generally categorized as mixtures or organic plastics. A separate section has been provided for these materials at the end of this compendium.

Techniques

Experimental:

- A. Pin: The surface accelerated by the shock wave under study may be used as one pole of a switch. The value and complexity of this technique for locating shock and free-surface position as a function of time are well illustrated by the experiments on iron of Bancroft et al.⁷ In these experiments, the switch completes a current path and causes a capacitor to discharge. Alternatively, a voltage or current pulse may be produced by a ferroelectric or piezoelectric crystal on impact with a surface. A description of some pin types has been given by van Thiel and Alder.⁸
- B. Optical flash: The moving surface may be used to compress a thin layer (0.01-0.1 mm) of argon or xenon gas. This shock compression results in a flash which is then recorded by a smear camera. This technique has been adequately described by Rice et al.¹ and by McQueen et al.⁹
- C₁. Reflectivity change of a mirror: An optically reflecting surface will turn opaque when impacted with sufficient strength by a shocked material. A smear camera is focused on a mirror placed on the surface to be shocked or some distance from it and records arrival of the shock wave or the free surface, respectively.²

- C₂. **Turning mirror:** Frequently surface structure of shock strength is such that the surface of the mirror does not change reflectivity rapidly enough. In this case, the mirror may be placed at an angle to the shock front or surface. Impact then causes the mirror to turn through an angle, thus removing a parallel beam of light from the optical path. This technique is more reliable for smooth surfaces than technique C₁ but requires more care in the optical setup.²
- C₃. **Total reflection of a prism¹⁰:** This is a variation of the mirror technique. Closing a void (as small as a micron) between a sample and a glass or other transparent prism will change the angle of total internal reflection. A beam of light reflected internally close to the critical angle will, therefore, be rapidly cut off when the void is closed. The prism may be placed either on the surface or at some angle to it to determine either the arrival time of the shock or the velocity of the surface.
- D. **"Optical lever":** The optical-lever technique employs the change of position of an optical image in a reflecting surface when this surface is turned by a shock wave. In this case, the shock front makes a known finite angle with the surface. The measurement of the displacement of a series of images made by different parts of the surface and the observation of the time of displacement of these images yields the required shock and free-surface parameters.²
- E. **"Knife edge":** The knife-edge or moving-image technique yields higher spatial resolution than the optical-lever technique. Here a thin wire or knife edge is suspended over the polished surface of the specimen. The separation of the knife edge and its image may be photographed as a function of time to yield the surface velocity. This technique was first reported by Davis and Craig.¹¹
- F. **Resistance wire:** This technique, first described in a Sandia Laboratories report,¹² has been used to measure free-surface velocity. Here a change of resistance is measured as the moving surface shorts out a resistance wire positioned at an angle to the surface. The velocity of the free surface is then obtained from the resistance-time record. Shock velocity may be obtained if the wire is parallel to the surface and the shock direction is other than normal.

- G. Capacitor: The capacitor technique measures the change of capacitance of a parallel plate capacitor, one plate of which is the surface whose velocity is to be measured.¹³
- H. Polarization: The polarization technique¹⁴ uses the shape of an electrical pulse produced by the shock to measure shock velocity in cases where the shock wave changes the dielectric polarization of the sample.
- I. Transducer: While the above techniques measure shock or surface motion, several transducer techniques that give pressure directly have been developed or are under development. In these cases, a measured voltage change is related to pressure by calibration against one of the primary techniques mentioned above. Some examples are:
- Conductivity gauge: The sulfur gauge was first developed by Eichelberger and Hauver.¹⁵ It is based on the change of conductivity of a sulfur wafer with pressure. Another useful material for pressure measurement is manganin wire, used previously in static work because of its linear pressure versus resistivity curve. Some work has been done to extend this curve to higher pressures.¹⁶
 - Piezoelectric gauge: The quartz gauge¹⁷ and the lithium niobate gauge,¹⁸ which are based on the piezoelectric coefficient, yield useful pressure measurements up to about 50 kbar and 15 kbar, respectively. A calibration curve may be obtained from R. A. Graham, Sandia Laboratories, Albuquerque. Tourmaline has been used and a calibration up to 21 kbar¹⁹ has been reported.
- J. Radiographic methods: These may measure either one of two sets of parameters.
- The shock velocity and density change across the shock front. This method has been described by Schall.²⁰
 - The shock velocity and the velocity of an embedded foil of higher specific absorption index. This method is valuable in that it yields P, V, and E points independently of a model or a standard material. Its limitations lie in the relatively poor shock front resolution.
- K. Magnetic induction: This may be used in interface and free-surface velocity measurements.

- A current loop made of foil or flattened wire may be embedded in a nonconducting sample or placed at an interface.^{21,22} The voltage induced in the wire when the wire is moved through an externally applied uniform magnetic field is a measure of material velocity.
- Similarly, a voltage may be induced in a current loop placed concentric to a magnetic pole and close to a moving surface.²³ This method is relatively inexpensive and may be used to measure the motion of any metallic-nonmetallic interface.

Data Reduction:

The parameters obtained from most experimental techniques used to date are shock velocity (U_s) and surface velocity (U_{fs}). The surface velocity may be either the velocity of a projectile generating pressure in the sample (Fig. 2) or the velocity of the sample surface after the shock has traveled through the sample. The pressure change $P - P_0 = \rho_0 U_s U_p$ is a function of shock velocity and the change of mass velocity across the front. Techniques for calculating this latter velocity are discussed below.

A. If the projectile and the sample (Fig. 2(a)) are made of the same material, the shock velocity in the sample and in the projectile (relative to the unshocked medium in front of the wave) is the same. Therefore, the dynamic impedance $\rho_0 U_s = P/U_p$, $\tan\theta_1 = \tan\theta_2$, where from symmetry (Fig. 2(b)) U_p is equal to one-half the projectile velocity before impact and equal to the change of mass velocity across the wave. If the projectile and the sample are not the same, the projectile Hugoniot (H_2) is used as a known standard (Fig. 2(c)).

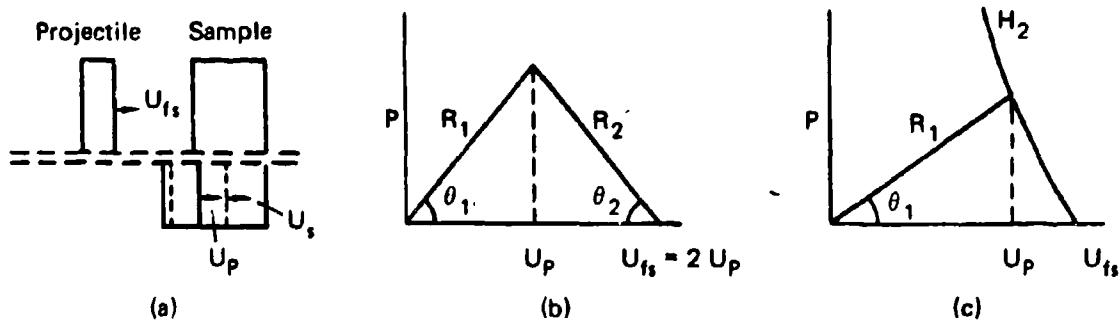


Fig. 2. Velocity diagrams.

The solution is then given by the intersection of R_1 and H_2 . The equations to be solved are therefore:

In Fig. 2(b):

$$P - P_0)_{\text{sample}} = (\rho_0 U_s) \quad (1a)$$

and

$$P - P_0)_{\text{projectile}} = (\rho_0 U_s) (U_{fs} - U_p). \quad (1b)$$

Or i g. 2(c):

$$P_0)_{\text{sample}} = (\rho_0 U_s) U_p, \quad (1c)$$

and the equation of the projectile Hugoniot

$$(P - P_0)_{\text{projectile}} = f(U_{fs} - U_p). \quad (2)$$

- B₁. Where the velocity of a projectile is not measured, the sample is positioned on a plate (the standard) with a known Hugoniot locus (H) and unloading isentropes (a). As is shown in Fig. 3, the solution is given by the dynamic impedance condition of the sample

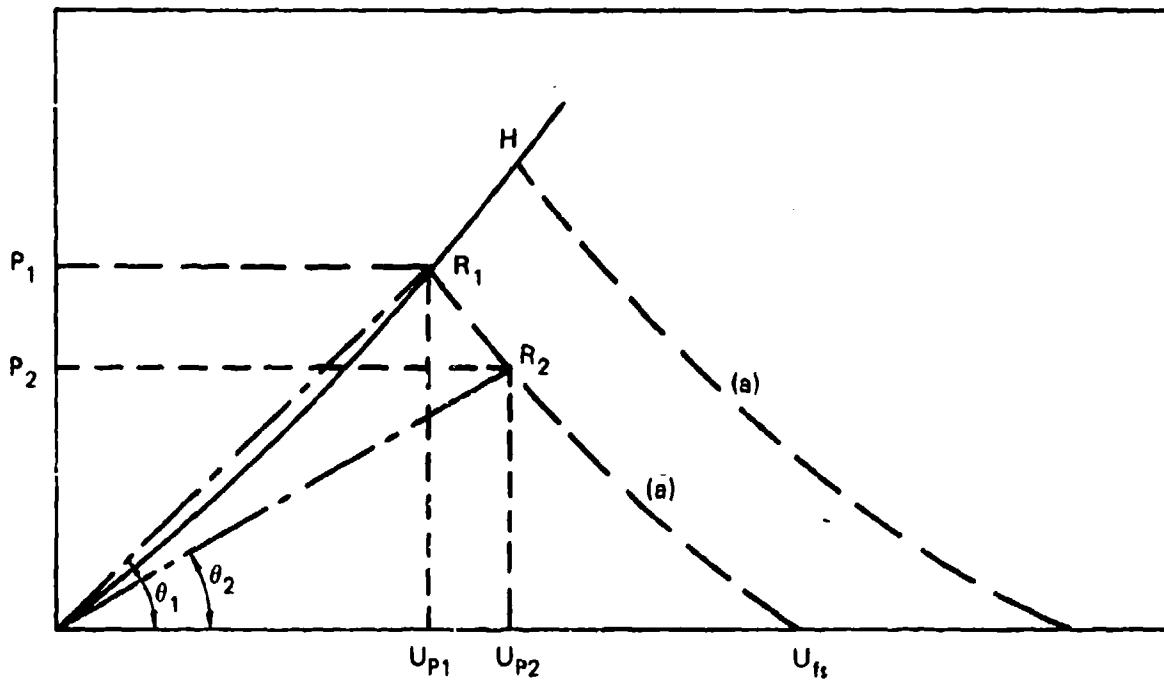


Fig. 3. Projectile velocity.

$(\tan\theta_1 = \rho_0 U_s)$, and the particular unloading path (a) that crosses the Hugoniot point of the plate at the experimental pressure (P_1). The latter is selected on the basis of a measurement of U_{fs} or $\rho_0 U_s = \tan\theta_1$ of the plate. The solution of P_1 and U_{p1} is, therefore, obtained from:

$$P_2 - P_0 = \Delta P \text{ (sample)} = (\tan\theta_2) U_{p2} \quad (1d)$$

and the equation of the unloading adiabat

$$\begin{aligned} P_2 &= P_a \text{ (plate)} = f_a (U_s, U_{fs} - U_{p2}) \\ &\quad = f'_a (U_{fs}, U_{fs} - U_{p2}). \end{aligned} \quad (3)$$

- B₂. A modification of technique B₁ places the material with known equation of state on the sample. The measured quantities are usually the shock and (or) free-surface velocity of the known standard and the shock velocity of the unknown. Consider P_2 as the known pressure and P_1 as the pressure to be determined (Fig. 3). Since the cross curve (a) is not known in this case, the acoustic approximation angle

$$P_2 R_2 R_1 \approx \theta_1 = \rho_1 U_{s1} \quad (4)$$

may be made to yield a first-order solution

$$P_1/P_2 = (\theta_1 + \theta_2)/2\theta_2. \quad (5)$$

After several points are thus obtained, the solution may be improved by iteration. When the shock front is not parallel to the sample-standard interface, appropriate components must be used.²⁴

- C. Where the pressure is measured directly with a transducer, the other parameter usually measured is the shock velocity through the sample. The conservation of mass and momentum conditions mentioned before may be rearranged to give

$$V/V_0 = 1 - \frac{U_p}{U_s} = 1 - \frac{P - P_0}{\rho_0 U_s^2}. \quad (6)$$

Once P and V are known, conservation of energy then yields the energy change

$$E - E_0 = 1/2 (P_0 + P) (V_0 - V). \quad (7)$$

- D. If only U_s and U_{fs} are available from a set of experiments, a Hugoniot may be obtained with:
- The assumption that $U_p = 1/2 U_{fs}$. This is a good approximation at sufficiently low shock compression.
 - Otherwise, a model of the thermal expansion or thermal pressure coefficient is used to calculate adiabats from an assumed Hugoniot. A first-order estimate may be the first assumption above.

The Riemann integral

$$U_r = \int_{V_0}^V \sqrt{\left| \frac{\partial P}{\partial V} \right|} dV$$

along an isentrope is then assumed to yield the increase in mass velocity when shocked material, initially at some pressure P_1 and particle velocity U_p (Fig. 3), unloads to P_0 and $U_{fs} = U_p + U_r$. These techniques have been discussed in detail by Walsh²⁵ and by Al'tshuler.²⁶

This model assumes equilibrium within the state considered. Some uncertainty exists at the time of this writing on how to treat small but measurable effects due to slow relaxation processes behind or in a shock wave. These may be important near-phase transitions or in the pressure range of elastic yield.

- E. Where a radiographic method has given U_s and V/V_0 , U_p is obtained directly from Eq. (6) and P from the pressure equation given on page 1.
- F. The before-mentioned conservation conditions may be applied without rearranging where U_s and U_p are given directly.
- G. If a transducer with a known Hugoniot equation ($U_s = F(P)$, $U_p = F(P)$) like quartz is impacted onto the sample or is impacted by the unknown, all required quantities can be calculated from a measure of P and impact velocity, since $P(tr) = P(sa)$, and $U_p(sa) = U - U_p(tr)$, where tr is the transducer, sa is the sample and U is the impact velocity.

Table 1. Order of the elements.

Actinium	Ac	91	Indium	In	31
Aluminum	Al	29	Iodine	I	12
Americium	Am	85	Iridium	Ir	46
Antimony (Stibium)	Sb	21	Iron	Fe	41
Argon	Ar	5	Krypton	Kr	6
Arsenic	As	20	Lanthanum	La	76
Astatine	At	13	Lead	Pb	27
Barium	Ba	96	Lithium	Li	98
Berkelium	Bk	83	Lutetium	Lu	62
Beryllium	Be	92	Lawrencium	Lw	77
Bismuth	Bi	22	Magnesium	Mg	93
Boron	B	28	Manganese	Mn	48
Bromine	Br	11	Mendelevium	Mv	79
Cadmium	Cd	34	Mercury (Hydrargyrum)	Hg	35
Calcium	Ca	94	Molybdenum	Mo	52
Californium	Cf	82	Neodymium	Nd	73
Carbon	C	23	Neon	Ne	4
Cerium	Ce	75	Neptunium	Np	87
Cesium	Cs	102	Nickel	Ni	39
Chlorine	Cl	10	Niobium (Columbium)	Nb	55
Chromium	Cr	51	Nitrogen	N	18
Cobalt	Co	40	Nobelium	No	78
Copper	Cu	36	Osmium	Os	47
Curium	Cm	84	Oxygen	O	1
Dysprosium	Dy	67	Palladium	Pd	42
Einsteinium	D	81	Phosphorus	P	19
Erbium	Er	65	Platinum	Pt	45
Europium	Eu	70	Plutonium	Pu	86
Fermium	Fm	80	Polonium	Po	17
Fluorine	F	9	Potassium (Kalium)	K	74
Francium	Fr	103	Promethium	Pm	72
Gadolinium	Gd	69	Protactinium	Pa	89
Gallium	Ga	30	Praseodymium	Pr	74
Germanium	Ge	25	Radium	Ra	97
Gold	Au	38	Radon	Rn	8
Hafnium	Hf	59	Rhenium	Re	50
Helium	He	3	Rhodium	Rh	43
Hydrogen	H	2	Rubidium	Rb	101
Holmium	Ho	66	Ruthenium	Ru	44

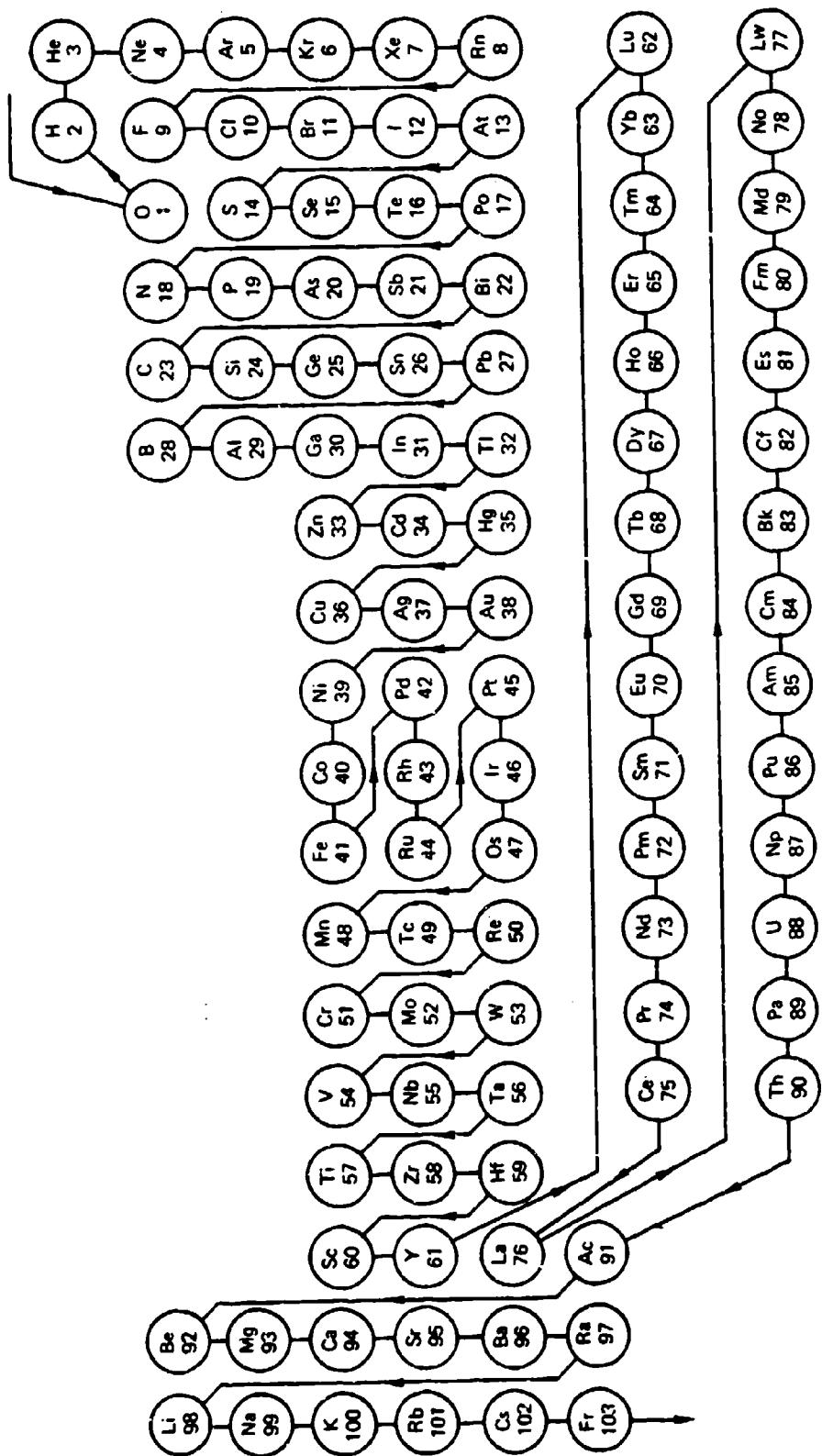


Fig. 4. Standard order of arrangement of the chemical elements.

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C	R	L	-	S	O	1	0	3	R	E	V				
A	T	A	I	N	T	R	O	A	N	O	5	S	C	T	1
A	T	A	I	N	T	R	O	A	N	O	7	U	N	C	A
A	T	A	I	N	T	R	O	A	N	7	7	7	7	7	7
1	2	3	4	5	6	7	8	9	10	11	12	13	14		

SECTION A1

ELEMENTS

1---1
OXYGEN, LIQUID

O2

WT. PERCENT

$$T_0 = 76.9 + \text{OR} - 0.2 \text{ DEG. KELVIN}$$

$$V_0 = 0.8319 + \text{OR} - 0.0006 \text{ CC/G}$$

$$C_0 = 1.01 \text{ KM/SEC}$$

IN TABLE I BELOW, DENSITY IS GIVEN IN G/CC AND VELOCITIES IN KM/SEC. MAT. DESIGNATES THE ANVIL MATERIAL FROM WHICH THE FIRST SHOCK IS REFLECTED. NO IS THE EXPERIMENT IDENTIFICATION NUMBER AND P(ST) DENOTES THE STANDARD BASE PLATE PRESSURE. IN TABLE I AND II, (1) AND (2) REPRESENT THE SINGLE AND REFLECTED-SHOCK STATES. IN TABLE II, PRESSURE IS GIVEN IN KILOBARS.

TABLE I

NO	RHO0	US1	UP1	US2	UP2	MAT	P(ST)
1A	1.202	4.60	2.05	3.42	1.51	MG	272
1B	-	-	-	4.13	1.17	AL	-
2A	-	4.60	2.07	3.73	1.17	AL	276
2B	-	-	-	4.86	0.71	CU	-
2C	-	-	-	5.23	0.49	AU	-
3A	-	4.61	2.11	2.25	1.50	MG	283
3B	-	-	-	3.31	1.17	AL	-
3C	-	-	-	4.15	0.69	CU	-
4A	-	5.64	2.62	3.31	2.00	MG	386
4B	-	-	-	4.16	1.58	AL	-
4C	-	-	-	4.25	0.92	CU	-
5A	-	5.41	2.65	4.65	2.08	MG	391
5B	-	-	-	3.97	1.59	AL	-
5C	-	-	-	4.61	0.95	CU	-
6A	-	5.48	2.65	4.31	2.07	MG	392
6B	-	-	-	5.08	1.65	AL	-
6C	-	-	-	4.95	0.66	AU	-
7A	-	5.47	2.69	4.28	0.95	CU	400
7B	-	-	-	4.68	0.66	AU	-
8A	-	5.72	2.90	4.81	1.82	AL	444
8B	-	-	-	4.81	1.08	CU	-
8C	-	-	-	4.77	0.74	AU	-
9A	-	5.89	2.95	3.34	2.29	MG	458
9B	-	-	-	4.66	1.09	CU	-
9C	-	-	-	4.81	0.75	AU	-
10A	-	5.87	2.95	2.43	2.25	MG	461
10B	-	-	-	4.29	1.83	AL	-
10C	-	-	-	3.93	1.05	CU	-
11A	-	6.70	3.61	8.24	3.05	MG	616
11B	-	-	-	4.88	1.41	CU	-
11C	-	-	-	5.16	0.99	AU	-
12A	-	6.75	3.66	4.34	2.96	MG	630
12B	-	-	-	4.98	2.40	AL	-
12C	-	-	-	4.41	1.41	CU	-
13A	-	6.89	3.66	4.56	2.37	AL	633
13B	-	-	-	4.69	1.42	CU	-
13C	-	-	-	4.44	0.97	AU	-

OXYGEN, LIQUID

NO	RHO0	US1	UP1	US2	UP2	MAT	P(ST)
14	-	6.92	3.67	3.29	2.30	AL	636
15A	-	7.50	4.13	6.23	3.47	MG	762
15B	-	-	-	3.58	2.70	AL	-
15C	-	-	-	4.68	1.65	CU	-
16A	-	7.54	4.17	13.20	3.68	MG	772
16B	-	-	-	4.45	1.65	CU	-
16C	-	-	-	5.35	1.19	AU	-
17A	-	7.49	4.22	3.91	2.76	AL	784
17B	-	-	-	4.14	1.66	CU	-
17C	-	-	-	4.11	1.15	AU	-
18A	-	8.06	4.66	5.50	3.20	AL	915
18B	-	-	-	5.07	1.94	CU	-
18C	-	-	-	5.20	1.36	AU	-
19A	-	8.09	4.67	2.21	3.78	MG	919
19B	-	-	-	5.37	1.97	CU	-
19C	-	-	-	5.68	1.39	AU	-
20A	-	8.08	4.72	3.72	3.90	MG	931
20B	-	-	-	4.77	3.20	AL	-
20C	-	-	-	5.49	2.00	CU	-

$$US1 = 1.171 + 1.788 \cdot UP1 - 0.6709 \cdot UP1^{1/2} \text{ KM/SEC}$$

$$\text{SIGMA US1} = 0.075 \text{ KM/SEC.}$$

TABLE II

NO	P1	P2	V1/V0	V2/V0
1A	113.	177.	0.554	0.500
1B	-	231.	0.554	0.475
2A	115.	229.	0.550	0.464
2B	-	321.	0.550	0.442
2C	-	367.	0.550	0.431
3A	117.	176.	0.542	0.466
3B	-	230.	0.542	0.448
3C	-	314.	0.542	0.419
4A	172.	257.	0.520	0.465
4B	-	335.	0.520	0.440
4C	-	442.	0.520	0.391
5A	172.	270.	0.510	0.470
5B	-	339.	0.510	0.428
5C	-	463.	0.510	0.391
6A	174.	268.	0.516	0.473
6B	-	354.	0.516	0.450
6C	-	526.	0.516	0.381
7A	177.	464.	0.509	0.381
7B	-	531.	0.508	0.368
8A	199.	402.	0.493	0.423
8B	-	541.	0.493	0.377
8C	-	603.	0.493	0.354
9A	209.	307.	0.499	0.446
9B	-	550.	0.499	0.377

9C	-	620.	0.499	0.358
10A	209.	300.	0.497	0.433
10B	-	405.	0.497	0.421
10C	-	528.	0.497	0.360
11A	290.	463.	0.461	0.439
11B	-	777.	0.461	0.342
11C	-	889.	0.461	0.323
12A	297.	440.	0.458	0.418
12B	-	583.	0.458	0.391
12C	-	774.	0.458	0.330
13A	303.	575.	0.469	0.395
13B	-	783.	0.469	0.343
13C	-	862.	0.469	0.313
14	305.	549.	0.470	0.377
15A	372.	555.	0.449	0.421
15B	-	687.	0.449	0.366
15C	-	957.	0.449	0.323
16A	377.	606.	0.447	0.434
16B	-	961.	0.447	0.316
16C	-	1140.	0.447	0.307
17A	380.	707.	0.436	0.358
17B	-	969.	0.436	0.303
17C	-	1084.	0.436	0.277
18A	451.	874.	0.422	0.361
18B	-	1205.	0.422	0.304
18C	-	1378.	0.422	0.281
19A	454.	628.	0.423	0.368
19B	-	1225.	0.423	0.309
19C	-	1419.	0.423	0.289
20A	458.	658.	0.416	0.375
20B	-	875.	0.416	0.349
20C	-	1261.	0.416	0.305

COMMENTS:

1) SOURCE: WACKERLE, J., SEITZ, W. L. AND JAMIESON, C. C.
SYMPOSIUM HIGH DYNAMIC PRESSURE, I.U.T.A.M., SEPT. 11-15, 1967
PARIS, FRANCE.

2) EXPERIMENTAL TECHNIQUE B. EACH SHOT CONFIGURATION HAD FOUR FLASH-GAP ASSEMBLIES INCORPORATED INTO A COMMON BASE PLATE. ONE ASSEMBLY WAS USED TO MEASURE THE SHOCK STRENGTH IN THE BASE PLATE; THE OTHERS MEASURED INDEPENDENTLY THE SINGLE- AND DOUBLE-SHOCK STATES IN OXYGEN.

DATA REDUCTION TECHNIQUE B. STANDARD MATERIALS USED WERE:

ALUMINUM 2024 ALLOY WAS USED AS THE BASE PLATE MATERIAL. THE COVER PLATES WERE AZ31B MAGNESIUM (Mg) ALLOY, 2024 ALUMINUM (Al) ALLOY, COPPER (Cu) AND GOLD (Au).

- 3) CO WAS OBTAINED FROM VAN ITTERBEK, A. AND VAN DAEL, W., ADV. CRYOGENIC ENG., VOL. 9, PAGE 207 (1963).
- 4. LIQUID NITROGEN COOLANT WAS USED IN THIS SERIES OF EXPERIMENTS.
- 5) THE AVERAGE OF THE ESTIMATED ERRORS IN US1 AND UP1 ARE 1.9 AND 1.3 PERCENT RESPECTIVELY, WHILE THE AVERAGE ERROR IN UP2 IS ABOUT 2.0 PERCENT.

TABLE I

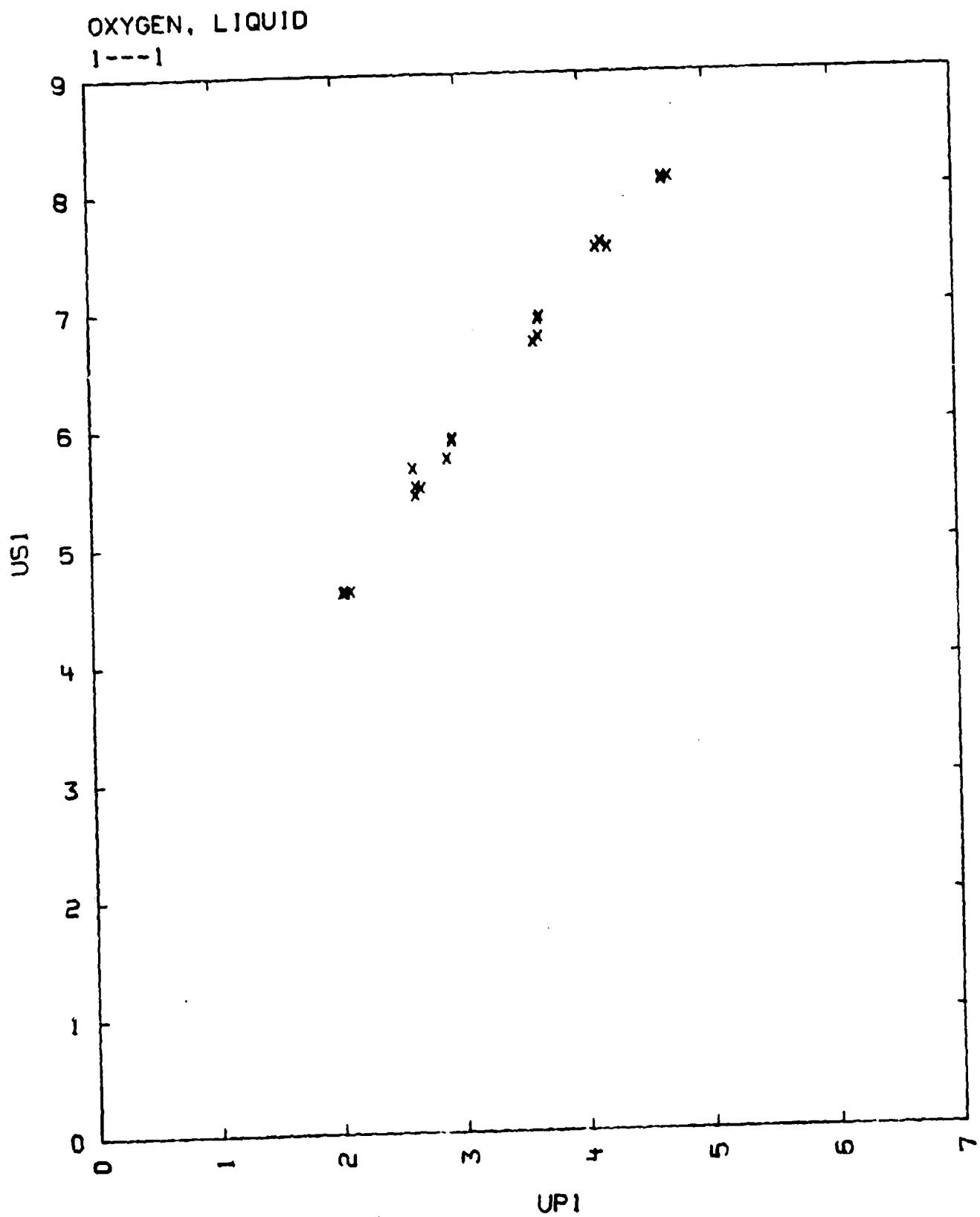
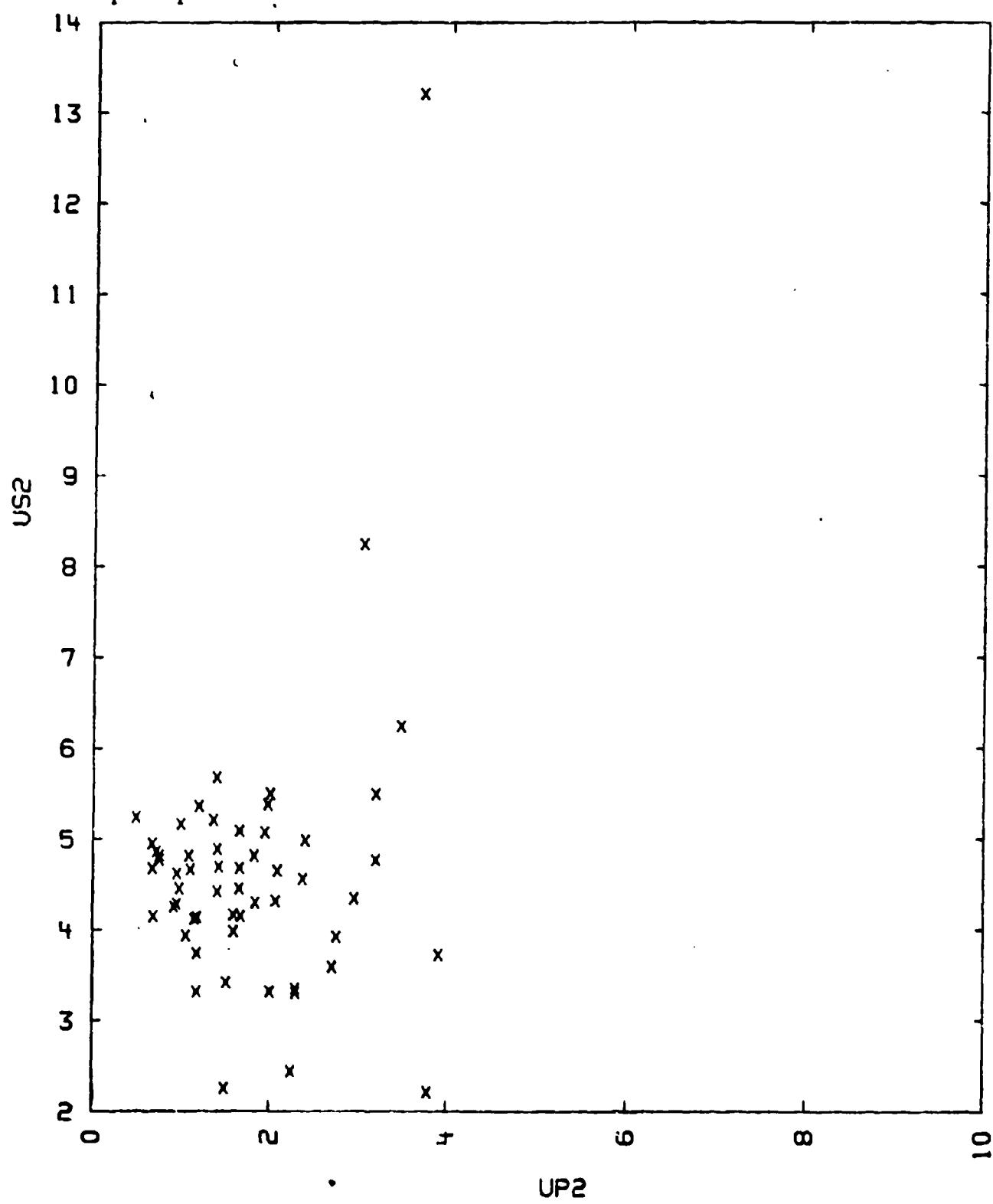


TABLE I

OXYGEN, LIQUID

1---1



5---1
ARGON

AR GREATER THAN 99.9 PERCENT
N₂ LESS THAN .1 PERCENT

$$T_0 = 86 \text{ DEGR. K.} \quad P_0 = 1.9 \text{ BARS}$$

$$V_0 = 0.7117 \text{ CC/G} \quad C_0 = 0.848 \text{ KM/SEC}$$

IN THE TABLE, DENSITY IS GIVEN IN G/CC., VELOCITIES IN MM/MICROSEC., AND PRESSURE IN KILOBARS.

TABLE

RHO0	US	UP	P	V/V0
1.408	5.44	2.52	194.	.536
1.400	3.40	1.23	58.2	.638
1.403	5.41	2.57	195.	.525
1.405	5.26	2.54	188.	.517
1.408	5.18	2.58	188.	.503
1.406	7.15	3.65	363.	.490
1.409	6.36	3.20	286.	.497
1.401	6.76	3.40	322.	.497
1.405	2.70	0.88	33.0	.673
1.401	4.98	2.32	162.	.535
1.407	1.52	0.301	6.4	.802
1.404	2.19	0.674	20.7	.692
1.406	6.44	3.30	289.	.513
1.407	5.40	2.58	196.	.480
1.405	3.34	1.34	63.0	.519
1.405	3.45	1.35	65.6	.608
1.403	3.38	1.33	63.1	.607
1.403	3.34	1.32	62.1	.603
1.402	3.34	1.33	62.1	.602

$$US = 0.84 + 2.07 UP \text{ MM/MICROSEC} \quad \text{FROM UP} = 0.1 \text{ TO } 1.0$$

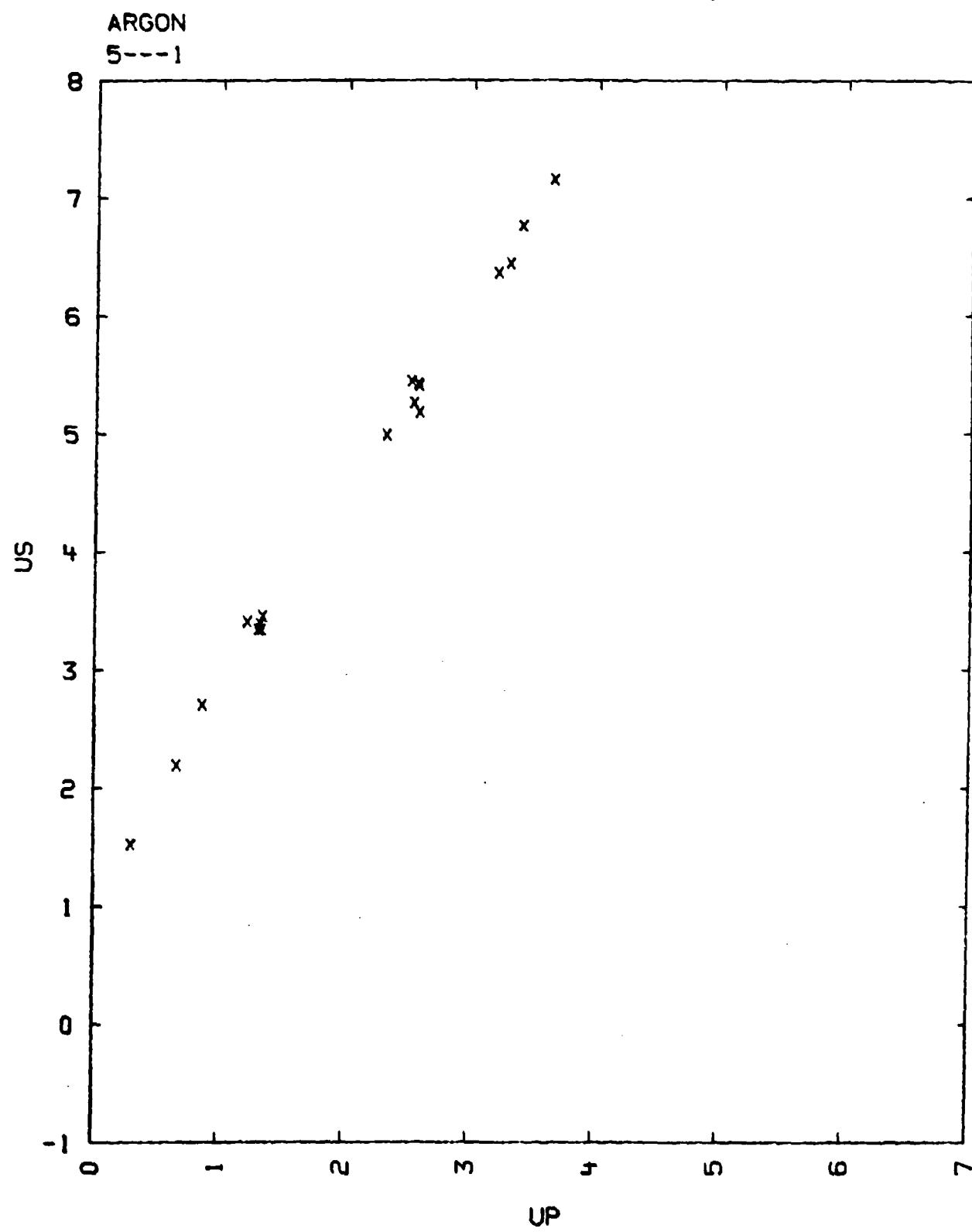
$$US = 1.19 + 1.627 UP \text{ MM/MICROSEC} \quad \text{FROM UP} = 1.0 \text{ TO } 4.0$$

COMMENTS:

- 1) SOURCE: VAN THIEL, M. AND ALDER, B.
J. CHEM. PHYS., VOL. 44, P. 1056 (1966)
LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
(STANDARD MATERIALS: 2024 AL AND AZ-31G Mg)
- 3) THE DATA STRONGLY SUGGEST THE EXISTENCE OF A TRANSITION CLOSE TO THE LIQUIDUS.
- 4) THE INITIAL SLOPE (US-C01/UP FOR SMALL VALUES OF UP WAS OBTAINED FROM SONIC MEASUREMENTS BY DOBBS AND FINEGOLD, J. ACoustICAL SOC. OF AMERICA, NO. 32, P 1215 (1960), AND IS 4.0
- 5) US IS DETERMINED TO A PRECISION OF 2 PERCENT ASSUMING LINEARITY OF

THE FITS

TABLE I



5---2

ARGON

AR GREATER THAN 99.9 PERCENT
 N2 LESS THAN 0.1 PERCENT

 $T_0 = 148.2 \text{ DEGR. K.}$ $P_0 = 70.0 \text{ BARS}$ $V_0 = 1.088 \text{ CC/G}$ $C_0 = 0.35 \text{ KM/SEC}$

IN THE TABLE DENSITY IS LISTED IN G/CC. VELOCITY IN MM/MICROSEC AND
 PRESSURE IN KILOBARS.

TABLE

RHO0	US	UP	P	V/V0
.919	3.700	1.923	65.4	.480
.919	5.424	3.188	159.	.412
.919	6.497	4.04	241.	.378
.919	3.183	1.592	46.5	.500
.919	4.934	2.876	130.3	.417
.919	6.54	4.005	240.7	.388

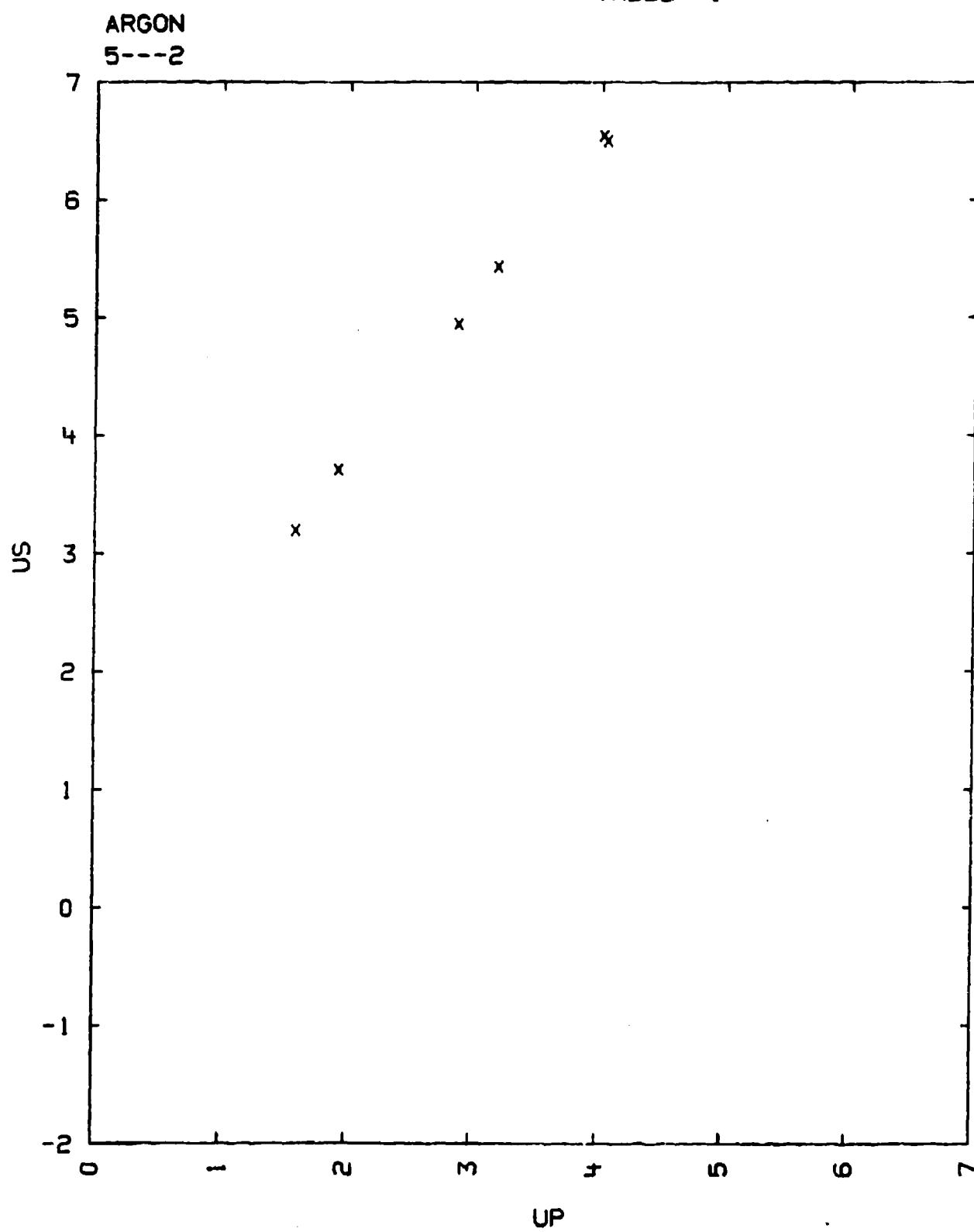
 $US = 1.036 + 1.368 UP \text{ MM/MICROSEC}$

FROM UP = 1.6 TO 4.2 MM/MICROSEC

COMMENTS:

- 1) SOURCE: VAN THIEL, M. AND ALDER, B.
J. CHEM. PHYS., VOL. 44, P. 1056 (1966)
 LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA
- 2) EXPERIMENTAL TECHNIQUE A
 DATA REDUCTION TECHNIQUE B
 (STANDARD MATERIAL AZ-31B MG)
- 3) THE INITIAL SLOPE $(C_0 - US)/UP$ FOR SMALL UP WAS CALCULATED FROM P.V.T.
 DATA PUBLISHED BY MICHELS ET AL., *PHYSICA* NO. 24, P 659 FF (1958).
 THIS YIELDS A SLOPE OF 2.21 AT THE INITIAL WAVE VELOCITY $US_0 = C_0 = .35$
 MM/MICROSEC.

TABLE I



5--3
ARGON

AR

$T_0 = 288-301 \text{ DEG. K.}$
 $V_0 = 750.6-801.9 \text{ CC/G.}$

$C_D = 0.330 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURES IN KILOBARS, DENSITY IN G/CC. AND TEMPERATURE IN DEG. KELVIN.

TABLE

RHO0	T0	P0	US	UP	P	V/V0
0.001307	294	0.0008002	3.877	3.030	0.1543	0.218
0.001249	301	0.0007830	4.076	3.291	0.1683	0.192
0.001328	297	0.0008211	5.290	4.461	0.3142	0.157
0.001248	301	0.0007822	5.365	4.461	0.2995	0.168
0.001307	294	0.0008000	6.230	5.400	0.4405	0.133
0.001298	296	0.0008000	6.550	5.776	0.4922	0.119
0.001332	288	0.0007988	8.260	7.380	0.8128	0.107
0.001247	301	0.0007815	8.700	7.810	0.8481	0.102

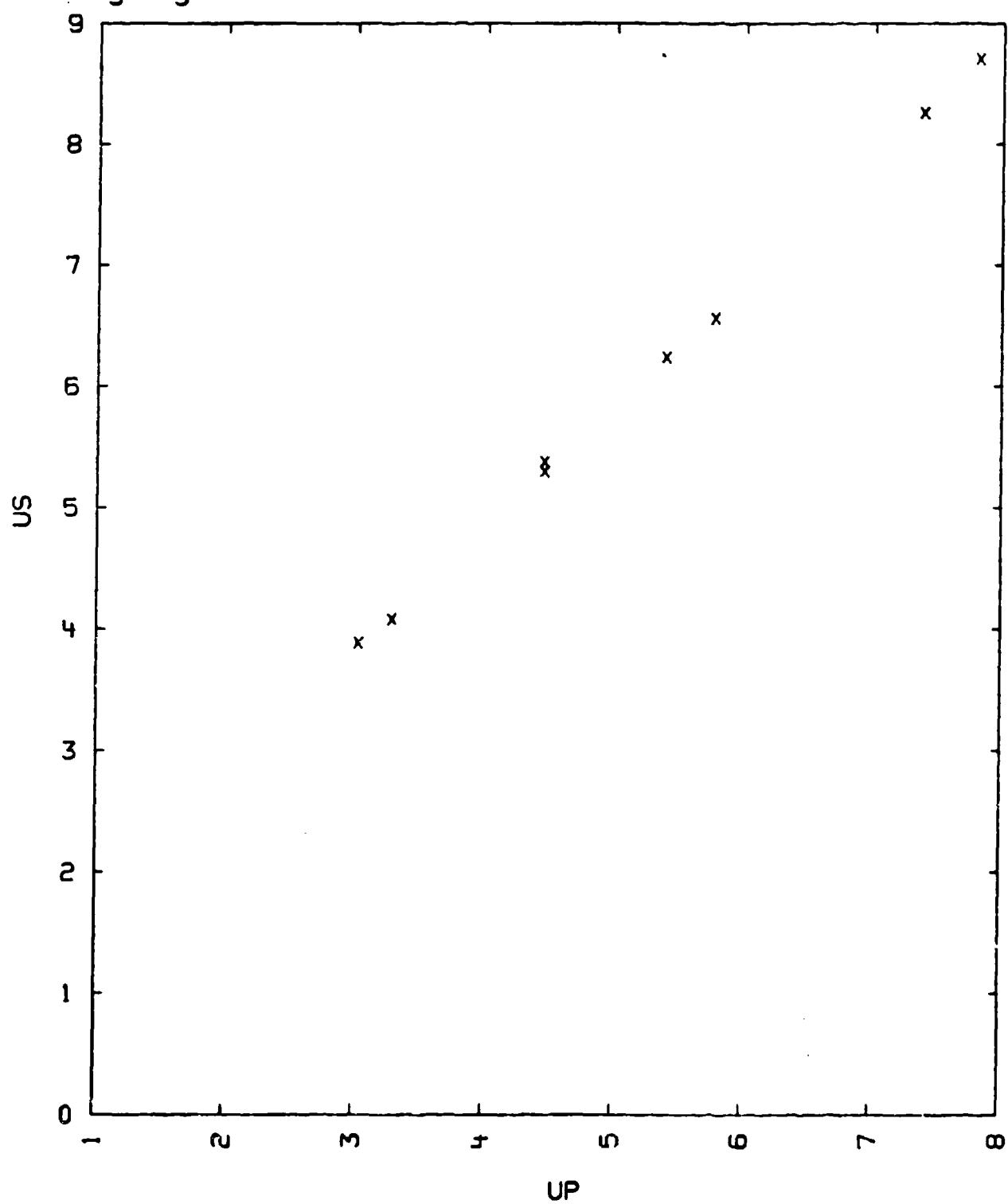
US =

COMMENTS:

- 1) SOURCE: CHRISTIAN, R. H. AND YARGER, F. L.
JOURNAL OF CHEMICAL PHYS., VOL. 23, P. 2042 (1955)
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE EXPERIMENTAL ERROR IN THE US MEASUREMENTS IS 0.5 PERCENT.
- 4) IN THE ABOVE TABLE, RHO0 WAS RECOMPUTED FROM THE MEASURED VALUES OF T0 AND P0 USING THE PERFECT GAS LAW. MEASURED P AND V/V0 WERE ALSO RECALCULATED USING THE CONSERVATION CONDITIONS.

TABLE I

ARGON
5---3



3---4
ARGON

AR

$T_0 = 298.3-307.2$ DEG. KELVIN
 $V_0 = 0.879-1.030$ CC/G.

IN THE TABLE BELOW, DENSITIES ARE GIVEN IN G/CC., TEMPERATURE IN DEG. KELVIN, VELOCITIES IN KM/SEC., AND PRESSURES IN KILOBARS.

TABLE

RHO0	T0	P0	US	UP	P	V/V0
0.971	302.2	1.034	2.87	1.42	40	0.546
0.994	299.0	1.074	3.85	1.86	71	0.518
0.972	303.0	1.040	4.72	2.53	117	0.465
0.988	302.7	1.081	4.84	2.66	127	0.460
0.986	307.7	1.110	6.873	4.10	276	0.411
1.138	305.4	1.662	4.968	2.52	144	0.493
1.135	298.3	1.586	6.779	3.97	307	0.414

$$US = 0.32 + 1.25 \cdot (RHO0 - 1.) + 2.03 \cdot UP - 0.11 \cdot UP^2. \text{ KM/SEC}$$

SIG.US = 0.12 KM/SEC

COMMENTS:

- 1) SOURCE: WACKERLE, J.
(PRIVATE COMMUNICATION) LOS ALAMOS SCIENTIFIC LAB.,
P.O. BOX 1663, LOS ALAMOS, NEW MEXICO.
- 2) EXPERIMENTAL TECHNIQUE A AND B.
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE MEASURED VALUES FOR THE BULK SOUND VELOCITIES (C0) ARE GIVEN IN THE TABLE BELOW.

TABLE

RHO0	C0
0.971	0.75
0.994	0.77
0.972	0.775
0.988	0.772
0.986	0.775
1.138	0.91
1.135	0.91

- 4) THE AVERAGE EXPERIMENTAL ERROR IN THE US MEASUREMENTS DUE TO PROBE SCATTERING IS 1.5 PERCENT, THE LARGEST ERROR IS 4 PERCENT.

5--5

ARGON

AR

$$T_0 = 86 \text{ DEG K}$$

$$V_0 = 0.714 \text{ G/CC}$$

THE TABLE LISTS DENSITY IN G/CC, VELOCITIES IN KM/SEC AND PRESSURES IN KBAR. 1 AND 2 REFER TO SHOCKS INCIDENT ON AND REFLECTED FROM A HIGHER IMPEDANCE ANVIL. MAT = ANVIL MATERIAL.

TABLE

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0	MAT
1.40	6.052	3.075	261.	0.492	-3.94	2.034	469.	0.419	AL
-	-	-	-	-	-4.26	1.226	647.	0.368	CU
-	-	-	-	-	-4.67	0.843	753.	0.350	AU
-	6.069	3.093	263.	0.490	-2.95	1.986	454.	0.400	AL
-	-	-	-	-	-3.89	1.213	638.	0.358	CU
-	-	-	-	-	-4.64	0.873	753.	0.350	AU

US=

COMMENTS

- 1) SOURCE: WACKERLE, J.
PRIVATE COMMUNICATION. GMX-7, LOS ALAMOS SCIENTIFIC LAB.
BOX 1663, LOS ALAMOS, NEW MEXICO, 87544
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B; STANDARDS: 2024 AL FOR THE BASE PLATE
AND 2024 AL, CU AND AU FOR THE anvils
USED TO REFLECT THE SHOCK.
- 3) DETAILS OF THE TECHNIQUE ARE DESCRIBED BY: WALSH J. M. AND RICE M. H.
J. CHEM. PHYS., VOL. 26, P. 815, (1957)
- 4) PRECISION: DEFINE FSIG.A = (SIG.A)/A

BASE PLATE=AL	ANVIL = AL	CU	AU
FSIG.US1=0.02	FSIG.UP2= 0.023	0.015	0.015
FSIG.UP1=0.013	FSIG.P2 = 0.027	0.02	0.018
FSIG.P1 = 0.022	FSIG.(V2-V0)=0.053	0.04	0.04
FSIG.(V1-V0)=0.03			

TABLE I

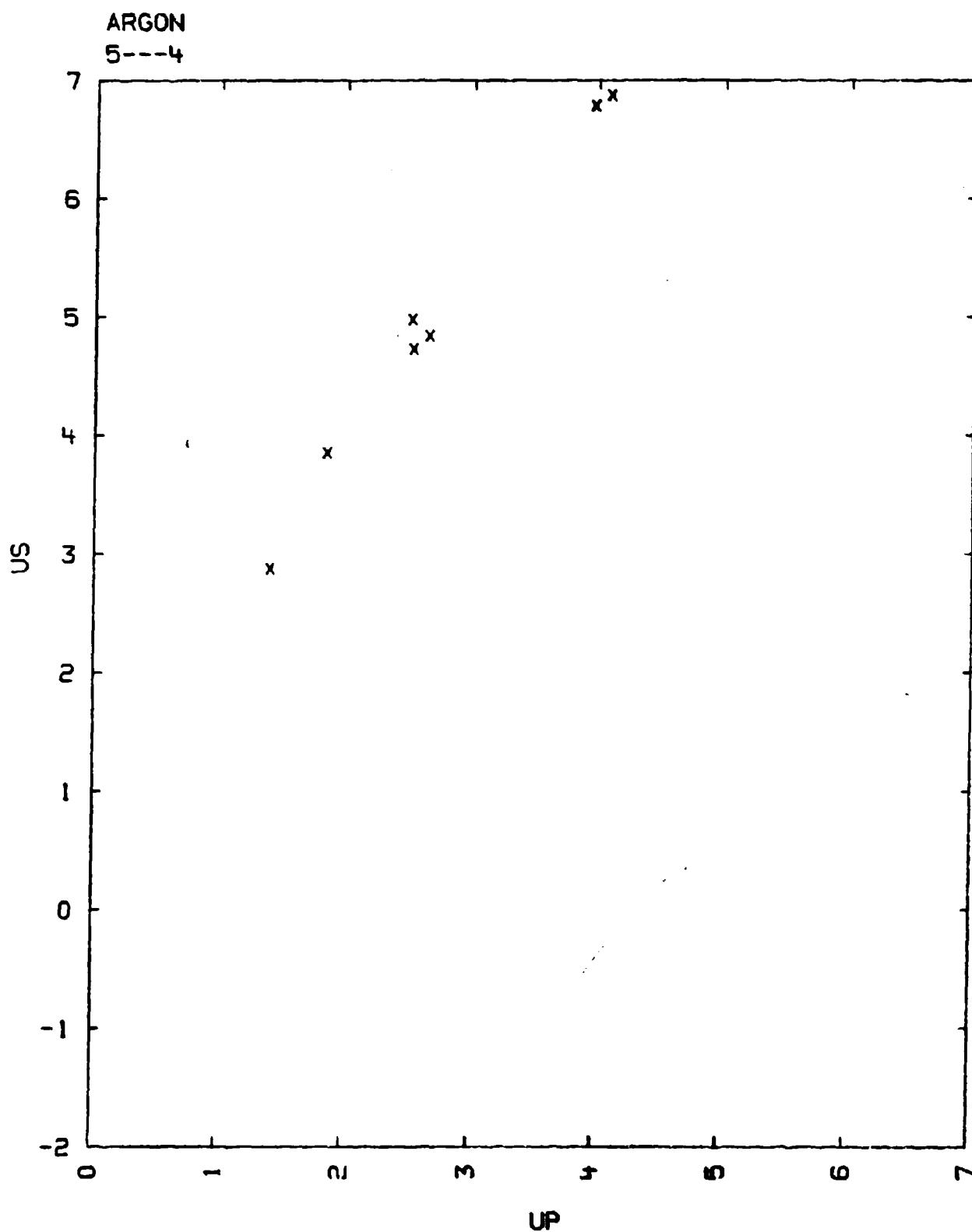


TABLE I

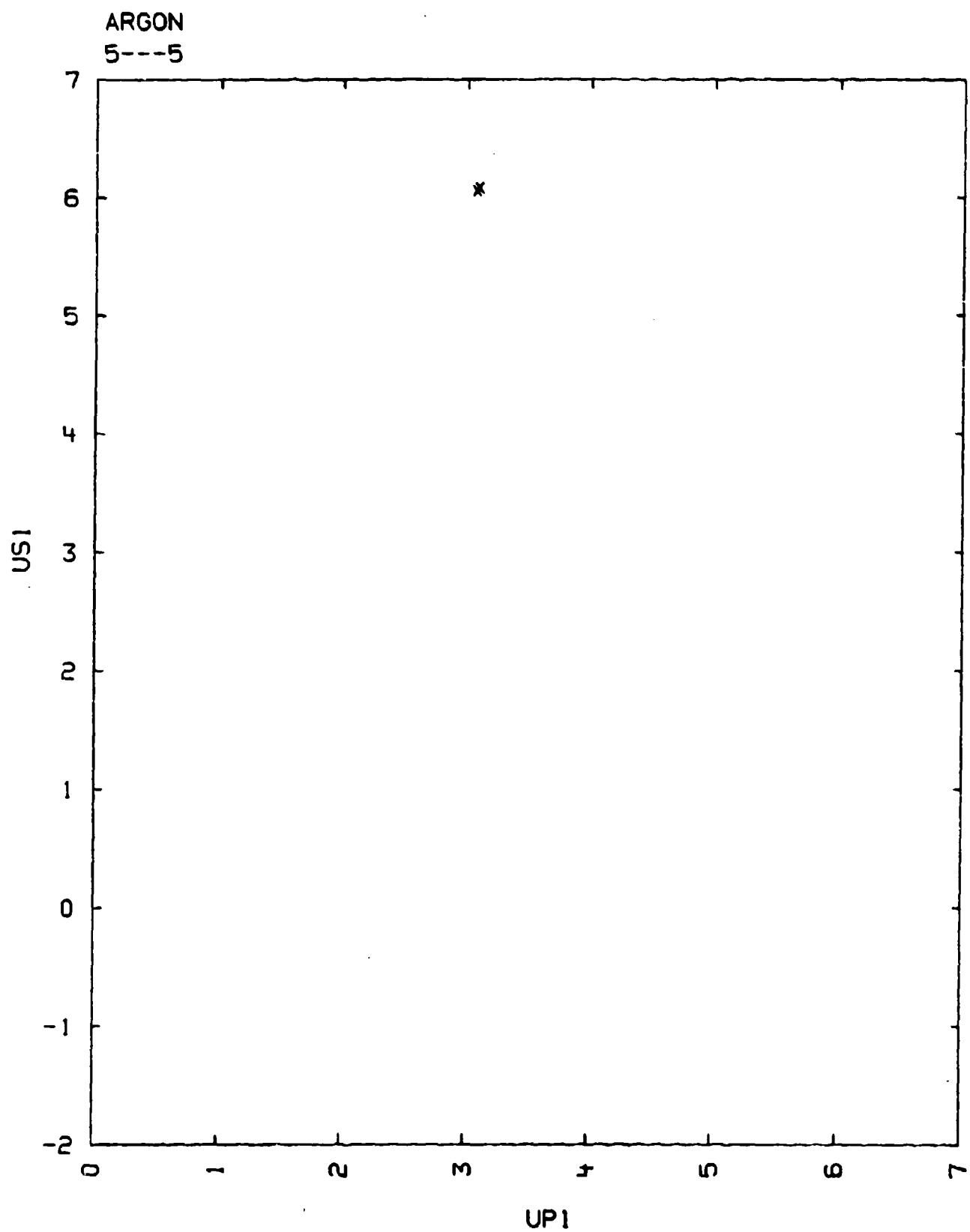
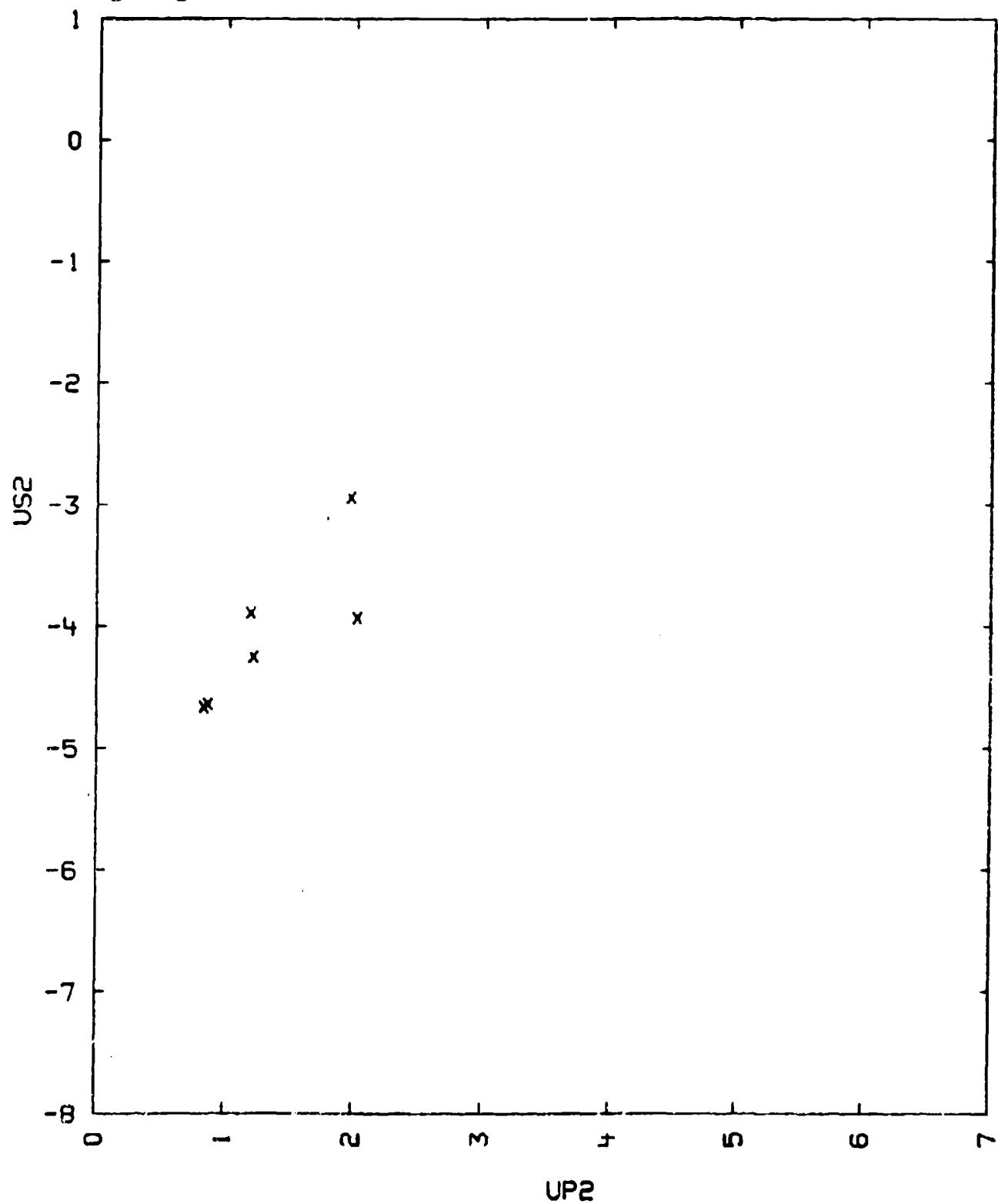


TABLE I

ARGON
5---5



7---1

XENON

XE 99.9 PERCENT

$T_0 = 165.2$ DEG. K. $P_0 = 2.24$ ATM
 $V_0 = 0.3231$ CC/G

IN THE TABLE BELOW DENSITY IS GIVEN IN G/CC., VELOCITIES IN KM/SEC AND PRESSURE IN KBARS

TABLE

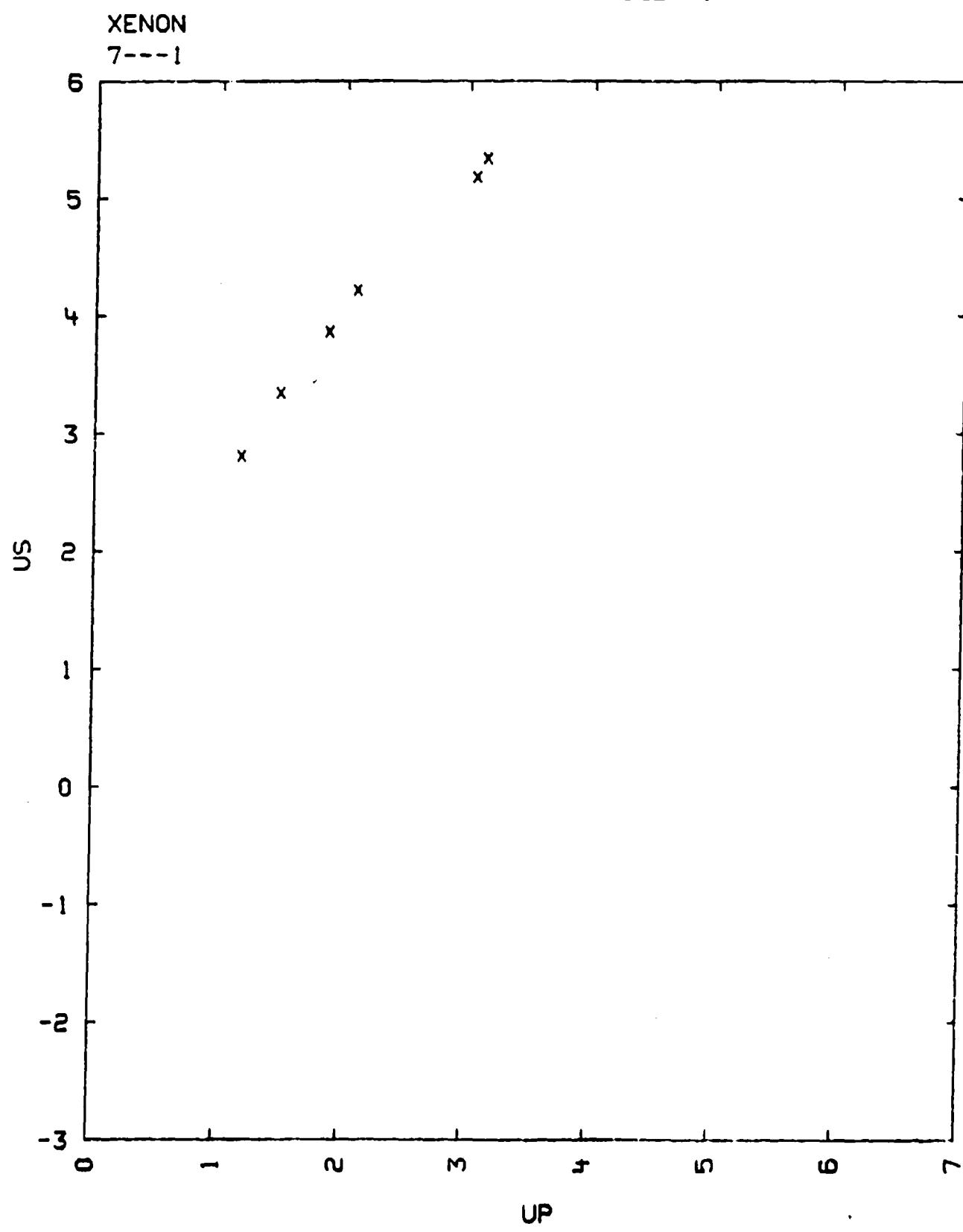
SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	US	UFS
3.095	2.803	1.185	103	0.578	1.787	
3.095	3.334	1.482	153	0.556	6.968	2.329
3.095	3.856	1.87	224	0.510	7.45	3.06
3.095	4.206	2.10	268	0.501	7.654	3.423
3.095	5.177	3.055	487	0.410		5.38
3.095	5.335	3.14	511	0.411	8.912	5.527

US =

COMMENTS:

- 1) SOURCE: KEELER, R. N.
 PRIVATE COMMUNICATION.
 LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE: A.
 DATA REDUCTION METHOD : B. STANDARD MATERIAL AL 2024.
- 3) THE SHOCK VELOCITY OF THE STANDARD IN THE FIRST AND FIFTH TABLE ENTRY WAS RELATIVELY INACCURATE AND THEREFORE DISREGARDED IN THE DETERMINATION OF THE ALUMINUM PRESSURE. THE SHOCK AND FREE SURFACE VELOCITY OF THE STANDARD WERE CORRECTED IN THE THIRD TABLE ENTRY (BY -1 AND +1 PERCENT RESPECTIVELY) TO YIELD AGREEMENT WITH THE KNOWN AVERAGE AL. HUGONIOT.
- 4) A PRELIMINARY DISCUSSION OF THE SIGNIFICANCE OF THIS DATA HAS BEEN GIVEN: R. N. KEELER, M. VAN THIEL AND B. J. ALDER
 PHYSICA, VOL. 31, P. 1437 (1965).

TABLE I



12---1
100INEV0 = 0.2041 CC/G
V01 = 0.2019 CC/GIN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC.
AND PRESSURE IN KILOBARS.

TABLE

SAMPLE						BASE PLATE
RHO0	US	UFS	UP	P	V/V0	PRESSURE
4.902	3.78	3.30	1.51	280	0.601	295
4.905	3.74	3.27	1.51	278	0.595	295
4.923	3.40	2.69	1.23	206	0.638	224
4.928	3.34	2.71	1.24	204	0.629	224
4.901	3.04	2.21	1.00	150	0.671	169
4.903	3.02	2.22	1.01	149	0.666	169
4.924	2.98		1.00	147	0.665	168
4.916	2.46	1.48	0.68	82	0.724	101
4.926	2.47	1.48	0.67	82	0.726	101
4.915	3.03	2.26	1.05	156	0.656	177
4.919	3.01	2.22	1.05	155	0.651	177
4.905	3.43	2.75	1.26	212	0.633	230
4.884	5.39	6.11	2.45	644	0.545	616
4.878	4.67	4.79	2.10	478	0.551	479
4.861	4.51	4.28	1.93	423	0.572	423
4.869	2.74	1.98	0.90	120	0.674	138
4.939	4.03	3.90	1.78	354	0.558	369
4.888	5.22	5.74	2.42	617	0.536	596
4.884	4.04	3.70	1.67	330	0.587	343
4.905	3.91	3.59	1.55	298	0.603	311
4.909	3.68	3.10	1.51	273	0.590	292
4.901	3.70	3.17	1.51	273	0.592	292
4.915	5.53	6.21	2.66	723	0.519	688
4.895	5.55	6.28	2.65	720	0.523	688
4.888	5.92	7.10	3.17	917	0.465	888
4.918	5.92	7.09	3.18	926	0.463	888
4.896	6.15	7.50	3.45	1039	0.439	1000
4.878	6.19	7.49	3.43	1036	0.446	993
4.899	4.71	4.96	2.13	491	0.548	488
4.896	4.69	4.72	2.13	489	0.546	488
4.873	5.15		2.48	617	0.522	605
4.844	5.19		2.48	618	0.526	605
4.875	2.94	2.25	1.03	147	0.651	171
4.891	2.90	2.23	1.04	147	0.642	171
4.902	5.16		2.39	604	0.537	584
4.910	5.21		2.37	608	0.545	587
4.890	2.24	1.02	0.49	54	0.780	69
4.879	2.23	1.00	0.50	54	0.777	69
4.916	3.44		1.38	233	0.599	256

IODINE

RHO0	US	UFS	UP	P	V/V0	PRESSURE
4.917	3.47		1.37	234	0.605	256
4.921	3.48		1.37	235	0.606	256
4.871	3.67		1.48	265	0.597	284
4.883	3.60		1.49	262	0.586	284
4.903	3.84		1.58	298	0.589	314
4.916	3.80		1.59	297	0.582	314
4.924	3.82		1.59	298	0.584	314

US = 1.38 + 1.56 UP KM/SEC FOR UP BETWEEN 0.5 AND 2.5 KM/SEC
 SIGMA US = 0.073 KM/SEC

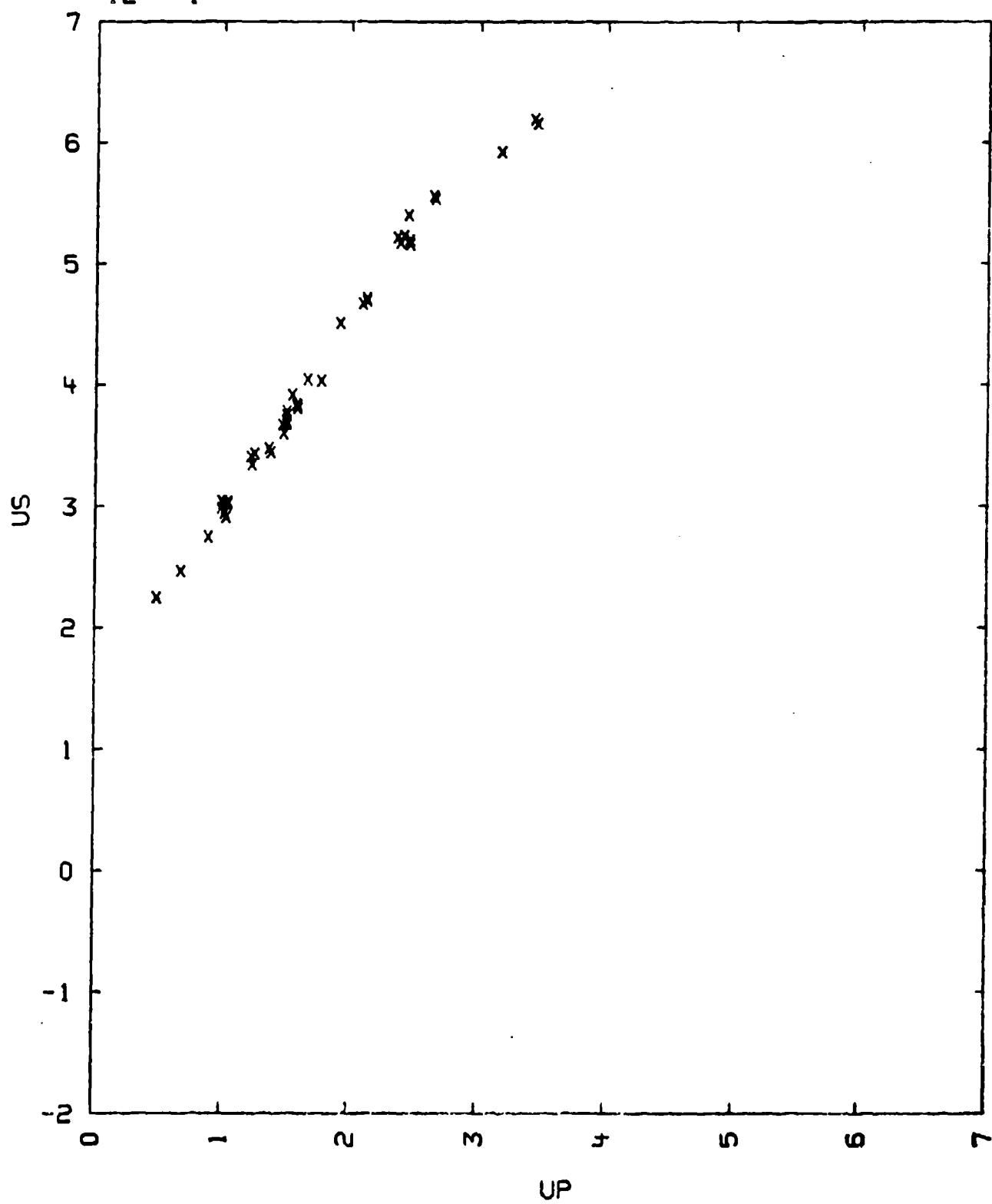
US = 3.43 + 0.79 UP KM/SEC FOR UP BETWEEN 2.5 AND 3.5 KM/SEC
 SIGMA US = 0.031 KM/SEC

COMMENTS:

- 1) SOURCE: COMPILER
 L. R. L. EQUATION OF STATE FILE
 LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B. (ALUMINUM STANDARD BASE PLATE).
 DATA REDUCTION TECHNIQUE B.
- 3) A PHASE TRANSITION IS OBSERVED AT APPROXIMATELY 644 KBAR AND A
 RELATIVE VOLUME OF 0.545
- 3) AN EARLIER ANALYSIS AND DISCUSSION OF THIS DATA WAS REPORTED BY
 B. J. ALDER AND R. H. CHRISTIAN, PHYS. REV. LETTERS VOL. 4, P. 450
 (1960).
- 4) THE VALUE OF V01 WAS OBTAINED FROM A.C.A. MONOGRAPH NO.5,
 1963) 2ND EDITION.
 CRYSTAL DATA DETERMINATIVE TABLES (AMERICAN CRYSTALLOGRAPHIC ASSOC.)

TABLE I

IODINE
12---1



18---1
NITROGEN LIQUID

N₂

T₀ = 77.4 DEGREES K.

V₀ = 1.24 CC/G.

C₀ = 0.849 KM/SEC.

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	UP STANDARD
0.808	3.14	1.17	29.6	0.625	0.69
-	3.74	1.57	47.4	0.580	0.845
-	5.00	2.50	101	0.498	1.55
-	7.45	4.27	257	0.424	2.80
-	9.05	5.52	404.	0.390	3.68

US = 1.588 + 1.360 UP

SIGMA = 0.08 PERCENT

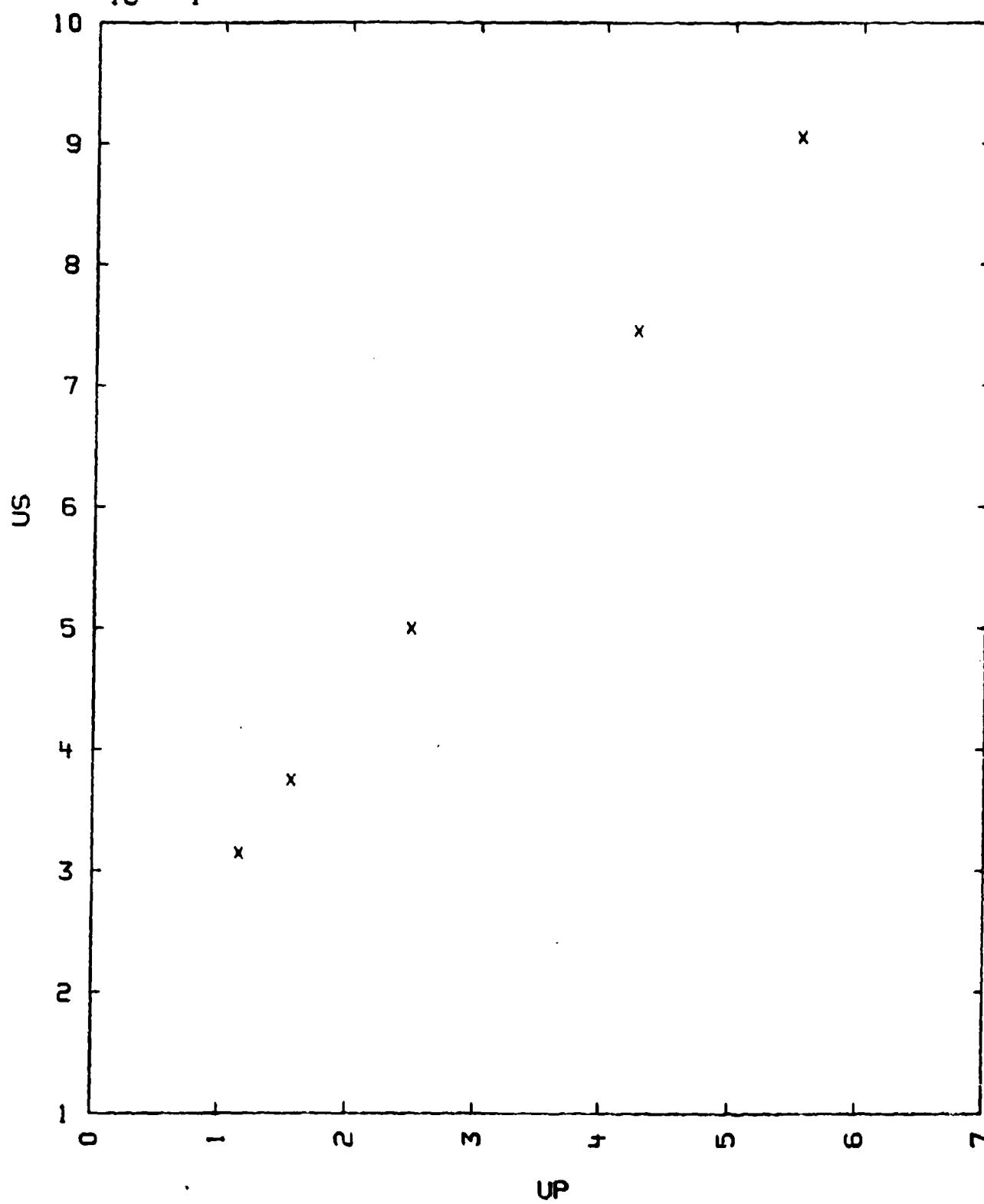
COMMENTS:

- 1) SOURCE: ZUBAREV, V.N. AND TELEGIN, G.S.
SOVIET PHYS.-DOKLADY, VOL. 7, P. 34 (1962)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
STANDARD MATERIALS ARE ALUMINUM AND COPPER
- 3) EACH POINT IS AN AVERAGE OF 4 TO 12 MEASUREMENTS. THE SQUARE
DEVIATION FROM THE MEAN IS 1 TO 2 PERCENT.
- 4) C₀ IS OBTAINED FROM, DOBBS E. R. AND FINEGOLD L.
THE JOURNAL OF THE ACOUSTICAL SOCIETY OF AMERICA, VOL. 32, NO. 10,
P. 1215, OCTOBER, 1960.

TABLE I

NITROGEN LIQUID

18---1



18---2
NITROGEN, LIQUID

$N_2 = N-N$

$T_0 = 77$ DEG K
 $V_0 = 1.238$ CC/G

$C_0 = 0.857$ KM/SEC.

IN THE TABLE BELOW VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS, DENSITY IN G/CC. AND TEMPERATURE IN DEGREES KELVIN.

TABLE

T_0	ρ_{00}	U_S	U_P	P	V/V_0
77	0.808	3.79	1.72	52.6	0.546
-	-	3.81	1.44	44.4	0.622
-	-	2.97	1.10	26.4	0.630
-	-	4.11	1.74	71.5	0.577
-	-	4.92	2.32	92.4	0.528
-	-	2.20	0.64	11.4	0.710
-	-	5.68	2.74	126.0	0.518
-	-	5.66	2.70	123.0	0.523
-	-	6.47	3.26	170.0	0.496

$U_S = 1.219 + 1.622 \cdot U_P$ KM/SEC.

SIGMA $U_S = 0.13$ KM/SEC.

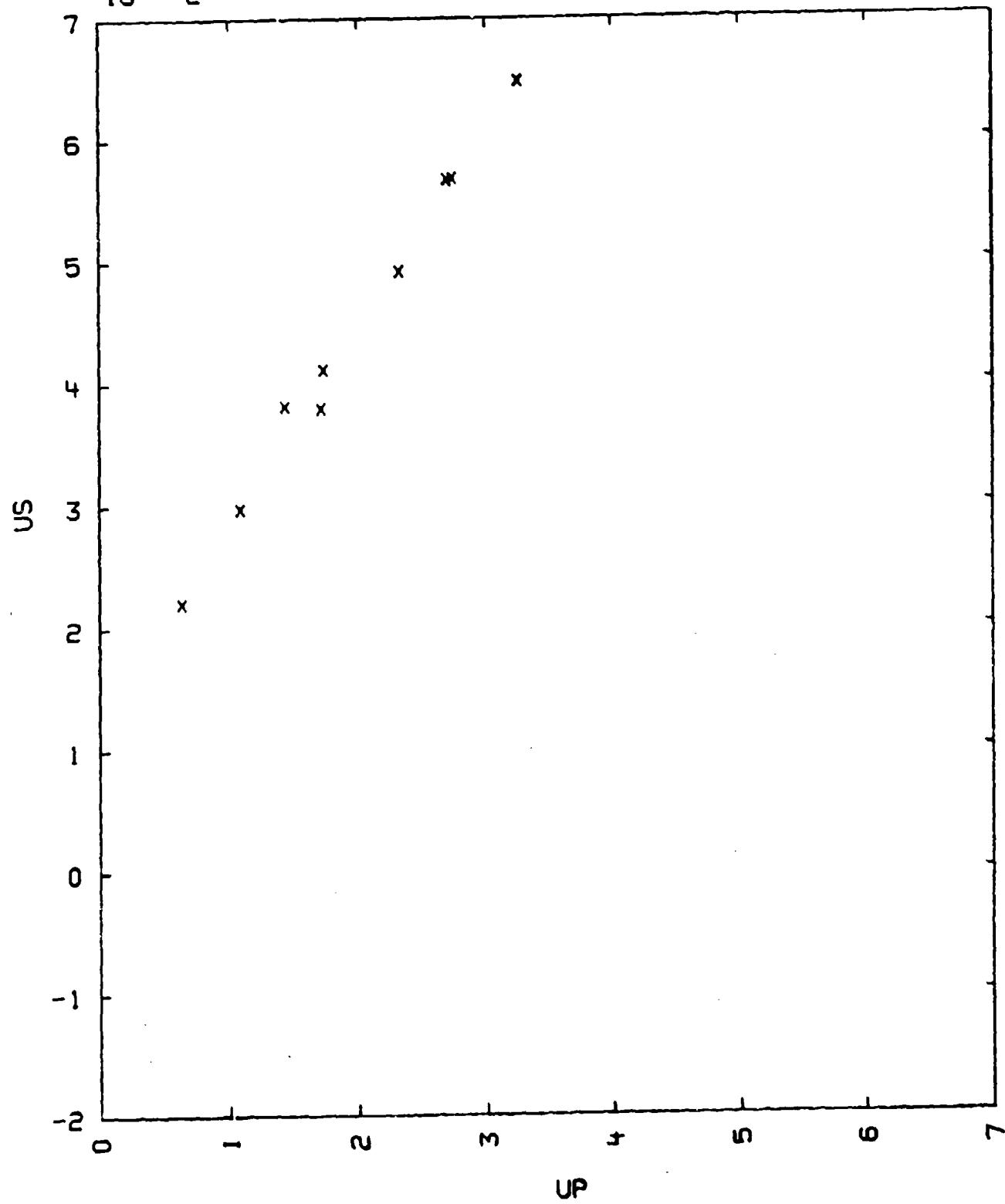
COMMENTS:

- 1) SOURCE: DICK R. D. AND SCHOENHERR G.
PRIVATE COMMUNICATION
GMX-7 LOS ALAMOS SCIENTIFIC LABORATORY, LOS ALAMOS, N. M.
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE :
IN THESE EXPERIMENTS THE VELOCITY OF THE INTERFACE BETWEEN THE SAMPLE AND THE FLAT BOTTOM OF THE METAL CONTAINER WAS MADE DIRECTLY. A CORRECTION WAS MADE FOR THE EXPECTED MOTION OF THE PROBE. THE METHOD APPEARED TO YIELD RESULTS COMPARABLE TO THE MORE USUAL DATA REDUCTION METHOD B WITH AN UNCERTAINTY OF +OR- 3 PERCENT. MAX ERROR = 6 PERCENT.
- 3) THE INITIAL SOUND SPEED C_0 WAS OBTAINED FROM: VAN ITTERBEEK A. AND VAN DAEL W., PHYSICA, VOL. 28, PAGE 861 (1962)

TABLE I

NITROGEN, LIQUID

18---2



18---3
NITROGEN

N₂ 99. WT. PERCENT
O₂ 1. - -

T₀ = 75 DEG. K.
V₀ = 1.220 CC/G.

C₀ = 0.88 KM/SEC.

THE TABLE LISTS DENSITY IN G/CC., VELOCITIES IN KM/SEC. AND PRESSURE IN KBAR. DUS AND DUP ARE THE UNCERTAINTIES IN US AND UP.

TABLE

RH00	SAMPLE					P	V/V ₀	- STANDARD -	
	US	DUS	UP	DUP	US(ST)			US(ST)	DUS(ST)
0.820	2.58	0.02	0.75	0.10	16.	0.708	6.07	0.07	
0.820	2.97	0.02	0.97	0.05	24.	0.674	6.24	0.04	
0.820	3.53	0.01	1.39	0.04	40.	0.606	6.58	0.03	
0.820	4.19	0.02	1.85	0.03	63.	0.558	6.97	0.03	
0.820	5.09	0.02	2.44	0.02	102.	0.521	7.48	0.02	
0.820	5.50	0.02	2.69	0.02	121.	0.512	7.71	0.02	
0.820	5.92	0.03	2.90	0.12	141.	0.510	7.91	0.01	
0.820	6.66	0.03	3.54	0.07	193.	0.469	8.48	0.05	
0.820	6.98	0.08	3.64	0.09	208.	0.479	8.58	0.07	
0.820	7.34	0.04	3.98	0.08	239.	0.458	8.89	0.06	
0.820	7.52	0.05	4.26	0.07	263.	0.434	9.13	0.06	
0.820	7.59	0.03	4.32	0.12	269.	0.431	9.19	0.10	
0.820	7.43	0.03	4.38	0.08	267.	0.410	9.22	0.06	
0.820	7.73	0.03	4.50	0.10	285.	0.418	9.35	0.08	
0.820	8.17	0.06	4.55	0.09	305.	0.443	9.43	0.07	
0.820	7.92	0.05	4.60	0.11	299.	0.420	9.44	0.09	
0.820	8.36	0.06	4.69	0.08	321.	0.440	9.56	0.06	
0.820	8.51	0.08	4.86	0.05	339.	0.429	9.71	0.04	
0.820	8.48	0.03	5.04	0.11	350.	0.406	9.85	0.09	
0.820	8.82	0.04	5.20	0.10	376.	0.411	10.02	0.08	
0.820	8.92	0.06	5.30	0.12	388.	0.406	10.11	0.10	

US = A + B*UP, WITH A = 1.49 KM/SEC. B = 1.49
SIG.A = 0.06 KM/SEC. SIG.B = 0.02

FOR US BETWEEN 2.3 AND 7.4 KM/SEC.,

A = -1.0 KM/SEC. B = 2.0
SIG.A = 2.0 KM/SEC. SIG.B = 0.4

FOR US BETWEEN 7.4 AND 8.4 KM/SEC. AND

A = 4.06 KM/SEC B = 0.92
SIG.A = 0.05 KM/SEC. SIG.B = 0.01

FOR US BETWEEN 8.4 AND 9.0 KM/SEC.

COMMENTS:

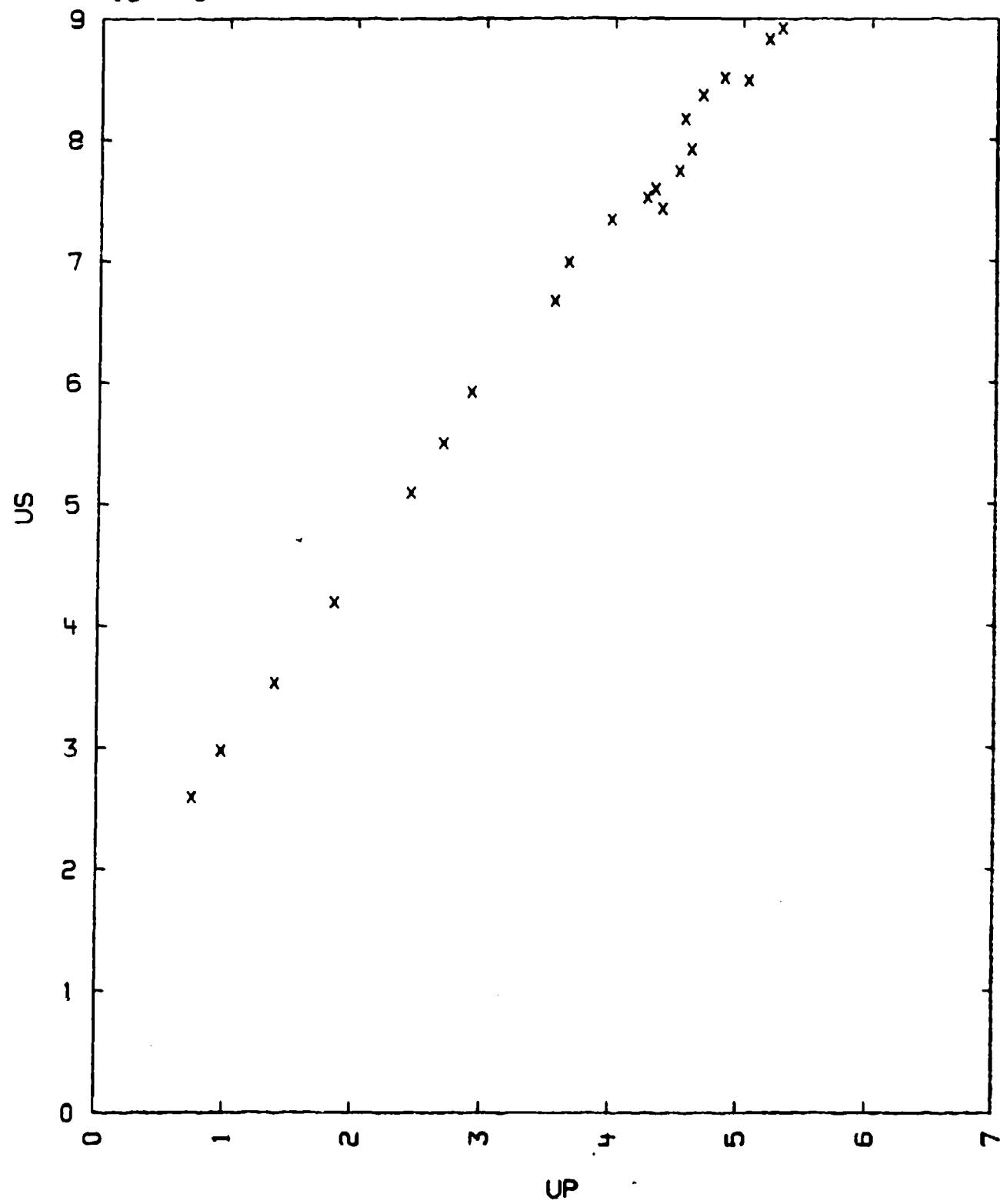
- 1) SOURCE: DICK, R.D.
LOS ALAMOS SCIENTIFIC LAB. (THESIS)
LOS ALAMOS SCIENTIFIC LAB.,

LOS ALAMOS, BOX 1663, NEW MEXICO 87544

- 2) EXPERIMENTAL TECHNIQUE: A
DATA REDUCTION METHOD: B STANDARD MATERIAL 2024 AL ALLOY WITH
 $US = 5.460 + 1.318 \cdot UP$ $\rho_{H0} = 2.785G/CC$
AND GRUNEISEN GAMMA = 2.22
- 3) THE SOUND VELOCITY WAS TAKEN FROM: A. VAN ITTERBEEK AND H. VAN DAEL
PHYSICA, VOL. 29, P. 861 (1962)

TABLE I

NITROGEN
18---3



21---1
ANTIMONY

SB

$$V_0 = 0.1493 \text{ CC/G.}$$

$$V_{01} = 0.1485 \text{ CC/G.}$$

$$C_0 = 2.37 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC

TABLE

SAMPLE					STANDARD	
RHO0	US	UP	P	VIVO	P	MTRL
6.70	3.61	1.03	248	0.716	220	AL
-	3.59	1.03	248	0.713	220	-
-	3.63	1.02	249	0.720	220	-
-	4.30	1.39	400	0.678	339	-
-	4.33	1.38	401	0.681	339	-
-	5.12	1.96	673	0.617	869	BRASS
-	5.06	1.88	637	0.629	819	-
-	5.64	2.19	828	0.611	1040	-
-	5.72	2.19	838	0.618	1042	-
-	5.71	2.20	843	0.614	1051	-
-	6.31	2.70	1142	0.572	1422	-
-	6.34	2.73	1158	0.569	1442	-
-	6.43	2.73	1178	0.576	1452	-

$$US = 1.989 + 1.634 UP \quad \text{SIGMA US} = 0.2 \text{ PERCENT}$$

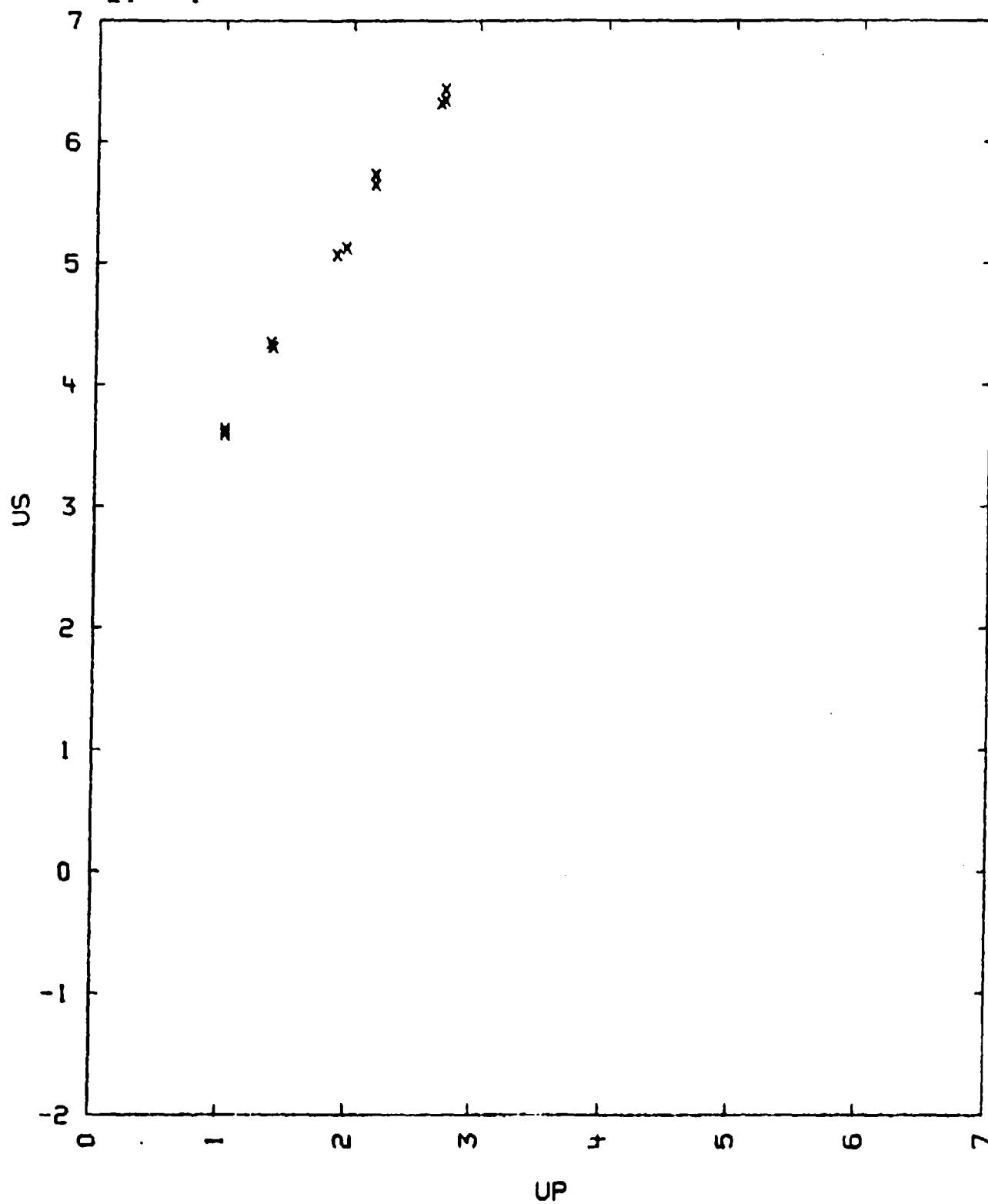
COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIALS: BRASS SEE MTRL. NO. 36--33
2024 ALUMINUM ALLOY SEE MTRL. NO. 29
- 3) C0 CALCULATED FROM THE BULK MODULUS IN
C. KITTEL: INTRODUCTION TO SOLID STATE PHYSICS.
(JOHN WILEY AND SONS INC. N.Y.) 2ND ED. 1956
- 4) V01 OBTAINED FROM: AM. INST. OF PHYS. HANDBOOK (MCGRAW HILL-BOOK CO.,
N.Y., 1963) 2ND ED.
- 5) AN ESTIMATE OF THE GRUNEISEN GAMMA HAS BEEN MADE BY: D.B. SIRDESMUKH
AND KAMASHWARA RAO. J. APPLIED PHYSICS, V. 44, P. 894, (1973)

TABLE I

ANTIMONY

21---1



22---0
BISMUTH, SUMMARY

B1

$$\begin{array}{lll} V_0 = 0.1016 - 0.1021 \text{ CC/G} & C_L = 2.18 \text{ KM/SEC} & C_0 = 1.771 \text{ KM/SEC} \\ V_{01} = 0.10197 & \text{CC/G} & C_S = 1.10 \text{ KM/SEC} \\ & & C_B = 1.86 \text{ KM/SEC} \end{array}$$

THE TABLE LISTS POINTS CALCULATED FROM THE US VS. UP FITS GIVEN BELOW.
DENSITY IS IN G/CC, VELOCITIES IN KM/SEC, PRESSURE IN KBAR AND E-E0 IS
INTERNAL ENERGY CHANGE IN KBAR CC/G. TP = TRANSITION POINT.

TABLE I

FIT	RHO0	US	UP	P	V/V0	E-E0	COMMENTS
	9.807	2.30	0.009	2.0	0.9962	0.00040	1ST WAVE
1	-	1.79	0.009	2.0	0.9962	0.00040	2ND -
1	-	1.86	.040	7.6	0.9795	0.00520	- -
1	-	1.95	.080	15.5	0.9597	0.0256	- -
1	-	2.05	0.126	25.5	0.9391	0.0688	- - TP
2	-	1.755	0.30	55.0	0.8388	0.220	3RD -
2	-	2.183	0.50	107.	0.771	1.205	2ND -
2	-	2.718	0.75	200.	0.724	2.81	1ST -
2	-	3.253	1.00	319.	0.693	5.00	- -
3	-	3.66	1.2	431.	0.672	7.20	- -
3	-	4.73	2.0	928.	0.577	20.0	- -
3	-	6.07	3.0	1787	0.506	45.0	- -
3	-	7.41	4.0	2908	0.460	80.0	- -
3	-	8.08	4.5	3567	0.443	101.2	- -

US = 1.771 + 2.21*UP + OR- 0.02 KM/SEC FIT 1
FOR UP BETWEEN .009 AND .126 KM/SEC.

US = 1.113 + 2.14*UP, SIG.US = .03 KM/SEC FIT 2
FOR UP BETWEEN .3 AND 1.1 KM/SEC.

US = 2.05 + 1.341*UP, SIG.US = 0.07 KM/SEC FIT 3
- FOR UP BETWEEN 1.1 AND 4.5 KM/SEC.

TABLE II LISTS CORRECTED DATA FOR THE TRANSITION WAVE OF 22---6.
P IN KBARS, AND V/V0 ARE CALCULATED ASSUMING A 2 KBAR ELASTIC WAVE
WITH PARTICLE VELOCITY = 0.009 KM/SEC.

TABLE II

RHO0	T0	T	US	UP	P	V/V0
9.804	19	40	2.050	0.126	25.5	0.9391
9.813	-48	-29	2.089	0.143	29.4	0.9820

BISMUTH. SUMMARY

RHO0	T0	T	US	UP	P	V/V0
9.804	62	85	2.033	0.118	23.7	0.9425
9.813	268	233	1.990	0.086	17.0	0.9584
US =						

COMMENTS:

- 1) SOURCE: COMPILER
- 2) THE DATA OF 22---1,2,3,4,6,7 WERE USED IN THIS SUMMARY.
- 3) THE FIRST WAVE PRESSURE OF CAST BISMUTH FROM 22---7 IS GIVEN HERE. THE MAXIMUM PRESSURE OF THIS RAMP-SHAPED PRECURSOR WAS USED WITH ITS AVERAGE VELOCITY TO CALCULATE V/V0 AND UP FOR THE FIRST WAVE USING A SINGLE STEP APPROXIMATION.
THIS SUMMARY SHOULD NOT BE EXPECTED TO YIELD AN ACCURATE DESCRIPTION OF THIS HIGHLY ANISOTROPIC MATERIAL NEAR THE 2 KBAR REGION.
- 4) FIT 1: THE TRANSITION PRESSURE OF 22---7 WAS COMBINED WITH THE US VALUE OF 22---6 TO GIVE A STRAIGHT US-UP LINE TO THE INTERCEPT CO.0.
- 5) THE ADOPTION OF THE 22---7 PHASE POINT CORRESPONDED TO THE UP1 VALUES OF 22---6 BY 6.6 PERCENT TO THOSE OF TABLE
- 6) THE SECOND SHOCK PARAMETERS OF 22---6 HAVE BEEN USED IN THE 2ND FIT.
- 7) A NUMBER OF GRUNEISEN GAMMAS HAVE BEEN ESTIMATED FOR THIS- AND OTHER V.VI ELEMENTS: D.B. SIRDESHMUKH AND B. KAMAN, J. APPL. PHYS., P. 894 (1973)

BISMUTH, SUMMARY
22---0

TABLE I

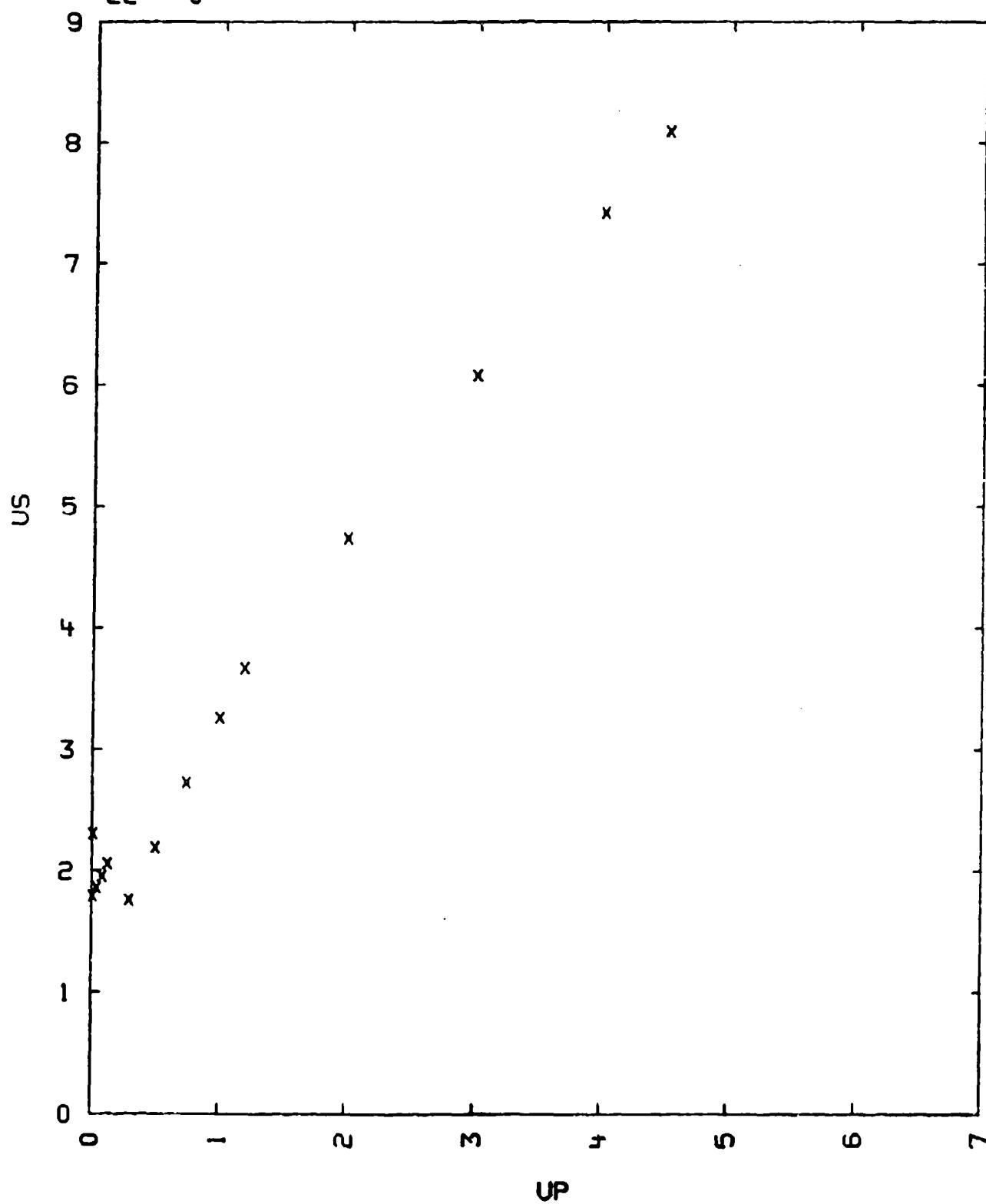
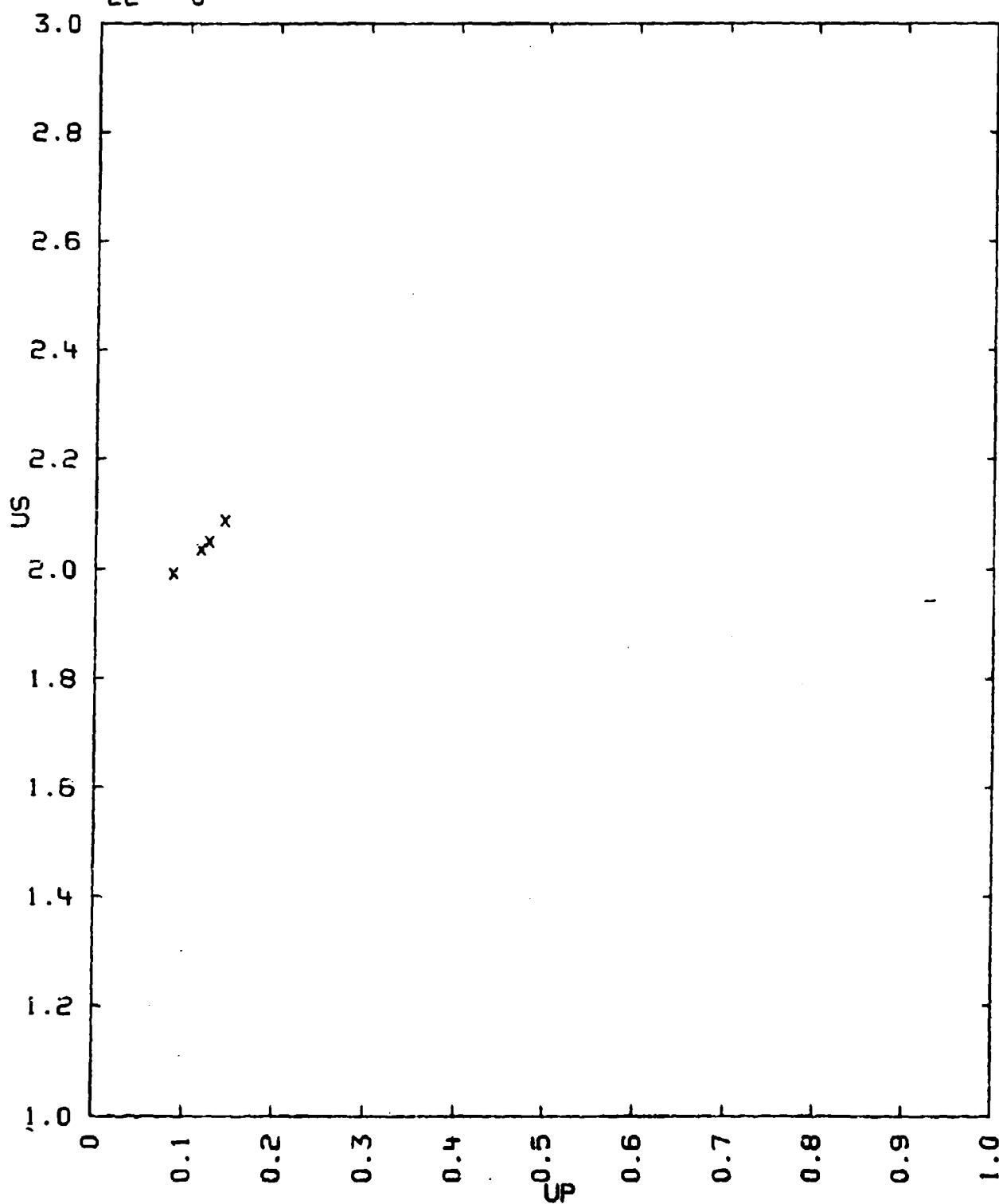


TABLE II

BISMUTH, SUMMARY

22---0



22---1
BISMUTH

SI 99.8 PER CENT OR GREATER

$$V_0 = 0.1021 \text{ CC/G} \quad C_L = 2.18 \text{ KM/SEC.} \quad C_0 = 1.77 \text{ KM/SEC.}$$

$$V_{01} = 0.1014 \text{ CC/G} \quad C_S = 1.10 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UFS	UP	P	V/V0
9.79	2.696	1.401	0.718	189.5	0.7337
-	2.585	1.318	0.676	171.1	0.7385
-	3.075	1.793	0.914	275.2	0.7028
-	3.084	1.800	0.922	278.4	0.7010
-	3.682	2.476	1.212	436.9	0.6708
-	3.659	2.564	1.222	437.7	0.6660

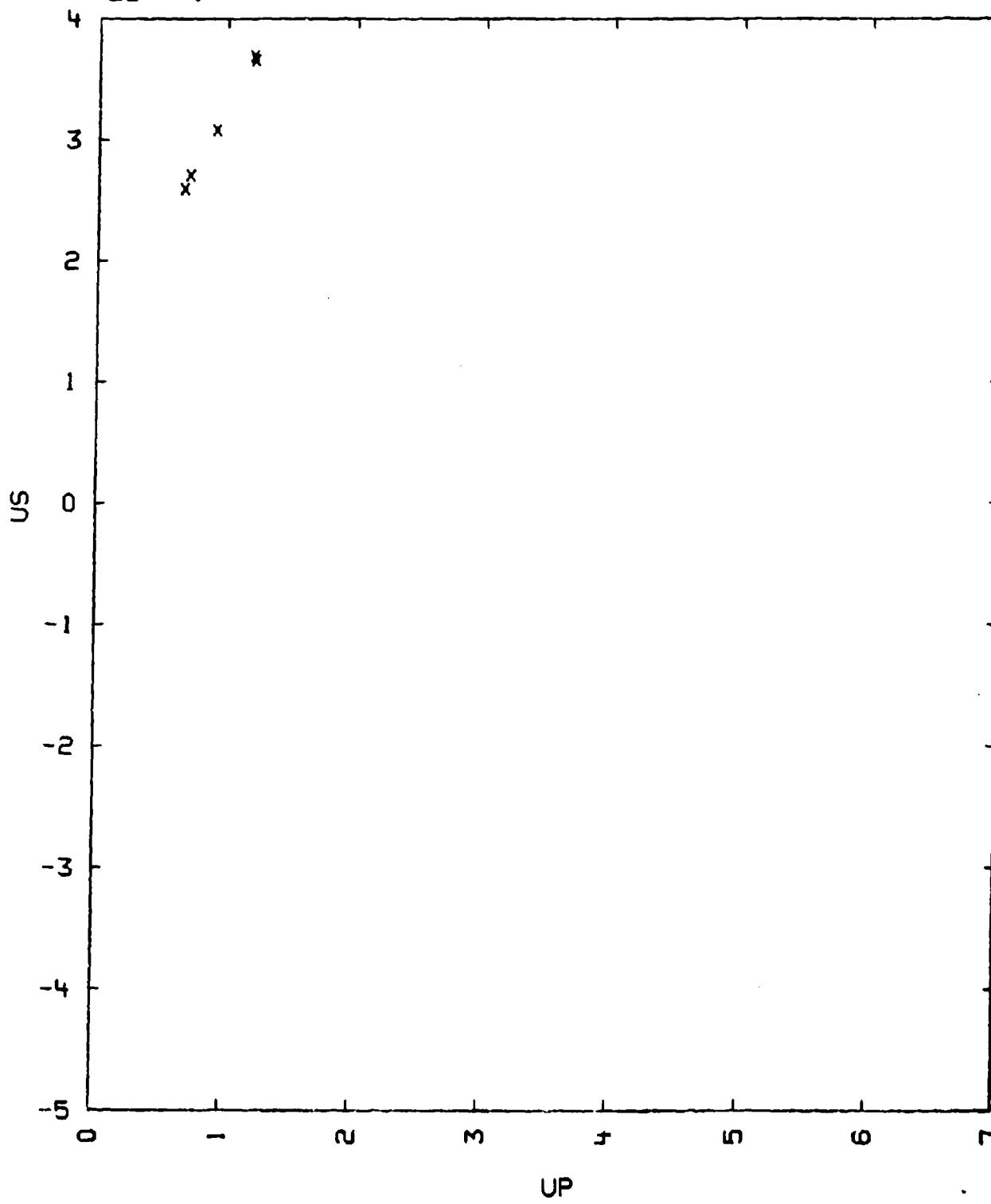
$$US = 1.259 + 1.982 UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.6 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J.M., RICE, M.H., MCQUEEN, R.G. AND YARGER, F.L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS
- 4) CL AND CS VALUES WERE OBTAINED FROM L. BERGMAN, DER ULTRASCHALL,
S. HIRZEL VERLAG, STUTTGART, 1954, 6TH ED., P. 850
- 5) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.

TABLE I

BISMUTH
22---1



22---2
BISMUTH

B1

$V_0 = 0.1015 \text{ CC/G.}$
 $V_{01} = 0.1020 \text{ CC/G.}$

 $C_0 = 1.771 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC..
PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
9.85	3.67	1.23	444	0.665	339
-	3.64	1.23	443	0.661	339
-	4.44	1.80	786	0.596	882
-	4.42	1.79	781	0.594	867
-	4.46	1.79	787	0.599	869
-	4.75	1.97	923	0.585	1008
-	4.66	2.06	945	0.558	1050
-	4.73	2.00	931	0.577	1020
-	5.33	2.47	1299	0.536	1410
-	5.36	2.47	1303	0.540	1410
-	5.51	2.51	1360	0.545	1453
-	5.49	2.48	1344	0.548	1436
-	5.49	2.47	1335	0.551	1423

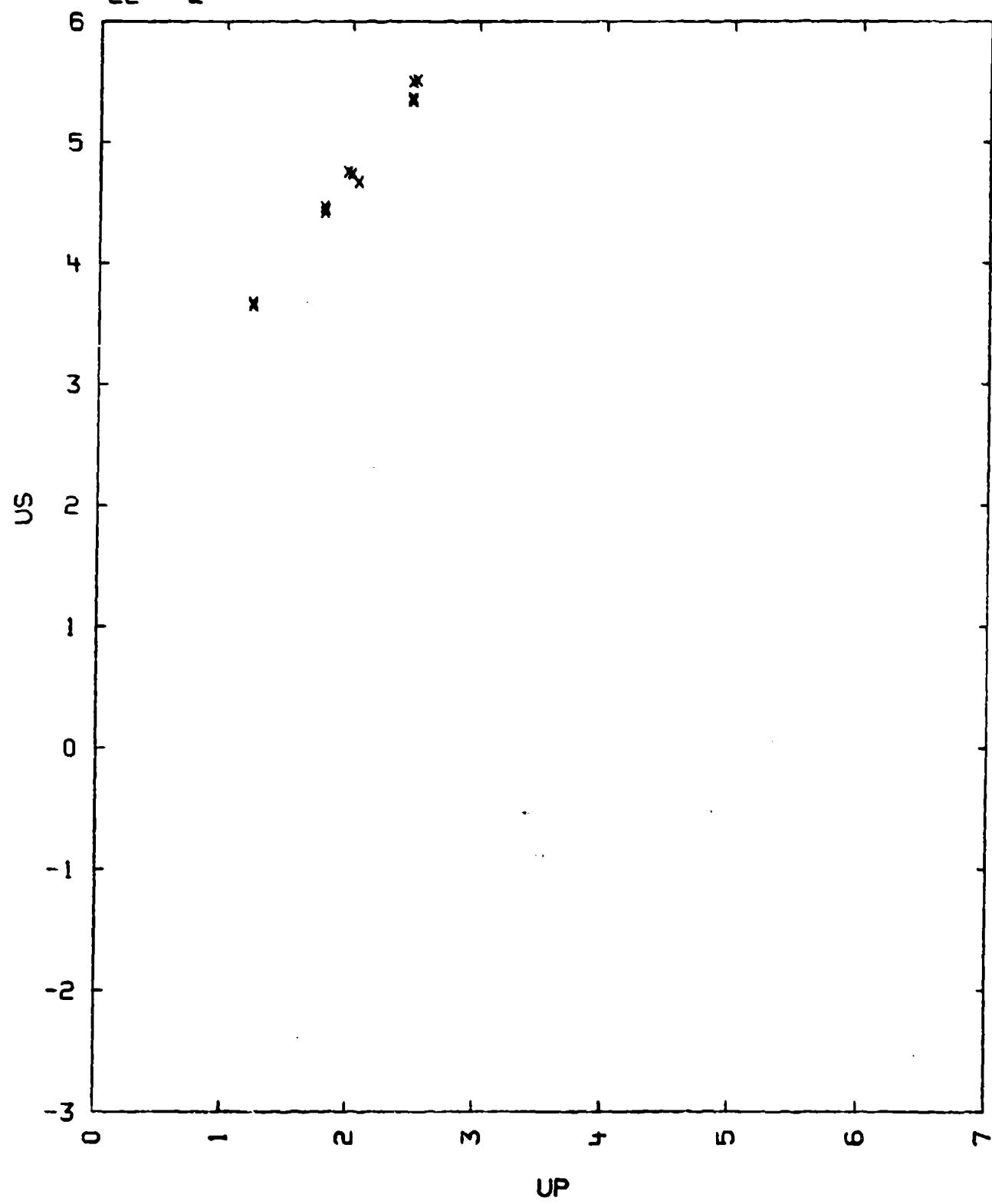
$$US = 1.88 + 1.42 UP \text{ KM/SEC. } SIG.US = 0.07 \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36--33---2
- 4) C_0 IS OBTAINED FROM DR. LUDWIG BERGMANN, DER ULTRASCHALL, 2N ED.,
(S. HIRZEL VERLAG STUTTGART, GERMANY, 1954)

TABLE I

BISMUTH
22---2



22---3
BISMUTH

81

$V_0 = 0.102 \text{ CC/G}$
 $V_{01} = 0.1020 \text{ CC/G}$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC AND PRESSURE IN KILOBARS.

TABLE

RH00	SAMPLE				BASE PLATE		
	US	UFS	UP	P	V/V0	MATERIAL	PRESSURE
9.772	3.69	2.50	1.19	428	0.678	AL	324
9.775	2.56	1.29	0.69	173	0.730	AL	146
9.781	2.54	1.34	0.69	173	0.727	AL	146
9.79	4.60	4.07	1.90	855	0.587	AL	618
9.77	2.08	0.89	0.46	94	0.778	BRASS	133
9.767	1.82	0.72	0.32	74	0.870	BRASS	95

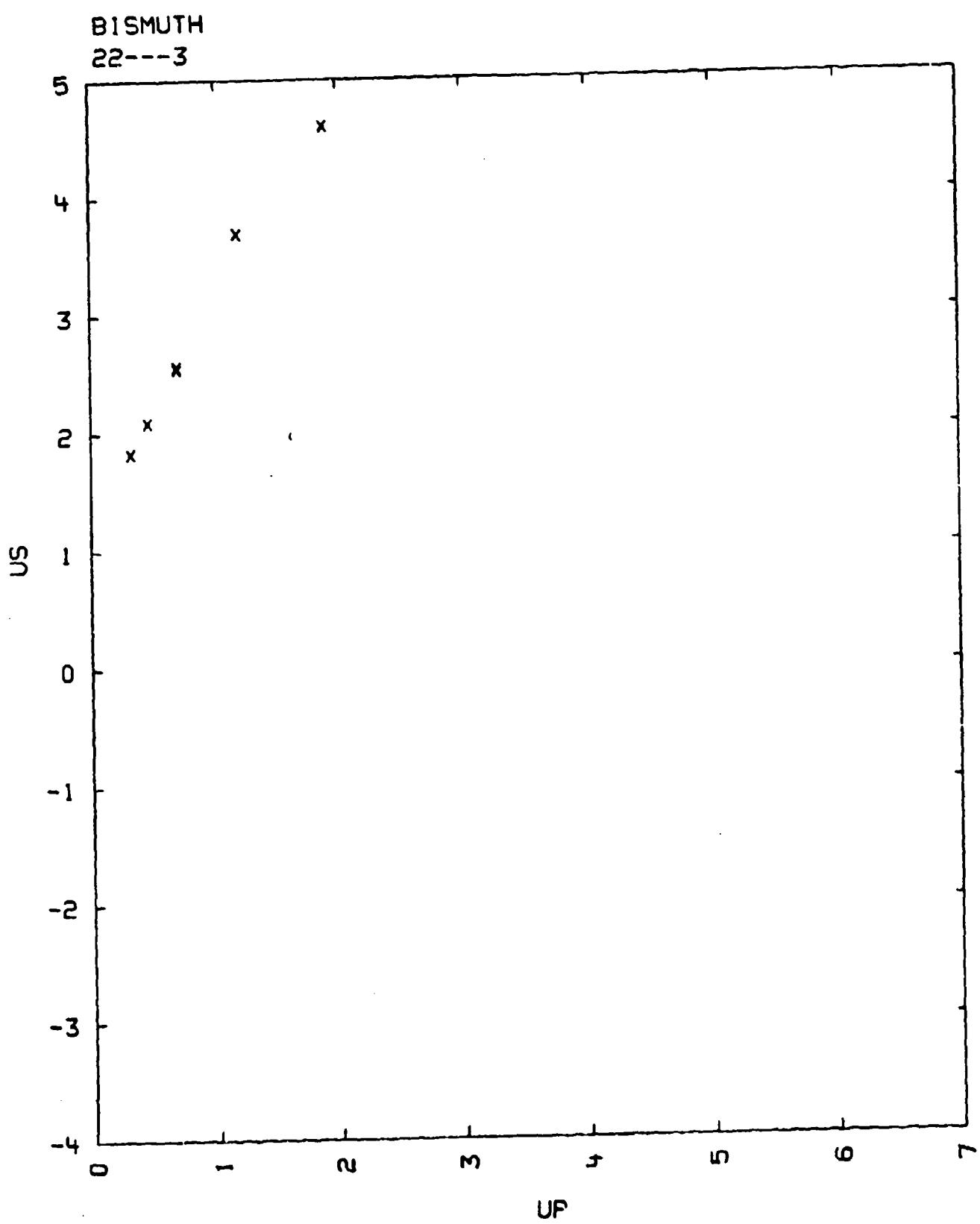
$US = 1.58 + 1.61 UP \text{ KM/SEC}$

$\Sigma US = 0.246 \text{ KM/SEC}$

COMMENTS:

- 1) SOURCE: COMPILER
 L.R.L. EQUATION OF STATE FILE
 LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B. STANDARD MATERIALS: 2024 AL ALLOY
 ASTM B16 BRASS
 DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF V_{01} WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
 CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES
 (DOVER PUBLICATIONS, INC., NEW YORK, N.Y., 1963).

TABLE I



22---4
BISMUTH

81

$V_0 = 0.1020 \text{ g/cc.}$
 $V_{01} = 0.1020 \text{ g/cc.}$

$C_B = 1.86 \text{ km/sec.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

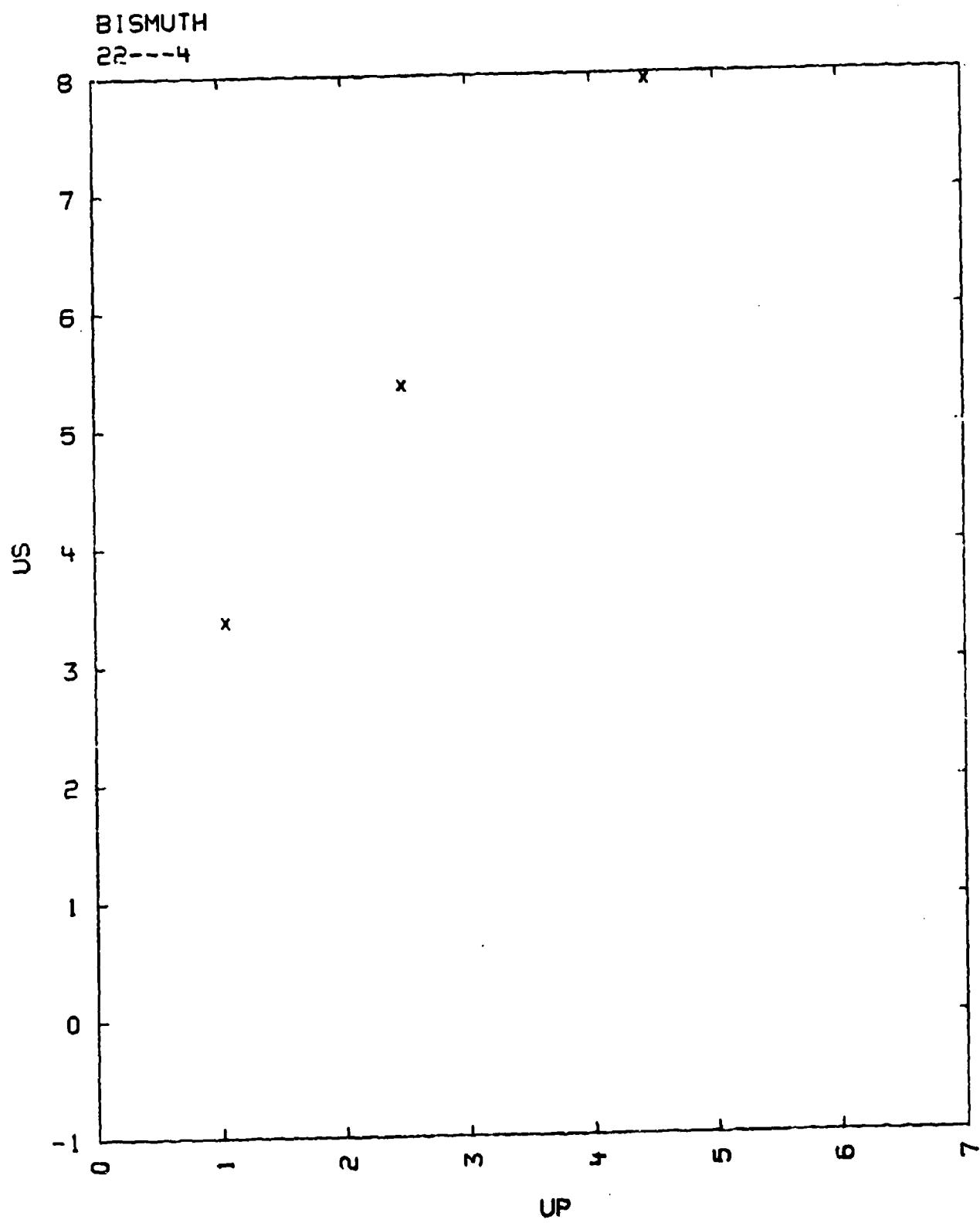
RHO0	US	UP	P	V/V0
9.80	3.37	1.05	350	0.690
	5.36	2.47	1300	0.538
	7.94	4.45	3450	0.439

$$US = 2.00 + 1.34 UP \text{ km/sec.}$$

COMMENTS:

- 1) SOURCE: AL'TSHULER, L. V., KRUPNIKOV, K. K. AND BRAZHNICK, M. I.
SOVIET PHYS.-JETP, VOL. 34, P. 614 (1958)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL FE
- 3) CB IS CALCULATED USING THE VOLUME COMPRESSION COEFFICIENT TAKEN FROM
P. W. BRIDGMAN, THE PHYSICS OF HIGH PRESSURE (G. BELL AND SONS LTD.
LONDON 1958)
- 4) VO1 WAS OBTAINED FROM THE HANDBOOK OF CHEMISTRY AND PHYSICS (THE
CHEMICAL RUBBER PUBLISHING CO. 1962-1963) 44TH ED.
- 5) THE SPECIMEN WITH A THICKNESS BETWEEN 6-8MM. WAS ATTACHED TO A IRON
PLATE OF COMPARABLE THICKNESS 6-9MM.

TABLE I



22---5
BISMUTH

B1

$$V_0 = 0.1016 - 0.1020 \text{ g/cc}$$

$$V_{01} = 0.1014 \text{ g/cc}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	-STANDARD-
					UFS
9.80	3.511	1.139	392	0.676	2.980
-	3.521	1.130	390	0.679	2.970
9.84	3.558	1.105	387	0.689	2.920
-	3.568	1.105	388	0.690	2.920
9.84	3.411	1.044	350.5	0.694	2.720
-	3.376	1.096	364	0.675	2.820
9.82	3.451	1.045	354	0.697	2.730
-	3.425	1.048	352.5	0.694	2.730
9.84	3.385	1.025	339	0.697	2.260
-	3.376	1.030	342	0.695	2.270
9.82	3.259	1.044	334	0.680	2.660
-	3.268	0.989	317.5	0.697	2.540
9.83	3.305	0.957	311	0.710	2.490
-	3.222	0.998	316	0.690	2.550
9.82	3.221	0.980	310	0.695	2.505
-	3.276	0.970	312	0.704	2.500
9.82	3.119	0.997	306.5	0.680	2.510
-	3.149	0.999	309	0.682	2.520
9.83	3.180	0.968	302.5	0.696	2.460
-	3.213	0.987	303	0.699	2.480
9.82	3.033	0.866	258	0.714	2.185
-	3.033	0.873	260	0.712	2.200
9.82	2.919	0.830	238	0.716	2.070
-	2.957	0.836	242.5	0.717	2.095
9.83	2.763	0.696	189	0.748	1.715
-	2.667	0.703	184	0.736	1.710
9.84	2.392	0.588	138.5	0.754	1.385
-	2.438	0.582	139.5	0.761	1.375
9.82	2.224	0.615	135	0.723	1.390
-	2.265	0.600	134	0.735	1.370
9.83	2.303	0.537	121.5	0.767	1.245
-	2.283	0.530	119	0.768	1.225
9.81	2.271	0.532	118.5	0.766	1.220
-	2.271	0.530	118	0.767	1.215
9.82	2.170	0.554	118	0.745	1.245
-	2.185	0.556	119	0.746	1.250
9.83	2.104	0.455	94	0.784	1.015
-	2.102	0.463	95.5	0.780	1.030
-	2.091	0.465	95.5	0.778	1.027
-	2.088	0.463	95	0.778	1.025

BISMUTH

RHO0	US	UP	P	V/V0	UFS
-	1.938	0.430	82	0.778	0.925
-	1.925	0.423	80	0.780	0.900

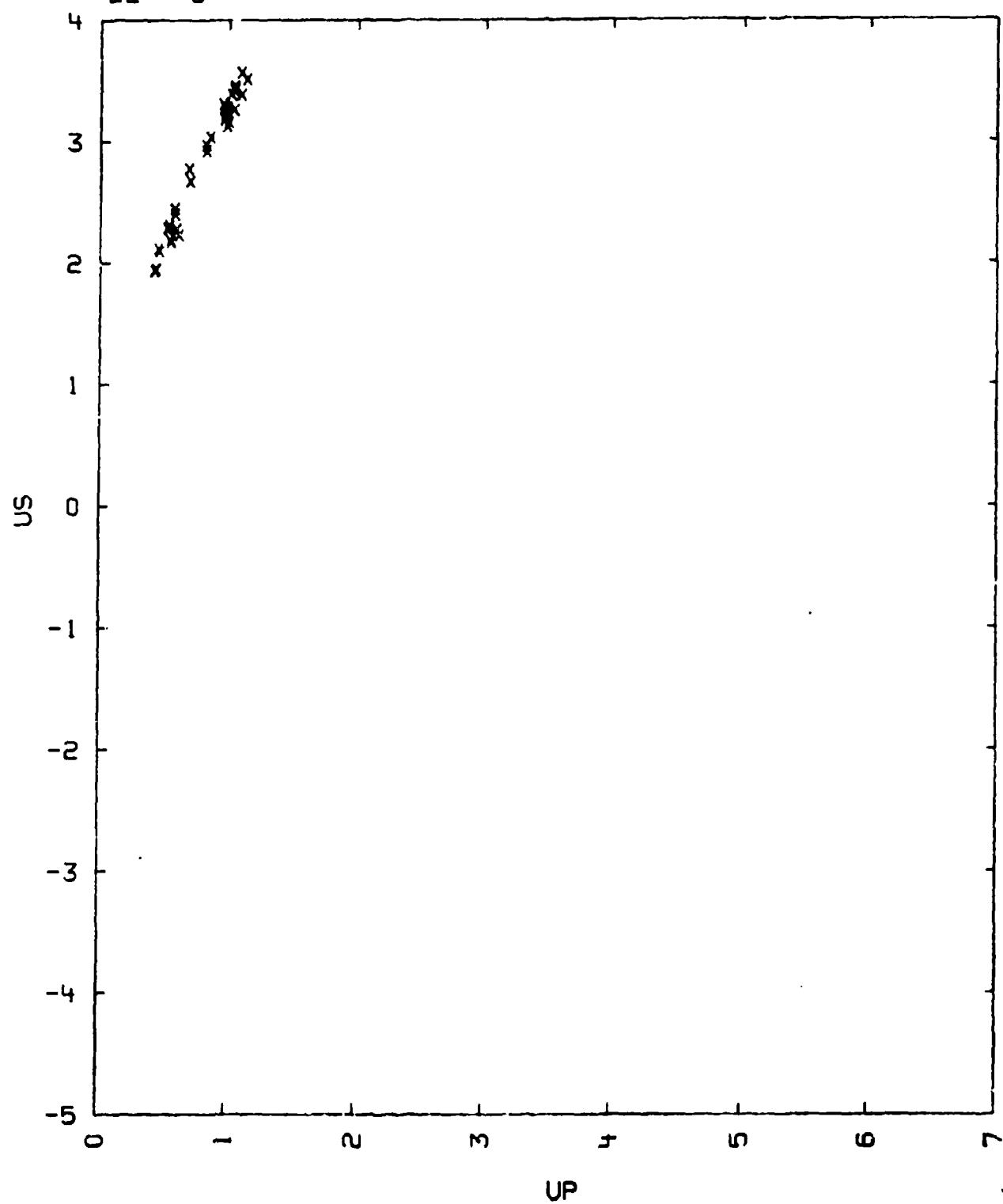
$$US = 1.050 + 2.224 UP \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: BERGER J. AND FAUQUIGNON C.
PRIVATE COMMUNICATION (1964), B.P. NO. 7, SEVRAN, FRANCE
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION B
STANDARD MATERIAL AU4G
- 3) SAMPLE DIMENSIONS WERE: 2.0 CM DIAMETER
0.5 CM THICKNESS
- 4) VOI WAS CALCULATED FROM THE CONSTANTS, A = 4.53674 AND C = 11.83834
ANGSTROMS, OF THE HEXAGONAL UNIT CELL.
AMERICAN INST. OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO. 1963) 2ND ED.

TABLE I

BISMUTH
22---5



22---6

BISMUTH

81

$V_0 = 0.1020 - 0.1021 \text{ CC/G}$ $C_L = 2.15 \text{ KM/SEC.}$
 $V_{01} = 0.1020 \text{ CC/G}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURES IN KILOBARS, DENSITY IN G/CC AND TEMPERATURE IN DEGREES CENTIGRADE.
 THE NUMBERS 1 AND 2 DESIGNATE THE FIRST AND SECOND SHOCK WAVES.

TABLE

RHO0	T0	T1	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0
9.794	19	42	2.043	0.136	27.2	0.9328	1.308	0.239	39.9	0.8508
9.804	19	42	2.041	0.135	27.0	0.9346	1.262	0.217	36.7	0.8666
9.804	19	42	2.048	0.134	26.9	0.9349	1.232	0.199	34.5	0.8787
9.804	19	42	2.063	0.136	27.5	0.9344	1.203	0.183	42.8	0.8933
9.813	-48	-27	2.088	0.153	31.3	0.9246				
9.804	62	87	2.033	0.127	25.3	0.9388				
9.813	208	236	1.990	0.092	17.5	0.9617				

US =

COMMENTS:

- 1) SOURCE: CUFF, R. E. AND MINSHALL, F. S.
 PHYS. REV., VOL. 108, P. 1207 (1957)
- 2) EXPERIMENTAL TECHNIQUE A
 DATA REDUCTION TECHNIQUE D. THE ASSUMPTION $2\bar{U}_P = UFS$ WAS USED.
- 3) V_{01} WAS OBTAINED FROM THE CRYSTAL DATA DETERMINATIVE TABLES, AMERICAN CRYSTALLOGRAPHIC ASSOCIATION, 2ED. (1963)

TABLE I

BISMUTH
22---6

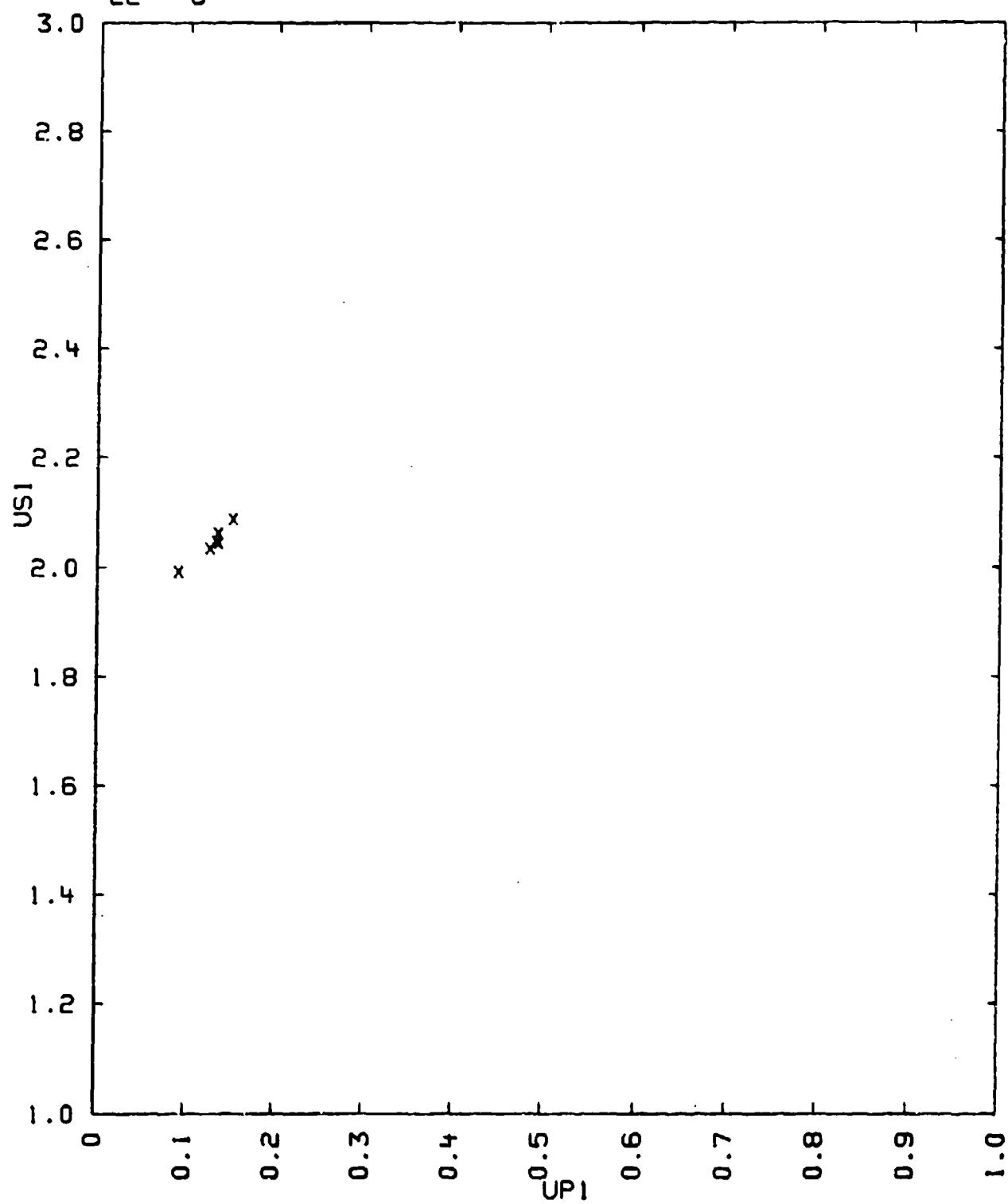
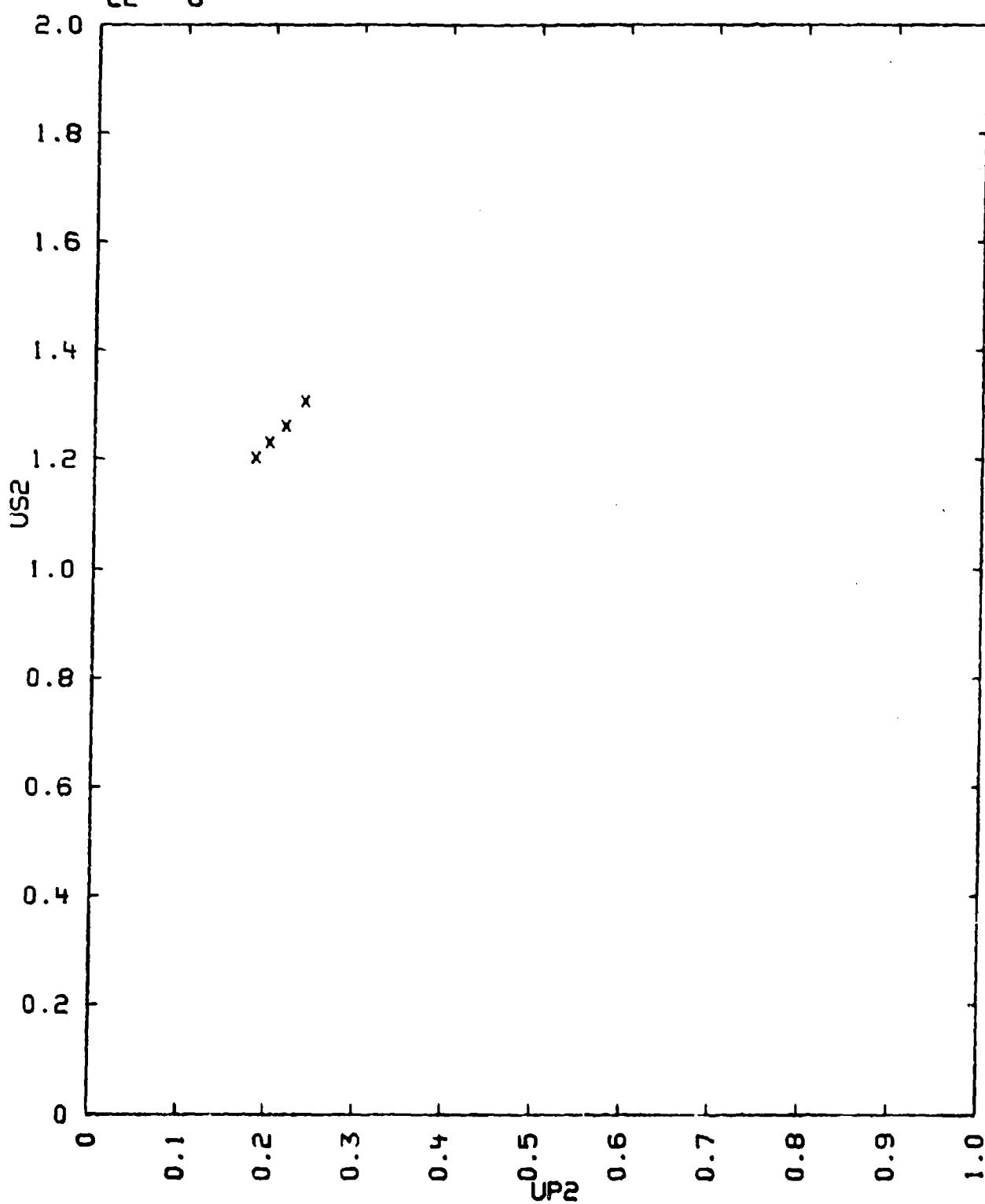


TABLE I

BISMUTH
22---6



22---7
BISMUTH

BI

VO1 = 0.101968 CC/G

CAST

VO = 0.1020 CC/G

PRESSED

VO = 0.1025 CC/G CL = 2.27 KM/SEC CO = 1.86 KM/SEC
CS = 1.13 KM/SEC

SINGLE CRYSTAL A AXIS

VO = 0.1020 CC/G CL = 2.54 KM/SEC

SINGLE CRYSTAL C AXIS

VO = 0.1020 CC/G CL = 1.98 KM/SEC
CS = 1.02 KM/SEC

IN THE TABLES BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE I
(CAST BISMUTH)

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0
9.804	2.30	0.0089	2.0	0.9961	2.06	0.126	25.5	0.939
-	-	0.0093	2.1	0.9960	2.06	0.130	26.5	0.937
-	-	0.0075	1.7	0.9967	2.06	0.126	25.6	0.939
-	-	0.0098	2.2	0.9958	2.06	0.122	24.6	0.941

US =

TABLE II
(PRESSED BISMUTH)

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0
9.756	2.16	0.0119	2.5	0.9945	2.06	0.126	25.4	0.939
-	-	0.0114	2.4	0.9947	2.06	0.124	25.1	0.940
-	-	0.0100	2.1	0.9954	2.06	0.131	26.5	0.937

US =

TABLE III

005/14/77

(SINGLE CRYSTAL BISMUTH A-AXIS)

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0
9.804	2.54	0.0104	2.6	0.9959	2.06	0.119	24.6	0.943
US =								

TABLE IV
(SINGLE CRYSTAL BISMUTH C-AXIS)

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0	
9.804	-	1.97	0.0161	3.1	0.9918	1.83	0.067	12.2	0.964
US =									

COMMENTS:

- 1) SOURCE: LARSON, D. B.
J. APPL. PHYS. (1967), IN PRINT
UCRL-14853
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
LIVERMORE, CALIFORNIA. 94550
- 2) EXPERIMENTAL TECHNIQUE I
DATA REDUCTION TECHNIQUE C.
- 3) THE TRANSITION REPORTED IN THIS PAPER IS UNDOUBTEDLY THE BI I TO
BI II TRANSFORMATION OBSERVED STATICALLY BY B. W. BRIDGMAN. PHYS.
REV. VOL. 48, P.836 (1935) AND DYNAMICALLY BY R. E. DUFF AND
F. S. MINSHALL. PHYS. REV. VOL. 108, P.1207 (1957).
- 4) THE VALUE OF V01 WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF A HEXAGONAL UNIT CELL CONTAINING SIX ATOMS
 $A = 4.54590 \text{ ANGSTROMS}$, $C = 11.86225 \text{ ANGSTROMS}$ (25 DEG. C.)
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. 1, P. 32.
- 5) THE VALUE OF US1 IS THE AVERAGE VELOCITY OF A PRESSURE RAMP WITH
LEADING EDGE TRAVELING AT 2.54 KM/SEC FOR CAST BISMUTH AND
2.27 KM/SEC FOR PRESSED BISMUTH. THE SINGLE CRYSTAL SAMPLES HAVE
WELL-DEFINED ELASTIC SHOCK WAVES.

TABLE I

BISMUTH
22---7

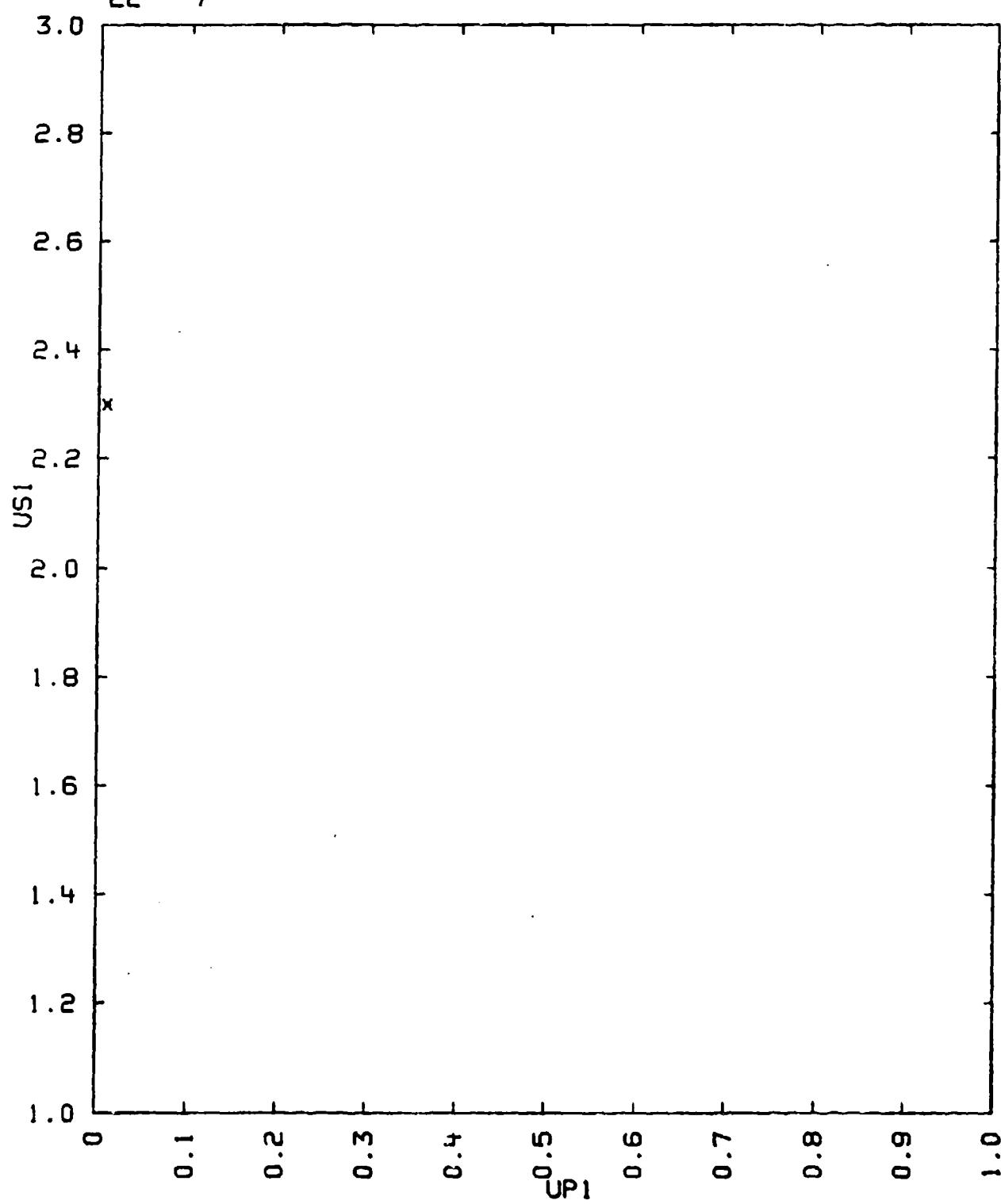


TABLE I

BISMUTH
22---7

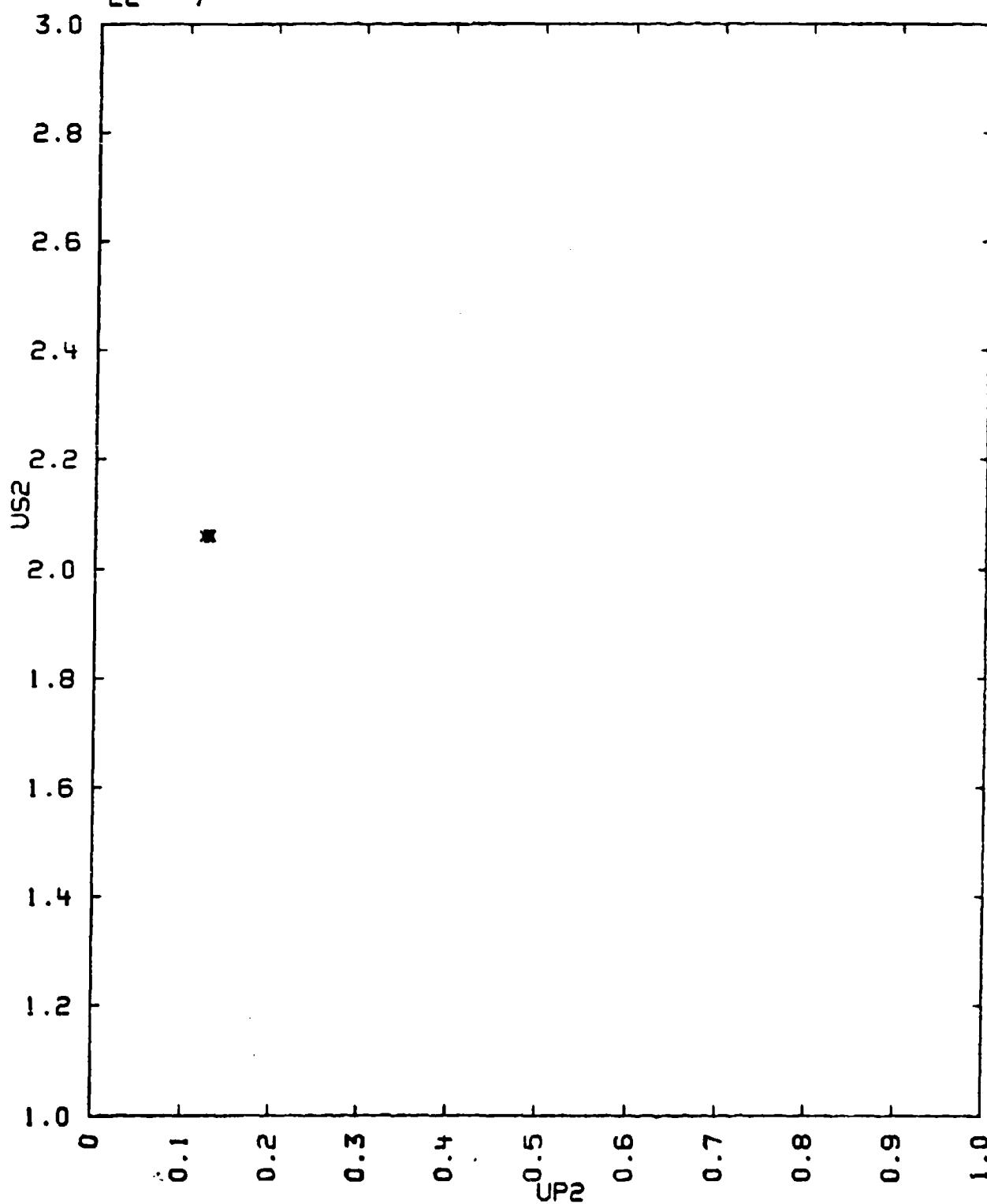


TABLE II

BISMUTH
22---7

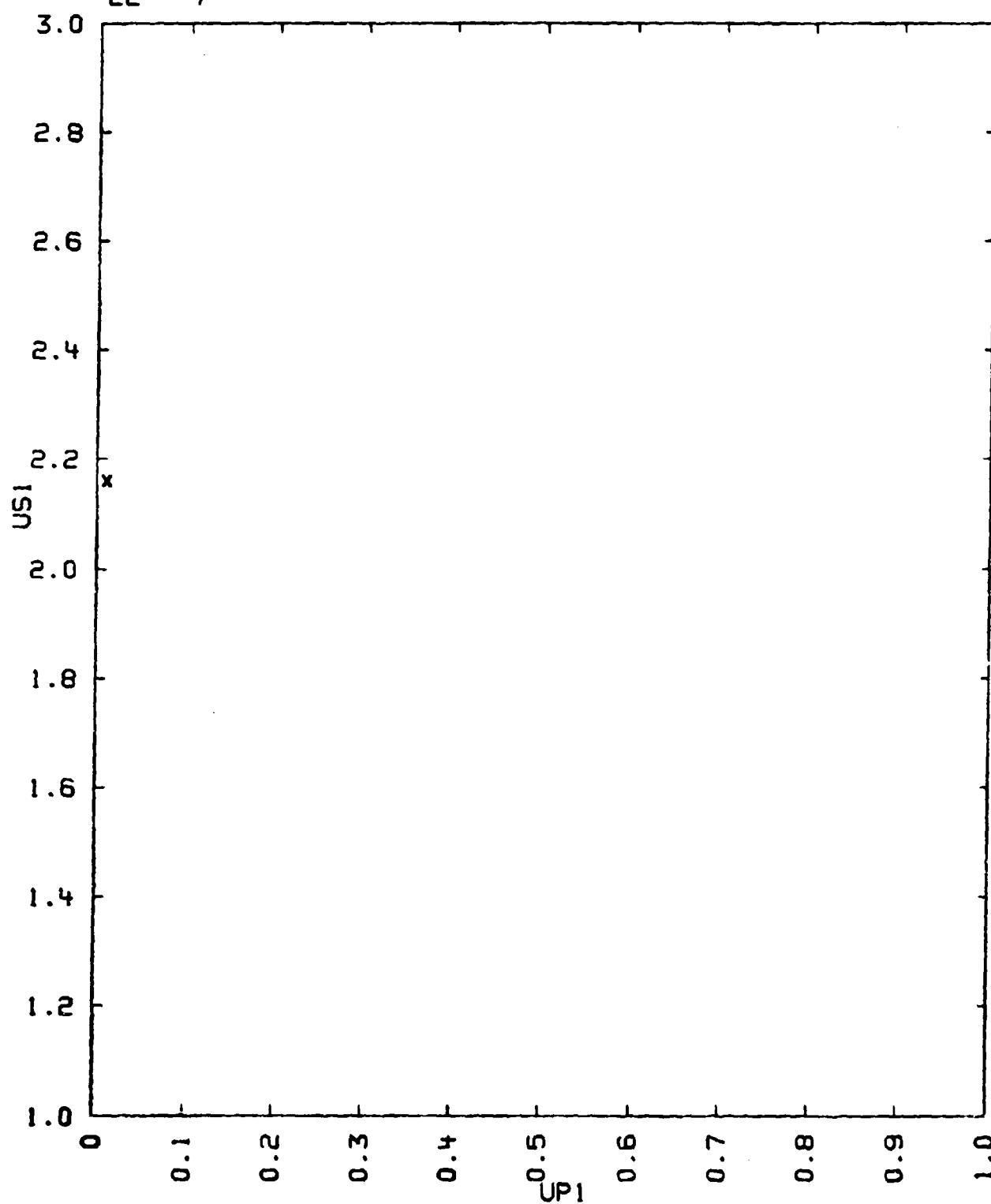


TABLE II

BISMUTH
22---7

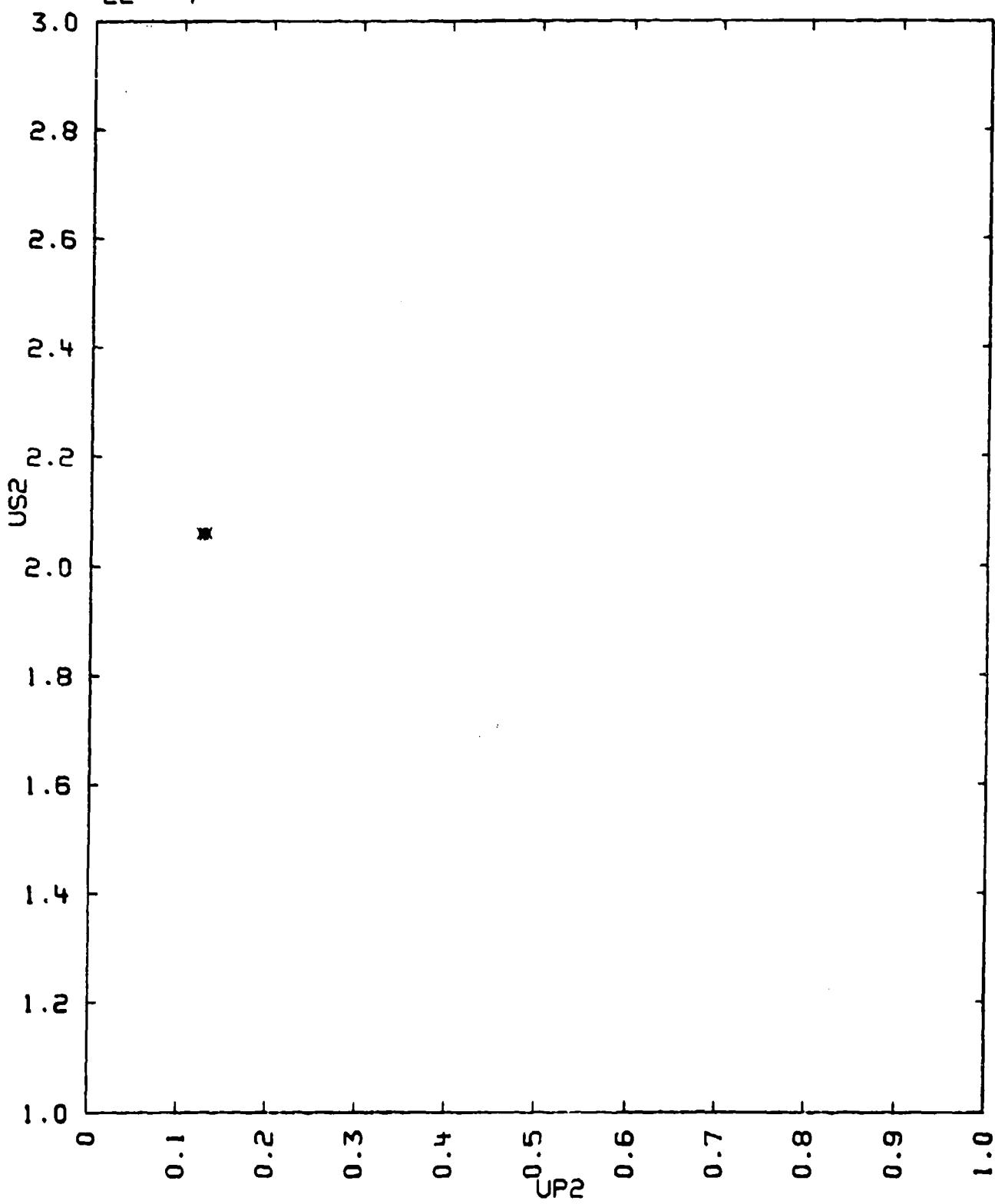


TABLE III

BISMUTH
22---7

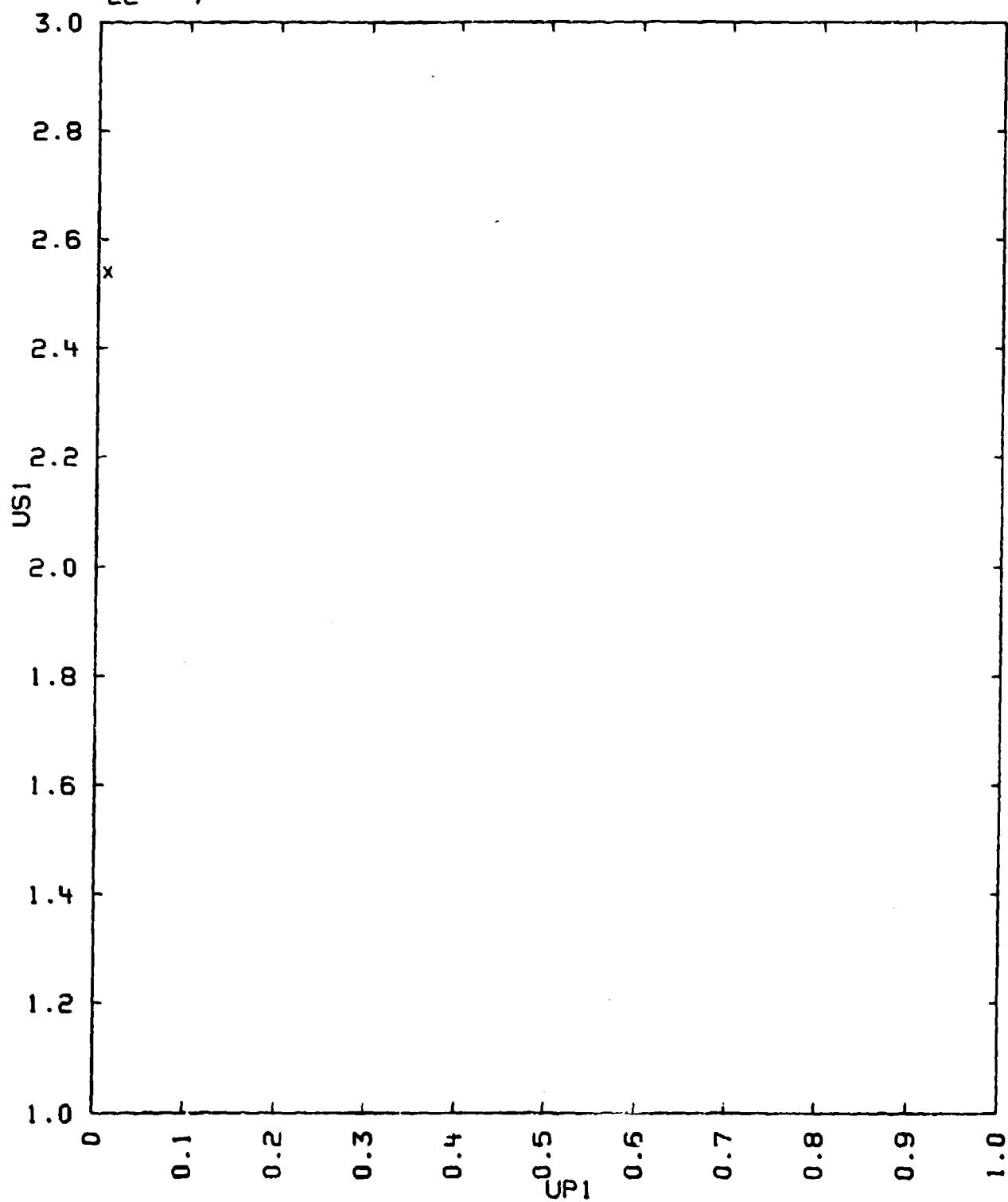


TABLE III

BISMUTH

22---7

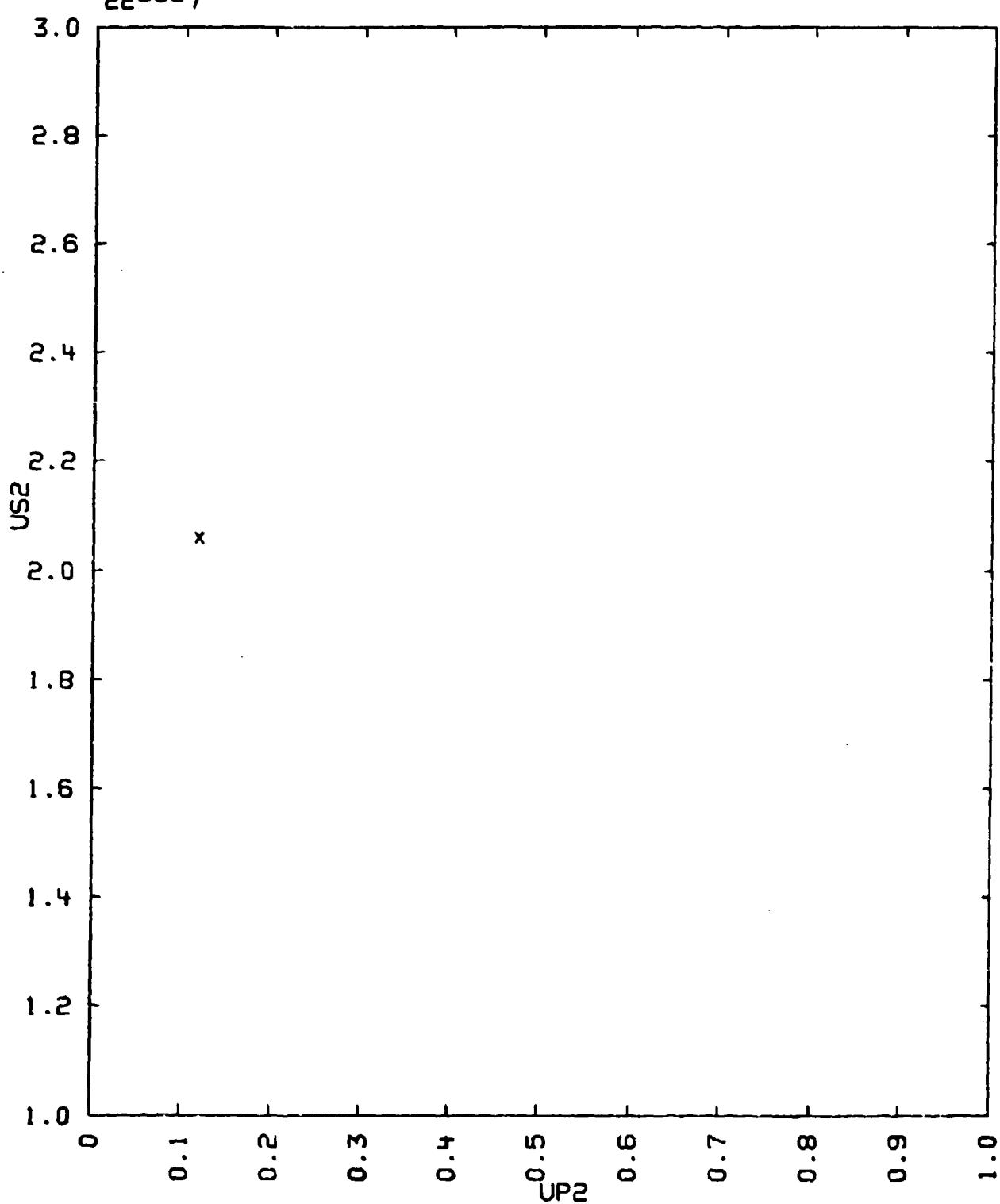


TABLE IV

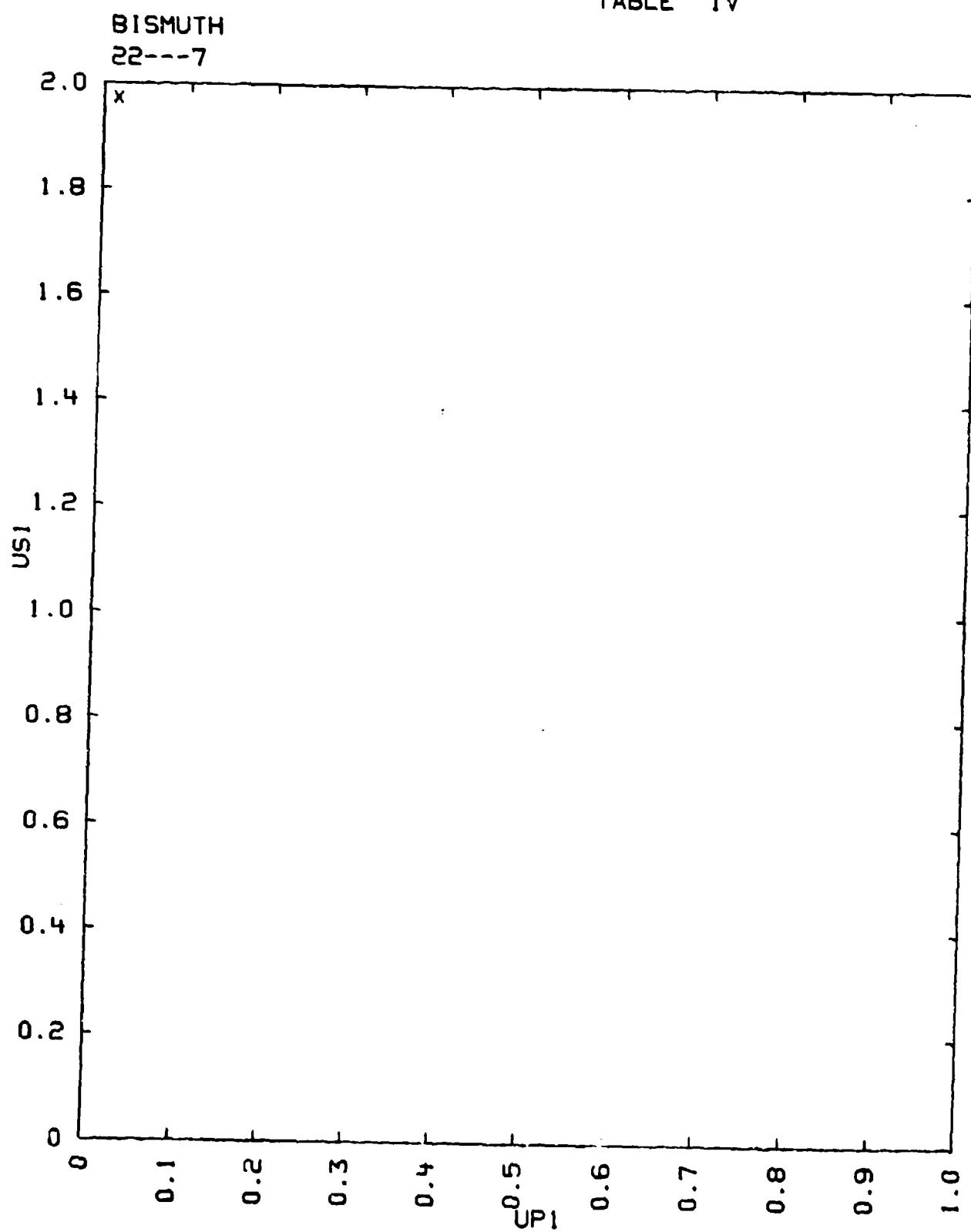
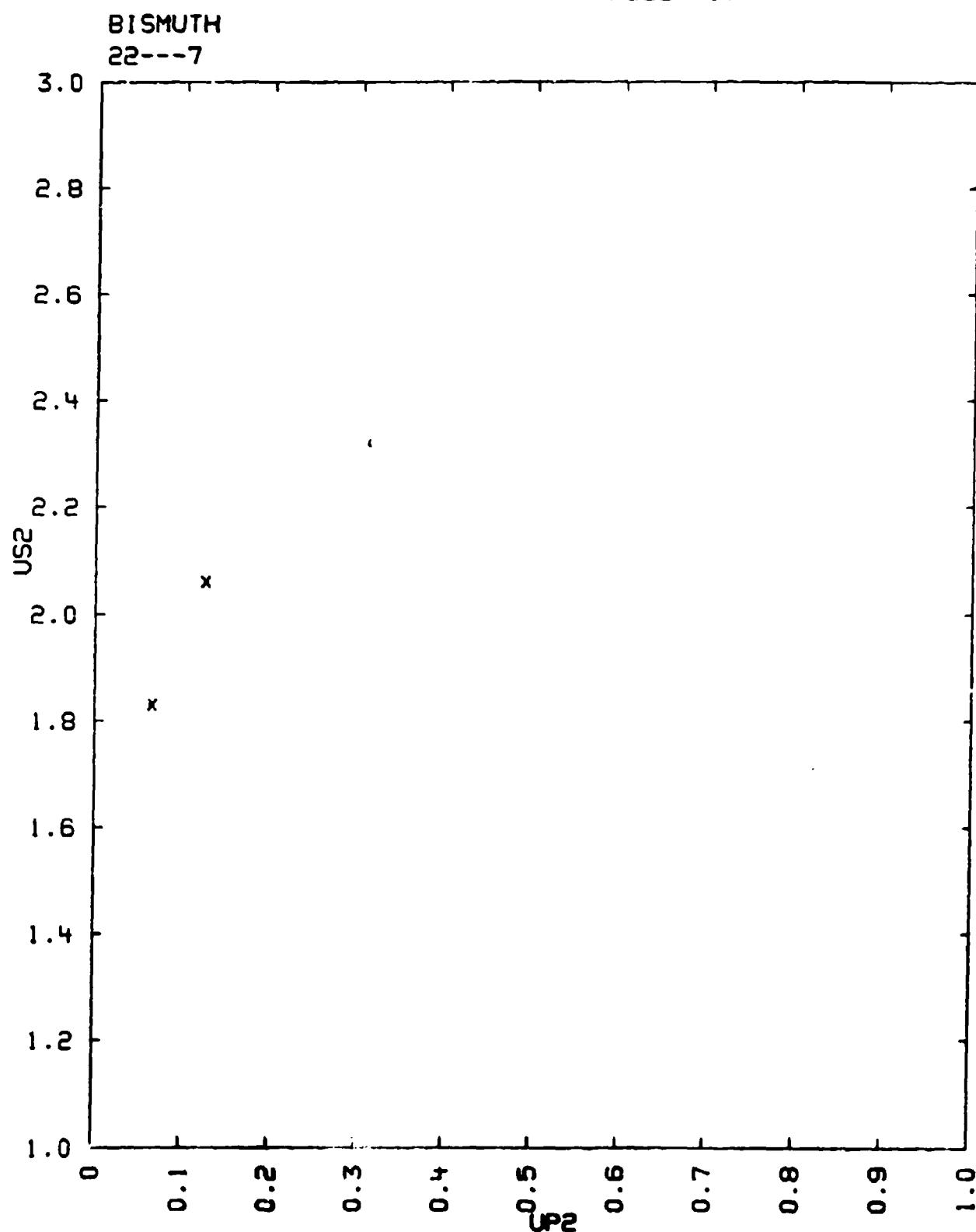


TABLE IV



22---8
BISMUTH CAST

B1

$$V_0 = 0.1020 \text{ CC/G}$$

$$V_{01} = 0.1020 \text{ CC/G}$$

IN THE TABLE BELOW, VELOCITIES ARE IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0
9.806	2.051	0.133	26.7	0.9351	1.540	0.262	45.7	0.8494
-	2.056	0.116	23.4	0.9436	1.470	0.235	40.1	0.8607
-	2.056	0.113	22.7	0.9450	1.512	0.239	41.0	0.8599
-	2.052	0.148	29.8	0.9279	1.535	0.236	42.7	0.8690

US =

COMMENTS

- 1) SOURCE: HUGHES D. S., GOURLEY L. E. AND GOURLEY M. F.
J. APPL. PHYS., VOL. 32, P. 624 (1961)
- 2) EXPERIMENTAL TECHNIQUE: A AND G
DATA REDUCTION METHOD: DI
- 3) V01 WAS OBTAINED FROM WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTER-SCIENCE PUBL., NEW-YORK, 1963) 2ND. EDITION, VOL. 1, P. 32.
- 4) THE ELASTIC WAVE OBSERVED IN ---7 WAS NEGLECTED IN CALCULATING THE ABOVE POINTS.

TABLE I

BISMUTH CAST
22---8

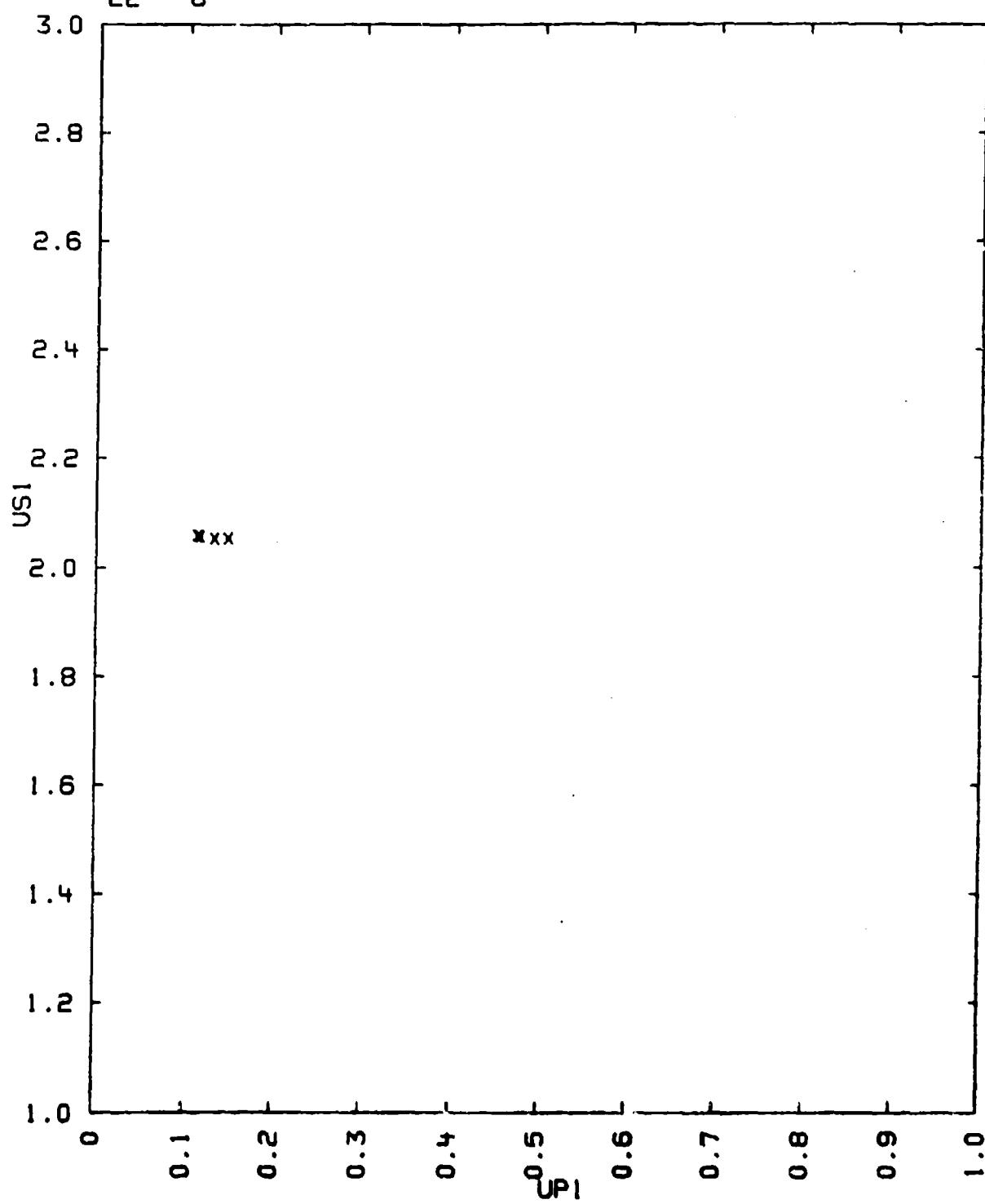
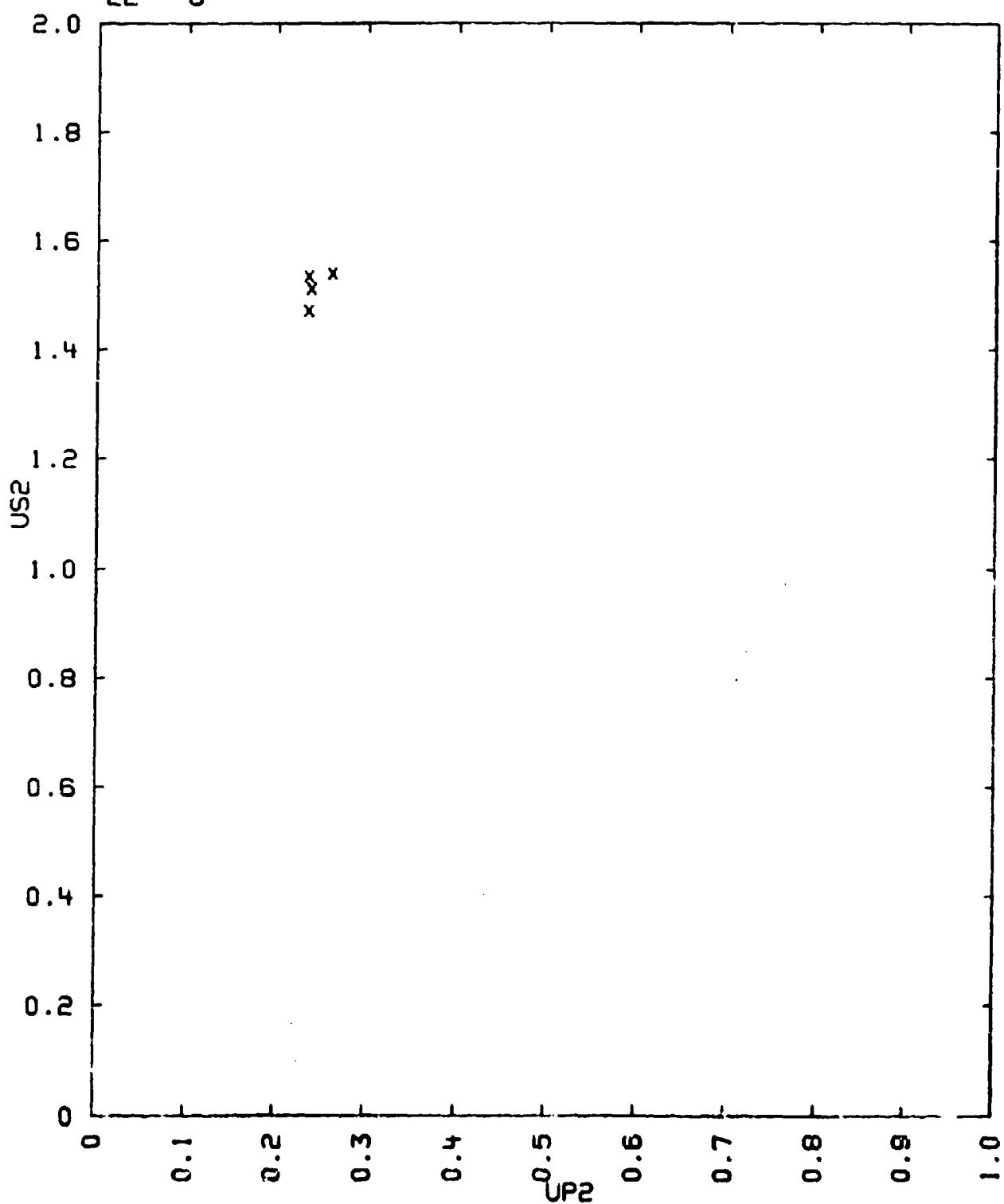


TABLE I

BISMUTH CAST

22---8



23---1
GRAPHITE PYROLYTIC

C 100 PERCENT

$$V_0 = .455 \text{ CC/G} \quad C_L = 3.45 \text{ KM/SEC}$$

$$V_{01} = .444 \text{ CC/G}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC. AND PRESSURE IN KILOBARS.

TABLE

US1	UFS1	UP1	P1	V1/V0	US2	UFS2	UP2	P2	V2/V0
7.19	3.92	1.78	281.0	0.752					
6.65	2.62	1.33	193.0	0.802					
5.95	1.93	0.93	122.0	0.844					
5.57	1.62	0.80	98.0	0.858					
5.91	1.97	0.91	118.0	0.847					
5.91	1.93	0.95	123.0	0.842					
6.66	2.82	1.42	207.0	0.788					
7.34		1.73	280.0	0.764					
5.88	1.89	0.92	119.0	0.844					
5.97	1.92	0.90	117.0	0.851					
3.28		0.038	2.7	0.988	4.72		0.515	35.5	0.895
3.2		0.027	1.9		4.30	0.65	0.287	27.2	0.933
3.2		0.027	1.9		4.40	0.63	0.304	29.4	0.930
					5.35	1.17	0.638	75.0	0.881
3.2		0.027	1.9		4.14		0.174	15.7	0.957
3.2		0.012	0.8		5.75		0.830	105.0	0.854
3.2		0.012	0.8		5.54		0.800	98.0	0.855
3.2		0.012	0.8		4.88		0.440	47.2	0.909
3.2		0.012	0.8		4.81		0.415	43.9	0.913

US = $3.70 + 2.41 UP$, SIG.US = 0.1 KM/SEC. FOR UP LESS THAN 1.0
 US = $4.57 + 1.53 UP$, SIG.US = 0.13 KM/SEC. FOR UP GREATER THAN 1.0

COMMENTS:

- 1) SOURCE: DORAN, D.G.
 J. APPL. PHYS., VOL. 34, PP. 844-850 (1963)
 HUGONIOT EQUATION OF STATE OF PYROLYTIC GRAPHITE TO 300 KBARS
 (POULTER LABORATORIES, STANFORD RESEARCH INSTITUTE, MENLO
 PARK, CALIFORNIA)
- 2) EXPERIMENTAL TECHNIQUE C AND D
 DATA REDUCTION TECHNIQUE B
 (STANDARD MATERIAL ALUMINUM)
- 3) THE DOUBLE WAVE REPORTED IN GRAPHITE ORIGINATED IN THE DRIVER PLATE,
 NOT IN THE GRAPHITE. THE SECOND SHOCK AND PARTICLE VELOCITY REPRE-
 SENT THE FINAL STATE OF THE SHOCKED GRAPHITE.

TABLE I

GRAPHITE PYROLYTIC
23---1

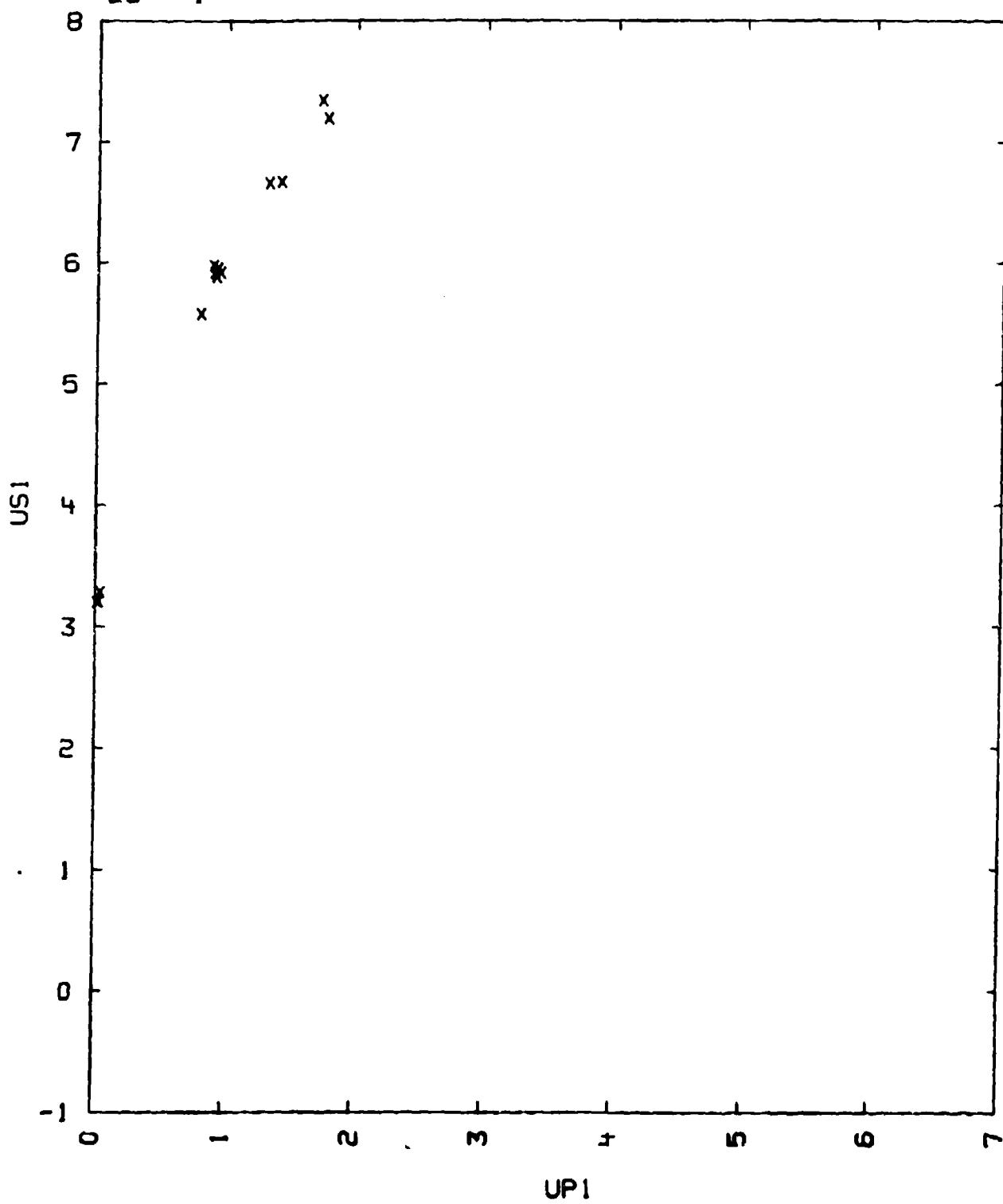
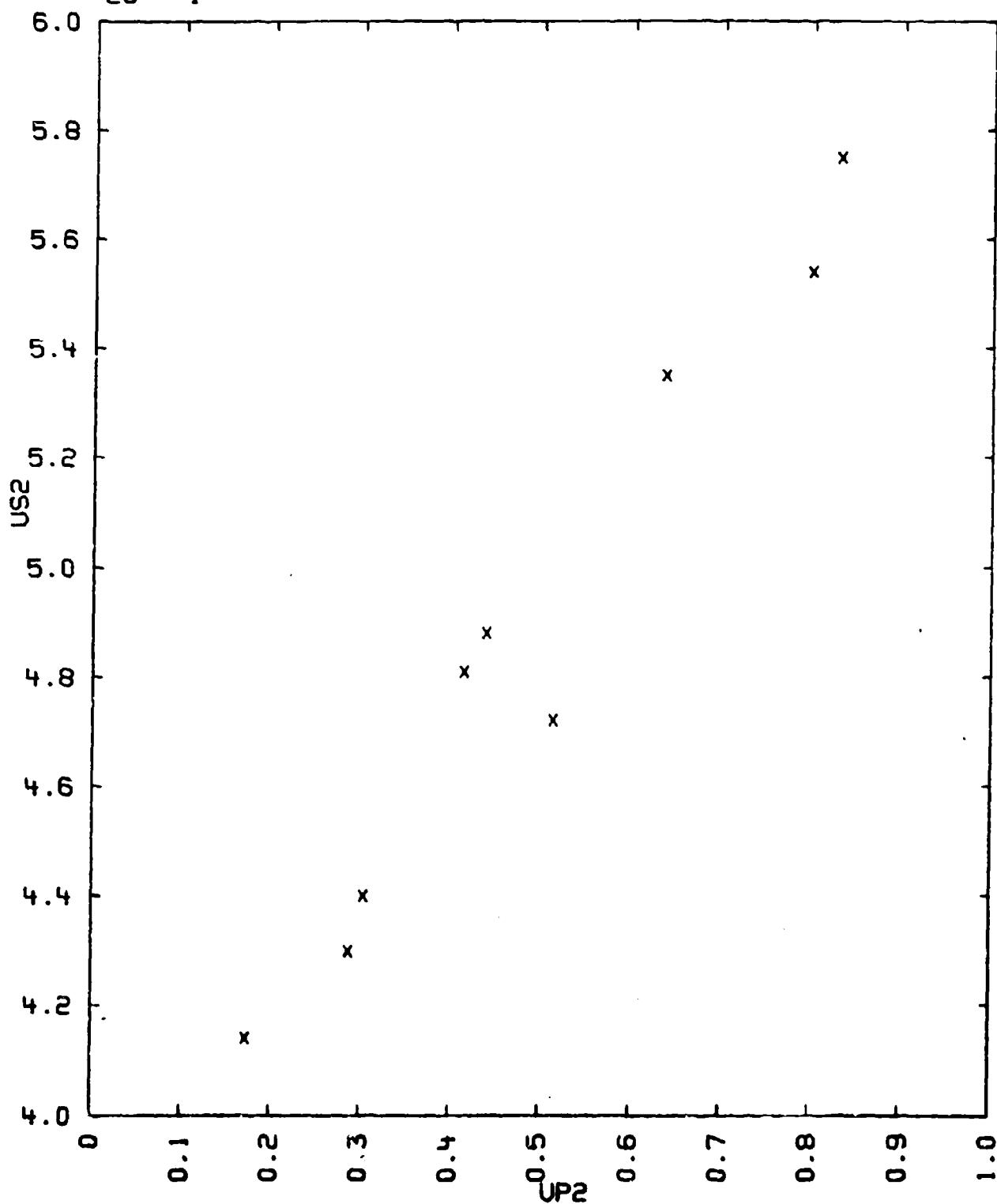


TABLE I

GRAPHITE PYROLYTIC
23---1

23---2
GRAPHITE PYROLYTIC

C 100 PERCENT

$$V_0 = 0.456 \text{ CC/G}$$

$$V_{01} = 0.444 \text{ CC/G}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC., PRESSURE IN KILOBARS, AND SPECIFIC VOLUME IN CC/G.

TABLE

US	UFS	UP	P	V/V0	BASE PLATE	BASE PLATE PRESSURE
5.901	2.360	0.893	115.9	0.847	BRASS	240.0
5.955	2.340	0.885	115.9	0.851	BRASS	240.0
5.790	1.760	0.855	108.9	0.852	BRASS	227.0
5.790	1.760	0.860	109.5	0.851	BRASS	227.0
5.514	1.586	0.787	95.5	0.857	BRASS	202.0
4.932	-	0.464	50.3	0.906	BRASS	108.0
4.932	-	0.464	50.3	0.906	BRASS	108.0
4.420	-	0.305	29.7	0.931	BRASS	95.4
4.420	-	0.305	29.7	0.931	BRASS	95.4
5.016	-	0.404	44.6	0.919	BRASS	108.0
5.003	-	0.479	52.7	0.904	BRASS	108.0
4.750	-	0.350	36.6	0.926	BRASS	95.4
5.750	1.762	(0.881)	111.4	0.847	BRASS	-
5.700	1.768	(0.884)	110.9	0.845	BRASS	-
5.440	1.668	0.798	95.5	0.853	BRASS	200.0
5.440	1.760	0.802	96.0	0.853	BRASS	200.0
6.550	2.796	1.504	216.7	0.770	AL	254.0
6.617	2.786	1.487	216.5	0.775	AL	254.0
6.337	2.574	1.237	172.5	0.805	AL	206.0
6.284	2.512	1.239	171.3	0.803	AL	206.0
7.803	4.656	2.252	386.6	0.711	BRASS	752.0
8.370	5.032	(2.516)	463.3	0.699	AL	318.0
6.342	4.980	2.501	459.0	0.700	AL	543.0
7.445	-	1.900	311.2	0.745	AL	356.0
7.628	-	2.180	365.8	0.714	BRASS	566.0
8.322	4.780	2.332	427.0	0.720	AL	318.0
7.577	4.118	(2.059)	343.2	0.728	BRASS	-

US = 3.91 + 2.14 UP, SIG.US = 0.14 BELOW UP = 1.0 KM/SEC.

US = 5.01 + 1.05 UP, SIG.US = 0.05 FOR UP BETWEEN 1.0 AND 1.7
KM/SEC.

US = 4.20 + 1.68*UP, SIG.US = 0.20 ABOVE UP = 1.7 KM/SEC.

COMMENTS:

- 1) SOURCE: COLEBURN, N. L.
J. CHEM. PHYS., V.40, P.71 (1964)

REPORT NO. NOLTR 63-73 (1963)

THE COMPRESSIBILITY OF PYROLYTIC GRAPHITE

UNITED STATES NAVAL ORDNANCE LABORATORY, WHITE OAK, SILVER
SPRING, MARYLAND.

- 2) EXPERIMENTAL TECHNIQUE B AND C
DATA REDUCTION TECHNIQUE B AND D
UP OBTAINED BY TECHNIQUE D IS ENCLOSED IN PARENTHESES.
- 3) THE PYROLYTIC GRAPHITE WAS OBTAINED FROM HIGH TEMPERATURES MATERIALS,
INC., LOWELL, MASS.
- 4) LONGITUDINAL SOUND VELOCITY:
C-AXIS NORMAL TO THE ORIGINAL PLANE OF DEPOSITION OF GRAPHITE
A-AXIS PARALLEL TO ORIGINAL PLANE OF DEPOSITION

TABLE

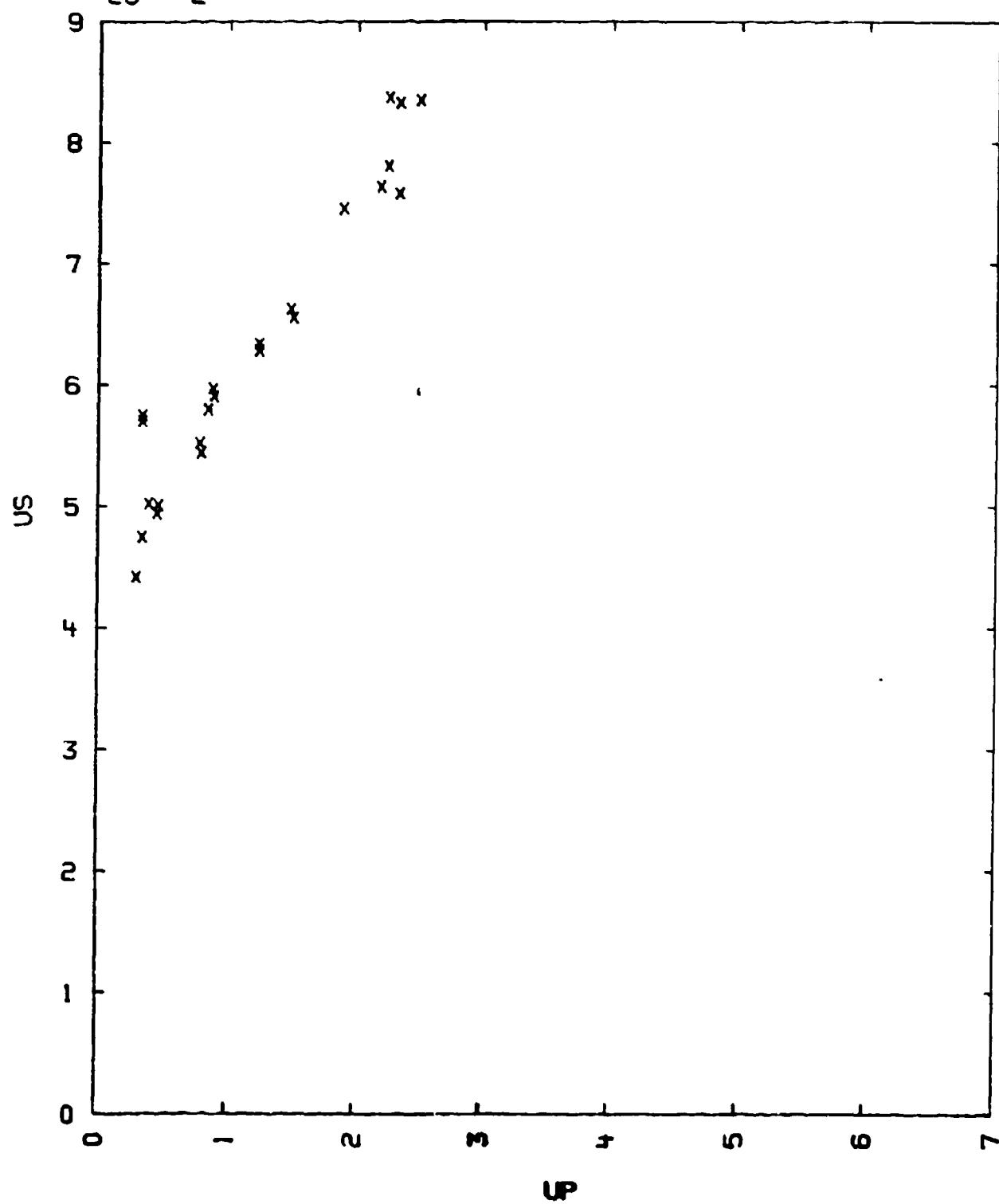
C-AXIS		A-AXIS	
TEMP. DEG. K.	VELOCITY KM/SEC.	TEMP. DEG. K.	VELOCITY KM/SEC.
51	3.518	298	4.791
62	3.516		
78	3.512		
103	3.507		
130	3.495		
165	3.483		
199	3.465		
234	3.460		
254	3.454		
273	3.446		
293	3.443		
313	3.433		
334	3.430		

- 5) THERE WAS NO MEASURABLE DIFFERENCE BETWEEN THE SHOCK VELOCITIES ALONG
THE C AND A AXES.

TABLE I

GRAPHITE PYROLYTIC

23---2



23---3
GRAPHITE COMMERCIAL

C

$$V_0 = 0.6143 \text{ CC/G}$$

$$V_{01} = 0.4412 \text{ CC/G}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	UFS(ST) (ALUMINUM)	UFS(ST) (MAGNESIUM)
1.628	5.312	2.331	200.5	0.561		4.080
-	5.192	2.347	199.5	0.548		4.080
-	5.263	2.442	210	0.536		4.260
-	5.263	2.343	201.5	0.565		4.100
-	4.960	2.062	167	0.584	2.990	
-	4.907	2.100	168	0.572	3.030	
-	5.071	2.048	170	0.578	3.020	
-	5.192	2.011	171	0.594	2.980	
-	4.634	1.893	142	0.591	2.700	
-	4.634	1.880	141	0.594	2.690	
-	4.726	1.909	147	0.596	2.740	
-	4.726	1.896	146	0.599	2.720	
-	4.583	1.913	143.5	0.583	2.740	
-	4.583	1.967	147.5	0.571	2.800	
-	4.695	1.829	139	0.610	2.630	
-	4.695	1.829	139	0.610	2.630	
-	4.241	1.739	120	0.590	2.430	
-	4.184	1.728	117.5	0.587	2.415	
-	4.464	1.760	128.5	0.606	2.505	
-	4.378	1.746	124	0.601	2.485	
-	4.115	1.537	103	0.626	2.160	
-	4.115	1.537	103	0.626	2.160	
-	4.039	1.576	104	0.610	2.200	
-	4.115	1.530	102.5	0.628	2.140	
-	3.720	1.336	81.5	0.641	1.835	
-	3.782	1.331	82.5	0.648	1.835	
-	3.660	1.350	81	0.631	1.835	
-	3.660	1.350	81	0.631	1.835	
-	3.448	1.286	72	0.627	1.720	
-	3.390	1.300	71.5	0.617	1.745	
-	3.472	1.246	71	0.641	1.695	
-	3.526	1.246	71	0.647	1.695	
-	3.526	1.246	71	0.647	1.695	
-	3.185	1.135	59	0.644	1.495	
-	3.123	1.137	58	0.536	1.495	
-	2.761	0.911	41	0.670	1.185	
-	2.778	0.933	42	0.664	1.187	
-	3.023	1.020	50	0.663	1.335	

GRAPHITE COMMERCIAL

RHO0	US	UP	P	V/V0	UFS(ST)	UFS(ST)
-	3.003	1.041	51	0.653	1.350	
-	3.043	1.010	50.5	0.668	1.330	
-	3.003	1.010	49.5	0.664	1.330	
-	2.726	0.955	42	0.650	1.215	
-	2.743	0.956	43	0.651	1.215	
-	2.761	0.911	41	0.670	1.175	
-	2.737	0.912	41.5	0.663	1.190	

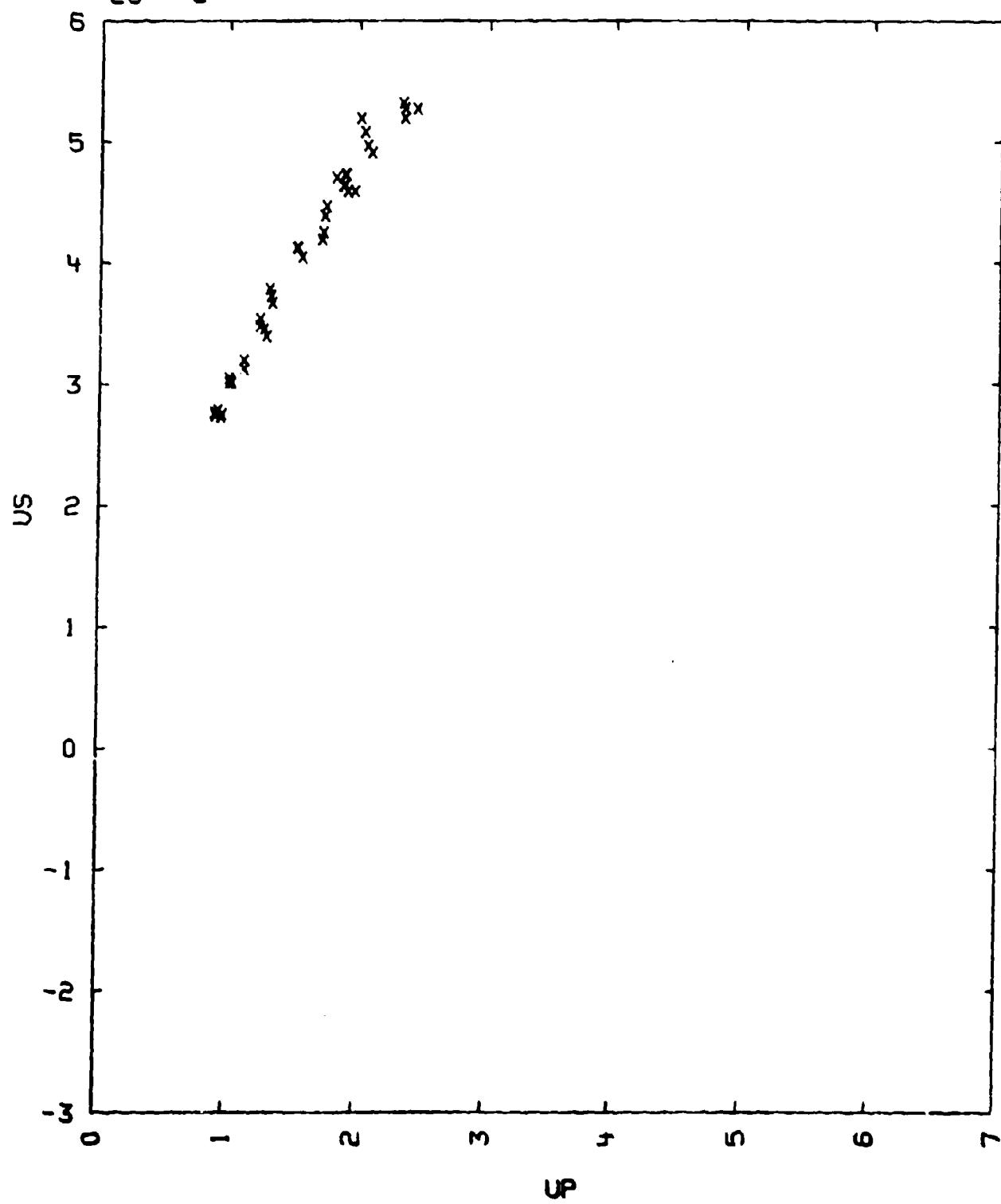
$$US = 1.162 + 1.822 UP \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: BERGER J. AND FAUQUIGNON C.
PRIVATE COMMUNICATION (1964), B.P. NO. 7, SEVRAN, FRANCE
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIALS, MAGNESIUM AND ALUMINUM AU4G ALLOY
- 3) VOI WAS OBTAINED FROM CRYSTAL DATA DETERMINATIVE TABLE (AMERICAN CRYSTALLOGRAPHIC ASSOCIATION) 2ND ED., APRIL 1, 1963.
- 4) SAMPLE DIMENSIONS WERE: 2.0 CM DIAMETER
0.5 CM THICKNESS

TABLE I

GRAPHITE COMMERCIAL
23---3



23---4
CARBON (GRAPHITE, CEYLON-PRESSED)

C

$$V_0 = 0.460 \text{ CC/G}$$

$$V_{01} = 0.4412 \text{ CC/G}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC. D IS SAMPLE THICKNESS IN MM.

TABLE

SAMPLE							BASEPLATE
RHO0	US	UFS	UP	P	V/V0	D	P
2.185	7.60		2.94	468	.613	6.2	612
2.185	7.43	5.19				3.1	612
2.181	7.41		3.02	488	.593	6.2	624
2.181	7.34	5.15				3.1	624
2.180	7.87		3.36	576	.573	6.2	730
2.180	7.99	5.45				3.1	730
2.170	6.91		2.72	408	.606	6.2	532
2.170	6.68	4.72				3.1	532
2.180	7.93		3.26	564	.589	6.2	707
2.180	7.79	5.52				3.1	707
2.179	6.47		2.21	311	.659	6.2	395
2.179	6.36	3.77				3.1	395
2.172	7.78		3.38	571	.566	6.2	730
2.170	7.65		3.31	549	.567	6.2	707
2.181	6.50	3.76	2.20	312	.661	6.2	401
2.179	6.52		2.20	312	.662	6.2	401

$$US = 3.90 + 1.177 \cdot UP \text{ KM/SEC.}$$

$$\text{SIG.US} = 0.14 \text{ KM/SEC.}$$

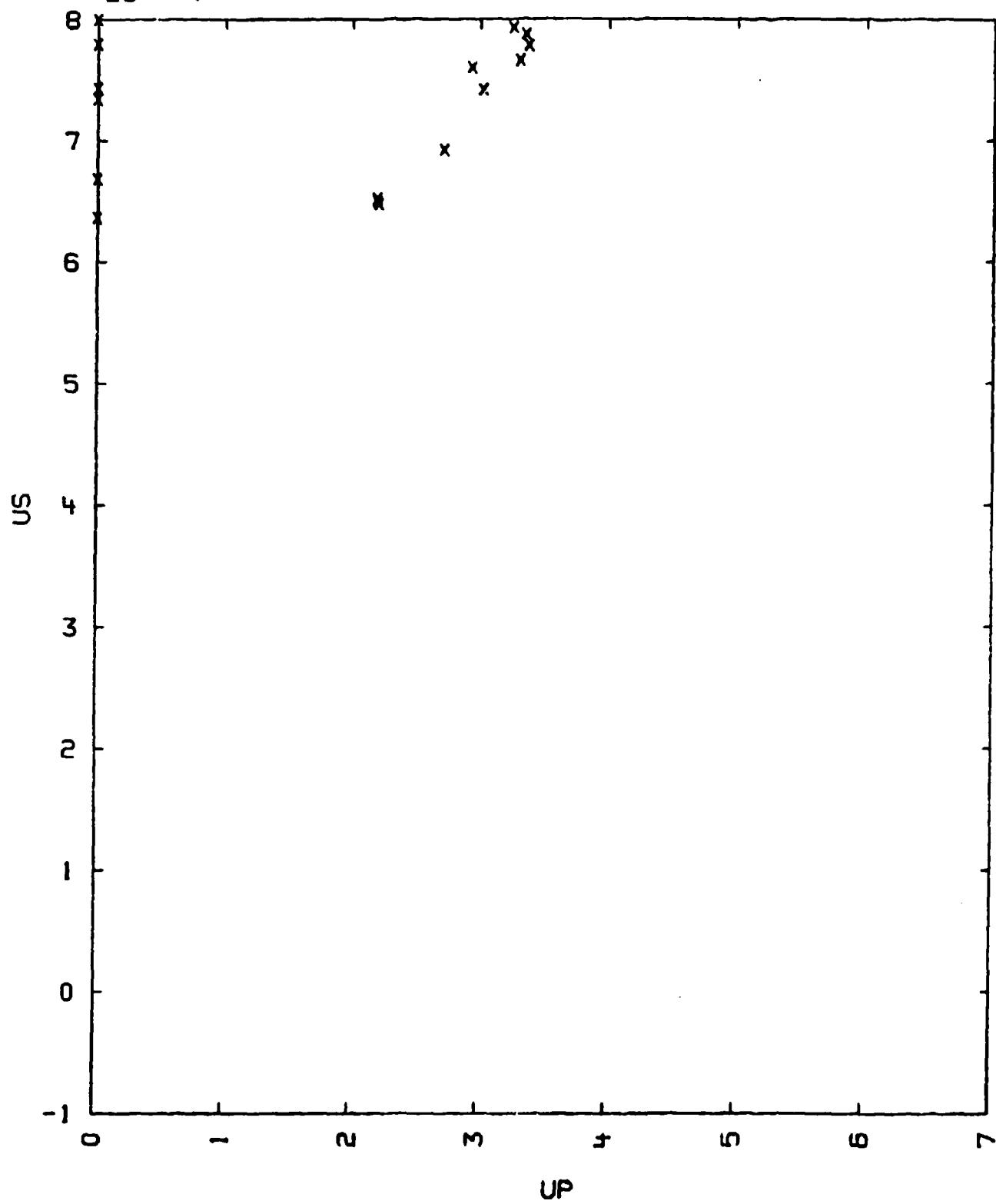
COMMENTS:

- 1) SOURCE: COMPILER
L. R. L. EQUATION OF STATE FILE
LAURENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B. (ALUMINUM STANDARD BASE PLATE)
DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF V₀₁ WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS OF THE HEXAGONAL UNIT CELL:
A = 2.4612 ANGSTROMS, C = 6.7079 ANGSTROMS. A.C.A. MONOGRAPH NO.5 CRYSTAL DATA DETERMINATIVE TABLES, 2ND EDITION (AMERICAN CRYSTALLOGRAPHIC ASSOCIATION 1963)
- 4) THE ABOVE DATA AND DATA FOR COMMERCIAL CARBON, (23---5), HAVE BEEN DISCUSSED BY B. J. ALDER AND R. H. CHRISTIAN IN PHYS. REV. LETTERS VOL. 7, P. 387 (1963).

- 5) DATA ABOVE 650 KILOBARS DISCUSSED BY ALDER AND CHRISTIAN ARE OMITTED BECAUSE OF EXPECTED INTERFERENCE BY AN ELASTIC RELEASE WAVE CAUSED BY A VERY THIN PROJECTILE PLATE THAT WAS USED TO OBTAIN HIGHER PRESSURES.

TABLE I
CARBON (GRAPHITE, CEYLON-PRESSED)

23---4



23---5

CARBON (GRAPHITE, COMMERCIAL) (ATJ)

C

$V_0 = 0.568 \text{ CC/G}$ $C_L = 2.54 \text{ KM/SEC}$ $C_0 = 1.68 \text{ KM/SEC}$
 $V_{01} = 0.4412 \text{ CC/G}$ $C_S = 1.65 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS
AND DENSITY IN G/CC. UFS OBTAINED ON 3MM SAMPLES ONLY.

TABLE

SAMPLE							BASEPLATE
RHO0	US	UFS	UP	P	V/V0	D	P
1.673	5.05		2.00	171	0.604	6	296
1.673	5.09	3.22				3	296
1.685	5.44		2.23	202	0.590	6	351
1.685	5.35	3.56				3	351
1.761	7.13		3.76	472	0.603	6	738
1.761		7.08				3	738
1.775	6.39		3.35	374	0.610	6	612
1.755		5.5				3	612
1.739	5.64		2.26	217	0.599	6	356
1.739	5.78	3.62				3	356
1.796	5.30		1.78	169	0.664	6	268
1.796	5.25	3.09				3	268
1.802	7.61	5.96	4.55	617	0.402	3	958
1.812	6.68		3.22	319	0.521	6	590
1.812	6.51	5.18				3	590

$US = 3.32 + 0.99 \cdot UP \text{ KM/SEC}$

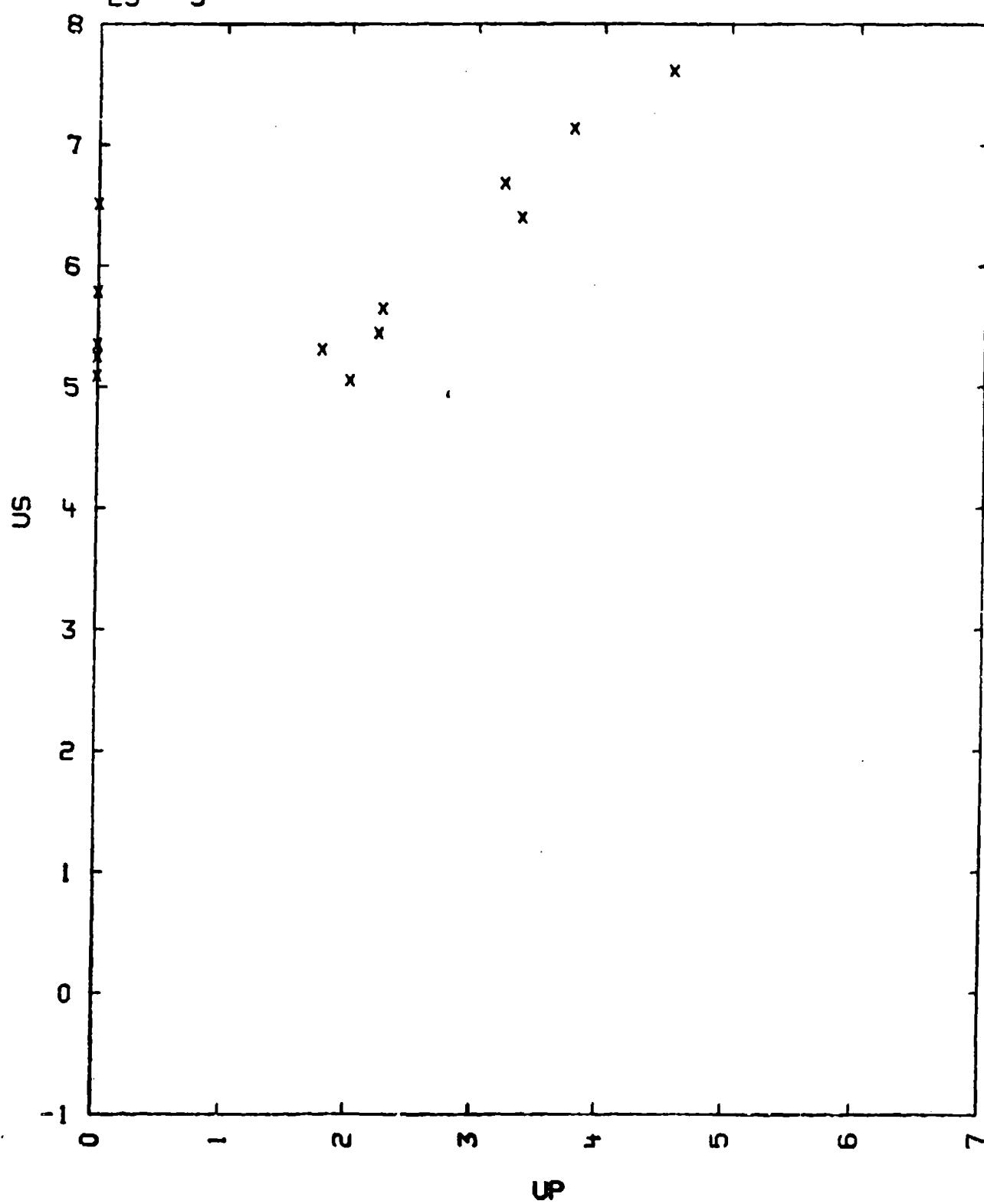
SIG.US = 0.21 KM/SEC.

COMMENTS:

- 1) SOURCE: COMPILER
L. R. L. EQUATION OF STATE FILE
LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B (ALUMINUM STANDARD BASE PLATE).
DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF VOI WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL UNIT CELL:
 $A = 2.4612 \text{ ANGSTROMS}$, $C = 6.7079 \text{ ANGSTROMS}$. A.C.A. MONOGRAPH NO.5
CRYSTAL DATA DETERMINATIVE TABLES, 2ND EDITION
AMERICAN CRYSTALLOGRAPHIC ASSOCIATION (1963).
- 4) SOUND VELOCITY WAS MEASURED BY H. L. DUNEGAN, SUPPORT ENGINEERING
ULTRASONICS GROUP, LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIF.
- 5) THE ABOVE DATA AND DATA FOR CEYLON GRAPHITE (23---4), HAVE BEEN
DISCUSSED BY B. J. ALDER AND R. H. CHRISTIAN IN PHYS. REV. LETTERS

- VOL. 7, P. 367 (1963).
- 6) DATA ABOVE 530 KILOBARS DISCUSSED BY ALDER AND CHRISTIAN ARE OMITTED BECAUSE OF EXPECTED INTERFERENCE BY AN ELASTIC RELEASE WAVE CAUSED BY A VERY THIN PROJECTILE PLATE THAT WAS USED TO OBTAIN HIGHER PRESSURES.

TABLE I
CARBON (GRAPHITE, COMMERCIAL) (ATJ)
23---5



23--6
CARBON, (GRAPHITE PYROLYTIC)

C

$$V_0 = 0.455 \text{ CC/G} \quad C_L = 3.45 \text{ KM/SEC}$$

$$V_{01} = 0.4412 \text{ CC/G} \quad C_B = 3.9 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD
RHO0	US	UP	P	V/V0	US(ST)
2.20	3.90	0.	0.	1.0000	0.
2.20	5.18	0.37	43.	0.9286	5.88
2.20	5.17	0.37	43.	0.9284	5.88
2.20	5.14	0.37	42.	0.9280	5.88
2.20	6.17	0.83	114.	0.8655	6.44
2.20	6.05	0.84	112.	0.8612	6.44
2.20	6.01	0.84	112.	0.8602	6.44
2.23	6.19	0.99	137.	0.8401	6.62
2.20	6.41	1.01	142.	0.8424	6.64
2.22	6.76	1.30	196.	0.8077	7.00
2.22	6.78	1.30	196.	0.8083	7.00
2.20	6.75	1.33	197.	0.8030	7.02
2.20	6.72	1.33	197.	0.8021	7.02
2.20	6.66	1.33	195.	0.8003	7.02
2.20	7.01	1.46	225.	0.7917	7.18
2.20	6.97	1.46	224.	0.7905	7.18
2.20	6.97	1.46	224.	0.7905	7.18
2.22	7.64	1.81	308.	0.7631	7.64
2.20	7.64	1.82	306.	0.7618	7.63
2.20	7.57	1.83	305.	0.7583	7.64
2.20	7.55	1.83	304.	0.7576	7.64
2.20	7.50	1.83	303.	0.7560	7.64
2.23	7.85	2.00	351.	0.7452	7.87
2.21	8.02	2.16	383.	0.7307	8.05
2.20	8.28	2.30	420.	0.7222	8.23
2.20	8.24	2.31	419.	0.7197	8.23
2.20	8.19	2.31	417.	0.7179	8.23
2.20	8.29	2.33	425.	0.7189	8.26
2.20	8.24	2.33	424.	0.7172	8.26
2.20	8.21	2.34	423.	0.7150	8.26
2.22	8.11	2.42	436.	0.7016	8.34
2.21	8.18	2.49	451.	0.6956	8.42
2.21	8.36	2.58	478.	0.6914	8.54
2.20	8.33	2.81	515.	0.6627	8.77
2.20	8.26	2.82	513.	0.6586	8.77
2.20	8.24	2.82	512.	0.6578	8.77
2.23	8.28	3.17	585.	0.6171	9.14
2.22	8.06	3.22	577.	0.6009	9.16
2.21	8.21	3.37	612.	0.5895	9.33
2.20	8.43	3.46	642.	0.5896	9.45

CARBON, (GRAPHITE PYROLYTIC)

RHO0	US	UP	P	V/V0	US(ST)
2.20	8.38	3.46	640.	0.5871	9.45
2.20	8.36	3.47	639.	0.5849	9.45
2.20	8.37	3.55	652.	0.5759	9.52
2.21	8.57	3.58	676.	0.5923	9.59
2.20	8.56	3.65	687.	0.5736	9.66
2.23	9.13	3.68	749.	0.5969	9.79
2.21	8.86	3.68	720.	0.5847	9.74
2.21	9.17	3.69	747.	0.5976	9.79
2.22	9.17	3.72	759.	0.5943	9.84
2.20	9.07	3.76	751.	0.5854	9.85
2.24	9.21	3.77	777.	0.5907	9.90
2.21	9.47	3.95	828.	0.5829	10.11
2.23	9.41	3.96	831.	0.5792	10.13

US = $4.74 + 1.53 \cdot UP$ KM/SEC .8 < UP < 2.3

SIG US = 0.067 KM/SEC

US = $3.9 + 2.3 \cdot UP$ KM/SEC 3.25 < UP < 4.

SIG US = 0.22 KM/SEC

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
HUGONIOTS OF GRAPHITES OF VARIOUS INITIAL DENSITIES AND THE EQUATION OF STATE OF CARBON.
SYMPOSIUM ON THE 'BEHAVIOR OF DENSE MEDIA UNDER HIGH DYNAMIC PRESSURE', I.U.T.A.M. PARIS, 1967
(GORDON AND BREACH, N.Y. 1968)
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION METHOD B.
STANDARD MATERIAL 2024 ALUMINUM WITH LINEAR SHOCK VELOCITY-PARTICLE VELOCITY HUGONIOT RELATIONSHIP $US = 5.46 + 1.318 \cdot UP$ KM/SEC
 $\rho_{H0} = 2.785$ G/CC, GRUNEISEN RATIO = $1.89 = V \cdot (DP/DE)$
- 3) THE VALUE OF V01 WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS:
 $A = 2.4612$ ANGSTROMS, $C = 6.7079$ ANGSTROMS: A.C.A. MONOGRAPH NO.5 CRYSTAL DATA DETERMINATIVE TABLES, 2ND EDITION,
AMERICAN CRYSTALLOGRAPHIC ASSOCIATION (1963).
- 4) THE GRUNEISEN RATIO OF THE LOW PRESSURE PHASE = $1.71/\rho_0$ YIELDS A GOOD FIT TO THE POROUS DATA OF 23---7 AND WITH A PYROLYTIC EOS CENTERED ON THE IDEAL DENSITY.

TABLE I

CARBON, (GRAPHITE PYROLYTIC)

23---6

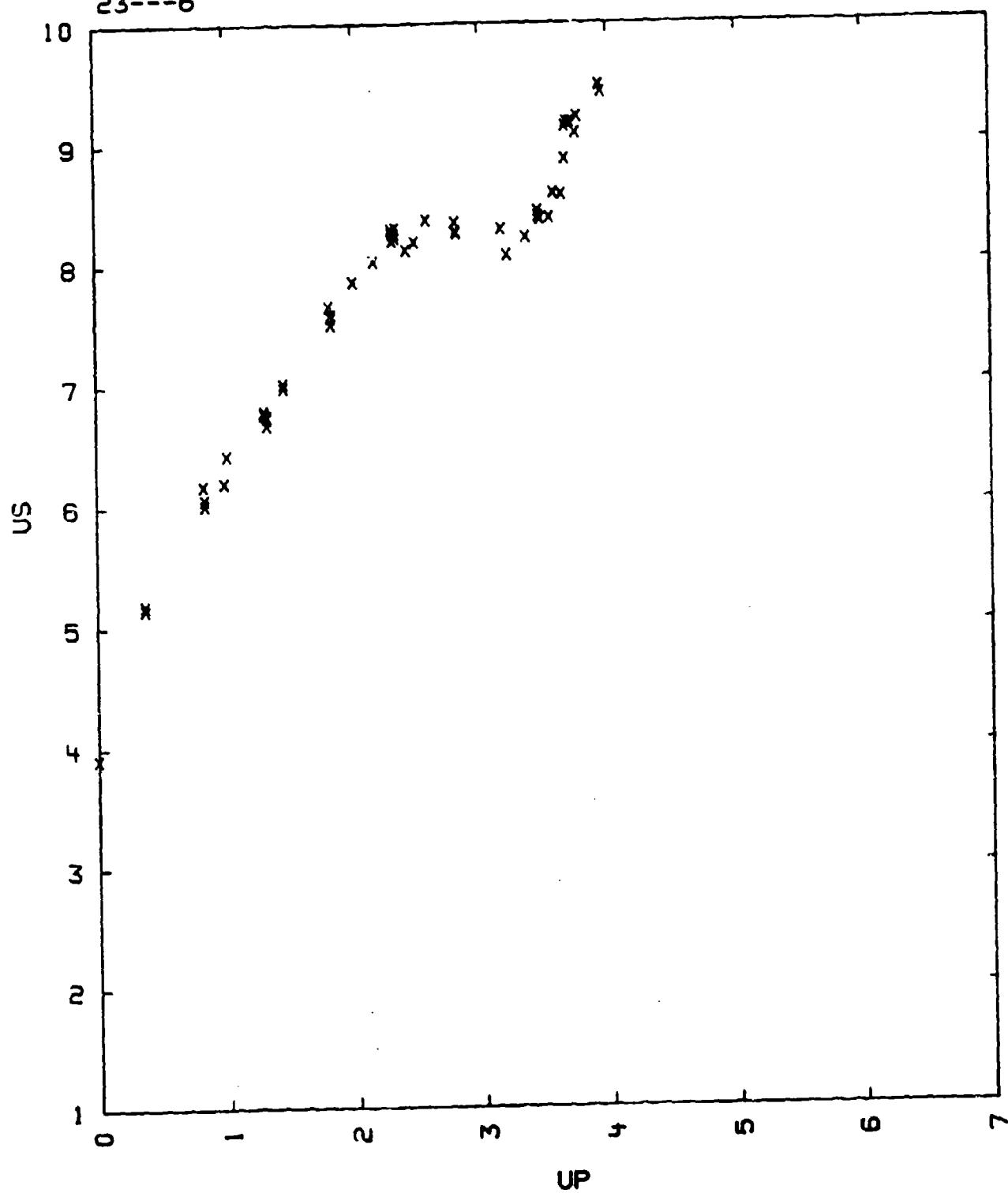
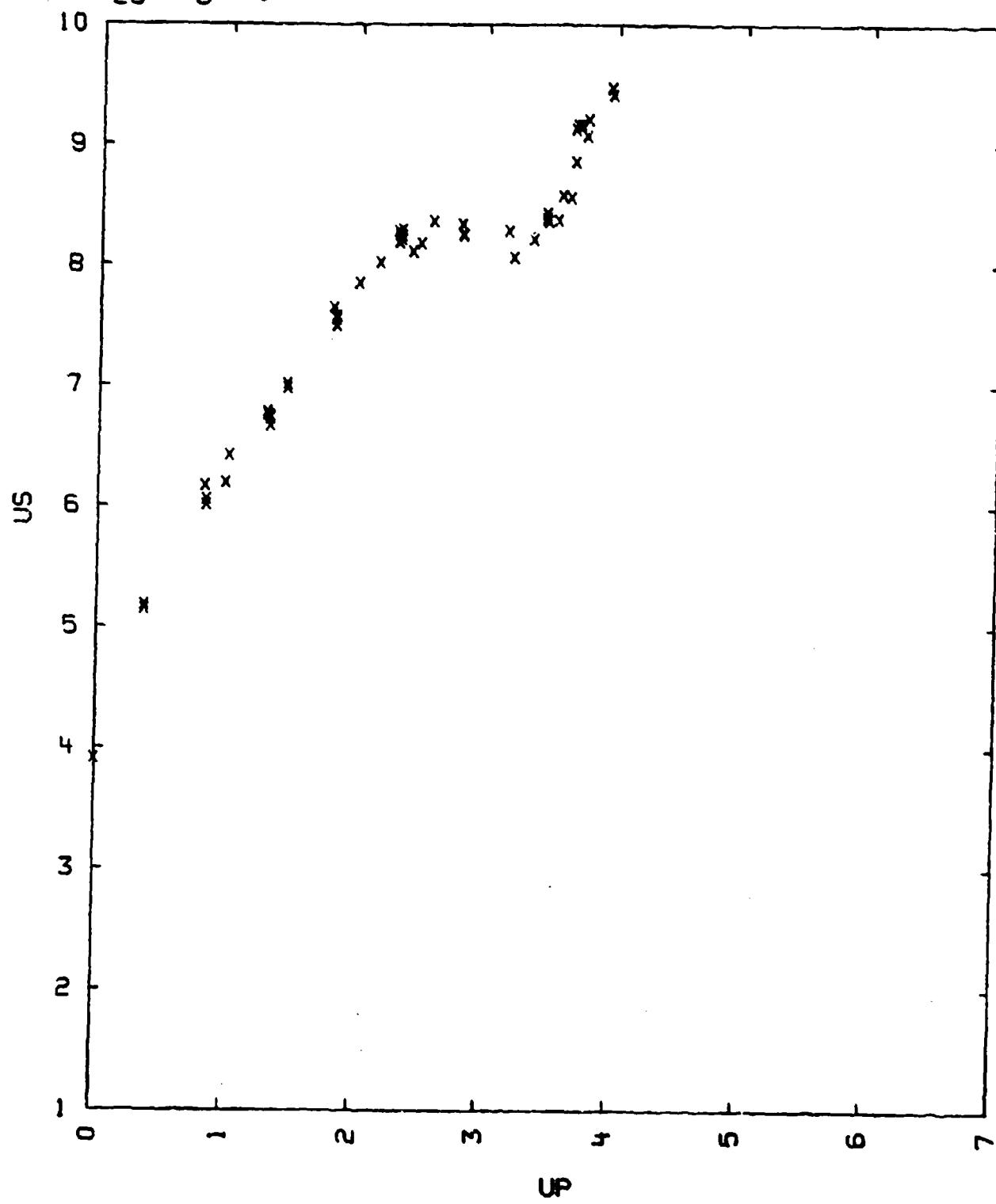


TABLE I

CARBON, (GRAPHITE PYROLYTIC)
23---6



23---7

CARBON (GRAPHITE POROUS)

C

$$V_0 = 1/\rho_{H0} \text{ CC/G}$$

$$V_{01} = 0.4412 \text{ CC/G}$$

IN THE TABLES BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE I

SAMPLE					STANDARD
RHO0	US	UP	P	V/V0	US(ST)
2.11	5.31	1.01	113.	0.8098	6.55
2.12	6.10	1.37	178.	0.7754	6.99
2.12	6.41	1.98	270.	0.6911	7.65
2.14	6.65	2.64	376.	0.6030	8.34
2.14	6.80	2.81	410.	0.5868	8.54
2.15	8.08	3.42	594.	0.5767	9.33
2.14	8.89	3.81	726.	0.5714	9.84
2.13	9.96	3.87	740.	0.5681	9.90
2.13	9.34	4.02	802.	0.5696	10.11
2.14	9.77	4.22	880.	0.5681	10.38
2.14	9.70	4.37	907.	0.5495	10.53

$$US = 3.1 + 2.2 \cdot UP \text{ KM/SEC} \quad 1.0 < UP < 1.4$$

$$\text{SIG US} = ?$$

$$US = 1.30 + 1.98 \cdot UP \text{ KM/SEC} \quad 2.7 < UP < 4.5$$

$$\text{SIG US} = 0.23 \text{ KM/SEC}$$

TABLE II

SAMPLE					STANDARD
RHO0	US	UP	P	V/V0	US(ST)
2.04	3.48	0.35	24.	0.8994	5.79
2.03	3.51	0.37	26.	0.8946	5.81
2.04	4.41	0.90	81.	0.7959	6.37
2.03	4.41	0.91	81.	0.7937	6.37
2.02	4.91	1.06	106.	0.7841	6.56
2.04	4.94	1.07	108.	0.7834	6.57
2.01	5.58	1.39	156.	0.7509	6.34
2.03	5.62	1.40	160.	0.7509	6.96
2.04	6.10	1.82	226.	0.7016	7.42
2.03	6.11	1.82	225.	0.7021	7.42
2.01	6.11	1.84	227.	0.6989	7.44
2.04	6.14	1.85	232.	0.6987	7.46
2.04	6.11	1.86	231.	0.6956	7.46
2.01	6.15	1.80	235.	0.8911	7.50

CARBON (GRAPHITE POROUS)

RHO0	US	UP	P	V/VO	US(ST)
2.01	6.96	2.97	417.	0.5733	8.66
2.00	6.98	2.98	416.	0.5731	8.66
2.03	7.30	3.19	472.	0.5630	8.92
2.02	7.28	3.20	469.	0.5604	8.92
2.03	7.68	3.44	536.	0.5521	9.22
2.01	7.70	3.44	533.	0.5532	9.22
2.03	7.93	3.71	598.	0.5322	9.53

$$US = 2.72 + 2.02 \cdot UP \text{ KM/SEC} \quad 0.3 < UP < 1.6$$

$$\text{SIG US} = 0.20 \text{ KM/SEC}$$

$$US = 2.87 + 1.38 \cdot UP \text{ KM/SEC}$$

$$\text{SIG US} = 0.12 \text{ KM/SEC} \quad 2.6 < UP < 4.$$

TABLE III

SAMPLE					STANDARD
RHO0	US	UP	P	V/VO	US(ST)
1.96	4.15	0.92	75.	0.7783	6.36
1.94	4.45	1.10	95.	0.7528	6.55
1.95	5.36	1.48	155.	0.7239	6.99
1.96	5.94	1.83	214.	0.6919	7.40
1.95	5.90	1.84	212.	0.6881	7.40
1.95	6.08	2.08	246.	0.6579	7.65
1.95	6.13	2.20	263.	0.6411	7.77
1.94	6.11	2.21	262.	0.6383	7.77
1.95	6.23	2.37	288.	0.6196	7.95
1.96	6.51	2.74	340.	0.5791	8.34
1.95	6.42	2.79	348.	0.5654	8.37
1.94	6.38	2.80	346.	0.5611	8.37
1.94	6.53	2.95	373.	0.5482	8.54
1.95	6.97	3.15	428.	0.5481	8.80
1.95	7.03	3.24	443.	0.5391	8.89
1.94	7.01	3.24	441.	0.5378	8.89
1.94	7.44	3.55	513.	0.5228	9.24
1.93	7.46	3.55	511.	0.5241	9.24
1.94	7.60	3.61	534.	0.5250	9.33
1.95	8.50	3.99	664.	0.5306	9.84
1.95	8.52	4.06	674.	0.5235	9.90
1.96	8.91	4.21	734.	0.5275	10.11
1.95	8.78	4.23	726.	0.5159	10.13
1.95	9.43	4.55	838.	0.5175	10.53

$$US = 2.34 + 1.97 \cdot UP \text{ KM/SEC} \quad 0.9 < UP < 1.85$$

$$\text{SIG US} = 0.20 \text{ KM/SEC}$$

$$US = 1.50 + 1.72 \cdot UP \text{ KM/SEC}$$

$$\text{SIG US} = 0.24 \text{ KM/SEC} \quad 2.8 < UP < 4.6$$

TABLE IV

SAMPLE					STANDARD
RHO0	US	UP	P	V/V0	US(ST)
1.88	3.85	0.95	69.	0.7532	6.36
1.88	4.23	1.13	90.	0.7329	6.55
1.88	5.07	1.52	145.	0.7002	6.99
1.88	5.68	1.89	201.	0.6673	7.40
1.88	5.98	2.11	238.	0.6472	7.65
1.88	6.06	2.24	255.	0.6304	7.77
1.88	6.27	2.39	282.	0.6188	7.95
1.88	6.32	2.60	332.	0.5570	8.34
1.88	6.42	2.82	340.	0.5607	8.37
1.87	6.48	2.98	363.	0.5401	8.54
1.88	6.89	3.20	414.	0.5356	8.80
1.88	6.98	3.28	430.	0.5301	8.89
1.88	7.33	3.60	497.	0.5089	9.24
1.88	7.45	3.68	514.	0.5060	9.33
1.88	8.37	4.13	649.	0.5066	9.90
1.87	8.62	4.32	697.	0.4988	10.11
1.87	8.58	4.34	697.	0.4942	10.13
1.88	8.87	4.55	757.	0.4870	10.38
1.87	9.27	4.64	806.	0.4995	10.53

US = 2.00 + 1.97*UP KM/SEC

0.9<UP<1.9

SIG US = 0.19 KM/SEC

US = 1.73 + 1.59*UP KM/SEC

2.9<UP<4.7

SIG US = 0.26 KM/SEC

TABLE V

SAMPLE					STANDARD
RHO0	US	UP	P	V/V0	US(ST)
1.79	3.57	0.98	63.	0.7255	6.36
1.76	3.83	1.18	79.	0.6919	6.55
1.77	4.15	1.26	92.	0.6964	6.64
1.77	4.69	1.59	132.	0.6610	6.99
1.73	4.75	1.60	132.	0.6632	7.00
1.76	4.77	1.65	139.	0.6541	6.87
1.76	4.74	1.65	138.	0.6519	6.87
1.81	5.44	1.94	190.	0.6434	7.40
1.77	5.77	2.16	221.	0.6256	7.63
1.79	5.70	2.18	223.	0.6175	7.65
1.76	5.64	2.23	221.	0.6046	7.52
1.76	5.62	2.23	220.	0.6032	7.52
1.78	5.78	2.31	238.	0.6003	7.77
1.80	6.03	2.45	267.	0.5937	7.95
1.79	6.07	2.59	282.	0.5733	8.08
1.77	6.25	2.67	294.	0.5720	8.03
1.77	6.04	2.69	288.	0.5546	8.03
1.78	6.13	2.86	311.	0.5334	8.33
1.77	6.12	2.86	211.	0.5327	8.33

CARBON (GRAPHITE POROUS)

RHO0	US	UP	P	V/V0	US(ST)
1.75	6.20	2.88	312.	0.5395	8.34
1.78	6.23	2.89	320.	0.5361	8.37
1.75	6.32	2.94	325.	0.5348	8.42
1.75	6.28	3.08	338.	0.5096	8.54
1.77	6.69	3.12	370.	0.5336	8.54
1.77	6.61	3.13	367.	0.5265	8.54
1.79	6.72	3.27	394.	0.5134	8.80
1.81	6.78	3.35	411.	0.5059	8.89
1.76	6.53	3.40	390.	0.4793	8.79
1.76	6.49	3.40	388.	0.4761	8.79
1.76	7.14	3.60	453.	0.4958	9.14
1.79	7.17	3.68	472.	0.4868	9.24
1.76	7.24	3.78	481.	0.4779	9.33
1.76	7.90	4.12	574.	0.4785	9.74
1.76	8.15	4.14	593.	0.4920	9.79
1.74	8.09	4.21	592.	0.4796	9.84
1.75	8.28	4.24	615.	0.4879	9.90
1.74	8.39	4.45	651.	0.4696	10.11
1.74	8.45	4.46	655.	0.4722	10.13
1.75	8.82	4.65	720.	0.4728	10.38
1.76	9.10	4.76	763.	0.4769	10.53

US = $1.68 + 1.90 \cdot UP$ KM/SEC .9 < UP < 2.18

SIG US = 0.20 KM/SEC

US = $1.70 + 1.51 \cdot UP$ KM/SEC 3.05 < UP < 5.

SIG US = 0.5 KM/SEC

TABLE VI

SAMPLE					STANDARD
RHO0	US	UP	P	V/V0	US(ST)
1.54	2.76	0.96	41.	0.6522	6.26
1.56	4.39	1.75	119.	0.6014	7.05
1.55	4.31	1.76	117.	0.5916	7.05
1.55	5.25	2.34	190.	0.5543	7.66
1.54	5.24	2.34	190.	0.5534	7.66
1.54	5.27	2.38	194.	0.5484	7.69
1.55	6.17	3.20	307.	0.4814	8.54
1.54	6.16	3.21	304.	0.4789	8.54
1.55	6.24	3.41	328.	0.4535	8.73
1.54	6.25	3.41	328.	0.4544	8.73
1.53	7.29	4.28	476.	0.4129	9.64
1.52	7.22	4.29	471.	0.4058	9.64

US = $0.842 + 2.00 \cdot UP$ KM/SEC 0.8 < UP < 2.15

SIG US = 0.25 KM/SEC

COMMENTS:

1) SOURCE: MCQUEEN, R.O. AND MARSH, S.P.

HUGONIOTS OF GRAPHITES OF VARIOUS INITIAL DENSITIES AND THE
EQUATION OF STATE OF CARBON
SYMPOSIUM ON THE 'BEHAVIOR OF DENSE MEDIA UNDER
HIGH DYNAMIC PRESSURE', I.U.T.A.M., PARIS, 1967
(GORDON AND BREACH, N.Y. 1968)

- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION METHOD B.
STANDARD MATERIAL 2024 ALUMINUM WITH LINEAR SHOCK VELOCITY-PARTICLE
VELOCITY HUGONIOT RELATIONSHIP $VS = 5.46 + 1.31B^{1/2}$ KM/SEC.
 $\rho_{H0} = 2.785 \text{ g/cc}$, GRUNEISEN GAMMA = 1.89
- 3) THE VALUE OF ρ_{H0} WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS:
 $A = 2.4612 \text{ ANGSTROMS}$, $C = 6.7078 \text{ ANGSTROMS}$, A.C.A. MONOGRAPH NO.5
AMERICAN CRYSTALLOGRAPHIC ASSOCIATION (1963).
- 4) THE ABOVE DATA WERE DISCUSSED BY R. G. MCQUEEN IN METALLURGICAL
SOCIETY CONFERENCES, (GORDON AND BREACH SCIENCE PUBLISHERS N. Y. 1963
VOL. 22, CHAPTER 3.
- 5) MATERIAL OF TABLE I
PRESSED GRAPHITE
- - - III PRESSED GRAPHITE
- - - III UNION CARBIDE CO. GRADE ZTA: HOT WORKED GRAPHI-
TIZED TAR (99.8 WT. PERCENT C)
- - - IV PRESSED GRAPHITE
- - - V UNION CARBIDE CO. GRADE ATJ: GRAPHITIZED TAR
(99.8 PERCENT C)
- - - VI UNION CARBIDE CO. GRADE PT 0178: CELLULOSE
FIBRE MIXTURE.

TABLE I

CARBON (GRAPHITE POROUS)
23---7

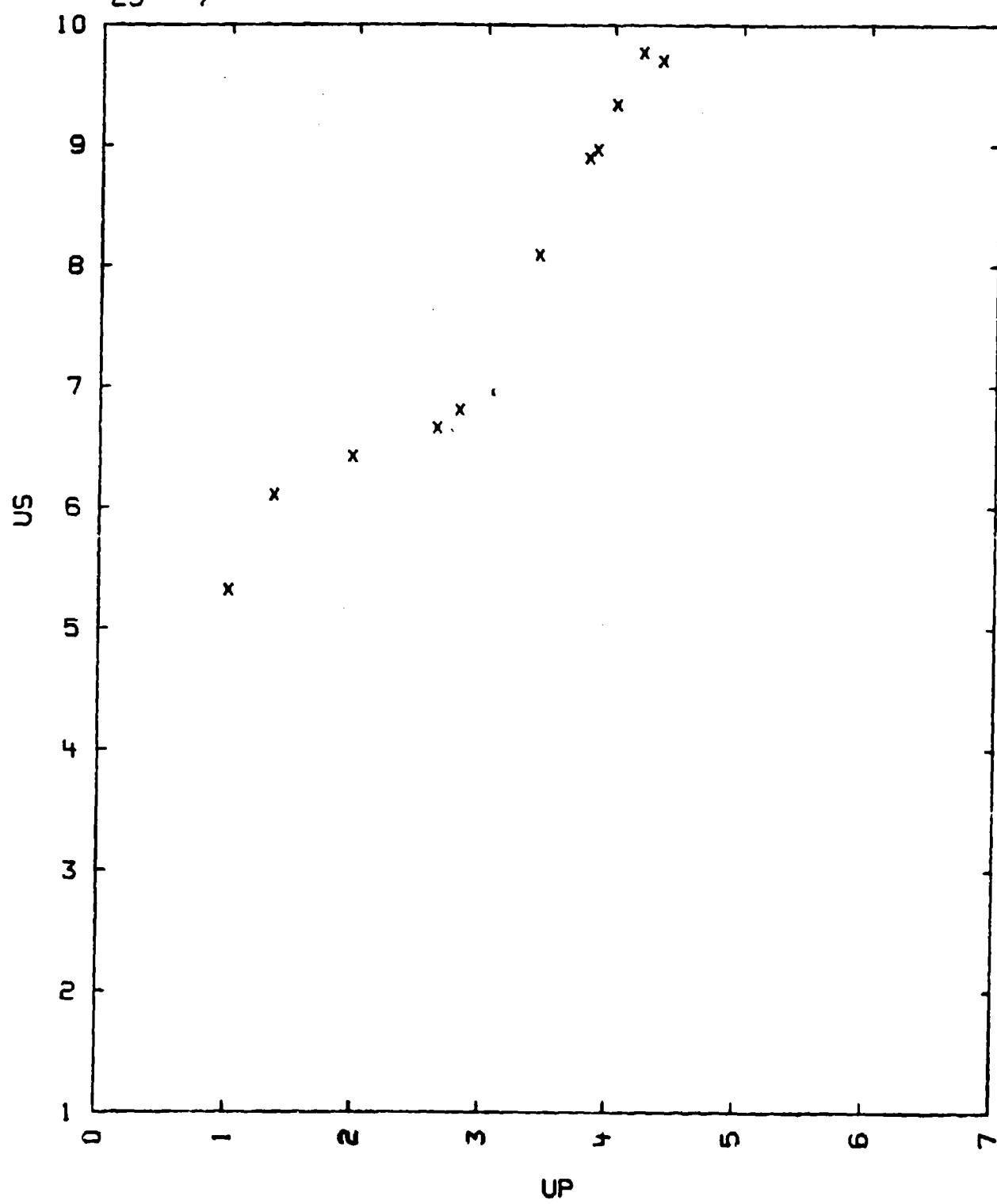


TABLE II

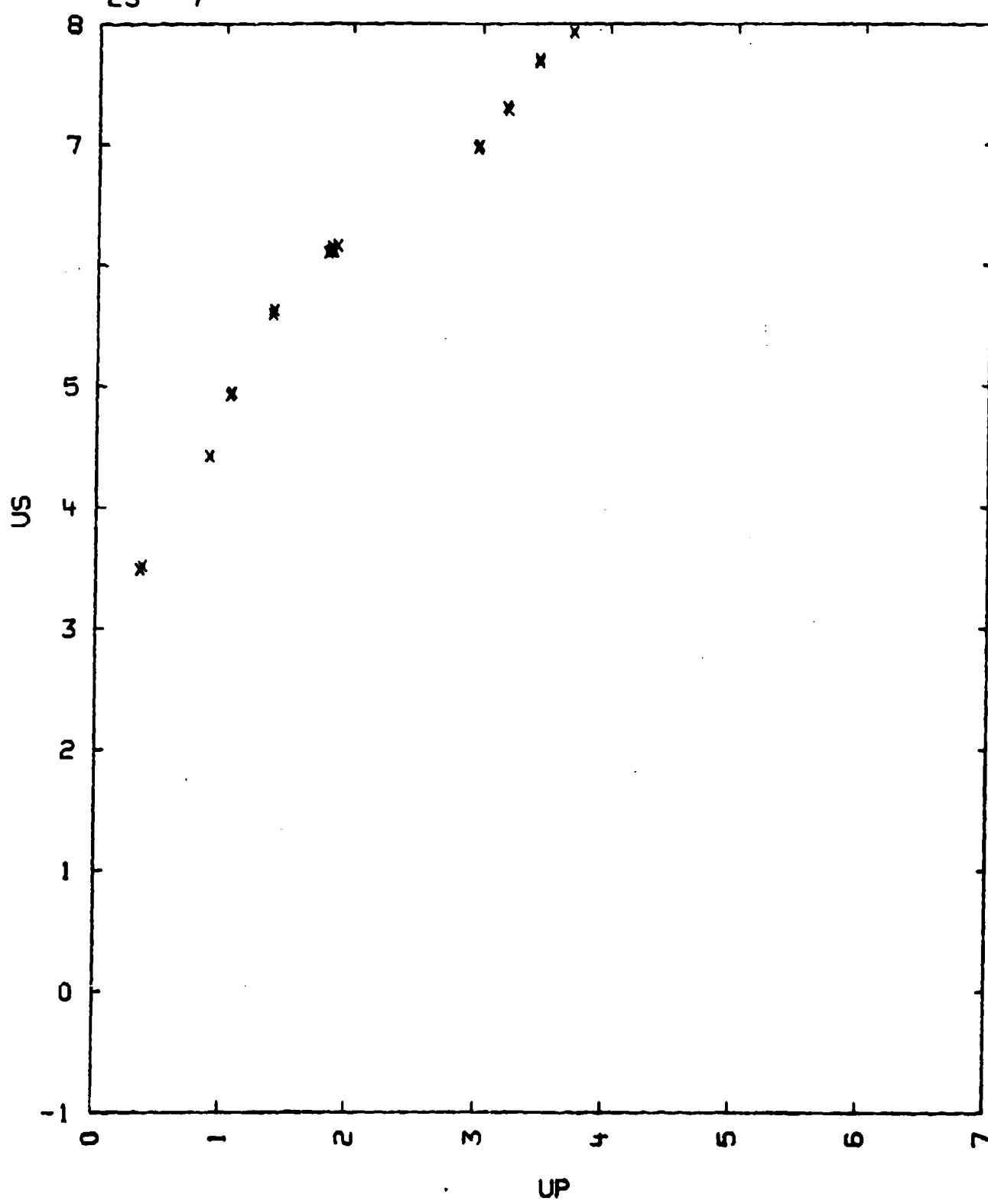
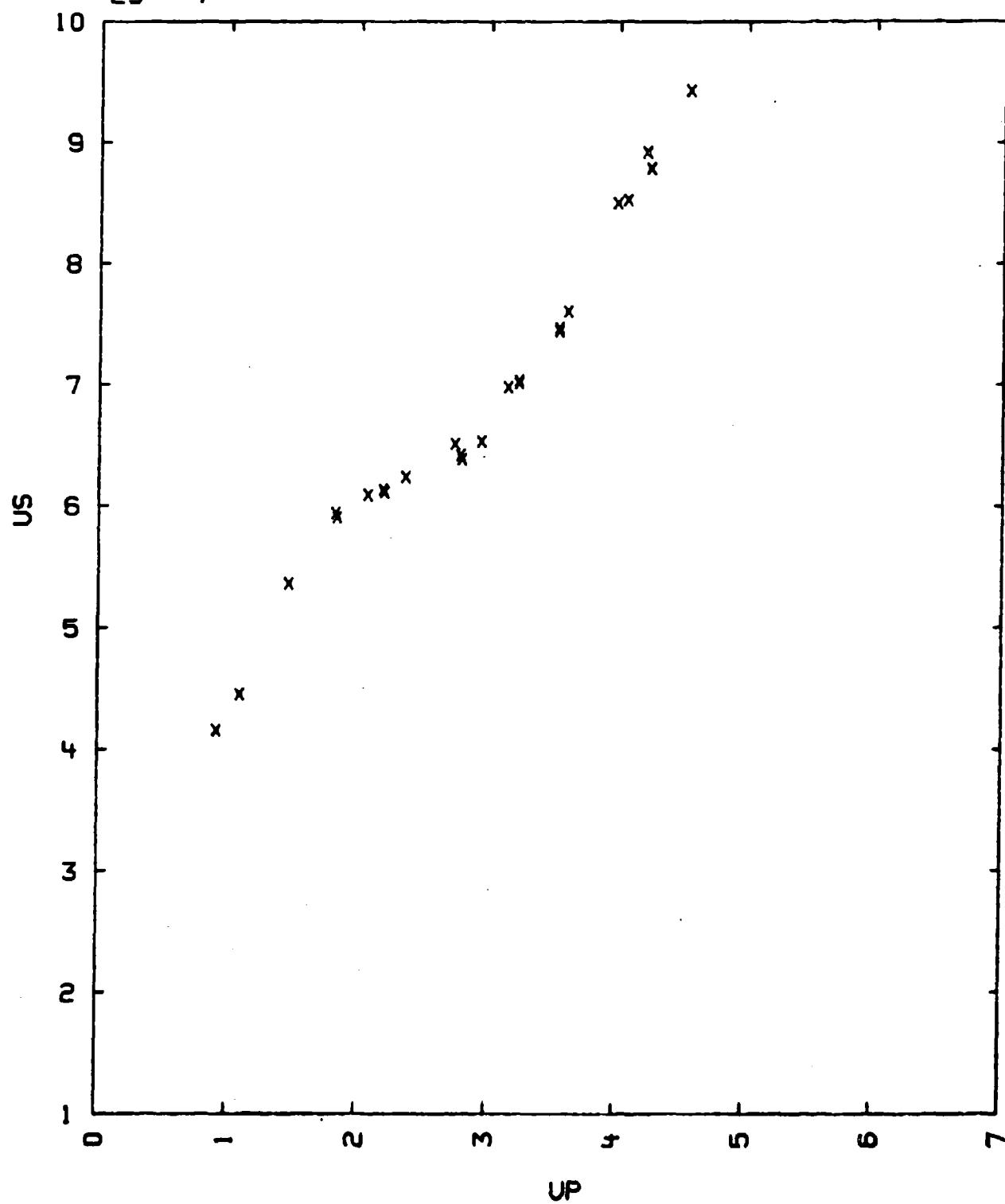
CARBON (GRAPHITE POROUS)
23---7

TABLE III

CARBON (GRAPHITE POROUS)
23---7



CARBON (GRAPHITE POROUS)
23---7

TABLE IV

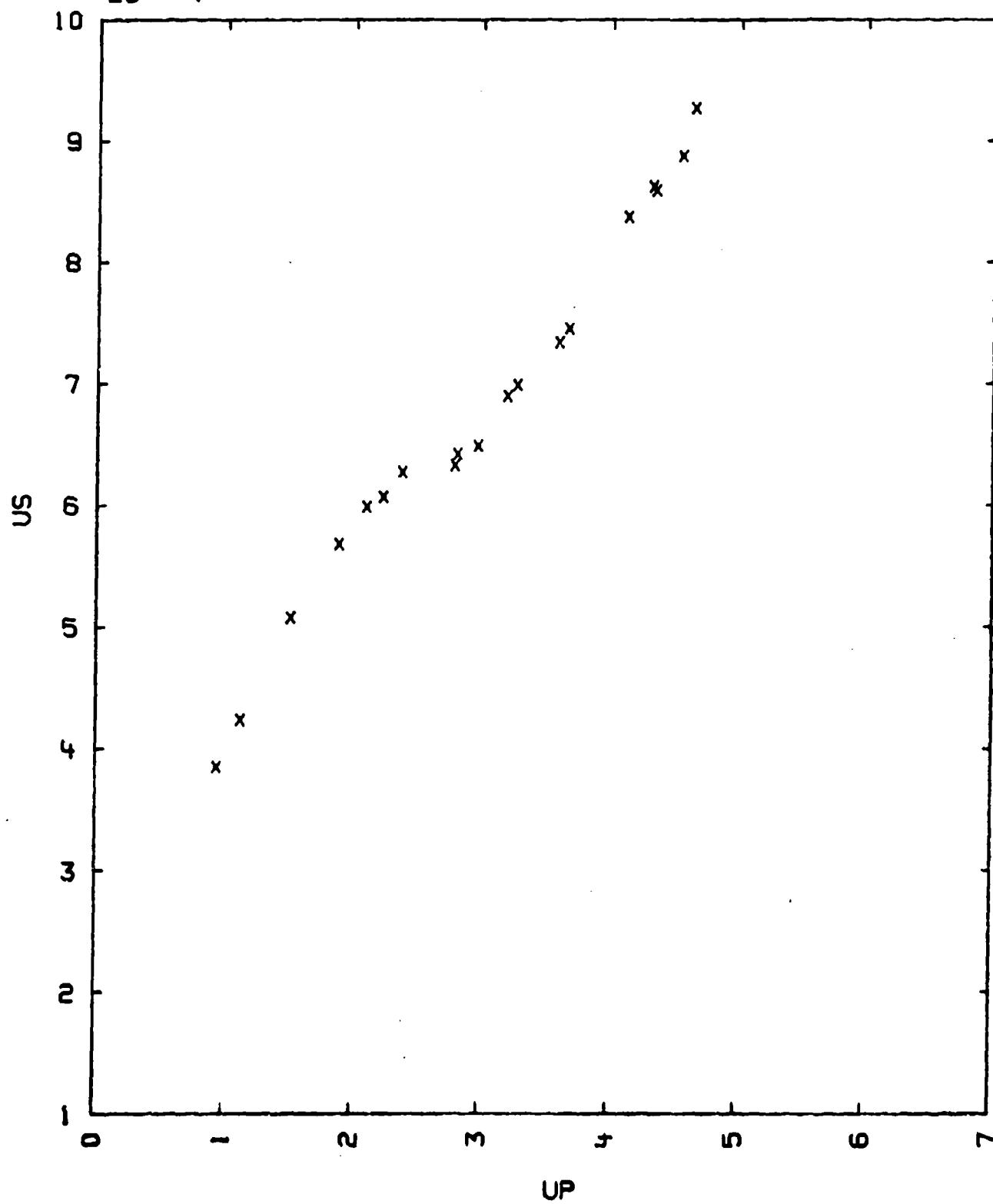


TABLE V

CARBON (GRAPHITE POROUS)

23---7

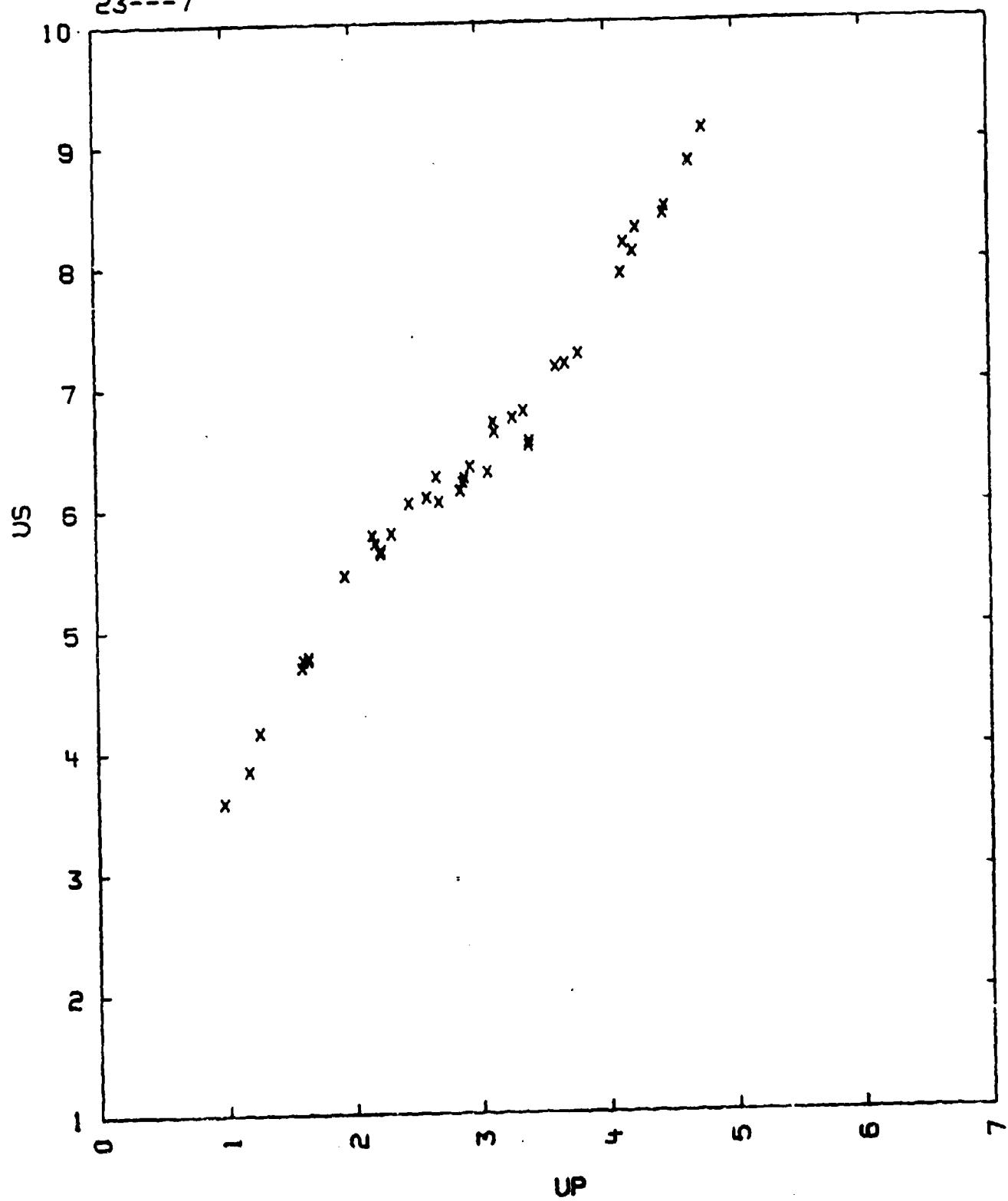
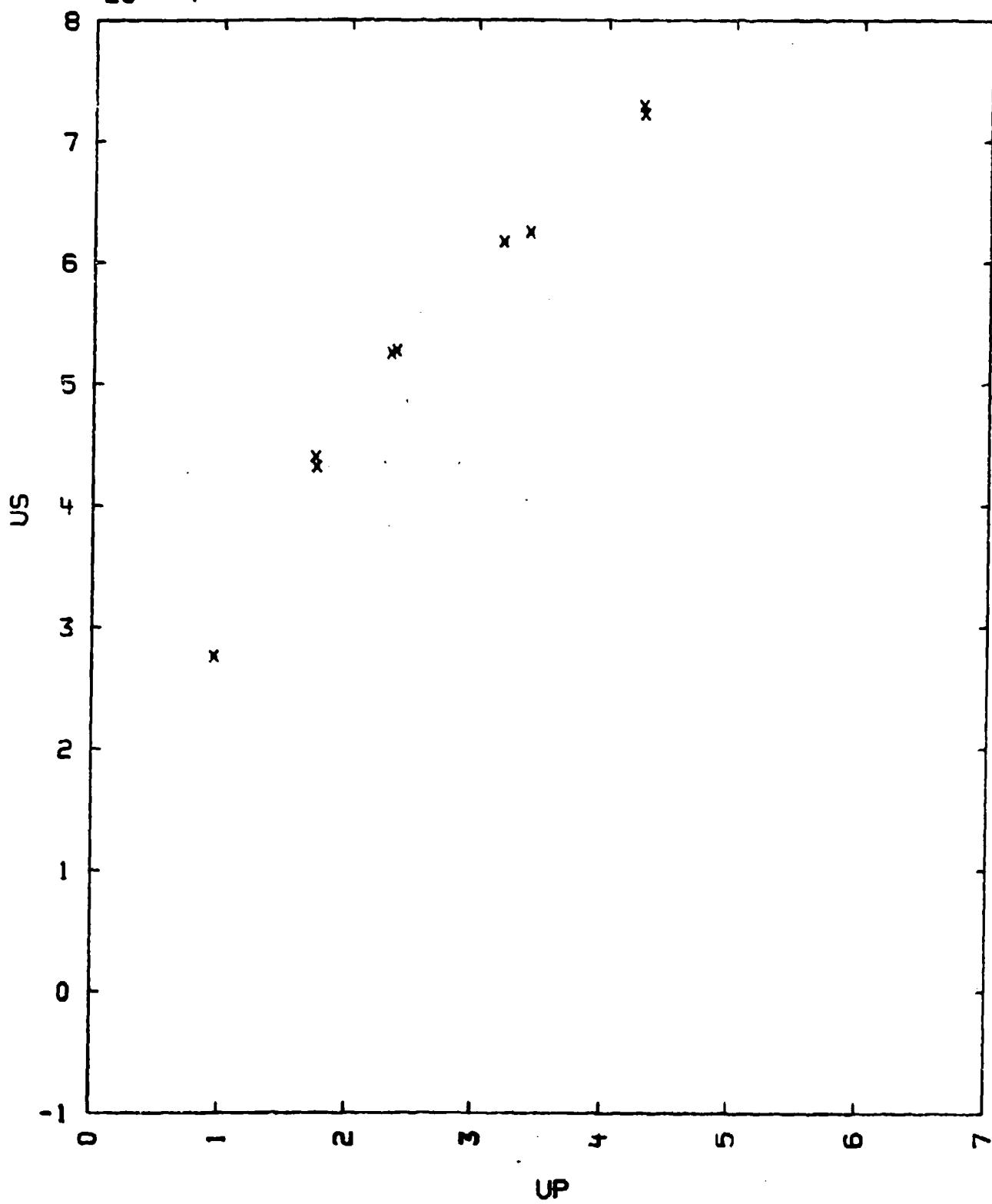


TABLE VI

CARBON (GRAPHITE POROUS)
23---7



23---8

CARBON (DIAMOND PRESSED)

C:		99+ WT. PERCENT.
DIAMOND	95	- -
GRAPHITE	5	- -
POROSITY	6	VOL. -

$$V_0 = 0.314 \text{ CC/G}$$

$$V_{OI} = 0.284404 \text{ CC/G}$$

$$C_0 = 11.22 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD
RHO0	US	UP	P	V/V0	US(ST)
3.20	9.91	1.37	434.	0.9618	7.65
3.17	12.10	2.94	1128.	0.7570	9.93
3.20	12.49	3.19	1275.	0.7446	10.31

$$US = 7.976 + 1.409 \cdot UP \text{ KM/SEC}$$

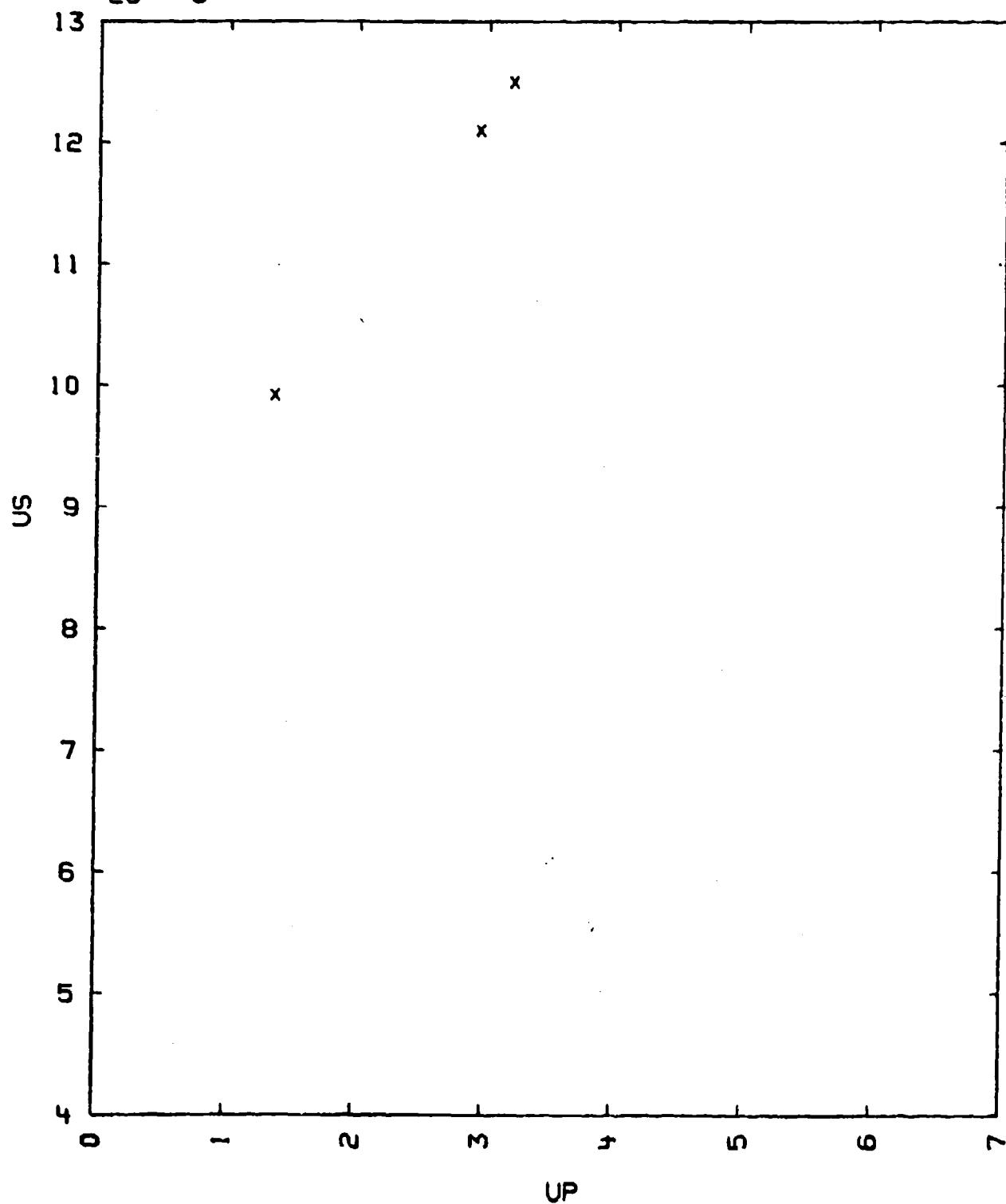
$$\text{SIG US} = 0.027 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
HUGONIOTS OF GRAPHITES OF VARIOUS INITIAL DENSITIES AND THE EQUATION OF STATE OF CARBON
IVTAM SYMPOSIUM: BEHAVIOR OF DENSE MEDIA UNDER HIGH DYNAMIC PRESSURES. (PARIS '67) (GORDON AND BREACH, NEW YORK 1968)
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION METHOD B.
STANDARD MATERIAL 2024 ALUMINUM WITH LINEAR SHOCK VELOCITY-PARTICLE VELOCITY HUGONIOT RELATIONSHIP $US=5.46+1.318 \cdot UP \text{ KM/SEC}$, $RHO0=2.785 \text{ G/CC}$, GRUNEISEN GAMMA = 1.89
- 3) THE VALUE OF V_{OI} WAS OBTAINED FROM HYCKOFF, CRYSTAL STRUCTURES, (INTERSCIENCE PUBL., N. Y. 1963) 2ND ED. VOL. 1: A = 3.56679 ANGSTR.
- 4) THE DIAMOND SAMPLES WERE PRESSED WITH GRAPHITE BINDER (CIRCA 5 WT. PERCENT) INTO 9 TO 12 MM DIAMETER DISCS.
- 5) $C_0 = ((C_{11}) + 2(C_{12})) / (3 \cdot RHO0)^{0.5}$ WITH $C_{11} = 10.79 \pm 0.05 \text{ MBAR}$
 $T = 25 \text{ DEG. C.}$, $RHO0 = 3.512 \text{ G/CC.}$, $C_{12} = 1.24 \pm 0.05$ -
 H.J. MCSKIMMIN AND P. ANDREATCH,J.R. PHYS. REV. 43, 2944 (1972)

TABLE I

CARBON (DIAMOND PRESSED)
23---8



25---1
GERMANIUM

OE

$$V_0 = 0.1885 \text{ CC/G}$$

$$V_{01} = 0.1878 \text{ CC/G}$$

$$C_0 = 3.77 \text{ KM/SEC}$$

$$C_B = 3.82 \text{ KM/SEC}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC.
AND PRESSURE IN KILOBARS.

TABLE

SAMPLE						BASE PLATE PRESSURE
RHO0	US	UFS	UP	P	V/V0	
5.324	3.81	1.32	0.73	148	0.808	139
5.317	4.43	2.93	1.63	384	0.632	363
5.239	4.38	2.84	1.56	358	0.645	341
5.308	3.87	1.75	0.97	199	0.749	192
5.327	4.71	3.25	1.72	430	0.635	398
5.312	5.82	4.54	2.34	723	0.598	632
5.317	6.28	5.12	2.62	875	0.583	756

$$US = 1.52 + 1.83 UP \text{ KM/SEC} \quad (\text{FOR UP FROM } 1.56 \text{ TO } 2.62 \text{ KM/SEC})$$

$$\Sigma US = 0.053 \text{ KM/SEC}$$

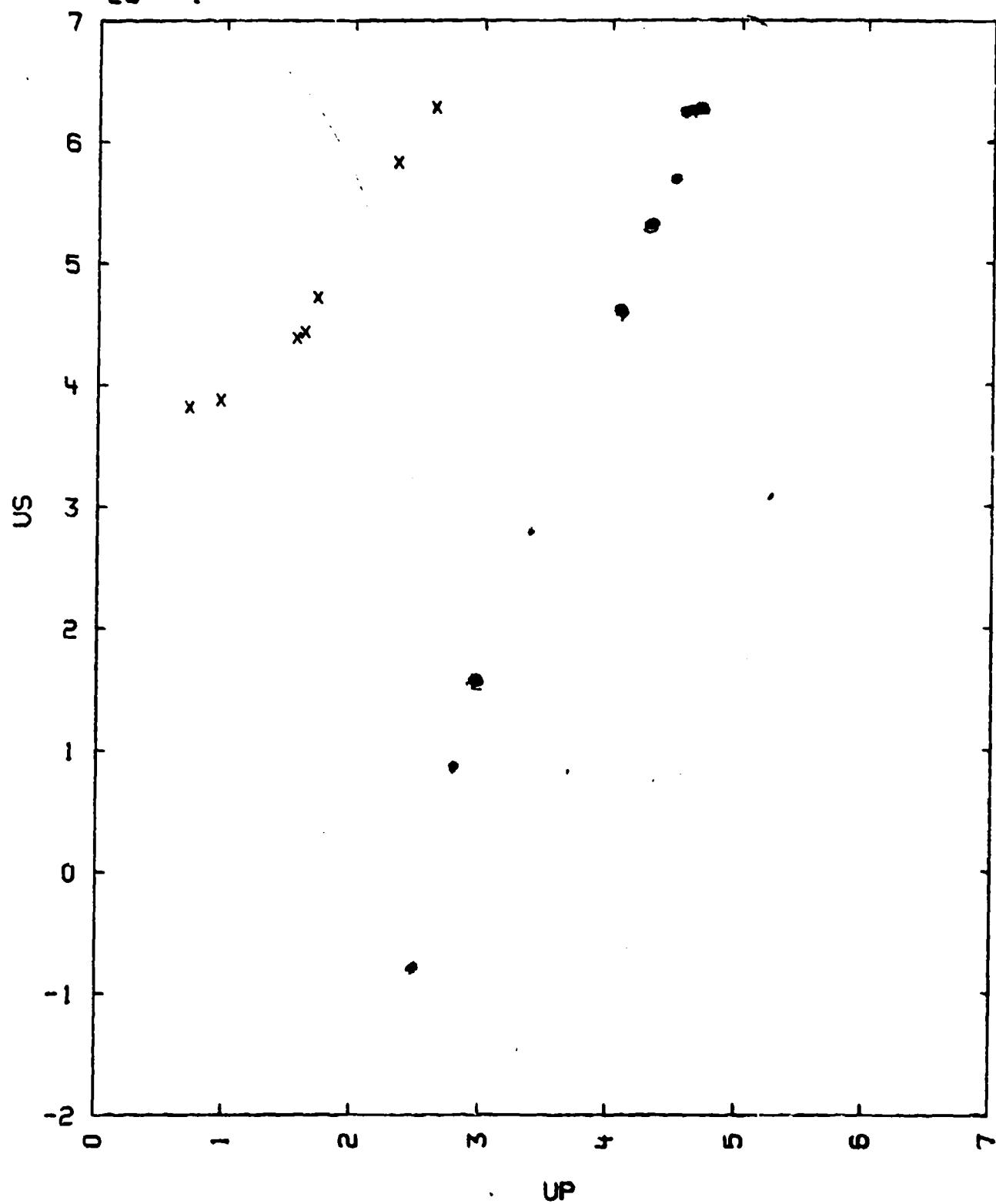
COMMENTS:

- 1) SOURCE: COMPILER
L.R.L. EQUATION OF STATE FILE
LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B. (ALUMINUM STANDARD BASE PLATE).
DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF V01 WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CHRISTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES
(DOVER PUBLICATIONS, INC., NEW YORK, N. Y., 1963).
- 4) THE VALUE OF CO WAS CALCULATED FROM DATA LISTED BY H. L. BOND ET AL.
ELASTIC CONSTANTS OF GERMANIUM SINGLE CRYSTALS PHYS. REV. LETTERS
VOL. 78 P. 176 (1950).
THE FOLLOWING CONSTANTS WERE LISTED:
C11 = 1.298 MEGABARS
C12 = 0.488 -
C44 = 0.873 -
- 5) A DYNAMIC YIELD POINT AT P = 46 KBAR, V/V0 = 0.975 AND A DYNAMIC
TRANSITION POINT AT P = 140 KBAR, V/V0 = 0.875 HAS BEEN OBSERVED
BY R. A. GRAHAM ET AL., SANDIA CORPORATION, ALBUQUERQUE, NEW MEXICO
(PREPRINT).
PREVIOUS WORK SUGGESTS THAT THE UPPER TRANSITION IS DUE TO MELTING:
TO 180,000 ATMOSPHERES, J. PHYS. CHEM. VOL. 59 P. 1144 (1955)
OR A SOLID-SOLID TRANSITION NEAR THE MELTING POINT REPORTED BY
S. MINOMURA AND H. G. DRICKAMER, PRESSURE INDUCED PHASE TRANSITION

- IN SILICON, GERMANIUM AND SOME III-V COMPOUNDS, J. PHYS. CHEM.
SOLIDS, VOL. 23 P. 451 (1962), IS NOT CLEAR.
- 6) THE EXTRAPOLATED V/V₀ OF THE TABLE SUGGESTS THAT THE VOLUME CHANGE
ACCOMPANYING THE UPPER TRANSITION OR TRANSITIONS IS OVER 10 PERCENT.

TABLE I

GERMANIUM
25---1



25---2

GERMANIUM, SINGLE CRYSTAL.

GE

$$V_0 = 0.1869 \text{ CC/G}$$

$$V_{01} = 0.1878 \text{ CC/G}$$

IN TABLE I BELOW, DENSITY IS GIVEN IN G/CC., VELOCITIES IN KM/SEC. AND (*) INDICATES ASSUMED VALUES. IN TABLE II, PRESSURES ARE GIVEN IN KILOBARS.

TABLE I

RHO0	US1	UP1	US2	UP2	US3	UP3
5.35	5.63	0.0775				
5.35	5.78	0.1286				
5.35	5.71	0.1443*	3.43	0.1580		
5.35	5.79	0.1443*	3.41	0.1620		
5.35	5.75	0.1443*	3.58	0.1700		
5.35	5.75	0.1443*	3.54	0.2132		
5.35	5.78	0.1443*	3.63	0.3432		
5.35	5.79	0.1443*	4.13	0.5540		
5.35	5.79	0.1443*	4.26	0.5778*	1.17	0.6015

$$US = 3.16 + 1.79 \cdot UP + OR - 0.09 \text{ KM/SEC}$$

TABLE II

P1	V1/V0	P2	V2/V0	P3	V3/V0
23.3	0.9864				
39.8	0.9778				
44	0.9750	46.5	0.9705		
44	0.9751	49.0	0.9697		
44	0.9749	50.3	0.9675		
44	0.9749	58.3	0.9551		
44	0.9750	83.5	0.9193		
44	0.9751	136.	0.8748		
44	0.9751	140.	0.8724	142	0.8413

COMMENTS:

- 1) SOURCE: GRAHAM, R. A., JONES, O. E. AND HOLLAND, J. R.
JOURNAL OF PHYSICS AND CHEMISTRY OF SOLIDS, VOL. 27, P 1519
(1968).
- 2) EXPERIMENTAL TECHNIQUE A: SHOCK VELOCITY WAS DETERMINED FROM THE
DEPENDENCE OF CONDUCTIVITY ON TIME DURING
SHOCK TRANSIT. PROJECTILE VELOCITY WAS

DETERMINED WITH SHEAR PINS.

- DATA REDUCTION TECHNIQUE A
STANDARD MATERIAL GERMANIUM, ASSUMING UFS = 2UP.
3) THE GERMANIUM CRYSTALS WERE HIGH PURITY, N-TYPE.
4) THE CRYSTALS WERE ORIENTED WITH THEIR FACES PARALLEL TO A (111)
CRYSTAL PLANE.
5) THE GERMANIUM PROJECTILES WERE ACCELERATED BY MEANS OF A COMPRESSED
GAS GUN. THE ANGULAR MISALIGNMENT BETWEEN THE IMPACTING SURFACES IS
APPROXIMATELY 0.0005 RADIANS.
6) FOR PARTICLE VELOCITY (UP) LESS THAN 0.1286 KM/SEC A SINGLE WAVE IS
OBSERVED AND AT UP = 0.1580 KM/SEC TWO WAVES ARE OBSERVED. WHEN
UP = 0.6015 KM/SEC A THIRD WAVE APPEARS. THE ASSUMED VALUES OF (UP)
FOR THE FIRST AND SECOND WAVES ARE 0.1443 AND 0.5778 KM/SEC
RESPECTIVELY. THE TRANSITION PRESSURES 44 AND 140 KILOBARS ARE
PROBABLY ACCURATE TO +OR- 10 PERCENT.
7) VOI WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CHRISTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES (DOVER
PUBLICATIONS, INC., NEW YORK, N. Y., 1963).

TABLE I

GERMANIUM, SINGLE CRYSTAL.

25---2

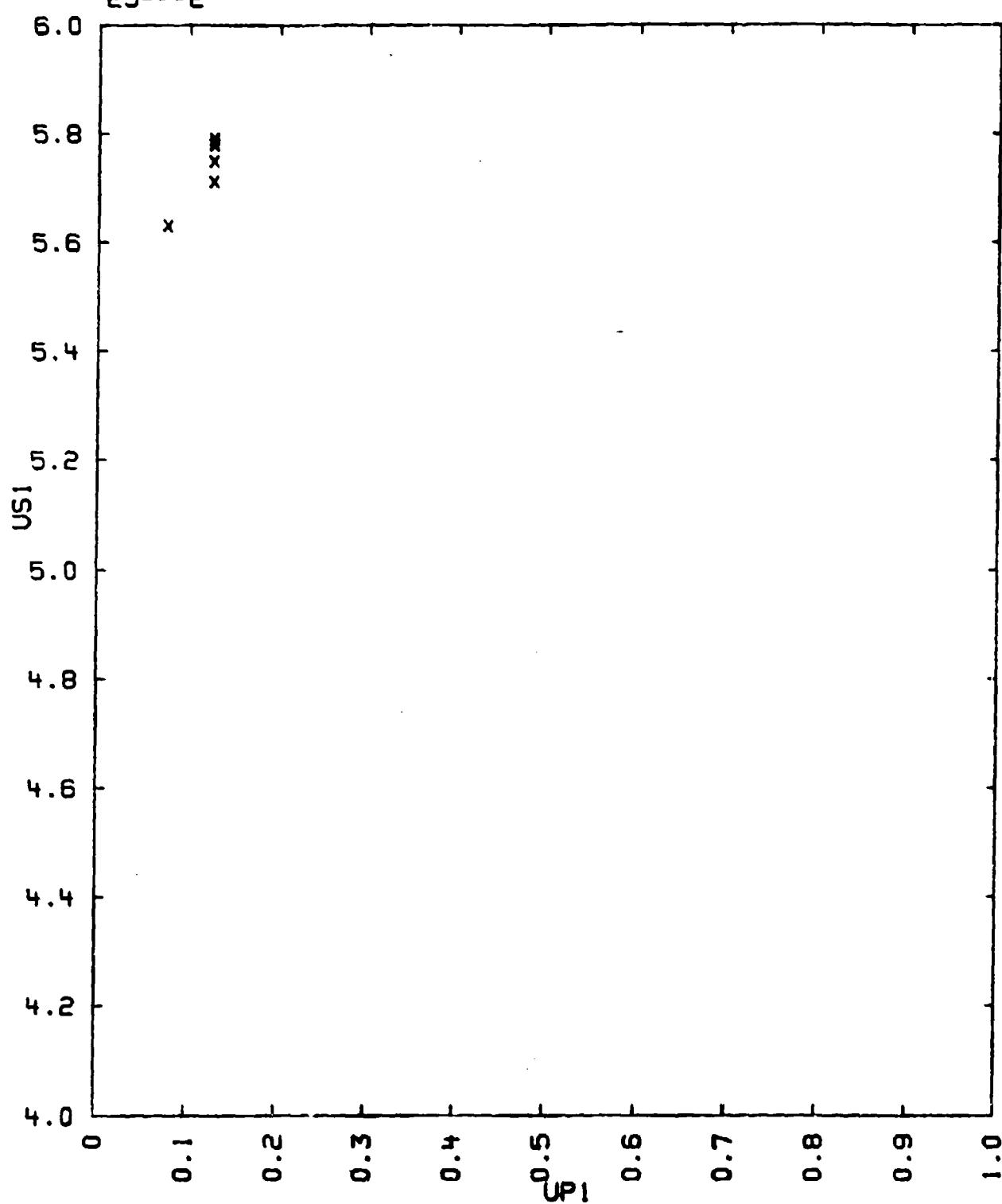


TABLE I

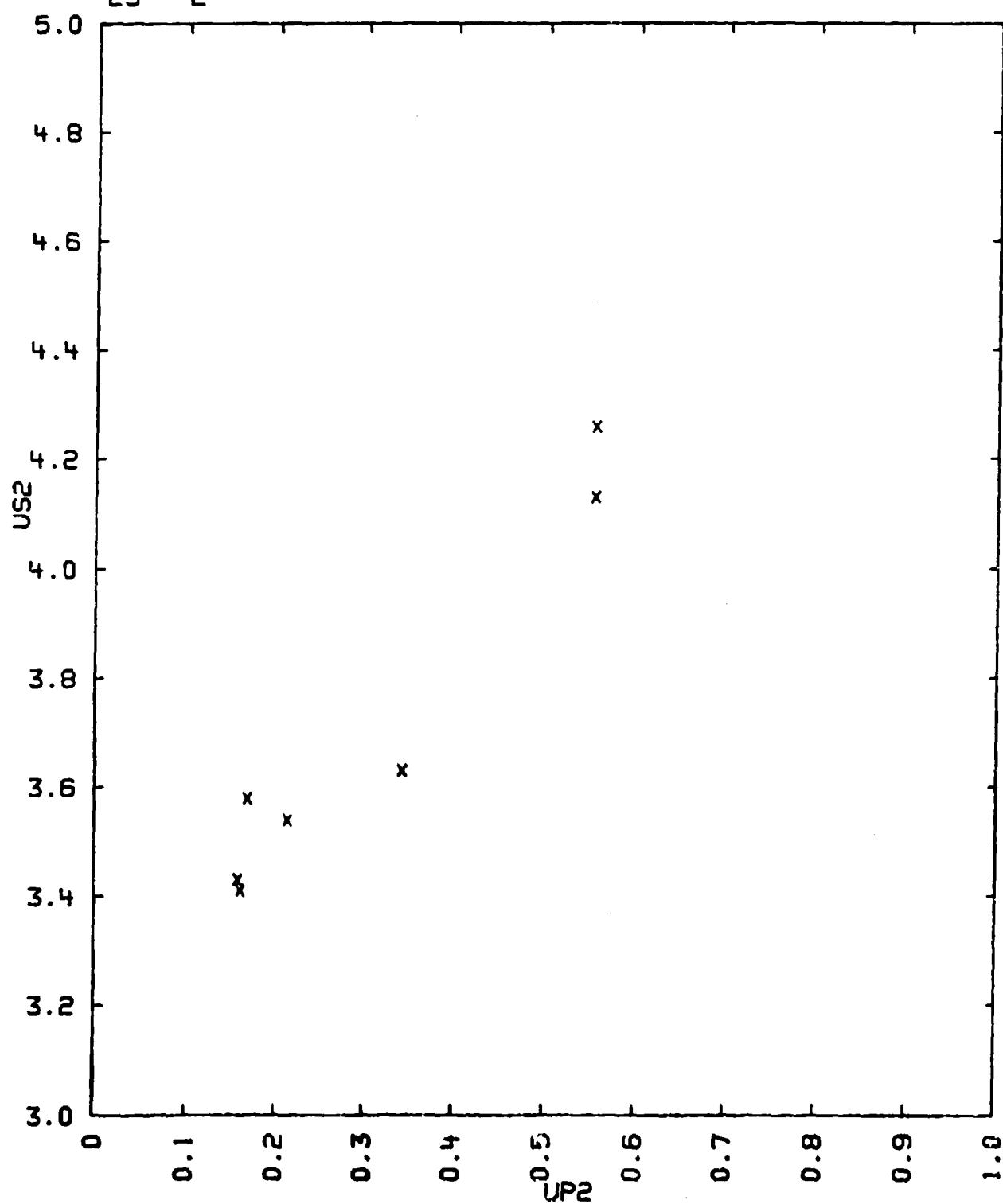
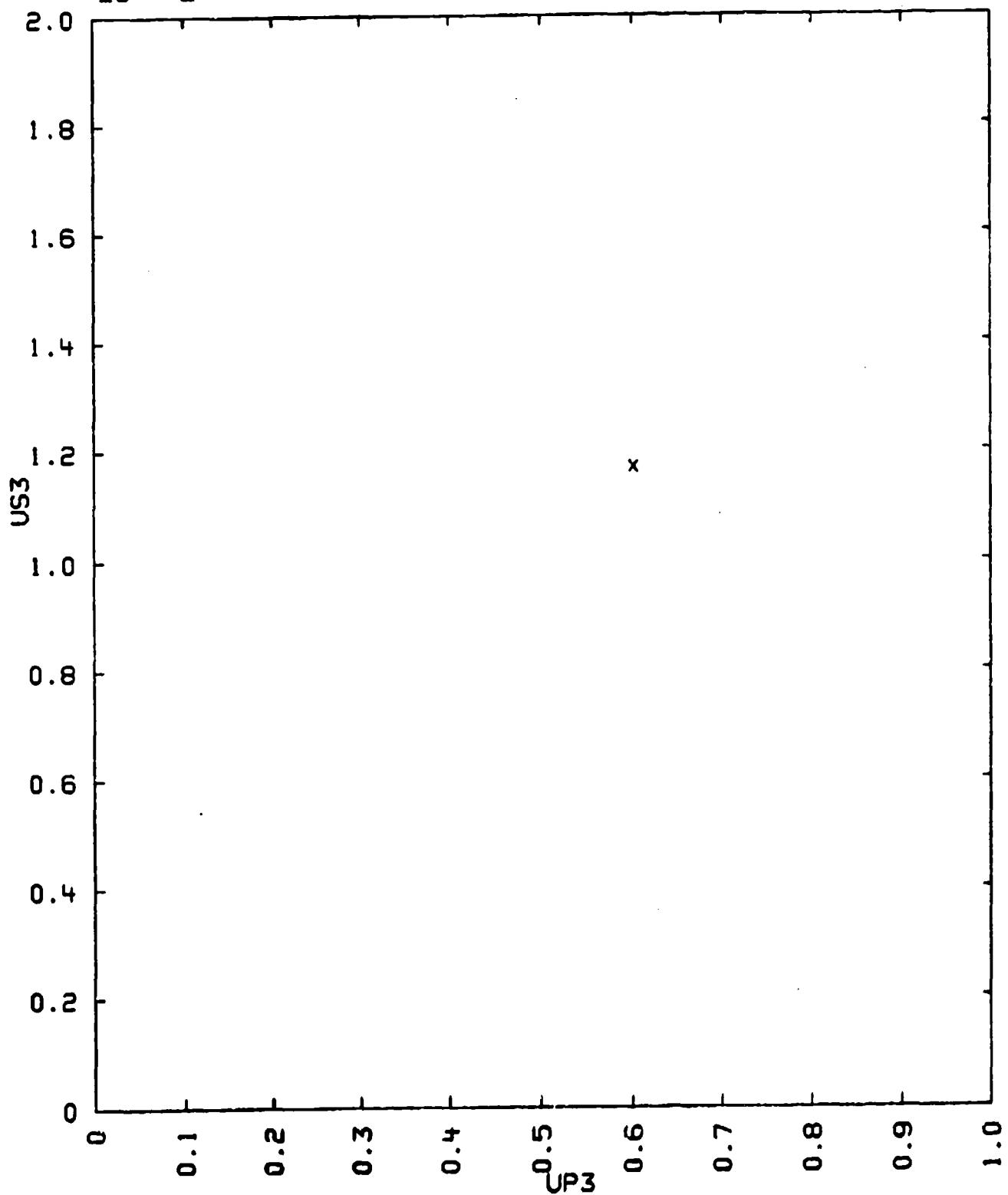
GERMANIUM, SINGLE CRYSTAL.
25---2

TABLE I

GERMANIUM, SINGLE CRYSTAL.

25---2



25---3
GERMANIUM

GE

$V_0 = 0.1878 \text{ CC/G}$
 $V_{01} = 0.1879 \text{ CC/G}$

 $C_0 = 3.754 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----		
RHO0	US	UP	P	V/V0	MATERIAL	US(IST)	
5.330	4.25	0.60	136.	0.8588	2024 AL	6.24	
5.330	4.22	0.75	169.	0.8223	2024 AL	6.45	
5.311	4.25	0.75	168.	0.8235	2024 AL	6.46	
5.311	4.14	0.78	174.	0.8092	2024 AL	6.49	
5.330	4.19	1.05	234.	0.7494	2024 AL	6.84	
5.330	4.16	1.07	237.	0.7428	2024 AL	6.86	
5.330	4.09	1.07	233.	0.7384	2024 AL	6.86	
5.311	4.12	1.09	239.	0.7354	2024 AL	6.87	
5.330	4.17	1.23	273.	0.7050	2024 AL	7.06	
5.311	4.18	1.46	324.	0.6507	2024 AL	7.38	
5.311	4.30	1.47	336.	0.6581	2024 AL	7.38	
5.311	4.18	1.50	333.	0.6411	2024 AL	7.40	
5.330	4.57	1.59	387.	0.6521	2024 AL	7.80	
5.330	4.55	1.60	388.	0.6484	2024 AL	7.60	
5.330	4.49	1.61	385.	0.6414	2024 AL	7.61	
5.330	4.82	1.75	450.	0.6369	2024 AL	7.35	
5.330	4.91	1.79	468.	0.6354	2024 AL	7.92	
5.330	5.22	1.95	543.	0.6264	2024 AL	8.17	
5.330	5.22	1.96	543.	0.6245	2024 AL	8.19	
5.330	5.14	1.97	540.	0.6187	2024 AL	8.19	
5.330	5.45	2.09	607.	0.6165	2024 AL	8.41	
5.330	5.38	2.11	605.	0.6078	2024 AL	8.41	
5.330	5.77	2.15	661.	0.6274	2024 AL	8.55	
5.330	5.68	2.16	654.	0.6197	2024 AL	8.55	
5.330	5.72	2.16	659.	0.6224	2024 AL	8.55	
5.330	5.73	2.19	669.	0.6178	2024 AL	8.59	
5.330	5.72	2.24	683.	0.6094	2024 AL	8.66	
5.330	5.50	2.25	680.	0.5909	2024 AL	8.61	
5.330	5.39	2.27	652.	0.5788	2024 AL	8.61	
5.330	5.65	2.27	684.	0.5982	2024 AL	8.68	
5.330	6.07	2.40	776.	0.6046	2024 AL	8.93	
5.330	6.12	2.41	786.	0.6062	2024 AL	8.96	
5.330	5.97	2.64	840.	0.5578	2024 AL	8.99	
5.330	5.91	2.47	778.	0.5821	2024 AL	8.99	
5.330	6.16	2.48	814.	0.5974	2024 AL	9.08	
5.330	6.06	2.56	827.	0.5776	2024 AL	9.13	
5.330	6.20	2.56	846.	0.5871	2024 AL	9.17	
5.330	6.19	2.57	848.	0.5848	2024 AL	9.17	
5.330	6.10	2.58	839.	0.5770	2024 AL	9.18	

GERMANIUM

RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
5.330	6.37	2.60	883.	0.5918	2024 AL	9.25
5.330	6.39	2.61	889.	0.5915	2024 AL	9.27
5.330	6.36	2.66	902.	0.5918	2024 AL	9.32
5.330	6.32	2.17	731.	0.6566	2024 AL	9.37
5.330	6.52	2.74	952.	0.5798	2024 AL	9.46
5.330	6.56	2.94	1028.	0.5518	2024 AL	9.72
5.330	6.92	3.01	1110.	0.5650	2024 AL	9.89
5.330	6.96	3.04	1128.	0.5632	2024 AL	9.94
5.330	6.71	3.12	1116.	0.5350	2024 AL	9.98
5.330	6.91	3.19	1175.	0.5384	2024 AL	10.11

US = 0.264 + 3.27*UP - 0.371*UP**2 KM/SEC FOR UP ABOVE 1.5 KM/SEC
 SIG US = 0.19 KM/SEC

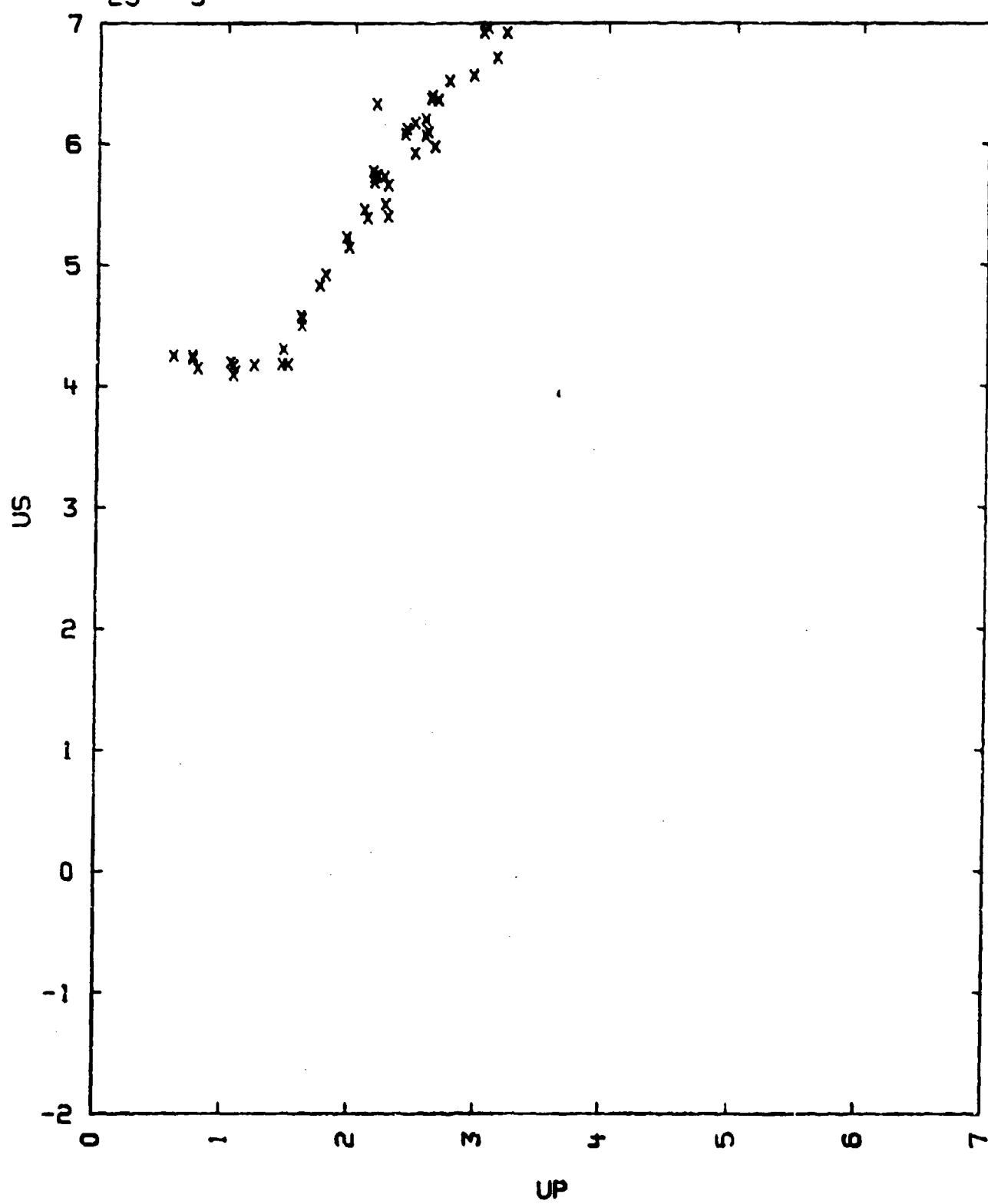
COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
 THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
 HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
 PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
 DATA REDUCTION TECHNIQUE: B
- 3) VOI FROM WYCKOFF, CRYSTAL STRUCTURES (JOHN WILEY, N.Y., 1963) VOL 1.
- 4) CO WAS CALCULATED FROM C11 = 12.8528E+11 DYNES/CM**2
 C12 = 4.82595E+11 DYNES/CM**2
 C44 = 6.680E+11 DYNES/CM**2
- 5) MC SKIMIN AND ANDREATCH, J. APL. PHYS. V. 34, P. 651, 01963).
 HUGONIOT ELASTIC LIMIT 52., 40., 37. KBAR (100, 114, 111 DIRECTIONS)

TABLE I

GERMANIUM

25---3



26---0
TIN SUMMARY

SN 99.8 PERCENT

$$V_0 = 0.1374 \text{ CC/G}$$

$$V_{01} = 0.1372 \text{ CC/G}$$

$$C_0 = 2.674 \text{ KM/SEC}$$

$$C_B = 2.76 \text{ KM/SEC}$$

THE TABLE BELOW LISTS THE HUGONIOT POINTS CALCULATED FROM THE US VS UP FITS GIVEN. VELOCITIES ARE IN KM/SEC, DENSITY IN G/CC, PRESSURE IN KBAR AND ENERGY DIFFERENCE IN KBAR.CC/G.

TABLE

FIT	RHO0	US	UP	P	V/V0	E-E0
1	7.28	3.44	0.6	150	0.926	1.80
1	-	4.08	1.0	297	0.755	5.00
1	-	4.87	1.5	532	0.692	11.25
1	-	5.67	2.0	825	0.647	20.0
2	-	6.27	2.4	1096	0.618	28.8
2	-	7.01	3.0	1530	0.572	45.0
2	-	7.61	3.5	1940	0.540	61.2
2	-	8.22	4.0	2395	0.514	80.0
2	-	8.83	4.5	2894	0.491	101.2
2	-	9.08	4.7	3106	0.482	110.5

$$US = 2.49 + 1.59 \cdot UP, \text{ SIG.US} = 0.04 \text{ KM/SEC} \quad (\text{FIT 1})$$

FOR UP BETWEEN 0.6 AND 2.2 KM/SEC

$$US = 3.35 + 1.219 \cdot UP, \text{ SIG.US} = 0.05 \text{ KM/SEC} \quad (\text{FIT 2})$$

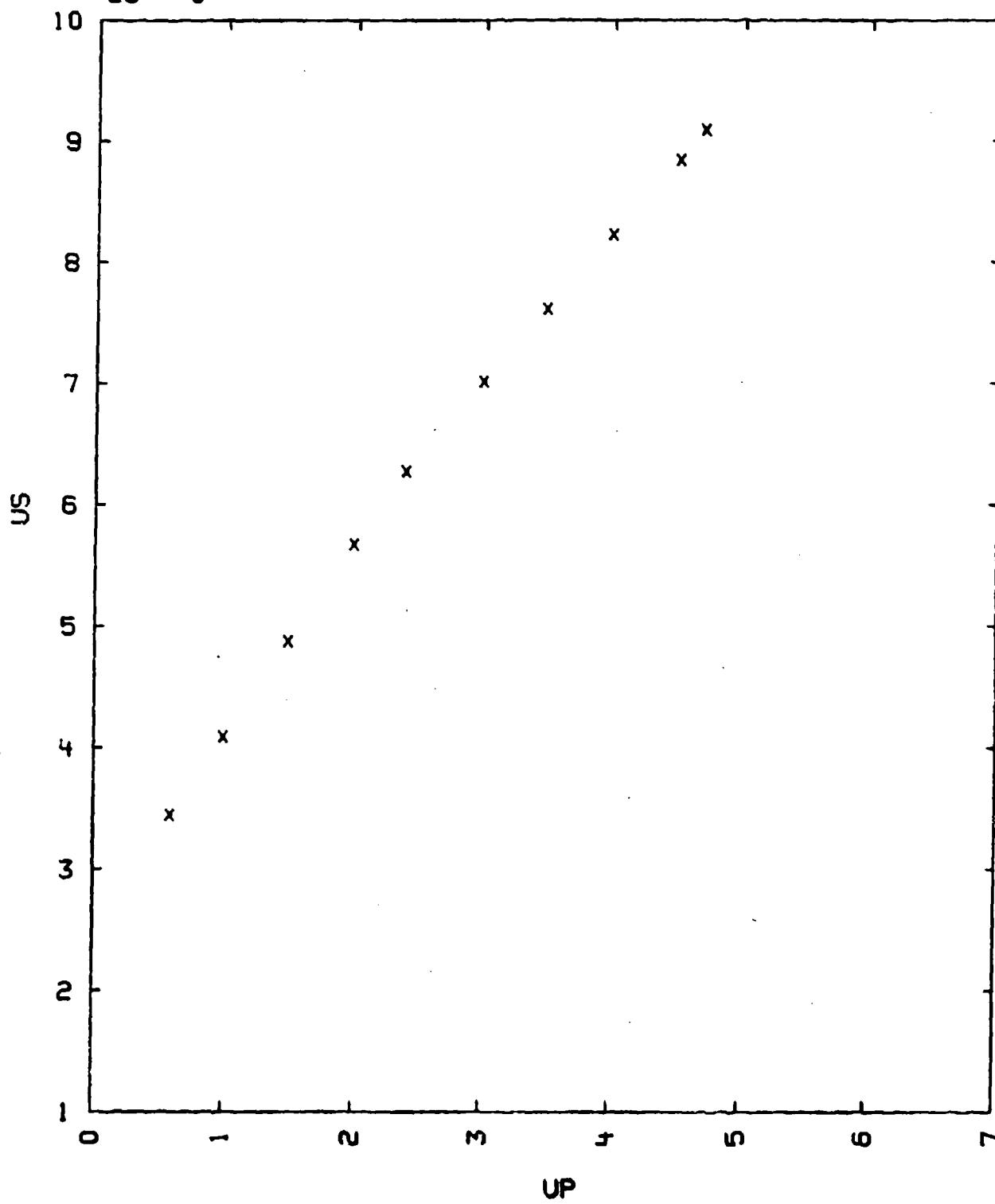
FOR UP BETWEEN 2.2 AND 4.7 KM/SEC

COMMENTS:

- 1) SOURCE: COMPILER.
- 1A) THE DATA OF MATERIALS 26---1,2 AND 3 WERE USED FOR THIS SUMMARY.
THE DATA OF 26---3 TABLE II AND 26---4 AGREE WITH THE OTHER DATA
BUT DO NOT IMPROVE THE ACCURACY.
- 2) THE CRYSTAL DENSITY WAS OBTAINED FROM THE AM. INST. OF PHYS. HANDBOOK
(MCGRAW-HILL BOOK CO INC. 1963) 2ND. ED.
- 3) C0 AND CB ARE VALUES USED ON PAGES 26---2 AND 3.

TABLE I

TIN SUMMARY
26---0



26---1

TIN

SN 99.8 PER CENT OR GREATER

$V_0 = 0.1374 \text{ CC/G.}$ $C_L = 3.32 \text{ KM/SEC.}$ $C_0 = 2.70 \text{ KM/SEC.}$
 $V_{01} = 0.1370 \text{ CC/G.}$ $C_S = 1.67 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

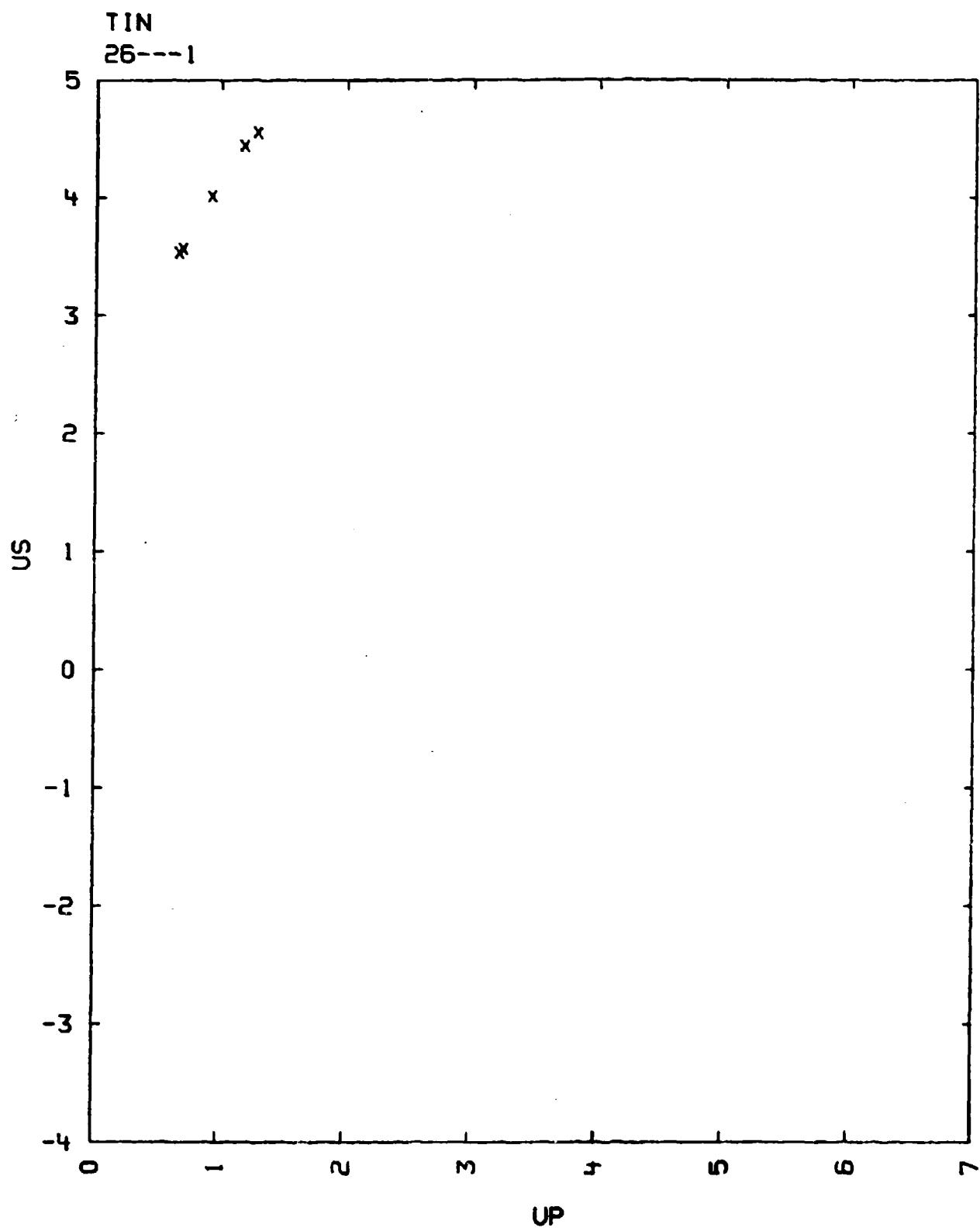
RHO0	US	UFS	UP	P	V/V0
7.28	4.555	2.704	1.290	427.8	0.7168
-	4.435	2.539	1.190	384.2	0.7317
-	4.004	1.912	0.925	269.6	0.7690
-	3.557	1.486	0.705	182.6	0.8018
-	3.524	1.364	0.670	171.9	0.8098

$$US = 2.378 + 1.712 UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.9 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS
- 5) CL AND CS WERE OBTAINED FROM L. BERGMAN, DE ULTRASCHALL,
S. HIRZEL VERLAG, STUTTGART, 1954, 6TH ED., P. 650
- 6) V01 WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW HILL BOOK CO.)
2ND ED.

TABLE I



26---2

TIN

SN

$$V_0 = 0.1374 \text{ CC/G}$$

$$V_{01} = 0.1372 \text{ CC/G}$$

$$C_B = 2.76 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
7.28	5.59	1.94	792	0.652	931
-	5.55	1.95	788	0.648	931
-	5.57	1.95	790	0.651	931
-	6.71	2.80	1364	0.583	1586
-	6.80	2.78	1377	0.591	1587
-	6.75	2.81	1378	0.584	1599

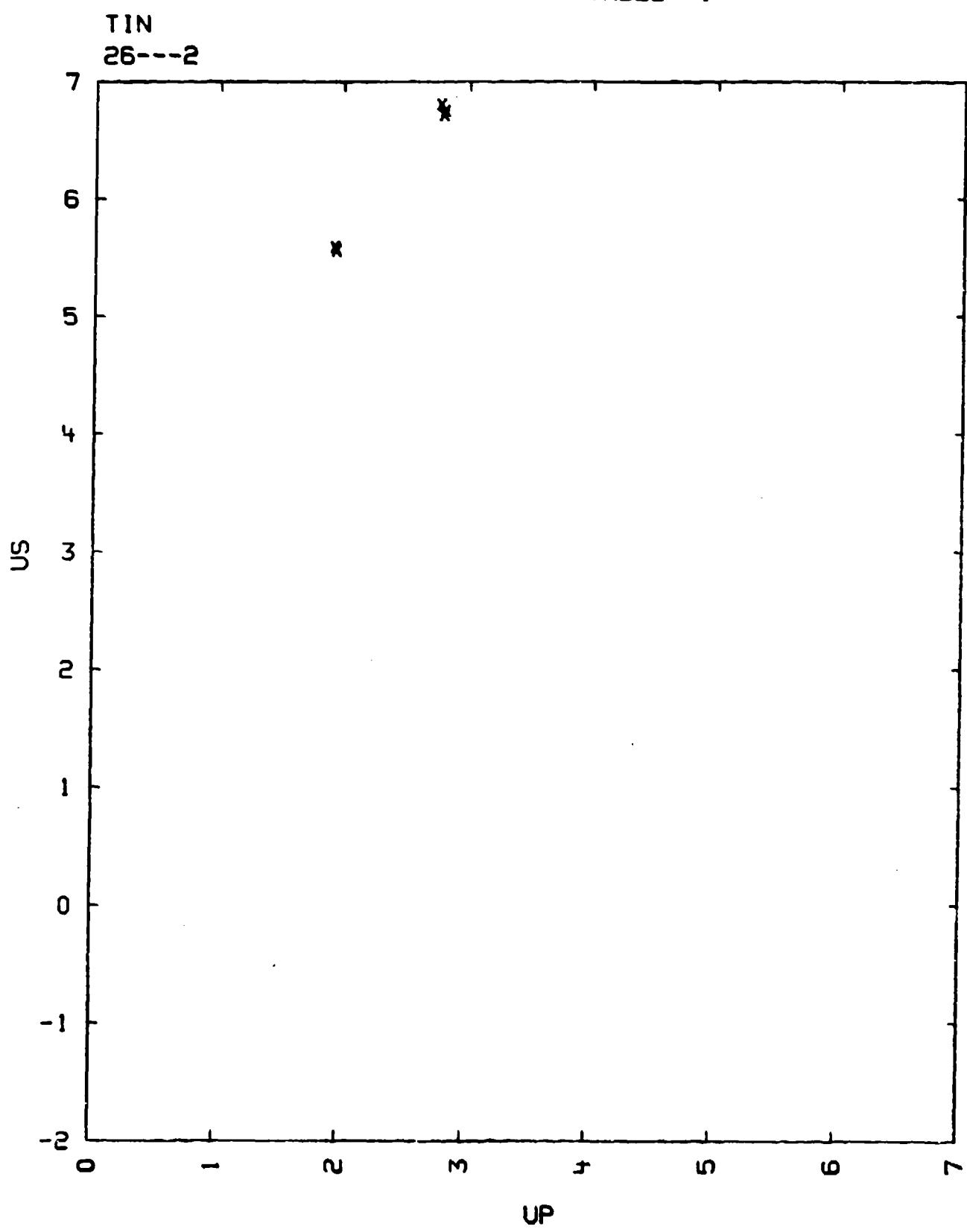
$$US = 2.64 + 1.476 UP \text{ KM/SEC.}$$

$$\Sigma US = 0.94 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 4) V01 WAS TAKEN FROM THE AM. INST. OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO., N.Y., 1963; 2ND ED.

TABLE I



26---3

TIN

SN

$$V_0 = 0.1374 \text{ CC/G.}$$

$$V_{01} = 0.1372 \text{ CC/G.}$$

$$C_0 = 2.674 \text{ KM/SEC}$$

IN THE TABLES BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL USED AS SAMPLE HOLDER IN TABLE I AND AS STRIKER OR PROJECTILE PLATE IN TABLE II. HERE $U(ST)$ IS TOTAL VELOCITY BEFORE IMPACT.

TABLE I

SAMPLE					HOLDER			
RHO0	US	UP	P	V/V0	ST	US(ST)	UP(ST)	
7.28	4.20	1.08	330	0.7407	FE	5.30	0.95	
-	6.36	2.46	1139	0.6131	FE	7.53	2.24	
-	9.02	4.65	3053	0.4845	FE	10.67	4.26	

US =

TABLE II

SAMPLE					STRIKER	
RHO0	US	UP	P	V/V0	ST	U(ST)
7.28	5.99	2.15	938	0.6410	AL	5.60
-	7.22	3.06	1608	0.5764	FE	5.60
-	9.50	4.97	3437	0.4769	FE	9.10
-	12.98	7.96	7520	0.3868	FE	14.68

$$\text{US} = 2.730 + 1.524 \cdot \text{UP} - 0.030 \cdot \text{UP}^2.$$

OR $\text{US} = 3.203 + 1.248 \cdot \text{UP}$

$$\text{SIG.US} = 0.14 \text{ KM/SEC.}$$

$$\text{SIG.US} = 0.21 \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: AL'TSHULER, L.V., BAKANOVA, A.A. AND TRUNIN, R.F.
SOVIET PHYS.-JETP, VOL. 15, P. 65 (1962).
J. EXPTL. THEORET. PHYS. (USSR) VOL. 42, P. 91, (1962)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
- 3) TABLE I IS CORRECTED DATA WHICH FIRST APPEARED IN THE SOVIET
PHYS.-JETP, VOL. 34, P. 614 (1960). THE \cdot VALUES WERE OBTAINED FROM
THE IRON STRIKER VELOCITY OF 8.84 KM/SEC.
- 4) THE INACCURACY IN THE DETERMINATION OF US IN TABLE II TO PRESSURES OF
4187 KILOBARS DID NOT EXCEED 1.0 PERCENT. EACH POINT IS THE AVERAGE

- OF 4 TO 6 EXPERIMENTS:
- 5) THE UNCERTAINTY OF THE MEASURED VALUES IN THE LAST TABLE ENTRY ARE:
1 PERCENT IN US, OBTAINED FROM 6-8 EXPERIMENTS AND 1.5 PERCENT IN
U(ST) OBTAINED FROM 12 EXPERIMENTS.
 - 6) VOI WAS CALCULATED FROM THE TETRAGONAL UNIT CELL CONSTANTS
 $A = 5.81970 \text{ KX}$ AND $C = 3.17488 \text{ KX}$
AMERICAN INSTITUTE OF PHYSICS HANDBOOK (MCGRAW-HILL BOOK CO., N.Y.,
1963) 2ND ED.
 - 7) CD IS GIVEN BY: K. H. SCHRAMM, Z. METALKUNDE VOL. 53, P. 729 (1962)

TABLE :

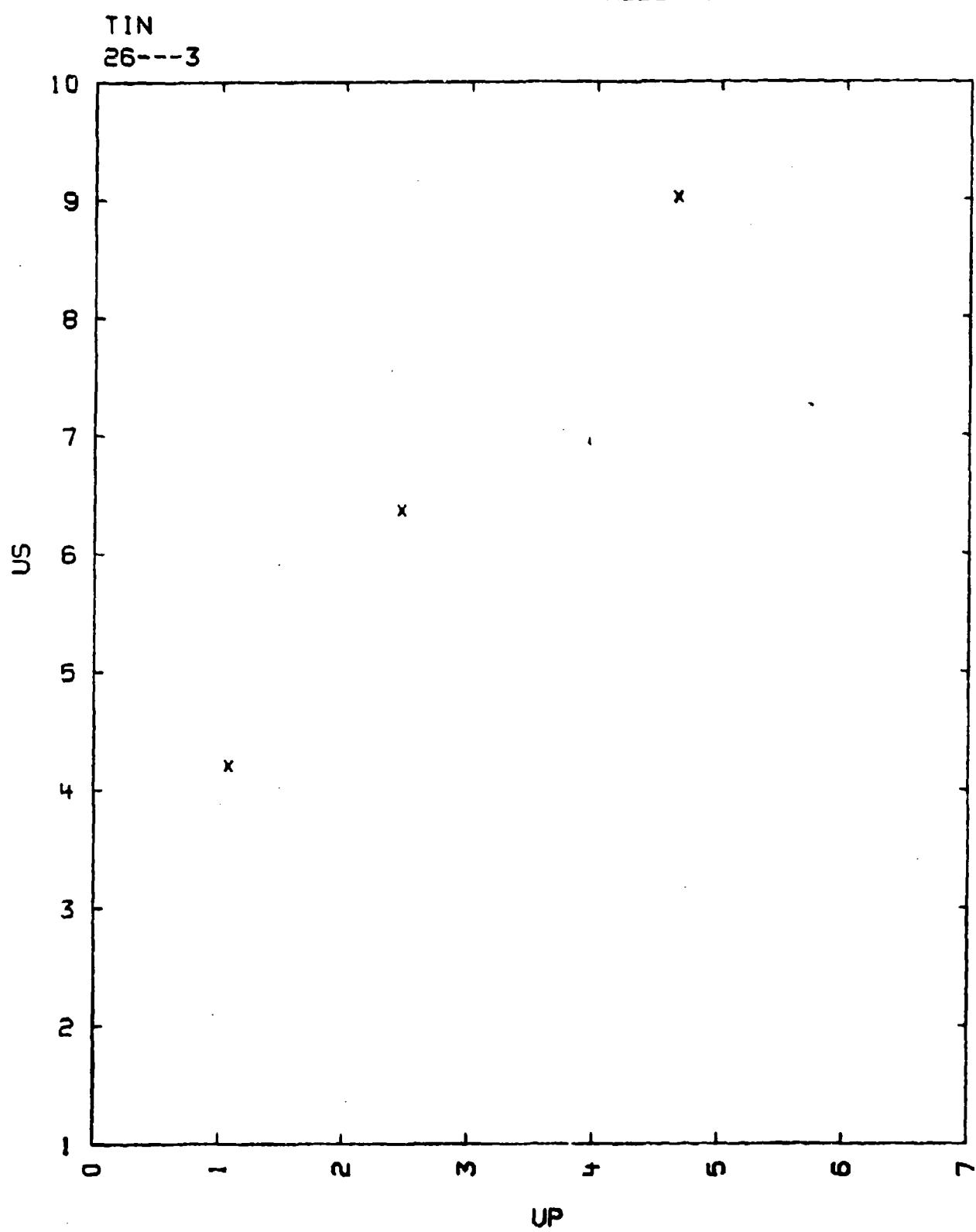
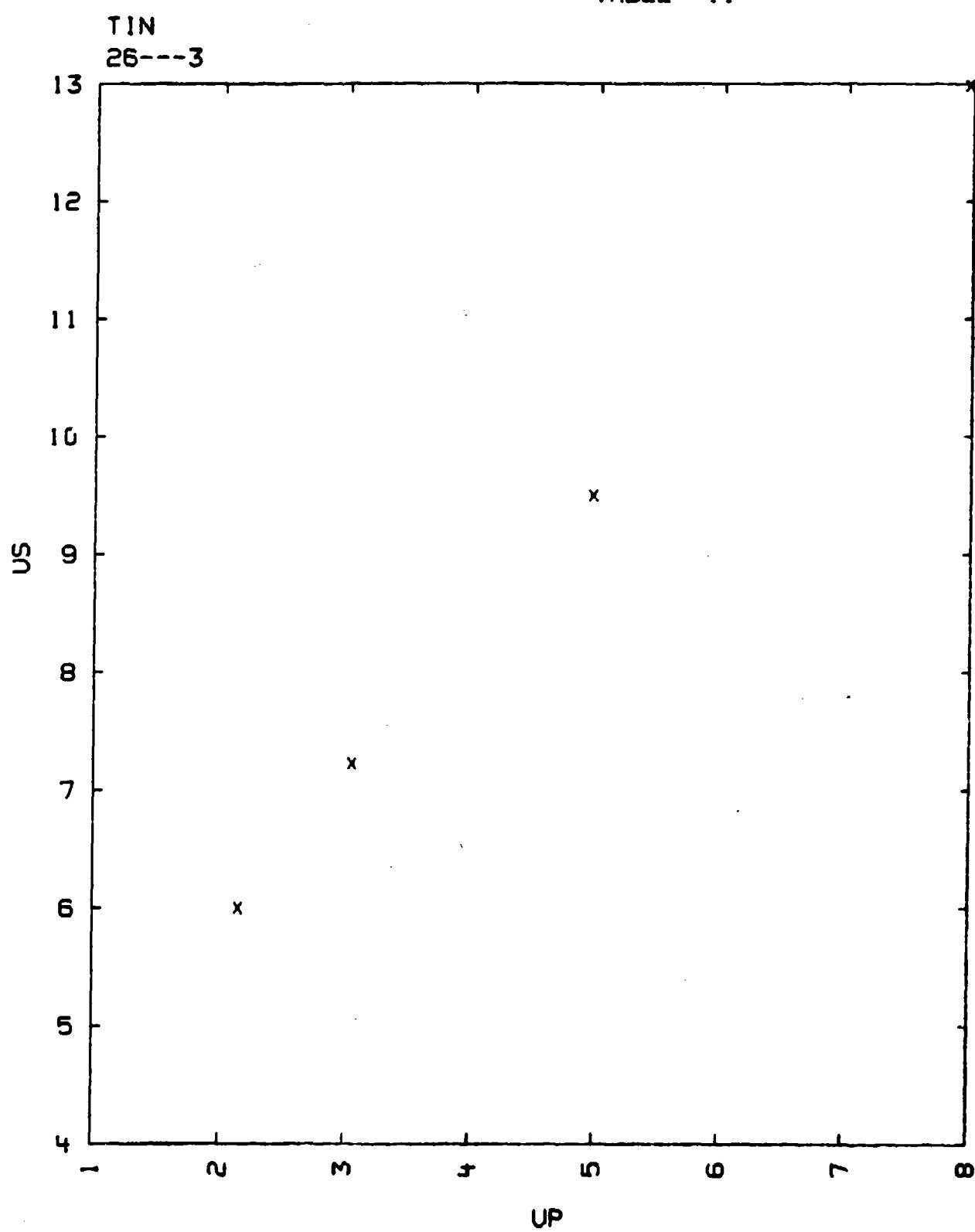


TABLE II



26---4

TIN

SN

$$V_0 = 0.1368 - 0.1390 \text{ CC/G}$$

$$V_{01} = 0.1372 \text{ CC/G}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

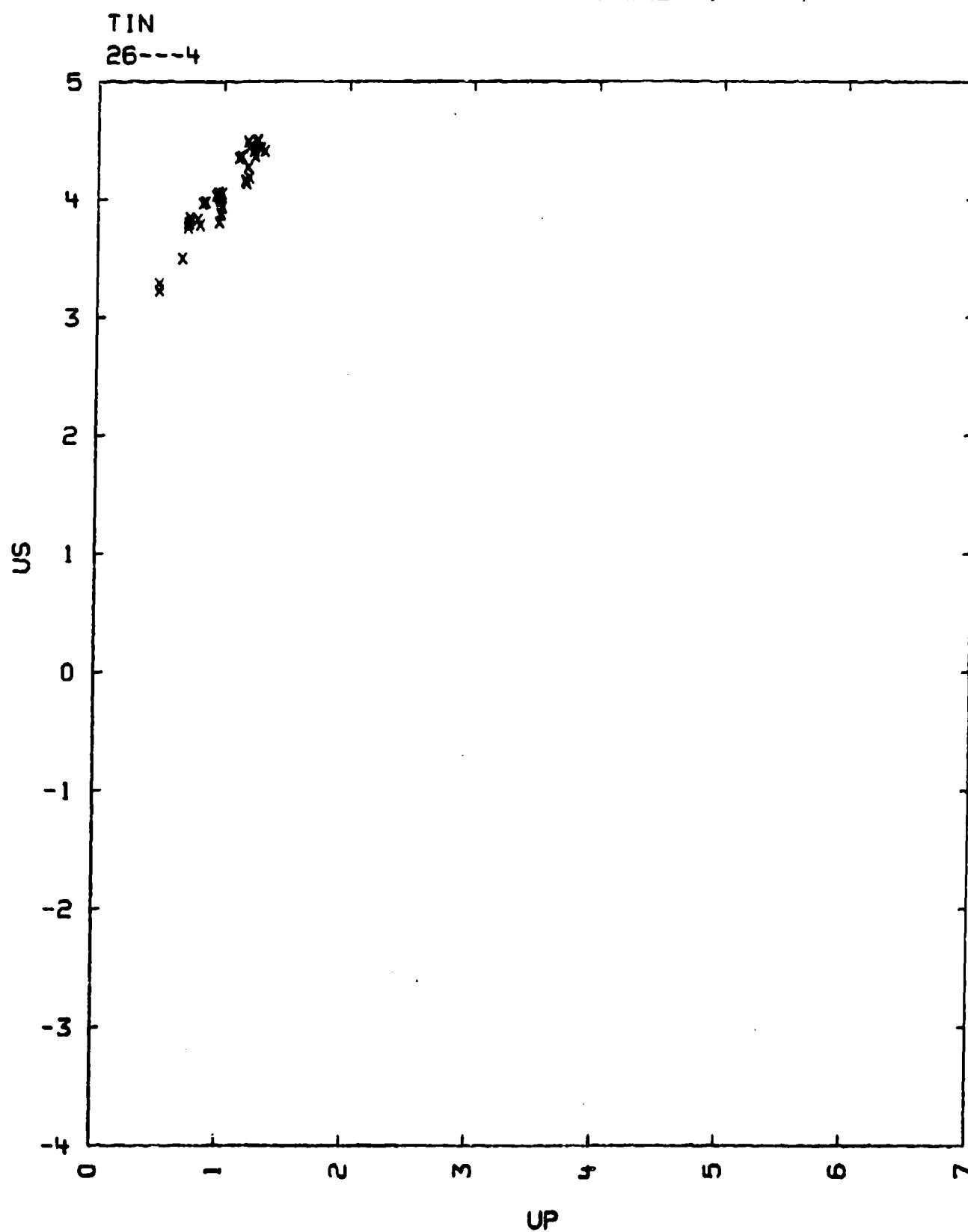
TABLE

SAMPLE				STANDARD	
RHO0	US	UP	P	V/V0	UFS
7.30	4.401	1.330	427	0.698	3.290
7.30	4.425	1.291	417	0.708	3.220
7.31	4.484	1.271	417	0.716	3.200
7.31	4.500	1.270	418	0.718	3.200
7.31	4.405	1.236	398	0.719	3.095
7.31	4.359	1.251	398	0.713	3.110
7.31	4.433	1.213	393	0.726	3.050
7.31	4.488	1.195	392	0.734	3.025
7.27	4.266	1.200	372	0.719	2.960
7.27	4.181	1.204	368	0.712	2.940
7.31	4.344	1.126	358	0.741	2.830
7.31	4.359	1.144	385	0.739	2.880
7.30	4.149	1.175	356	0.716	2.875
7.30	4.132	1.182	357	0.713	2.890
7.31	3.980	0.983	298	0.753	2.415
7.31	4.042	0.990	292	0.755	2.440
7.31	4.055	0.986	286	0.762	2.400
7.31	4.032	0.953	281	0.764	2.360
7.31	4.023	0.273	285	0.757	2.400
7.31	4.010	0.973	286	0.758	2.400
7.31	3.931	0.889	284	0.745	2.400
7.31	3.873	0.883	278	0.750	2.400
7.31	3.884	0.875	277	0.749	2.365
7.31	3.798	0.971	269	0.744	2.335
7.31	3.965	0.848	248	0.798	2.115
7.31	3.971	0.862	250	0.782	2.145
7.31	3.824	0.804	225	0.790	1.985
7.31	3.779	0.819	226	0.783	2.000
7.31	3.791	0.738	204	0.806	1.830
7.31	3.837	0.739	207	0.807	1.840
7.31	3.759	0.720	198	0.808	1.780
7.31	3.791	0.722	200	0.810	1.795
7.31	3.492	0.686	175	0.804	1.640
7.31	3.484	0.686	175	0.803	1.640
7.31	3.274	0.497	118	0.848	1.195
7.31	3.222	0.499	117.5	0.845	1.190

$$US = 2.65 + 1.38 \cdot UP \text{ KM/SEC.}$$

$$\pm 10 US = 0.11$$

TABLE I



COMMENTS:

- 1) SOURCE: BERGER J. AND FAUQUIGNON C.
PRIVATE COMMUNICATION (1964), B.P. NO. 7, SEVRAN, FRANCE
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL ALUMINUM AU4G ALLOY
- 3) VOI WAS DETERMINED FROM THE TETRAGONAL UNIT CELL CONSTANTS
 $A = 5.81970 \text{ KX}$ AND $C = 3.17488 \text{ KX}$
CRYSTAL DATA DETERMINATIVE TABLES (AMERICAN CRYSTALLOGRAPHIC
ASSOCIATION, 1963) 2ND ED.

27---0
LEAD. SUMMARY

PB

$$\begin{array}{lll} V_0 = 0.9882 \text{ CC/G} & C_L = 2.16 \text{ KM/SEC} & C_0 = 2.00 \text{ KM/SEC} \\ V_{01} = 0.08818 \text{ CC/G} & C_S = 0.70 \text{ KM/SEC} & C_B = 2.00 \text{ KM/SEC} \end{array}$$

THE TABLE LISTS HUGONIOT POINTS OBTAINED FROM THE LISTED US VS. UP FITS. DENSITY IS IN G/CC, VELOCITIES IN KM/SEC, PRESSURE IN KBAR AND ENERGY DIFFERENCE IN KBAR.CC/G. THE FIRST 8 ENTRIES ARE FROM 27---8. EL = ELASTIC LIMIT.

TABLE

FIT	RHO0	US	UP	P	V/V0	E-E0	COMMENTS
11.34	2.3	.00093		0.24	0.9996	0.0004	1ST WAVE EL
-	1.90	.0193		4.20	0.9901	0.0021	2ND -
-	1.95	.0292		6.28	0.9853	0.0041	- -
-	2.08	.0500		11.82	0.9736	0.0124	- -
-	2.15	.1146		27.95	0.9466	0.0657	- -
-	2.26	.1834		47.24	0.9191	0.1682	1ST -
-	2.37	.2546		68.55	0.8926	0.324	- -
-	2.44	.2822		78.12	0.8842	0.398	- -
1	-	2.467	.3	83.9	0.8784	45	- -
1	-	2.929	.6	199.3	0.7951	1.80	- -
1	-	3.544	1.0	402	0.7178	5.0	- -
1	-	4.313	1.5	733	0.6522	11.2	- -
1	-	5.082	2.0	1152	0.6064	20.0	- -
2	-	5.364	2.2	1338	0.5898	24.2	- -
2	-	6.363	3.0	2165	0.5285	45.0	- -
2	-	7.613	4.0	3453	0.4746	80.0	- -
2	-	8.882	5.0	5024	0.4358	125.0	- -
2	-	10.11	6.0	6879	0.4066	180.0	- -
2	-	11.36	7.0	9017	0.3838	245.0	- -

US = $2.005 + 1.538 \cdot UP \text{ KM/SEC}$ (FIT 1)

SIG US = 0.014 KM/SEC

US = $2.616 + 1.249 \cdot UP \text{ KM/SEC}$ (FIT 2)

SIG US = 0.065 KM/SEC

COMMENTS

- 1) SOURCE: COMPILER
- 1A) THE FITS WERE MADE FROM THE DATA OF 27---1, 2, 4 AND 7. THE LOW PRESSURE DATA LISTED IN 27---8, WERE TRANSFERRED DIRECTLY WITHOUT INCLUDING THE RECALCULATED US AND UP POINTS IN THE FIT.
- 2) FIT 1 AGREES ESSENTIALLY WITH THE DATA OF PAGE 27---8 ABOVE 10 KBAR BUT THE LOWEST POINTS SUGGEST A LOWER INTERCEPT. THEIR UP VALUES

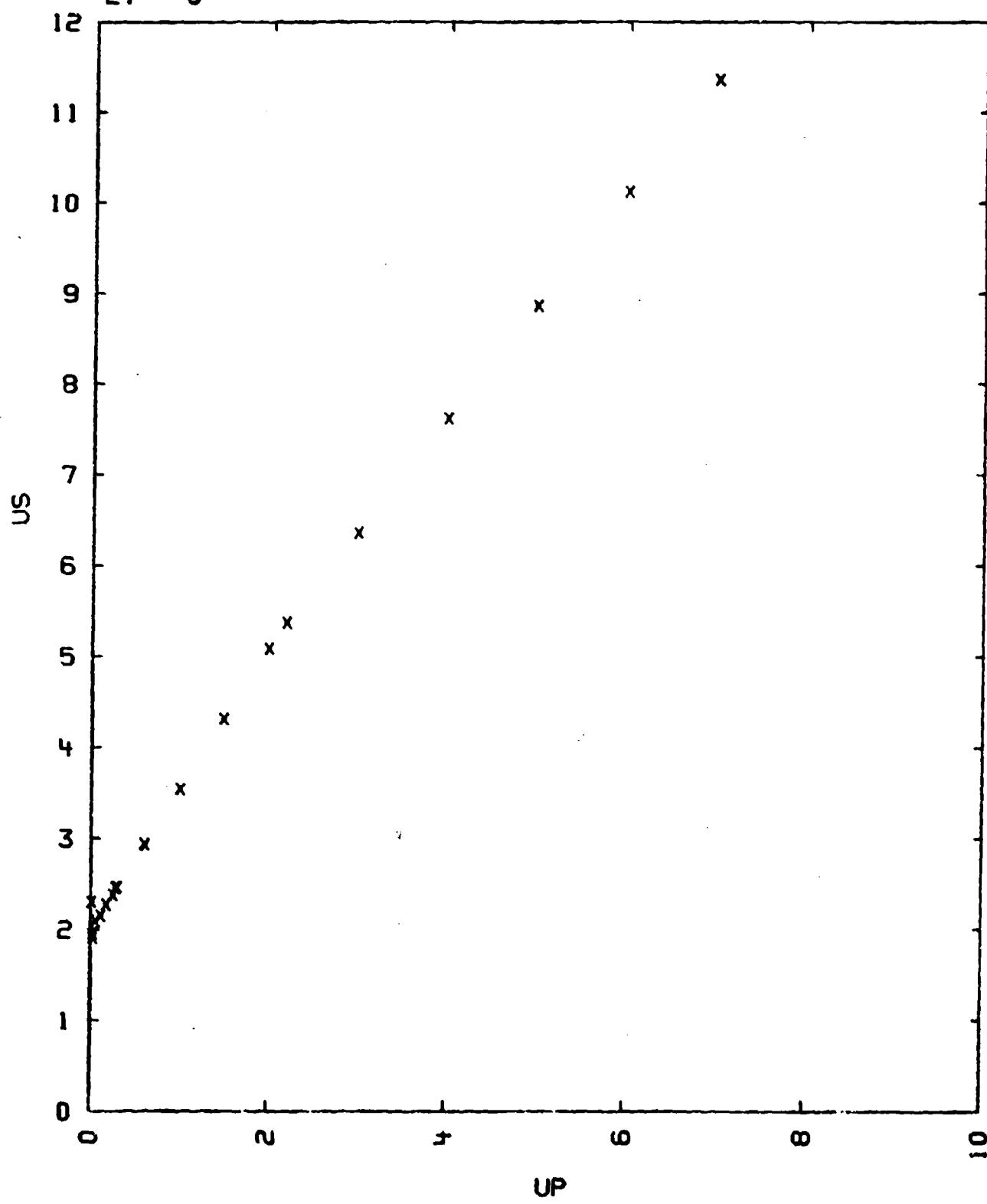
HAVE BEEN SLIGHTLY READJUSTED TO YIELD THE PROPER VALUE OF P IN A SIMPLE TWO-STEP PROCESS(SEE COMMENTS OF 27---8).

- 3) DATA OF 27---3 ARE IN GREATER DISAGREEMENT, WHILE US UP VALUES OF 27---6 ARE CALCULATED POINTS.

TABLE I

LEAD, SUMMARY

27---0



27---1

LEAD

PB 99.8 PER CENT OR GREATER

$$V_0 = 0.0882 \text{ CC/G} \quad C_L = 2.16 \text{ KM/SEC.} \quad C_0 = 2.00 \text{ KM/SEC.}$$

$$V_{01} = 0.0881 \text{ CC/G} \quad C_S = 0.70 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

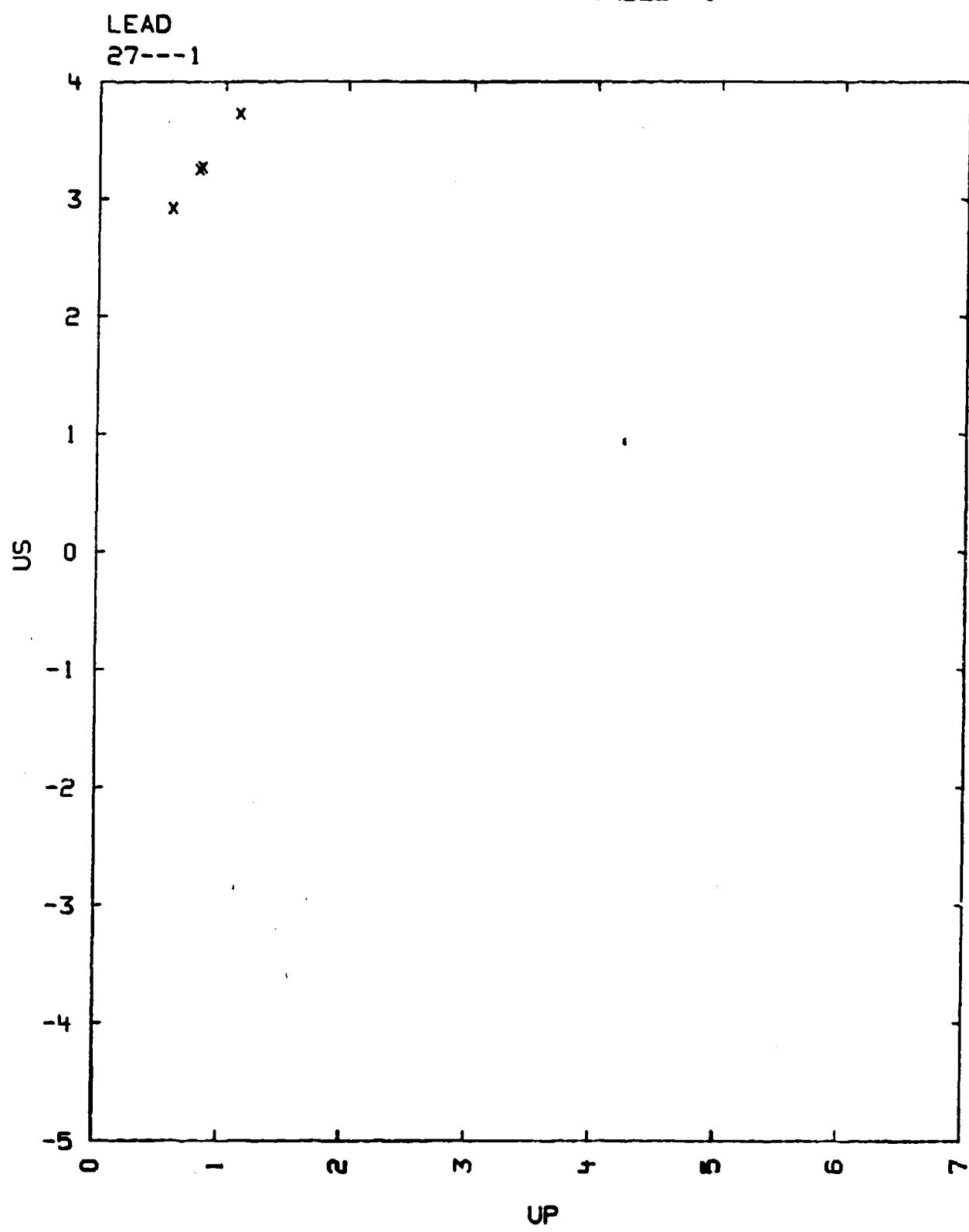
RHO0	US	UFS	UP	P	V/V0
11.34	2.914	1.230	0.590	194.8	0.7975
-	3.268	1.745	0.819	303.2	0.7494
-	3.250	1.731	0.802	295.3	0.7532
-	3.724	2.420	1.118	471.7	0.6898

$$US = 2.092 + 1.452 UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.4 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
STANDARD MATERIAL 24ST ALUMINUM.
- 3) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS.
- 4) CL AND CS VALUES WERE OBTAINED FROM L. BERGMANN, DER ULTRASCHALL,
S. HIRZEL VERLAG, STUTTGART, 1954, 6TH ED., P. 650
- 5) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.

TABLE I



27---2
LEAD

PB

$$V_0 = 0.0882 \text{ CC/G}$$

$$V_{01} = 0.0881 \text{ CC/G}$$

$$C_B = 2.02 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

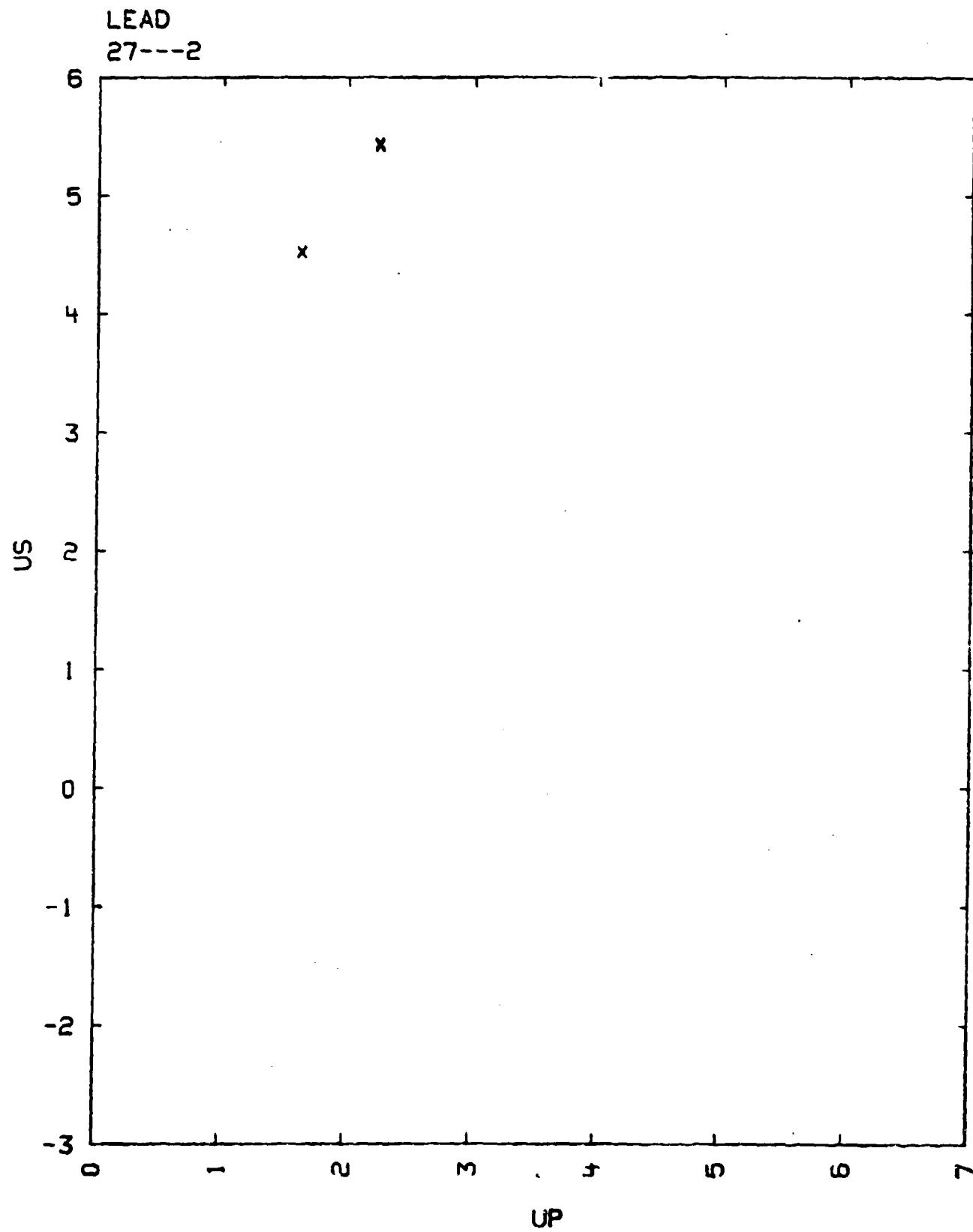
RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
11.34	4.52	1.64	838	0.638	842
			837	0.638	842
-	5.44	2.25	1388	0.587	1354
-	5.42	2.25	1383	0.585	1352

$$US = 2.028 + 1.517 UP \text{ KM/SEC.} \quad SIGMA US = 0.11 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 4) V01 WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW HILL BOOK CO.)
2ND ED.

TABLE I



27---3

LEAD

PB

$$V_0 = 0.0882 \text{ CC/G.}$$

$$V_{01} = 0.0882 \text{ CC/G.}$$

$$C_B = 2.00 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN KM/SEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

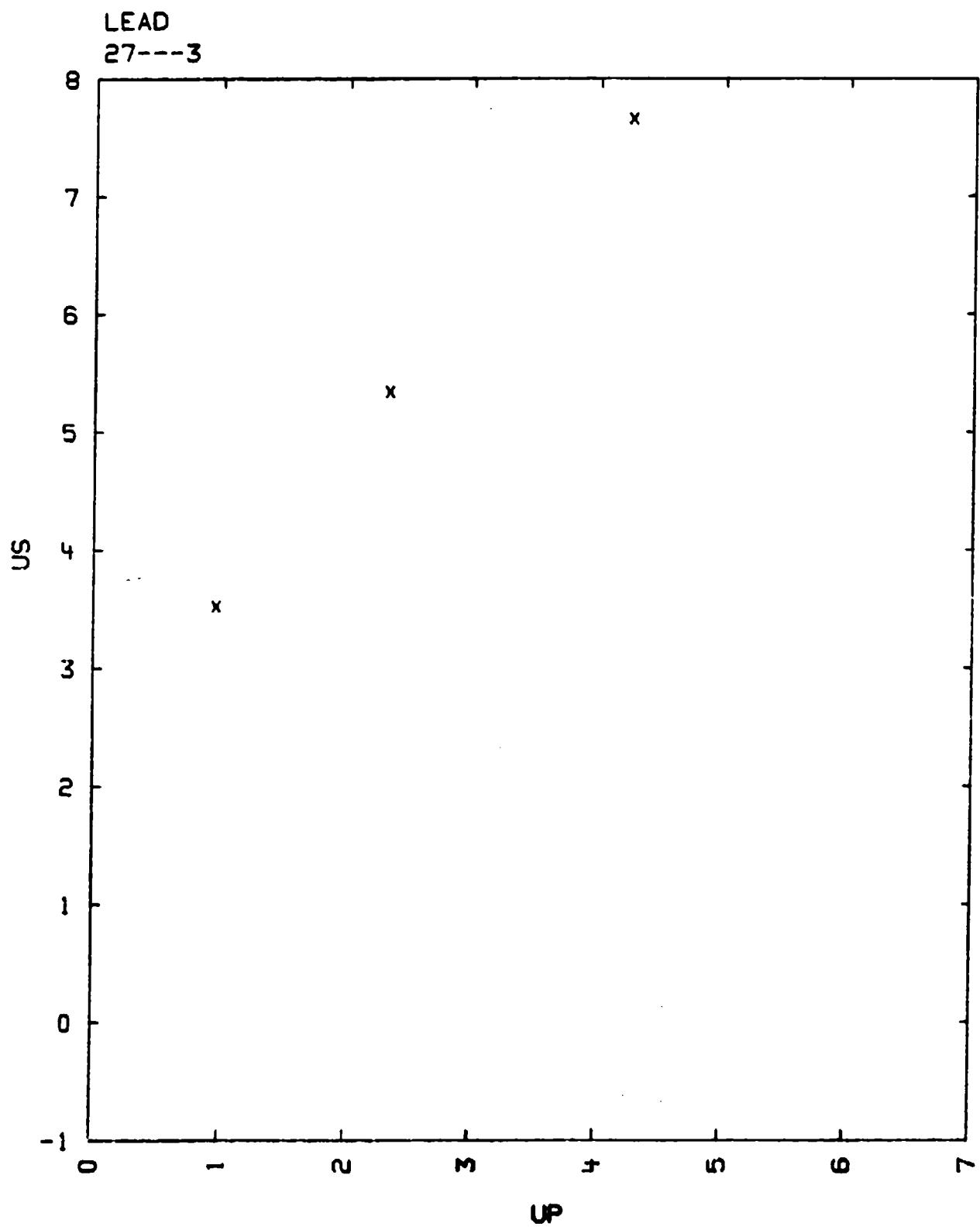
RHO0	US	UP	P	V/V0
11.34	3.52	0.97	390	0.725
-	5.33	2.34	1410	0.562
-	7.65	4.26	3700	0.442

$$US = 2.30 + 1.27UP \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: AL'TSHULER, L. V., KRUPNIKOV, K. K. AND BRAZHNICK, M. I.
SOVIET PHYS.-TEPP, VOL. 7, P. 614, (1958)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL FE.
- 3) CB IS CALCULATED USING THE VOLUME COMPRESSION COEFFICIENT TAKEN FROM
P. H. BRIDGMAN, THE PHYSICS OF HIGH PRESSURE, (G. BELL AND SONS LTD.
LONDON 1958)
- 4) V01 WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK
(MCGRAH-HILL BOOK CO. 1963) 2ND ED.
- 5) THE SPECIMEN WITH A THICKNESS BETWEEN 6-8MM. WAS ATTACHED TO AN IRON
PLATE OF COMPARABLE THICKNESS 6-9MM.

TABLE I



27---4
LEAD

PB

$$V_0 = 0.0882 \text{ CC/G.}$$

$$V_{01} = 0.0881 \text{ CC/G.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL USED AS STRIKER PLATE. U(ST) IS THE STRIKER VELOCITY BEFORE IMPACT.

TABLE

RHO0	US	UP	P	V/V0	ST	U(ST)
11.34	4.92	1.88	1049	0.6180	AL	5.60
-	6.07	2.76	1900	0.5453	FE	5.60
-	7.74	4.18	3669	0.4600	FE	8.64
-	8.26	4.45	4170	0.4613	FE	9.10
-	11.36	7.00	9150	0.3750	FE	14.68

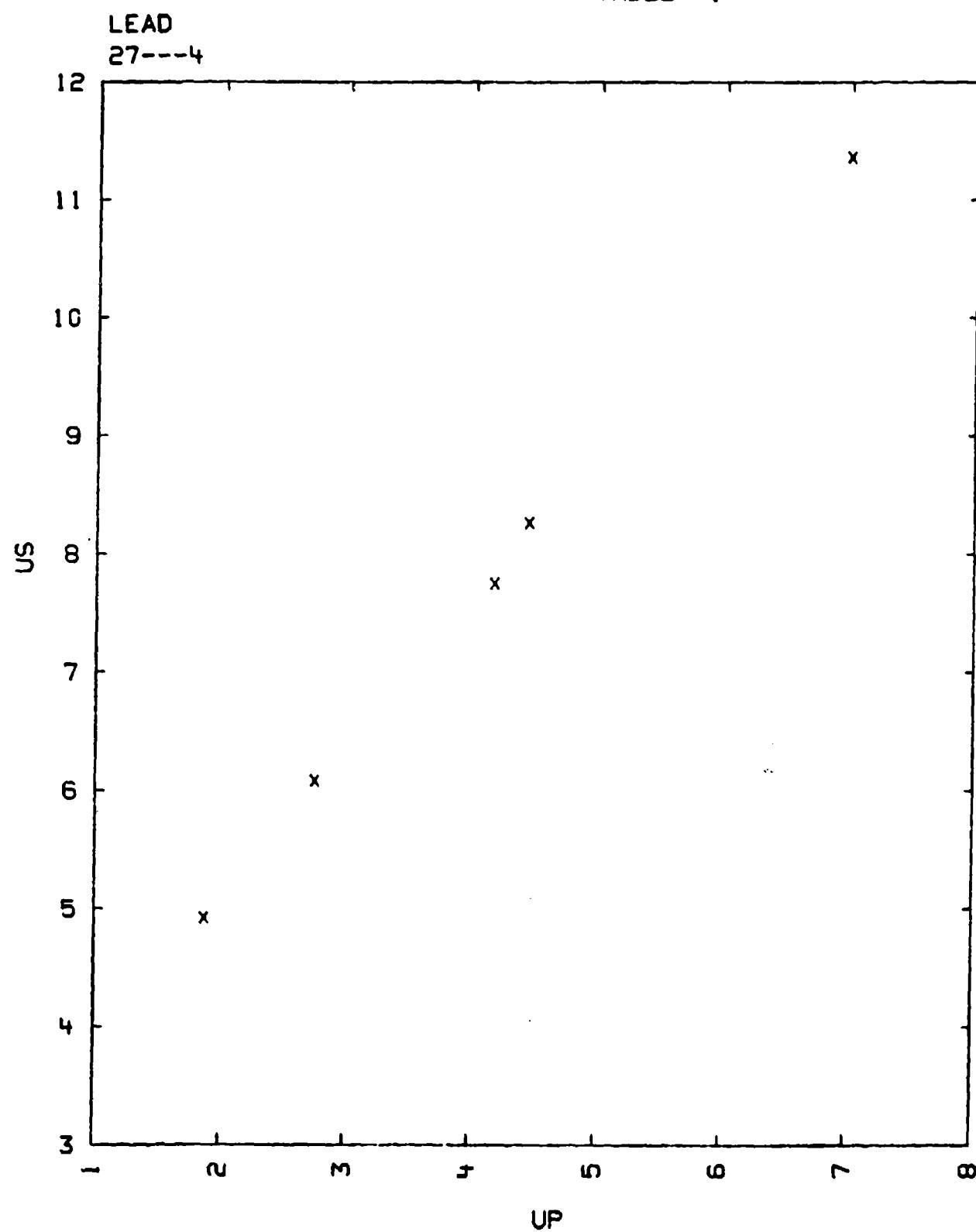
$$US = 2.58 + 1.26 UP \quad \text{FOR UP BETWEEN 1.88 TO 7.00 KM/SEC.}$$

SIGMA US = 0.08 KM/SEC.

COMMENTS:

- 1) SOURCE: AL'TSHULER, L.V., KORMER, S.B., BAKANOVA, A.A.
AND TRUNIN, R.F.
SOVIET PHYS-JETP, VOL. 11, PAGE 573 (1960)
AL'TSHULER, L. V. ET AL
SOVIET PHYS.-JETP, VOL. 15, P. 65 (1962)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE A
- 3) THE PRESSURES WERE PRODUCED BY AN EXPLOSIVELY ACCELERATED ALUMINUM OR STEEL PLATE AS INDICATED IN COLUMN 6.
- 4) THE INACCURACY IN THE DETERMINATION OF US AND UP FOR PRESSURES TO 4187 KILOBARS DID NOT EXCEED 1.0 PERCENT. EACH POINT IS THE AVERAGE OF 4 TO 6 EXPERIMENTS.
- 5) THE U UNCERTAINTY OF THE VALUES IN THE LAST TABLE ENTRY ARE:
1 PERCENT IN US OBTAINED FROM 6-8 EXPERIMENTS AND 1.5 PERCENT IN U(ST) OBTAINED FROM 12 EXPERIMENTS.
- 6) V01 WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK (MCGRATH-HILL BOOK CO. 1963) 2ND ED.
THE CORRESPONDING LATTICE PARAMETERS FOR THE FACE CENTERED CUBIC CELL IS 4.9496 ANGSTROMS.

TABLE I



27---5
LEAD POROUS

PAGE I

PB 99.5 PERCENT
AVERAGE GRAIN SIZE IS 8 MICRONS

VOI = 0.08818 CC/G.
VO = 1/RHO0 SEE TABLE

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURES IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL AND THE MATERIAL OF THE STRIKER PLATE. U(ST) IS THE STRIKER VELOCITY BEFORE IMPACT. ST LABELS THE HUGONIOT VARIABLES OF THE IRON SAMPLE HOLDER.

TABLE

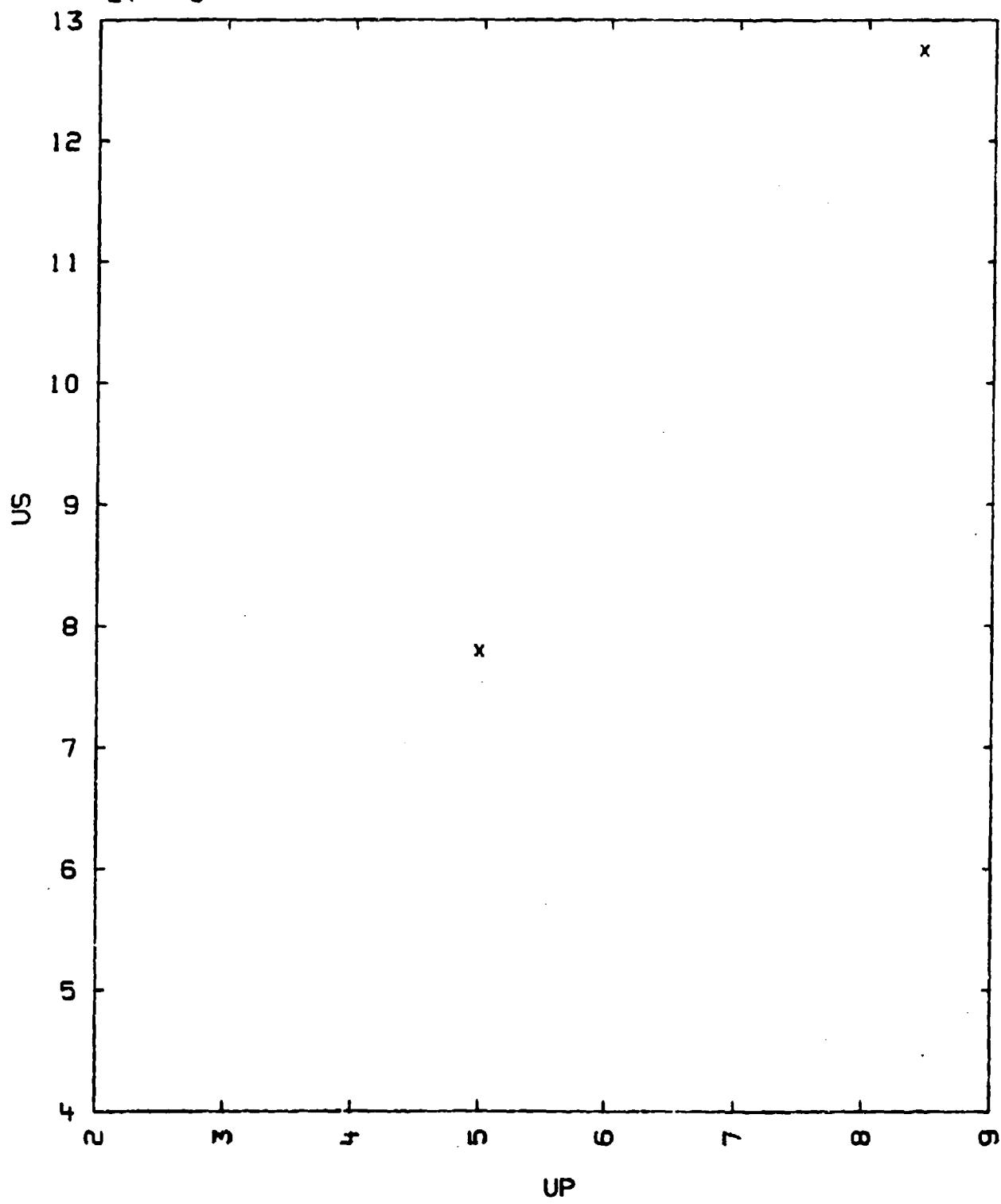
SAMPLE						DRIVER SYSTEM				
RHO0	US	UP	SIGMA	P	V/VO	ST	U(ST)	US(ST)	UP(ST)	P(ST)
US										
6.79	7.79	4.995	.04	2642	.5988	FE	8.64	10.67	4.26	3568
6.79	12.74	8.439	.16	7300	.5637	FE	15.45			
US =										

COMMENTS:

- 1) SOURCE: KORMER, S.B., FUNTIKOV, A.L., URLIN, V.D.
AND KOLESNIKOVA, A.N.
SOVIET PHYS.-JETP, VOL. 15, P. 477 (1962)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
- 3) US IS THE AVERAGE DETERMINATION TAKEN FROM 4-8 EXPERIMENTS, EACH
CONSISTING OF 4-6 READINGS.
- 4) VOI WAS OBTAINED FROM THE HANDBOOK OF CHEMISTRY AND PHYSICS (THE
CHEMICAL RUBBER PUBLISHING CO., CLEVELAND, OHIO, 1962-1963) 44TH ED.

TABLE I

LEAD POROUS
27---5



27---6
LEAD

PB

$$V_0 = 0.08826 \text{ g/cc.}$$

$$V_{01} = 0.08818 \text{ g/cc.}$$

THE TABLE BELOW GIVE THE VELOCITIES OF RELAXATION WAVES, C, AT VARIOUS PRESSURES. THE HUGONIOT STATE THE RELAXATION WAVE TRAVELS THROUGH IS GIVEN BY US, UP, P, OR V/V0. DENSITY IS GIVEN IN G/CC., VELOCITY IN KM/SEC. AND PRESSURE IN KILOBARS. ST DESIGNATES THE MATERIAL OF THE STRIKER PLATE, U(ST) IS THE VELOCITY BEFORE IMPACT AND US(ST) IS THE SHOCK VELOCITY IN THE PLATE AFTER IMPACT.

TABLE

SOUND VELOCITY
PARALLEL TO SHOCK DIRECTION

RH00	---HUGONIOT STATE---					--DRIVER SYSTEM--		
	US	UP	P	V/V0	C	ST	U(ST)	US(ST)
11.34	4.91	1.89	1052	0.6150		AL	5.60	10.34
-	4.16	1.20	656	0.6658	4.24	AL	5.60	10.34
-	3.72	1.10	464	0.7042	3.85	AL	5.60	10.34
-	6.17	2.82	1973	0.5429		FE	5.71	8.61
-	5.53	2.34	1467	0.5767	5.16	FE	5.71	8.61
-	4.83	1.84	1008	0.6188	4.62	FE	5.71	8.61
-	4.41	1.56	780	0.6472	4.29	FE	5.71	8.61
-	7.92	4.28	3850	0.4593	6.56	FE		
-	7.01	3.51	2795	0.4993	5.92	FE		

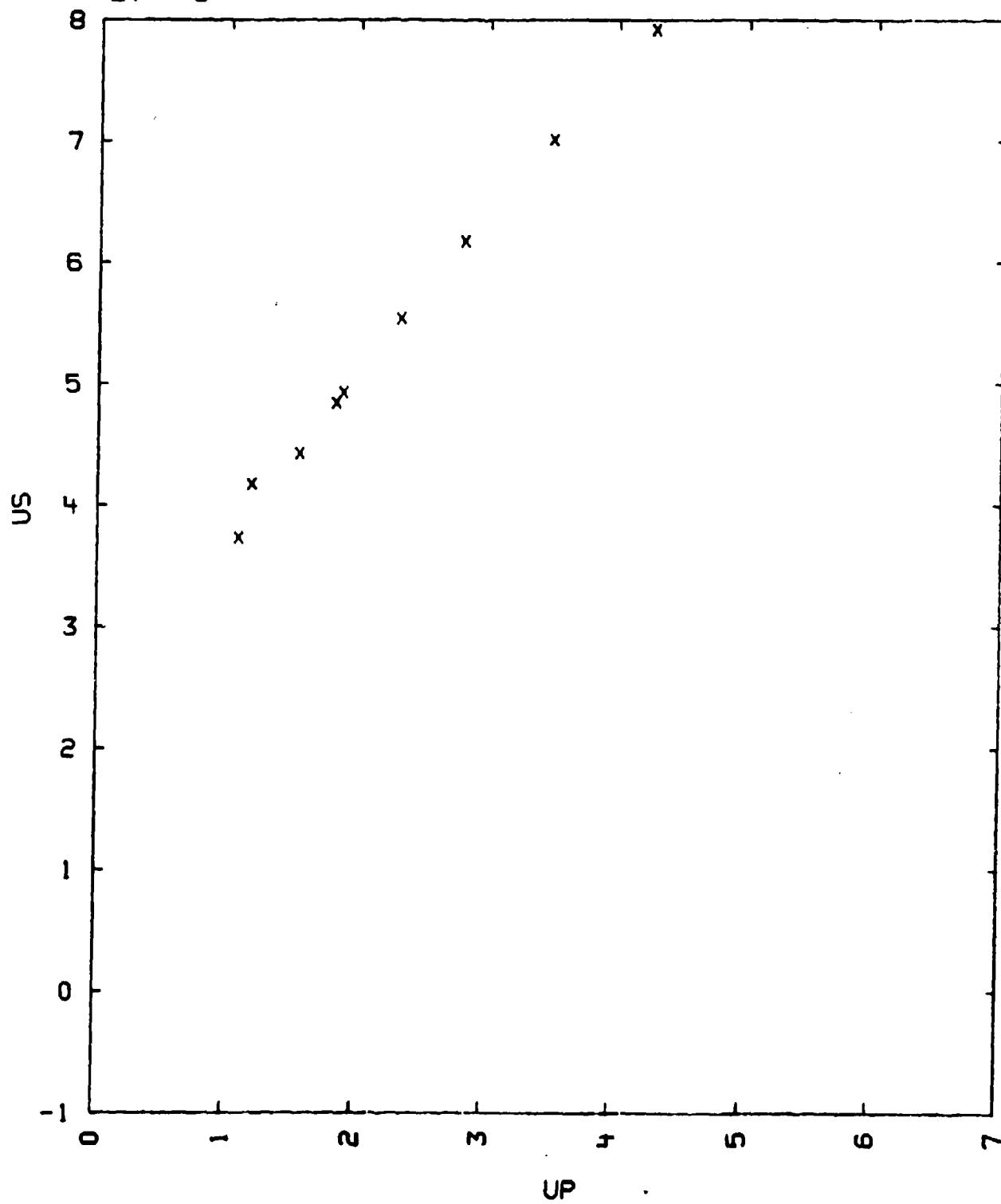
US =

COMMENTS:

- 1) SOURCE: AL'TSHULER, L. V., KORMER, S. B., BRAZHNICK, M. I., VLADIMIROV, L. A. AND FUNTIKOV, A. I. SOVIET PHYS.-JETP, VOL. 11, P. 768 (1960)
- 2) EXPERIMENTAL TECHNIQUE A WAS USED FOR THE FIRST THREE ENTRIES. FOR ALL OTHER ENTRIES THE EXPERIMENTAL TECHNIQUE B WAS USED. THE HUGONIOT STATE WAS DETERMINED FROM ONE MEASURED PARAMETER AND THE KNOWN EQUATION OF STATE. AL'TSHULER ET AL. SOVIET PHYS-JETP, VOL. 11, P. 573 (1960).
- 3) THE SOUND VELOCITY WAS DETERMINED FROM THE ATTENUATION IN THE MEASURED SHOCK OR FREE SURFACE VELOCITY PRODUCED BY THE RAREFACTION WAVE ORIGINATING AT THE REAR SURFACE OF THE STRIKER PLATE.
- 4) VO1 WAS OBTAINED FROM THE HANDBOOK OF CHEMISTRY AND PHYSICS (THE CHEMICAL RUBBER PUBLISHING CO., CLEVELAND, OHIO, 1962-1963) 44TH ED.

TABLE I

LEAD
27---6



27---7

LEAD

PB

$$V_0 = 0.08811 \text{ CC/G}$$

$$V_{OI} = 0.088176 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC.
AND PRESSURE IN KILOBARS.

TABLE

RH00	SAMPLE				BASE PLATE		
	US	UFS	UP	P	V/V0	MATERIAL	PRESSURE
11.35	4.66	3.77	1.74	920	0.526	AL	618
11.35	2.91	1.22	0.59	195	0.797	AL	146
11.35	3.70	2.30	1.09	457	0.705	AL	324
11.35	2.63	0.82	0.40	120	0.848	BRASS	133
11.34	2.45	0.61	0.30	83	0.879	BRASS	95

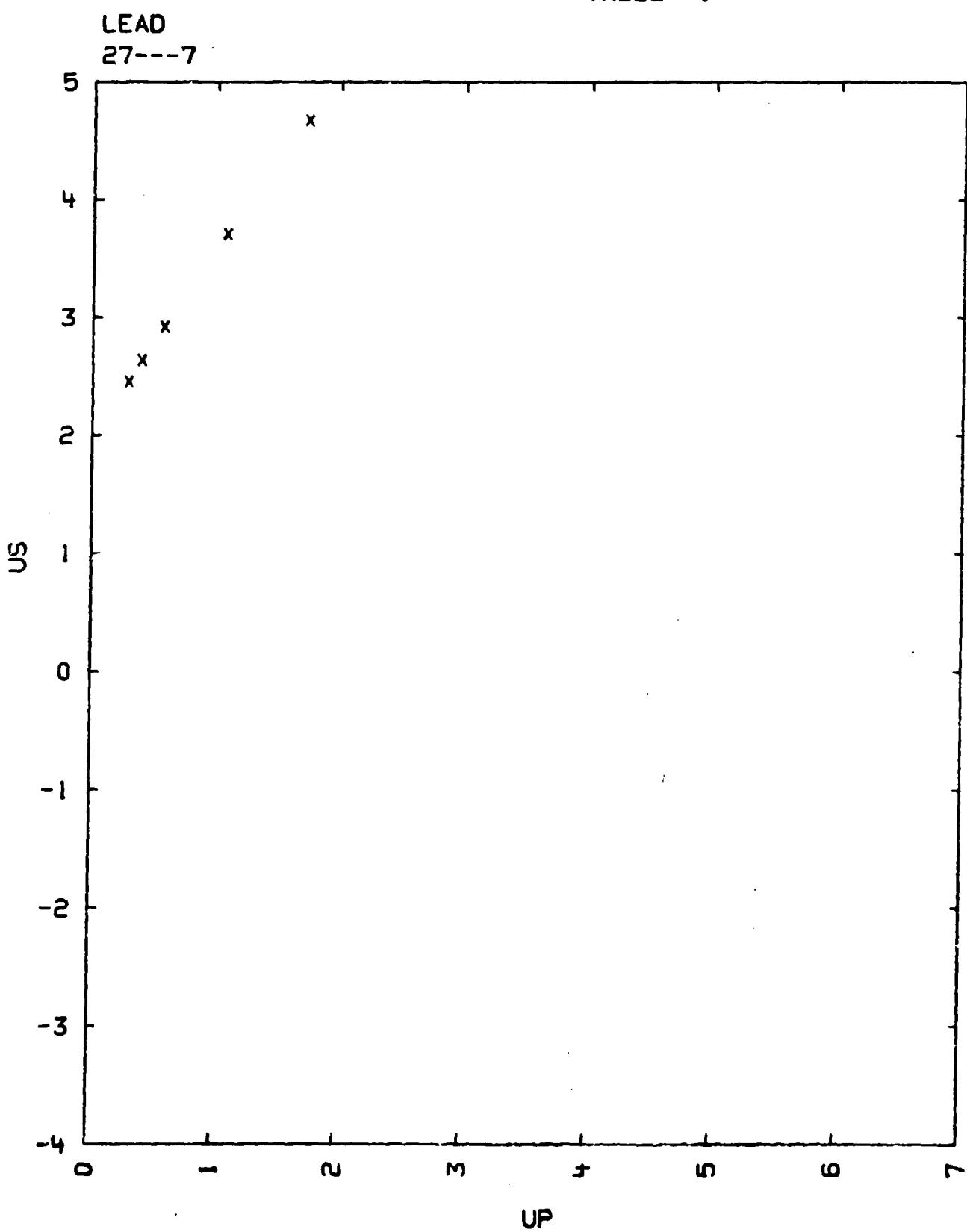
$$US = 2.01 + 1.53 UP \text{ KM/SEC}$$

$$\text{SIGMA US} = 0.019 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: COMPILER
L.R.L. EQUATION OF STATE FILE
LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF VOI WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES
(DOVER PUBLICATIONS, INC., NEW YORK, N. Y., 1963).

TABLE I



27---B
LEAD

CU, FE, ZN, BI, AG AND OTHERS LESS THAN 0.082 WT PERCENT
PB
REMAINDER - - -

$V_0 = 0.08807 \text{ g/cc}$
 $V_{01} = 0.08816 \text{ g/cc}$

$C_0 = 1.972 \text{ km/sec.}$

THE TABLE LISTS ALL THE PRESSURES AND COMPRESSIONS OF THE MULTIPLE WAVE SYSTEMS USED IN THE DATA ANALYSIS OF THESE RELATIVELY BROAD SHOCK WAVES, TOGETHER WITH THE HYDROSTATIC PRESSURE (PH) AND THE APPROXIMATE USI AND UPI VALUES CALCULATED FROM THEM. DENSITY IS IN G/CC, VELOCITIES IN KM/SEC AND PRESSURE OR STRESS IN Kbars. NO = EXPERIMENT NUMBER. 1 INDICATES THE 1ST, 2ND, 3RD ETC. WAVE OF THE SYSTEM. U = PROJECTILE VELOCITY AND MAT IS THE PROJECTILE MATERIAL.

TABLE

U	RHO0	US	UP	P	V/V0	I	PH	MAT
.0401	11.355	2.3	0.00093	.25	0.9996	1		PB
		1.94	0.0092	2.08	0.9953	2		
		1.85	0.0179	3.90	0.9906	3		
		2.	0.0191	4.20	0.9901	4	4.15	
.0599	-	2.3	0.00091	0.24	0.9996	1		-
		1.96	0.0089	2.01	0.9955	2		
		1.95	0.0190	4.23	0.9903	3		
		1.91	0.0285	6.28	0.9853	4	6.22	
.1264	-	2.08	0.0438	10.13	0.9785	1		
		1.76	0.0523	11.82	0.9736	2	11.75	-
.3025	-	2.15	0.1146	27.95	0.9466	1	27.96	AL 6061-T6
.5020	-	2.26	0.1834	47.24	0.9191	1	47.13	- -
.4270	-	2.37	0.2546	58.55	0.8926	1	58.42	NI
.4940	-	2.44	0.2822	76.12	0.8842	1	77.99	-

US = SEE COMMENT 3

COMMENTS

- 1) SOURCE: MUNSON, D. E. AND BARKER, L. M.
J. APPL. PHYS., VOL. 37, P. 1652, (1966)
SANDIA CORPORATION, ALBUQUERQUE, NEW MEXICO.
- 2) EXPERIMENTAL TECHNIQUE F
DATA REDUCTION METHOD D WITH $Z_{UP} = UFS$

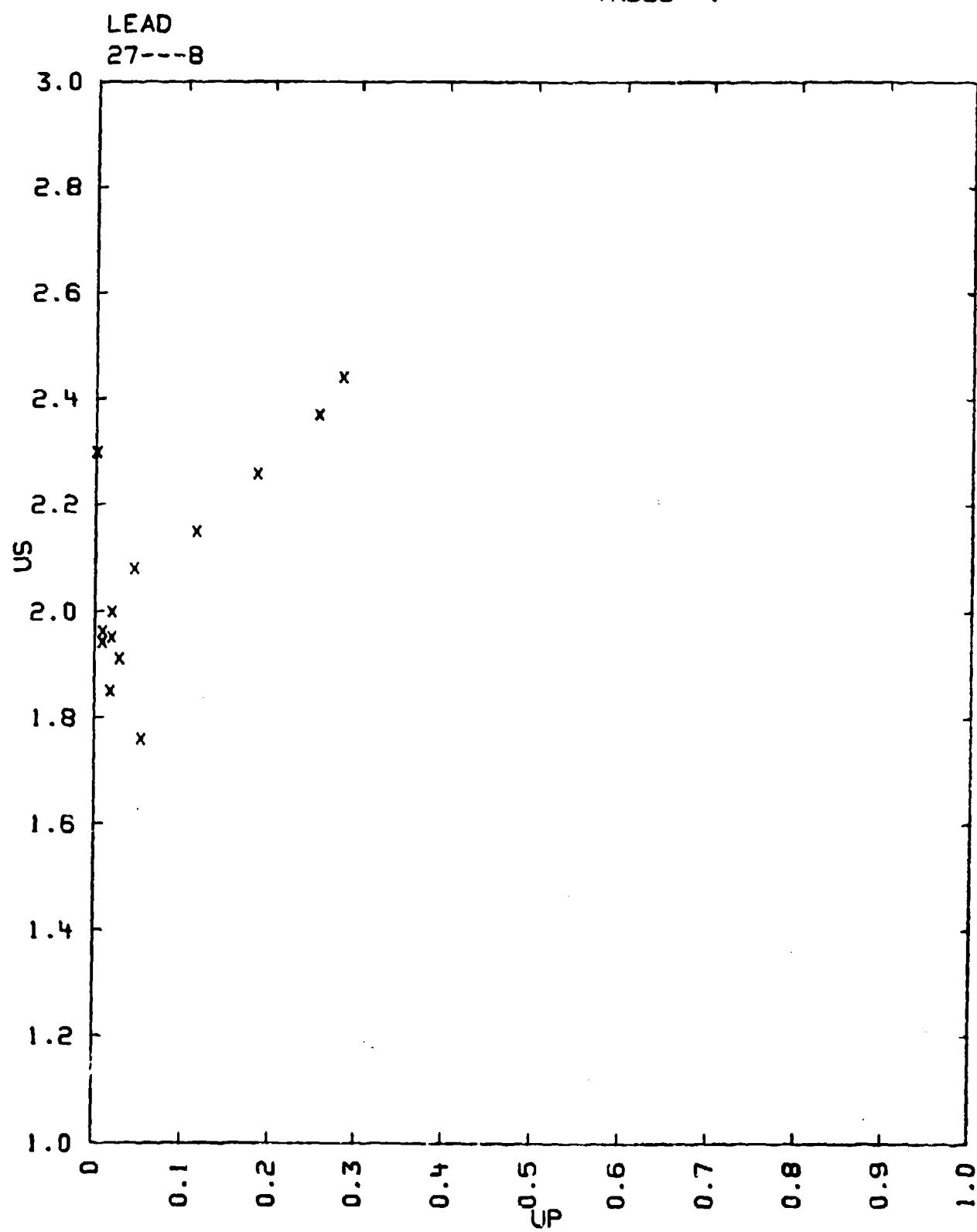
WITH THE MODIFICATION THAT A SMOOTH BROAD WAVE IS APPROXIMATED BY A SET OF DISCREET PRESSURE STEPS. THE SHOCK VELOCITY MEASUREMENTS WERE CORRECTED FOR ERRORS INTRODUCED BY SURFACE INTERACTIONS.

- 3) NOTE THAT US_I AND UP_I ARE ONLY APPROXIMATE NUMBERS CALCULATED FROM THE PI AND VI/V₀ VALUES TO ILLUSTRATE THE WAVE THICKNESS. IN THE FIT BELOW US_I IS OBTAINED FROM THE MAXIMUM VALUE OF PH OF THE PARTICULAR WAVE SYSTEM AND THE CORRESPONDING VALUE OF VI, USING THE EQUALITY US_I = V₀(PH/(V₀-VI))^{1/2}, WHILE UP_I WAS TAKEN TO BE 1/2 THE MAXIMUM FREE SURFACE VELOCITY. THESE POINTS ARE COMBINED WITH DATA FROM LOS ALAMOS SCIENTIFIC LABORATORY TO GIVE:

US_I = 1.952 + 1.562·UP_I KM/SEC, UP_I FROM 0 TO 2.3 KM/SEC.
MAX. DEVIATION 3.4 PERCENT

- 4) PH = PI - 2/3·Y, WHERE THE YIELD POINT Y IS A FUNCTION OF THE DEGREE OF WORK HARDENING.
5) V₀I HAS CALCULATED FROM THE CUBIC UNIT CELL CONSTANT GIVEN IN HYCKOFF, CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, 1963)
A = 4.9505 ANGSTROM.

TABLE I



29---1
ALUMINUM

AL

$$V_0 = 0.370 \text{ CC/G} \quad C_L = 6.260 \text{ KM/SEC}$$

$$V_{01} = 0.370 \text{ CC/G} \quad C_S = 3.080 \text{ KM/SEC}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN MM/MICRO-SEC, AND PRESSURE IN KILOBARS.

TABLE

RHO0	US	UP	P	V/V0
2.70	8.74	2.70	640	.691
-	9.20	2.58	641	.719
-	9.48	2.60	665	.726
-	10.14	3.53	974	.652
-	10.04	3.58	977	.643
-	10.24	3.45	958	.606
-	10.67	3.56	1030	.665
-	11.95	5.13	1650	.571
-	13.09	6.38	2250	.513
-	13.26	6.73	2410	.493
-	13.67	7.02	2590	.487

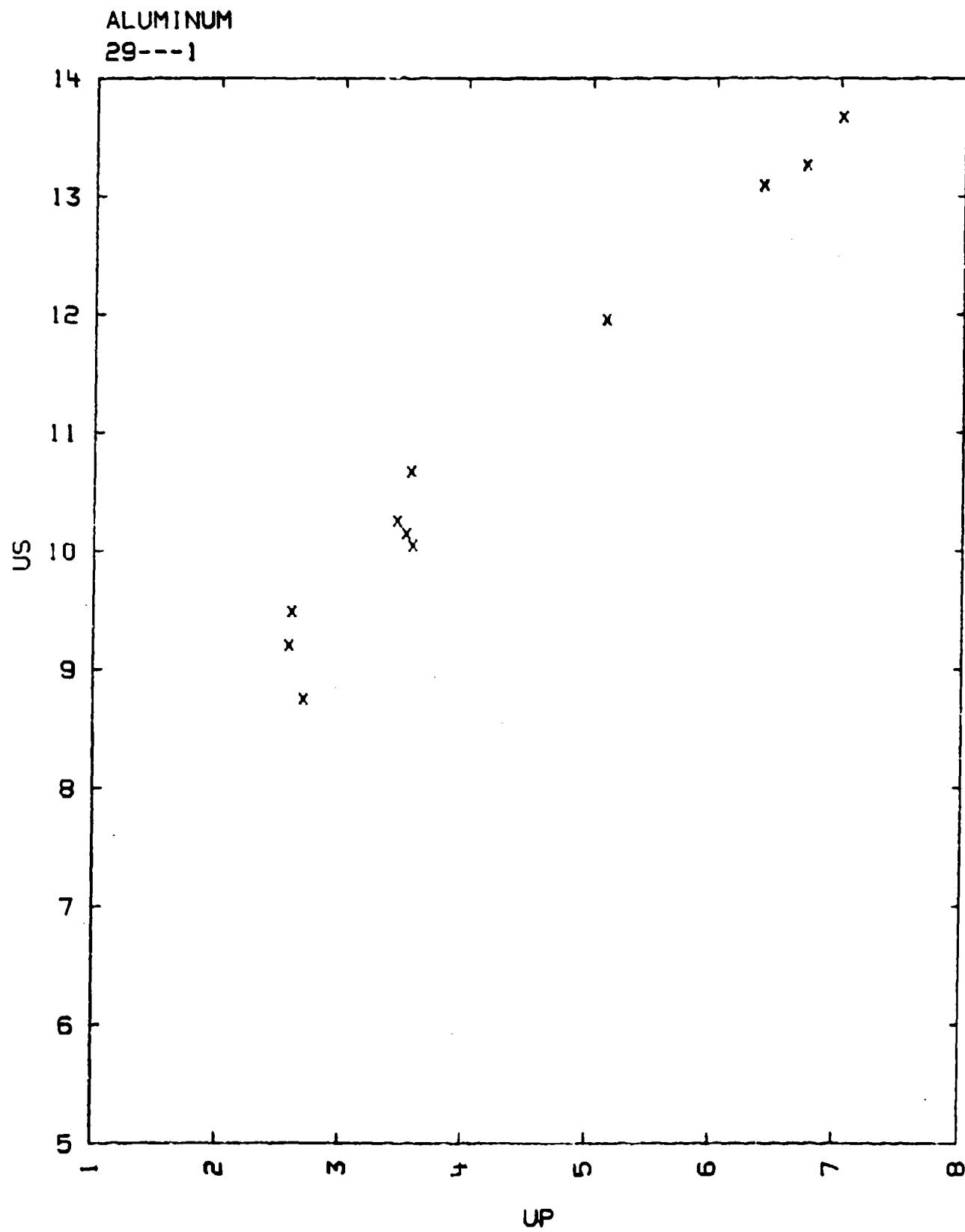
$$US = 5.38 + 1.337UP \text{ MM/MICROSEC} \quad \text{FOR UP LESS THAN } 3.24$$

$$US = 5.38 + 1.337UP - 0.0830(UP - 3.24)^{0.2} \text{ MM/MICROSEC} \quad \text{FOR UP GREATER THAN } 3.24$$

COMMENTS:

- 1) SOURCE: SKIDMORE, I.C. AND MORRIS, E.
THERMODYNAMICS OF NUCLEAR MATERIALS, P. 173 FF. (1962)
INTERN. AT. ENERGY AGENCY, VIENNA
ATOMIC WEAPONS RESEARCH ESTABLISHMENT, ALDERMASTON, ENGLAND
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE A AND B
THE SAMPLES WERE POSITIONED ON A TARGET PLATE OF STEEL (MTRL. 41---2)
AND THE PRESSURE WAS PRODUCED BY AN EXPLOSIVELY ACCELERATED PLATE OF
STEEL.
- 3) THE VELOCITY OF THE FLYING PLATE AND THE SHOCK AND SURFACE VELOCITY
OF THE TARGET PLATE WERE MEASURED AS WELL AS THE SAMPLE SURFACE AND
SHOCK VELOCITIES.
- 4) DATA SCATTER WAS ABOUT 0.03 MICROSEC.
- 5) CORRECTIONS WERE MADE FOR FLYING PLATE CURVATURE OF UP TO 1 MICROSEC.
- 6) THE HIGHER PRESSURES WERE OBTAINED BY A SPHERICALLY CONVERGING
SYSTEM.
- 7) CL AND CS FROM L. BERGMAN: DER ULTRASCHALL. S. HIRZEL VERLAG, STUTTGART, (1954).

TABLE I



29--2
ALUMINUM

AL 97.9 PERCENT
MG 1.0 PERCENT
SI 0.6 PERCENT
CU 0.25 PERCENT
CR 0.25 PERCENT

$$V_0 = 0.3707 \text{ CC/G.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC..
PRESSURE IN KBARS AND DENSITY IN G/CC. D IS SAMPLE THICKNESS IN MM.
PROJ. DESIGNATES THE PROJECTILE MATERIAL: ALUMINUM(AL) OR 4340 STEEL(FE)

TABLE I

NO	RHO0	US1	UP1	UFS1	US2	UP2	UFS2	D
1	2.698	5.99	0.010	0.021				2.5
2	-	6.09	0.014	0.030				2.5
3	-	6.05	0.026	0.054				1.25
4	-	5.84	0.031	0.063				1.25
5	-	6.02	0.031	0.064				2.5
6	-	6.23	0.038	0.075				2.5
7	-	6.23			5.47	0.043	0.087	2.5
8	-	6.27			5.87	0.044	0.086	2.5
9	-	5.21			5.60	0.046	0.088	1.25
10	-	6.49			5.78	0.047	0.089	2.5
11	-	6.33			5.36	0.060	0.118	1.25
12	-	6.19			5.36	0.063	0.118	2.5
13	-	6.39			5.32	0.074	0.135	1.25
14	-	6.39			5.53	0.074	0.128	2.5
15	-	6.31			5.42	0.075	0.131	2.5
16	-	6.23			5.48	0.075	0.126	2.5
17	-	6.31			5.50	0.088	0.152	2.5
18	-	6.28			5.40	0.088	0.155	2.5
19	-	6.34			5.50	0.092	0.173	1.25
20	-	6.54			5.51	0.092	0.169	1.25
21	-	6.18			5.28	0.097	0.181	1.25
22	-	6.51			5.59	0.102	0.162	2.5
23	-	6.20			5.34	0.102	0.187	1.25
24	-	5.96			5.25	0.116	0.217	1.25
25	-	6.30			5.47	0.116	0.186	2.5
26	-	6.23			5.37	0.121	0.213	2.5
27	-	6.23			5.36	0.122	0.232	1.25
28	-	6.14			5.25	0.123	0.235	1.25
29	-	6.25			5.41	0.130	0.229	2.5
30	-	6.22			5.36	0.136	0.258	1.25
31	-	6.33			5.42	0.136	0.244	2.5
32	-	6.35			5.47	0.141	0.245	2.5
33	-	6.38			5.48	0.145	0.252	2.5
34	-	6.38			5.47	0.154	0.261	2.5
35	-	6.26			5.49	0.166	0.317	1.25

ALUMINUM

NO	RHO0	US1	UP1	UFS1	US2	UP2	UFS2	D	
36	-	6.25				5.45	0.184	0.353	1.25
37	-	6.19				5.44	0.208	0.401	1.25

US -

TABLE II

NO	P1	V1/V0	P2	V2/V0	PROJ.
1	1.58	0.9985			AL
2	2.42	0.9977			-
3	4.42	0.9958			-
4	5.20	0.9950			-
5	5.32	0.9949			-
6	6.33	0.9940			-
7			7.12	0.9930	-
8			7.43	0.9928	-
9			7.59	0.9925	-
10			7.81	0.9923	-
11			9.58	0.9898	-
12			10.01	0.9892	-
13			11.50	0.9872	-
14			11.78	0.9873	-
15			11.80	0.9870	-
16			11.88	0.9871	-
17			13.79	0.9848	-
18			13.76	0.9846	-
19			14.27	0.9842	-
20			14.32	0.9842	-
21			14.74	0.9828	-
22			15.96	0.9825	-
23			15.53	0.9820	-
24			17.38	0.9791	-
25			17.98	0.9796	-
26			18.40	0.9784	-
27			18.52	0.9782	-
28			18.37	0.9777	-
29			19.81	0.9768	-
30			20.60	0.9755	-
31			20.77	0.9757	-
32			21.42	0.9752	FE
33			22.22	0.9743	AL
34			23.45	0.9727	-
35			25.29	0.9706	FE
36			27.79	0.9672	-
37			31.29	0.9627	-

COMMENTS:

- 1) SOURCE: LUNDERGAN, C. D. AND HERRMANN, W.
J. APPL. PHYS. VOL. 34 P. 2046 (1963)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE A
STANDARD MATERIAL 6061-T6 ALUMINUM
- 3) 2UP IS THE ALUMINUM PROJECTILE VELOCITY. -- INDICATES EXPERIMENTS IN
WHICH 4340 STEEL PROTECTILES WERE USED
- 4) THE DIFFERENCE BETWEEN THE PROJECTILE VELOCITY (2UP) AND THE MEASURED
UPS BEHIND THE SAMPLE RANGES FROM 0.001 AT 2.42 KILOBARS TO 0.047 AT
6.67 KILOBARS.

TABLE I

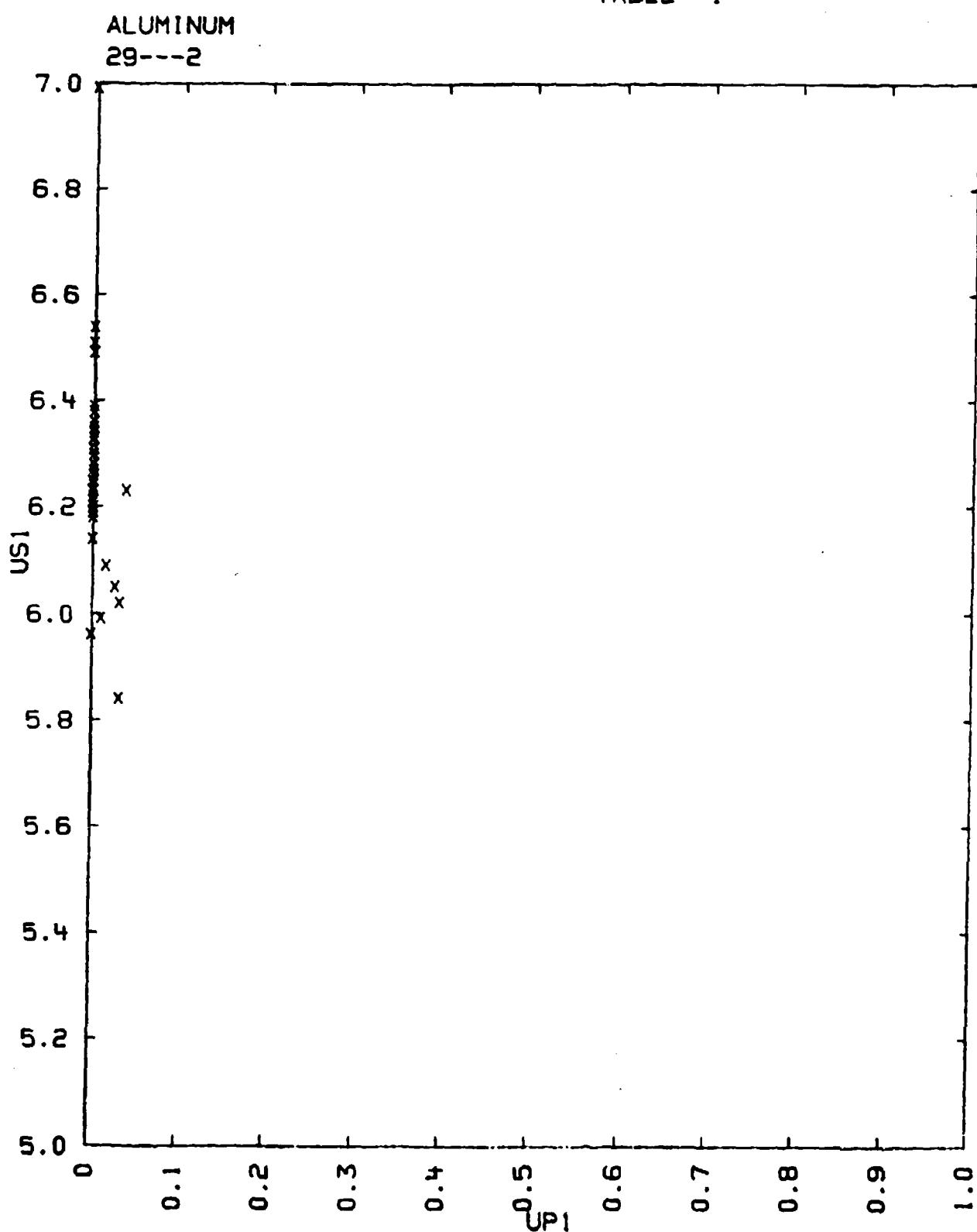
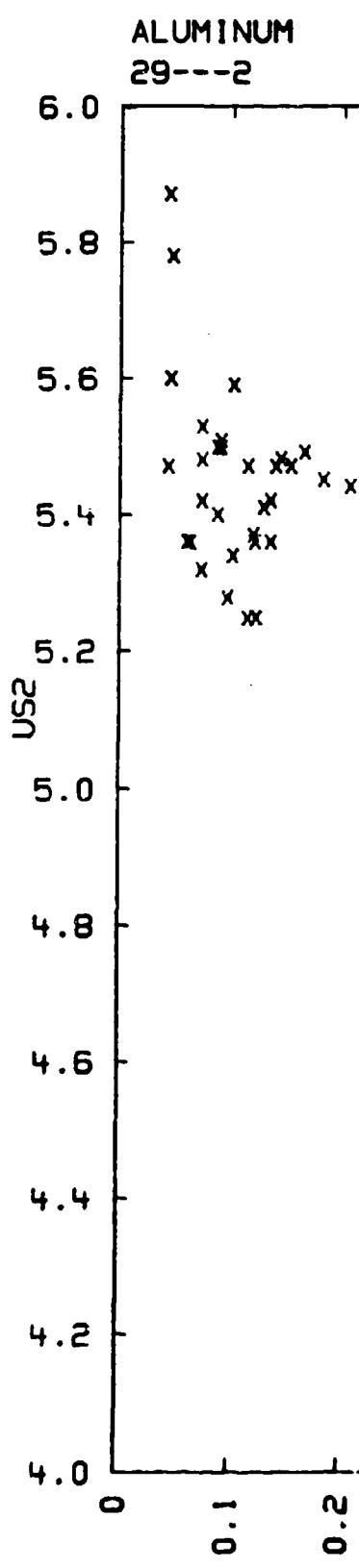


TABLE I



29---3
ALUMINUM

AL

$$V_0 = 0.369 \text{ CC/G.}$$

$$V_{01} = 0.3715 \text{ CC/G.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL USED AS STRIKER. U(ST) IS THE VELOCITY OF STRIKER BEFORE IMPACT

TABLE

RHO0	US	UP	P	V/V0	ST	V(ST)
2.71	9.13	2.80	693	0.6935	AL	5.60
-	10.39	3.70	1042	0.6439	FE	5.60
-	12.94	5.62	1972	0.5659	FE	8.64

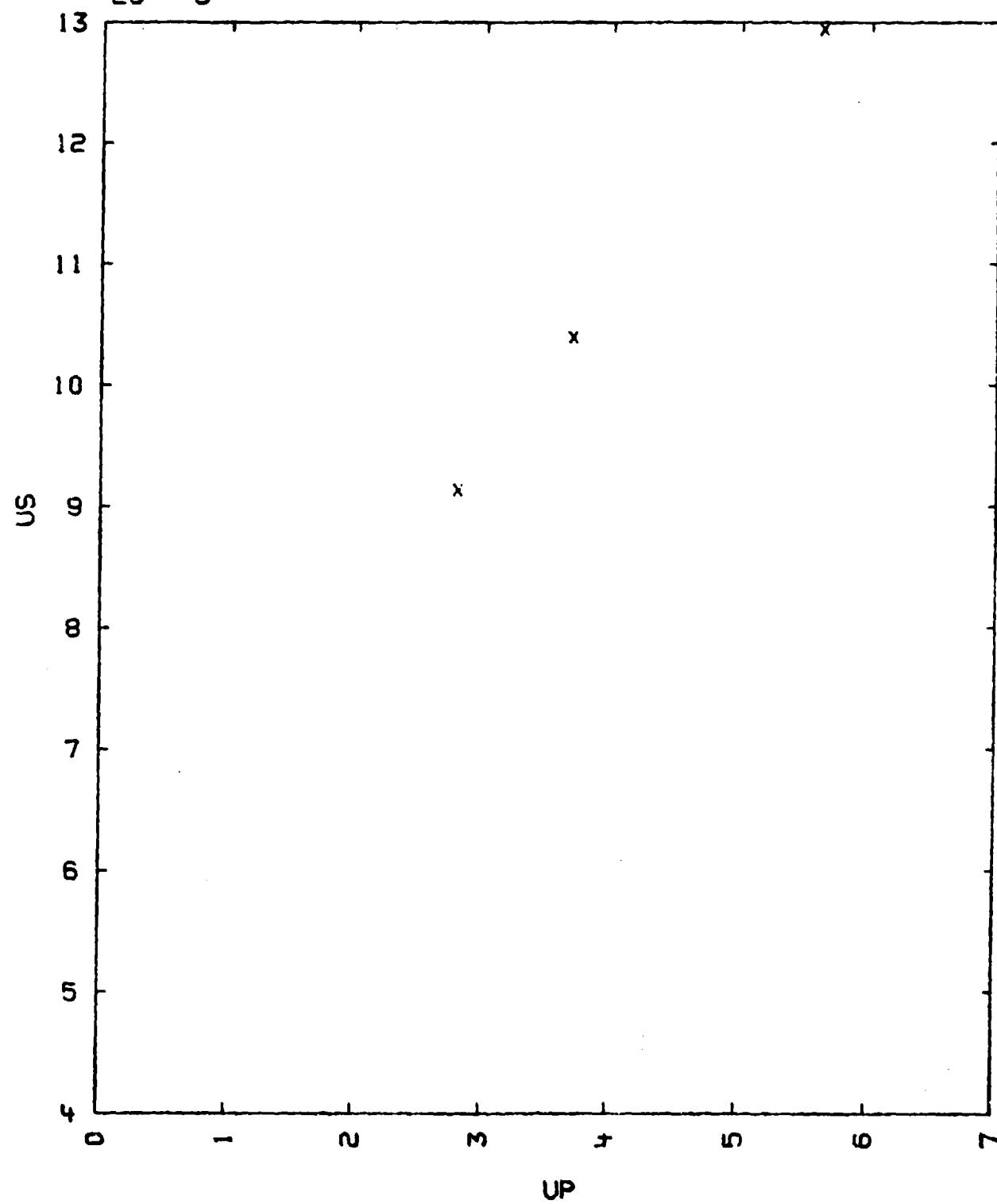
US = $5.38 + 1.35 \text{ UP KM/SEC.}$ FOR UP BETWEEN 2.80 TO 5.62 KM/SEC.
 $\Sigma \Delta US = 0.04 \text{ KM/SEC.}$

COMMENTS:

- 1) SOURCE: AL'TSHULER, L.V., KORMER, S.B., BAKANOVA, A.A.
AND TRUNIN, R.F.
SOVIET PHYS.-JETP, VOL. 11, P. 573 (1960)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION A
- 3) THE PRESSURES WERE PRODUCED BY AN EXPLOSIVELY ACCELERATED ALUMINUM OR STEEL PLATE AS INDICATED IN COLUMN 6.
- 4) THE INACCURACY IN THE DETERMINATION OF US AND UP DID NOT EXCEED 1.0 PERCENT. EACH POINT IS THE AVERAGE OF 4 TO 6 EXPERIMENTS.
- 5) V01 WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK.
(MCGRAW-HILL BOOK CO. 1963) 2 ED.

TABLE I

ALUMINUM
29---3



29---4

ALUMINUM POROUS

AL 92.5 PERCENT
 AVERAGE GRAIN SIZE IS 1 MICRON

$\text{VOL} = 0.3701 \text{ CC/G.}$
 $\text{VO} = 1/\text{RHOO} \text{ SEE TABLE}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURES IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL AND THE MATERIAL OF THE STRIKER PLATE. U(ST) IS THE STRIKER VELOCITY BEFORE IMPACT. ST LABELS THE HUGONIOT VARIABLES OF THE IRON SAMPLE HOLDER.

TABLE

- - - - - SAMPLE - - - - -							- - DRIVER SYSTEM - -		
RHOO	US	SIG.US	UP	P	V/VOL	U(ST)	US(ST)	UP(ST)	
1.89	11.74	0.10	6.27	1391	0.6676	8.64	10.67	4.26	
1.30	11.42	0.09	6.76	1003	0.8503	8.64	10.67	4.26	
0.907	10.75	0.08	7.20	702	0.9852	8.64	10.67	4.26	
2.70	18.31	0.16	9.94	4930	0.4577	15.45			

US =

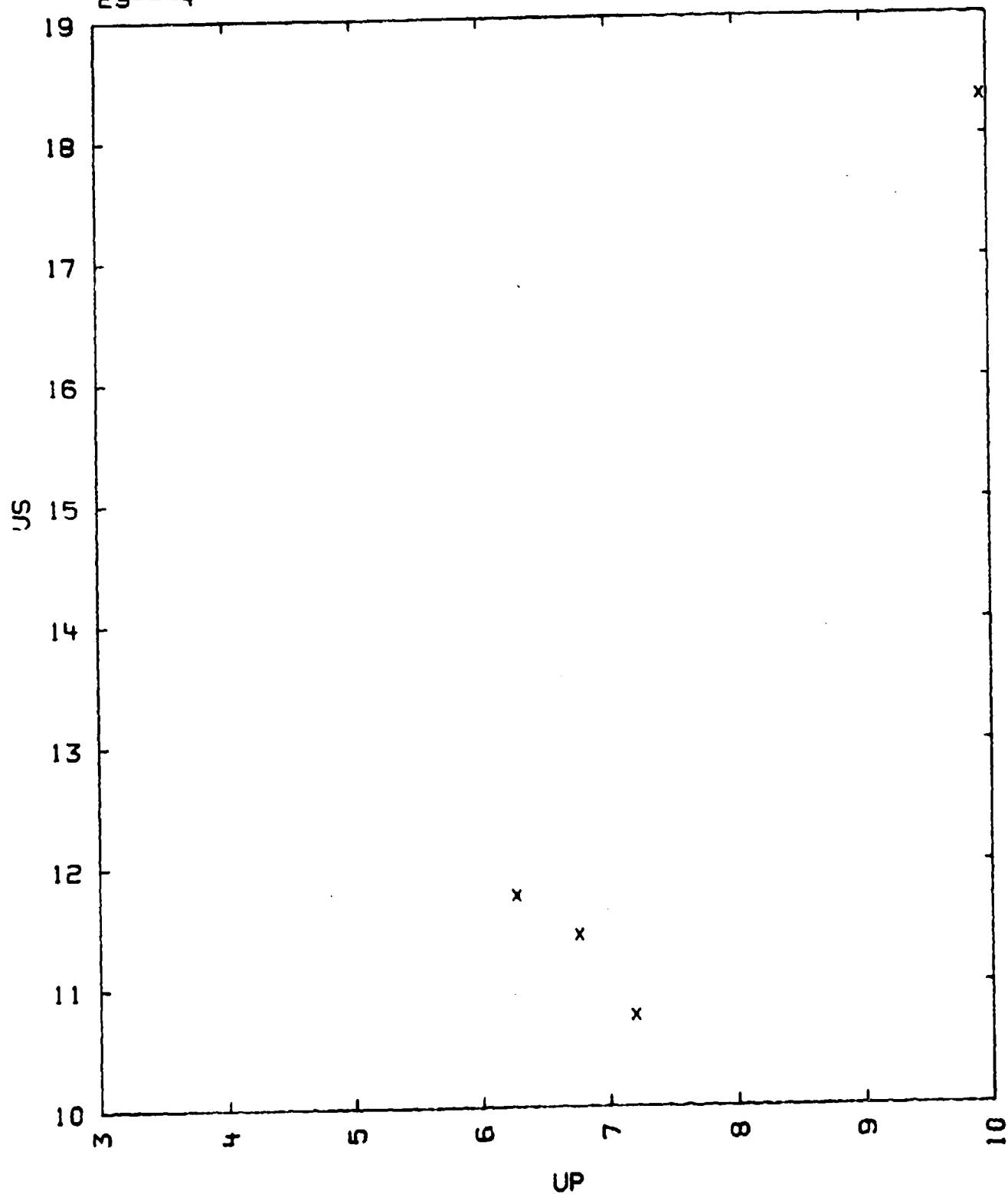
COMMENTS:

- 1) SOURCE: KORMER, S.B., FUNTIKOV, A.I., URLIN, V.D.
 AND KOLESNIKOVA, A.N.
 SOVIET PHYS.-JETP, VOL. 15, P. 477 (1962)
- 2) EXPERIMENTAL TECHNIQUE A
 DATA REDUCTION TECHNIQUE B
- 3) US IS THE AVERAGE DETERMINATION TAKEN FROM 4-8 EXPERIMENTS, EACH
 CONSISTING OF 4-6 READINGS.
- 5) VOL IS THE SPECIFIC VOLUME OF PURE ALUMINUM OBTAINED FROM THE
 HANDBOOK OF CHEMISTRY AND PHYSICS (THE CHEMICAL RUBBER PUBLISHING
 CO., CLEVELAND, OHIO, 1962-1963) 44TH ED.
- 6) UP WAS NOT GIVEN BY THE SOURCE BUT CALCULATED FROM THEIR TABULATION.

TABLE I

ALUMINUM POROUS

29---4



29---5
ALUMINUM

AL

$$V_0 = 0.369 \text{ g/cc.}$$

$$V_{01} = 0.370 \text{ g/cc.}$$

THE TABLE BELOW GIVES THE VELOCITIES OF RELAXATION WAVES, C, AT VARIOUS PRESSURES. THE HUGONIOT STATE THE RELAXATION WAVE TRAVELS THROUGH IS GIVEN BY US, UP, P, OR V/V0. DENSITY IS GIVEN IN G/CC., VELOCITY IN KM/SEC. AND PRESSURE IN KILOBARS. ST DESIGNATES THE MATERIAL OF THE STRIKER PLATE, U(ST) IS THE VELOCITY BEFORE IMPACT AND US(ST) IS THE SHOCK VELOCITY IN THE PLATE AFTER IMPACT.

TABLE

SOUND VELOCITY
PARALLEL TO SHOCK DIRECTION

RH00	---HUGONIOT STATE-----					--DRIVER SYSTEM--		
	US	UP	P	V/V0	C	ST	U(ST)	US(ST)
2.71	9.13	2.80	693	0.6935		AL	5.60	9.13
-	8.52	2.35	545	0.7231	8.84	AL	5.60	9.13
-	7.92	1.92	412	0.7576	8.13	AL	5.60	9.13
-	10.5	3.77	1072	0.6406		FE	5.71	7.06
-	10.2	3.54	975	0.6519	9.76	FE	5.71	7.06
-	12.9	5.59	1955	0.5679	11.74	FE		
-	12.0	4.93	1600	0.5879	11.23	FE		

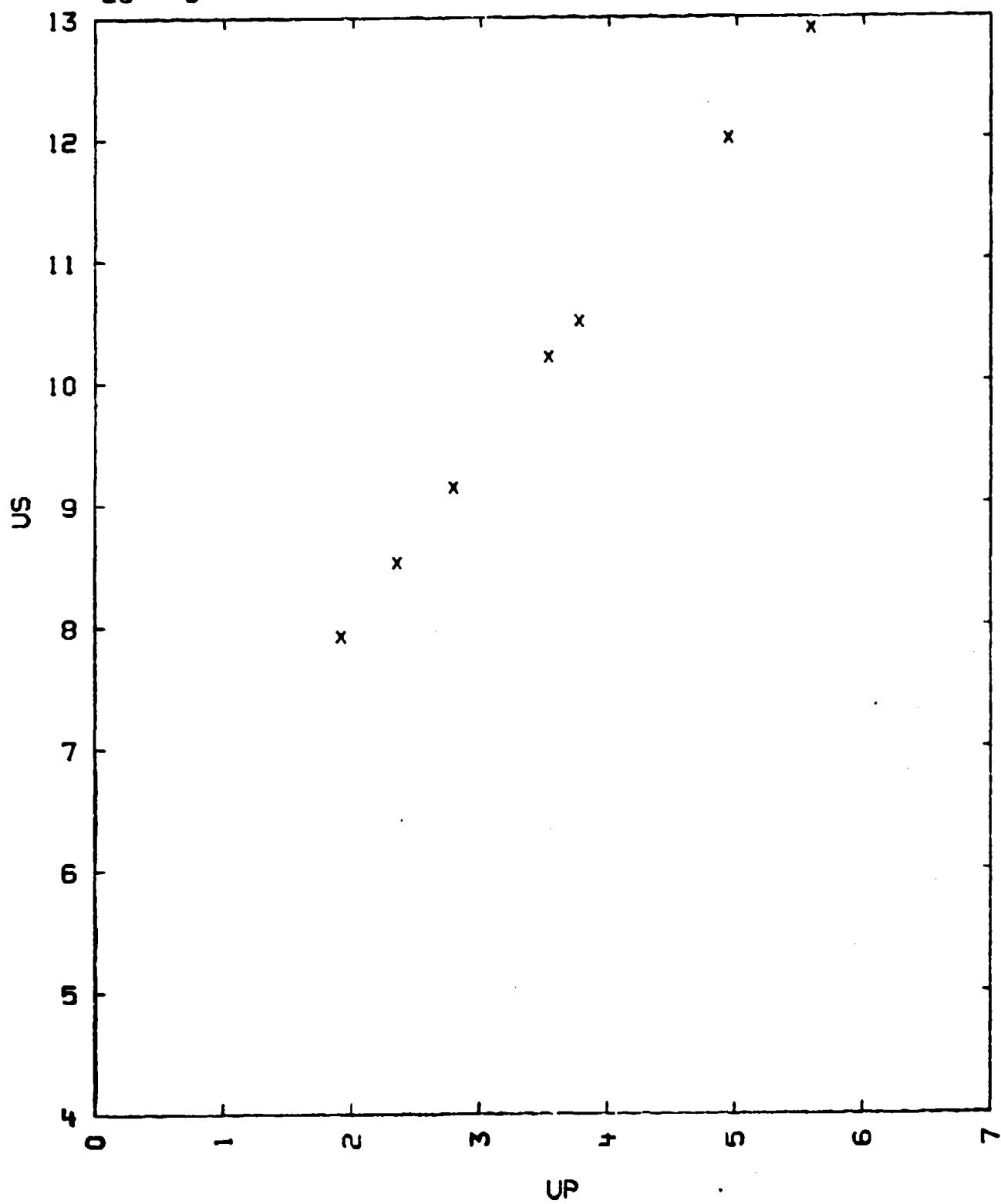
US =

COMMENTS:

- 1) SOURCE: AL'TSHULER, L. V., KORMER, S. B., BRAZNIK, M. I., VLADIMIROV, L. A. AND FUNTIKOV, A. I., SOVIET PHYS.-JETP, VOL. 11, P. 766 (1960)
- 2) EXPERIMENTAL TECHNIQUE A WAS USED FOR THE FIRST THREE ENTRIES. FOR ALL OTHER ENTRIES THE EXPERIMENTAL TECHNIQUE B WAS USED. THE HUGONIOT STATE WAS DETERMINED FROM ONE MEASURED PARAMETER AND THE KNOWN EQUATION OF STATE. AL'TSHULER ET AL, SOVIET PHYS-JETP, VOL. 11, P. 573 (1960).
- 3) THE SOUND VELOCITY WAS DETERMINED FROM THE ATTENUATION IN THE MEASURED SHOCK OR FREE SURFACE VELOCITY PRODUCED BY THE RAREFACTION WAVE ORIGINATING AT THE REAR SURFACE OF THE STRIKER PLATE.
- 4) THE MEASURED SOUND VELOCITY IS HIGHER THAN THE CALCULATED SOUND VELOCITY FOR A PURELY NONELASTIC SAMPLE.
- 5) VOI WAS OBTAINED FROM THE HANDBOOK OF CHEMISTRY AND PHYSICS (THE CHEMICAL RUBBER PUBLISHING CO. 1962-1963) 44TH ED.

TABLE I

ALUMINUM
29---5



29---6
ALUMINUM 2024 ALLOY

AL	93	PERCENT BY WEIGHT			
CU	3.8 - 4.9	-	-	-	
MG	1.2 - 1.8	-	-	-	
MN	0.3 - 0.9	-	-	-	
FE	< 0.5	-	-	-	
SI	< 0.5	-	-	-	
ZN	< 0.25	-	-	-	
CR	< 0.1	-	-	-	

$$V_0 = 0.3591 \text{ CC/G}$$

THE TABLE BELOW LISTS PRESSURE (P IN KILOBARS), PARTICLE-SHOCK- AND FREE SURFACE VELOCITY (UP, US AND UFS IN KM/SEC.) OF THE HUGONIOT. THIS HUGONIOT MAY BE CALCULATED FROM THE LINEAR FIT BELOW THE TABLE. THE OTHER COLUMNS ARE THE COEFFICIENTS OF THE FIT TO THE ISENTROPIC CURVES

$P = A_0 + A_1 \cdot (UFS-U) + A_2 \cdot (UFS-U)^2 + A_3 \cdot (UFS-U)^3 + A_4 \cdot (UFS-U)^4$
WHICH IN THE P VS U PLANE PASS THROUGH THE HUGONIOT AT THE LISTED POINTS.
THE DENSITY AT THE FOOT OF THE HUGONIOT IS 2.785 G/CC.

TABLE

US	UP	UFS	P	A0	A1	A2	A3	A4
5.489	0.100	0.200	15.3	0.0	149.38	36.73	-0.160	0.170
6.834	1.100	2.210	209.4	0.0	147.71	36.89	-0.291	0.183
7.910	1.900	3.848	418.6	0.0	143.57	36.75	-0.456	0.196
8.717	2.500	5.168	606.9	-9.25	135.60	35.14	-0.169	0.149
9.524	3.100	6.566	822.3	-23.66	122.26	34.25	-0.229	0.140
10.197	3.600	7.752	1022.3	-33.53	110.97	33.43	-0.283	0.131

$$US = 5.3550 + 1.345 UP \text{ KM/SEC.}$$

COMMENTS:

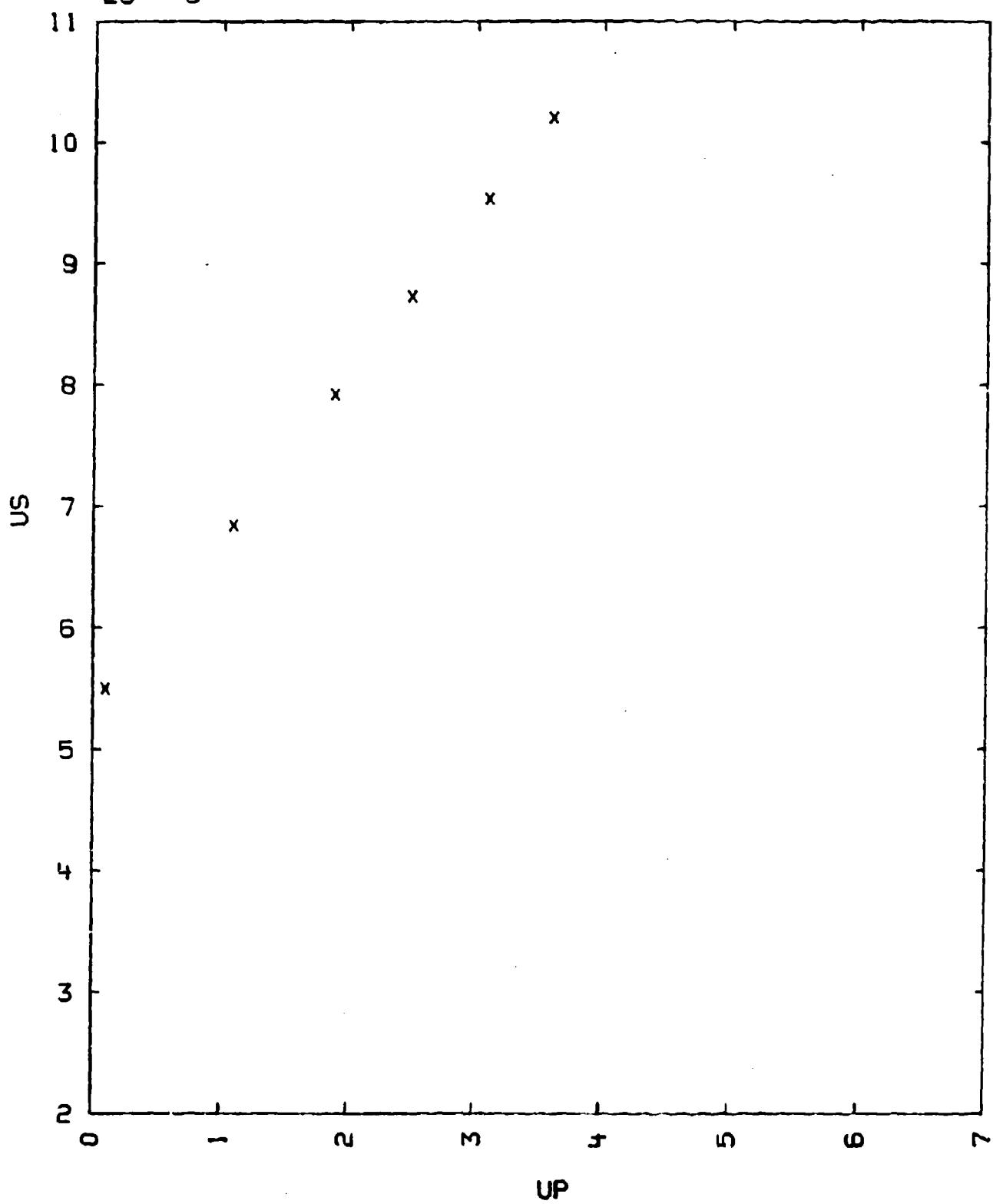
- 1) SOURCE: MARSH, S. P.
PRIVATE COMMUNICATION
GMX-6, LOS ALAMOS SCIENTIFIC LABORATORY
LOS ALAMOS, NEW MEXICO.
- 2) DATA REDUCTION METHOD: 0
THE ISENTROPES WERE CALCULATED WITH A CONSTANT VALUE OF
0.1620 CC/G FOR THE PARTIAL DERIVATIVE AT CONSTANT VOLUME
(DE/DP). E = INTERNAL ENERGY AND P = PRESSURE.
- 3) THE ACCURACY OF THE FIT TO THE COMPUTER OUTPUT FROM THE ABOVE SOURCE
IS AS FOLLOWS:

A DEVIATION OF LESS THAN 0.03 KBAR FOR THE FIRST ADIABAT	SECOND	-
- - - - - 0.10 - - - - -	THIRD	-
- - - - - 0.13 - - - - -	FOURTH	-
- - - - - 0.24 - - - - -	ABOVE 10 KB	-
- - - - - 0.03 - - - - -	33 -	-
- - - - - 0.24 - - - - -	FIFTH	31 -

- | | | | |
|-----------|------|-----------|----|
| - - - - - | 0.03 | - - - - - | 54 |
| - - - - - | 0.2 | - - - - - | 50 |
| - - - - - | 0.03 | - - - - - | 76 |
- SIXTH
- 4) THE FIT FOR THE LAST THREE ADIABATS WAS IMPROVED BY ALLOWING A NON-ZERO INTERCEPT. THE DEVIATION AT UFS-U = 0 IS THEREFORE THE VALUE OF THIS INTERCEPT. THIS ERROR DECREASES NEARLY LINEARLY TO THE LARGER UNCERTAINTY LISTED ABOVE.
- 5) THE MAXIMUM PRESSURE TO WHICH THE ABOVE FITS ARE VALID RANGE FROM 1100 KBARS FOR THE FIRST LINE TO 1250 KBARS FOR THE LAST TABLE ENTRY

TABLE I

ALUMINUM 2024 ALLOY
29---6



29---7

ALUMINUM 2024 ALLOY

	AL 93	PERCENT BY WEIGHT					
CU	3.8 - 4.9	-	-	-	-	-	-
MG	1.2 - 1.8	-	-	-	-	-	-
MN	0.3 - 0.9	-	-	-	-	-	-
FE	<0.5	-	-	-	-	-	-
SI	<0.5	-	-	-	-	-	-
ZN	<0.25	-	-	-	-	-	-
CR	<0.1	-	-	-	-	-	-

$$V_0 = 0.3591 \text{ CC/G} \quad C_L = 6.300 \text{ SEC} \quad C_D = 5.339 \text{ KM/SEC}$$

$$C_S = 2.895 \text{ SEC}$$

THE TABLE BELOW LISTS PRESSURE (P IN KILOBARS), PARTICLE- SHOCK- AND FREE SURFACE VELOCITY (UP, US AND UFS IN KM/SEC.) OF THE HUGONIOT. THIS HUGONIOT MAY BE CALCULATED FROM THE LEAST SQUARE FIT BELOW THE TABLE. OTHER COLUMNS ARE THE COEFFICIENTS OF THE FIT TO THE ISENTROPIC CURVES

$P = A_1 \cdot (UFS-U) + A_2 \cdot (UFS-U)^2 + A_3 \cdot (UFS-U)^3 + A_4 \cdot (UFS-U)^4 + A_5 \cdot (UFS-U)^5$

WHICH IN THE P VS U PLANE PASS THROUGH THE HUGONIOT AT THE LISTED POINTS
THE DENSITY AT THE FOOT OF THE HUGONIOT IS 2.785 G/CC.

TABLE

US	UP	UFS	P	A1	A2	A3	A4	A5
6.307	0.688	1.381	120.8	146.677	40.576	-0.969	-0.1114	
6.919	1.140	2.294	219.7	145.945	38.828	-0.016	-0.2619	
7.257	1.398	2.828	262.5	141.834	43.640	-5.966	2.5537	-0.4384
7.624	1.681	3.417	356.8	138.205	44.767	-7.214	3.0070	-0.4781
8.026	1.993	4.080	445.3	133.033	45.229	-7.609	3.0555	-0.4528
8.478	2.341	4.839	552.6	129.740	45.017	-7.537	2.9445	-0.4056
8.730	2.532	5.267	615.5	121.215	44.160	-6.932	2.6771	-0.3529
9.154	2.846	5.977	725.4	113.401	43.276	-6.570	2.4953	-0.3055
9.478	3.079	6.538	812.5	104.909	43.985	-7.959	2.9091	-0.3302
9.850	3.337	7.153	915.2	97.414	43.155	-7.982	2.8328	-0.2993
10.293	3.631	7.899	1040.3	87.492	41.468	-7.955	2.6758	-0.2534
10.553	3.795	8.312	1115.2	83.355	40.218	-7.660	2.5273	-0.2275

$$US = 5.300 + 1.550 \cdot UP - 0.1442 \cdot UP^2 + 0.02648 \cdot UP^3 \text{ KM/SEC}$$

COMMENTS

1) SOURCE: COMPILER

EQUATION OF STATE FILE

LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.

2) EXPERIMENTAL TECHNIQUE &

DATA REDUCTION METHOD ID

OVER

THE VALUE OF THE DERIVATIVE AT CONSTANT VOLUME:

$$\gamma(DP/DE) = 4.213 - 1.4275 \cdot \rho_{H_2O} - 0.77141 \cdot \rho_{H_2O}^2 + 0.169378 \cdot \rho_{H_2O}^3$$

HAS OBTAINED BY USING THE DUGDALE MCDONALD EQUATION FOR THE GRUNDTSEN GAMMA IN THE METHOD OF

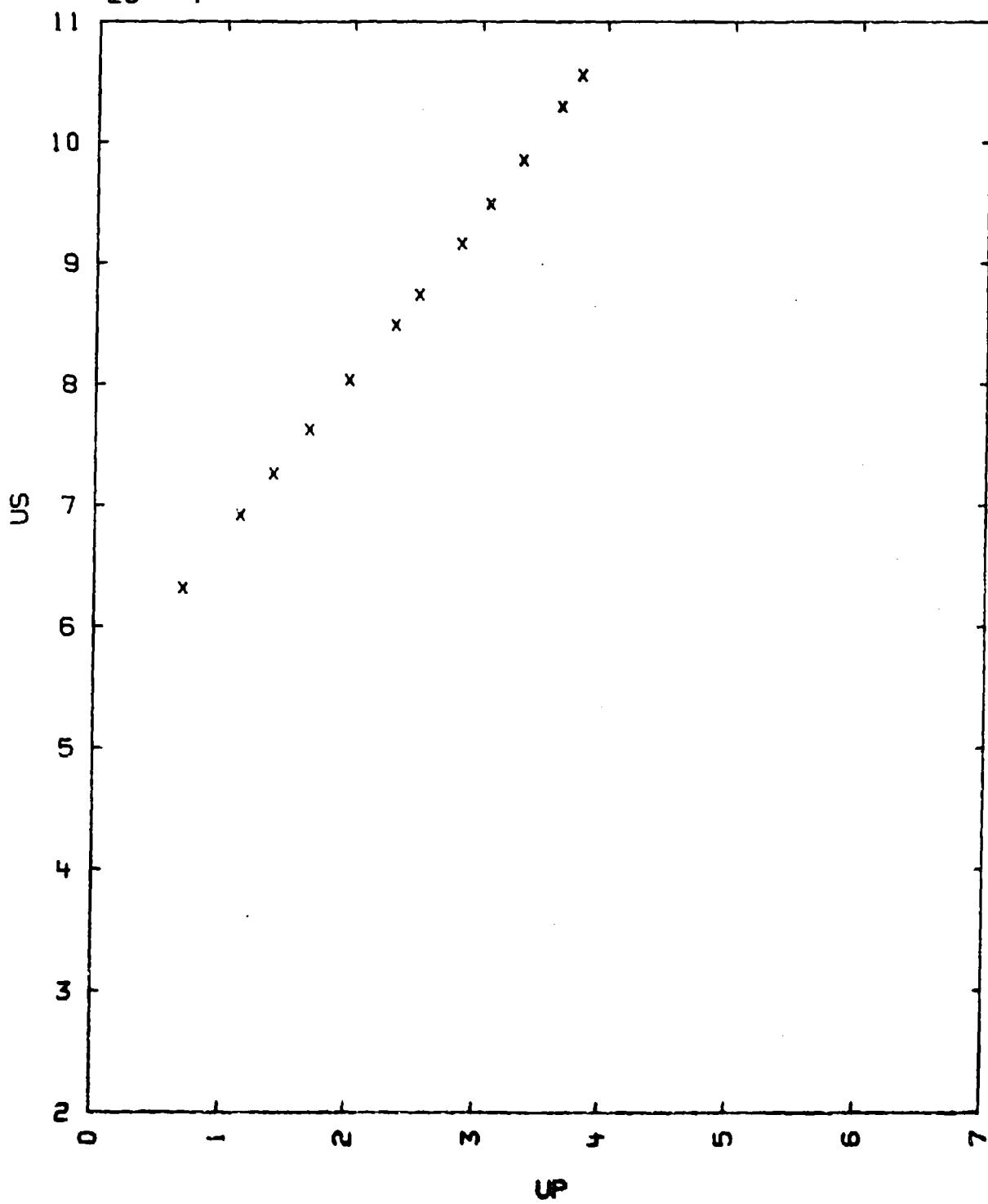
J. M. HALSH ET AL

PHYS. REV., VOL. 108, P. 196, (1957)

- 3) THE ACCURACY OF THE TABULATED LEAST SQUARE FIT TO THE COMPUTER OUTPUT OF P AND (UFS-U) IS BETTER THAN 0.2 KBAR. THE CALCULATION USED ABOUT 200 EXPERIMENTAL POINTS.
- 4) THE PRESSURE RANGE OVER WHICH THE TABULATED FITS ARE VALID RANGES FROM 0 TO 900 KBAR FOR THE FIRST ENTRY AND UP TO 1300 KBARS FOR THE LAST ENTRY.

TABLE I

ALUMINUM 2024 ALLOY
29---7



29---8

ALUMINUM AU4G ALLOY

AL = 92.9 PERCENT BY WEIGHT

CU = 4.0 - - -
 MG = 1.2 - - -
 MN = 1.3 - - -
 SI = 0.6 - - -

V0 = 0.356-0.357 CC/G

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	UFS	P	V/V0
2.80	7.309	1.374	2.747	281	0.812
-	7.142	1.400	2.799	280	0.804
-	7.353	1.381	2.762	284	0.812
-	7.353	1.381	2.762	284	0.812
-	7.143	1.592	3.184	318	0.777
-	7.396	1.539	3.078	319	0.792
-	7.235	1.501	3.001	304	0.793
-	7.564	1.590	3.180	337	0.790
-	7.032	1.377	2.753	271	0.804
-	7.032	1.402	2.803	276	0.801
-	7.310	1.429	2.857	293	0.805
-	7.386	1.449	2.897	300	0.804
-	7.032	1.227	2.453	242	0.828
-	7.032	1.209	2.418	238	0.825
-	6.983	1.237	2.473	242	0.823
-	7.143	1.265	2.530	253	0.823
-	7.299	1.446	2.892	296	0.802
-	7.386	1.428	2.855	295	0.806
-	7.496	1.447	2.896	306	0.808
-	7.331	1.519	3.038	312	0.792
2.81	7.194	1.235	2.469	250	0.828
-	7.112	1.240	2.479	248	0.826
-	7.022	1.241	2.482	245	0.823
-	7.257	1.209	2.418	247	0.833
-	6.859	1.213	2.426	234	0.823
-	6.793	1.116	2.232	213	0.836
-	6.451	0.972	1.944	176	0.849
-	6.451	0.938	1.875	170	0.855
-	6.321	0.725	1.449	129	0.885
-	6.321	0.738	1.476	131	0.883
-	6.427	0.638	1.275	115	0.901
-	6.361	0.634	1.267	113	0.900
-	6.410	0.636	1.272	115	0.901
-	6.410	0.646	1.291	116	0.899
-	6.265	0.700	1.393	123	0.888
-	6.321	0.711	1.421	126	0.888
-	6.127	0.715	1.430	123	0.883
-	6.127	0.735	1.466	127	0.880

ALUMINUM AU4G ALLOY

RHO0	US	UP	UFS	P	V/V0
------	----	----	-----	---	------

-	6.024	0.659	1.318	112	0.891
-	5.807	0.654	1.307	107	0.887
-	6.045	0.639	1.277	109	0.894
-	5.938	0.632	1.263	105	0.894
-	6.188	0.708	1.416	123	0.886
-	6.188	0.693	1.386	121	0.880
-	6.180	0.690	1.379	120	0.888
-	6.180	0.695	1.390	121	0.888
-	6.031	0.721	1.441	122	0.880
-	6.150	0.708	1.416	122	0.885
-	6.165	0.657	1.314	114	0.893
-	6.165	0.647	1.293	112	0.895
-	5.903	0.622	1.244	103	0.895
-	5.903	0.622	1.244	103	0.895

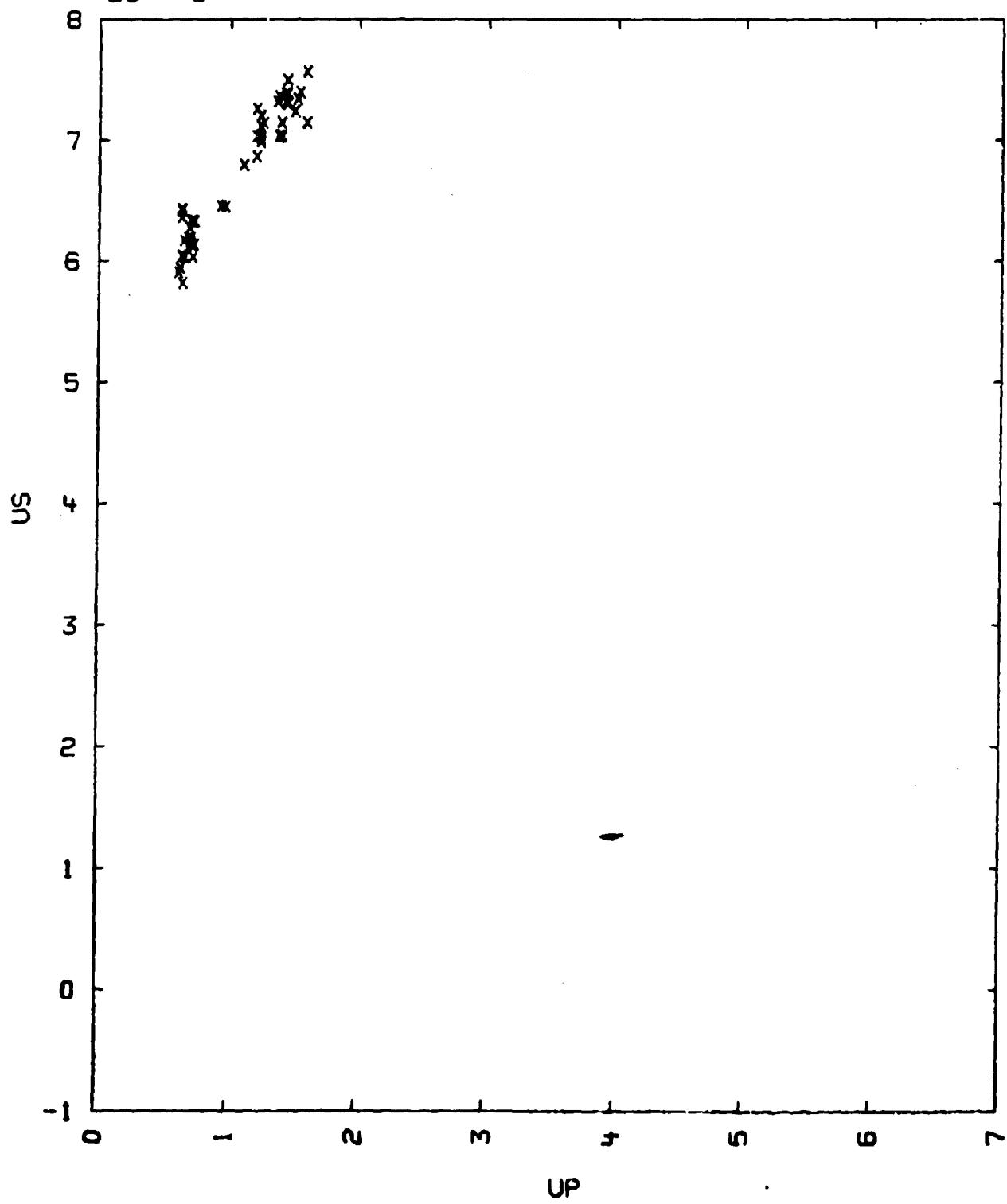
US =

COMMENTS:

- 1) SOURCE: BERGER J. AND FAUQUIGNON C.
PRIVATE COMMUNICATION (1964), B.P. NO. 7, SEVRAN, FRANCE
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE D
UP WAS DETERMINED BY USING THE FREE-SURFACE APPROXIMATION WHERE:
 $UFS = 2UP$
- 3) SAMPLE DIMENSIONS FOR THE US MEASUREMENTS: 2.0 CM DIAMETER
0.5 CM THICKNESS
SAMPLE DIMENSIONS FOR THE UFS MEASUREMENTS: 2.0 CM DIAMETER
0.25 CM THICKNESS

TABLE I

ALUMINUM AU4G ALLOY
29---8



29---9

ALUMINUM POROUS

AL 99.6 WT PERCENT
 CU, FE AND MN TOTAL 0.4 WT PERCENT
 PARTICLE SIZE: ABOUT 4 MICRON. SEE COMMENTS

$$V_0 = 0.55 - 0.75 \text{ CC/G}$$

$$V_{01} = 0.3715 \text{ CC/G}$$

THE TABLE LISTS DENSITY IN G/CC, VELOCITIES IN KM/SEC AND PRESSURE IN KBAR. U(ST) DESIGNATES THE VELOCITY OF THE 1100 ALUMINUM PROJECTILE

TABLE I

RHO0	US	UP	P	V/V0	U(ST)
1.816	3.23	1.197	70.22	0.630	1.64
-	2.97	1.206	65.07	0.594	1.62
-	3.94	1.425	102.0	0.638	2.04
-	4.28	1.543	119.9	0.640	2.25
-	5.00	1.901	172.6	0.619	2.86

$$US = -0.128 + 2.76 \cdot UP \text{ KM/SEC}$$

$$\text{SIGMA US} = 0.13 \text{ KM/SEC}$$

TABLE II

RHO0	US	UP	P	V/V0	U(ST)
1.348	2.63	1.523	46.9	0.497	1.63
-	2.65	1.329	47.5	0.498	1.64
-	3.50	1.772	83.7	0.494	2.29
-	4.38	2.179	129.7	0.502	2.93

$$US = -0.057 + 2.03 \cdot UP \text{ KM/SEC}$$

$$\text{SIGMA US} = 0.030 \text{ KM/SEC}$$

COMMENTS:

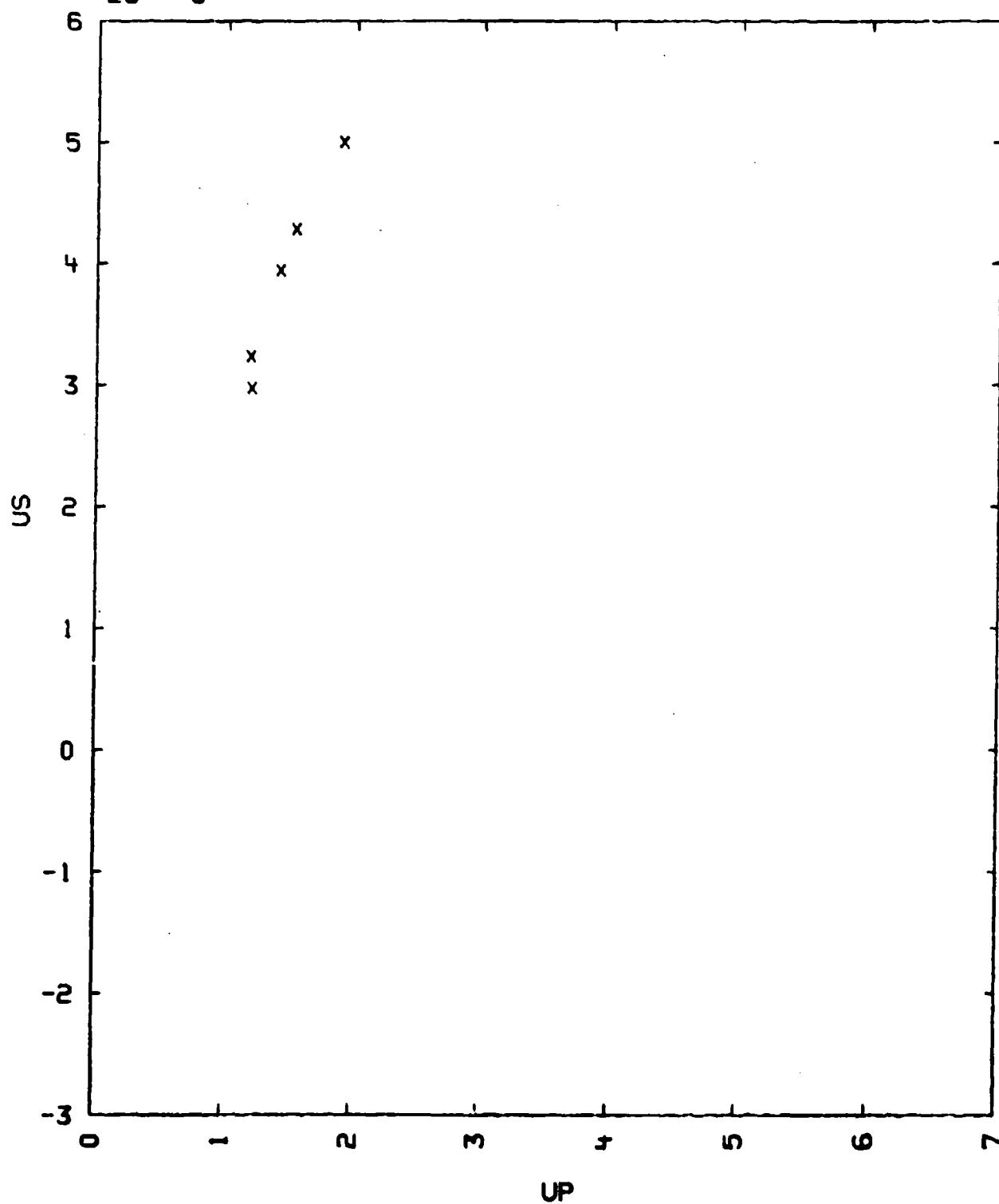
- 1) SOURCE: MORGAN D. T., ROCKOWITZ M., ATKINSON A. L.
 AVCO CORP. REPORT NO. AFHL-TR-65-117
 RESEARCH AND DEVELOPMENT DIVISION
 AVCO CORP., WILMINGTON, MASS., USA
- 2) EXPERIMENTAL TECHNIQUE: (SEE 3)
 DATA REDUCTION METHOD: B
 THE STANDARD MATERIAL WAS 1100 AL ALLOY WITH
 $RHO0 = 2.70 \text{ G/CC}$, $US = C + S \cdot UP \text{ KM/SEC}$
 $C = 3.144 + 0.093 \text{ KM/SEC}$
 $S = 1.528 + 0.041$
- 3) THE PRESSURE WAS GENERATED WITH ALUMINUM 1100 ALLOY PROJECTILES
 THE PROJECTILE VELOCITY AND SAMPLE SURFACE MOTION WERE DETECTED BY

REFLECTING A LASER BEAM OFF THESE SURFACES AND OBSERVING THE LIGHT
THROUGH A SET OF SLITS PHOTOMETRICALLY. THESE MEASUREMENTS TOGETHER
WITH THE OBSERVATION OF AN IMPACT FLASH GENERATED WHEN THE PROJECTILE
AND SAMPLE COLLIDE, YIELD US AND $U(ST)$

- 4) ACCURACY OF THE VELOCITY MEASUREMENTS WAS TYPICALLY 1 PERCENT

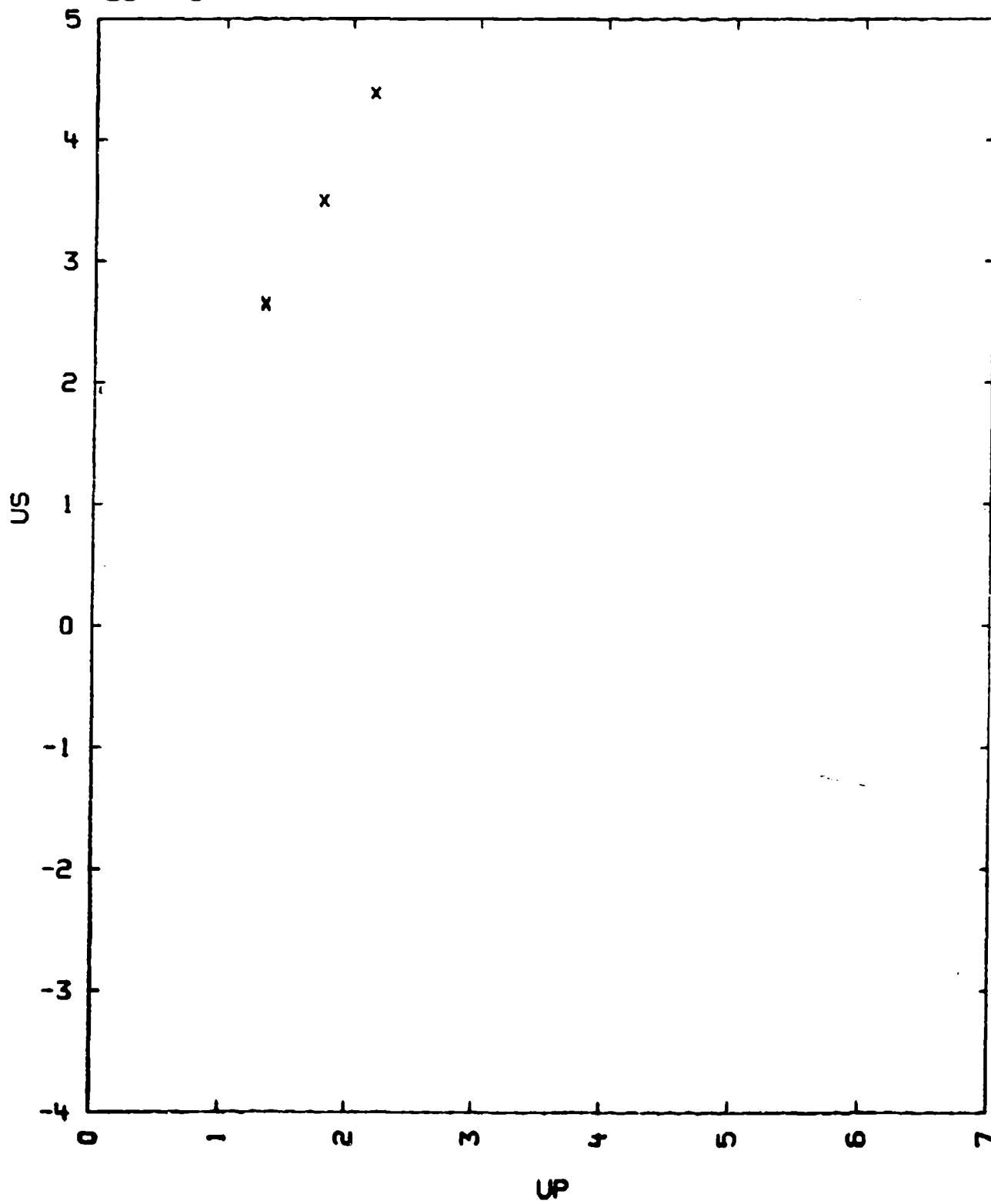
TABLE I

ALUMINUM POROUS
29---9



ALUMINUM POROUS
29---9

TABLE II



29---10
ALUMINUM POROUS

AL 100 PERCENT
PARTICLE SIZE: 0.208 MM MAX. 0.030 MM AVERAGE IN TABLES I AND II
0.300 MM MAX. 0.208 MM MIN. IN TABLE III

TABLE I, II AND III GIVE THE EXPERIMENTAL HUGONIOT POINTS FOR SAMPLES OF COMPRESSED POWDER AT VARIOUS DENSITIES. DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC AND PRESSURE IN KILOBARS.

$$V_0 = 0.519 - 0.521 \text{ CC/G}$$

TABLE I

SAMPLE						STANDARD
RHO0	US	UP	UFS	P	V/V0	UFS(ST)
1.93	5.08	1.88	3.238	184	0.630	2.89
1.930	5.17	1.89		189	0.634	
-	5.36	1.84		191	0.657	
1.927	6.43	2.45		304	0.619	
1.930	6.83	2.70		357	0.605	
-	6.90	2.77		368	0.598	
1.92	7.15	3.03	5.43	418	0.576	5.07C
1.92	7.79	3.42	5.93	511	0.561	5.86
1.919	7.97	3.45		528	0.567	

$$US = 2.08 + 1.71 \cdot UP \text{ KM/SEC.}$$

$$\text{SIG US} = 0.15 \text{ KM/SEC.}$$

$$V_0 = 0.629 - 0.635 \text{ CC/G}$$

TABLE II

SAMPLE						STANDARD
RHO0	US	UP	UFS	P	V/V0	UFS(ST)
1.580	5.43	2.43	3.968	210	0.552	3.56
1.585	5.82	2.67	4.148	246	0.540	3.96
1.580	5.93	2.69	4.53	254	0.547	4.02
1.587	6.58	3.05	5.21	319	0.536	4.67
1.580	6.50	3.06	5.25	316	0.529	4.67
1.585	6.83	3.20	5.53	347	0.531	4.97
1.585	6.90	3.25	5.52	355	0.529	5.04
1.580	6.91	3.26	5.258	356	0.528	5.07C
1.587	7.29	3.44	5.95	397	0.520	5.43
1.585	7.49	3.50		426	0.523	5.72
1.573	7.55	3.66	6.24	436	0.519	5.86

ALUMINUM POROUS

RHO0	US	UP	UFS	P	VVO	UFS(ST)
------	----	----	-----	---	-----	---------

$$US = 1.12 + 1.78 \cdot UP \text{ KM/SEC}$$

SIGMA US = 0.044 KM/SEC

$$V0 = 0.740 - 0.744 \text{ CC/G}$$

TABLE III

SAMPLE						STANDARD
RHO0	US	UP	UFS	P	V/V0	UFS(ST)
1.350	5.09	2.58	4.518	177	0.493	3.56
-	5.57	2.82	4.168	212	0.494	3.96
-	5.64	2.85	5.11	218	0.495	4.02
1.350	6.72	3.37	6.08	305	0.498	4.97
1.351	6.79	3.41	5.848	313	0.498	5.04
1.350	7.16	3.60	6.88	348	0.497	5.43
1.344	7.29	3.85		377	0.472	5.84
1.350	7.32	3.78		373	0.484	5.72

$$US = 0.426 + 1.83 \cdot UP \text{ KM/SEC}$$

SIGMA US = 0.12 KM/SEC

TABLE IV SUMMARIZES THE RESULTS OF SOME SHOCKS THAT WERE REFLECTED FROM A TUNGSTEN COVER. RHO0 IN G/CC, P IN KBAR AND E IN KBAR*CC/G

TABLE IV

RHO0	P1	V1/V0	E1-E0	P2	V2/V0	E2-E0
1.587	.319	0.536	46.6	984	0.428	92.5
1.585	426	0.523	64.1	1261	0.414	124.0
1.331	373	0.479	71.0	1244	0.375	135.8

COMMENTS:

- 1) SOURCE: ANDERSON, G. D., DORAN, D. G. AND FAHRENBRUCH, A. L.
AIR FORCE WEAPONS LAB. REPORT: AFWL-TR-65-147 (1965)
AFWL-TR-67-43
STANFORD RES. INST., MENLO PARK, CALIFORNIA, U.S.A.
- 2) EXPERIMENTAL METHOD C
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 2024 ALUMINUM
- 3) AVERAGE PARTICLE SIZE OF 0.030 MM. THE SIZE LIMITS GIVEN ABOVE WERE OBTAINED BY APPROPRIATELY SIEVING THE POWDER BEFORE PRESSING.
- 4) SAMPLE DIMENSIONS: 3.8 CM DIAMETER BY 0.48 CM THICK

5) IN TABLE I, II AND III:

C REFERS TO CASES WHERE THE UFS OF THE STANDARD WAS CALCULATED FROM ITS MEASURED SHOCK VELOCITY AND ITS KNOWN EQUATION OF STATE.
B REFERS TO EXPERIMENTS IN WHICH THE SURFACE OF THE SAMPLE HAS COVERED BY AN ALUMINUM SHIM.

TABLE I

ALUMINUM POROUS
29---10

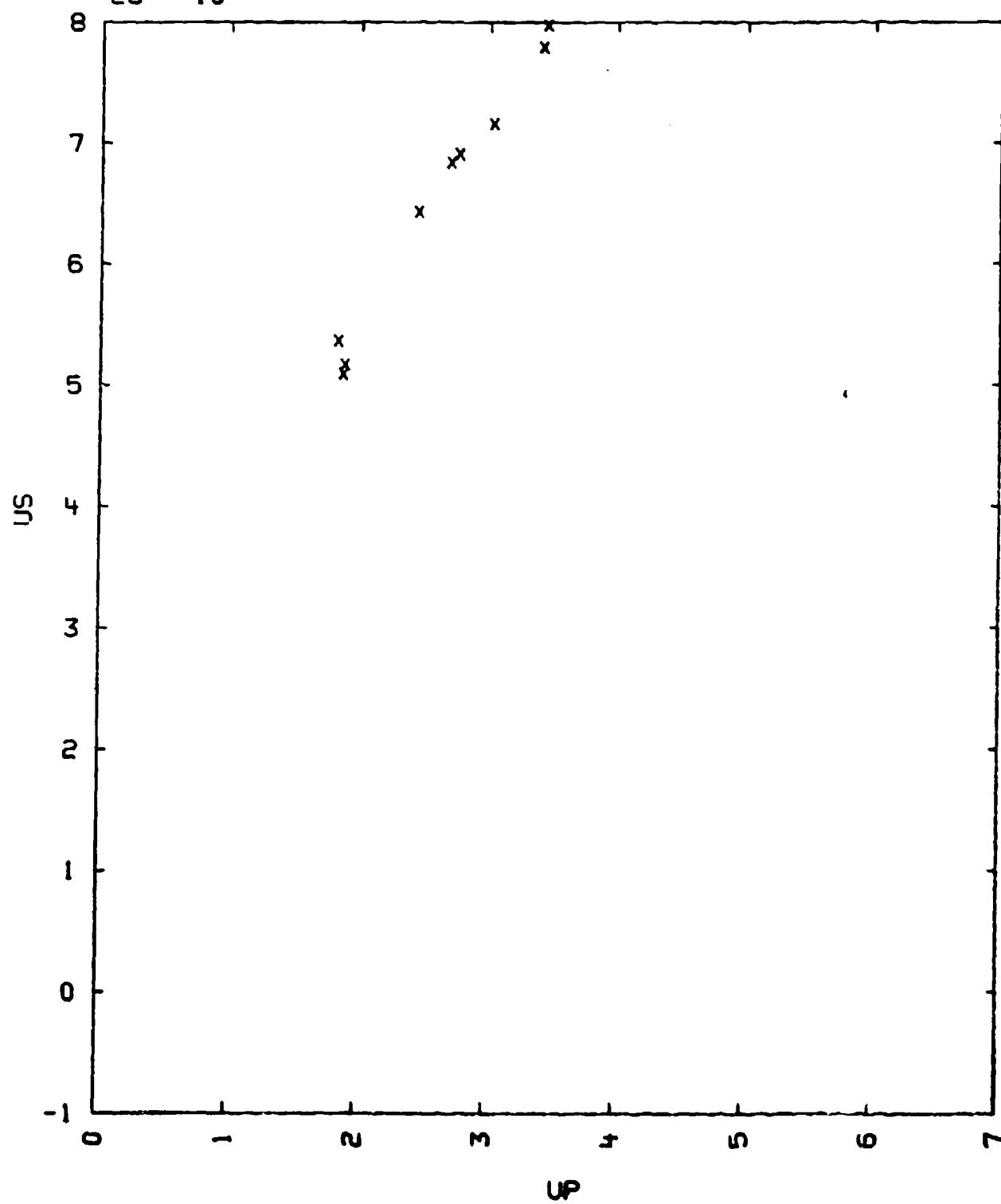


TABLE II

ALUMINUM POROUS

29---10

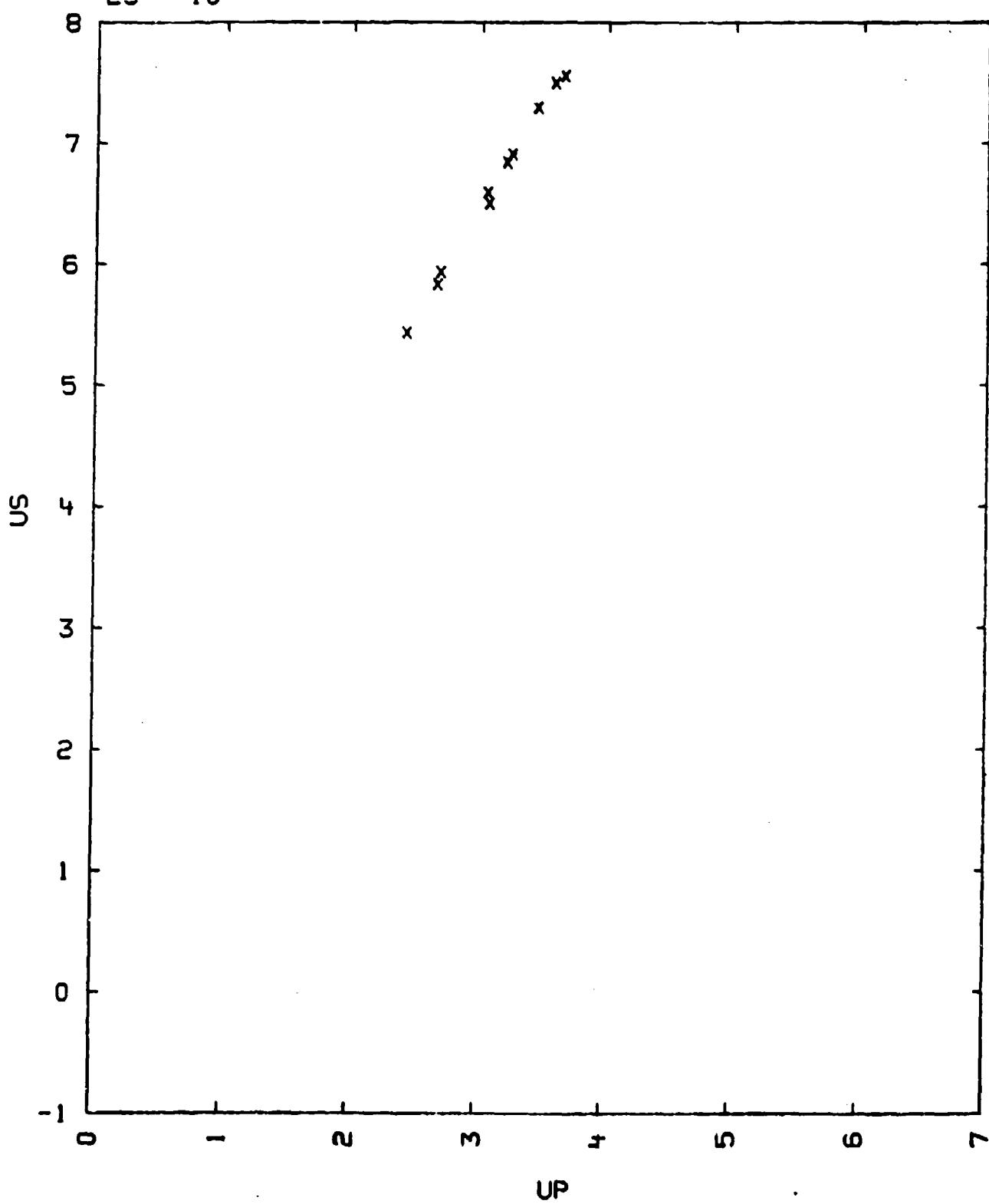
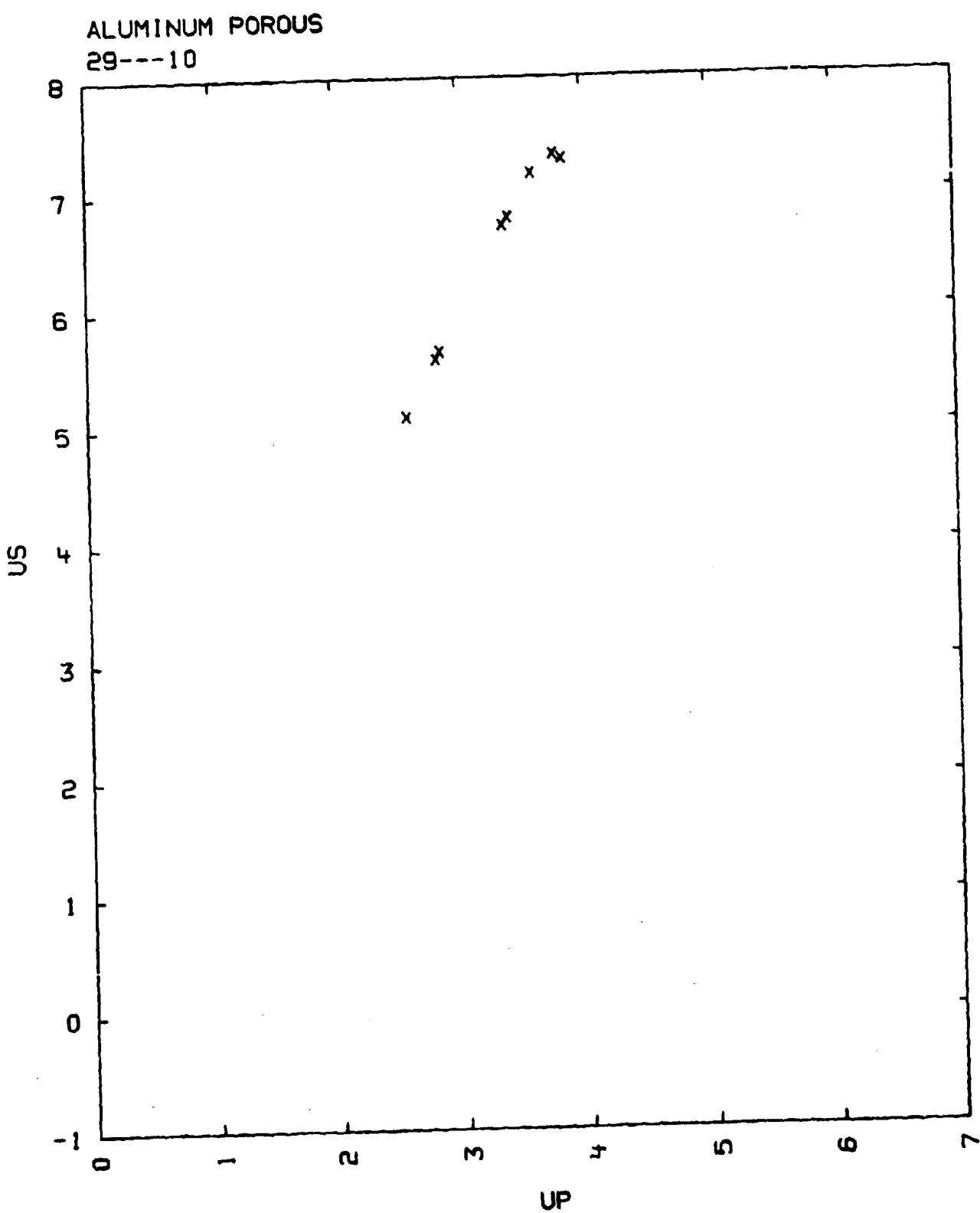


TABLE III



29---11
ALUMINUM 1060 ALLOY

SILICON	SI	0.25 WT PERCENT MAX.
IRON	FE	0.35 - - -
COPPER	CU	0.05 - - -
MANGANESE	MN	0.03 - - -
MAGNESIUM	MG	0.03 - - -
ZINK	ZN	0.05 - - -
TITANIUM	TI	0.03 - - -
OTHER		0.03 - - -
ALUMINUM	AL	REMAINDER

V0 = 0.370 CC/G

V01= 0.3706 CC/G

THE TABLE BELOW LISTS DENSITY IN G/CC, VELOCITIES IN KM/SEC AND PRESSURE IN KILOBARS

TABLE I

SAMPLE							STANDARD
T0	RHO0	US	UP	UFS	P	V/V0	UFS
17	2.70	7.31	1.45	2.97	286	0.802	2.89
-	-	7.75	1.80		377	0.767	3.60
-	-	8.77	2.46		583	0.718	5.03
-	-	9.06	2.58	5.42	633	0.716	5.35
-	-	9.13	2.71		667	0.702	5.61
-	-	8.76	2.77	5.48	656	0.684	•
-	-	8.82	2.70	5.14	647	0.694	•
17	-	8.14	2.21	4.56	486	0.728	•
542	2.580	8.82	2.79	5.47	636	0.684	•
555	2.586	8.56	2.76	5.27	611	0.678	•
543	2.586	8.24	2.23	4.66	475	0.729	•

US = 5.39 + 1.32*UP KM/SEC

SIGMA US = 0.20 KM/SEC

FOR THE COLD SAMPLES

COMMENTS:

1) SOURCE: ANDERSON, G.D., DORAN, D.G. AND FAHRENBRUCH, A.L.
AIR FORCE WEAPONS LAB REPORT AFWL-TR-65-147 (1965)
STANFORD RES. INST., MENLO PARK, CALIFORNIA

2) EXPERIMENTAL METHOD C
DATA REDUCTION TECHNIQUE B AND A (FOR THE ROWS WITH • IN THE UFS
COLUMN)

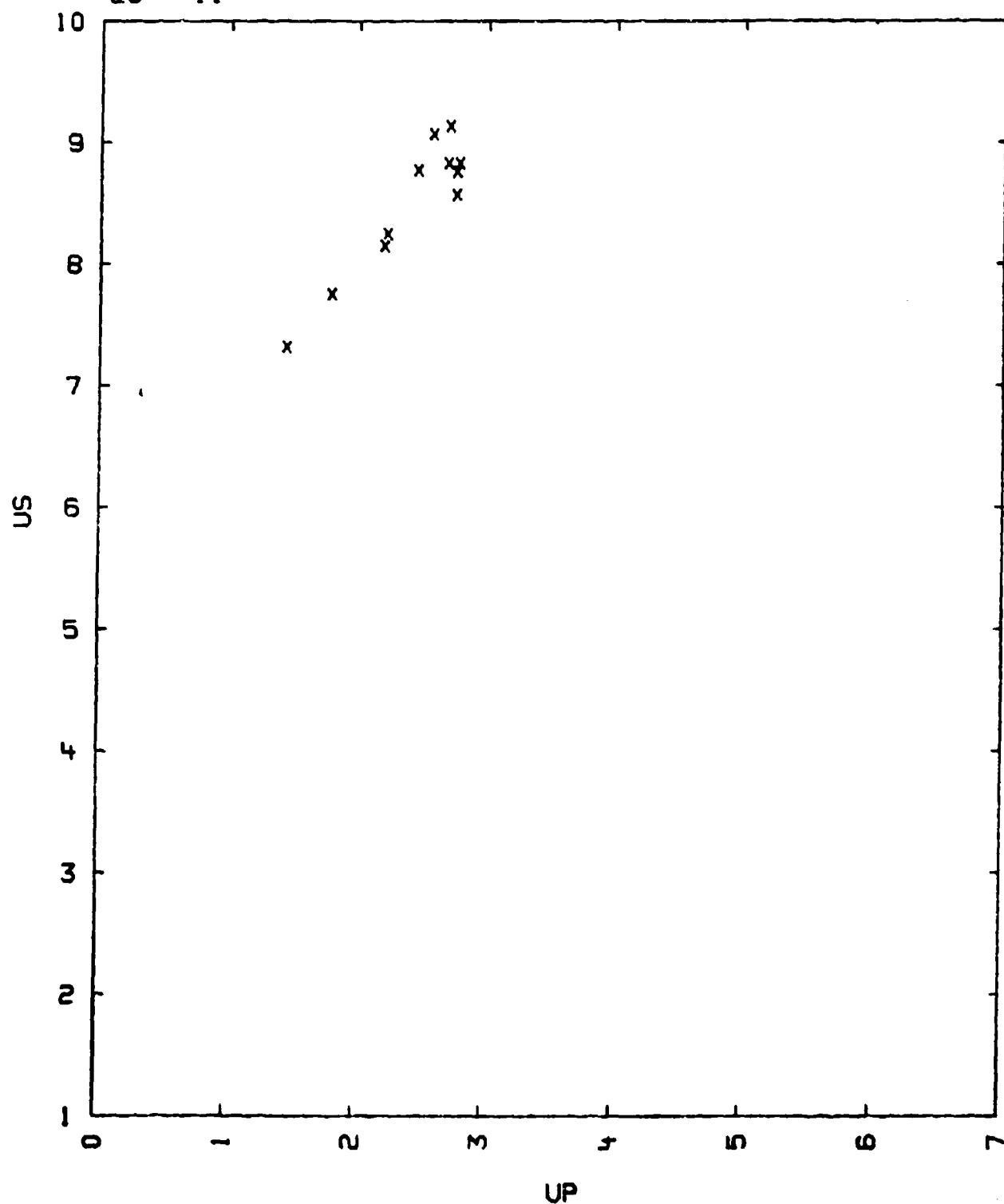
STANDARD MATERIAL 2024 ALUMINUM EXCEPT FOR
THE ROWS MARKED WITH A • WHERE BRASS HAS
USED. NO PARAMETER WAS LISTED FOR THIS
MATERIAL

3) SAMPLE DIMENSIONS: 3.8 CM DIAMETER BY 0.48 CM THICK.

TABLE I

ALUMINUM 1060 ALLOY

29---11



29---12
ALUMINUM 2024-T4

SI	0.50	WT PERCENT MAX.
FE	0.50	- - -
CR	0.10	- - -
ZN	0.25	- - -
OTHERS	0.15	- - -
CU	3.8-4.9	WT PERCENT
MN	0.3-0.9	- -
MG	1.3-1.8	- -

$$V_0 = 0.3579 \text{ CC/G} \quad C_L = 6.30 \text{ KM/SEC}$$

THE TABLE LISTS DENSITY IN G/CC VELOCITIES IN KM/SEC AND PRESSURES IN KBARS. SUBSCRIPTS 1 AND 2 REFER TO AN ELASTIC PRECURSOR AND THE MAIN HAVE.

TABLE

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0	
2.794	6.41	0.0304	5.44	0.99526					
-	6.46	0.0289	5.22	0.99553	5.801	0.609	0.311	50.9	0.94704
-	-	-	-	-	5.771	0.591	0.302	49.3	0.94836
-	-	-	-	-	5.745	0.572	0.293	47.5	0.94981
-	-	-	-	-	5.723	0.542	0.2775	45.0	0.95225
-	-	-	-	-	5.704	0.520	0.266	43.1	0.95405
-	-	-	-	-	5.686	0.496	0.254	41.1	0.95603
-	-	-	-	-	5.669	0.472	0.242	39.1	0.95804
-	-	-	-	-	5.658	0.447	0.2295	37.1	0.96018
-	-	-	-	-	5.647	0.421	0.2165	34.9	0.96242
-	6.40	0.0307	5.49	0.99520	5.721	0.535	0.274	44.5	0.95275
-	-	-	-	-	5.702	0.515	0.264	42.8	0.95428
-	-	-	-	-	5.683	0.493	0.253	40.9	0.95609
-	-	-	-	-	5.664	0.471	0.242	39.1	0.95791
-	-	-	-	-	5.643	0.449	0.231	37.2	0.95973
-	-	-	-	-	5.624	0.426	0.219	35.4	0.96157
-	-	-	-	-	5.603	0.403	0.208	33.4	0.96359
-	-	-	-	-	5.582	0.379	0.1955	31.4	0.96563
-	-	-	-	-	5.559	0.354	0.183	29.5	0.96767
-	-	-	-	-	5.535	0.329	0.170	27.4	0.96991
-	-	-	-	-	5.510	0.302	0.157	25.3	0.97216
-	6.42	0.0040	0.70	0.9994	5.671	0.573	0.287	45.5	0.94967
-	-	-	-	-	5.639	0.530	0.266	41.9	0.95312
-	-	-	-	-	5.617	0.502	0.252	39.6	0.95544
-	-	-	-	-	5.597	0.474	0.238	37.2	0.95778
2.794	-	-	-	-	5.558	0.444	0.223	34.6	0.96020
-	-	-	-	-	5.543	0.414	0.208	32.2	0.96280
-	6.41	0.0038	0.7	0.9994					
-	6.39	0.0072	1.30	0.9999	5.676	0.559	0.282	44.8	0.95061

ALUMINUM 2024-T4

RHO0	US1	UPI	P1	V1/V0	US2	UFS2	UP2	P2	V2/V0
-	-	-	-	-	5.659	0.546	0.276	43.7	0.95161
-	-	-	-	-	5.642	0.530	0.268	42.3	0.95209
-	-	-	-	-	5.626	0.513	0.259	40.8	0.95427
-	-	-	-	-	5.611	0.495	0.250	39.3	0.95576
-	-	-	-	-	5.595	0.480	0.243	38.0	0.95697
-	-	-	-	-	5.581	0.462	0.234	36.5	0.95849
-	-	-	-	-	5.567	0.443	0.224	35.0	0.96010
-	-	-	-	-	5.552	0.426	0.216	33.6	0.96152

US = $5.24 + 1.77 \cdot UP$ KM/SEC
SIG US = 0.009 KM/SEC
US = $5.19 + 1.68 \cdot P$ KM/SEC
SIG US = 0.005 KM/SEC
US = $5.15 + 1.83 \cdot UP$ KM/SEC
SIG US = 0.003 KM/SEC

FOR FIRST 20 POINTS WITH
UP = 0.15 TO 0.3 KM/SEC.
FOR THE FOLLOWING 6 POINTS
WITH UP = 0.2 TO 0.3 KM/SEC
FOR THE LAST 9 POINTS WITH
UP = 0.2 TO 0.58 KM/SEC.

COMMENTS:

- 1) SOURCE: FOWLES G. R.
J. APPL. PHYS. VOL. 32, P. 1475 (1961)
- 2) EXPERIMENTAL METHOD D
DATA REDUCTION TECHNIQUE D:
THE ASSUMPTION $UPI = 1/2 UFS1$ WAS USED
ARGUMENTS BASED ON ELASTIC THEORY YIELD $2UP = UFS + US(P1/K)$
 $-UPI$ WHERE K IS THE BULK MODULUS AND $P1/K$ IS THE SUM OF
 $(V1/V0)-1$ AND $(VFINAL/V0)-1$, WITH $V1$ THE VOLUME AT THE YIELD
POINT, AND $VFINAL$ THE VOLUME AFTER THE RELEASE WAVE FROM THE
FREE SURFACE REDUCED THE PRESSURE TO ZERO.
- 3) THE ABOVE TABLE WAS OBTAINED FROM 6 SEPARATE EXPERIMENTS USING
WEDGE SHAPED SAMPLES WITH A PRESSURE GRADIENT ACROSS ITS FACE.
SOME CORRECTIONS WERE MADE FOR LACK OF NORMAL INCIDENCE OF THE WAVES.
- 4) OTHER CONSTANTS LISTED: YOUNGS MODULUS E = 760 KBAR
BULK - K = 728 -
RIGIDITY - G = 286.6 -
HEAT CAPACITY CV = 0.925 JOULES/(G.DEG.C)
GRUNEISEN RATIO GAMMA = 2.13
 $(DP/DT)V = 0.055$ KBAR/DEG.C

- 5) THE FOLLOWING TENSILE YIELD POINTS WERE MEASURED:
HARDENED SAMPLE : 2.40 TO 2.90 KBAR
ANNEALED - : 0.70 TO 1.0 -
THE LARGER VALUE CORRESPONDS TO 2 PERCENT DEVIATION FROM THE LINEAR
STRESS STRAIN CURVE.
- 6) A CORRELATION BETWEEN THE HUGONIOT ELASTIC LIMIT, ROCKWELL HARDNESS
AND DISTANCE OF SHOCK TRAVEL IS GIVEN BY THE FOLLOWING TABLE:

P1	RH	X
3.4	80	7.9
2.0	80	15.9
12.6	885	3.3
10.7	885	7.9
7.9	885	15.9

SEE : STRESS RELAXATION IN THE SHOCK COMPRESSION OF SOLIDS

G. D. ANDERSON AND W. J. MURRI, TECHN. RPT. NO. AFL-TR-57-24
STANFORD RES. INST., 333 RAVENSWOOD AVE., MENLO PARK, CALIF.

TABLE I

ALUMINUM 2024-T4

29---12

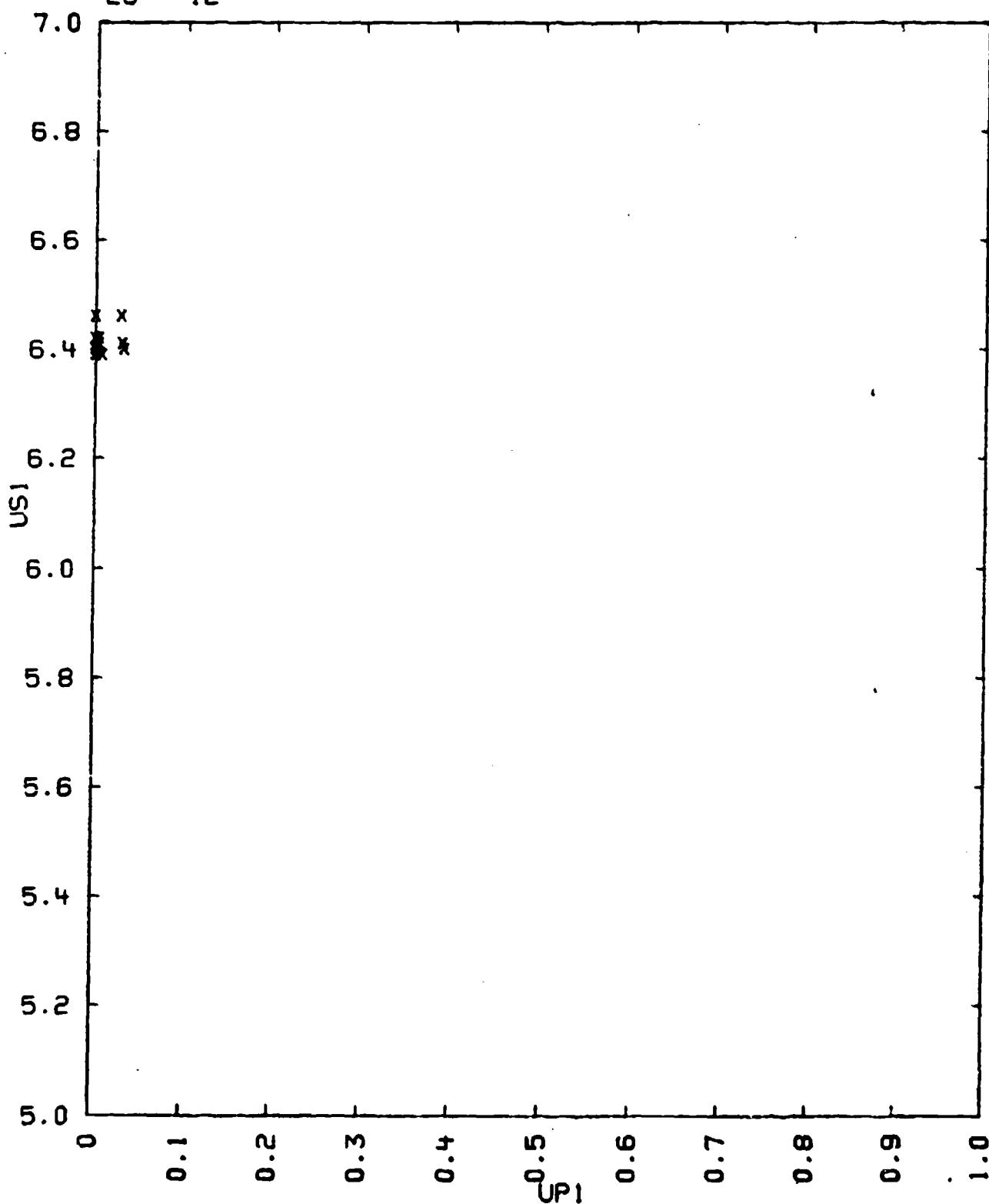
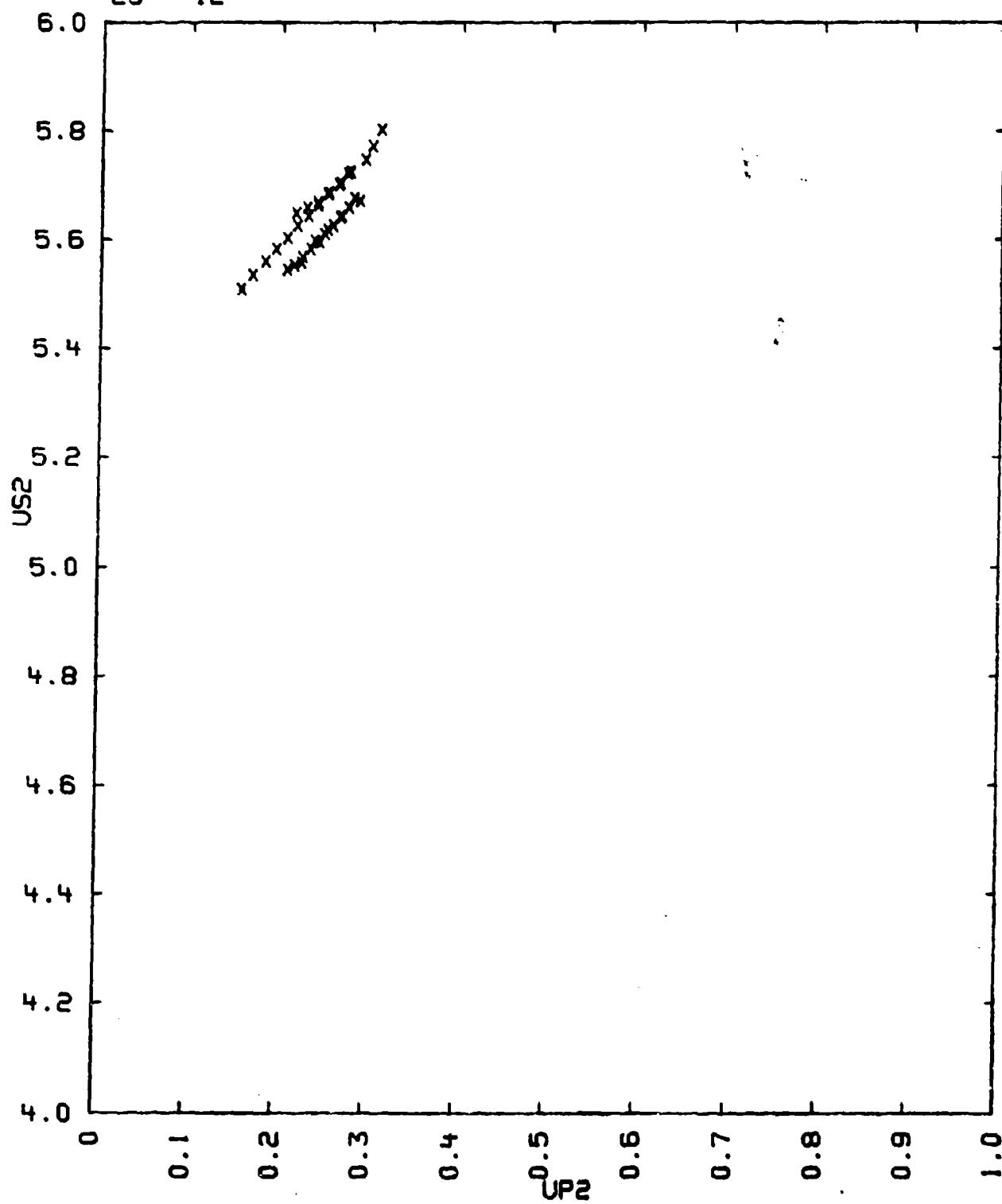


TABLE I

ALUMINUM 2024-T4
29---12



29---13
ALUMINUM 1060 ALLOY

FE	0.23	WEIGHT PERCENT
CU SI MN MG	LESS THAN 0.08	- -
AL	REMAINDER	- -

$$V_0 = 0.3695 \text{ CC/G}$$

$$V_{01} = 0.3706 \text{ CC/G}$$

$$C_0 = 5.240 \text{ KM/SEC}$$

THE TABLE LISTS ALL THE PRESSURES AND COMPRESSIONS OF THE MULTIPLE WAVE SYSTEMS USED IN THE DATA ANALYSIS OF THESE RELATIVELY BROAD SHOCK WAVES, TOGETHER WITH THE HYDROSTATIC PRESSURE (PH) AND THE APPROXIMATE USI AND UPI VALUES CALCULATED FROM THEM. DENSITY IS IN G/CC., VELOCITIES IN KM/SEC AND PRESSURE OR STRESS IN KBARS. NO = EXPERIMENT NUMBER. I INDICATES THE 1ST, 2ND, 3RD ETC. WAVE OF THE SYSTEM. U = PROJECTILE VELOCITY AND MAT IS THE PROJECTILE MATERIAL.

TABLE

U	RHO0	USI	UPI	P1	V1/V0	I	PH	MAT
.0086	2.705	6.7	0.0018	0.33	0.99973	1		AL
		5.4	0.0032	0.53	0.99948	2		
		5.1	0.0039	0.63	0.99934	3		
		3.8	0.0041	0.65	0.99929	4		
			0.0042	0.66	0.99928	5	0.53	
.0321	-	6.4	0.0017	0.30	0.99973	1		-
		6.1	0.0028	0.47	0.99956	2		
		5.7	0.0054	0.88	0.99909	3		
		5.5	0.0075	1.19	0.99871	4		
		5.3	0.0106	1.64	0.99812	5		
		5.2	0.0137	2.07	0.99753	6		
		5.0	0.0148	2.22	0.99731	7		
		4.6	0.0154	2.29	0.99719	8		
		4.7	0.0158	2.35	0.99709	9	2.17	
.0358	-	6.4	0.0021	0.37	0.99967	1		-
		5.9	0.0032	0.55	0.99948	2		
		5.7	0.0051	0.83	0.99916	3		
		5.5	0.0084	1.33	0.99854	4		
		5.3	0.0129	1.97	0.99770	5		
		5.1	0.0161	2.42	0.99707	6		
		4.4	0.0176	2.59	0.99675	7		
		4.3	0.0177	2.61	0.99671	8	2.44	
.0628	-	6.0	0.0024	0.39	0.9996	1		-
		5.4	0.0181	2.70	0.9967	2		
		5.2	0.0311	4.53	0.9942	3	4.34	
.0928	-	6.7	0.0020	0.36	0.9997	1		-
		5.4	0.0252	3.78	0.9954	2		
		5.2	0.0450	6.55	0.9916	3	6.35	

ALUMINUM 1060 ALLOY

U	RHO0	US1	UPI	PI	VI/VO	I	PH	MAT
.1620	-	6.0 5.6 5.9 1.8	0.0024 0.0317 0.0740 0.0770	0.39 4.79 11.52 11.66	0.9996 0.9943 0.9871 0.9854	1 2 3 4	11.43	-
.2197	-	6.4 5.6 5.5	0.0025 0.0359 0.106	0.44 5.46 15.84	0.9996 0.9936 0.9807	1 2 3	15.59	-
.3020	-	6.45 5.70	0.0026 0.148	0.45 22.36	0.9996 0.9734	1 2	22.05	-
.3995	-	6.13 5.82	0.0037 0.197	0.61 30.04	0.9994 0.9650	1 2	29.68	-
.4815	-	5.97 5.91	0.0065 0.237	1.06 36.42	0.9989 0.9584	1 2	36.09	-
.5210	-	6.44 5.92	0.0032 0.256	0.56 39.35	0.9995 0.9549	1 2	38.99	-

US = COMMENT 3

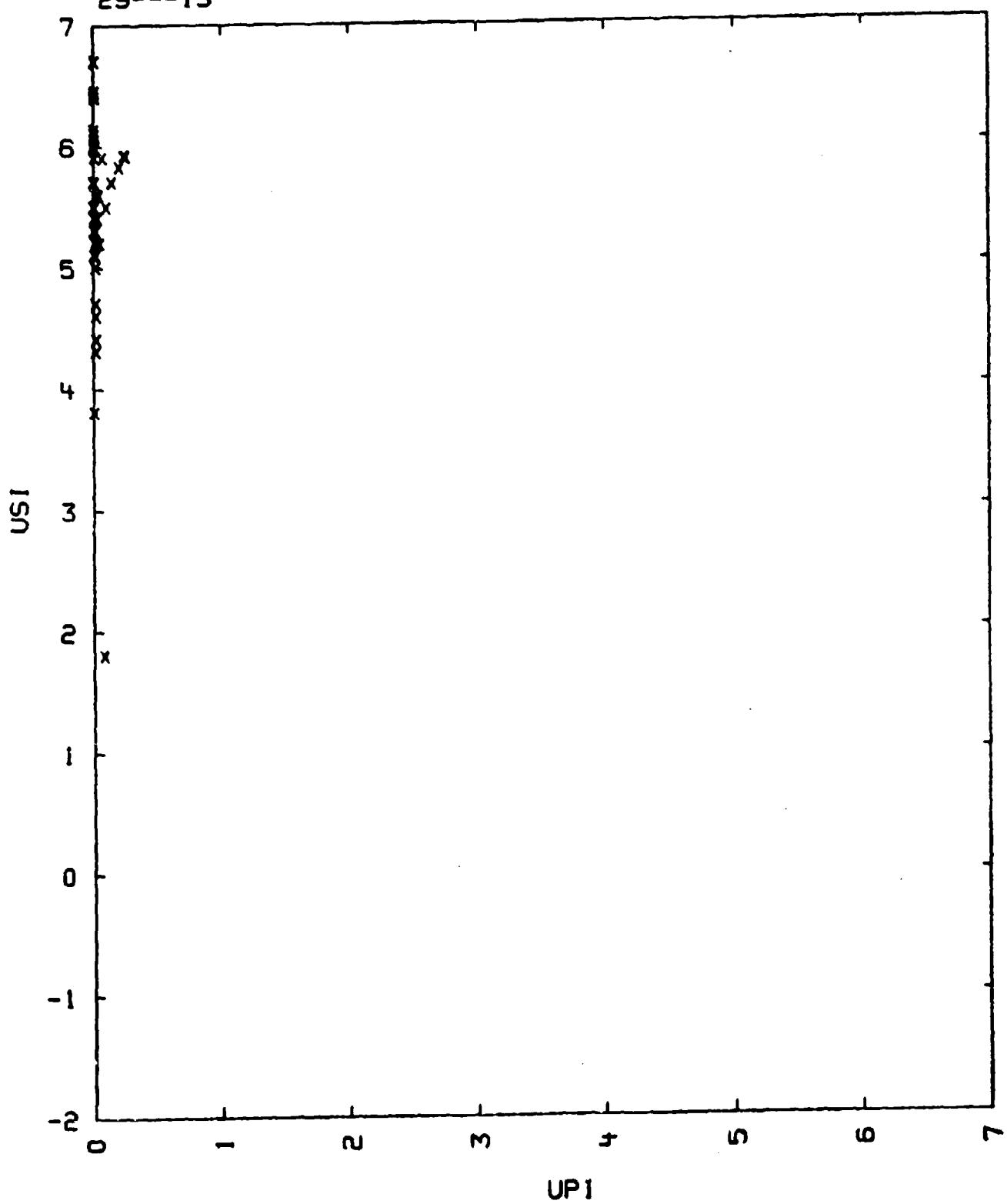
COMMENTS

- 1) SOURCE: MUNSON, D. E. AND BARKER, L. M.
J. APPL. PHYS., VOL. 37, P. 1652, (1966)
SANDIA CORPORATION, ALBUQUERQUE, NEW MEXICO.
- 2) EXPERIMENTAL TECHNIQUE F AND AN INTERFEROMETRIC METHOD
DATA REDUCTION METHOD D WITH 2UP = UFS
WITH THE MODIFICATION THAT A SMOOTH BROAD WAVE
IS APPROXIMATED BY A SET OF DISCREET PRESSURE STEPS. THE SHOCK VELO-
CITY MEASUREMENTS WERE CORRECTED FOR ERRORS INTRODUCED BY SURFACE
INTERACTIONS.
- 3) NOTE THAT US1 AND UPI ARE ONLY APPROXIMATE NUMBERS CALCULATED FROM
THE PI AND VI/VO VALUES TO ILLUSTRATE THE WAVE THICKNESS. IN THE
FIT BELOW US IS OBTAINED FROM THE MAXIMUM VALUE OF PH OF THE PARTICULAR
WAVE SYSTEM AND THE CORRESPONDING VALUE OF VI, USING THE EQUALITY
US = VO(PH/(VO-VI))^{1/2}, WHILE UP WAS TAKEN TO BE 1/2 THE MAXIMUM
FREE SURFACE VELOCITY. THESE POINTS ARE COMBINED WITH DATA FROM LOS-
ALAMOS SCIENTIFIC LABORATORY TO GIVE:
US = 5.270 + 1.429[•]UP - 0.0119[•]UP[•]2 KM/SEC, TO UP = 4.45 KM/SEC
MAX. DEVIATION 1.0 PERCENT.
- 4) PH = PI - 2/3[•]Y, WHERE THE YIELD POINT Y IS A FUNCTION OF THE DEGREE
OF WORK HARDENING.
- 5) VO1 WAS CALCULATED FROM THE CUBIC UNIT CELL CONSTANT GIVEN IN
HYCKOFF, CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, 1963)
A = 4.04958 ANGSTROM AT 25 DEG. CENTIGRATE.

TABLE I

ALUMINUM 1060 ALLOY

29---13



29---14
ALUMINUM 2024

CU	3.0-4.9	WT. PERCENT
MG	1.2-1.8	- -
MN	0.3-0.9	- -
SI	0.5	- -
FE	0.5	- -
ZN	0.25	- -
CR	0.10	- -
AL	REST	- -

$$V_0 = 0.3593 \text{ CC/C} \quad CL = 6.38 \text{ KM/SEC} \quad CO = 5.20 \text{ KM/SEC}$$

$$CS = 3.20 \text{ KM/SEC}$$

THE TABLE LISTS RHO0 IN G/CC, VELOCITIES IN KM/SEC AND P IN KBARS. CU= COPPER, FS= FANSTEEL, AL= ALUMINUM AND WF= WEIGHTING FACTOR.

TABLE								
SAMPLE					IMPACTOR			
RHO0	US	UP	P	V/V0	MAT	U	WF	
2.783	8.099	2.017	455.	.7510	FS	2.583	1	
2.783	9.504	3.081	815.	.6758	AL	6.161	1	
2.783	9.793	3.289	896.	.6642	CU	4.883	1	
2.783	9.839	3.439	942.	.6505	AL	6.876	1	
2.783	10.047	3.663	1024.	.6354	FS	4.775	1	
2.783	10.432	3.763	1092.	.6393	AL	7.526	1	
2.783	10.313	3.789	1088.	.6326	CU	5.621	1	
2.783	10.159	3.828	1082.	.6232	CU	5.654	1	
2.783	10.594	4.034	1189.	.6192	AL	8.067	1	
2.783	10.585	4.127	1216.	.6101	FS	5.400	1	
2.783	10.884	4.438	1344.	.5922	CU	6.567	1	
2.783	12.351	5.237	1800.	.5760	CU	7.833	1	
2.783	12.460	5.297	1837.	.5749	CU	7.929	1	
2.783	13.228	5.982	2196.	.5491	FS	7.947	1	

$$US = 8.471 + 1.310 \cdot UP \text{ KM/SEC}$$

$$SIG.US = 0.022 \text{ KM/SEC}$$

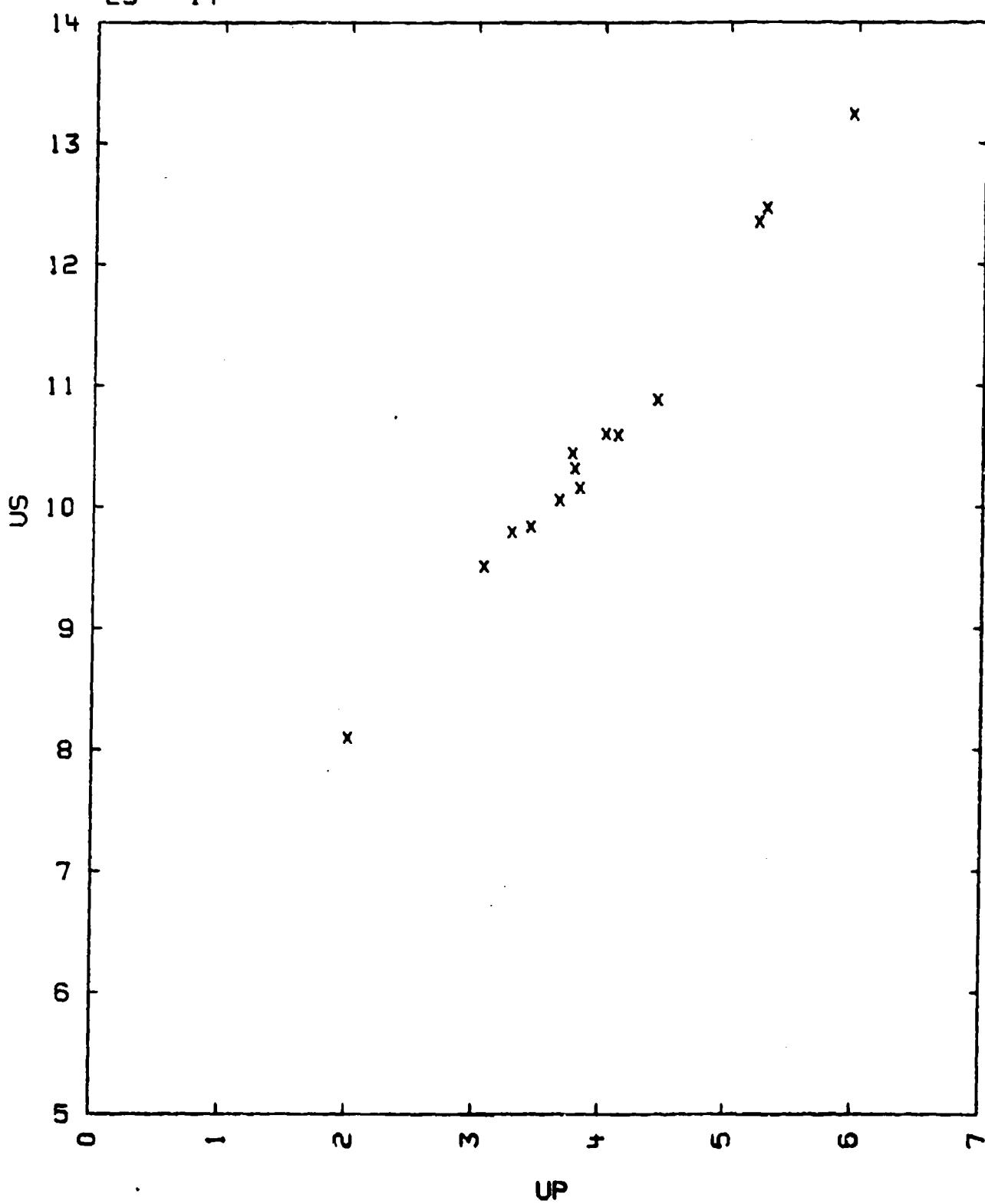
COMMENTS:

- 1) SOURCE: ISBELL H.M., SHIPMAN F.H. AND JONES A.H.
HUGONIOT EQUATION OF STATE OF ELEVEN MATERIALS TO FIVE MBARS
MATERIALS SCIENCE LABORATORY REPORT: MSL-58-13
- 2) EXPERIMENTAL TECHNIQUE: A
DATA REDUCTION METHOD : A
- 3) NOMINAL UNCERTAINTIES ARE: (SIG.US)/US = .005 AND (SIG.U)/U = .0005
- 4) POISSONS RATIO = 0.332 ; YIELD STRENGTH = 3.2 KBAR
ULTIMATE TENSILE STRENGTH = 4.7 KBAR
ULTIMATE TENSILE STRENGTH = 4.7 KBAR

TABLE I

ALUMINUM 2024

29---14



29---15
2024 ALUMINUM

AL	93.
CU	3.8 - 4.9
MG	1.2 - 1.8
MN	0.3 - 0.9
FE	0.5
SI	0.5
ZN	0.25
CR	0.1

$$V_0 = 0.3585 - 0.3591 \text{ CC/G} \quad C_L = 6.39 \text{ KM/SEC} \quad C_D = 5.25 \text{ KM/SEC}$$

$$C_S = 3.15 \text{ KM/SEC}$$

IN THE TABLES BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBAR
AND DENSITY IN G/CC. THE IMPACTOR IS THE STANDARD.

TABLE I

SAMPLE					STANDARD
RHO0	US	UP	P	V/V0	MATERIAL
2.785	6.02	0.50	84.	0.9169	2024 AL
2.785	6.10	0.50	85.	0.9180	2024 AL
2.785	6.05	0.51	86.	0.9157	2024 AL
2.782	6.16	0.67	115.	0.8912	2024 AL
2.785	6.35	0.77	136.	0.8787	2024 AL
2.782	6.49	0.86	155.	0.8675	2024 AL
2.785	6.52	0.87	158.	0.8666	2024 AL
2.782	6.61	0.97	178.	0.8533	2024 AL
2.789	6.86	1.15	220.	0.8324	2024 AL
2.789	6.89	1.16	223.	0.8316	2024 AL
2.780	6.91	1.16	223.	0.8321	2024 AL
2.789	6.86	1.21	232.	0.8236	2024 AL
2.789	7.27	1.43	290.	0.8033	2024 AL
2.789	7.33	1.58	323.	0.7844	2024 AL
2.779	7.51	1.62	338.	0.7843	2024 AL
2.789	7.69	1.72	368.	0.7760	2024 AL
2.789	7.86	1.77	378.	0.7689	2024 AL
2.789	7.80	1.73	367.	0.7724	2024 AL
2.789	7.77	1.81	392.	0.7671	2024 AL
2.789	7.69	1.85	397.	0.7594	2024 AL
2.779	7.87	1.85	432.	0.7553	2024 AL
2.789	8.13	2.13	483.	0.7390	2024 AL
2.789	8.33	2.16	502.	0.7407	2024 AL
2.789	8.23	2.21	507.	0.7315	2024 AL
2.779	8.40	2.31	539.	0.7250	2024 AL
2.789	8.53	2.45	583.	0.7129	2024 AL
2.789	8.83	2.59	638.	0.7067	2024 AL
2.789	8.76	2.60	635.	0.7032	2024 AL
2.789	8.80	2.65	650.	0.6969	2024 AL
2.789	9.14	2.82	719.	0.6915	2024 AL
2.789	9.40	2.89	784.	0.6919	2024 AL

2024 ALUMINUM

RHO0	US	UP	P	V/V0	MATERIAL
2.789	9.67	3.22	868.	0.6670	2024 AL
2.789	9.64	3.29	885.	0.6587	2024 AL
2.778	9.87	3.42	938.	0.6533	2024 AL
2.778	10.19	3.72	1053.	0.6349	2024 AL
2.789	10.37	3.75	1085.	0.6384	2024 AL

US = $5.328 + 1.338 \cdot UP$ KM/SEC HUGONIOT FIT ADOPTED BY THE SOURCE
COMPARISON LEAST SQUARE FITS:

US = A0 + A1 · UP KM/SEC

A0 = 5.343 KM/SEC SIG A0 = 0.022 KM/SEC

A1 = 1.325 - SIG A1 = 0.010 -

SIG US = 0.058 -

TABLE II

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
2.784	6.00	0.50	84.	0.9167	921-T AL	5.76
2.783	5.95	0.51	84.	0.9143	921-T AL	5.77
2.784	5.95	0.51	84.	0.9143	921-T AL	5.77
2.785	6.12	0.51	104.	0.9003	921-T AL	5.91
2.785	6.10	0.51	104.	0.9000	921-T AL	5.91
2.785	6.53	0.91	165.	0.8608	921-T AL	6.34
2.785	6.53	0.91	165.	0.8606	921-T AL	6.34
2.784	6.56	0.98	179.	0.8506	921-T AL	6.43
2.784	6.51	0.98	179.	0.8479	921-T AL	6.44
2.780	6.49	0.99	179.	0.8475	921-T AL	6.44
2.785	6.95	1.26	244.	0.8187	921-T AL	6.83
2.777	6.94	1.26	243.	0.8184	921-T AL	6.83
2.784	6.94	1.28	247.	0.8156	921-T AL	6.86
2.781	7.61	1.73	368.	0.7727	921-T AL	7.49
2.781	7.61	1.73	366.	0.7727	921-T AL	7.49
2.784	7.62	1.74	369.	0.7717	921-T AL	7.52
2.783	8.70	2.47	598.	0.7181	921-T AL	8.54
2.783	8.62	2.48	595.	0.7123	921-T AL	8.54
2.781	9.60	3.18	849.	0.6687	921-T AL	9.53
2.781	9.55	3.19	847.	0.6680	921-T AL	9.53
2.779	9.62	3.27	874.	0.6601	921-T AL	9.65
2.781	9.92	3.40	938.	0.6573	921-T AL	9.84
2.781	9.87	3.41	938.	0.6545	921-T AL	9.84

US = A0 + A1 · UP KM/SEC

A0 = 5.253 KM/SEC

A1 = 1.357 - SIG A0 = 0.020 KM/SEC

SIG A1 = 0.010 -

SIG US = 0.050 -

TABLE III

-----SAMPLE----- -----STANDARD-----

RHO0	US	UP	P	V/V0	MATERIAL US(ST)
2.784	5.81	0.28	45.	0.9518	CU 4.24
2.784	5.78	0.28	45.	0.9516	CU 4.24
2.784	6.02	0.44	74.	0.9269	CU 4.41
2.782	6.05	0.47	79.	0.9223	CU 4.45
2.784	6.23	0.65	113.	0.8957	CU 4.64
2.782	6.28	0.68	119.	0.8917	CU 4.67
2.785	6.37	0.72	128.	0.8870	CU 4.72
2.784	6.32	0.73	128.	0.8845	CU 4.72
2.784	6.31	0.73	128.	0.8843	CU 4.72
2.783	6.39	0.78	139.	0.8779	CU 4.78
2.784	6.52	0.86	156.	0.8681	CU 4.87
2.782	6.59	0.89	163.	0.8649	CU 4.90
2.784	6.59	0.90	165.	0.8634	CU 4.91
2.784	6.58	0.90	165.	0.8632	CU 4.91
2.782	6.68	0.97	180.	0.8544	CU 4.98
2.782	7.06	1.32	259.	0.8130	CU 5.37
2.781	7.09	1.35	266.	0.8096	CU 5.40
2.782	7.22	1.38	277.	0.8089	CU 5.44
2.785	7.16	1.44	287.	0.7989	CU 5.49
2.782	7.21	1.45	291.	0.7989	CU 5.51
2.782	7.85	1.86	406.	0.7631	CU 5.97
2.782	8.11	2.09	472.	0.7423	CU 6.23
2.781	8.08	2.10	472.	0.7401	CU 6.22
2.782	8.36	2.33	542.	0.7213	CU 6.48
2.785	8.42	2.33	546.	0.7233	CU 6.49
2.784	8.44	2.37	557.	0.7192	CU 6.53
2.781	8.57	2.45	584.	0.7141	CU 6.62
2.782	8.75	2.60	633.	0.7029	CU 6.79
2.782	8.91	2.73	677.	0.6936	CU 6.93
2.782	8.92	2.74	680.	0.6928	CU 6.94
2.782	9.23	2.93	752.	0.6826	CU 7.16
2.782	9.77	3.35	911.	0.6571	CU 7.62
2.780	9.75	3.36	911.	0.6554	CU 7.63
2.780	9.80	3.38	921.	0.6551	CU 7.65
2.782	9.75	3.38	917.	0.6533	CU 7.67
2.782	9.82	3.39	926.	0.6548	CU 7.92
2.780	10.04	3.62	1010.	0.6394	CU 7.95
2.784	10.24	3.63	1035.	0.6455	CU 7.95

US = A0 + A1 * UP KM/SEC

A0 = 5.398 KM/SEC

A1 = 1.298 -

SIG A0 = 0.014 KM/SEC

SIG A1 = 0.007 -

SIG US = 0.044 -

TABLE IV

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL US(ST)	
2.784	6.82	0.95	175.	0.8585	FE	4.83
2.784	6.82	0.95	175.	0.8585	FE	4.83
2.784	7.14	1.38	270.	0.8095	FE	5.49

2024 ALUMINUM

RHO0	US	UP	P	V/VO	MATERIAL US(ST)
2.784	7.14	1.36	270.	0.8095	FE 5.49
2.784	8.15	2.15	488.	0.7362	FE 6.53
2.784	8.15	2.15	488.	0.7362	FE 6.53
2.784	8.85	2.64	650.	0.7017	FE 7.15
2.784	8.80	2.64	1627.	0.2455	FE 7.15
2.784	9.76	3.24	880.	0.6690	FE 7.90
2.784	9.43	3.27	858.	0.8532	FE 7.90
2.784	9.76	3.29	894.	0.5629	FE 7.96
2.784	9.72	3.30	893.	0.6605	FE 7.96
2.784	9.67	3.38	910.	0.6505	FE 8.05
2.784	9.61	3.39	907.	0.6472	FE 8.05
2.784	10.46	3.77	1098.	0.6396	FE 8.54
2.784	10.41	3.78	1095.	0.6369	FE 8.54
2.784	10.43	3.78	1098.	0.6376	FE 8.54
2.784	10.34	3.79	1091.	0.6335	FE 8.54

US = A0 + A1*UP KM/SEC

A0 = 5.333 KM/SEC

A1 = 1.323 -

SIG A0 = 0.074 KM/SEC

SIG A1 = 0.025 -

SIG US = 0.106 -

TABLE V

SAMPLE					STANDARD	
RHO0	US	UP	P	V/VO	MATERIAL	US(ST)
2.782	6.26	0.63	110.	0.8994	TU-3 MO	3.20
2.782	6.23	0.63	109.	0.8989	TU-3 MO	3.20
2.784	6.36	0.79	140.	0.8758	TU-3 73	3.37
2.782	6.46	0.80	144.	0.8762	TU-3 MO	3.38
2.782	6.39	0.80	142.	0.8748	TU-3 MO	3.38
2.782	6.82	1.08	205.	0.8416	TU-3 MO	3.66
2.784	6.84	1.12	213.	0.8363	TU-3 MO	3.70
2.784	7.01	1.22	238.	0.8260	TU-3 MO	3.81
2.784	6.98	1.22	237.	0.8252	TU-3 MO	3.81
2.785	7.69	1.74	373.	0.7737	TU-3 MO	4.34
2.784	7.76	1.78	385.	0.7706	TU-3 MO	4.38
2.785	7.77	1.94	420.	0.7503	TU-3 MO	4.53
2.782	8.05	1.95	437.	0.7578	TU-3 MO	4.56
2.782	8.01	1.96	437.	0.7553	TU-3 MO	4.56
2.784	9.24	2.97	764.	0.6786	TU-3 MO	5.59
2.784	10.39	3.71	1073.	0.6429	TU-3 MO	6.35
2.789	10.14	3.73	1055.	0.6321	TU-3 MO	6.35

US = A0 + A1*UP KM/SEC

A0 = 5.393 KM/SEC

A1 = 1.308 -

SIG A0 = 0.038 KM/SEC

SIG A1 = 0.020 -

SIG US = 0.078 -

TABLE VI

-----SAMPLE-----					-----STANDARD-----		
RHO	US	UP	P	V/V0	MATERIAL	US(ST)	
2.571	4.78	0.72	88.	0.8494	2024 AL	6.17	
2.647	5.36	1.00	142.	0.8134	2024 AL	6.53	
2.524	5.16	1.03	134.	0.8004	2024 AL	6.52	
2.575	5.98	1.33	205.	0.7775	2024 AL	6.94	
2.566	6.85	1.76	309.	0.7431	2024 AL	7.52	
2.532	6.00	1.92	292.	0.6800	2024 AL	7.58	
2.472	6.93	1.99	341.	0.7128	2024 AL	7.75	
2.554	7.55	2.23	430.	0.7046	2024 AL	8.12	
2.465	7.91	2.58	503.	0.6738	2024 AL	8.52	
2.554	8.32	2.70	574.	0.6755	2024 AL	8.74	
2.605	8.37	2.79	608.	0.6667	2024 AL	8.87	
2.583	8.81	3.03	690.	0.6561	2024 AL	9.19	
2.506	8.72	3.05	666.	0.6502	2024 AL	9.17	
2.603	9.09	3.23	764.	0.6447	2024 AL	9.47	
2.535	9.15	3.30	765.	0.6393	2024 AL	9.52	
2.585	9.59	3.55	880.	0.6298	2024 AL	9.89	
2.543	9.58	3.59	875.	0.6253	2024 AL	9.90	
2.522	9.54	3.62	871.	0.6205	2024 AL	9.92	
2.510	9.63	3.64	880.	0.6220	2024 AL	9.95	
2.631	9.84	3.73	966.	0.6209	2024 AL	10.15	
2.618	9.84	3.80	979.	0.6138	2024 AL	10.22	
2.617	9.90	3.82	990.	0.6141	2024 AL	10.25	

US =

TABLE VII

-----SAMPLE-----					-----STANDARD-----		
RHO	US	UP	P	V/V0	MATERIAL	US(ST)	
2.241	3.47	0.85	66.	0.7550	2024 AL	6.17	
2.261	4.30	1.14	111.	0.7349	2024 AL	6.52	
2.183	4.10	1.18	106.	0.7122	2024 AL	6.53	
2.062	4.89	1.55	156.	0.6830	2024 AL	6.94	
2.271	6.12	1.93	268.	0.6848	2024 AL	7.52	
2.136	6.54	1.98	277.	0.6972	2024 AL	7.58	
2.247	6.40	2.12	305.	0.6687	2024 AL	7.75	
2.193	6.90	2.44	369.	0.6464	2024 AL	8.12	
2.260	7.63	2.70	468.	0.6461	2024 AL	8.52	
2.268	7.91	2.88	516.	0.6359	2024 AL	8.74	
2.188	7.88	3.04	524.	0.6142	2024 AL	8.87	
2.301	8.47	3.21	626.	0.6210	2024 AL	9.19	
2.203	8.30	3.26	596.	0.6072	2024 AL	9.17	
2.280	8.53	3.48	677.	0.5920	2024 AL	9.47	
2.198	8.80	3.53	683.	0.5989	2024 AL	9.52	
2.287	9.27	3.77	799.	0.5933	2024 AL	9.89	
2.208	9.27	3.83	784.	0.5868	2024 AL	9.90	
2.217	9.26	3.84	788.	0.5893	2024 AL	9.92	
2.229	9.26	3.86	798.	0.5841	2024 AL	9.95	
2.211	9.84	4.01	895.	0.5840	2024 AL	10.15	

2024 ALUMINUM

RHO0	US	UP	P	V/V0	MATERIAL US(ST)
2.265	9.65	4.04	883.	0.5813	2024 AL 10.22
2.289	9.57	4.06	889.	0.5758	2024 AL 10.25

US =

TABLE VIII

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL US(ST)	
1.993	2.91	0.92	53.	0.6838	2024 AL	6.17
1.957	2.77	0.93	50.	0.6643	2024 AL	6.16
1.966	3.62	1.25	89.	0.6547	2024 AL	6.52
1.929	3.62	1.27	89.	0.6492	2024 AL	6.53
1.908	4.40	1.64	138.	0.6273	2024 AL	6.94
2.010	5.52	2.07	234.	0.6317	2024 AL	7.52
1.970	5.49	2.16	234.	0.6066	2024 AL	7.58
1.995	5.99	2.26	270.	0.6227	2024 AL	7.75
1.981	6.61	2.57	337.	0.6112	2024 AL	8.12
1.916	7.05	2.94	397.	0.5830	2024 AL	8.52
1.959	7.47	3.09	452.	0.5863	2024 AL	8.74
1.880	7.56	3.25	462.	0.5701	2024 AL	8.87
2.019	8.00	3.40	549.	0.5750	2024 AL	9.17
2.037	8.04	3.42	560.	0.5746	2024 AL	9.19
1.933	8.48	3.69	605.	0.5649	2024 AL	9.47
1.888	8.38	3.78	598.	0.5489	2024 AL	9.52
2.065	8.92	3.99	735.	0.5527	2024 AL	9.92
1.996	9.02	4.01	722.	0.5554	2024 AL	9.90
1.983	9.03	4.06	727.	0.5504	2024 AL	9.95
1.825	8.76	4.15	663.	0.5263	2024 AL	9.89
1.916	9.16	4.29	753.	0.5317	2024 AL	10.15
1.951	9.32	4.33	787.	0.5354	2024 AL	10.25
1.899	9.25	4.36	766.	0.5286	2024 AL	10.22

US =

TABLE IX

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL US(ST)	
1.853	2.47	0.98	40.	0.6032	2024 AL	6.16
1.622	2.31	1.01	38.	0.5828	2024 AL	6.17
1.699	3.33	1.34	76.	0.5975	2024 AL	6.53
1.618	2.98	1.38	67.	0.5359	2024 AL	6.52
1.648	4.05	1.75	117.	0.5879	2024 AL	6.94
1.663	5.15	2.25	193.	0.5831	2024 AL	7.52
1.687	5.13	2.31	200.	0.5497	2024 AL	7.58
1.626	5.04	2.45	225.	0.5658	2024 AL	7.75

2024 ALUMINUM

RHO0	US	UP	P	V/V0	MATERIAL US(ST)
1.700	6.26	2.74	292.	0.5623	2024 AL 8.12
1.661	6.68	3.12	346.	0.5329	2024 AL 8.52
1.671	7.09	3.30	391.	0.5346	2024 AL 8.74
1.687	7.41	3.38	423.	0.5439	2024 AL 8.87
1.695	7.60	3.57	470.	0.5171	2024 AL 9.17
1.691	7.74	3.57	480.	0.5258	2024 AL 9.19
1.657	8.18	3.92	531.	0.5208	2024 AL 9.47
1.656	8.20	3.97	539.	0.5159	2024 AL 9.52
1.707	8.76	4.24	634.	0.5160	2024 AL 9.89
1.596	8.71	4.35	605.	0.5006	2024 AL 9.90
1.603	8.65	4.37	606.	0.4948	2024 AL 9.92
1.544	8.64	4.45	584.	0.4850	2024 AL 9.95
1.673	9.00	4.50	678.	0.5000	2024 AL 10.15
1.713	9.02	4.54	701.	0.4967	2024 AL 10.22
1.694	9.12	4.57	706.	0.4989	2024 AL 10.25

US =

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, H.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES.
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: I B
DATA REDUCTION TECHNIQUE: A (TABLE I) B (THE REST)
- 3) TABLE I IS THE HUGONIOT OBTAINED WITH SYMMETRIC IMPACT. TABLE II THROUGH V CHECK THE INTERNAL CONSISTENCY OF THE SYMMETRIC IMPACT HUGONIOTS OF CU, FE U-MO ALLOY AND 921T AL.
- 4) THE GRUNEISEN GAMMA OF 2.00 (= GAM) WITH THE ASSUMPTION THAT RHO0.GAM = CONSTANT REPRESENTS THE BEHAVIOR OF THE POROUS SAMPLES. ALSO GIVEN ARE (DV/DT)/V=6.54 E-5 PER DEG. AND CP = 0.879 JOULES PER G. DEG.
- 5) HUGONIOT ELASTIC LIMIT 5.4 KBAR.
- 6) THE 921-T AL PLATE STOCK USED TO OBTAIN THE CROSS CHECK DATA IN TABLE II IS SLIGHTLY POROUS.

TABLE I

2024 ALUMINUM
29---15

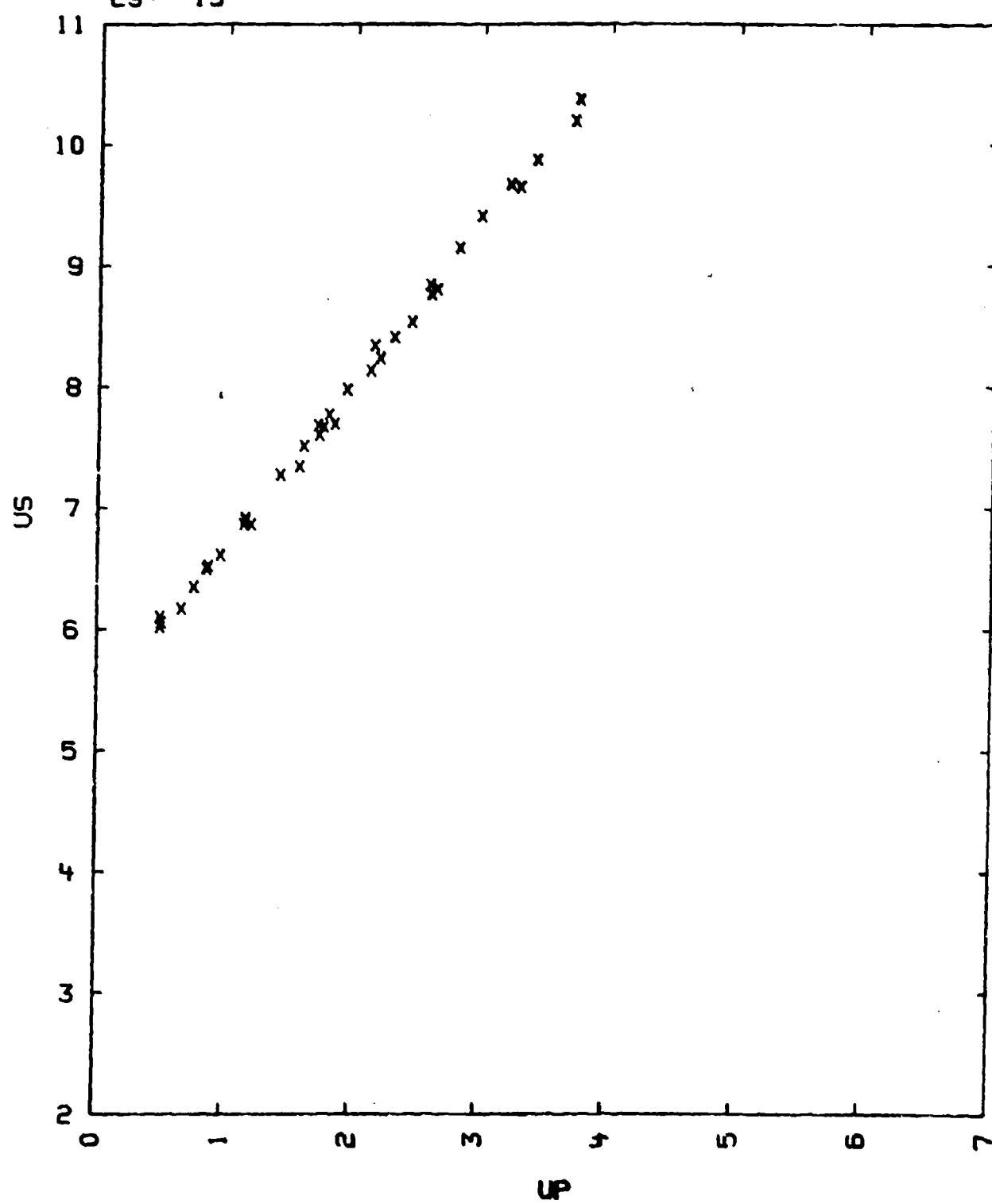


TABLE II

2024 ALUMINUM

29---15

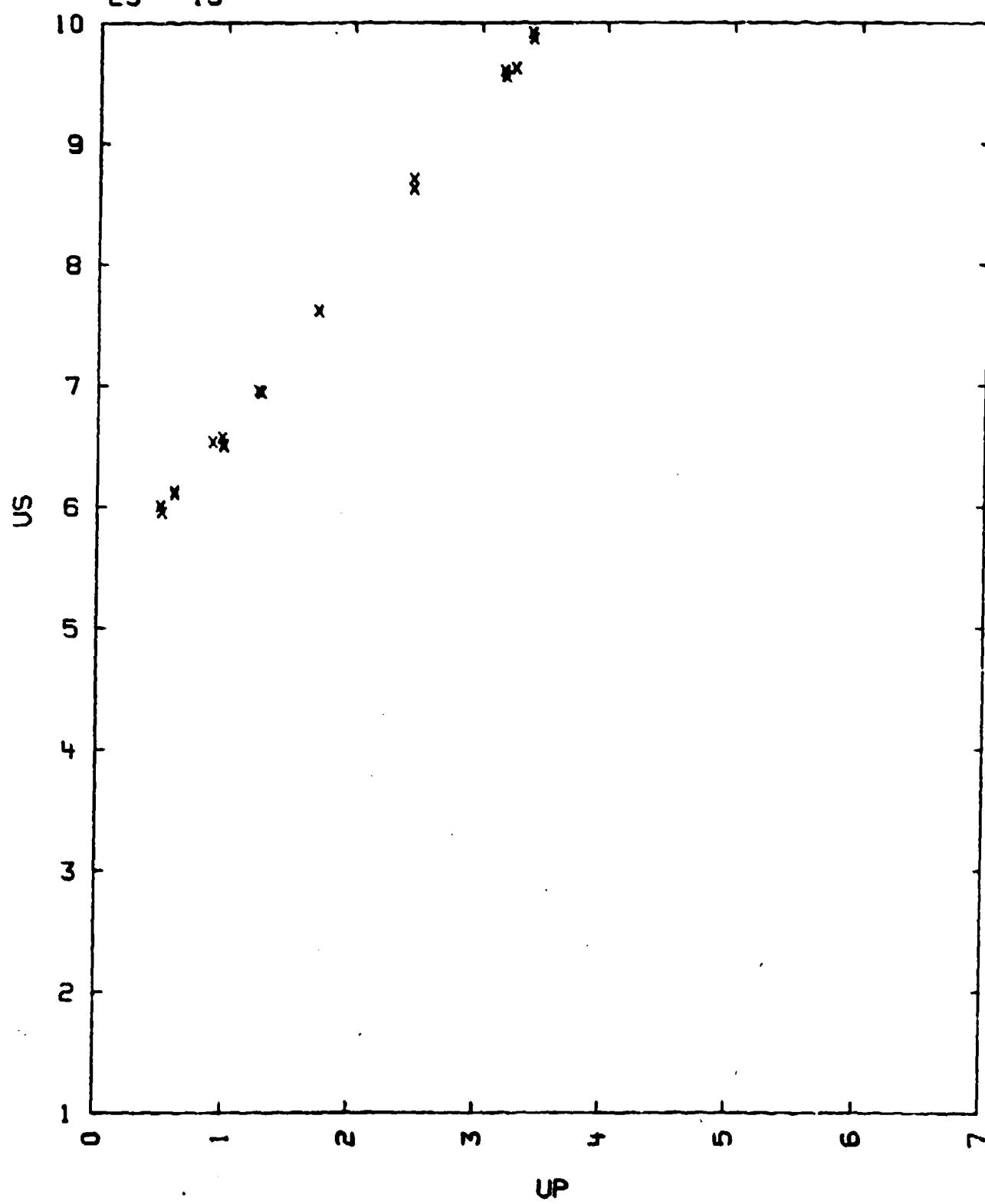


TABLE III

2024 ALUMINUM
29---15

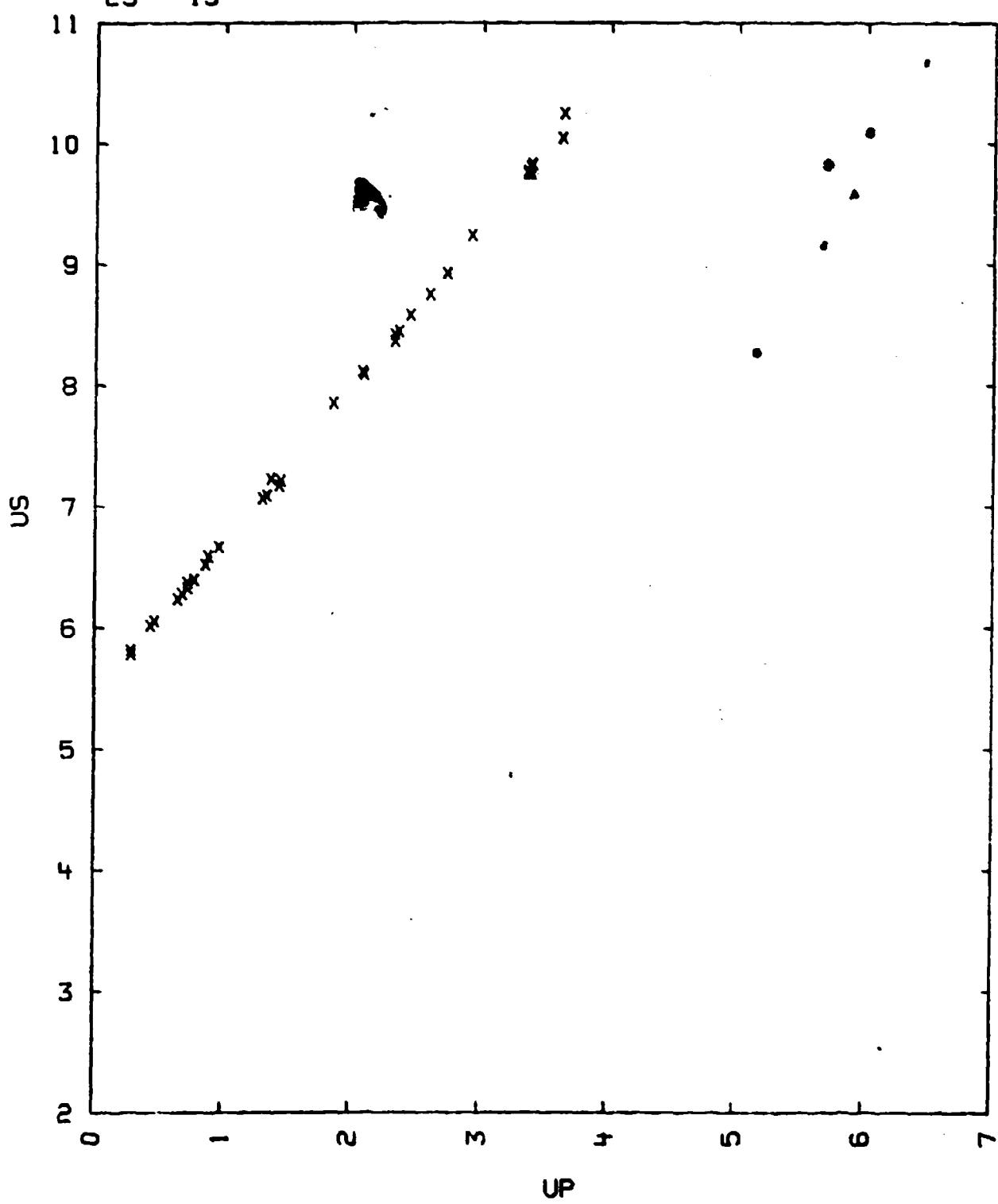


TABLE IV

2024 ALUMINUM
29---15

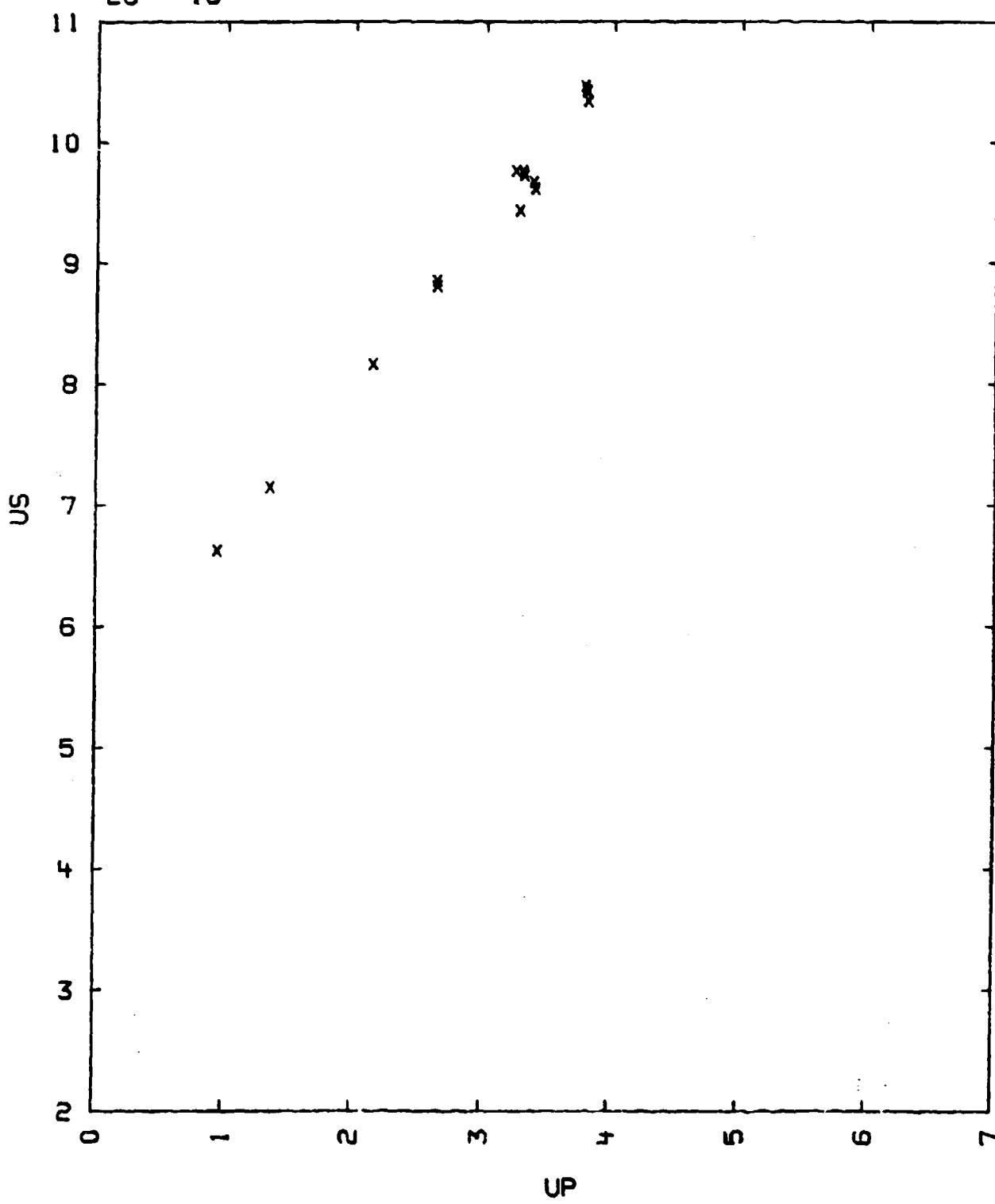


TABLE V

2024 ALUMINUM
29---15

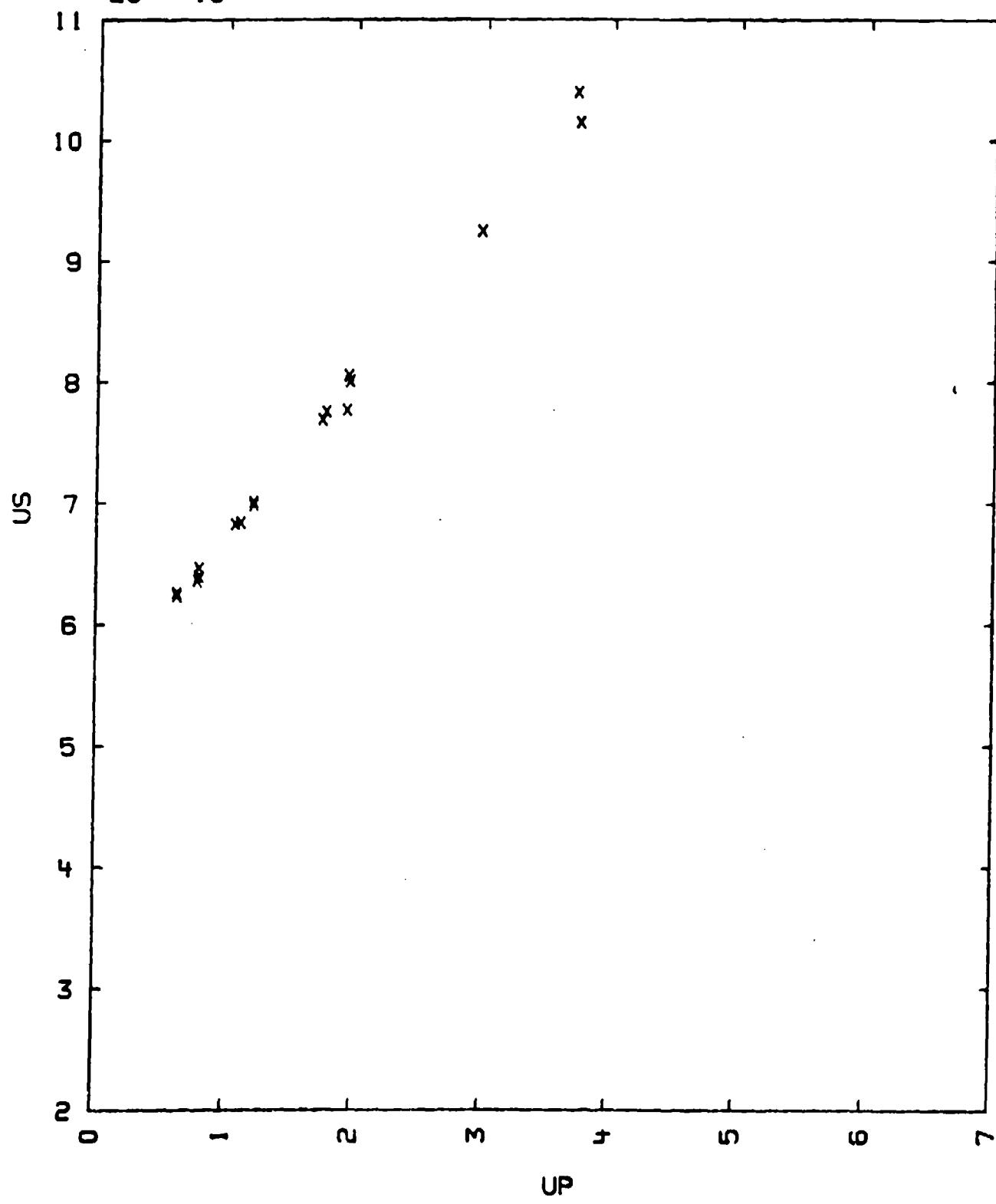


TABLE VI

2024 ALUMINUM

29---15

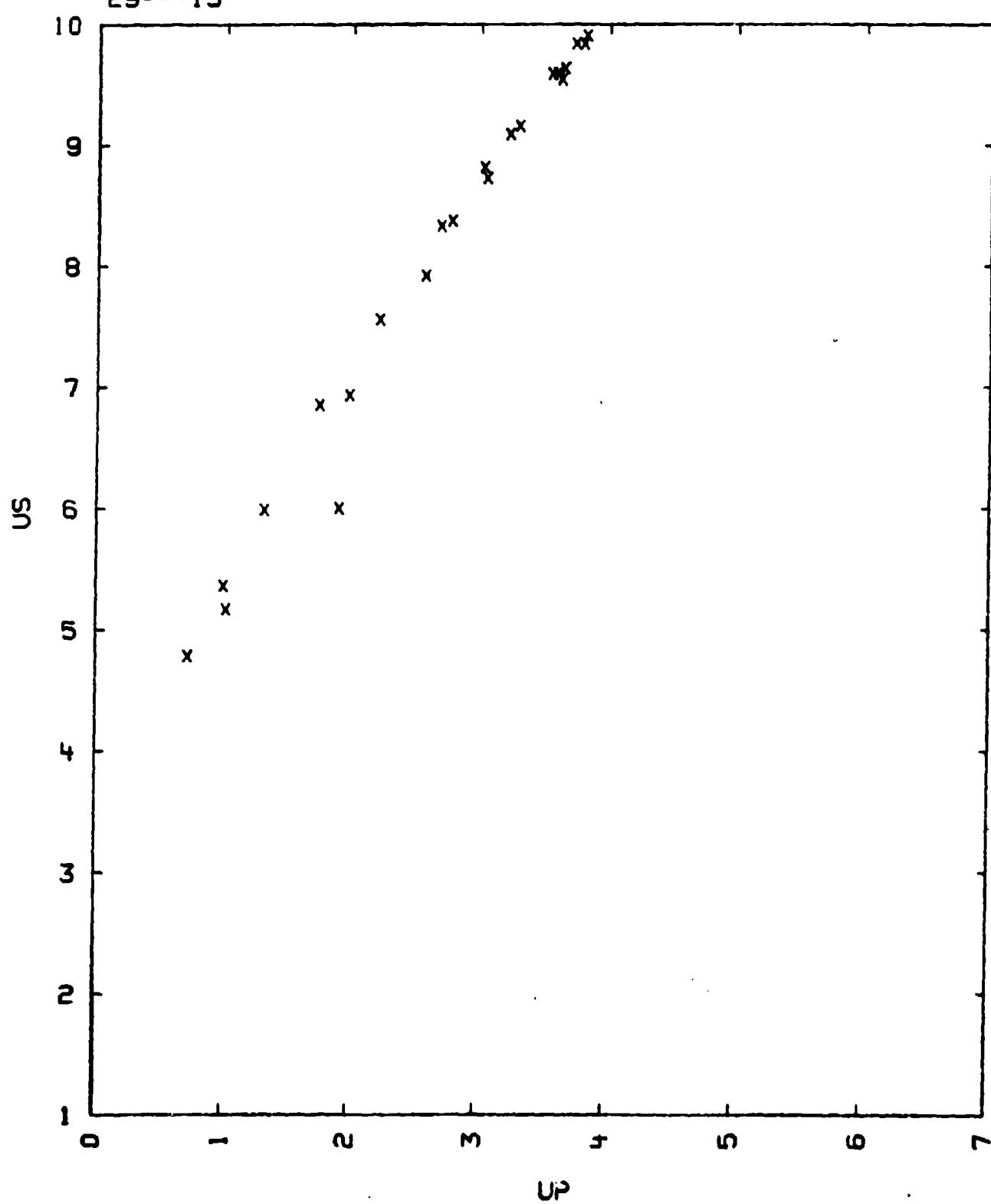


TABLE VII

2024 ALUMINUM
29---15

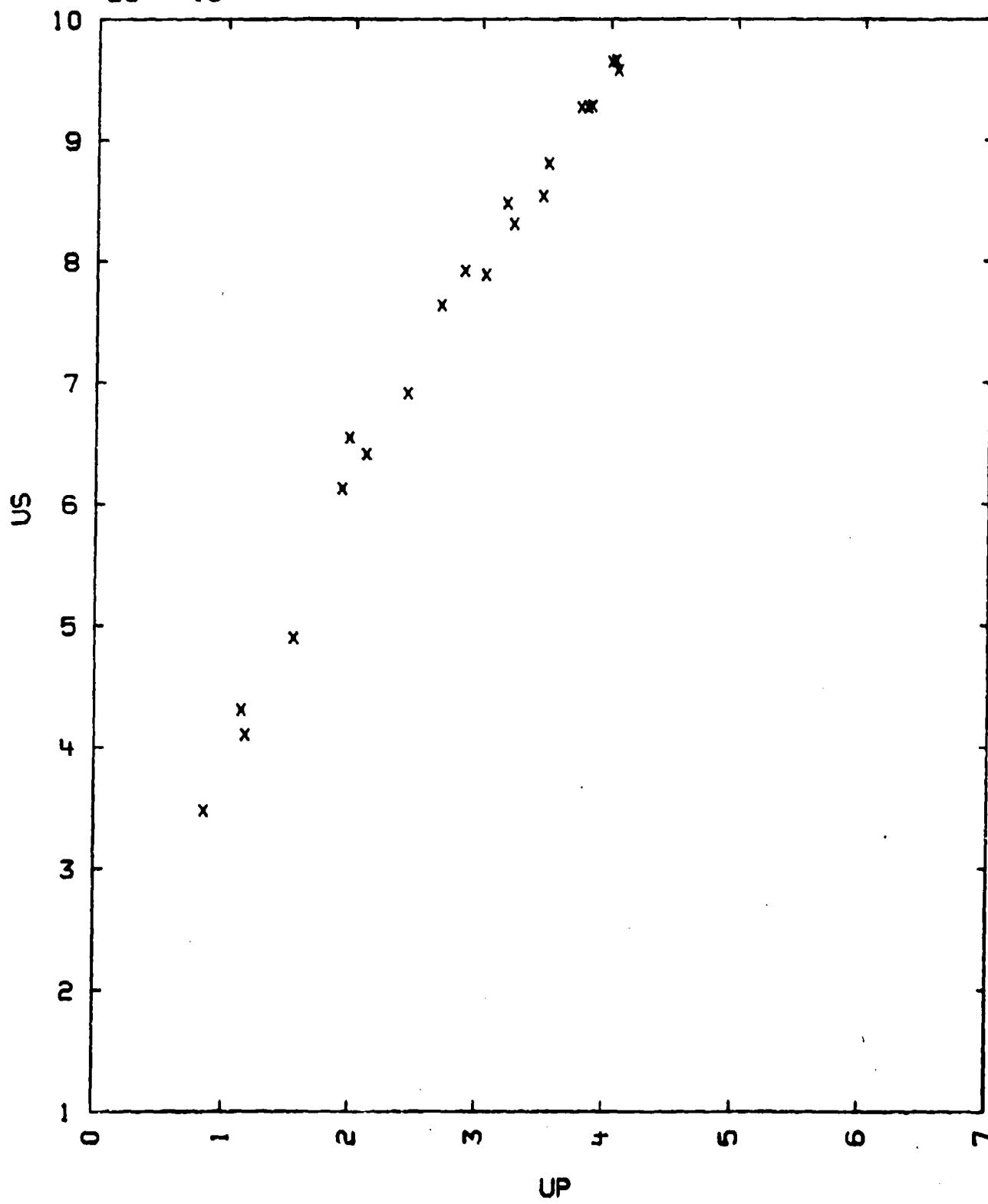


TABLE VIII

2024 ALUMINUM

29---15

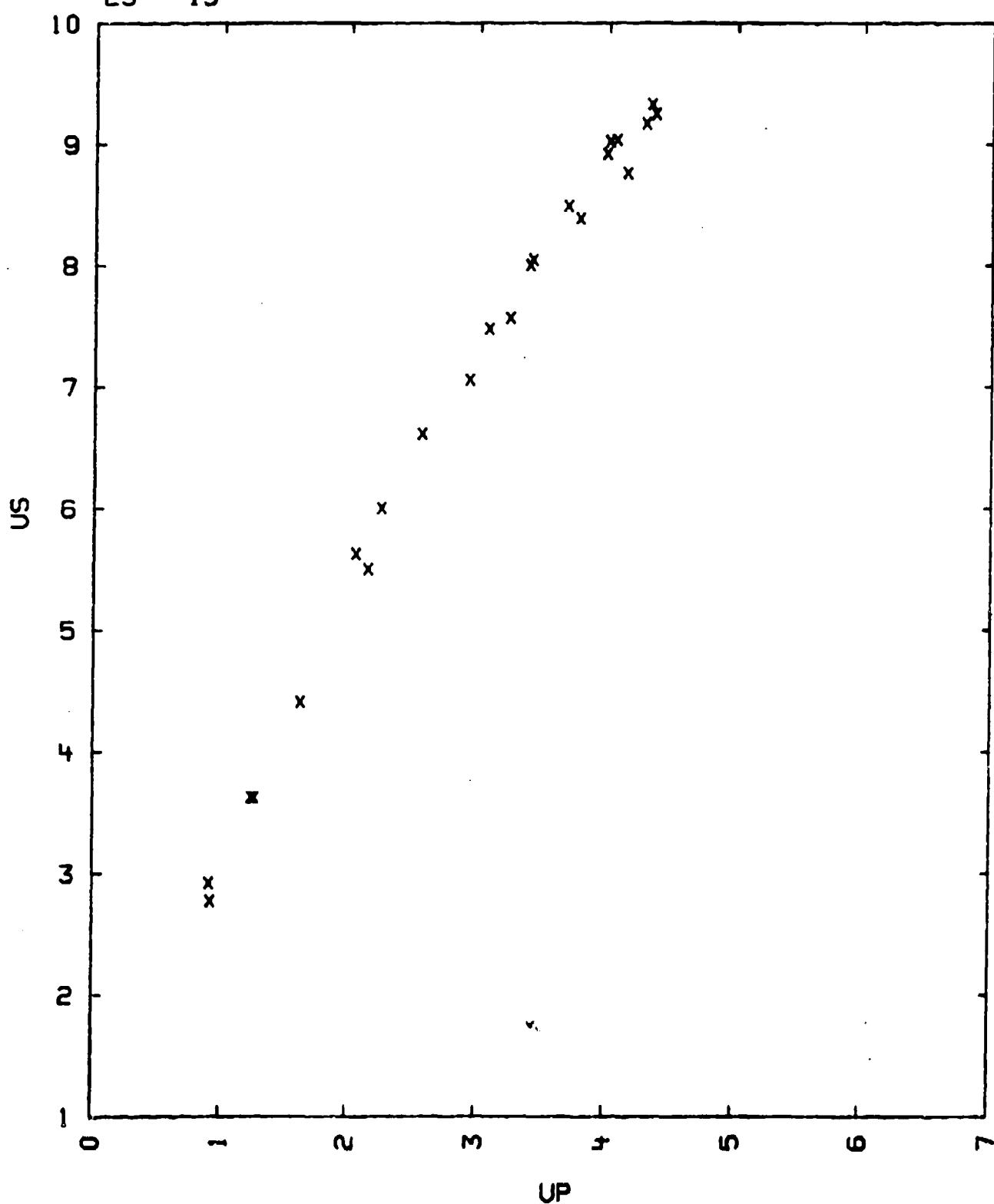
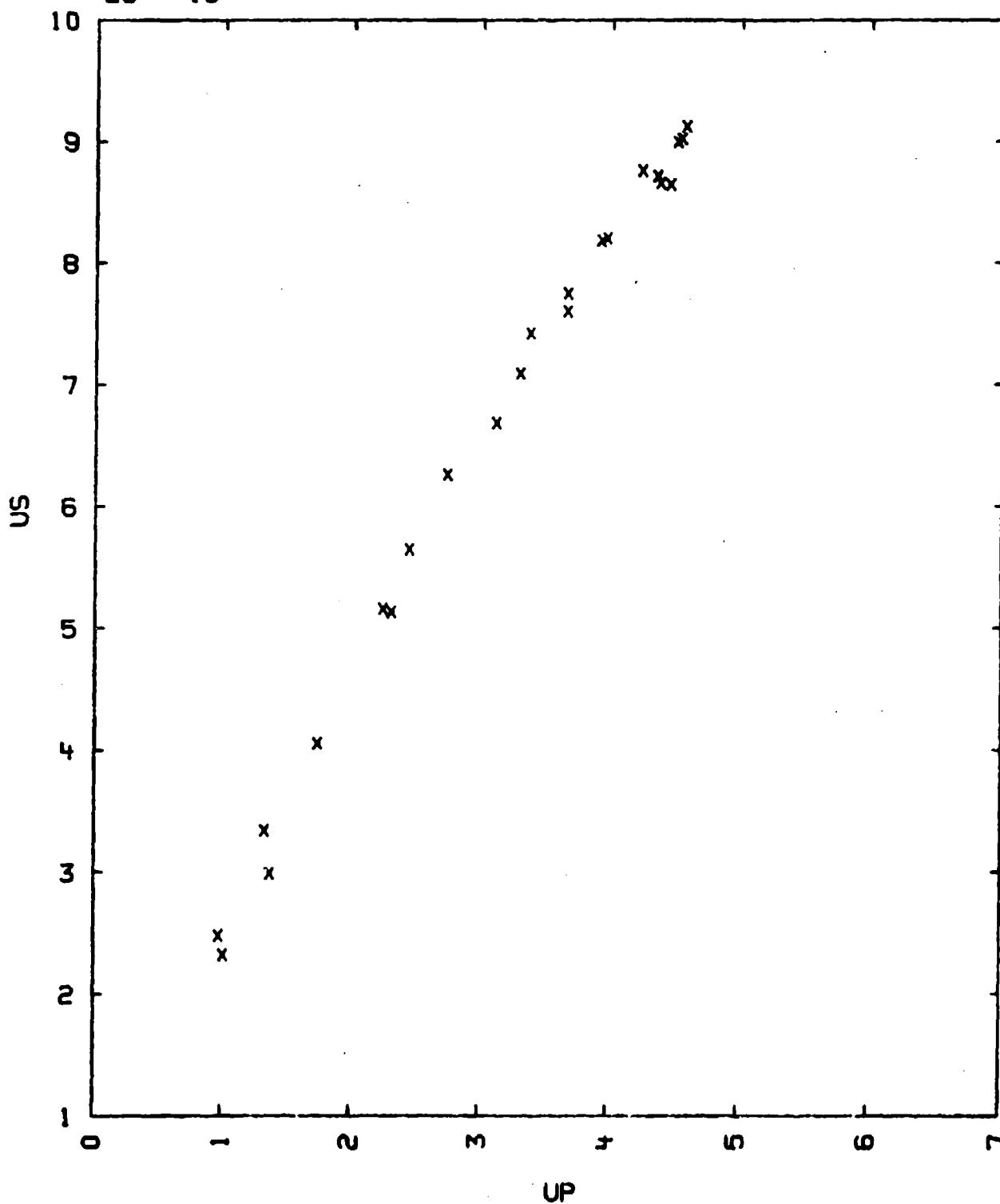


TABLE IX

2024 ALUMINUM
29---15



29---16
921T ALUMINUM

AL	REST				
ZN	0.5 - 1.	WT. PERCENT			
CU	3. - 4.	WT. PERCENT			
SI	1.1 - 1.35	WT. PERCENT			
FE	1.0 MAX	WT. PERCENT			
MN	0.5 MAX	WT. PERCENT			
MG	0.3 - 0.7	WT. PERCENT			
CR	0.15 MAX	WT. PERCENT			
TI	0.15 MAX	WT. PERCENT			

$$V_0 = 0.3519 - 0.3559 \text{ CC/G} \quad C_L = 6.34 \text{ KM/SEC} \quad C_D = 5.20 \text{ KM/SEC}$$

$$C_S = 3.14 \text{ KM/SEC}$$

IN THE TABLES BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBAR AND DENSITY IN G/CC. THE IMPACTOR IS THE STANDARD.

TABLE I

-----SAMPLE-----					-----STANDARD-----
RHO0	US	UP	P	V/V0	MATERIAL
2.810	5.72	0.50	80.	0.9126	921T AL
2.810	5.74	0.53	85.	0.9077	921T AL
2.810	5.98	0.63	106.	0.8946	921T AL
2.811	5.88	0.65	107.	0.8895	921T AL
2.828	6.72	1.18	226.	0.8229	921T AL
2.828	7.04	1.40	279.	0.8011	921T AL
2.828	7.18	1.48	303.	0.7925	921T AL
2.828	7.21	1.57	320.	0.7822	921T AL
2.828	7.42	1.63	342.	0.7803	921T AL
2.828	7.42	1.63	342.	0.7803	921T AL
2.828	7.73	1.88	407.	0.7594	921T AL
2.828	8.06	2.13	486.	0.7357	921T AL
2.828	8.08	2.13	487.	0.7367	921T AL
2.828	8.35	2.28	538.	0.7269	921T AL
2.828	8.42	2.40	571.	0.7150	921T AL
2.828	8.51	2.44	587.	0.7133	921T AL
2.828	9.07	2.87	738.	0.6835	921T AL
2.828	9.38	3.08	820.	0.6705	921T AL
2.828	9.40	3.10	824.	0.6702	921T AL

US = $5.041 + 1.420 \cdot UP$ KM/SEC HUGONIOT FIT ADOPTED BY THE SOURCE
COMPARISON LEAST SQUARE FITS:

US = $A_0 + A_1 \cdot UP$ KM/SEC

$$A_0 = 5.038 \text{ KM/SEC} \quad S10 A_0 = 0.027 \text{ KM/SEC}$$

$$A_1 = 1.420 \quad - \quad S10 A_1 = 0.013 \quad -$$

$$S10 US = 0.049 \quad -$$

TABLE II

-----SAMPLE-----					-----STANDARD-----		
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)	
2.812	5.46	0.35	54.	0.9359	2024 AL	5.79	
2.812	5.51	0.35	54.	0.9365	2024 AL	5.79	
2.838	5.69	0.46	74.	0.9192	2024 AL	5.94	
2.812	5.78	0.46	75.	0.9204	2024 AL	5.94	
2.812	5.75	0.47	76.	0.9183	2024 AL	5.93	
2.812	5.71	0.48	77.	0.9159	2024 AL	5.98	
2.814	5.75	0.49	79.	0.9148	2024 AL	5.98	
2.812	5.83	0.49	80.	0.9160	2024 AL	5.98	
2.812	6.03	0.84	108.	0.8939	2024 AL	6.18	
2.812	6.33	0.86	153.	0.8641	2024 AL	6.48	
2.842	6.37	0.90	163.	0.8587	2024 AL	6.53	
2.814	6.42	0.90	163.	0.8598	2024 AL	6.53	
2.842	6.39	0.91	165.	0.8576	2024 AL	6.55	
2.842	6.37	0.91	165.	0.8571	2024 AL	6.55	
2.842	6.76	1.15	221.	0.8299	2024 AL	6.87	
2.842	6.74	1.15	220.	0.8294	2024 AL	6.87	
2.814	6.57	1.20	222.	0.8174	2024 AL	6.87	
2.842	7.35	1.61	336.	0.7810	2024 AL	7.48	
2.814	7.47	1.66	349.	0.7778	2024 AL	7.48	
2.842	8.45	2.34	562.	0.7231	2024 AL	7.55	
2.842	8.42	2.34	560.	0.7221	2024 AL	8.48	
2.803	8.82	2.65	655.	0.6993	2024 AL	8.48	
2.842	8.34	3.02	802.	0.6767	2024 AL	8.87	
2.842	9.33	3.03	803.	0.6752	2024 AL	9.40	
2.842	9.68	3.28	902.	0.6612	2024 AL	9.40	
2.842	9.67	3.28	901.	0.6608	2024 AL	9.74	
2.838	9.71	3.38	931.	0.6519	2024 AL	9.86	
2.838	10.21	3.68	1068.	0.6396	2024 AL	10.28	

US = A0 + A1 * UP KM/SEC

A0 = 5.089 KM/SEC

A1 = 1.400 -

SIG A0 = 0.021 KM/SEC

SIG A1 = 0.011 -

SIG US = 0.065 -

TABLE III

-----SAMPLE-----					-----STANDARD-----		
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)	
2.838	5.52	0.28	44.	0.9493	CU	4.24	
2.842	5.86	0.47	78.	0.9198	CU	4.45	
2.814	6.19	0.72	125.	0.8837	CU	4.72	
2.838	6.17	0.73	128.	0.8817	CU	4.72	
2.814	6.17	0.73	127.	0.8817	CU	4.73	
2.838	6.19	0.74	130.	0.8805	CU	4.78	
2.812	6.29	0.78	138.	0.8780	CU	4.90	
2.842	6.40	0.88	162.	0.8609	CU	4.91	
2.838	6.40	0.80	163.	0.8594	CU	4.91	
2.838	6.40	0.80	163.	0.8594	CU	4.98	
2.822	6.53	0.97	178.	0.8515	CU	5.37	
2.842	6.95	1.32	261.	0.8101	CU		

921T ALUMINUM

RHO0	US	UP	P	V/V0	MATERIAL US(ST)
2.842	7.05	1.38	276.	0.8043	CU 5.44
2.814	7.08	1.44	287.	0.7956	CU 5.49
2.822	7.22	1.44	293.	0.8006	CU 5.51
2.842	7.70	1.85	407.	0.7584	CU 5.97
2.822	8.06	2.09	475.	0.7407	CU 6.23
2.842	8.27	2.32	545.	0.7195	CU 6.48
2.814	8.36	2.33	548.	0.7213	CU 6.49
2.812	8.28	2.51	584.	0.6969	CU 6.68
2.842	8.69	2.59	640.	0.7020	CU 6.79
2.842	8.82	2.73	684.	0.6905	CU 6.93
2.842	8.81	2.73	684.	0.6901	CU 6.94
2.822	9.15	2.93	757.	0.6798	CU 7.16
2.842	9.75	3.33	923.	0.6585	CU 7.62
2.822	9.66	3.37	919.	0.6511	CU 7.65
2.842	9.76	3.38	938.	0.6537	CU 7.67

US = A0 + A1*UP KM/SEC

A0 = 5.207 KM/SEC

A1 = 1.333 -

SIG A0 = 0.027 KM/SEC

SIG A1 = 0.014 -

SIG US = 0.069 -

TABLE IV

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL US(ST)	
2.838	6.49	0.95	175.	0.8536	FE	4.93
2.838	7.08	1.36	273.	0.8079		5.49
2.838	8.16	2.14	496.	0.7377	FE	6.53
2.838	8.73	2.64	654.	0.6976	FE	7.15
2.838	9.59	3.24	882.	0.6621	FE	7.90
1.838	9.73	3.28	587.	0.6629	FE	7.96
2.838	9.75	3.36	930.	0.6554	FE	8.05
2.838	10.35	3.76	1104.	0.6367	FE	8.54
2.838	10.30	3.77	1102.	0.6340	FE	8.54

US = A0 + A1*UP KM/SEC

A0 = 5.218 KM/SEC.

A1 = 1.358 -

SIG A0 = 0.044 KM/SEC

SIG A1 = 0.015 -

SIG US = 0.045 -

TABLE V

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL US(ST)	
2.812	6.29	0.79	140.	0.8744	TU-3 MO	3.37
2.812	6.30	0.79	140.	0.8746	TU-3 MO	3.37
2.812	6.76	1.12	213.	0.8343	TU-3 MO	3.70

921T ALUMINUM

RHO0	US	UP	P	V/V0	MATERIAL US(ST)
2.842	7.05	1.38	276.	0.8043	CU 5.44
2.814	7.08	1.44	287.	0.7966	CU 5.49
2.822	7.22	1.44	293.	0.8006	CU 5.51
2.842	7.70	1.86	407.	0.7584	CU 5.97
2.822	8.06	2.09	473.	0.7407	CU 6.23
2.842	8.27	2.32	545.	0.7195	CU 6.48
2.814	8.36	2.33	548.	0.7213	CU 6.49
2.812	8.28	2.51	584.	0.6969	CU 6.66
2.842	8.69	2.59	640.	0.7020	CU 6.79
2.842	8.82	2.73	684.	0.6905	CU 6.93
2.842	8.81	2.73	684.	0.6901	CU 6.94
2.822	9.15	2.93	757.	0.6798	CU 7.16
2.842	9.75	3.33	923.	0.6585	CU 7.62
2.822	9.66	3.37	919.	0.6511	CU 7.65
2.842	9.76	3.38	938.	0.6537	CU 7.67

US = A0 + A1*UP KM/SEC

A0 = 5.207 KM/SEC

A1 = 1.333 -

SIG A0 = 0.027 KM/SEC

SIG A1 = 0.014 -

SIG US = 0.069 -

TABLE IV

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL US(ST)	
2.838	6.49	0.95	175.	0.8536	FE	4.93
2.838	7.08	1.36	273.	0.8079		5.49
2.838	8.16	2.14	496.	0.7377	FE	6.53
2.838	8.73	2.64	654.	0.6976	FE	7.15
2.838	9.59	3.24	882.	0.6621	FE	7.90
1.838	9.73	3.28	587.	0.6629	FE	7.96
2.838	9.75	3.36	930.	0.6554	FE	8.05
2.838	10.35	3.76	1104.	0.6367	FE	8.54
2.838	10.30	3.77	1102.	0.6340	FE	8.54

US = A0 + A1*UP KM/SEC

A0 = 5.218 KM/SEC

A1 = 1.356 -

SIG A0 = 0.044 KM/SEC

SIG A1 = 0.015 -

SIG US = 0.045 -

TABLE V

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL US(ST)	
2.812	6.29	0.79	140.	0.8744	TU-3 MO	3.37
2.812	6.30	0.79	140.	0.8746	TU-3 MO	3.37
2.812	6.76	1.12	213.	0.8343	TU-3 MO	3.70

921T ALUMINUM

RHO0	US	UP	P	V/V0	MATERIAL US(ST)
2.812	8.87	1.22	236.	0.8224	TU-3 MO 3.81
2.812	6.88	1.22	236.	0.8227	TU-3 MO 3.81
2.814	7.82	1.74	373.	0.7717	TU-3 MO 4.34
2.814	7.60	1.74	372.	0.7711	TU-3 MO 4.34
2.838	7.79	1.77	391.	0.7728	TU-3 MO 4.38
2.838	7.70	1.77	387.	0.7701	TU-3 MO 4.38
2.814	7.94	1.92	429.	0.7582	TU-3 MO 4.53
2.814	7.88	1.93	428.	0.7551	TU-3 MO 4.53
2.838	9.12	2.97	769.	0.6743	TU-3 MO 5.59
2.838	9.10	2.87	767.	0.6736	TU-0 MO 5.59
2.838	10.07	3.72	1063.	0.6306	TU-3 MO 6.35

$$US = A0 + A1 \cdot UP \text{ KM/SEC}$$

$$A0 = 5.358 \text{ KM/SEC} \quad SIG A0 = 0.052 \text{ KM/SEC}$$

$$A1 = 1.282 \quad - \quad SIG A1 = 0.026 \quad -$$

$$SIG US = 0.080 \quad -$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE A (TABLE I), B (THE REST)
- 3) TABLE I LISTS THE STANDARD HUGONIOT OBTAINED WITH SYMMETRIC IMPACT.
TABLES II THROUGH V CHECK THE INTERNAL CONSISTENCY OF THE OTHER
STANDARDS OBTAINED WITH SYMMETRIC IMPACT.
- 4) ALSO GIVEN ARE $V(DP/DE) = 2.10$, $(DV/DT)/V = 6.87E-5$ PER DEG AND
 $CP = 0.877$ JOULES PER G. DEG.
- 5) THE DESIGNATION 921T IS A PIONEER ALUMINUM NOT AN A.S.M. DESIGNATION
THE COMPOSITION WAS OBTAINED FROM WAYNE HERRIN, PIONEER ALUMINUM CO.
PRIVATE COMMUNICATION (1975).
- 6) THE 921-T AL PLATE STOCK USED IN TABLE I IS SOMEWHAT POROUS. THE
PELLETS ON WHICH THE DATA IN TABLES II THROUGH V WERE OBTAINED ARE
MORE NEARLY IDEAL DENSITY

TABLE I

921T ALUMINUM
29---16

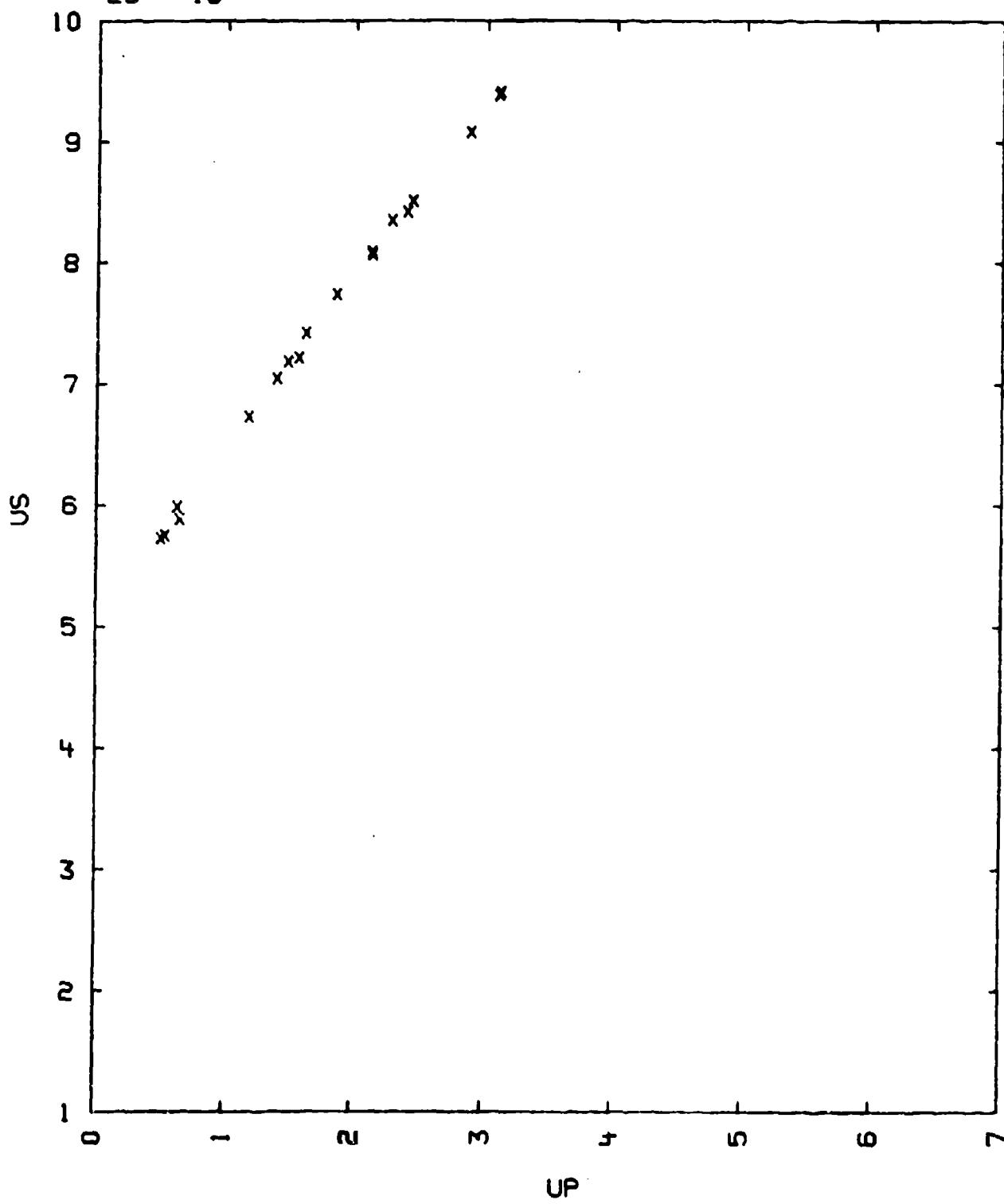


TABLE II

921T ALUMINUM
29---16

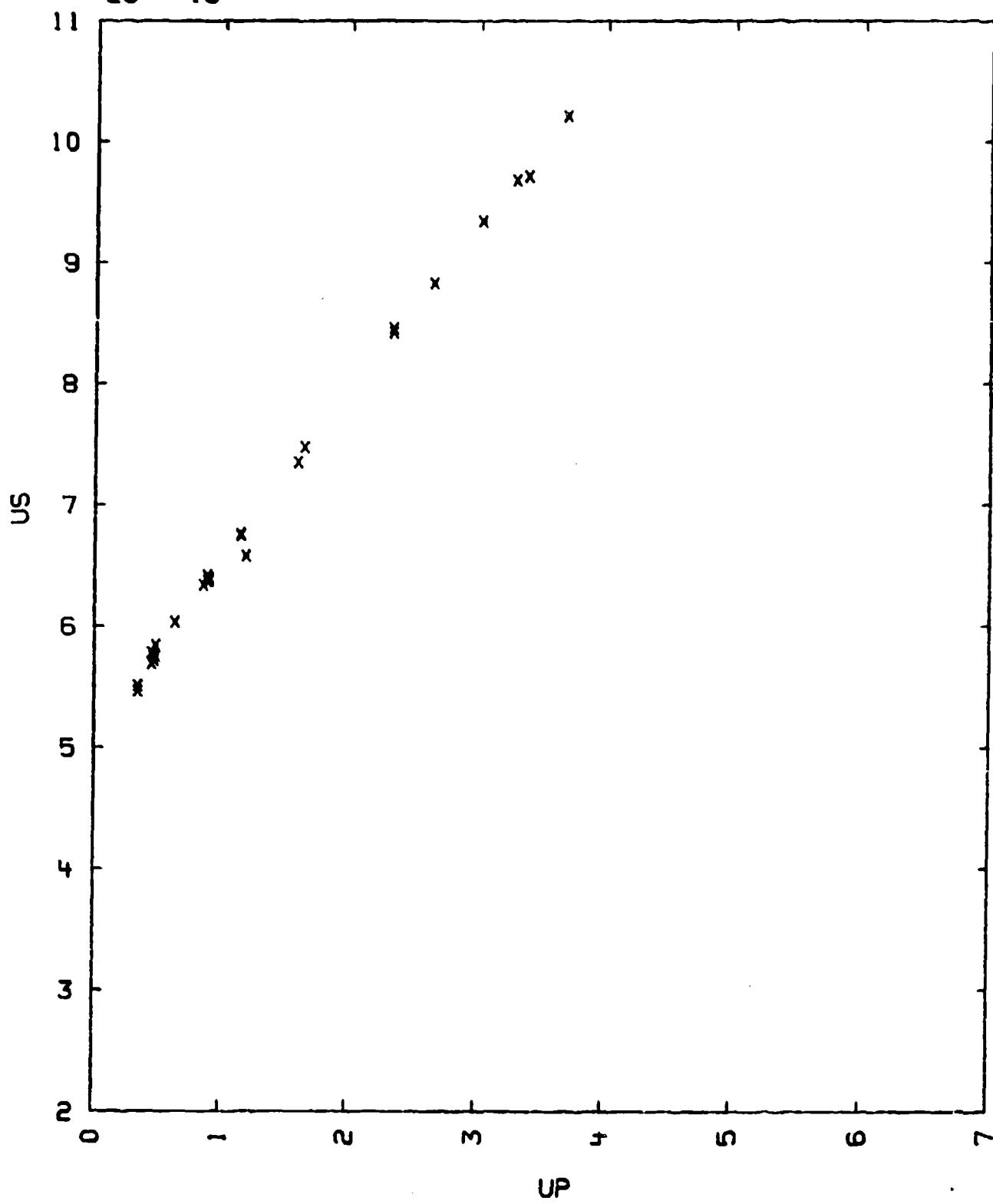


TABLE III

921T ALUMINUM
29---16

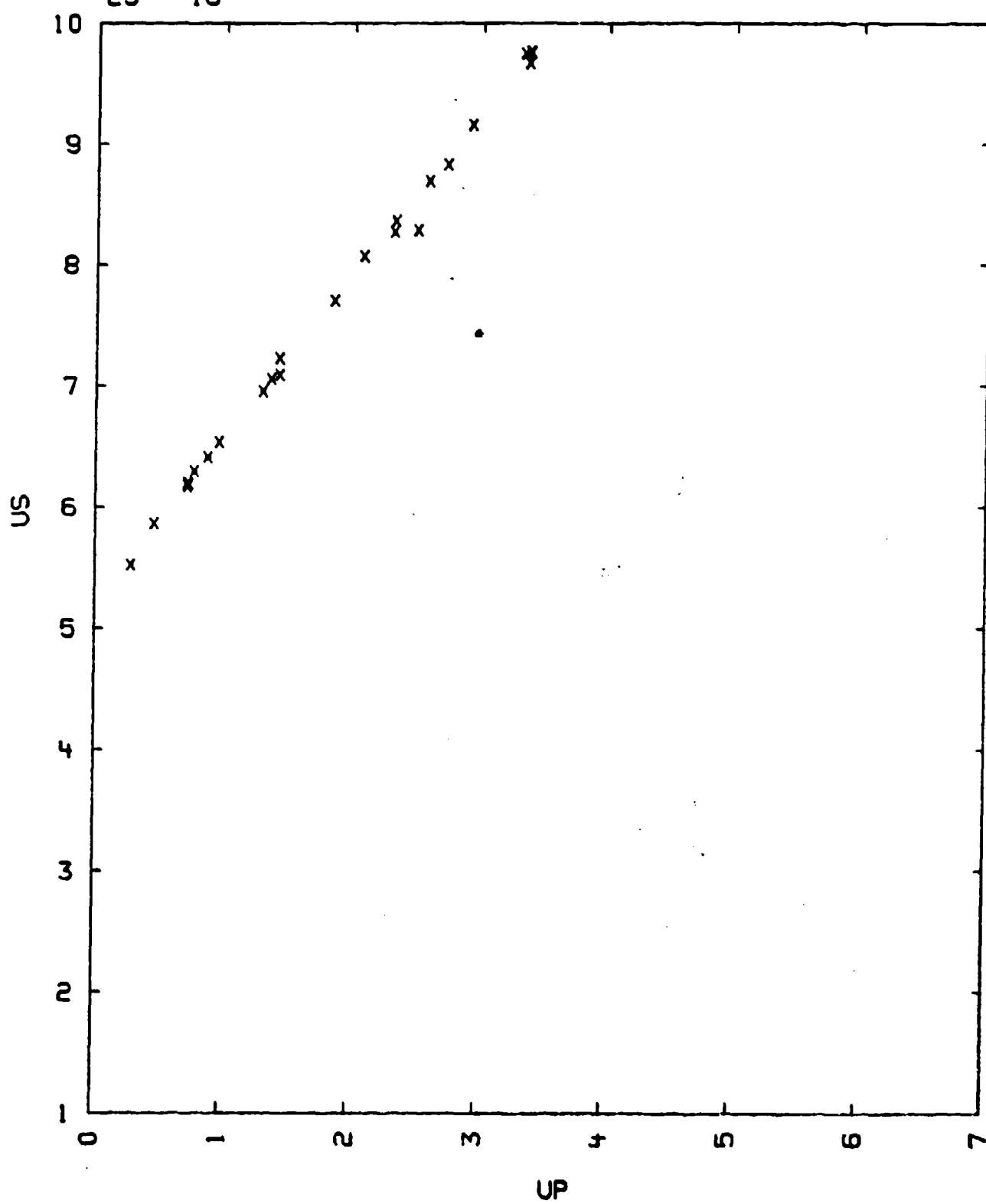


TABLE IV

921T ALUMINUM
29---16

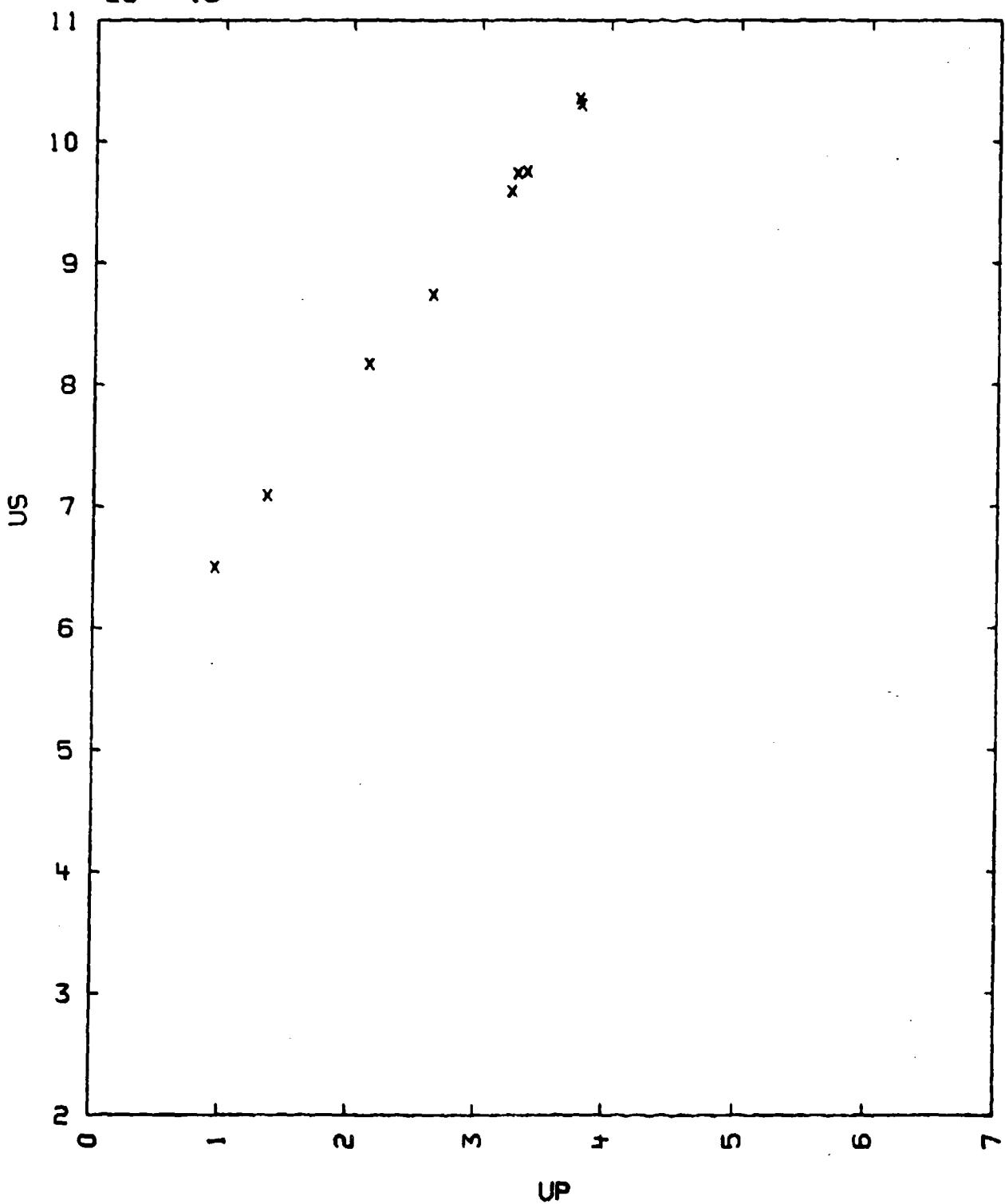
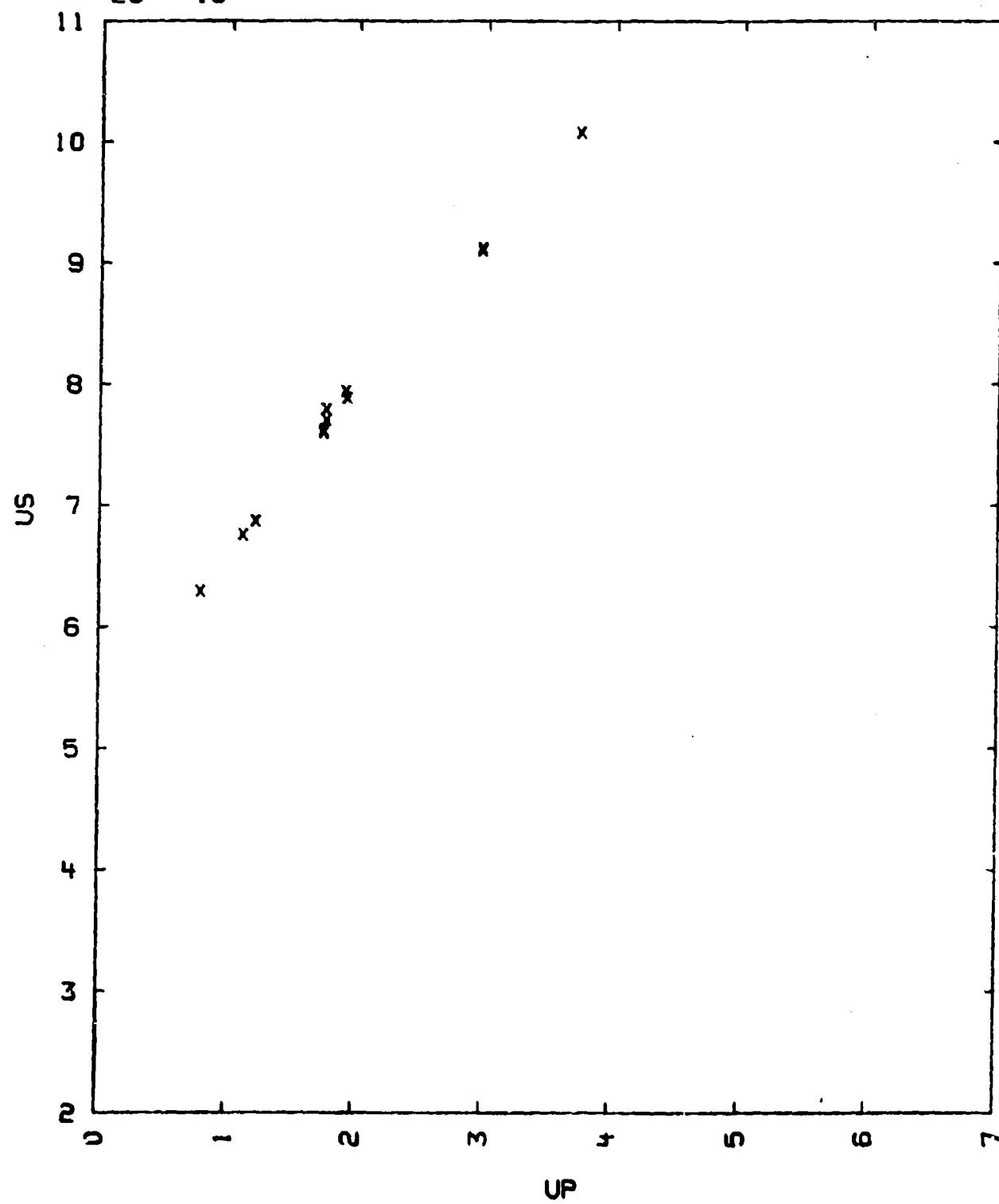


TABLE V

921T ALUMINUM
29---16



31---1

INDIUM

IN 99.8 PERCENT OR GREATER

$$V_0 = 0.1378 \text{ CC/G}$$

$$V_{01} = 0.1346 \text{ CC/G}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

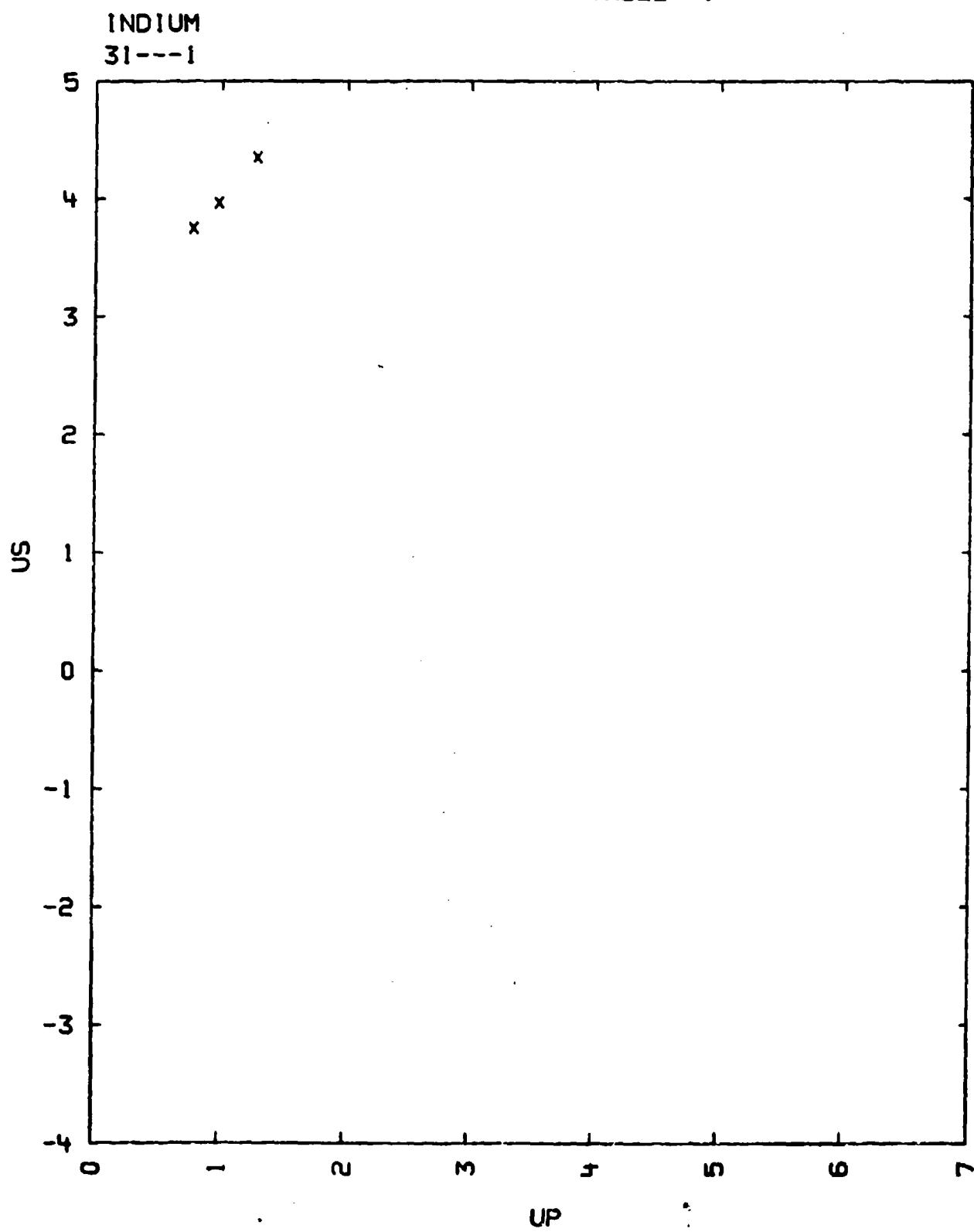
RHO0	US	UP	P	V/V0
7.27	3.745	0.7837	213.5	0.7907
-	3.965	0.9812	283	0.7525
-	4.348	1.281	.405	0.7054

$$US = 2.783 + 1.218 UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.4 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS.
- 4) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 3 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.

TABLE I



31--2

INDIUM

IN 99.5 PERCENT OR GREATER

$$V_0 = 0.1374 \text{ CC/G.}$$

$$V_{01} = 0.1346 \text{ CC/G.}$$

$$C_0 = 2.39 \text{ KM/SEC.}$$

$$C_B = 2.33 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITIES IN G/CC.

TABLE

	SAMPLE					STANDARD	
RHO 0	US	UP	P	V/V0	RHO0	US	
7.28	3.785	0.880	242.5	0.7675	8.934	5.040	
-	4.214	1.176	360.8	0.7209	-	5.426	
-	4.352	1.274	403.8	0.7073	-	5.556	
-	5.346	1.881	732.1	0.6481	-	6.387	
-	5.397	1.886	741.0	0.6506	-	6.400	
-	5.367	1.827	752.9	0.6410	-	6.443	
-	6.172	2.420	1087.4	0.6079	-	7.133	
-	6.526	2.875	1270.9	0.5901	-	7.482	
-	6.945	2.988	1510.7	0.5698	-	7.910	

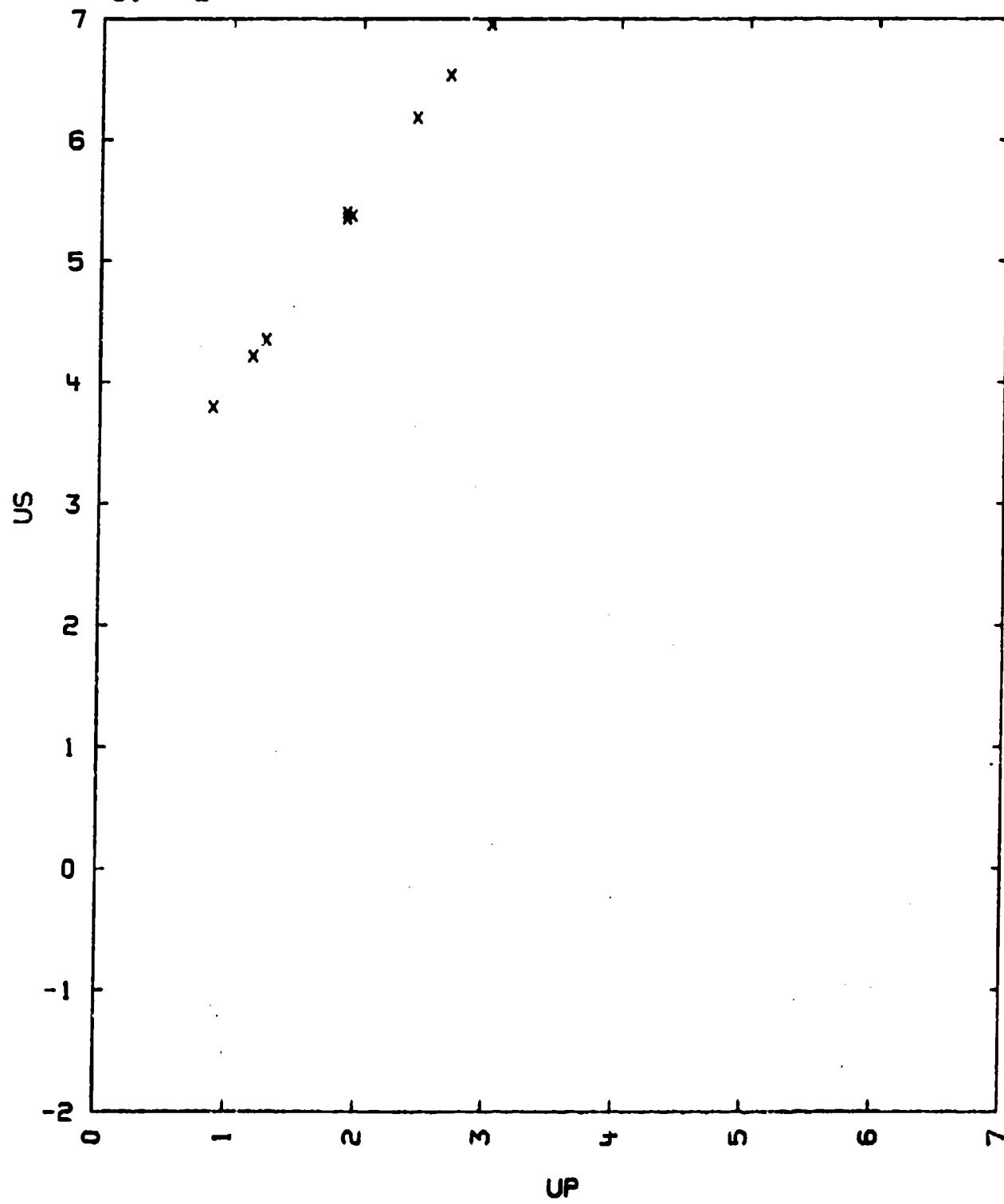
$$US = 2.388 + 1.552 \cdot UP \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
REPORT NO. GMX-5-568, PP. 51-62 (1964)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS COPPER
- 3) THE COPPER HUGONIOT IS REPRESENTED BY
US = 3.958 + 1.497 UP KM/SEC.
THE PARTIAL $(\partial E / \partial P) V = 0.0562$ CUBIC CM.
- 4) VO1 WAS TAKEN FROM THE AM. INST. OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO., N.Y., 1963) 2ND ED.

TABLE I

INDIUM
31---2



32---1
THALLIUM

TL 99.8 PER CENT OR GREATER

$$V_0 = 0.08446 \text{ CC/G.}$$

$$V_{01} = 0.08547 \text{ CC/G.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	VIVO
11.84	2.804	0.6416	213	0.7712
-	2.817	0.6388	213	0.7733
-	3.120	0.8446	312	0.7293
-	3.145	0.8406	313	0.7327
-	3.538	1.090	456.5	0.6919
-	3.541	1.089	456.5	0.6925

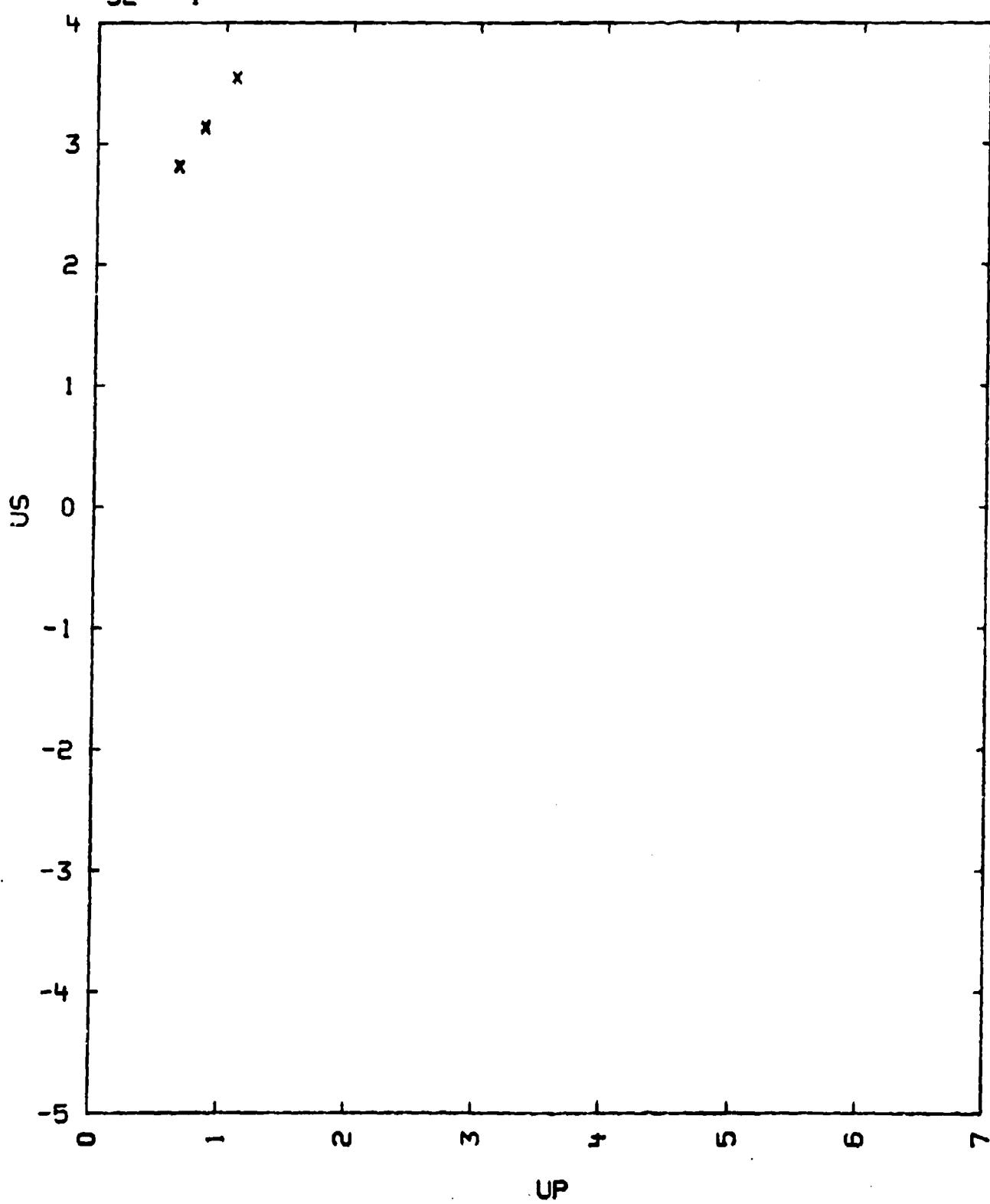
$$US = 1.770 + 1.623 UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.4 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J.M., RICE, M.H., MCQUEEN, R.G. AND YARGER, F.L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 3 PERCENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS

TABLE I

THALLIUM
32---1



32---2
THALLIUM

TL

$$V_0 = 0.0845 \text{ CC/G.}$$

$$V_{01} = 0.0844 \text{ CC/G}$$

$$C_0 = 1.83 \text{ KM/S}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
11.84	4.42	1.65	864	0.627	860
-	4.41	1.65	862	0.626	860
-	4.41	1.65	862	0.626	860
-	5.13	2.15	1306	0.581	1270
-	5.39	2.37	1515	0.560	1478
-	5.40	2.37	1516	0.561	1478
-	5.40	2.37	1517	0.561	1478

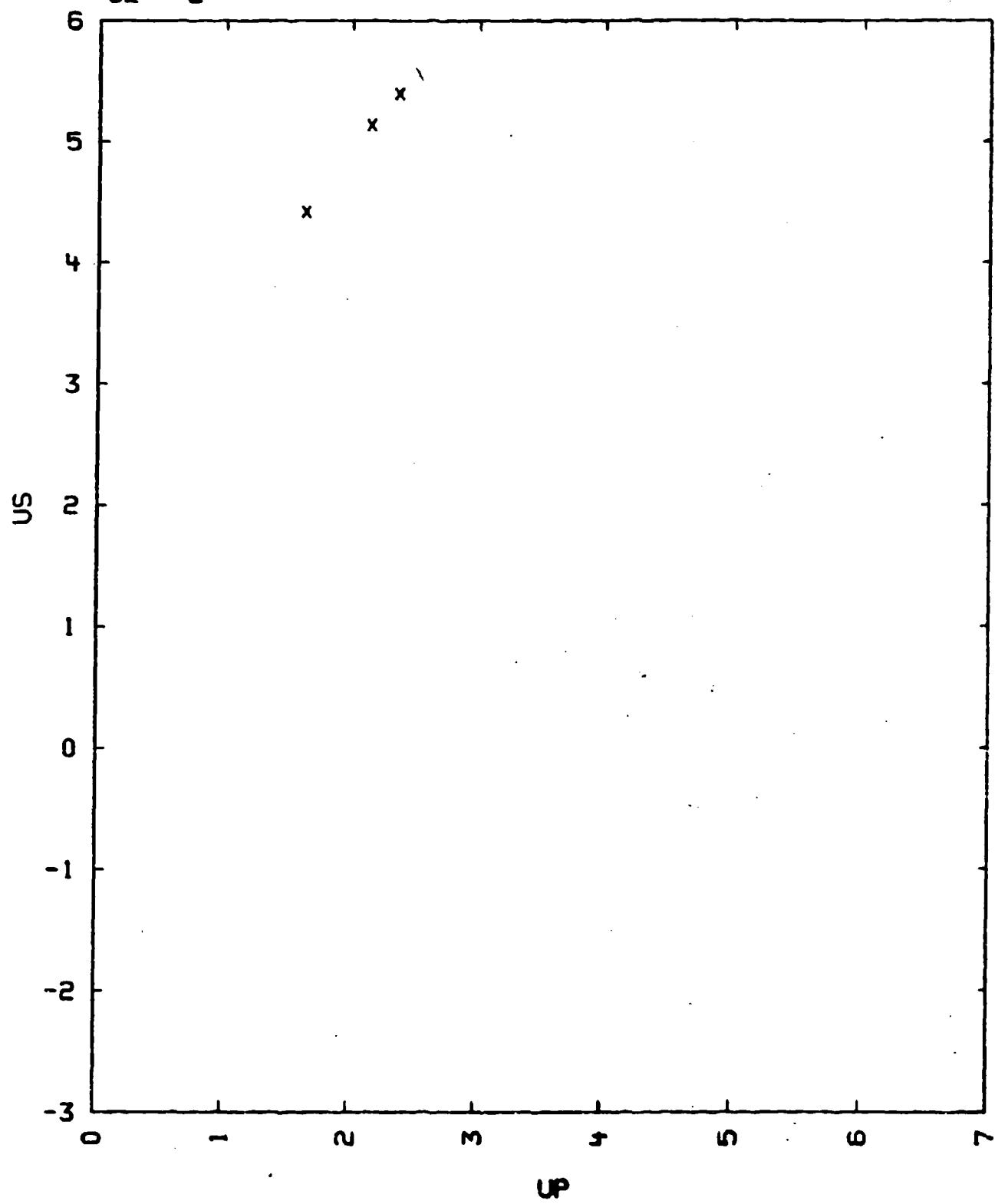
$$US = 1.859 + 1.515 UP \text{ KM/SEC.} \quad \text{SIGMA US} = 0.63 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 6) V01 WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW HILL BOOK CO.)
2ND ED.

TABLE I

THALLIUM
32---2



33---0
ZINC. SUMMARY

ZN

$V_0 = 0.1402 \text{ CC/G}$
 $V_{01} = 0.1401 \text{ CC/G}$

$C_B = 3.03 \text{ KM/SEC}$

THE TABLE LISTS HUGONIOT POINTS CALCULATED FROM THE LEAST SQUARE FIT GIVEN BELOW. DENSITY IS IN G/CC, PRESSURE IN KBAR, VELOCITY IN KM/SEC AND ENERGY DIFFERENCE IN KBAR.CC/G.

TABLE

RHO0	US	UP	P	V/V0	E-E0
7.135	3.963	0.6	170.	0.843	1.80
-	4.606	1.0	329.	0.783	5.00
-	5.240	1.4	523.	0.733	9.80
-	6.175	2.0	881.	0.676	20.0
-	7.693	3.0	1647.	0.610	45.0
-	9.158	4.0	2614.	0.563	80.0
-	11.932	6.0	5109.	0.497	180.
-	14.497	8.0	8275.	0.448	320.

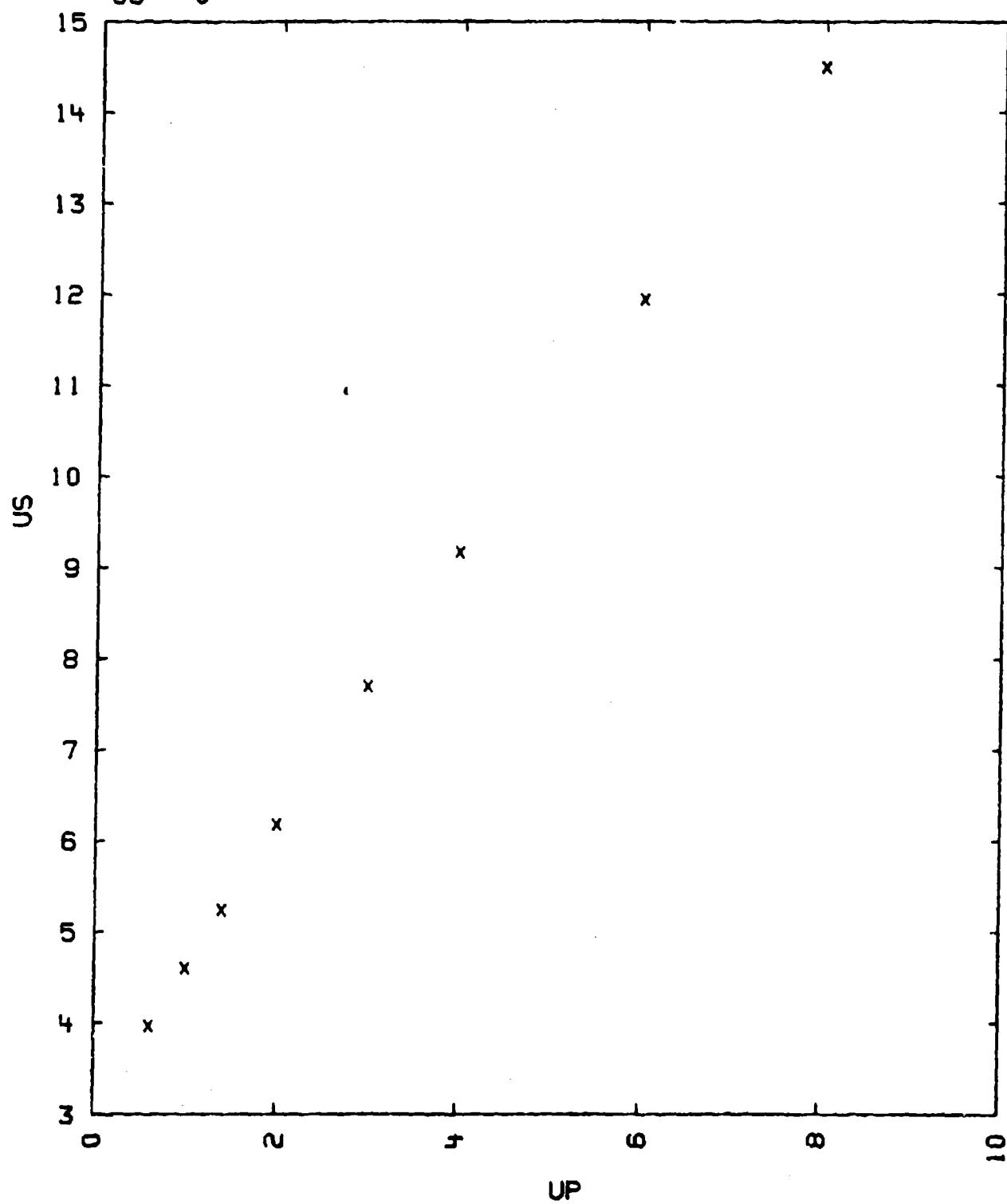
$US = 2.984 + 1.648 \cdot UP - 0.0261 \cdot UP^{1.2}$, SIG.US = 0.11 KM/SEC.
FOR UP BETWEEN 0.6 AND 8. KM/SEC.

COMMENTS:

- 1) SOURCE: COMPILER
DATA FROM 33---1,2,3 AND 4 WERE USED FOR THE LEAST SQUARE FIT
- 2) IF 33---3 TABLE I DATA ARE EXCLUDED THE DATA SUGGEST THAT THE BEST FIT IS OBTAINED FROM TWO STRAIGHT LINES. ADDITION OF TABLE I DATA INDICATES THAT THE UNCERTAINTY AT PRESSURES ABOVE 2 MBAR IS TOO LARGE TO RESOLVE THAT QUESTION.
- 3) V_{01} WAS CALCULATED FROM LATTICE CONSTANTS GIVEN BY WYCKOFF, CRYSTAL STRUCTURES VOL. I (INTERSCIENCE PUBL., N. Y. 1963).

TABLE I

ZINC, SUMMARY
33---0



33---1
ZINC

ZN 99.8 OR GREATER

V0 = 0.1402 CC/G
V0I = 0.1401 CC/G.IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

RHO0	US	UFS	UP	P	V/V0
7.135	5.014	2.589	1.250	447	0.7507
-	4.870		1.190	414	0.7555
-	4.481	1.795	0.88	281.4	0.8036
-	4.450	1.826	0.894	283.9	0.7991
-	4.053	1.355	0.650	188.0	0.8396
-	4.13		0.673	198.3	0.8370
-	4.022	1.295	0.630	180.8	0.8434

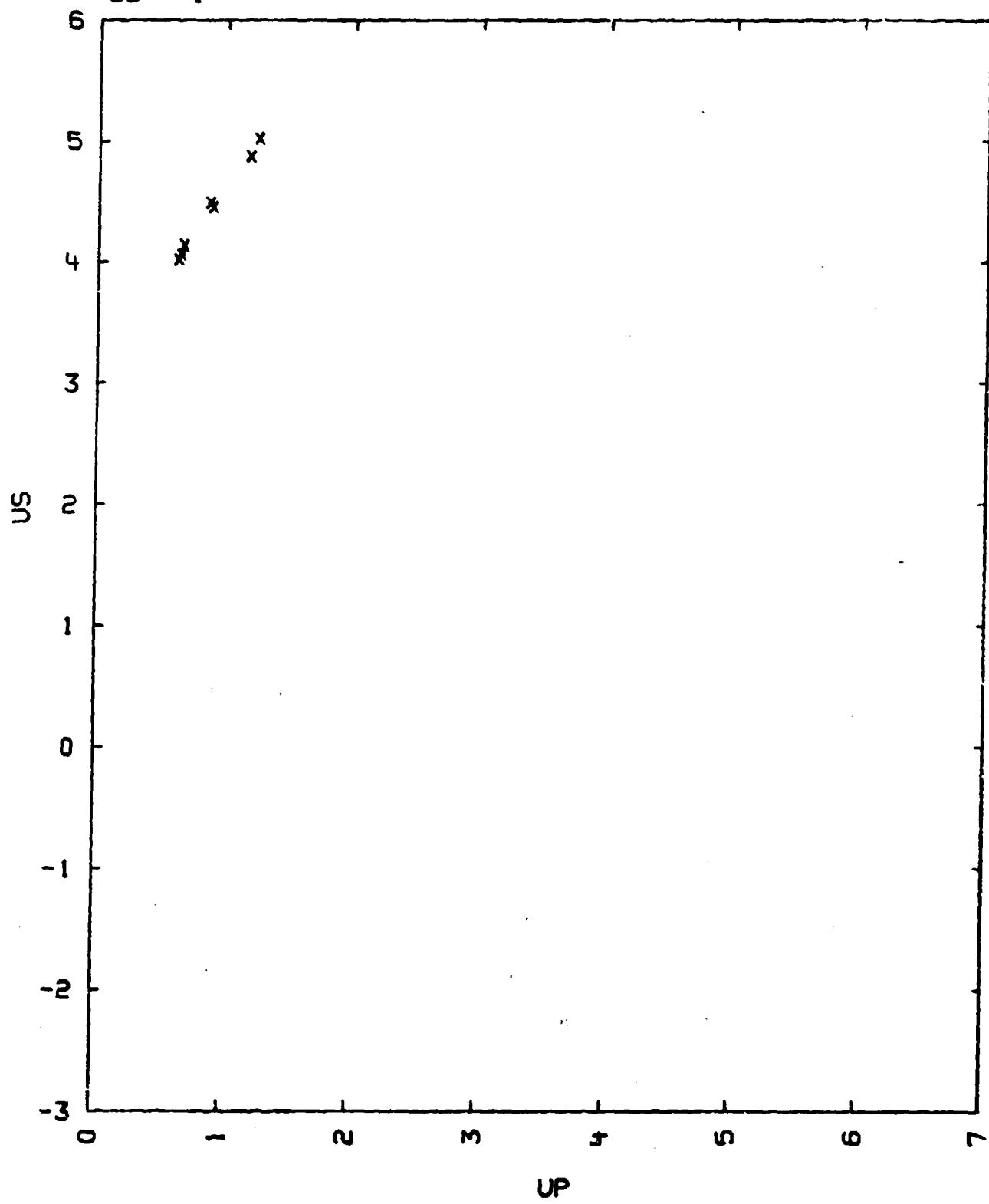
US = 3.073 + 1.542 UP KM/SEC SIGMA US = 0.8 PERCENT

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 109, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE TOTAL SPREAD IN SHOCK VELOCITIES AT HIGH PRESSURES WAS 1 PER CENT
AND THE SPREAD IN SHOCK VELOCITIES FOR PRESSURES BELOW 200 KILOBARS
WAS 1.5 PERCENT.
- 4) THREE ZINC CRYSTALS OF THE FOLLOWING ORIENTATION WERE USED ON EACH
SHOT:
 - A. THE SHOCK PROPAGATION DIRECTION ALONG THE C AXIS
 - B. THE SHOCK PROPAGATION DIRECTION ALONG THE A AXIS
 - C. THE SHOCK PROPAGATION DIRECTION BETWEEN TWO A AXES
- 5) WITHIN EXPERIMENTAL ERROR, THE RESULTS SHOW NO DEPENDENCE OF SHOCK
VELOCITY UPON CRYSTALLOGRAPHIC ORIENTATION.

TABLE I

ZINC
33---1



33---2
ZINC

ZN

$$V_0 = 0.1401 \text{ CC/G}$$

$$V_{01} = 0.1382 \text{ CC/G.}$$

$$C_B = 3.03 \text{ KM/SEC}$$

IN THE TABLE BELOW VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
7.14	5.81	1.80	745	0.690	853
-	5.82	1.80	747	0.691	853
-	5.78	1.80	743	0.688	853
-	7.22	2.71	1394	0.625	1563
-	7.31	2.71	1416	0.631	1573
-	7.30	2.69	1403	0.631	1560

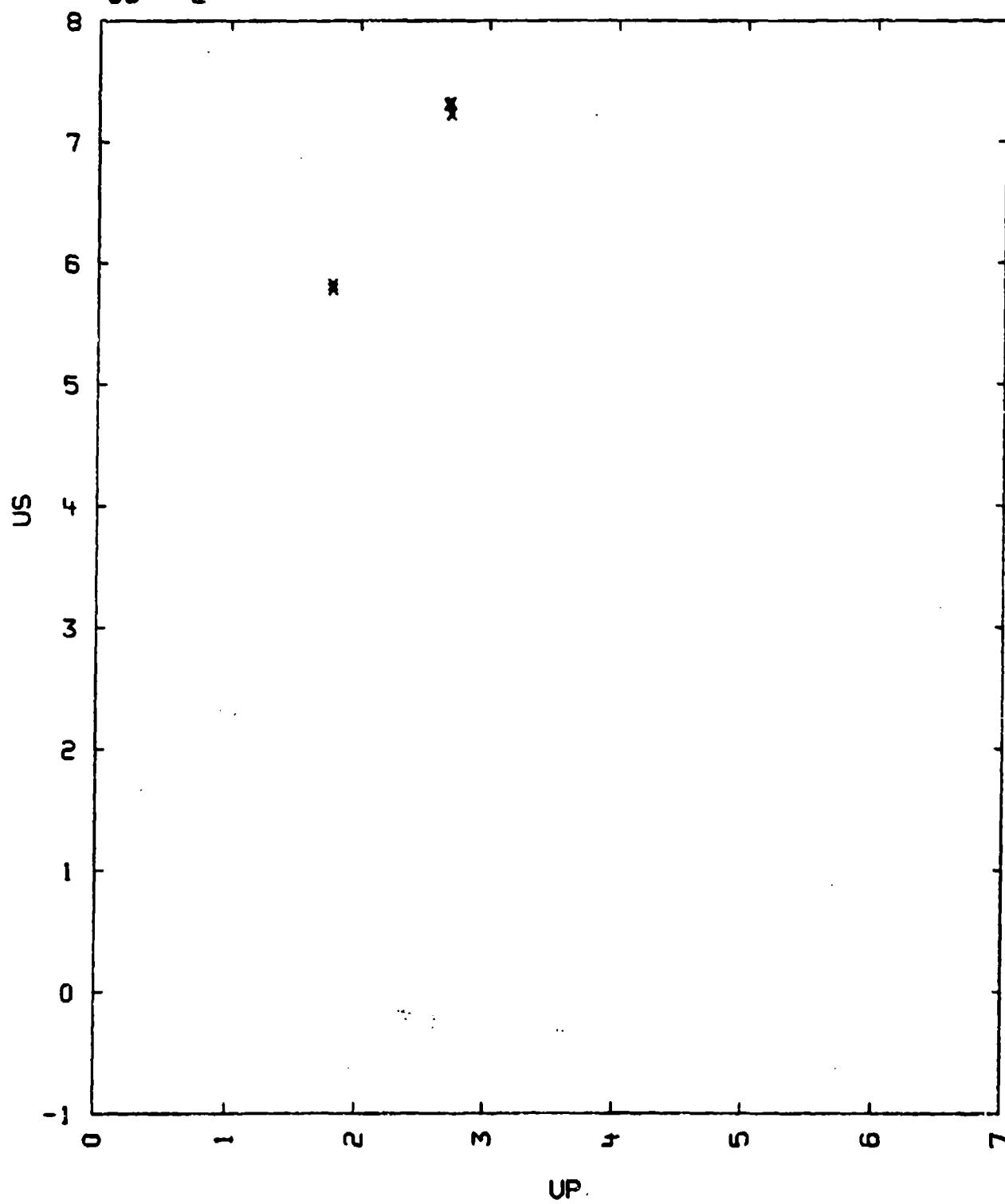
$$US = 3.050 + 1.559 UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.45 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R. G., AND MARSH, S. P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL USED IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
2ND ED.
- 6) V01 WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW HILL BOOK CO.)

TABLE I

ZINC
33---2



33---3
ZINC

ZN

$$V_0 = 0.1401 \text{ CC/G.}$$

$$V_{01} = 0.1392 \text{ CC/G.}$$

$$C_0 = 2.851 \text{ KM/SEC}$$

IN THE TABLES BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL USED AS SAMPLE HOLDER IN TABLE I OR AS STRIKER (PROJECTILE PLATE) IN TABLE II. IN TABLE II U(ST) REFERS TO THE TOTAL STRIKER VELOCITY BEFORE IMPACT.

TABLE I

----- SAMPLE -----					-- HOLDER --		
RHO0	US	UP	P	V/V0	ST	US(ST)	UP(ST)
7.14	4.70	1.04	350	0.7813	FE	5.30	0.95
-	6.85	2.41	1179	0.6481	FE	7.53	2.24
-	9.90	4.54	3209	0.5414	FE	10.67*	4.26*

US =

TABLE II

----- SAMPLE -----					STRIKER	
RHO0	US	UP	P	V/V0	ST	U(ST)
7.14	6.30	2.12	954	0.6636	AL	5.60
-	7.92	2.98	1685	0.6234	FE	5.60
-	10.39	4.83	3596	0.5328	FE	9.10
-	14.19	7.78	7860	0.4517	FE	14.68

$$US = 3.449 + 1.434 \cdot UP \quad \text{OR} \quad US = 2.976 + 1.687 \cdot UP - 0.032 \cdot UP^{1.2} \text{ KM/SEC}$$

$$\text{SIG US} = 0.20 \quad \text{SIG US} = 0.12$$

COMMENTS:

- 1) SOURCE: AL'TSHULER, L.V., BAKANOVA, A.A. AND TRUNIN, R.F.
SOVIET PHYS.-JETP, VOL. 15, P. 65 (1962)
J. EXPTL. THEORET. PHYS. (USSR) VOL. 42, P. 91, (1962)
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) TABLE I IS CORRECTED DATA WHICH FIRST APPEARED IN THE SOVIET PHYS.-JETP, VOL. 34, P. 814 (1959). THE * VALUES WERE CALCULATED FROM THE IRON STRIKER VELOCITY OF 8.64 KM/SEC.
- 4) THE INACCURACY IN THE DETERMINATION OF US IN TABLE II TO PRESSURES OF 4187 KILOBARS DID NOT EXCEED 1.0 PERCENT. EACH POINT IS THE AVERAGE OF 4 TO 6 EXPERIMENTS.
- 5) THE UNCERTAINTY OF THE MEASURED VALUES IN THE LAST ROW ARE:

- 1 PERCENT IN US, OBTAINED FROM 6-8 EXPERIMENTS AND 1.5 PERCENT IN
U(SI) OBTAINED FROM 12 EXPERIMENTS
- 6) VOI WAS CALCULATED FROM THE HEXAGONAL UNIT CELL WITH
 $A = 2.8990$ AND $C = 4.9351$ ANGSTROM
- AMERICAN INSTITUTE OF PHYSICS HANDBOOK (MCGRAW-HILL 1963) 2ND ED.
- 7) CO IS GIVEN BY: K. M. SCHRAMM, Z. METALKUNDE, VOL. 53, P. 729 (1982)

TABLE I

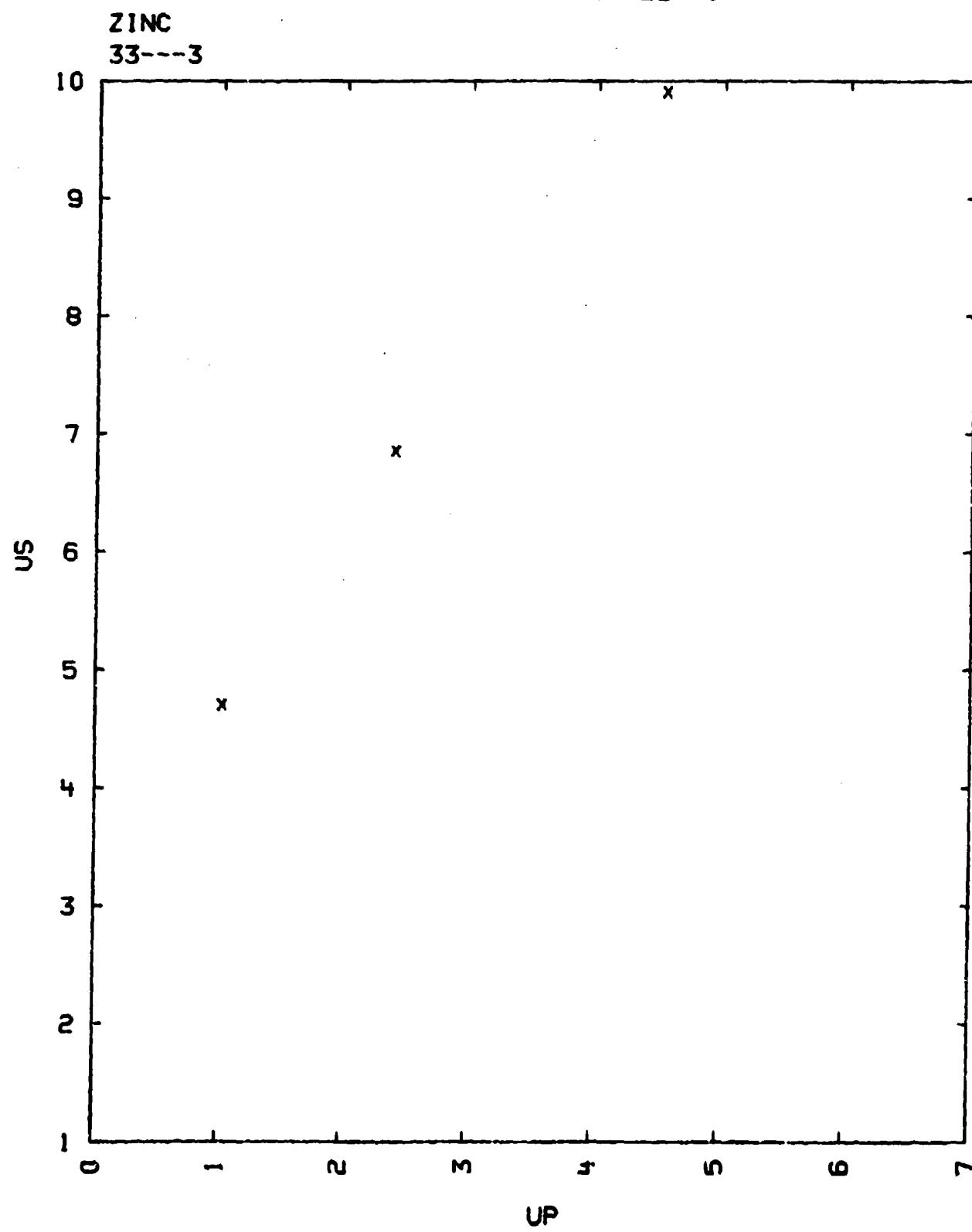
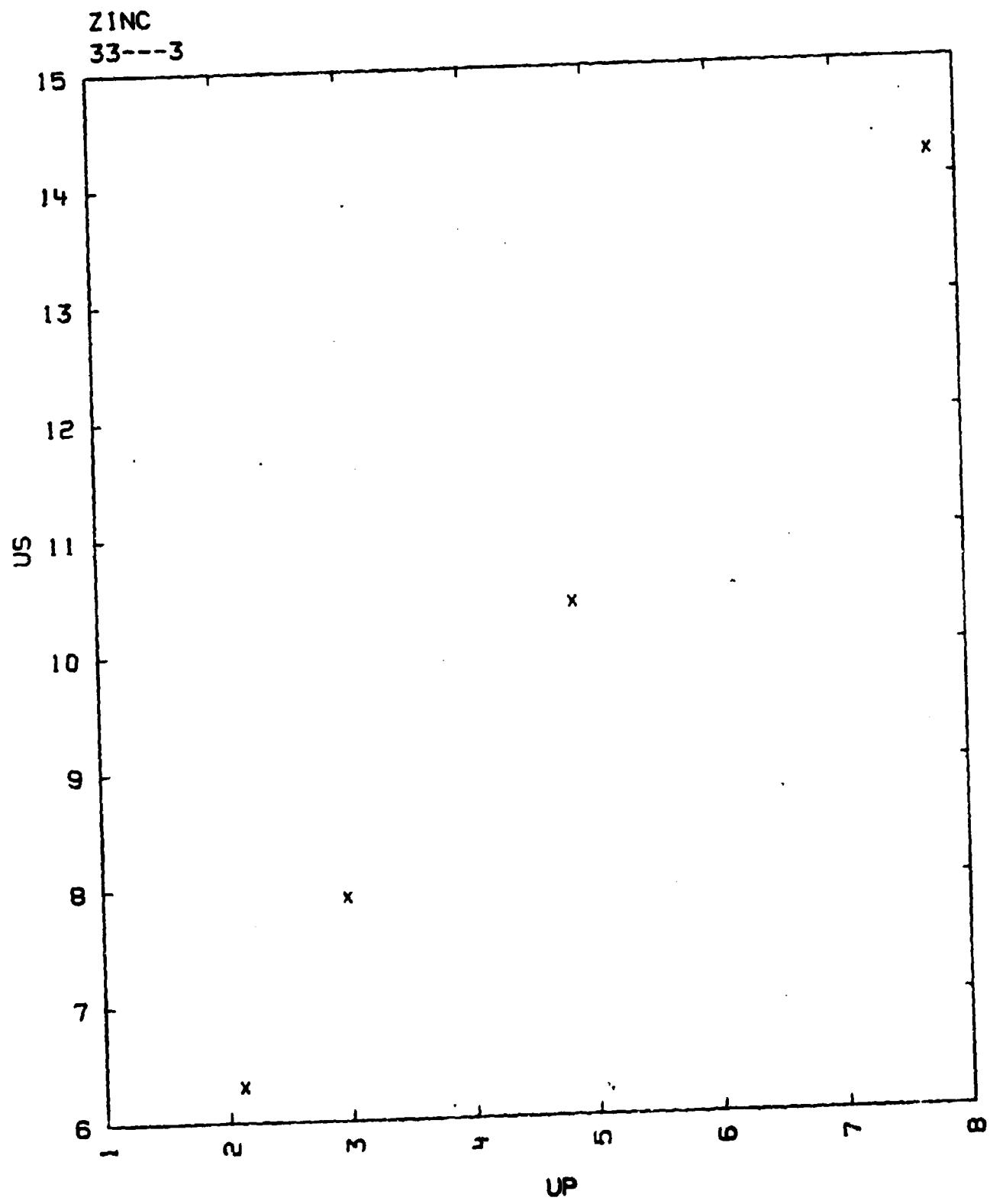


TABLE II



33---4
ZINC

ZN

$$V_0 = 0.1403 \text{ CC/G}$$

$$V_{01} = 0.1402 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC., VELOCITIES IN KM/SEC., AND PRESSURE IN KILOBARS.

TABLE

RHOD	SAMPLE				BASE PLATE	
	US	UFS	UP	P	V/V0	PRESSURE
7.118	4.51	1.83	0.89	286	0.802	224
7.121	4.45		0.91	287	0.797	224
7.129	4.81	2.26	1.13	387	0.765	291
7.138	4.69	2.26	1.13	378	0.760	281

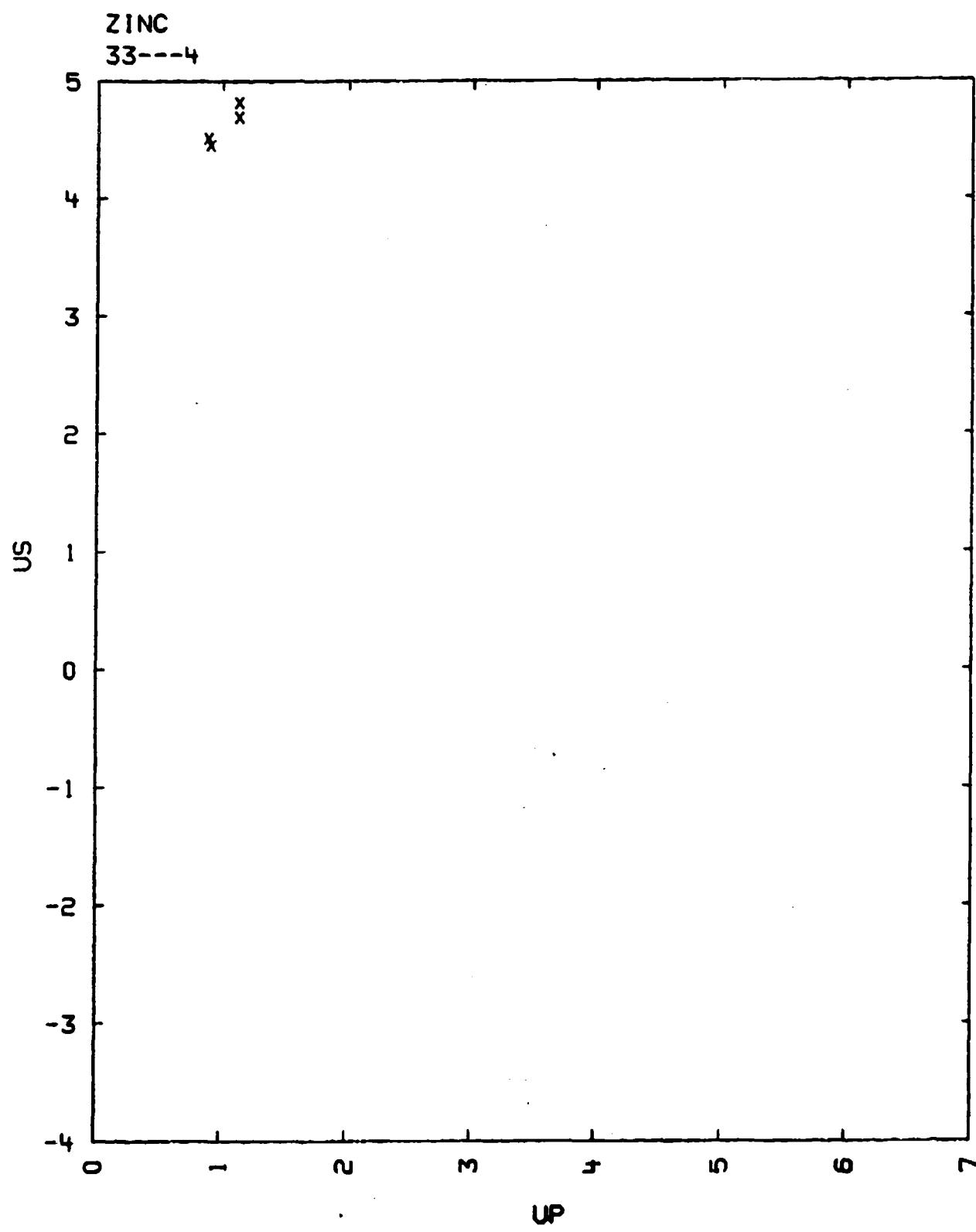
$$US = 3.44 + 1.16 UP \text{ KM/SEC}$$

$$\text{SIGMA US} = 0.073 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: COMPILER
L.R.L. EQUATION OF STATE FILE
LAURENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B. (ALUMINUM STANDARD BASE PLATE)
DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF VO1 WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES
(DOVER PUBLICATIONS, INC., NEW YORK, N.Y., 1963).

TABLE I



34---0
CADMIUM, SUMMARY

CD

$$V_0 = 0.1157 \text{ CC/G}$$

$$V_{01} = 0.1157 \text{ CC/G}$$

THE TABLE LISTS THE HUGONIOT POINTS CALCULATED FROM THE LEAST SQUARE FIT GIVEN BELOW. UNITS ARE: G/CC., KM/SEC., KBAR AND KBAR.CC/G.

TABLE

FIT	RHO0	US	UP	P	V/V0	E~E0
1	8.64	3.461	0.6	179	0.827	1.8
1	-	4.124	1.0	356	0.758	5.0
1	-	4.952	1.5	642	0.697	11.25
1	-	5.781	2.0	999	0.654	20.0
1	-	6.609	2.5	1428	0.622	31.25
1	-	7.438	3.0	1928	0.597	45.0
2	-	8.713	4.0	3011	0.541	80.0
2	-	11.165	6.0	5788	0.463	180.0
2	-	13.617	8.0	9412	0.412	320.

$$US = 2.467 + 1.657 \cdot UP, \quad SIG.US = 0.06 \text{ KM/SEC}$$

FOR UP BETWEEN 0.6 AND 3.0 KM/SEC

$$US = 3.809 + 1.226 \cdot UP, \quad SIG.UP = 0.04 \text{ KM/SEC}$$

FOR UP BETWEEN 4 AND 8

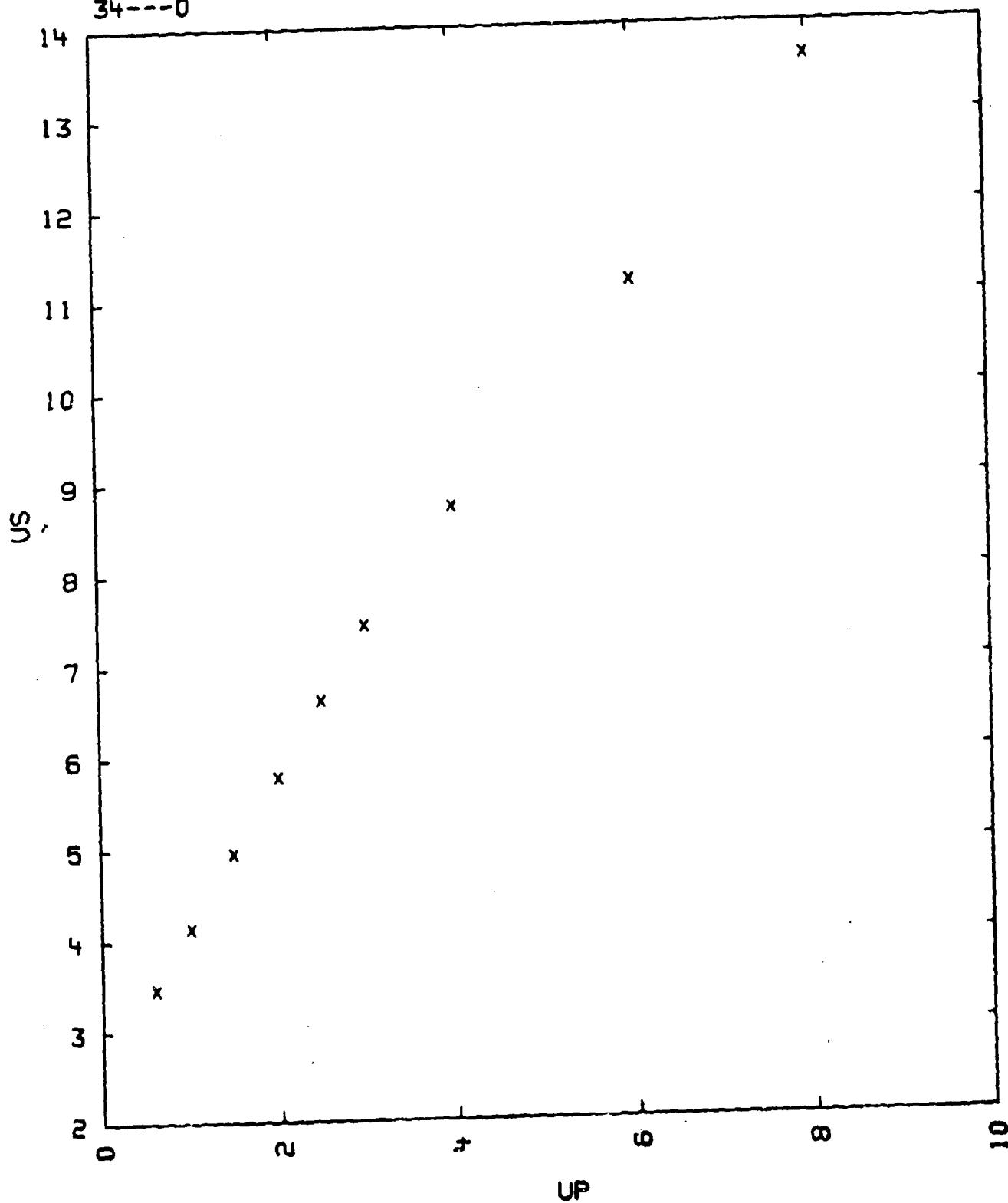
COMMENTS

- 1) SOURCE: COMPILER
- 1A) DATA FROM 34---1.2 AND 3 WERE USED FOR THIS FIT.
- 2) THE ACTUAL SHAPE OF THE HUGONIOT CURVE IS UNCERTAIN BETWEEN UP = 2.9 AND 4.3 KM/SEC. THE HIGHEST POINT ON THE FIRST FIT AND THE LOWEST EXPERIMENTAL POINT ON THE SECOND FIT RESPECTIVELY.

TABLE I

CADMUM, SUMMARY

34---0



J4---1
CADMIUM

CD 99.8 PER CENT OR GREATER

VO = 0.1157 CC/G. CL = 2.78 KM/SEC. CD = 2.17 KM/SEC.

VOI = 0.1168 CC/G. CS = 1.50 KM/SEC.

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

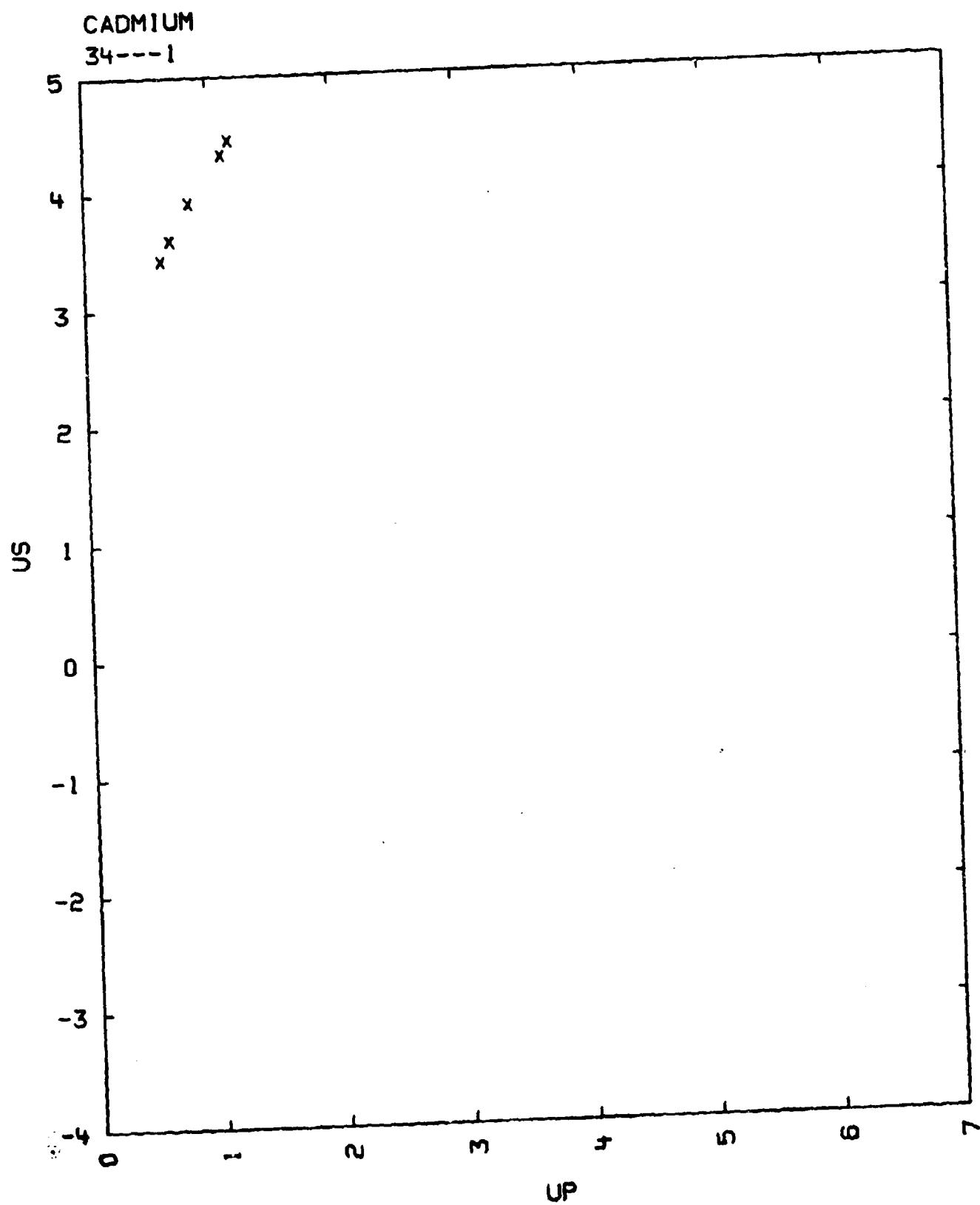
RHO0	US	UFS	UP	P	V/V0
8.64	3.599	1.464	0.690	214.4	0.8083
-	3.421	1.277	0.619	182.9	0.8191
-	3.918	1.757	0.850	287.6	0.7830
-	4.450	2.496	1.190	457.3	0.7326
-	4.324	2.400	1.120	418.2	0.7410

US = $2.379 + 1.749 \cdot UP$ KM/SEC SIGMA US = 1.0 PERCENT

COMMENTS

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
STANDARD MATERIAL 24ST ALUMINUM
- 3) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS.
- 4) CL AND CS WERE OBTAINED FROM L. BERGMAN, DER ULTRASCHALL,
S. HIRZEL VERLAG, STUTTGART, 1954, 6TH ED., P. 650
- 5) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.

TABLE I



34---2
CADMIUM

CD

$$V_0 = 0.116 \text{ CC/G}$$

$$V_{01} = 0.1157 \text{ CC/G}$$

$$C_B = 2.41 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
8.64	5.66	1.96	957	0.654	1022
-	5.78	1.96	976	0.662	1031
-	5.77	1.96	980	0.663	1034
-	5.77	1.97	982	0.668	1043
-	5.76	1.98	985	0.657	1045
-	6.48	2.40	1339	0.628	1403
-	6.48	2.40	1345	0.629	1406
-	6.39	2.43	1339	0.620	1410
-	6.43	2.43	1351	0.620	1421

$$US = 2.443 + 1.671 UP \text{ KM/SEC.}$$

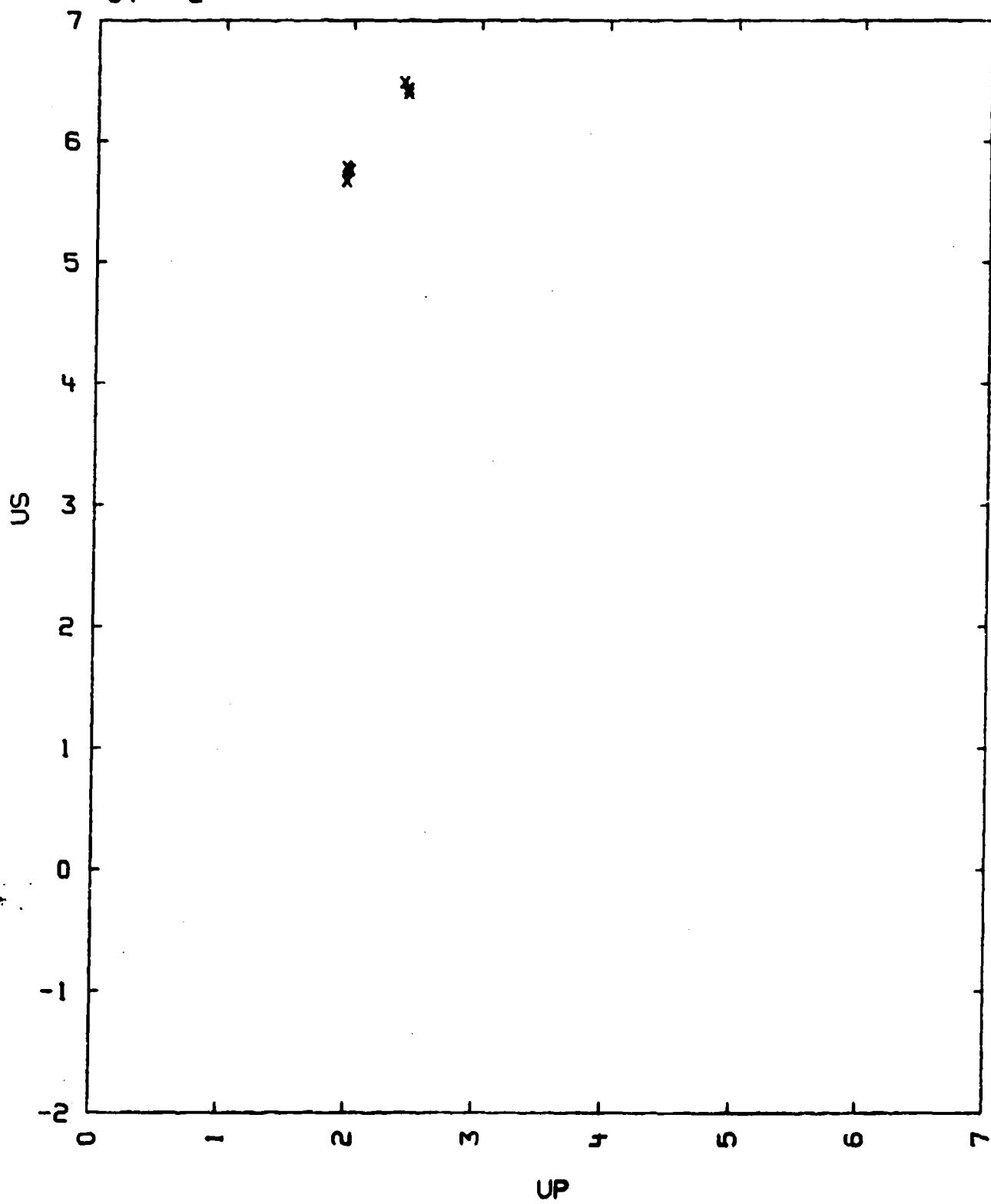
$$\text{SIGMA US} = 0.58 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 4) V01 WAS TAKEN FROM THE AM. INST. OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO., N.Y., 1963) 2ND ED.

TABLE I

CADMIUM
34---2



34--3
CADMIUM

CD

$$V_0 = 0.1157 \text{ CC/G.}$$

$$V_{01} = 0.1157 \text{ CC/G.}$$

$$C_0 = 2.245 \text{ KM/SEC.}$$

IN THE TABLES BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL USED AS SAMPLE HOLDER IN TABLE I OR AS STRIKER (PROJECTILE PLATE) IN TABLE II. U(ST) REFERS TO TOTAL STRIKER VELOCITY BEFORE IMPACT.

TABLE I

SAMPLE					HOLDER		
RHO ₀	US	UP	P	V/V ₀	ST	US(ST)	UP(ST)
8.64	4.10	1.02	360	0.7519	FE	5.30	0.95
-	6.32	2.31	1261	0.6345	FE	7.53	2.24
-	9.14	4.36	3443	0.5230	FE	10.67	4.26

US = SEE BELOW

TABLE II

SAMPLE					STRIKER	
RHO ₀	US	UP	P	V/V ₀	ST	U(ST)
8.64	5.88	1.99	1010	0.6614	AL	5.60
-	7.16	2.88	1788	0.5974	FE	5.60
-	9.55	4.67	3850	0.5107	FE	9.10
-	12.99	7.49	8410	0.4234	FE	14.68

$$US = 3.112 + 1.349 \cdot UP \quad \text{OR} \quad US = 2.486 + 1.739 \cdot UP - 0.045 \cdot UP^2 \text{ KM/SEC}$$

$$\text{SIG US} = 0.24 \quad \text{SIG US} = 0.10$$

COMMENTS:

- 1) SOURCE: AL'TSHULER, L.V., BAKANOVA, A.A. AND TRUNIN, R.F.
SOVIET PHYS.-JETP, VOL. 15, P. 65 (1962)
J. EXPTL. THEORET. PHYS. (USSR) VOL. 42, P. 91, (1962)
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B
- 3) TABLE I IS CORRECTED DATA WHICH FIRST APPEARED IN THE SOVIET PHYS.-JETP, VOL. 34, P. 614 (1958). THE * VALUES WERE CALCULATED FROM THE IRON STRIKER VELOCITY OF 8.64 KM/SEC.
- 4) THE INACCURACY IN THE DETERMINATION OF US IN TABLE 2 TO PRESSURES OF 4187 KILOBARS DID NOT EXCEED 1.0 PERCENT. EACH POINT IS THE AVERAGE OF 4 TO 6 EXPERIMENTS.
- 5) THE U UNCERTAINTY OF THE VALUES IN THE LAST TABLE ENTRY ARE:

1 PERCENT IN US, OBTAINED FROM 6-8 EXPERIMENTS AND 1.5 PERCENT IN
U(ST) OBTAINED FROM 12 EXPERIMENTS.

- 6) VOL WAS OBTAINED FROM THE HEXAGONAL UNIT CELL, WITH:
 $A = 2.9796$ AND $C = 5.6171$ ANGSTROM
AMERICAN INST. OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO., N.Y., 1963)
2ND ED.
- 7) CO IS GIVEN BY: K. M. SCHRAMM, 2. METALKUNDE, VOL. 53, P. 729 (1962)

TABLE I

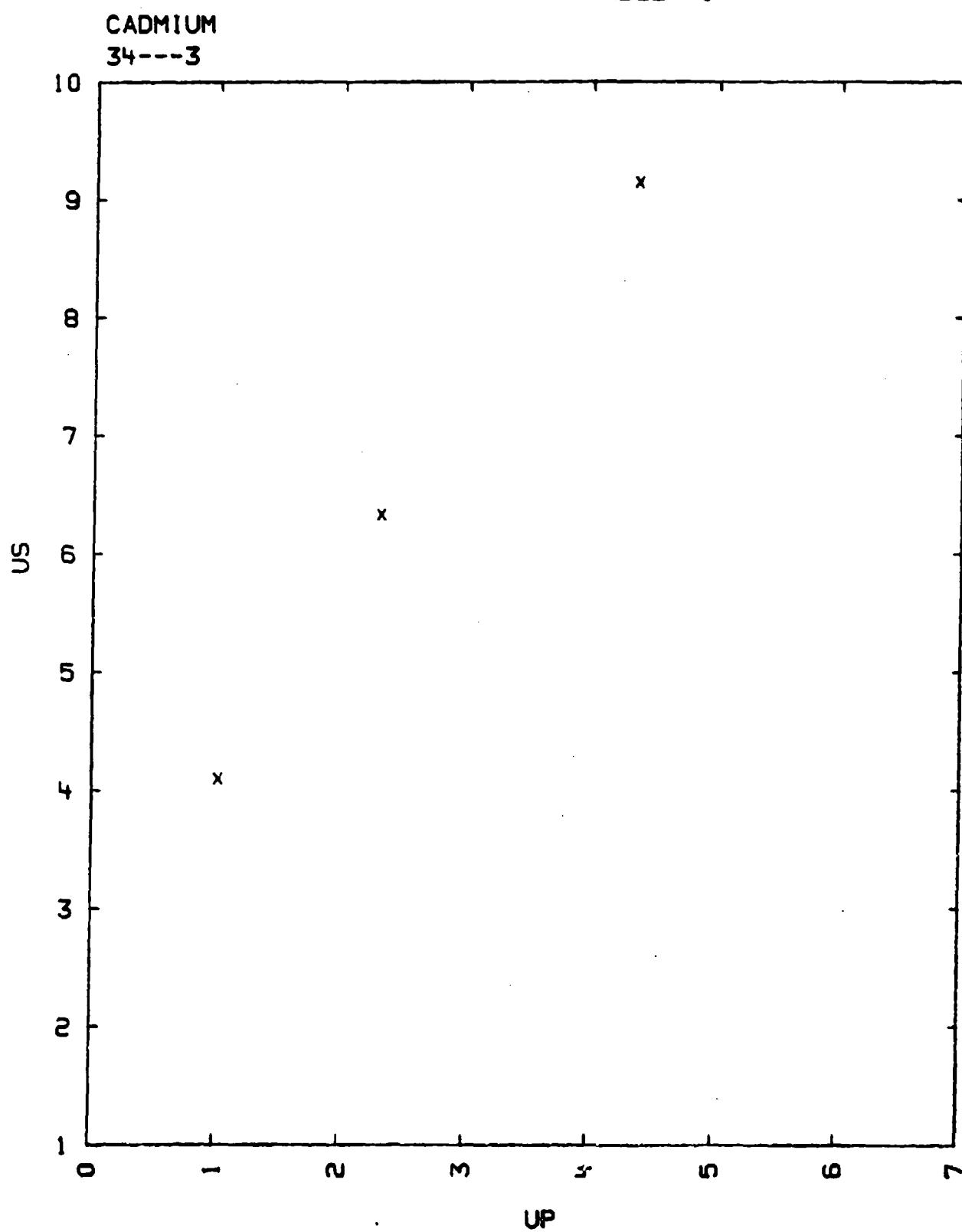
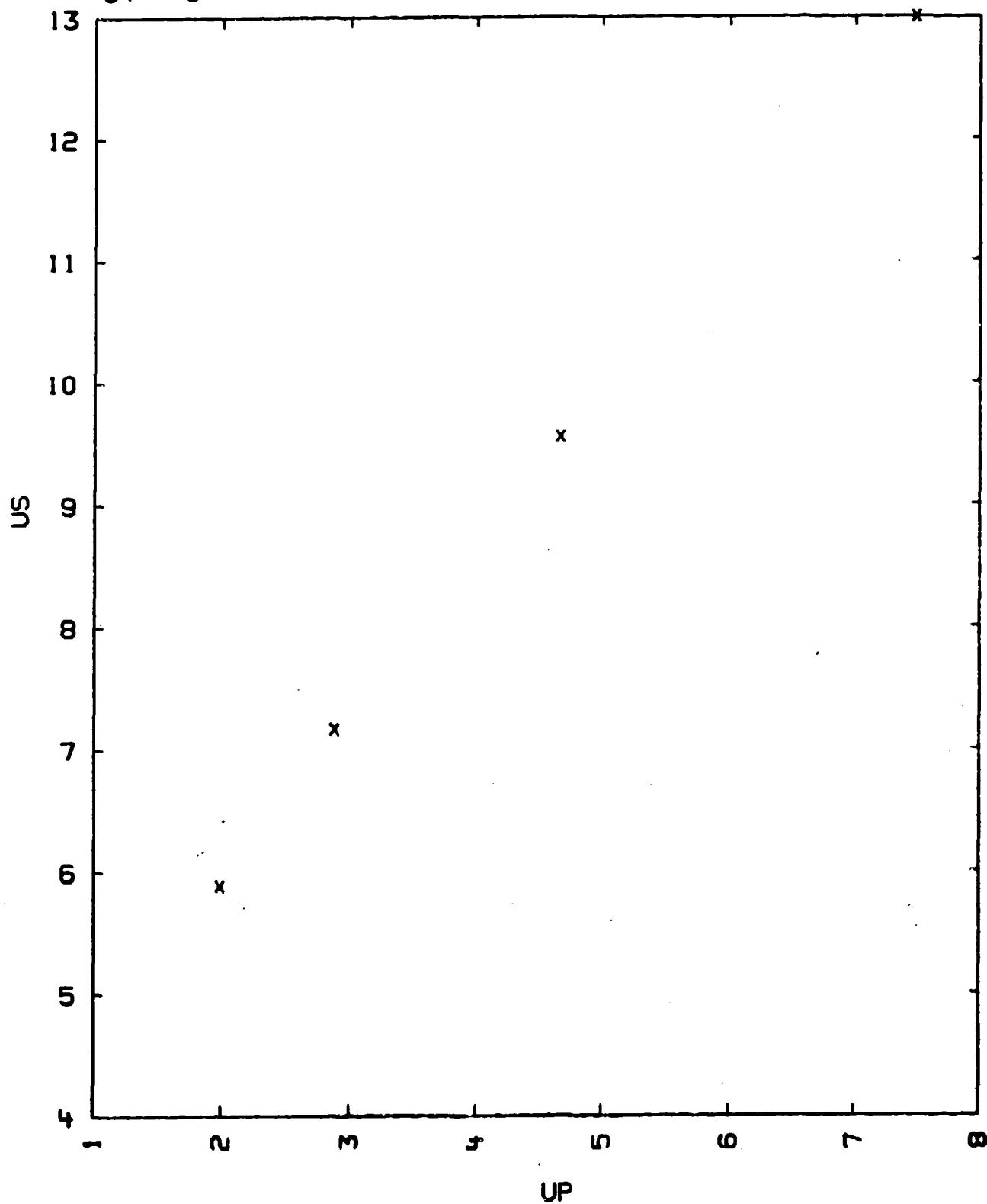


TABLE II

CADMUM
34---3



35---1
MERCURY

HG

$$T_0 = 17-25 \text{ DEG. CENTIGRADE}$$

$$V_0 = 0.0739 \text{ CC/G}$$

$$C_0 = 1.451 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS, DENSITY IN G/CC AND TEMPERATURE IN DEG. CENTIGRADE

TABLE

T ₀	RHO ₀	U _S	U _P	P	V/V ₀
25	13.53	2.752	0.608	226.4	0.779
17	-	3.101	0.772	324.0	0.751
24	-	3.504	0.978	463.7	0.721

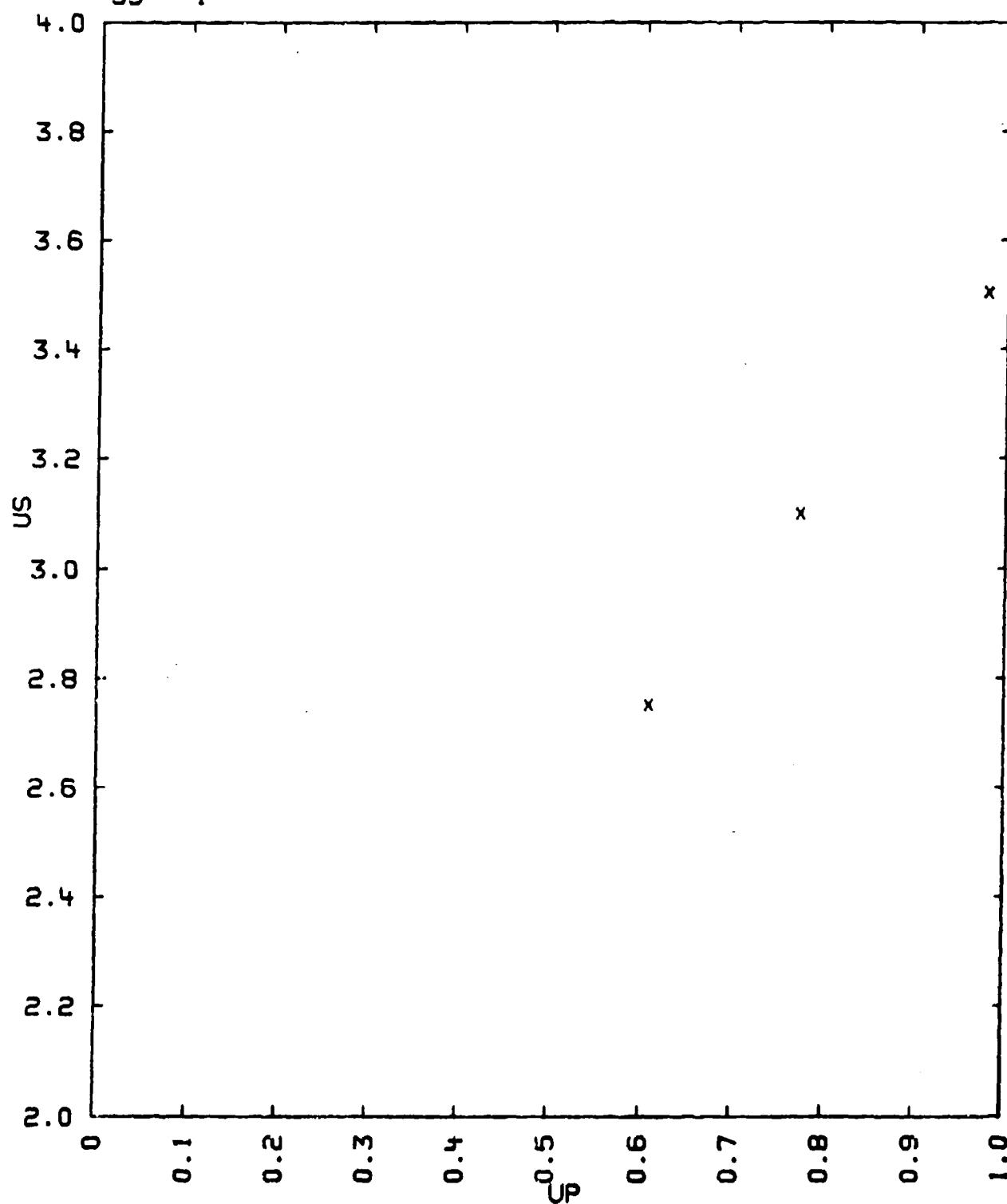
$$U_S = 1.524 + 2.029 U_P, \text{ SIG.U}_S = 0.013 \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: WALSH J. M. AND RICE M. H.
JOURNAL OF CHEMICAL PHYSICS, VOL. 26, P. 815 (1957)
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE VALUE FOR C₀ AT 20 DEG. CENTIGRADE WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK, (MCGRAW-HILL BOOK CO., N. Y. 1963)
2ND ED.

TABLE I

MERCURY
35---1



36---0
COPPER SUMMARY

CU

$$V_0 = 0.11120 - 0.448 \text{ CC/G}$$

$$V_{01} = 0.11119 \text{ CC/G}$$

$$C_0 = 3.92 \text{ KM/SEC.}$$

$$C_B = 3.98 \text{ KM/SEC.}$$

THE TABLE LISTS HUGONIOT POINTS CALCULATED FROM THE FIT GIVEN BELOW.
UNITS ARE G/CC., KM/SEC., KILOBARS AND KBAR-CC/G FOR THE ENERGY CHANGE.

TABLE

RH00	US	UP	P	V/V0	E-E0
8.93	4.740	0.5	212	0.8945	1.25
-	5.473	1.0	489	0.8173	5.
-	6.206	1.5	831	0.758	11.25
-	6.939	2.0	1239	0.712	20.0
-	7.672	2.5	1713	0.674	31.25
-	8.405	3.0	2252	0.643	45.0
-	9.871	4.0	3526	0.595	80.0
-	11.337	5.0	5062	0.559	125.
-	12.803	6.0	6860	0.531	180.
-	14.269	7.0	8920	0.509	245.
5.68	8.515	4.5	2176	0.471	101.
-	9.926	5.5	3101	0.446	151.
-	10.631	6.0	3623	0.436	180.
-	12.041	7.0	4788	0.419	245.
-	14.862	9.0	7398	0.394	405.
4.46	8.426	5.0	1879	0.406	125.
-	9.816	6.0	2626	0.389	180.
-	11.205	7.0	3498	0.375	245.
-	12.595	8.0	4493	0.365	320.
-	14.679	9.5	6219	0.353	451.
2.96	8.131	5.5	1324	0.324	151.
-	9.495	6.5	1827	0.315	211.
-	11.541	8.0	2733	0.307	320.
-	12.905	9.0	3438	0.302	405.
2.23	8.325	6.0	1114	0.279	180.
-	11.028	8.0	1987	0.274	320.
-	13.731	10.0	3062	0.272	500.
-	15.758	11.5	4041	0.270	861.

$$US = 4.007 - 0.5657 * (8.93 - RH00) + 1.466 * UP -$$

$$- 0.0171 * (8.93 - RH00) * UP$$

SIG.US = 0.14

FOR THE LIMITS GIVEN IN THE TABLE.

COMMENTS:

- 1) SOURCE: COMPILER
- 1A) THE DATA FROM 36---1,2,3,6 AND 7 WERE USED FOR THE ABOVE FIT.
- 2) THE POROUS DATA OF PAGE 36---6 AND THE DATA OF 36---2 WERE ASSIGNED A WEIGHT OF 1. THE REST A WEIGHT OF 2.
- 3) THE DATA OF 8 AGREES WITH THIS FIT.
- 4) VOI WAS OBTAINED FROM THE FACE CENTERED CUBIC CELL CONSTANT
 $A = 3.61496 \text{ ANGSTROM AT } 18 \text{ DEG. C.}$
GIVEN BY WYCKOFF, CRYSTAL STRUCTURES, VOL. 1, (INTERSCIENCE PUBL.
N. Y. 1963) 2ND. ED.
- 5) CO WAS CALCULATED FROM THE ELASTIC CONSTANTS OF
- 6) THE ONSET OF YIELDING OF SINGLE CRYSTALS OF COPPER WAS OBSERVED AT
SHOCK DIRECTION PI US1(CAL,C) US2

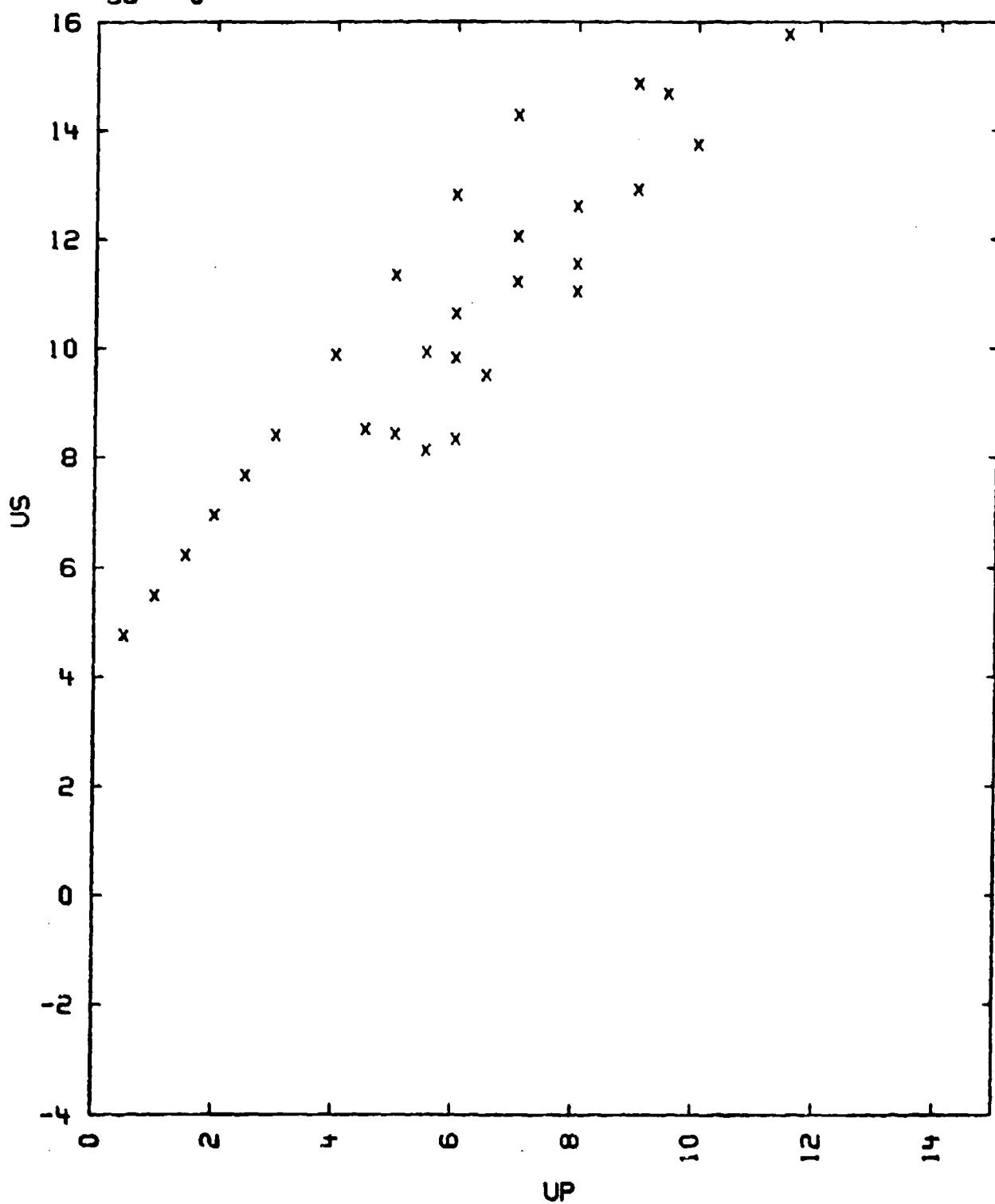
(100)	2.0	4.34	4.22
(110)	1.25	4.97	4.23
(111)	1.25	5.16	4.26

BY O. E. JONES AND J. D. MOTE, J. APPL. PHYS. 40, 4920, (1969)
H. B. DANIELS AND C. S. SMITH, PHYS. REV., VOL. III, P. 731, (1958).

TABLE I

COPPER SUMMARY

36---0



36---1
COPPER

CU 99.8 PER CENT OR GREATER

$V_0 = 0.1124 \text{ CC/G}$ $C_L = 4.70 \text{ KM/SEC.}$ $C_0 = 3.92 \text{ KM/SEC.}$
 $V_{01} = 0.1117 \text{ CC/G}$ $C_S = 2.26 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

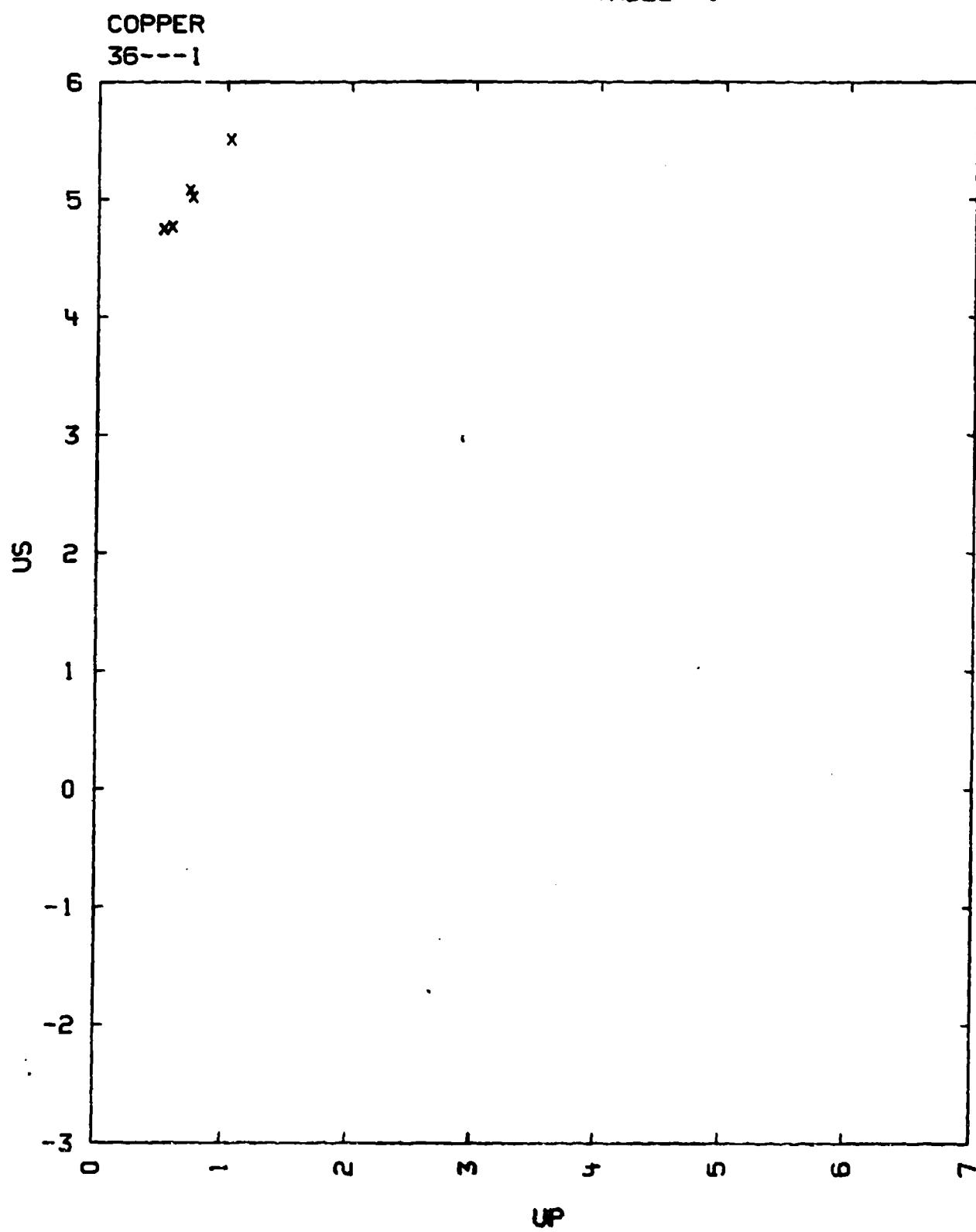
RHO0	US	UFS	UP	P	V/V0
8.90	4.744	1.024	0.511	215.8	0.8923
-	4.768	1.094	0.570	241.9	0.8804
-	5.070	1.440	0.711	320.8	0.8598
-	5.015	1.456	0.731	326.3	0.8542
-	5.508	2.079	1.032	505.9	0.8126

$$US = 3.944 + 1.51 \cdot UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.9 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-5, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS.
- 4) CL AND CS VALUES WERE OBTAINED FROM L. BERGMAN, DER ULTRASHALL,
S. HIRZEL VERLAG, STUTTGART, 1954, 6TH ED., P. 650
- 5) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.

TABLE I



38---2
COPPER

CU

$$V_0 = 0.1120 \text{ CC/G.}$$

$$V_{01} = 0.1121 \text{ CC/G.}$$

$$C_B = 3.97 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0
8.93	5.35	0.94	450	0.8264
-	7.13	2.29	1460	0.6803
-	10.16	4.19	3800	0.5882

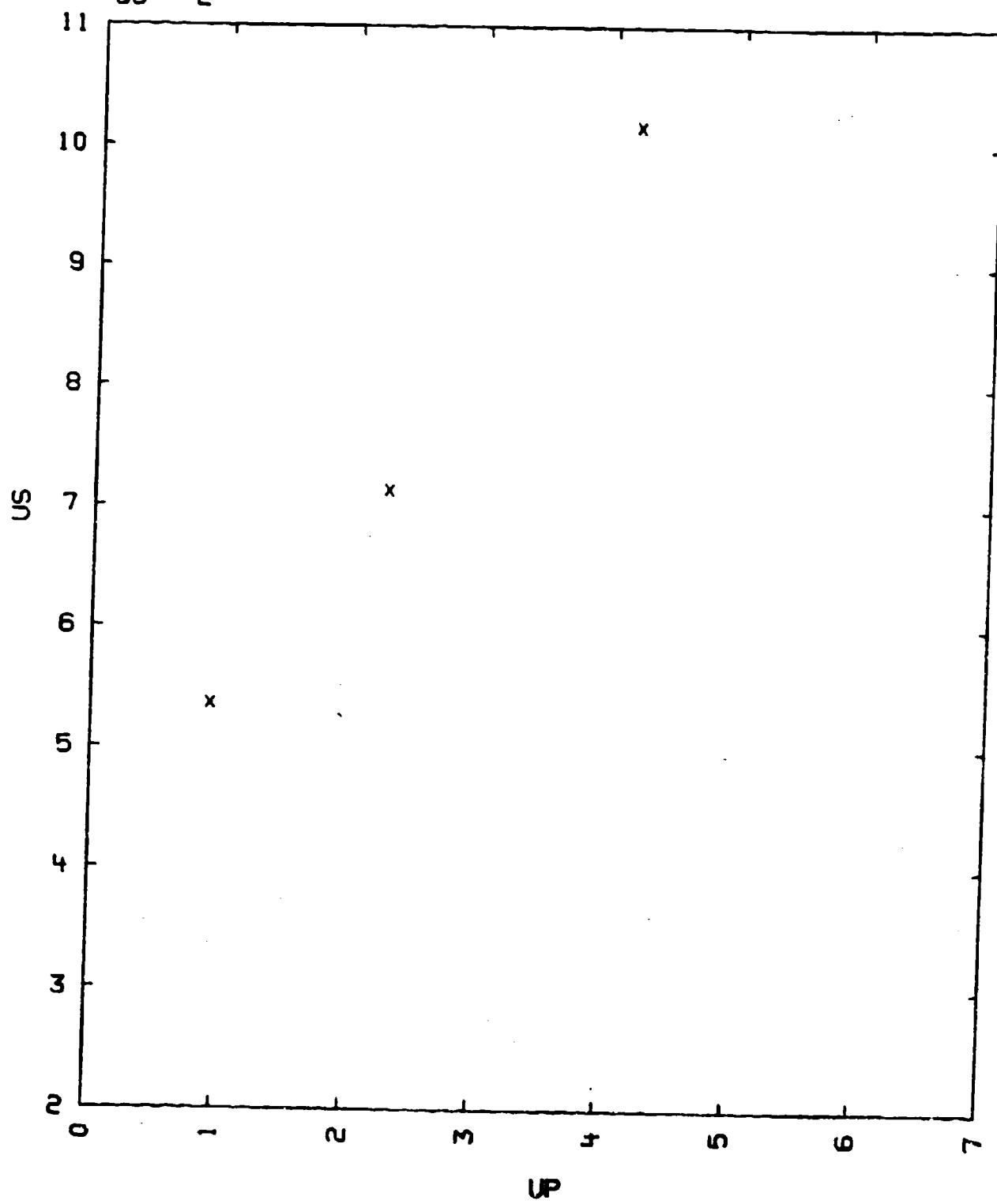
$$US = 3.90 + 1.46UP \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: AL'TSHULER, L. V., KRUPNIKOV, K. K. AND BRAZHIK, M. I.
SOVIET PHYS.-JETP, VOL. 34, P. 614 (1958)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL FE.
- 3) CB IS CALCULATED USING THE VOLUME COMPRESSION COEFFICIENT TAKEN FROM
P. W. BRIDGMAN, THE PHYSICS OF HIGH PRESSURE, (G. BELL AND SONS LTD.
LONDON 1958)
- 4) V01 WAS TAKEN FROM THE HANDBOOK OF CHEMISTRY AND PHYSICS,
(CHEMICAL RUBBER PUBLISHING CO. CLEVELAND, OHIO 1962-1963) 44TH ED.
- 5) THE SPECIMEN WITH A THICKNESS BETWEEN 6-8MM. WAS ATTACHED TO AN IRON
PLATE OF COMPARABLE THICKNESS 8-9MM.

TABLE I

COPPER
36---2



36---3
COPPER

CU

$V_0 = 0.1120 \text{ G/CC.}$
 $V_{01} = 0.1119 \text{ G/CC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL USED AS STRIKER PLATE. U(ST) IS THE STRIKER VELOCITY BEFORE IMPACT.

TABLE

RHO0	US	UP	P	V/V0	ST	U(ST)
8.93	6.64	1.82	1079	0.7262	AL	5.60
-	8.06	2.71	1950	0.6540	FE	5.60
-	10.12	4.14	3740	0.5910	FE	8.64
-	10.58	4.43	4187	0.5814	FE	9.10
-	14.20	7.15	9070	0.4965	FE	14.68

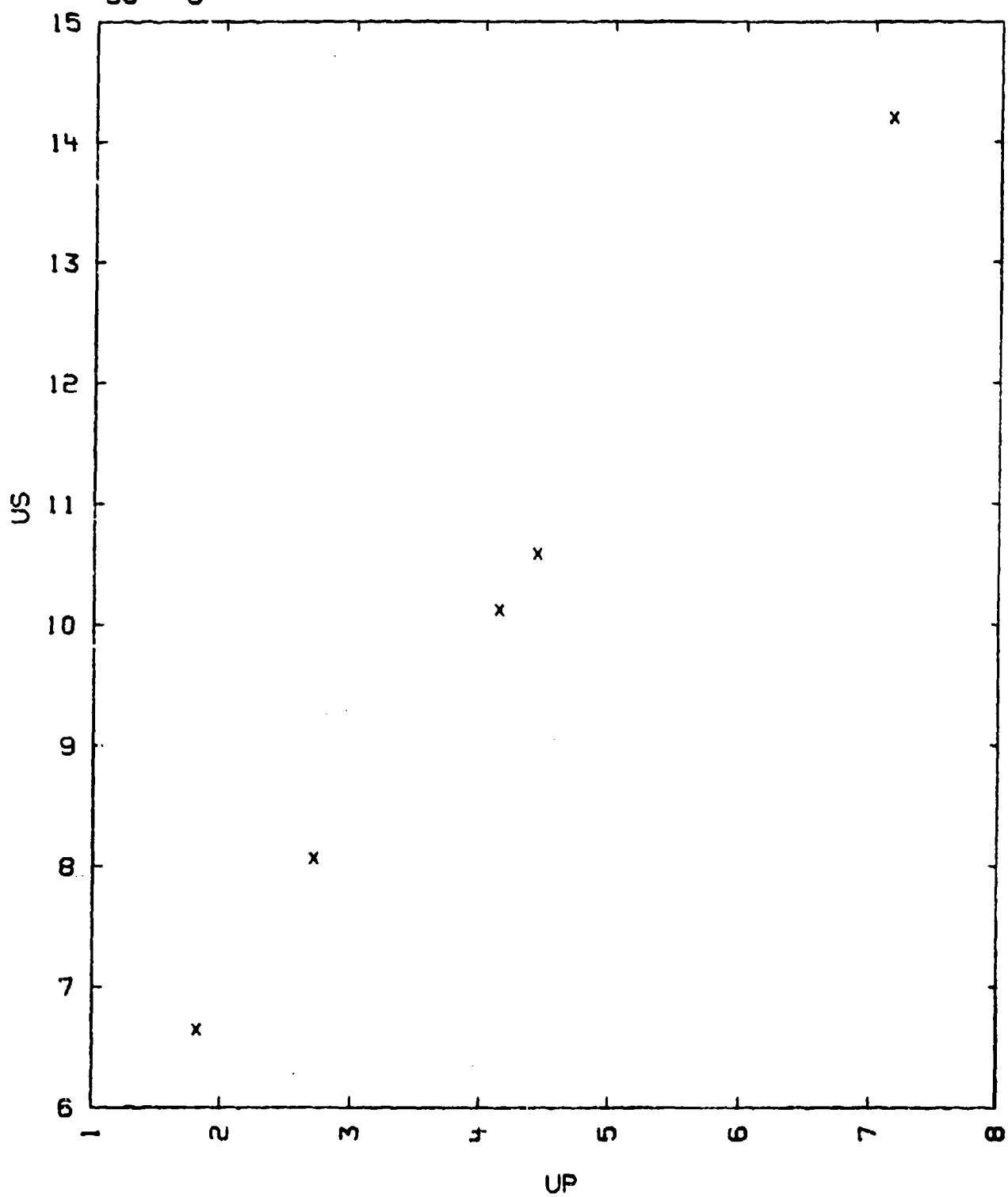
$US = 4.20 + 1.41 UP \text{ KM/SEC.}$ FOR UP BETWEEN 1.82 TO 7.15 KM/SEC.
 $\Sigma US = 0.13 \text{ KM/SEC.}$

COMMENTS:

- 1) SOURCE: AL'TSHULER, L.V., KORMER, S.B., BAKANOVA, A.A.
 SOVIET PHYS.-JETP, VOL. 11, PAGE 573 (1960)
 AL'TSHULER, L. V. ET AL
 SOVIET PHYS.-JETP, VOL. 15, P. 65 (1962)
- 2) EXPERIMENTAL TECHNIQUE A
 DATA REDUCTION TECHNIQUE A
- 3) THE PRESSURES WERE PRODUCED BY AN EXPLOSIVELY ACCELERATED ALUMINUM OR STEEL PLATE AS INDICATED IN COLUMN 6.
- 4) THE INACCURACY IN THE DETERMINATION OF US AND UP FOR PRESSURES TO 4187 KILOBARS DID NOT EXCEED 1.0 PERCENT. EACH POINT IS THE AVERAGE OF 4 TO 6 EXPERIMENTS.
- 5) THE U UNCERTAINTY OF THE VALUES IN THE LAST TABLE ENTRY ARE:
 1 PERCENT IN US OBTAINED FROM 6 TO 8 EXPERIMENTS AND 1.5 PERCENT IN U(ST) OBTAINED FROM 12 EXPERIMENTS.
- 6) V_{01} WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK (MCGRAH-HILL BOOK CO., N.Y., 1963) 2ND ED.
 THE CORRESPONDING LATTICE PARAMETER FOR THE FACE CENTERED CUBIC LATTICE IS 3.6147 ANGSTROMS

TABLE I

COPPER
36---3



36---4
COPPER

CU

$$V_0 = 0.111 \text{ CC/G}$$

$$V_{01} = 0.1120 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC., VELOCITIES IN KM/SEC., AND PRESSURE IN KILOBARS.

TABLE

SAMPLE						PROJECTILE
RHO0	US	UFS	UP	P	V/V0	VELOCITY
8.97	6.68	4.00	2.10	1260	0.686	4.20
-	6.19	3.56	1.76	977	0.717	3.51
-	6.49	3.56	1.75	1018	0.731	3.50
-	6.18	3.57	1.75	970	0.717	3.50
-	6.93	4.18	2.08	1274	0.696	4.16
-	6.16	2.76	1.35	745	0.782	2.69
-	6.22	3.35	1.68	937	0.732	3.35
-	5.90	2.90	1.36	719	0.770	2.72
-	6.23	3.55	1.69	942	0.730	3.37

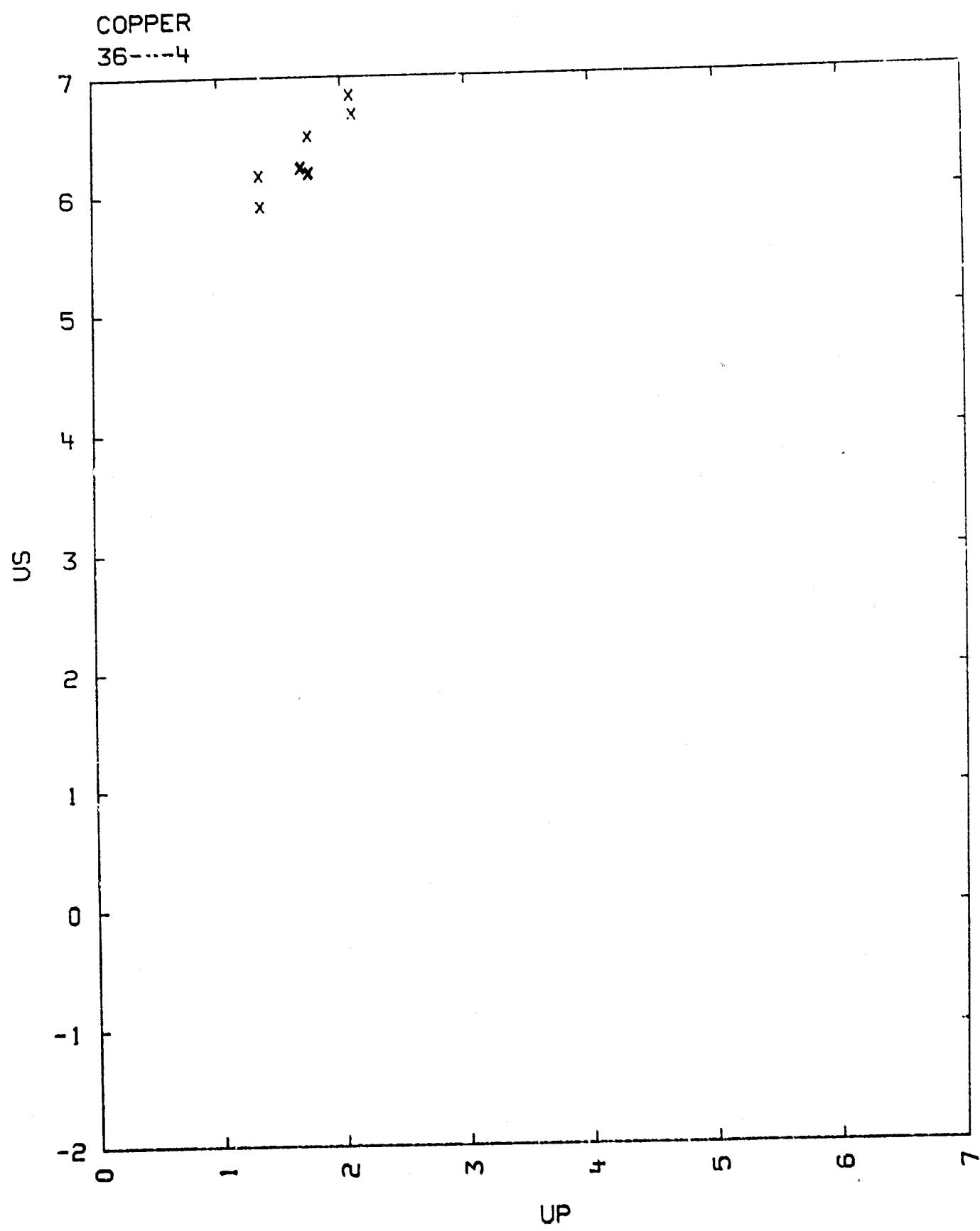
$$US = 4.64 + 0.98 \text{ UP KM/SEC}$$

$$\text{SIGMA US} = 0.149 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: PERSSON, P. A., PERSSON, I.
FOA 2 REPORT A 2299-222, SEPT. 1964.
FORSVARETS FORSKNINGSANSTALT AVDELNING, SHEDEN
UCRL TRANSLATION 1173(L).
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) PRECISION OF VELOCITY MEASUREMENTS PLUS OR MINUS 3 PERCENT.
- 4) THE PARTICLE VELOCITY UP OF THE SAMPLE WAS TAKEN AS HALF THE PROJECTILE VELOCITY. (COPPER PROJECTILE WAS MOST LIKELY USED).
- 5) THE VALUE OF V01 WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY ST
(DOVER PUBLICATIONS, INC., NEW YORK, N. Y., 1963).

TABLE I



36---5
COPPER

CU

$$V_0 = 0.112 \text{ g/cc.}$$

$$V_{CI} = 0.112 \text{ g/cc.}$$

THE TABLES BELOW GIVE THE VELOCITIES OF RELAXATION WAVES, C, AT VARIOUS PRESSURES. THE HUGONIOT STATE THE RELAXATION WAVE TRAVELS THROUGH IS GIVEN BY US, UP, P, OR V/V0. DENSITY IS GIVEN IN G/CC., VELOCITY IN KM/SEC. AND PRESSURE IN KILOBARS. IN TABLE II, ST DESIGNATES THE MATERIAL OF THE STRIKER PLATE, U(ST) IS THE VELOCITY BEFORE IMPACT AND US(ST) IS THE SHOCK VELOCITY IN THE PLATE AFTER IMPACT.

TABLE I

SOUND VELOCITIES
PERPENDICULAR TO SHOCK DIRECTION

RHO0	US	UP	P	V/V0	C
8.93	5.24	0.87	407	0.834	6.33
US =					

TABLE II

SOUND VELOCITY
PARALLEL TO SHOCK DIRECTION

---HUGONIOT STATE-----						--DRIVER SYSTEM--			
RHO0	US	UP	P	V/V0	C	ST	U(ST)	US(S1)	
8.93	6.69	1.82	1087	0.7278		AL	5.60	10.45	
-	5.92	1.31	692	0.7788	6.32	AL	5.60	10.45	
-	5.54	1.06	524	0.8084	5.95	AL	5.60	10.45	
-	8.11	2.77	2006	0.6583		FE	5.71	8.67	
-	7.67	2.47	1692	0.6780	7.78	FE	5.71	8.67	
-	7.14	2.12	1352	0.7032	7.13	FE	5.71	8.67	
-	6.62	1.77	1046	0.7326	6.71	FF	5.71	8.67	
-	10.2	4.18	3796	0.5903	9.48	FE			
-	9.43	3.70	3117	0.6106	8.93	FE			

US =

COMMENTS:

- 1) SOURCE: AL'TSHULER, L. V., KORMER, S. B., BRAZHNICK, M. I., VLADIMIROV, L. A. AND FUNTIKOV, A. I.
SOVIET PHYS.-JETP, VOL.11, P. 766 (1960)
- 2) EXPERIMENTAL TECHNIQUE A WAS USED IN TABLE II FOR THE FIRST THREE

ENTRIES. FOR ALL OTHER ENTRIES EXPERIMENTAL TECHNIQUE B WAS USED.
THE HUGONIOT STATE WAS DETERMINED FROM ONE MEASURED PARAMETER
AND THE KNOWN EQUATION OF STATE. AL'TSHULER ET AL., SOVIET PHYS-JETP,
VOL. 11, P. 573 (1960).

- 3) THE SOUND VELOCITY IN TABLE I WAS OBTAINED FROM THE SHAPE OF THE FREE SURFACE AFTER IT HAD MOVED SOME DISTANCE. IN TABLE II THE SOUND VELOCITY WAS DETERMINED FROM THE ATTENUATION IN THE MEASURED SHOCK OR FREE SURFACE VELOCITY PRODUCED BY THE RAREFACTION WAVE ORIGINATING AT THE REAR SURFACE OF THE STRIKER PLATE.
- 4) VOI WAS OBTAINED FROM THE HANDBOOK OF CHEMISTRY AND PHYSICS (THE CHEMICAL RUBBER PUBLISHING CO., CLEVELAND, OHIO 1962-1963) 44TH ED.
- 5) THE MEASURED SOUND VELOCITY IS HIGHER THAN THE CALCULATED SOUND VELOCITY FOR A PURELY NONELASTIC SAMPLE

TABLE I

COPPER
36---5

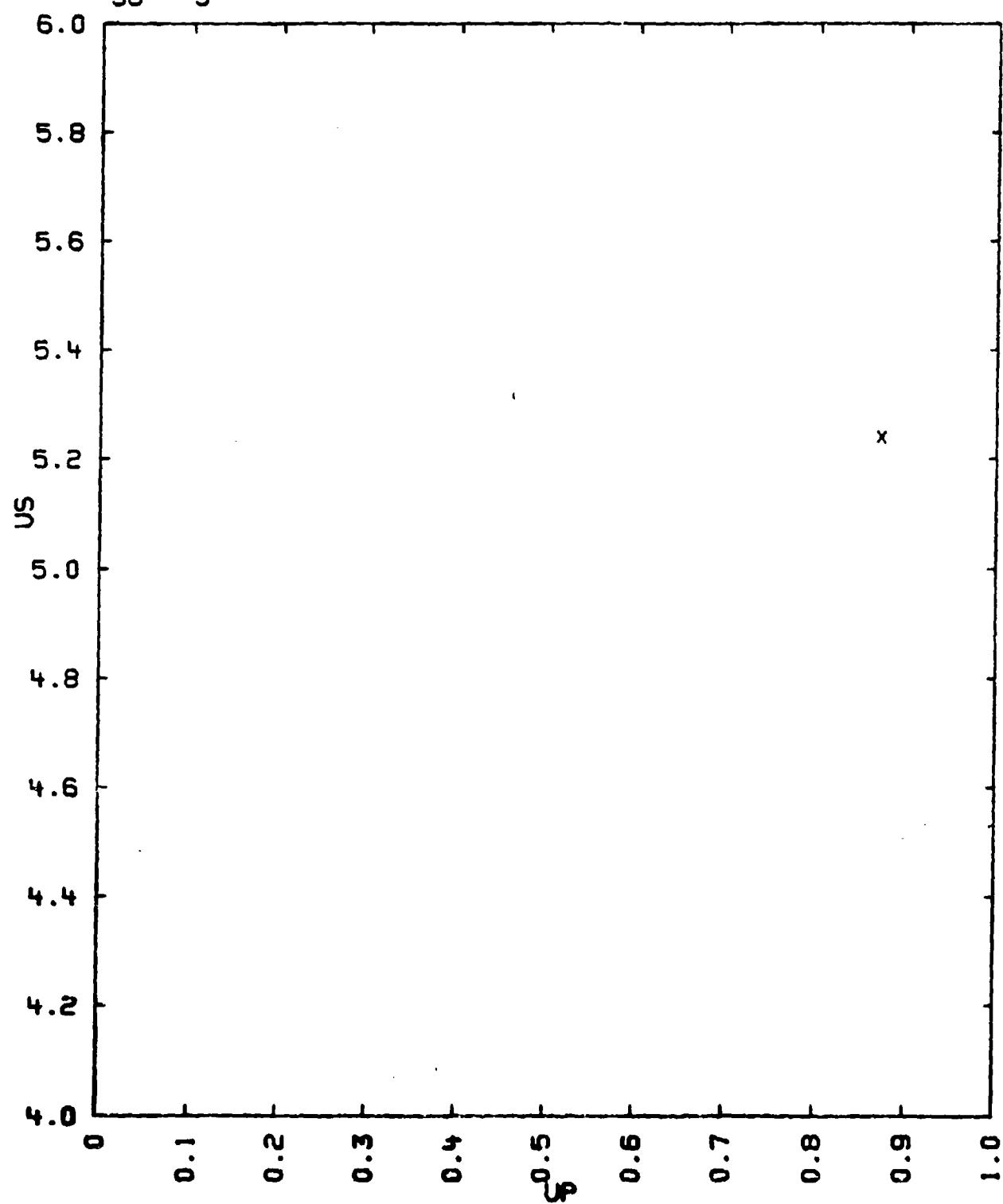
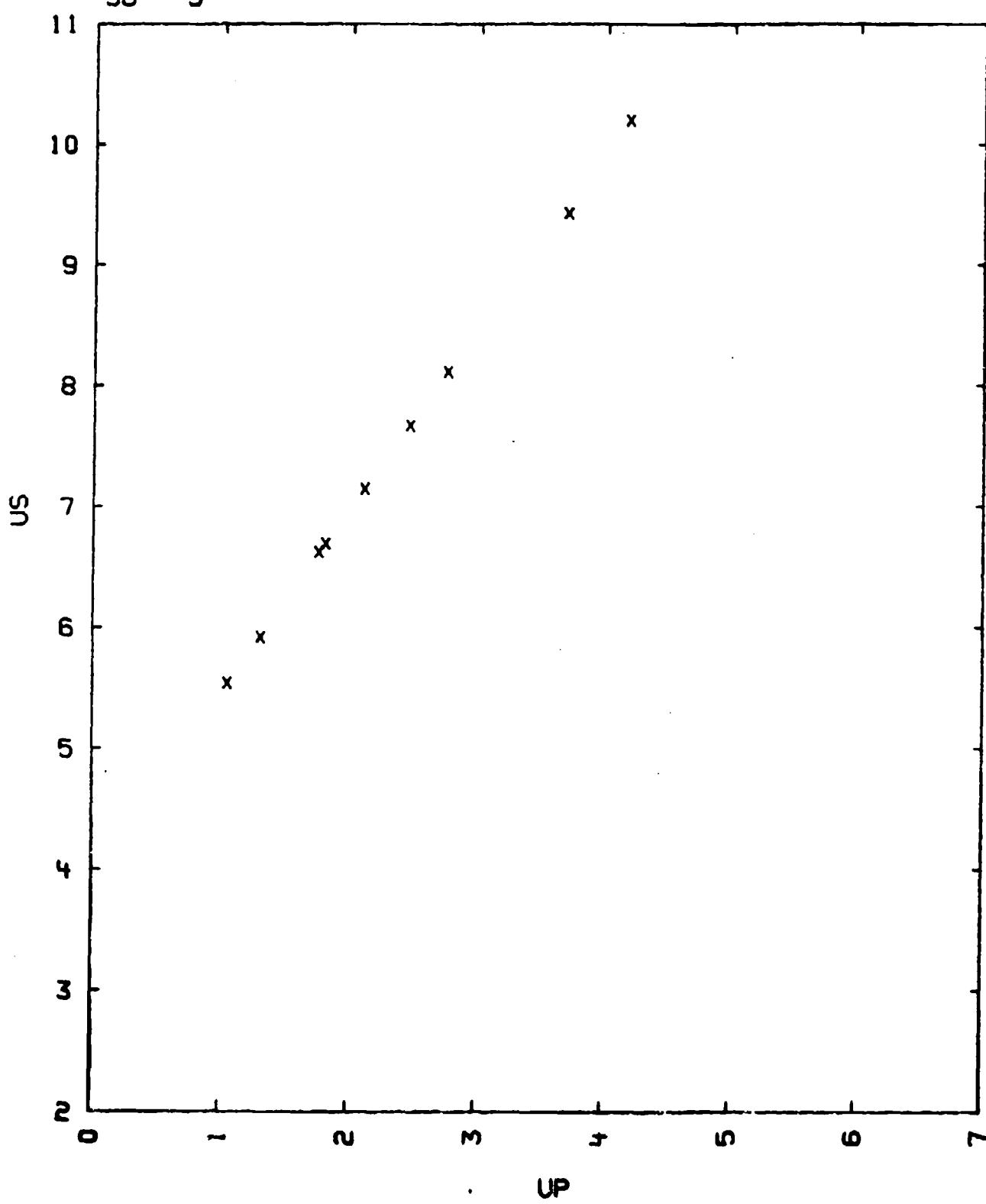


TABLE II

COPPER

36---5



36---6
COPPER POROUS

CU 99.7 PERCENT
AVERAGE GRAIN SIZE IS 5 MICRONS.

VOI = 0.112 CC/G.
VO = 1/RHO0 SEE TABLE

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURES IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL AND THE MATERIAL OF THE STRIKER PLATE. U(ST) IS THE STRIKER VELOCITY BEFORE IMPACT. SH LABELS THE HUGONIOT VARIABLES OF THE IRON SAMPLE HOLDER AFTER IMPACT.

TABLE

-----SAMPLE-----						----DRIVER SYSTEM----		
RHO0	US	SIGMA	UP	P	V/VOI	U(ST)	US(ST)	UP(ST)
5.68	9.22	0.04	5.01	2626	0.7168	8.64	10.67	4.26
4.46	9.15	0.07	5.40	2204	0.8203	8.64	10.67	4.26
2.96	8.85	0.07	6.04	1582	0.9569	8.64	10.67	4.26
2.23	8.79	0.03	6.43	1260	1.079	8.64	10.67	4.26
8.92	14.78	0.16	7.24	9550	0.5102	15.45		
5.68	14.32	0.34	8.62	7010	0.6270	15.45		
4.46	14.39	0.12	9.24	5950	0.7133	15.45		
2.23	14.50	0.14	10.95	3540	0.9823	15.5		

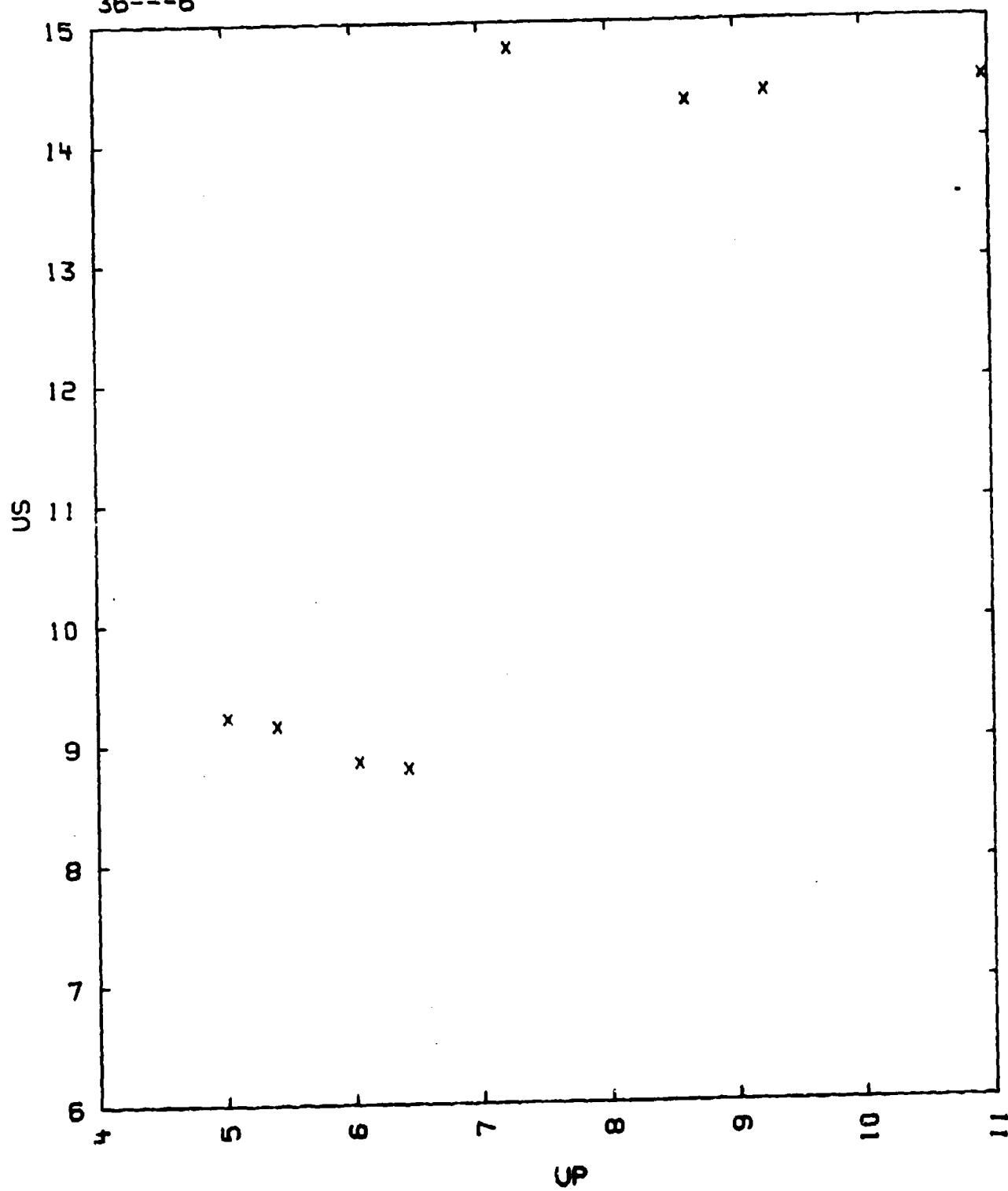
US =

COMMENTS:

- 1) SOURCE: KORMER, S. B., FUNTIKOV, A. I., URLIN, V. D.
AND KOLESNIKOVA, A. N.
SOVIET PHYS.-JETP, VOL. 15, P. 477 (1962)
- 2) EXPERIMENTAL TECHNIQUE A. (STANDARD MATERIAL IRON).
DATA REDUCTION TECHNIQUE B. (FIRST FOUR ENTRIES) AND A (OTHER ENTRIES)
- 3) US IS THE AVERAGE DETERMINATION TAKEN FROM 4-8 EXPERIMENTS, EACH
CONSISTING OF 4-6 READINGS.
- 4) VOI HAS OBTAINED FROM THE HANDBOOK OF CHEMISTRY AND PHYSICS (THE
CHEMICAL RUBBER PUBLISHING CO., CLEVELAND, OHIO 1962-1963) 44TH ED.

TABLE I

COPPER POROUS
36---6



36---7
COPPER

CU

$V_0 = .112 \text{ g/cc.}$
 $V_{01} = .112 \text{ g/cc.}$

$C_B = 3.98 \text{ km/sec.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

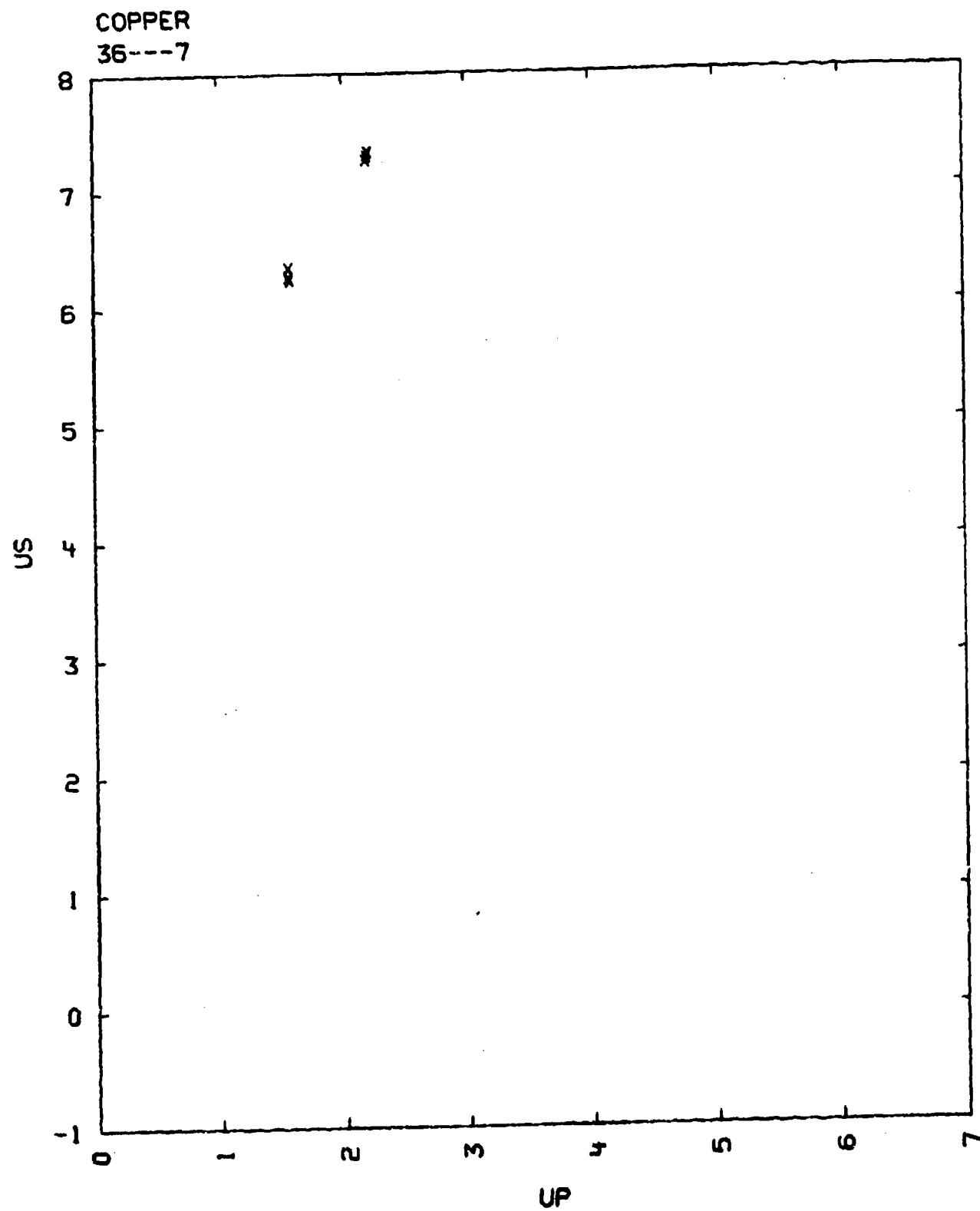
SAMPLE				STANDARD	
RHO0	US	UP	P	V/V0	P(ST)
8.90	6.33	1.57	883	0.752	842
-	6.23	1.58	875	0.746	842
-	6.26	1.57	877	0.749	842
-	7.26	2.20	1424	0.697	1362
-	7.29	2.21	1430	0.698	1366
-	7.32	2.22	1444	0.697	1378

$US = 3.958 + 1.497 UP$ $\Sigma \sigma_{US} = 0.42 \text{ percent}$

COMMENTS:

- 1) SOURCE: MCQUEEN, R. G. AND MARSH S. P.
JOURNAL OF APPLIED PHYSICS, VOL. 31, P. 1253, (1960)
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) V_{01} WAS OBTAINED FROM THE HANDBOOK OF CHEMISTRY AND PHYSICS (THE CHEMICAL RUBBER PUBLISHING CO., CLEVELAND, OHIO 1962-1963) 44TH ED.

TABLE I



36---B
COPPER

CU

V0 = 0.1112
V01 = 0.11120 CC/G

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHOO	US	UFS	UP	P	V/V0
8.93	7.150	4.710	2.290	1460	0.680
-	7.120	4.590	2.23	1420	0.686
-	7.540	5.020	2.43	1640	0.677
-	6.925	4.340	2.12	1310	0.696
-	7.235	4.780	2.32	1510	0.679
-	6.790	4.165	2.035	1230	0.700
-	7.240	4.55	2.215	1430	0.694
-	7.090	4.53	2.210	1400	0.689
-	7.225	4.75	2.310	1490	0.681
-	6.830	4.01	1.960	1190	0.713
-	6.640	3.62	1.780	1060	0.731
-	6.490	3.52	1.730	1000	0.733
-	7.160	4.51	2.195	1400	0.694
-	6.910	4.13	2.020	1240	0.708
-	6.700	3.90	1.910	1140	0.714
-	6.44	3.64	1.790	1030	0.721
-	6.88	3.95	1.930	1190	0.719
-	6.79	4.09	2.00	1210	0.706
-	6.66	3.515	1.73	1020	0.740
-	6.62	3.65	1.795	1060	0.729
-	6.845	4.20	2.05	1250	0.700
-	7.230	4.805	2.33	1505	0.677
-	7.260	4.750	2.31	1500	0.682
-	7.050	4.48	2.18	1370	0.690
-	6.890	4.37	2.13	1310	0.691
-	6.85	4.13	2.02	1235	0.705
-	6.86	4.07	1.99	1220	0.710
-	6.91	4.27	2.08	1290	0.699
-	6.775	3.70	1.82	1100	0.732
-	6.75	3.74	1.835	1110	0.728
-	6.68	3.78	1.85	1110	0.723
-	6.54	3.87	1.90	1110	0.709
-	6.555	3.525	1.73	1010	0.736
-	6.50	3.48	1.71	990	0.737
-	6.52	3.54	1.74	1010	0.734
-	6.26	3.46	1.70	950	0.728
-	6.20	2.95	1.48	810	0.764
-	6.27	3.21	1.53	850	0.756
-	6.50	3.70	1.815	1050	0.720
-	6.55	3.71	1.820	1060	0.722
-	6.39	3.50	1.720	980	0.731

COPPER

RHO0	US	UFS	UP	P	V/V0
-	6.445	3.34	1.646	950	0.744
-	6.40	3.615	1.78	1010	0.722
-	5.88	2.870	1.42	740	0.759
-	5.84	2.910	1.44	750	0.753

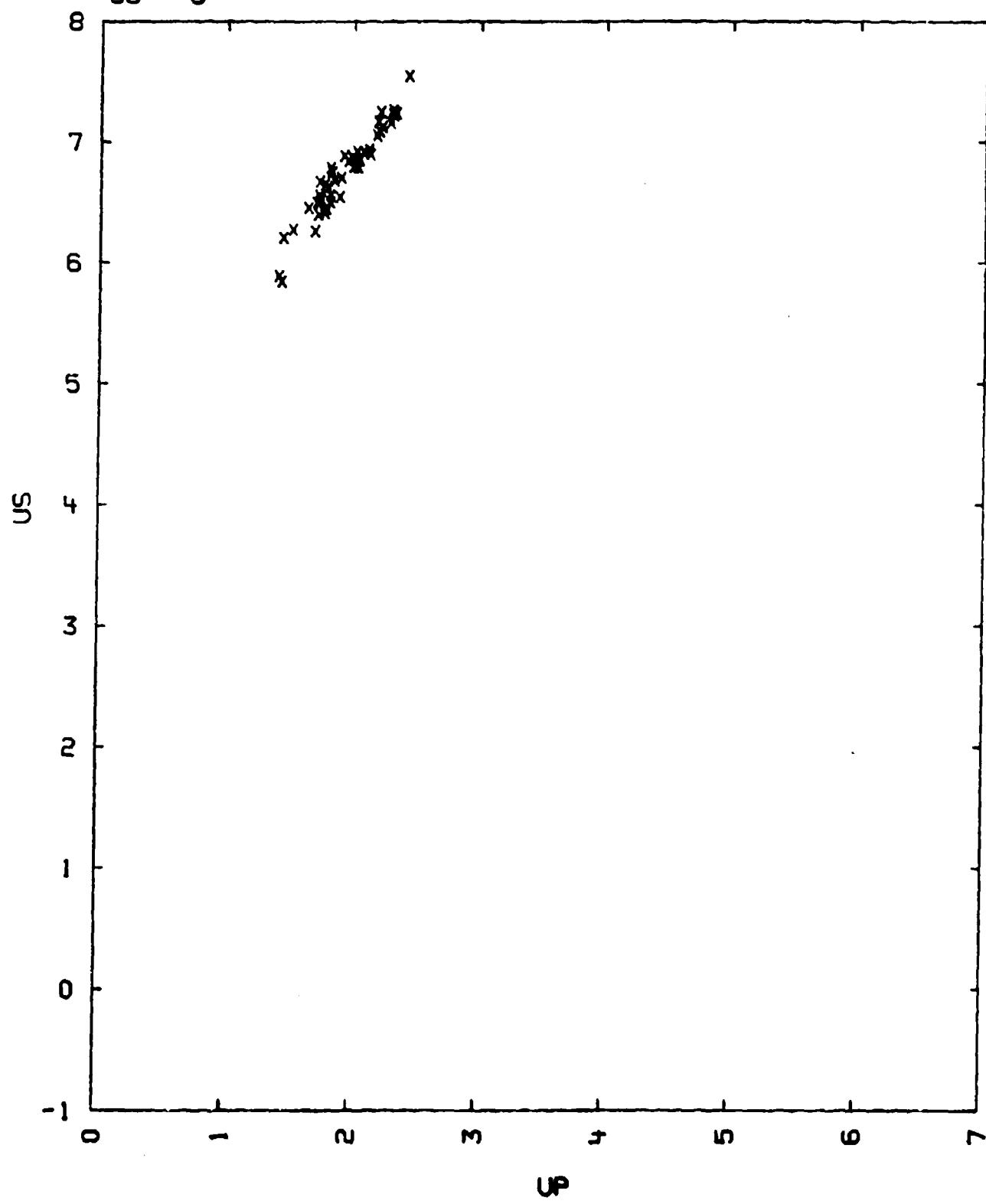
$$US = 4.12 + 1.81 UP, \quad S10.US = 0.15 \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: BERGER, J. AND FAUQUIGNON, B.P.
PRIVATE COMMUNICATION (1964), B.P. NO. 7, SEVRAN, FRANCE
- 2) EXPERIMENTAL TECHNIQUES A AND B
DATA REDUCTION TECHNIQUE
STANDARD MATERIAL
- 3) V01 WAS DETERMINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK
(MCGRAW-HILL BOOK CO., N.Y., 1963) 2ND ED.
THE CORRESPONDING LATTICE PARAMETER FOR THE FACE CENTER CUBIC
LATTICE IS 3.6147 ANGSTROMS.
- 4) SAMPLE DIMENSIONS FOR THE US MEASUREMENTS: 2.0 CM DIAMETER
0.5 CM THICKNESS
SAMPLE DIMENSIONS FOR THE UFS MEASUREMENTS: 2.0 CM DIAMETER
0.25 CM THICKNESS

TABLE I

COPPER
36---8



36---9
COPPER

PB,FE,SI,ZN,AG AND OTHERS LESS THAN 0.052 WT PERCENT
CU REMAINDER - -

$$V_0 = 0.1119 \text{ CC/G}$$

$$V_{01} = 0.1119 \text{ CC/G}$$

$$C_0 = 3.894 \text{ KM/SEC}$$

THE TABLE LISTS ALL THE PRESSURES AND COMPRESSIONS OF THE MULTIPLE WAVE SYSTEMS USED IN THE DATA ANALYSIS OF THESE RELATIVELY BROAD SHOCK WAVES, TOGETHER WITH THE HYDROSTATIC PRESSURE (PH) AND THE APPROXIMATE USI AND UPI VALUES CALCULATED FROM THEM. DENSITY IS IN G/CC, VELOCITIES IN KM/SEC AND PRESSURE (PH) OR STRESS (PI) IN KBARS. NO = EXPERIMENT NUMBER I INDICATES THE 1ST 2ND 3RD ETC. WAVE OF THE SYSTEM. U = PROJECTILE VELOCITY AND MAT = PROJECTILE MATERIAL.

TABLE

U	RHO0	USI	UPI	PI	V1/V0	I	PH	MAT
0.0315	8.939	4.5	0.0013	0.54	0.9997	1		CU
	-	4.0	0.0053	1.96	0.9987	2		
	-	3.9	0.0134	4.79	0.9966	3		
	-	3.8	0.0161	5.70	0.9959	4	5.65	
0.0374	-	4.9	0.0029	1.29	0.9994	1		
	-	3.9	0.0274	9.78	0.9931	2		
	-	3.4	0.0287	10.19	0.9927	3	10.09	
0.1223	-	4.5	0.0018	0.74	0.9996	1		
	-	4.1	0.0582	21.05	0.9856	2		
	-	3.5	0.0607	21.82	0.9849	3	21.62	
0.1735	-	4.6	0.0018	0.75	0.9996	1		
	-	4.13	0.0823	29.86	0.9797	2	29.68	
0.2440	-	4.4	0.0022	0.89	0.9995	1		
	-	4.20	0.0931	34.33	0.9774	2		
	-	4.03	0.1118	43.35	0.9711	3	43.14	
0.2995	-	4.4	0.0013	0.52	0.9997	1		
	-	4.2	0.0349	12.99	0.9916	2		
	-	4.19	0.144	52.92	0.9648	3	52.69	
0.5852	-	4.43	0.292	115.95	0.9340	1	115.63	-

US = (FOR SHOCK VELOCITY FIT, SEE COMMENT 3)

COMMENTS:

- 1) SOURCE: MUNSON, D. E. AND BARKER, L. M.
J. APPL. PHYS., VOL. 37, P. 1652, (1966)

SANDIA CORPORATION, ALBUQUERQUE, NEW MEXICO

- 2) EXPERIMENTAL TECHNIQUE F AND AN INTERFEROMETRIC METHOD.
DATA REDUCTION METHOD D WITH 2UP = UFS

WITH THE MODIFICATION THAT A SMOOTH BROAD WAVE
WAS APPROXIMATED BY A SET OF DISCRETE PRESSURE STEPS. THE SHOCK VELO-
CITY MEASUREMENTS WERE CORRECTED FOR ERRORS INTRODUCED BY SURFACE
INTERACTIONS.

- 3) NOTE THAT USI AND UPI ARE ONLY APPROXIMATE NUMBERS CALCULATED FROM
THE PI AND VI/VO VALUES. IN THE FIT BELOW US IS OBTAINED FROM THE
VALUE OF PH OF A PARTICULAR WAVE SYSTEM AND THE CORRESPONDING
VALUE OF VI USING THE EQUALITY $US = VO(PH/(VO-VI))^{1/2}$, WHILE UP WAS
TAKEN TO BE 1/2 THE MAXIMUM FREE SURFACE VELOCITY. THESE POINTS
ARE COMBINED WITH DATA FROM LOS ALAMOS TO GIVE:

$US = 3.917 + 1.520 \cdot UP$ KM/SEC. UP FROM 0 TO 2.2 KM/SEC
INCLUDING AL'TSHULER 1960 DATA UP TO UP = 4.45 KM/SEC YIELDS

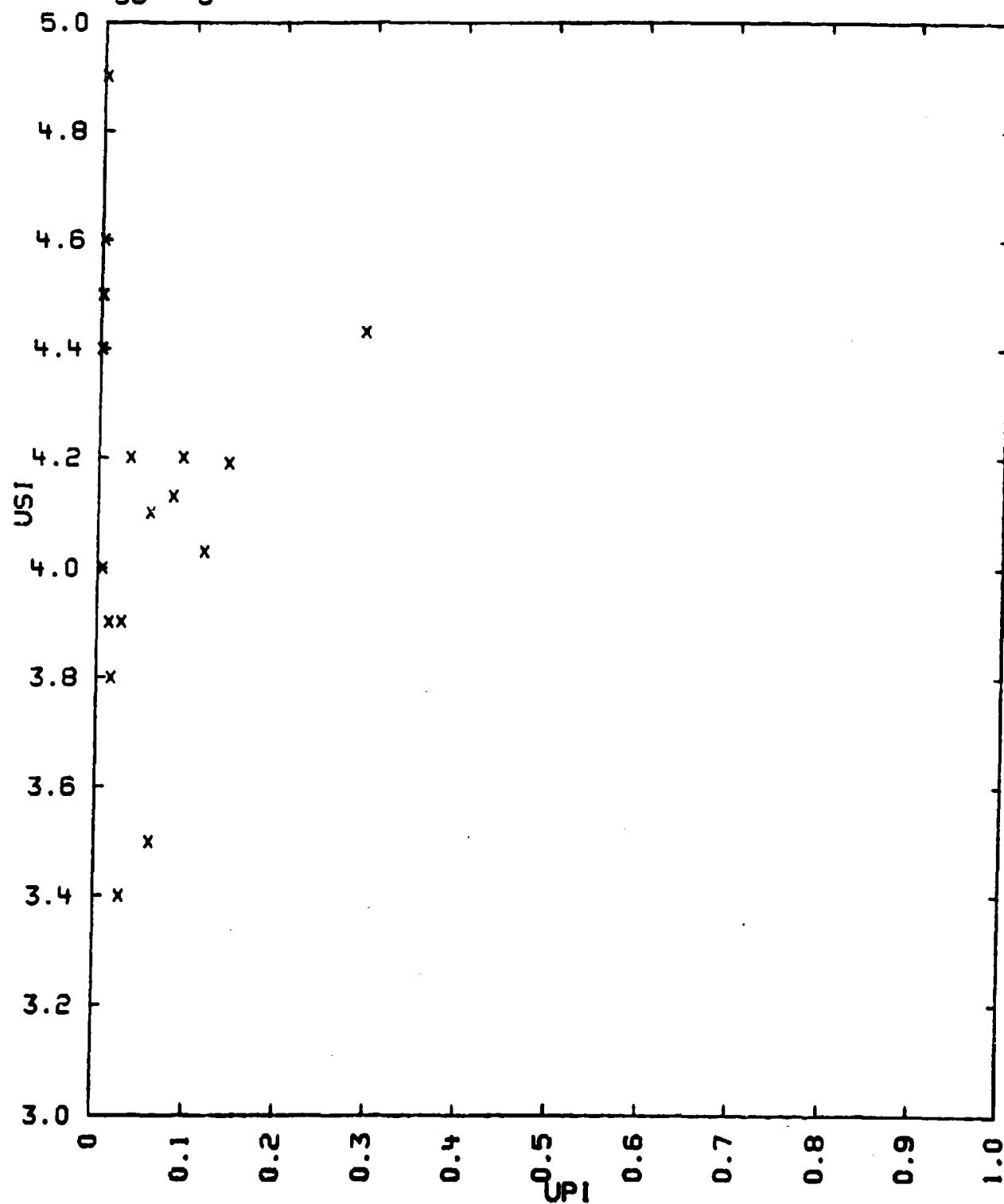
$US = 3.927 + 1.507 \cdot UP$ KM/SEC. MAX. DEV. IS 1.4 PERCENT

- 4) $PH = PI - 2/3 \cdot Y$, WHERE THE YIELD POINT Y IS A FUNCTION OF THE
DEGREE OF WORK HARDENING.

- 5) VOI WAS CALCULATED FROM THE CUBIC UNIT CELL CONSTANT GIVEN IN
WYCKOFF, CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, 1963)
 $A = 3.61496$ ANGSTROM AT 18 DEG. CENTIGRADE

TABLE I

COPPER
36---9



38--10
COPPER POROUS

CA	SI	FE	NI	SN	ZN	0.2 MOLE PERCENT	(0.05 IN SOLID SAMPLES)
C						0.01-0.1	- - -
O						0.28	- - -
CU						REST	
PARTICLE SIZE						5. TO 20 MICRONS	

$$V_0 = 0.1118 - 0.158 \text{ CC/G}$$

$$V_{01} = 0.1119 \text{ CC/G}$$

THE TABLE LISTS DENSITY IN G/CC, VELOCITIES IN KM/SEC AND PRESSURES IN KBAR. SEE COMMENT 7

TABLE I

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0
8.94	4.61	0.010	4.3	0.998	4.13	0.089	33.2	0.979
-	4.48	0.010	4.2	0.998	4.06	0.084	30.9	0.980
-	4.68	0.008	3.2	0.999	4.28	0.224	85.9	0.948
-					4.38	0.324	126.8	0.927

$$US = .386 + 2.8 \cdot UP \text{ KM/SEC}$$

$$\text{SIG US} = .24 \text{ KM/SEC}$$

TABLE II

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0
6.62	2.80	0.007	1.4	0.997	1.56	0.387	40.4	0.754
6.42	2.6	0.007	1.2	0.997	1.29	0.224	19.1	0.828
6.45	3.00	0.010	2.0	0.997	0.59	0.15	7.	0.75
6.72	2.	0.007	1.	0.996	0.59	0.13	6.	0.79
6.34	2.90	0.008	1.5	0.997	0.77	0.22	12.	0.72
6.40	2.93	0.004	0.7	0.999	1.78	0.491	56.1	0.723
6.56	3.23	0.002	0.6	0.999	1.78	0.492	57.6	0.724
6.31	3.09	0.004	0.8	0.999	1.43	0.401	36.6	0.720
6.82	3.03	0.005	1.0	0.998	1.50	0.205	21.4	0.865
6.84	2.4	0.009	1.4	0.996	0.55	0.09	4.	0.85
6.36	2.81	0.004	0.8	0.999	1.60	0.498	51.0	0.691

$$US =$$

COMMENTS:

1) SOURCE: SCHMIDT, D.N. AND LINDE, R.K.
AIR FORCE WEAPONS LAB. REP.: CONTRACT NO. AF 29(601)-7236
SRI PROJECT: POU-8127
STANFORD RES. INST., MENLO PARK, CALIFORNIA, U. S. A.

2) EXPERIMENTAL TECHNIQUE: 11 AND 12

DATA REDUCTION TECHNIQUE: C. WAVE PROFILES WERE BROAD BUT WERE APPROXIMATED BY A STEP FUNCTION MOVING AT THE AVERAGE WAVE VELOCITY LISTED

- 3) THE FOOT OF THE ELASTIC WAVE IS BETWEEN 1 AND 10 PERCENT FASTER THAN THE AVERAGE VELOCITY LISTED. THE PROFILE AND AVERAGE VELOCITY OF THE PLASTIC (CRUSHING) WAVE WAS INFERRED FROM THE STRESS-TIME PROFILE AT THE GAUGE-SAMPLE INTERFACE AND THE REVERBERATION TIME OF AN ELASTIC RELEASE WAVE REFLECTING OFF THE CRUSHING WAVE.
- 4) THE LAST SAMPLE LISTED, WITH A ρ_{H_2O} VALUE OF 6.38 G/CC WAS HEATED AT 1100 DEG. F. IN A HYDROGEN ATMOSPHERE TO REMOVE OXIDE AND VOLATILE ORGANIC MATERIALS. THIS YIELDED A COMPRESSION CONSISTENT WITH ALL EXCEPT THE FOUR SAMPLES WITH $\rho_{H_2O} = 6.62, 6.42, 6.31$ AND 6.82. THESE WERE CONSIDERABLY LESS COMPRESSIBLE, SUGGESTING A HIGHER IMPURITY LEVEL.
- 5) ALL POROUS SAMPLES WERE COMPACTED AND SINTERED AT 1700 DEG. F.
- 6) VOI WAS OBTAINED FROM WYCKOFF, CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, 1963). CUBIC CELL CONSTANT = 3.61496 ANGSTROM.
- 7) NOTE THAT THE SECOND WAVE PARAMETERS IN THE TABLE REPRESENT A CRUSHING CURVE. P IS AN EFFECTIVE LONGITUDINAL STRESS OBTAINED BY CONSIDERING THE SAMPLE AS A HOMOGENEOUS ISOTROPIC MEDIUM.

TABLE I

COPPER POROUS

36---10

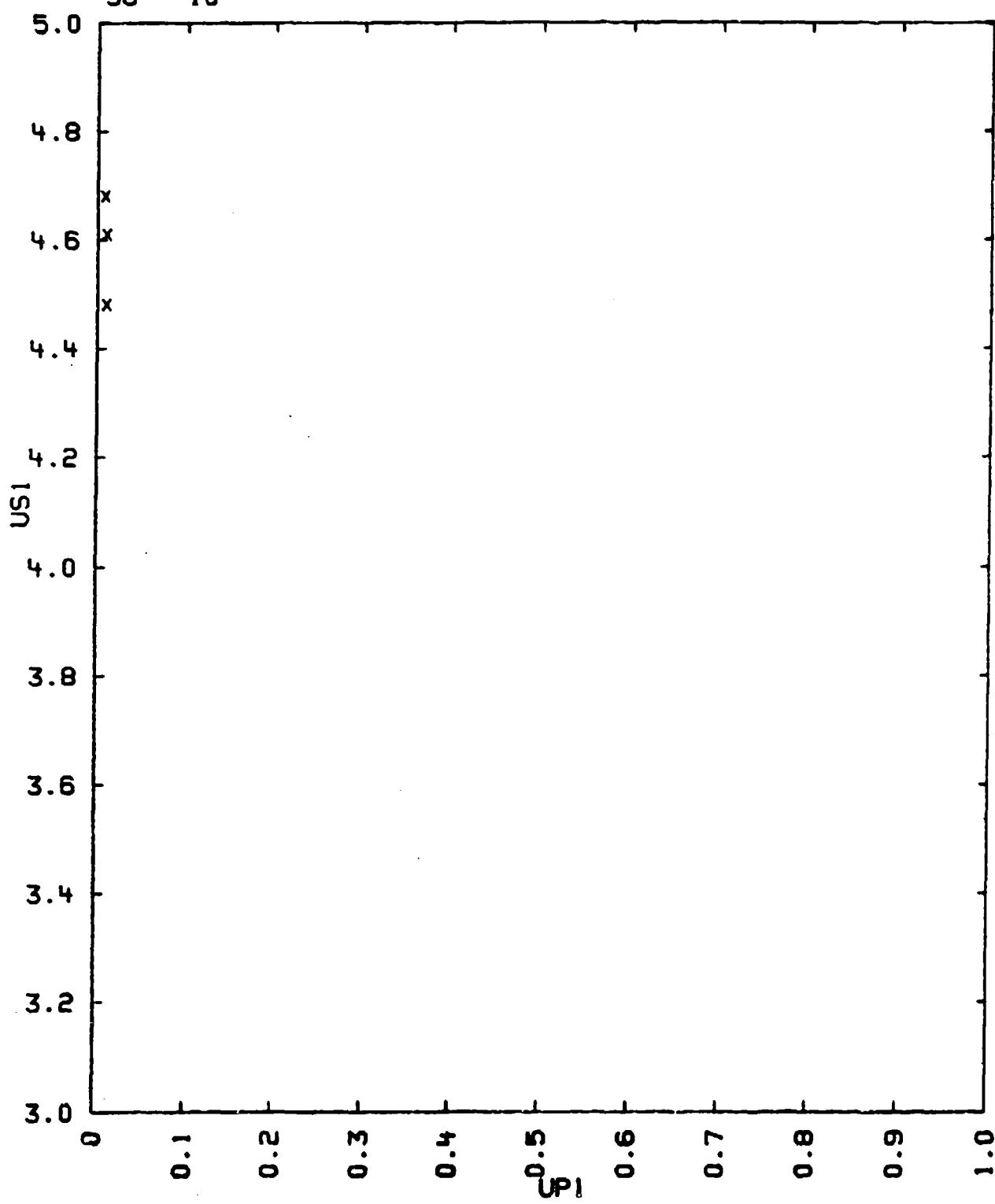


TABLE II

COPPER POROUS

36---10

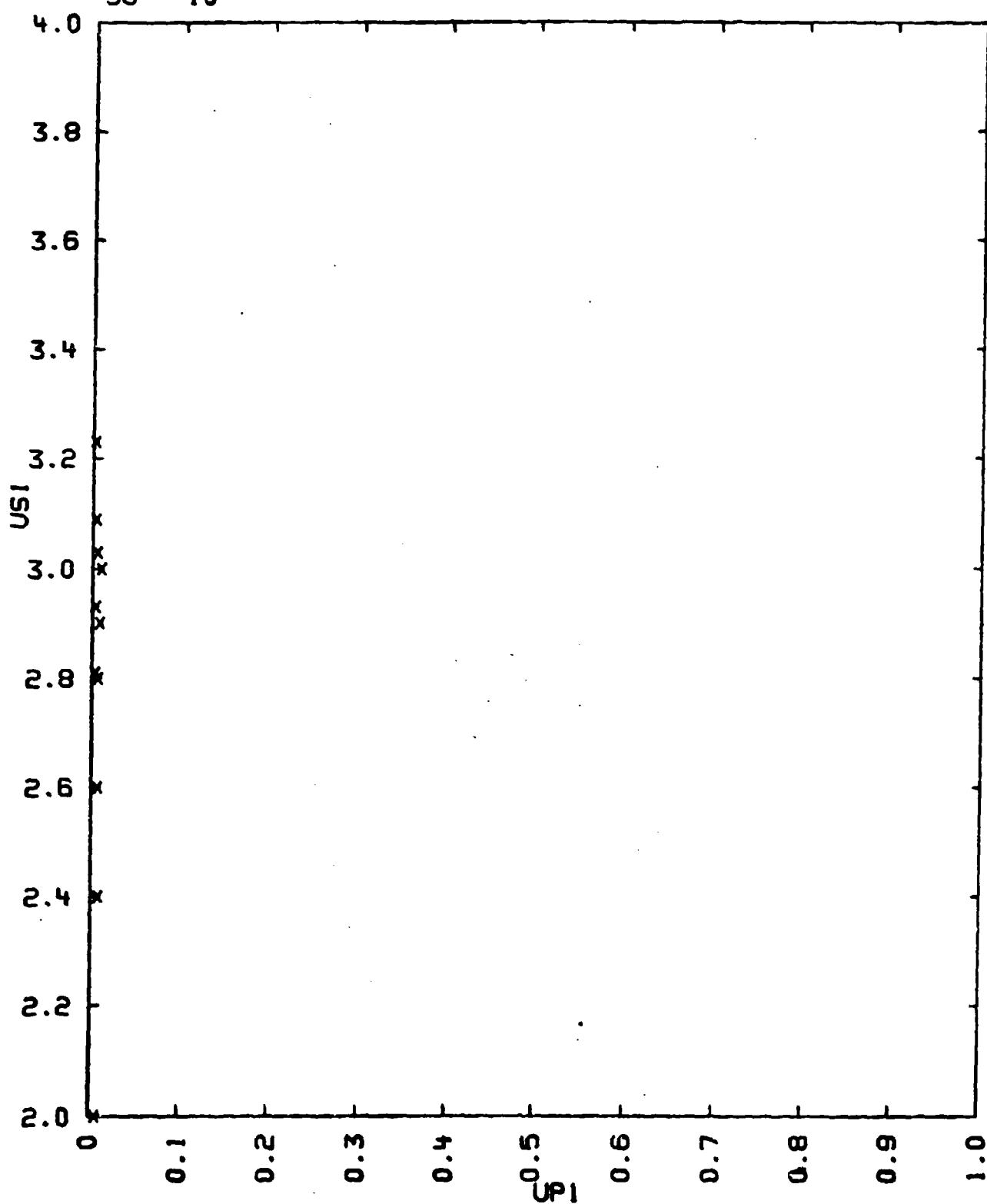


TABLE I

COPPER POROUS
36---10

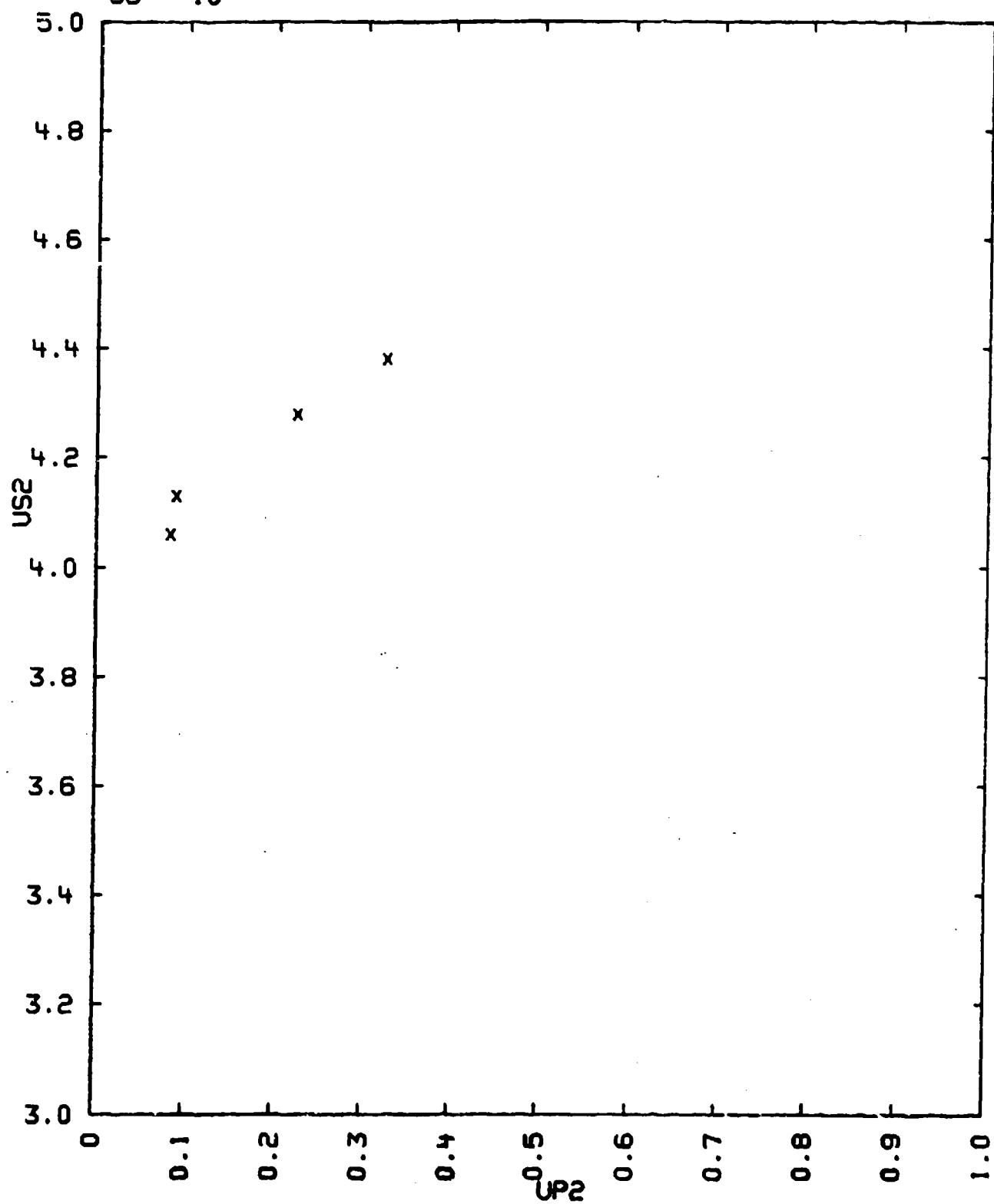
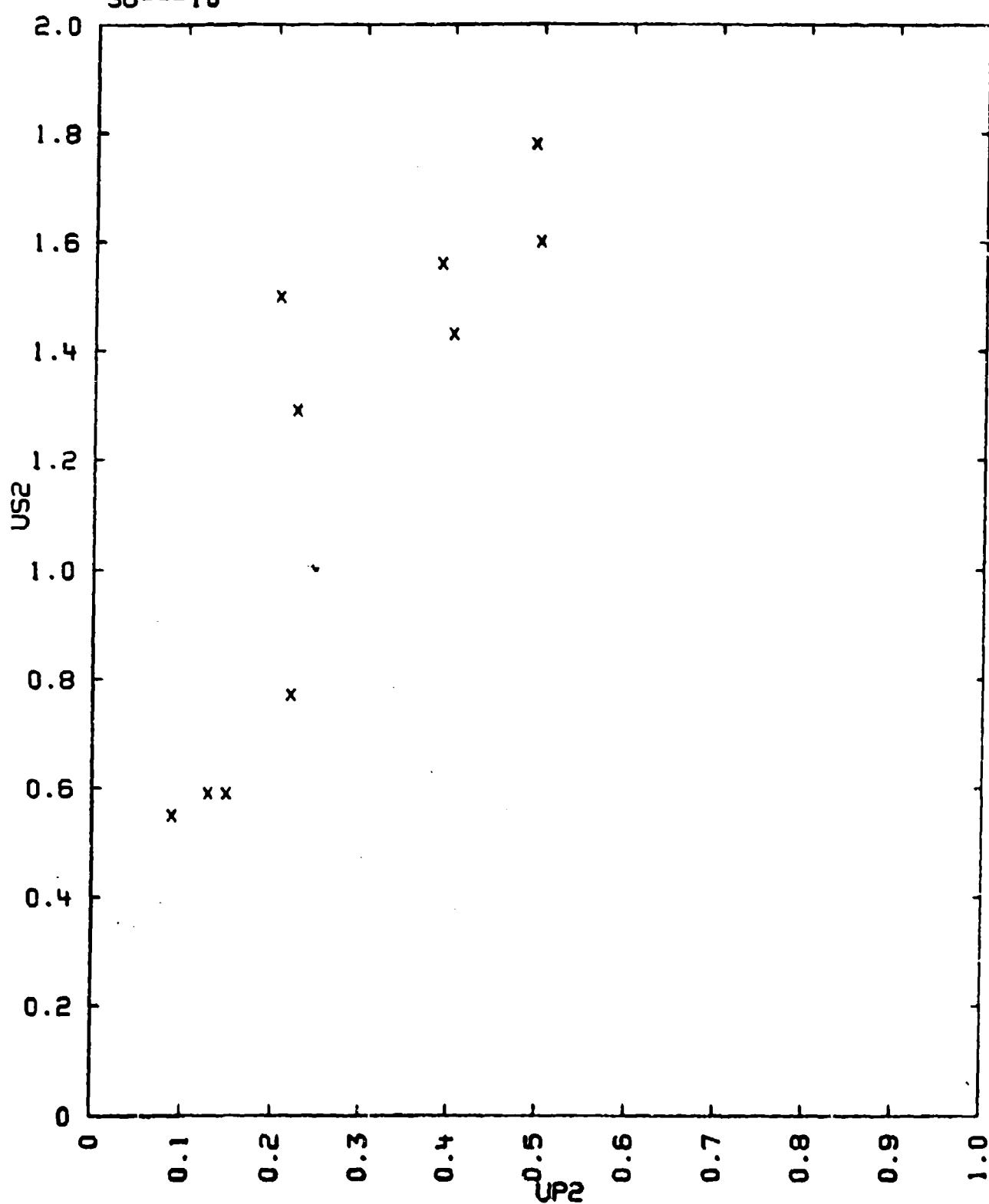


TABLE II

COPPER POROUS
36---10

38---11
COPPER, OFHC

FE	.0005	WT. PERCENT
S	.0025	- -
AG	.0010	- -
NI	.0006	- -
SB	.0005	- -
PB	.0006	- -
CU	REST	

V0 = 0.1120 CC/G

CL = 4.757 KM/SEC
CS = 2.247 KM/SEC

CO = 3.99 KM/SEC

THE TABLE LISTS RHO0 IN G/CC, VELOCITIES IN KM/SEC AND P IN KBARS. CU= COPPER, FS= FANSTEEL, AL= ALUMINUM AND WF= WEIGHTING FACTOR.

TABLE

SAMPLE					IMPACTOR			
RHO0	US	UP	P	V/V0	MAT	U	WF	
8.930	6.463	1.713	989.	.7350	CU	3.426	3	
8.930	7.414	2.343	1551.	.6840	FS	3.875	3	
8.930	7.815	2.631	1836.	.6633	AL	7.929	1	
8.930	8.062	2.808	2022.	.6517	FS	4.673	3	
8.930	8.669	3.217	2490.	.6289	CU	6.434	3	
8.930	9.076	3.509	2843.	.6134	CU	7.018	3	
8.930	9.149	3.549	2900.	.6121	CU	7.097	1	
8.930	9.196	3.561	2924.	.6128	CU	7.121	1	
8.930	9.785	3.988	3485.	.5924	FS	6.728	1	
8.930	9.802	3.971	3476.	.5949	CU	7.937	1	
8.930	10.390	4.368	4053.	.5796	FS	7.405	1	
8.930	10.785	4.674	4502.	.5666	FS	7.932	1	

$$US = 3.984 + 1.463 \cdot UP \text{ KM/SEC}$$

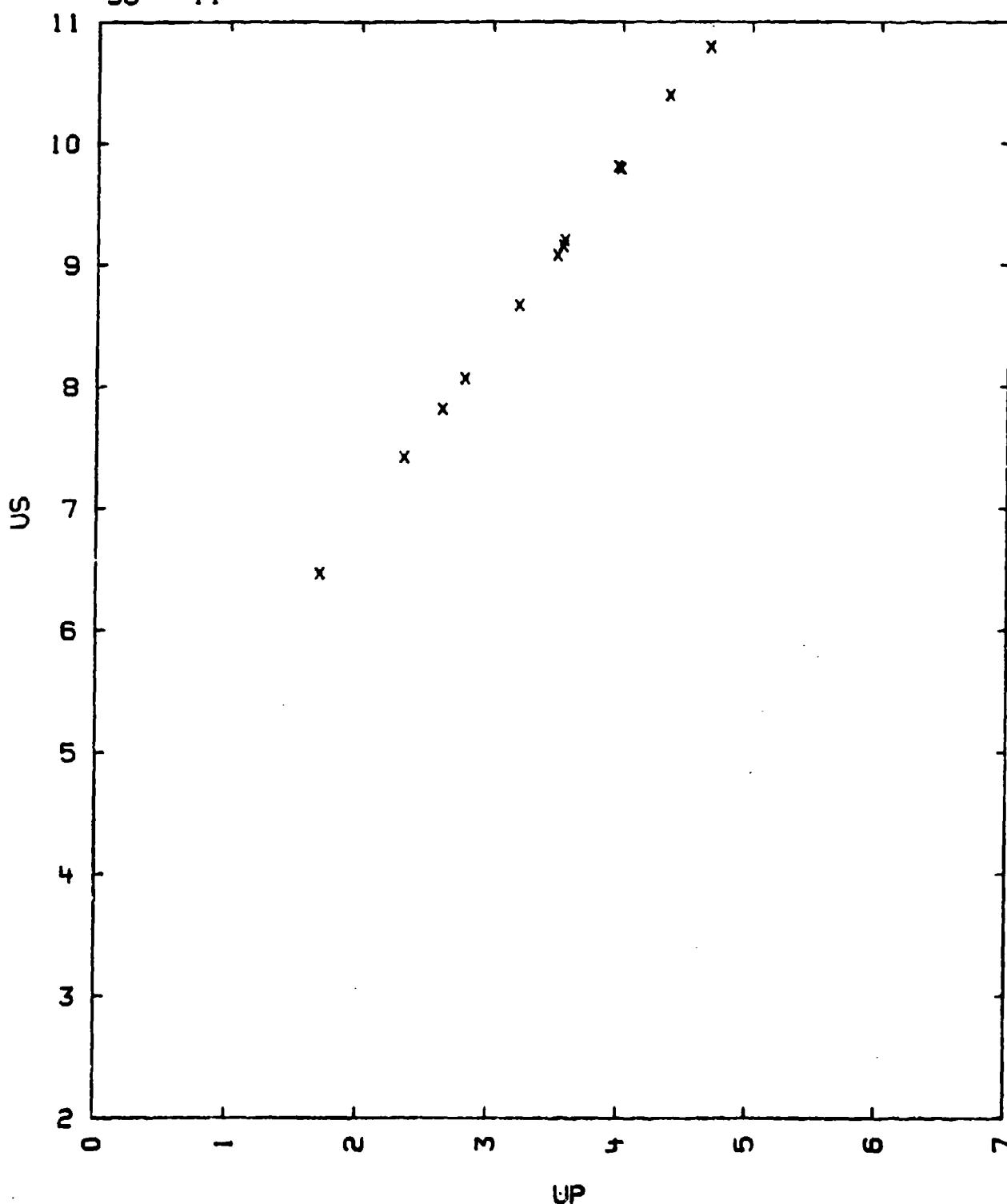
$$\text{SIG.US} = 0.009 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: ISBELL W.M., SHIPMAN F.H. AND JONES A.H.
HUGONIOT EQUATION OF STATE OF ELEVEN MATERIALS TO FIVE MBARS
MATERIALS SCIENCE LABORATORY REPORT: MSL-68-13
- 2) EXPERIMENTAL TECHNIQUE: A
DATA REDUCTION METHOD : A
- 3) NOMINAL UNCERTAINTIES ARE: $(\text{SIG.US})/\text{US} = .005$ AND $(\text{SIG.U})/\text{U} = .0005$
- 4) POISSON RATIO = 0.356

TABLE I

COPPER, OFHC
36---11



36---12
COPPER

CU

$$V_0 = 0.1120 - 0.225 \text{ CC/G} \quad C_L = 4.76 \text{ KM/SEC} \quad C_0 = 3.93 \text{ KM/SEC}$$

$$V_{01} = 0.1119 \quad C_S = 2.33 \text{ KM/SEC}$$

IN THE TABLES BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBAR AND DENSITY IN G/CC. STANDARD MATERIAL DESCRIBES THE IMPACTOR.

TABLE I

-----SAMPLE-----					-----STANDARD-----
RHO0	US	UP	P	V/V0	MATERIAL
8.930	4.48	0.41	164.	0.9085	COPPER
8.931	4.57	0.41	167.	0.9103	COPPER
8.928	4.89	0.62	271.	0.8732	COPPER
8.930	4.91	0.64	281.	0.8697	COPPER
8.930	4.91	0.73	320.	0.8513	COPPER
8.930	5.07	0.74	335.	0.8540	COPPER
8.930	4.89	0.74	323.	0.8487	COPPER
8.930	5.18	0.85	393.	0.8359	COPPER
8.930	5.36	0.95	455.	0.8228	COPPER
8.930	5.84	1.23	652.	0.7860	COPPER
8.930	6.19	1.48	818.	0.7609	COPPER
8.930	6.51	1.71	994.	0.7373	COPPER
8.928	6.55	1.74	1018.	0.7344	COPPER
8.930	6.91	1.99	1228.	0.7120	COPPER
8.930	6.88	2.04	1253.	0.7035	COPPER
8.930	7.03	2.10	1318.	0.7013	COPPER
8.928	7.46	2.37	1578.	0.6823	COPPER
8.930	7.47	2.37	1581.	0.6827	COPPER

US = $3.940 + 1.489 \cdot UP$ KM/SEC HUGONIOT FIT ADOPTED BY THE SOURCE
COMPARISON LEAST SQUARE FITS:

US = A0 + A1 · UP KM/SEC

A0 3.912 KM/SEC SIG A0 = 0.030 KM/SEC

A1 = 1.498 - SIG A1 = 0.021 -

SIG US = 0.059 -

TABLE II

-----SAMPLE-----					-----STANDARD-----
RHO0	US	UP	P	V/V0	MATERIAL US(ST)
8.920	4.31	0.21	81.	0.9513	2024 AL 5.79
8.925	4.21	0.21	79.	0.9501	2024 AL 5.79
8.900	4.22	0.22	83.	0.9479	2024 AL 5.81
8.928	4.34	0.28	108.	0.9355	2024 AL 5.84

RHO0	US	UP	P	V/V0	MATERIAL US(ST)
8.925	4.35	0.28	109.	0.9356	2024 AL 5.94
8.920	4.32	0.29	112.	0.9329	2024 AL 5.95
8.920	4.35	0.29	113.	0.9333	2024 AL 5.96
8.925	4.38	0.30	117.	0.9315	2024 AL 5.98
8.930	4.30	0.30	115.	0.9302	2024 AL 5.98
8.925	4.51	0.39	157.	0.9135	2024 AL 6.18
8.928	4.50	0.40	161.	0.9111	2024 AL 6.18
8.933	4.47	0.43	172.	0.9038	2024 AL 6.23
8.933	4.50	0.44	177.	0.9022	2024 AL 6.26
8.900	4.71	0.49	205.	0.8960	2024 AL 6.38
8.900	4.69	0.49	205.	0.8955	2024 AL 6.38
8.899	4.71	0.52	218.	0.8896	2024 AL 6.44
8.924	4.73	0.52	219.	0.8901	2024 AL 6.45
8.925	4.80	0.53	227.	0.8896	2024 AL 6.47
8.920	4.75	0.53	225.	0.8884	2024 AL 6.48
8.925	4.70	0.54	227.	0.8851	2024 AL 6.47
8.900	4.71	0.55	231.	0.8832	2024 AL 6.49
8.930	4.77	0.56	239.	0.8826	2024 AL 6.53
8.918	4.71	0.58	244.	0.8769	2024 AL 6.55
8.933	4.85	0.66	286.	0.8639	2024 AL 6.72
8.900	5.01	0.69	308.	0.8623	2024 AL 6.79
8.900	4.93	0.71	313.	0.8566	2024 AL 6.82
8.918	5.04	0.73	328.	0.8552	2024 AL 6.87
8.924	5.06	0.75	339.	0.8518	2024 AL 6.91
8.930	4.94	0.76	335.	0.8462	2024 AL 6.91
8.933	5.07	0.79	358.	0.8442	2024 AL 6.98
8.933	5.19	0.80	371.	0.8459	2024 AL 7.03
8.933	5.18	0.81	375.	0.8436	2024 AL 7.04
8.933	5.22	0.83	387.	0.8410	2024 AL 7.09
8.933	5.15	0.87	400.	0.8311	2024 AL 7.14
8.933	5.33	0.94	448.	0.8236	2024 AL 7.30
8.933	5.33	0.94	448.	0.8236	2024 AL 7.31
8.899	5.44	0.99	479.	0.8180	2024 AL 7.41
8.900	5.44	1.01	489.	0.8143	2024 AL 7.43
8.918	5.51	1.03	506.	0.8131	2024 AL 7.48
8.924	5.50	1.06	520.	0.8073	2024 AL 7.54
8.930	5.53	1.06	523.	0.8083	2024 AL 7.55
8.930	6.03	1.39	748.	0.7695	2024 AL 8.21
8.930	6.02	1.40	753.	0.7674	2024 AL 8.21
8.924	5.98	1.40	747.	0.7659	2024 AL 8.21
8.918	6.92	2.00	1234.	0.7110	2024 AL 9.40
8.918	7.23	2.17	1399.	0.6999	2024 AL 9.74
8.929	7.17	2.21	1415.	0.6918	2024 AL 9.79
8.924	7.65	2.41	1645.	0.6850	2024 AL 10.23
8.924	7.60	2.46	1668.	0.6763	2024 AL 10.29
8.924	7.73	2.56	1766.	0.6688	2024 AL 10.47

US = A0 + A1*UP KM/SEC

A0 = 3.910 KM/SEC

A1 = 1.510 -

SIG A0 = 0.012 KM/SEC

SIG A1 = 0.011 -

SIG US = 0.049 -

TABLE III

SAMPLE					STANDARD		
RHO	US	UP	P	V/VO	MATERIAL	US(ST)	
8.930	4.32	0.31	120.	0.9262	921-T AL	5.76	
8.925	4.53	0.41	166.	0.9095	921-T AL	5.98	
8.925	4.49	0.41	164.	0.9087	921-T AL	5.98	
8.930	4.76	0.62	264.	0.8697	921-T AL	6.43	
8.925	5.08	0.79	358.	0.8445	921-T AL	6.80	
8.925	5.07	0.79	357.	0.8442	921-T AL	6.80	
8.930	5.05	0.82	370.	0.8376	921-T AL	6.86	
8.930	5.05	0.82	370.	0.8376	921-T AL	6.86	
8.925	5.58	1.11	553.	0.8011	921-T AL	7.48	
8.925	5.53	1.12	553.	0.7975	921-T AL	7.48	
8.930	5.53	1.13	558.	0.7957	921-T AL	7.52	
8.928	6.12	1.46	798.	0.7614	921-T AL	8.21	
8.928	6.08	1.47	798.	0.7582	921-T AL	8.21	
8.928	6.56	1.78	1043.	0.7287	921-T AL	8.84	
8.928	6.56	1.78	1043.	0.7287	921-T AL	8.84	
8.928	7.24	2.16	1396.	0.7017	921-T AL	9.65	
8.928	7.14	2.18	1390.	0.6947	921-T AL	9.65	
8.928	7.67	2.48	1698.	0.6767	921-T AL	10.26	
8.928	7.65	2.48	1694.	0.6758	921-T AL	10.26	

US = A0 + A1 * UP KM/SEC

A0 = 3.836 KM/SEC

A1 = 1.539 -

SIG A0 = 0.021 KM/SEC

SIG A1 = 0.014 -

SIG US = 0.041 -

TABLE IV

SAMPLE					STANDARD		
RHO	US	UP	P	V/VO	MATERIAL	US(ST)	
8.930	5.01	0.68	304.	0.8643	FE	4.93	
8.930	4.99	0.68	303.	0.8637	FE	4.93	
8.930	5.47	0.98	479.	0.8208	FE	5.49	
8.930	5.47	0.98	479.	0.8208	FE	5.49	
8.930	6.31	1.57	885.	0.7512	FE	6.53	
8.930	6.87	1.94	1190.	0.7176	FE	7.15	
8.930	6.84	1.94	1185.	0.7184	FE	7.15	
8.929	7.59	2.39	1620.	0.6851	FE	7.90	
8.929	7.85	2.42	1653.	0.6837	FE	7.96	
8.929	7.68	2.48	1701.	0.6771	FE	8.05	
8.929	7.85	2.49	1701.	0.6745	FE	8.05	
8.929	8.23	2.78	2043.	0.6622	FE	8.54	
8.929	8.17	2.79	2035.	0.6585	FE	8.54	
8.929	8.16	2.79	2033.	0.6581	FE	8.54	
8.929	8.14	2.81	2042.	0.6548	FE	8.56	
8.929	8.10	2.82	2040.	0.6519	FE	8.56	
8.929	8.09	2.82	2037.	0.6514	FE	8.56	
8.929	8.08	2.82	2035.	0.6510	FE	8.56	
8.929	8.41	2.92	2193.	0.6528	FE	8.75	
8.929	8.35	2.93	2185.	0.6491	FE	8.75	

COPPER

RHO	US	UP	P	V/V0	MATERIAL	US(ST)
8.929	8.32	2.93	2177.	0.6478	FE	8.75
8.929	8.28	2.94	2174.	0.6449	FE	8.75

US = A0 + A1 * UP KM/SEC
A0 = 4.008 KM/SEC SIG A0 = 0.037 KM/SEC
A1 = 1.477 - SIG A1 = 0.015 -
SIG US = 0.055 -

TABLE V

SAMPLE					STANDARD	
RHO	US	UP	P	V/V0	MATERIAL	US(ST)
8.923	4.59	0.48	197.	0.8954	TU-3 MO	3.20
8.923	4.60	0.48	197.	0.8957	TU-3 MO	3.20
8.930	4.80	0.61	261.	0.8729	TU-3 MO	3.37
8.930	4.79	0.61	261.	0.8727	TU-3 MO	3.37
8.923	4.81	0.62	266.	0.8711	TU-3 MO	3.38
8.923	4.80	0.62	266.	0.8708	TU-3 MO	3.38
8.918	5.24	0.83	388.	0.8416	TU-3 MO	3.66
8.921	5.24	0.83	388.	0.8416	TU-3 MO	3.66
8.930	5.22	0.86	401.	0.8352	TU-3 MO	3.70
8.930	5.21	0.86	400.	0.8349	TU-3 MO	3.70
8.930	5.33	0.94	447.	0.8236	TU-3 MO	3.81
8.930	6.01	1.34	719.	0.7770	TU-3 MO	4.34
8.929	5.98	1.38	737.	0.7692	TU-3 MO	4.38
8.930	6.18	1.50	828.	0.7573	TU-3 MO	4.53
8.920	6.25	1.52	847.	0.7568	TU-3 MO	4.58
8.929	7.36	2.31	1518.	0.6861	TU-3 MO	5.59
8.931	7.72	2.46	1698.	0.6813	TU-3 MO	5.79
8.930	8.47	2.87	2171.	0.6812	TU-3 MO	6.36

US = A0 + A1 * UP KM/SEC
A0 = 3.852 KM/SEC SIG A0 = 0.025 KM/SEC
A1 = 1.575 - SIG A1 = 0.018 -
SIG US = 0.054 -

TABLE VI

SAMPLE					STANDARD	
RHO	US	UP	P	V/V0	MATERIAL	US(ST)
7.877	3.39	0.63	168.	0.8136	CU	4.71
7.885	3.32	0.65	170.	0.8042	CU	4.73
7.798	3.71	0.78	225.	0.7898	CU	4.92
7.880	3.77	0.82	244.	0.7825	CU	4.98
7.887	4.50	1.11	384.	0.7533	CU	5.40
7.908	4.55	1.18	403.	0.7539	CU	5.42
7.878	4.41	1.16	403.	0.7570	CU	5.45

COPPER

RHOO	US	UP	P	V/VO	MATERIAL	US(ST)
7.914	4.75	1.18	444.	0.7516	CU	5.51
7.880	5.00	1.39	548.	0.7220	CU	5.80
7.810	5.78	1.76	805.	0.6955	CU	6.35
7.932	5.69	1.78	808.	0.6954	CU	6.37
7.881	6.65	1.51	1211.	0.6526	CU	7.13
7.827	6.84	2.33	1226.	0.6491	CU	7.15
7.894	6.73	2.33	1242.	0.6548	CU	7.17
7.815	7.01	2.47	1370.	0.6476	CU	7.38
7.868	7.23	2.59	1492.	0.6418	CU	7.57
7.995	7.84	2.93	1837.	0.6263	CU	8.07
7.866	7.78	2.97	1813.	0.6173	CU	8.09

US =

TABLE VII

-----SAMPLE-----					-----STANDARD-----	
RHOO	US	UP	P	V/VO	MATERIAL	US(ST)
7.368	3.00	0.67	148.	0.7767	CU	4.71
7.384	2.78	0.70	144.	0.7482	CU	4.73
7.182	3.32	0.84	200.	0.7470	CU	4.82
7.364	3.32	0.88	215.	0.7349	CU	4.98
7.208	4.09	1.19	351.	0.7090	CU	5.40
7.201	4.12	1.21	359.	0.7063	CU	5.42
7.384	4.15	1.22	373.	0.7060	CU	5.45
7.193	4.33	1.27	398.	0.7067	CU	5.51
7.354	4.72	1.48	507.	0.6907	CU	5.80
7.216	5.44	1.88	730.	0.6581	CU	6.35
7.372	5.47	1.87	754.	0.6581	CU	6.37
7.489	6.52	2.39	1167.	0.6334	CU	7.15
7.386	6.42	2.40	1135.	0.6262	CU	7.13
7.219	6.53	2.44	1150.	0.6263	CU	7.17
7.201	6.62	2.61	1244.	0.6057	CU	7.38
7.384	6.98	2.70	1392.	0.6132	CU	7.57
7.376	7.51	3.06	1693.	0.5925	CU	8.07
7.374	7.59	3.08	1713.	0.5968	CU	8.09

US =

TABLE VIII

-----SAMPLE-----					-----STANDARD-----	
RHOO	US	UP	P	V/VO	MATERIAL	US(ST)
6.372	2.41	0.74	114.	0.6929	CU	4.71
6.382	2.28	0.78	112.	0.6579	CU	4.73
6.270	2.80	0.92	162.	0.6714	CU	4.92
6.325	2.79	0.97	171.	0.6523	CU	4.98

COPPER

RHO0	US	UP	P	V/V0	MATERIAL US(ST)
6.290	3.56	1.30	291.	0.6348	CU 5.40
6.309	3.64	1.31	301.	0.6401	CU 5.42
6.385	3.69	1.33	314.	0.6396	CU 5.45
6.279	3.89	1.37	335.	0.6478	CU 5.51
6.300	4.27	1.50	428.	0.6276	CU 5.80
6.278	5.08	2.00	638.	0.6063	CU 6.35
6.332	5.10	2.01	649.	0.6059	CU 6.37
6.392	6.17	2.56	1010.	0.5851	CU 7.13
6.350	6.13	2.56	1004.	0.5791	CU 7.15
6.278	6.14	2.61	1006.	0.5749	CU 7.17
6.264	6.39	2.77	1109.	0.5665	CU 7.38
6.327	6.70	2.89	1223.	0.5687	CU 7.57
6.342	7.37	3.26	1524.	0.5577	CU 8.09
6.343	7.21	3.27	1495.	0.5485	CU 8.07

US -

TABLE IX

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL US(ST)	
5.983	2.26	0.77	104.	0.6593	CU	4.71
5.933	2.22	0.80	104.	0.6398	CU	4.73
5.587	2.51	0.98	137.	0.6098	CU	4.92
5.718	? .81	1.01	151.	0.6130	CU	5.40
5.584	3.35	1.37	255.	0.5910	CU	5.45
5.830	3.52	1.38	283.	0.6080	CU	5.45
5.605	3.38	1.39	283.	0.5888	CU	2.42
5.605	3.65	1.45	297.	0.6027	CU	5.51
5.878	4.17	1.64	402.	0.6067	CU	5.80
5.712	4.85	2.10	594.	0.5758	CU	6.37
5.557	4.87	2.10	568.	0.5668	CU	6.35
5.978	6.10	2.62	955.	0.5705	CU	7.13
5.785	6.93	2.89	923.	0.5464	CU	7.15
5.627	5.99	2.72	917.	0.5459	CU	7.17
5.598	6.20	2.89	1003.	0.5339	CU	7.38
5.892	6.72	2.95	1186.	0.5610	CU	7.57
5.972	7.33	3.33	1458.	0.5457	CU	8.09
5.801	7.20	3.36	1403.	0.5333	CU	8.07

US -

TABLE X

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL US(ST)	
4.453	2.83	1.50	189.	0.4700	CU	5.40

COPPER

RHO0	US	UP	P	V/V0	MATERIAL US(ST)	
4.467	2.92	1.52	199.	0.4795	CU	5.42
4.533	3.21	1.58	230.	0.5078	CU	5.51
4.577	5.63	2.94	759.	0.4778	CU	7.17
4.513	5.86	3.13	828.	0.4659	CU	7.38
US =						

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: A (TABLE I) B (THE REST)
- 3) TABLE I IS THE STANDARD HUGONIOT OBTAINED FROM SYMMETRIC IMPACT.
TABLES II THROUGH V CHECK THE INTERNAL CONSISTENCY OF THE OTHER
HUGONIOTS OBTAINED WITH SYMMETRIC IMPACTS.
- 4) THE GRUNEISEN GAMMA OF 1.98 (= GAM) WITH THE ASSUMPTION THAT RHO.GAM
= CONSTANT REPRESENTS THE BEHAVIOR OF THE POROUS SAMPLES. ALSO LISTED
 $(DV/DT)/V = 4.89E-5$ PER DEG. AND CP= 0.389 JOULES PER G. DEG.
- 5) VOI WAS CALCULATED FROM THE LATTICE CONSTANTS IN WYCKOFF, CRYSTAL
STRUCTURES (JOHN WILEY AND SONS, N.Y., 1963) VOL. 1.
- 6) HUGONIOT ELASTIC LIMIT 0. = 6.3 KBAR (ANNEALED - 50 PERCENT COLD
HUGONIOT ELASTIC LIMIT WORKED RESP.)
- 7) THE 921-T AL PLATE STOCK USED TO OBTAIN THE CROSS CHECK DATA IN
TABLE III IS SLIGHTLY POROUS.

TABLE I

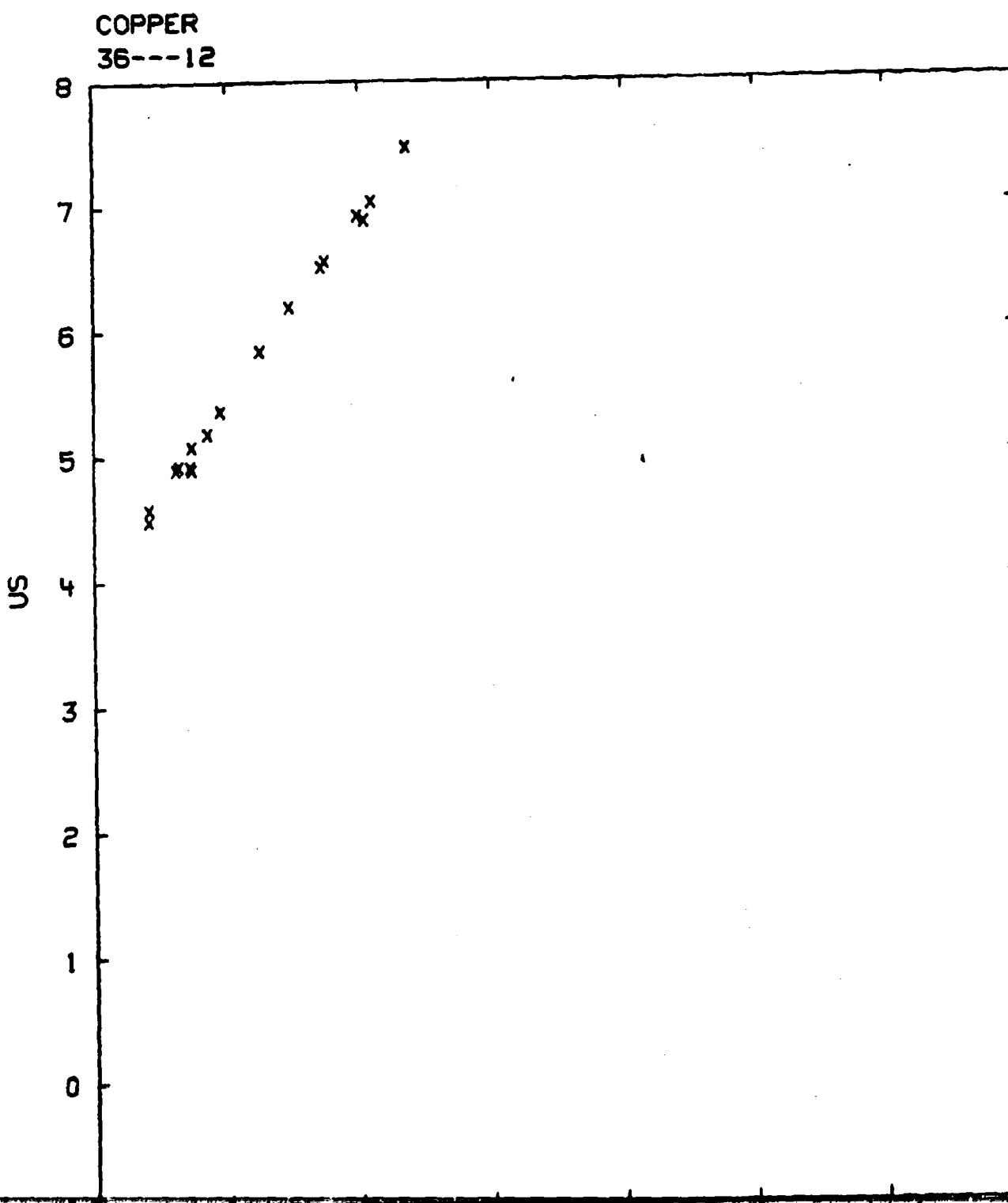


TABLE II

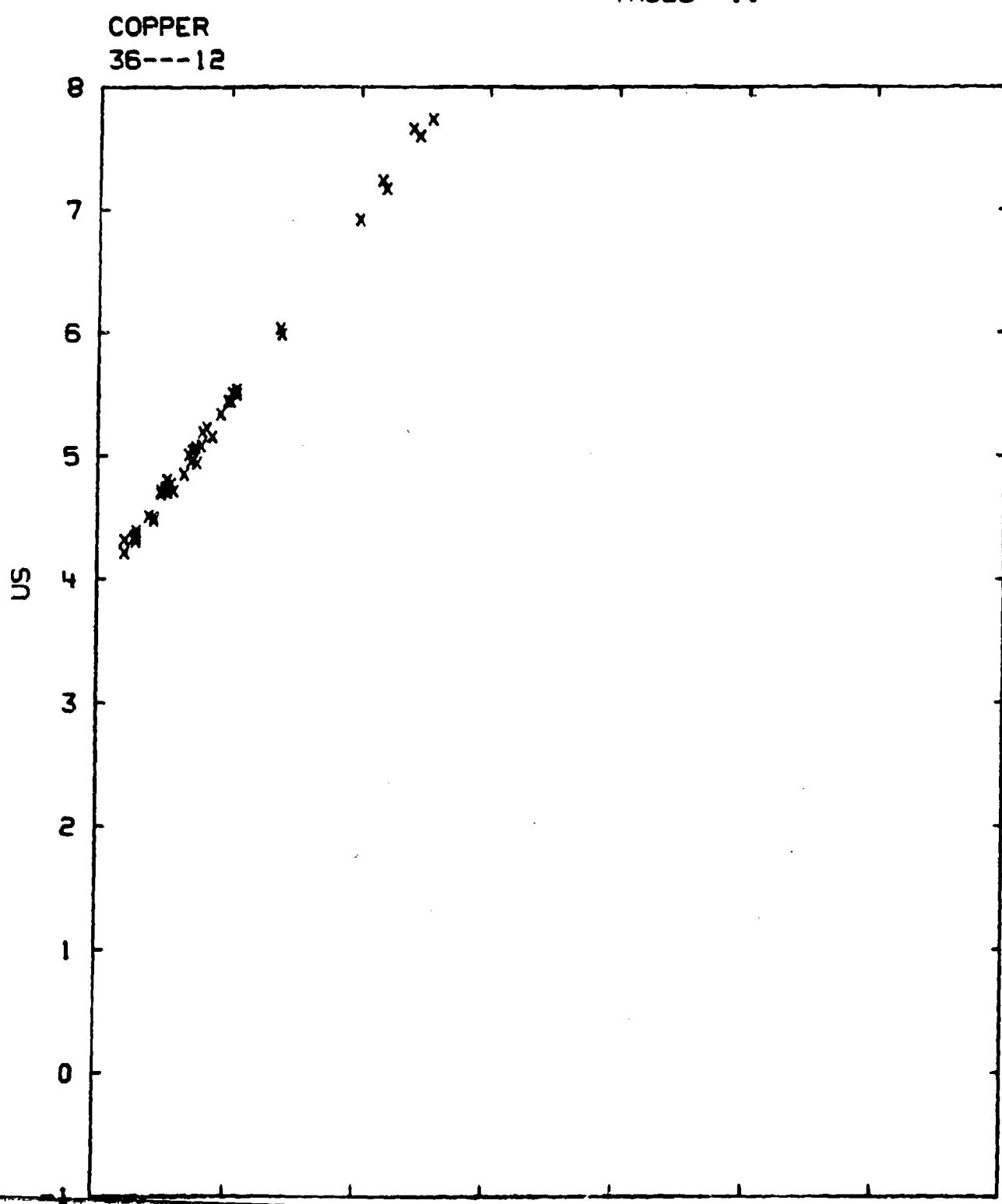


TABLE III

COPPER

36---12

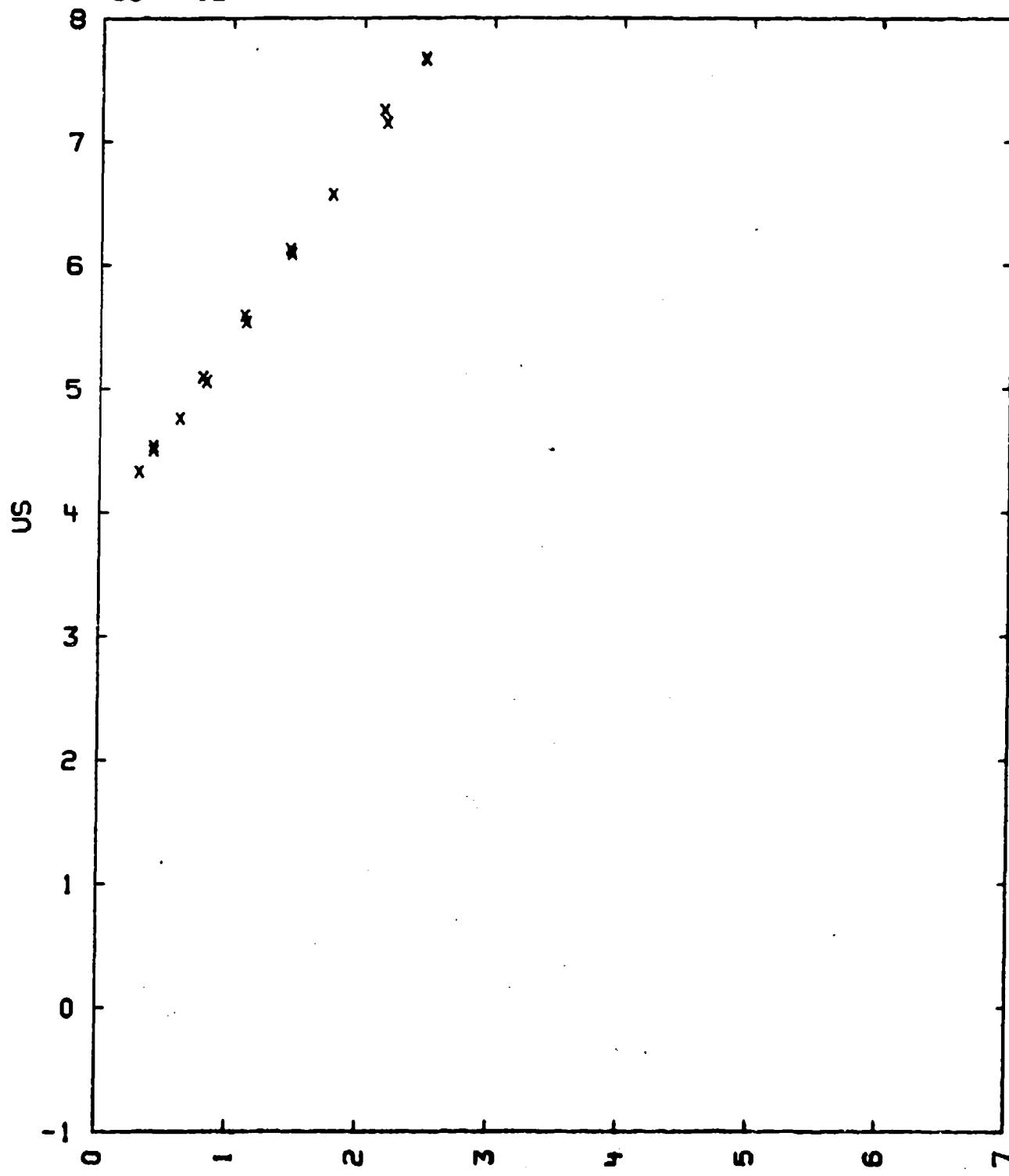


TABLE IV

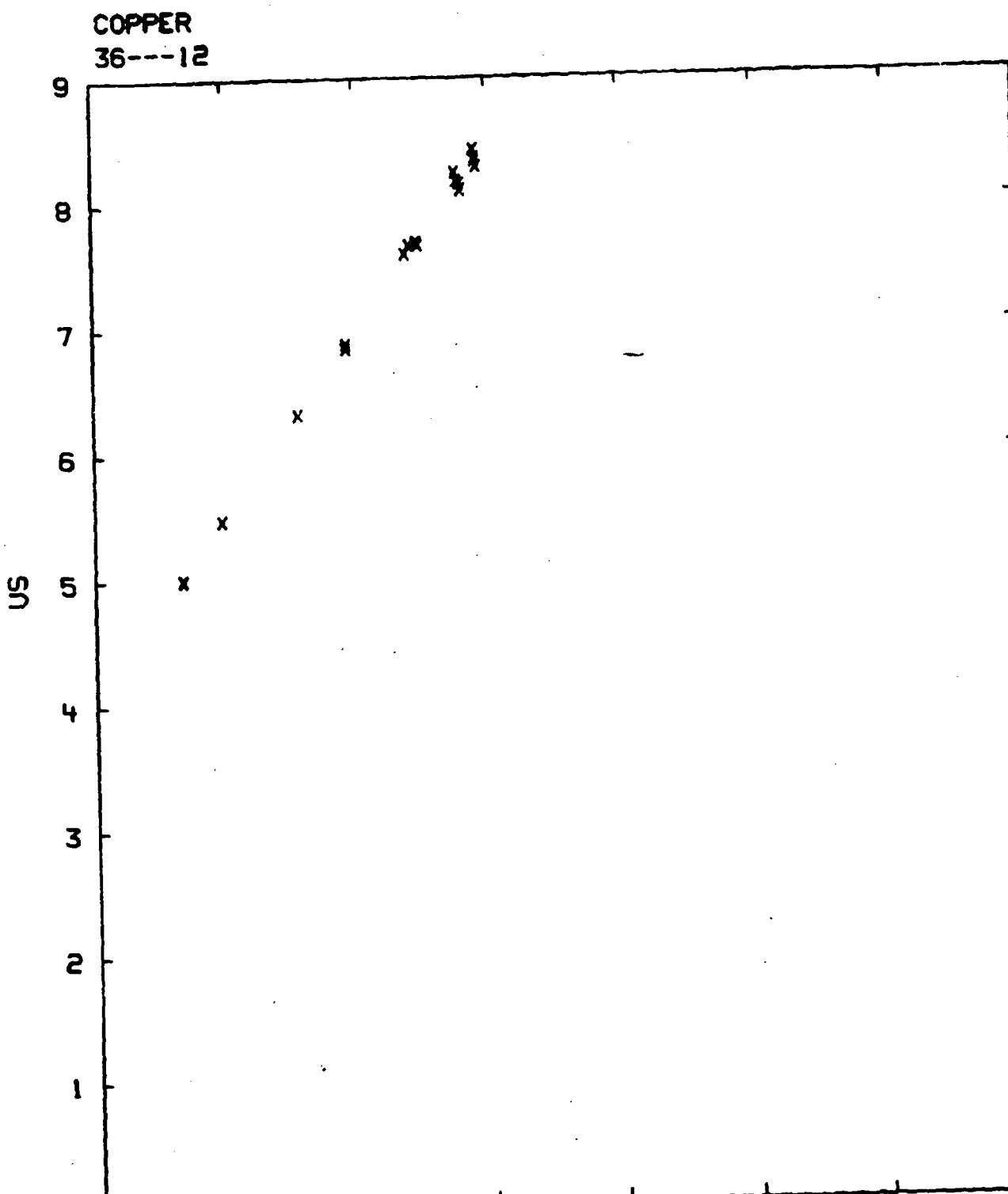


TABLE V

COPPER
36---12

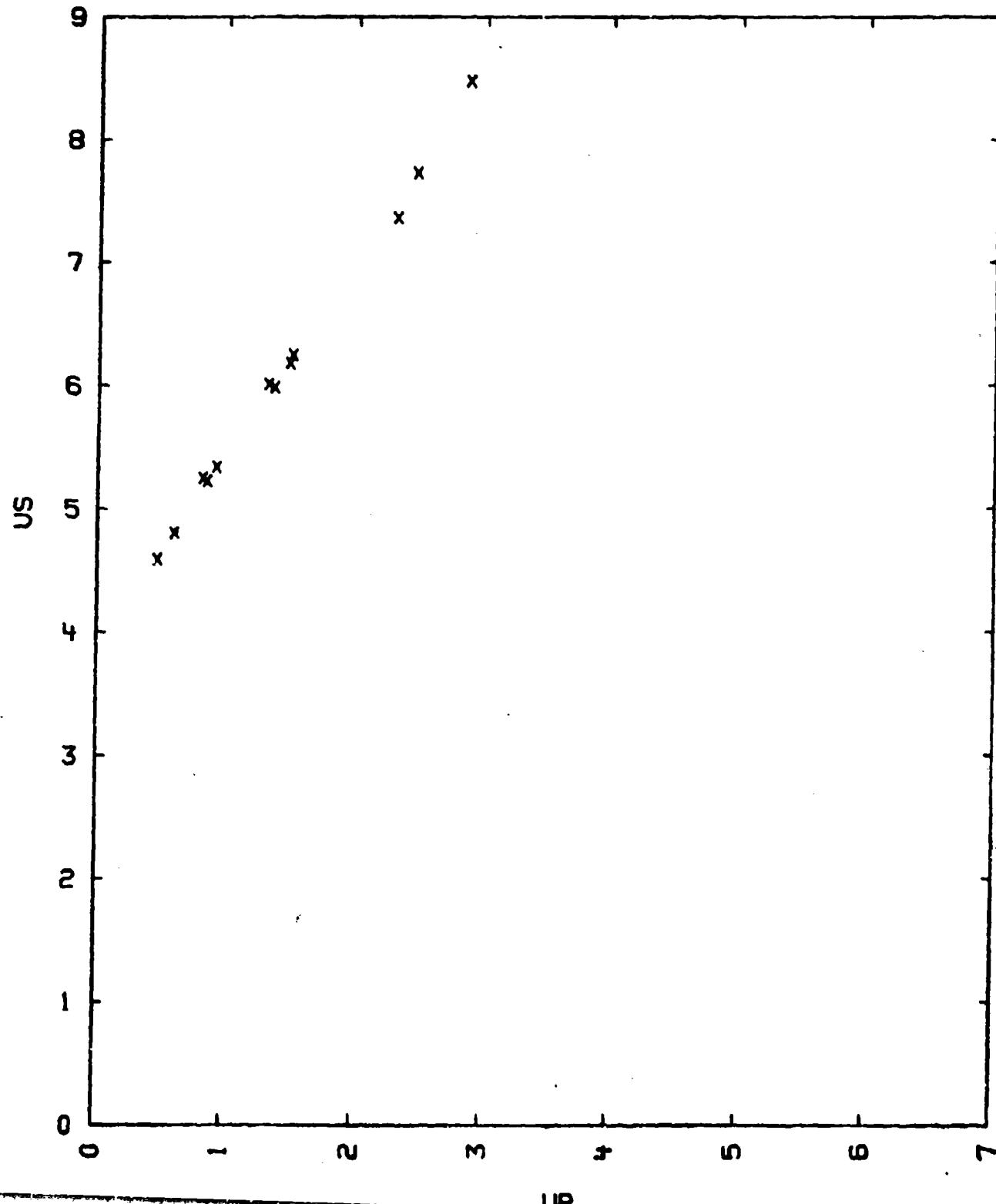


TABLE VI

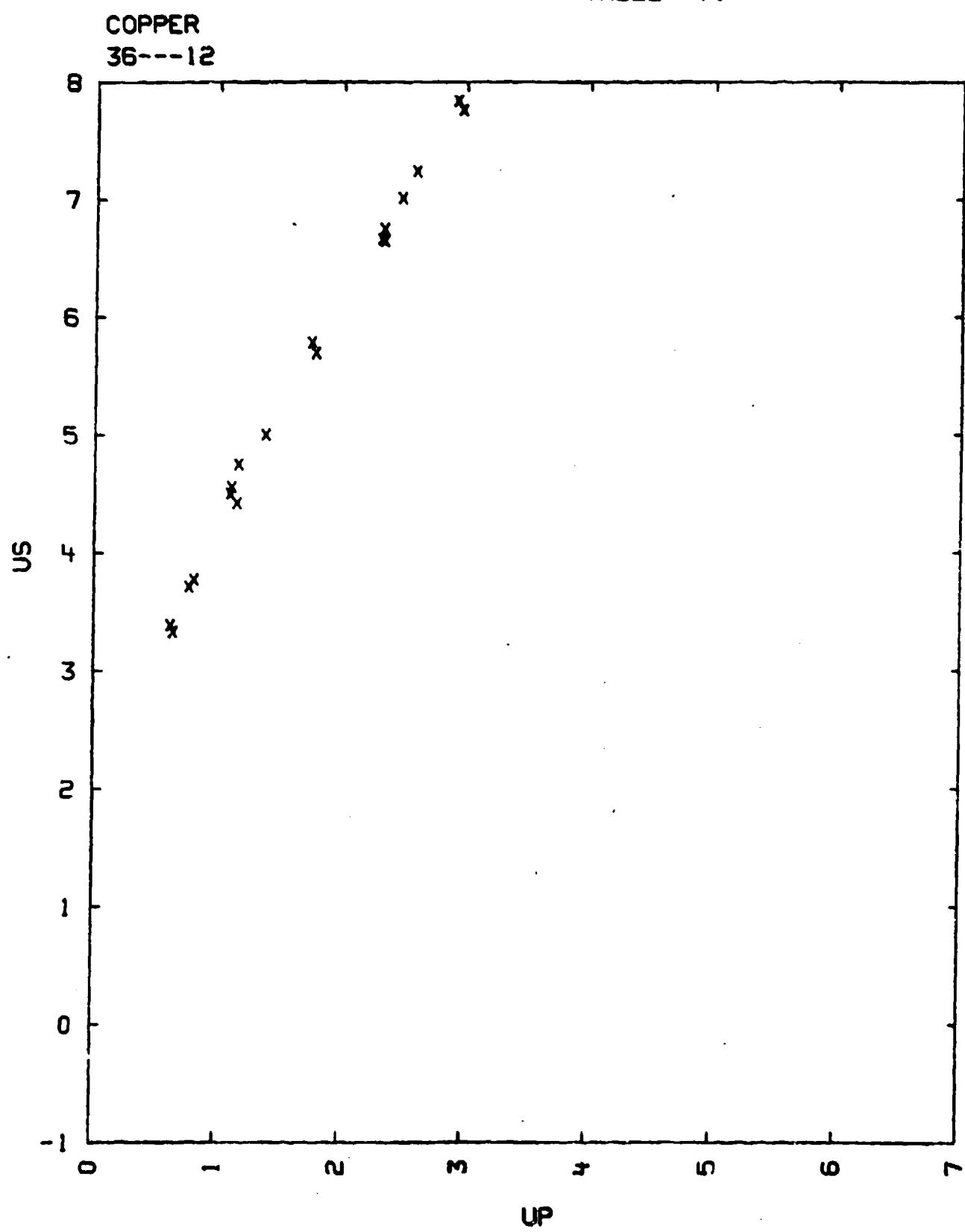


TABLE VII

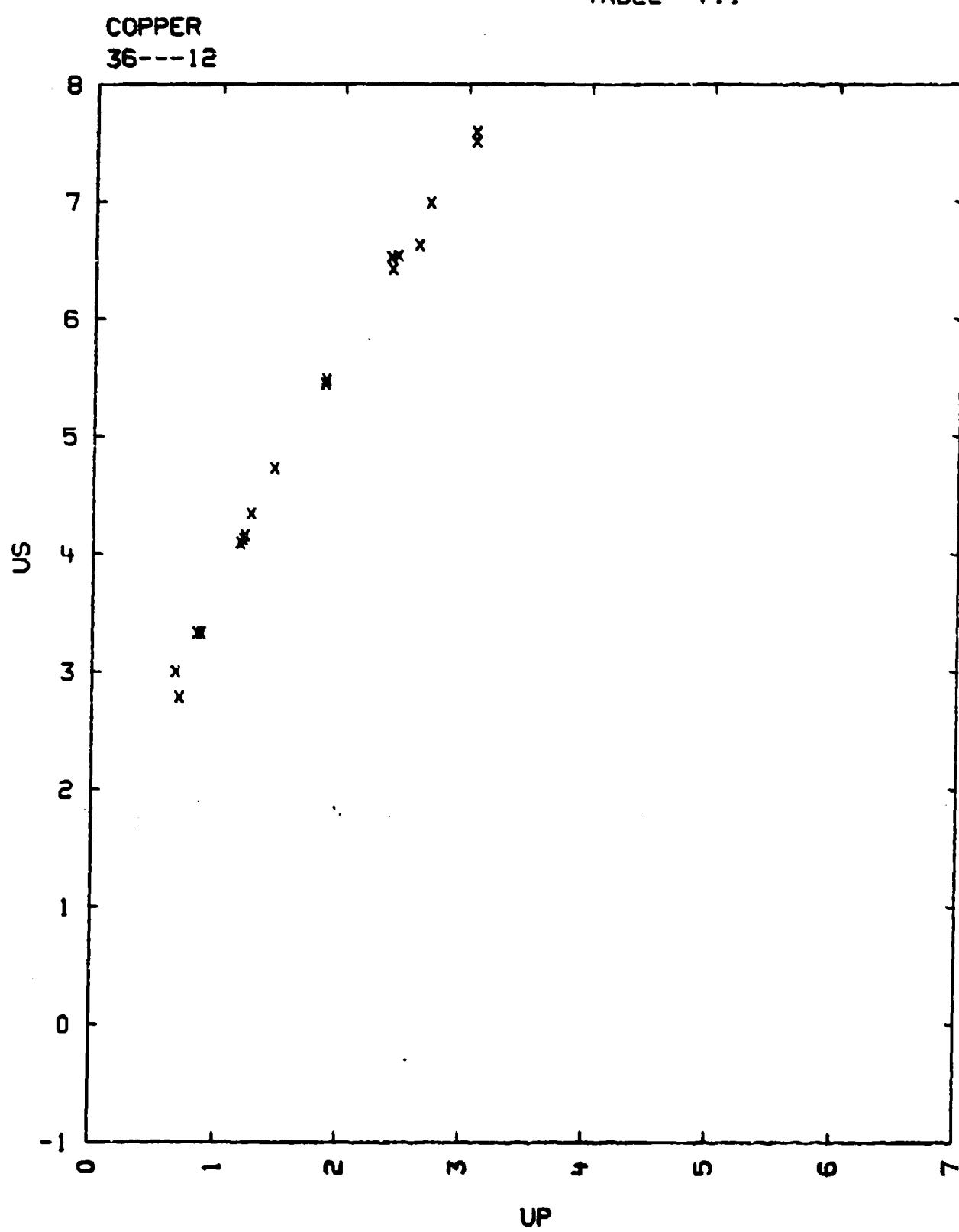


TABLE VIII

COPPER
36---12

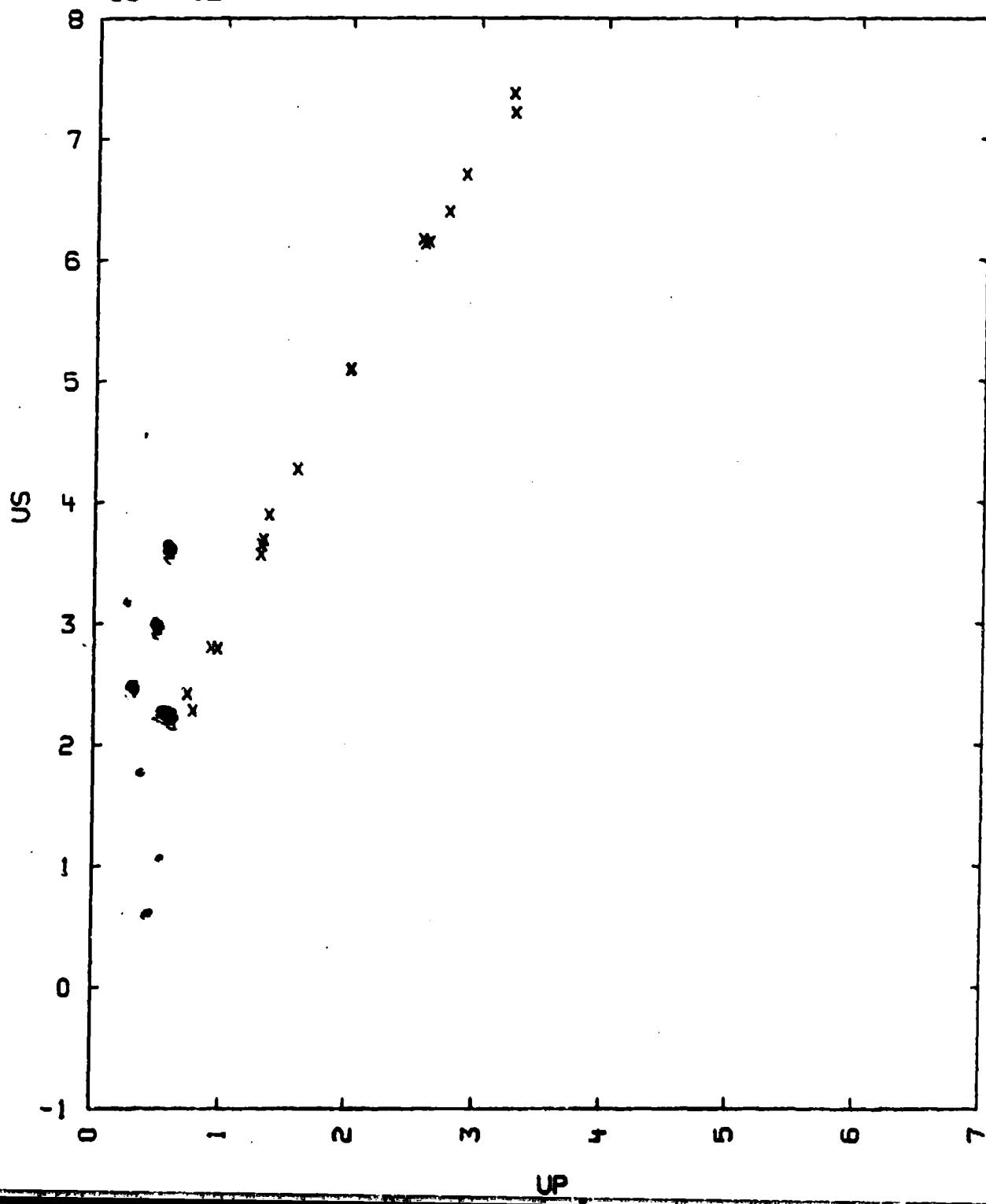


TABLE IX

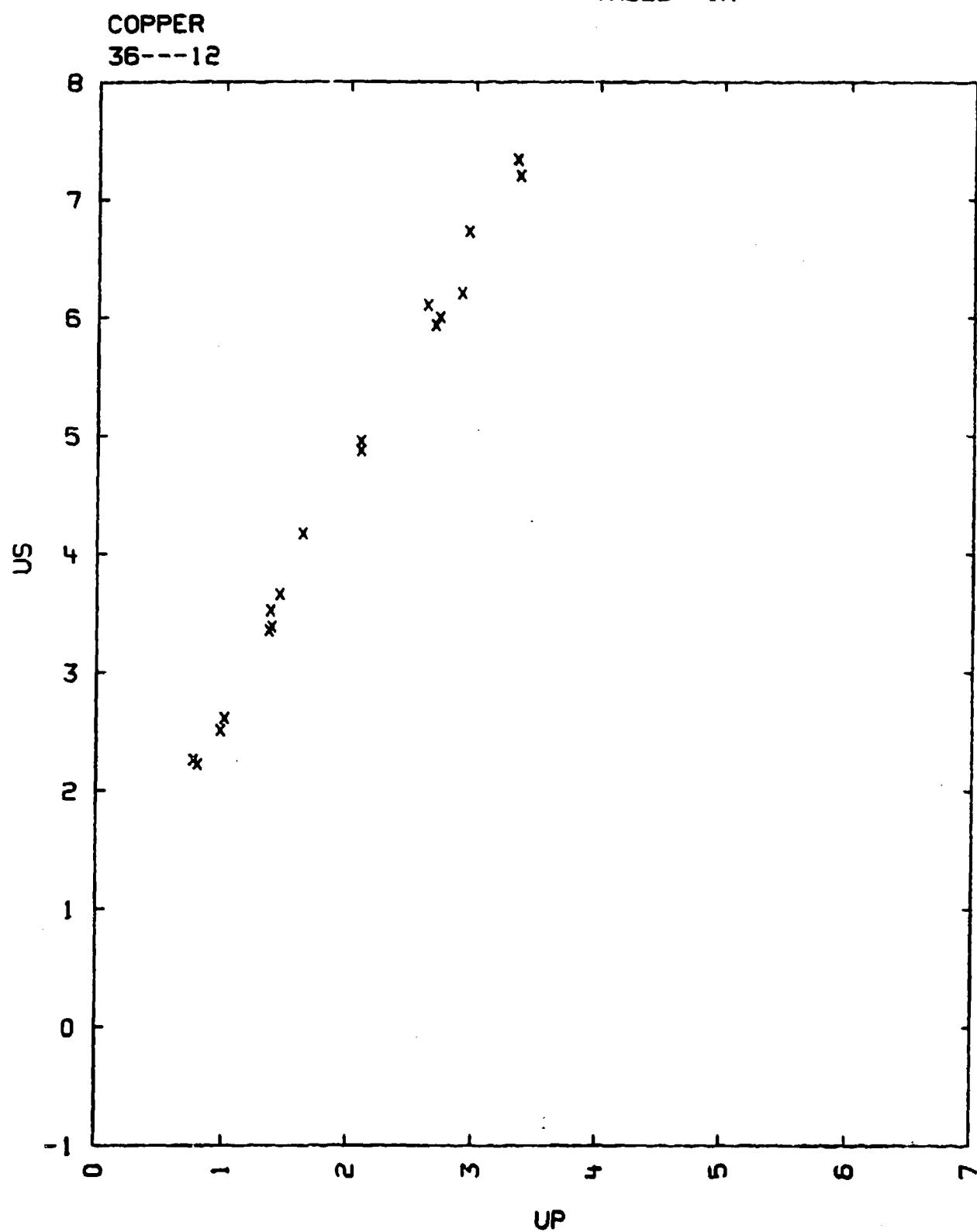
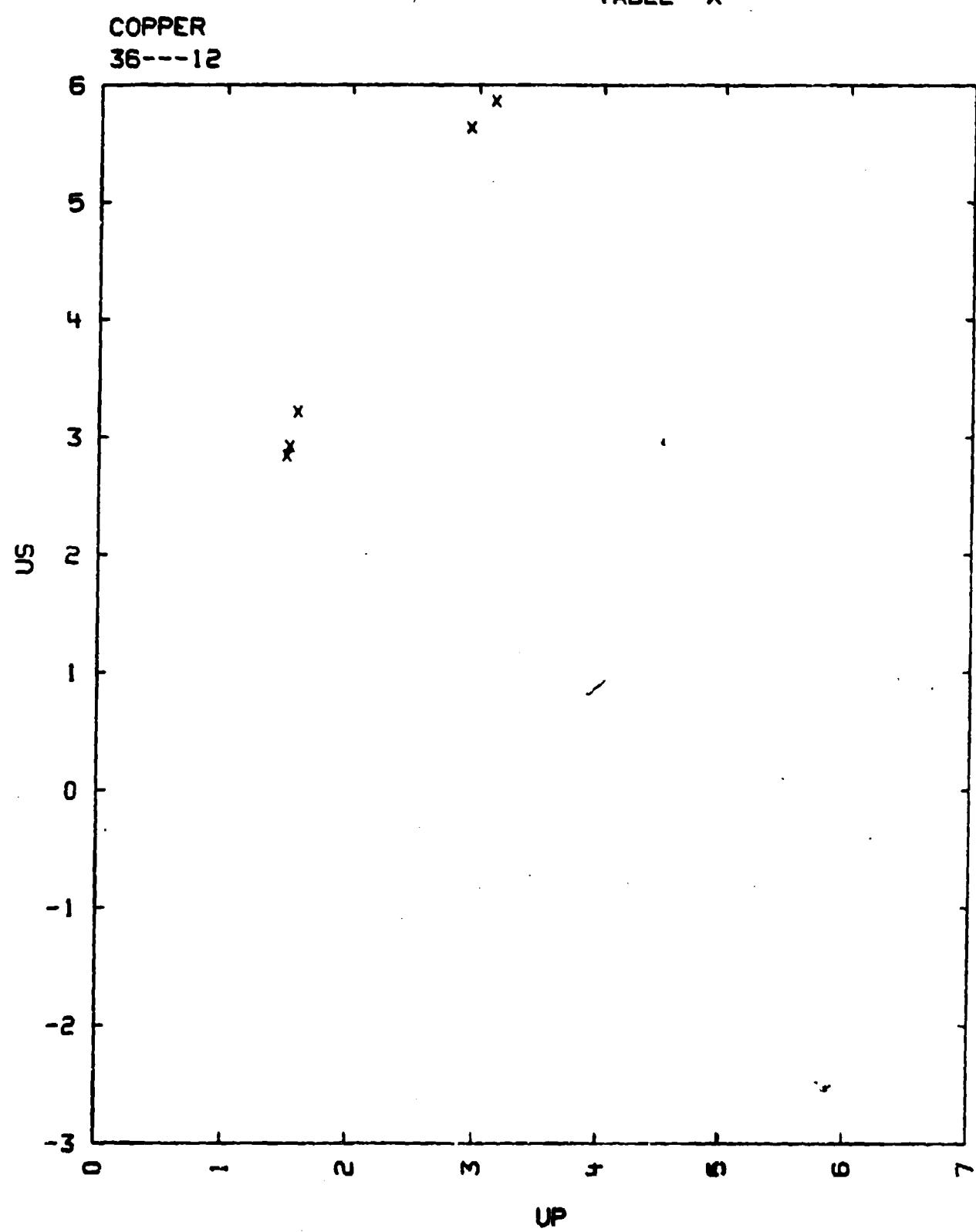


TABLE X



37---0
SILVER SUMMARY

AG

$$V_0 = 0.09533 \text{ CC/G} \quad C_L = 3.80 \text{ KM/SEC} \quad C_0 = 3.09 \text{ KM/SEC}$$

$$V_{01} = 0.09533 \text{ CC/G} \quad C_S = 1.59 \text{ KM/SEC} \quad C_B = 3.19 \text{ KM/SEC}$$

THE TABLE LISTS THE HUGONIOT POINTS CALCULATED FROM THE POLYNOMIAL FIT GIVEN BELOW. UNITS ARE: G/CC, KM/SEC, KILOBARS AND KBAR.CC/G FOR THE ENERGY DIFFERENCE.

TABLE

RHO0	US	UP	P	V/V0	E-E0
10.49	4.037	0.5	212	0.876	1.25
-	4.842	1.0	508	0.793	5.0
-	5.633	1.5	886	0.734	11.25
-	6.411	2.0	1345	0.688	20.0
-	7.928	3.0	2495	0.622	45.0
-	9.393	4.0	3941	0.574	80.0

$$US = 3.220 + 1.648 \cdot UP - 0.0262 \cdot UP^2, \quad SIG.US = 0.06 \text{ KM/SEC}$$

FOR UP BETWEEN 0.5 AND 4 KM/SEC

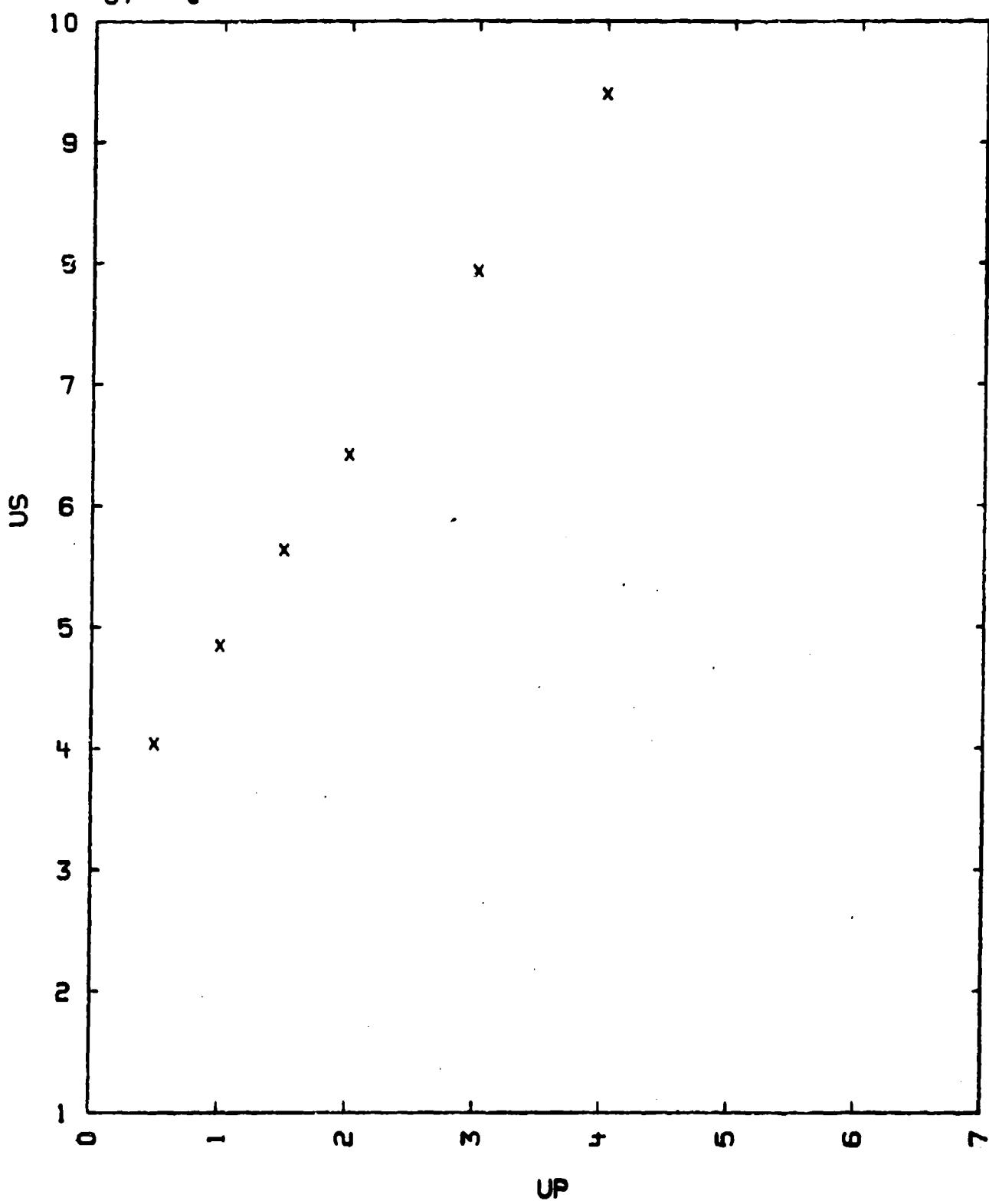
COMMENTS

- 1) SOURCE: COMPILER
- 2) THE QUADRATIC TERM IN THE FIT IS CAUSED BY THE SINGLE HIGHEST TV80L1B D000 OUTPUT..... 12:15:52U 06/14
THE FIT WAS MADE FROM THE DATA OF PAGE 37---1, 2 AND 3.
- 3) MORE RECENT SOUND VELOCITY DATA THAN THOSE LISTED ABOVE FROM PAGE 37---1 YIELD A VALUE OF $C_0 = 3.14 \text{ KM/SEC}$:
W. B. DANIELS AND C. S. SMITH, PHYS. REV., VOL. III, P. 713 (1958).

TABLE I

SILVER SUMMARY

37---0



37---I
SILVER

AG 99.8 PER CENT OR GREATER

$V_0 = 0.09533 \text{ CC}/\text{O}$. $C_L = 3.60 \text{ KM}/\text{SEC}$. $C_0 = 3.09 \text{ KM}/\text{SEC}$
 $V_{01} = 0.09533 \text{ CC}/\text{O}$. $C_S = 1.59 \text{ KM}/\text{SEC}$.

IN THE TABLE BELOW, VELOCITIES ARE GIVEN MM/MICROSEC.
PRESSURES IN KILOBARS, AND DENSITY IN G/CC.

TABLE

RHO0	US	UFS	UP	P	V/V0
10.49	4.065	1.015	0.504	214.9	0.8760
-	4.113	1.049	0.527	227.4	0.8719
-	4.378	1.448	0.717	329.3	0.8362
-	4.846	2.041	0.985	500.7	0.7967
-	4.848	2.074	1.010	513.6	0.7917

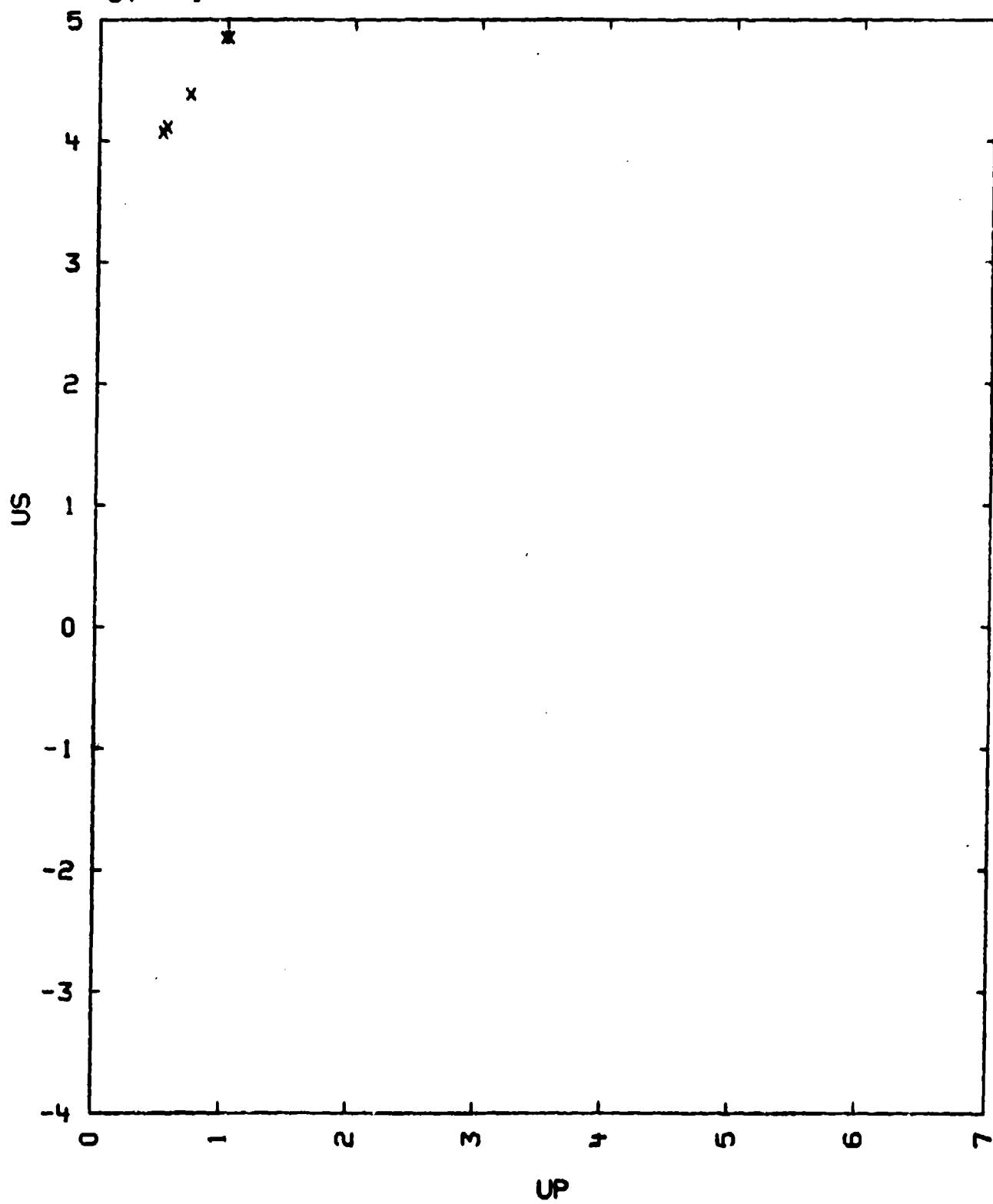
$US = 3.271 + 1.575 UP \text{ KM}/\text{SEC}$ SIGMA US = 0.5 PERCENT

COMMENTS:

- 1) SOURCE: HALSH, J. M., RICE, M. H., MCQUEEN, R. O. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 168 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE OBTAINED BY SPECTROCHEMICAL ANALYSIS
- 5) CL AND CS WERE OBTAINED FROM L. BERGMAN, DER ULTRASCHALL,
S. HIRZEL VERLAG, STUTTOART, 1954, 6TH ED., P. 850

TABLE I

SILVER
37---1



37---2
SILVER

AG

$$V_0 = 0.09532 \text{ CC/G.}$$

$$V_{01} = 0.09532 \text{ CC/G.}$$

$$C_0 = 0.319 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURES IN KILOBARS, AND DENSITY IN G/CC.

TABLE

RHO	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
10.49	5.98	1.77	1107	0.705	1027
-	5.98	1.78	1109	0.702	1034
-	6.73	2.14	1512	0.681	1378
-	6.63	2.17	1509	0.673	1388
-	6.68	2.16	1510	0.677	1393
-	6.72	2.17	1530	0.677	1399

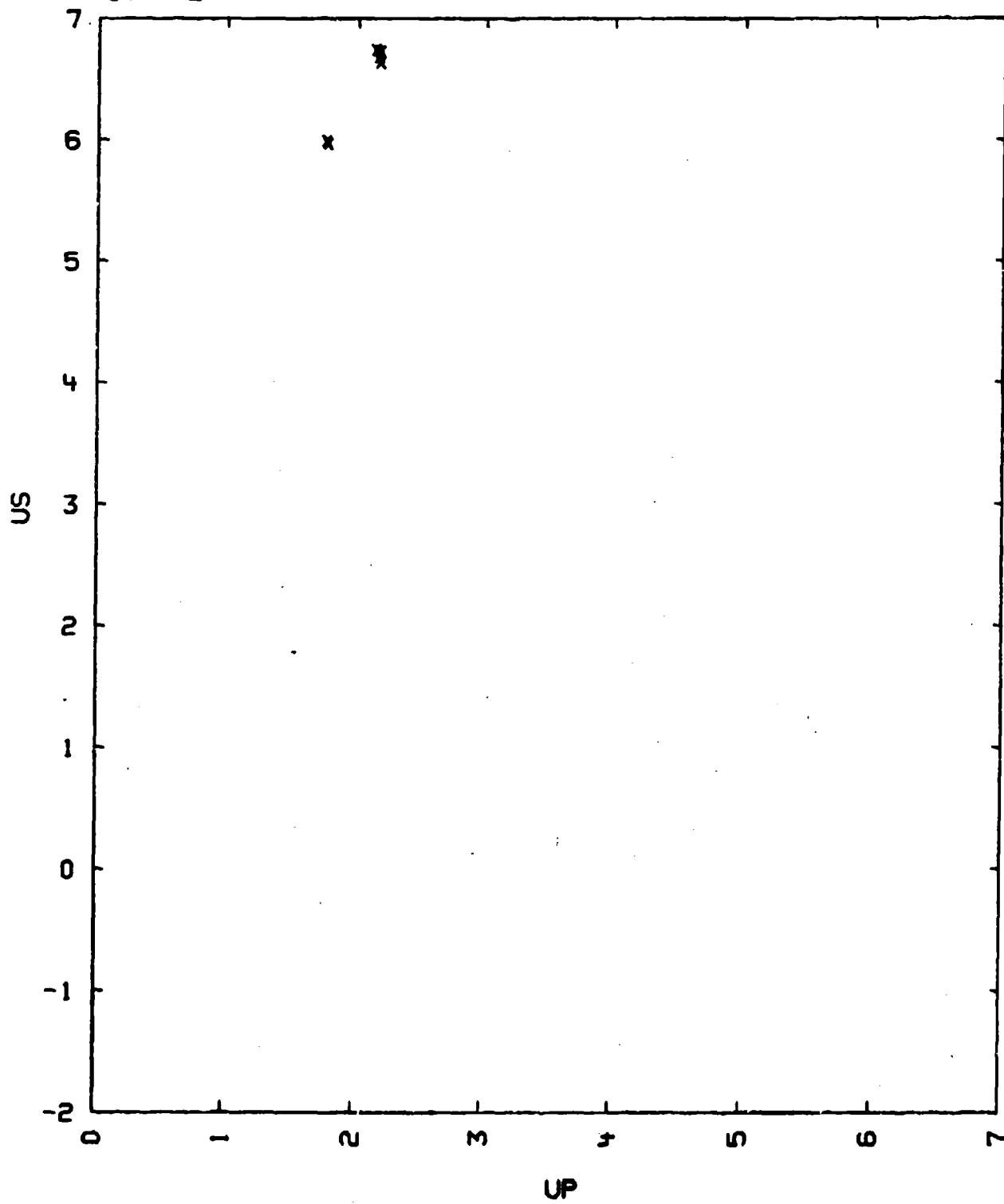
$$US = 3.243 + 1.586 UP \text{ KM/SEC.} \quad SIGMA US = 0.55 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R. G. AND MARSH, S. P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, OMX-8, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 4) V01 WAS TAKEN FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK
(MCGRAW-HILL BOOK CO., N.Y., 1963) 2ND ED.

TABLE I

SILVER
37---2



37---3
SILVER

AG

$$V_0 = 0.09533 \text{ CC/G.}$$

$$V_{01} = 0.09533 \text{ CC/G.}$$

$$C_B = 3.14 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC.,
PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0
10.49	4.69	0.93	460	0.800
-	6.76	2.19	1550	0.576
-	9.45	4.05	4010	0.571

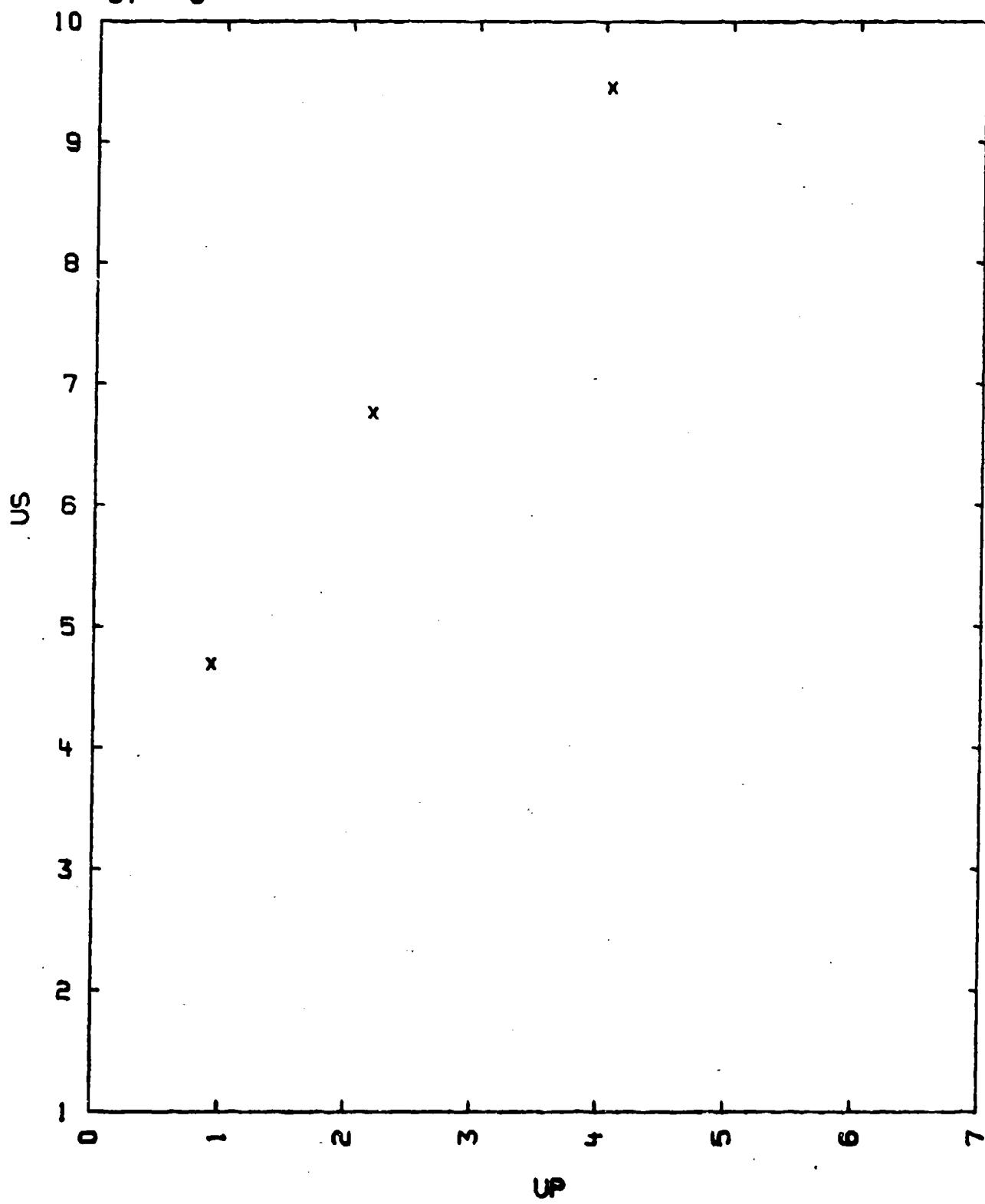
$$US = 3.30 + 1.54UP \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: AL'TSHULER, L. V., KRUPNIKOV, K. K. AND BRAZHNIK, M. I.
SOVIET PHYS.-JETP, VOL. 34, P. 614 (1958)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL Fe.
- 3) CB IS CALCULATED USING THE VOLUME COMPRESSION COEFFICIENT TAKEN FROM
P. W. BRIDGMAN, THE PHYSICS OF HIGH PRESSURE, (G. BELL AND SONS LTD.
LONDON 1958)
- 4) V01 WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK.
(MCGRAW-HILL BOOK CO., N.Y., 1963) 2ND ED.
- 5) THE SPECIMEN WITH A THICKNESS BETWEEN 6-8MM. WAS ATTACHED TO A IRON
PLATE OF COMPARABLE THICKNESS 6-8MM.

TABLE I

SILVER
37---3



38---0
GOLD SUMMARY

AU

$$V_0 = 0.0518 - 0.0520 \text{ CC/G} \quad C_L = 3.24 \text{ KM/SEC} \quad C_0 = 2.93 \text{ KM/SEC}$$

$$V_{01} = 0.05183 \text{ CC/G} \quad C_S = 1.20 \text{ KM/SEC} \quad C_B = 3.05 \text{ KM/SEC}$$

THE TABLE LISTS HUGONIOT POINTS CALCULATED FROM THE FIT GIVEN BELOW.
UNITS ARE: G/CC., KM/SEC., KBARS AND KBAR.CC/G FOR THE ENERGY DIFFERENCE.

TABLE

RHOO	US	UP	P	V/V0	E-E0
19.29	3.576	.3	207.	0.9181	0.45
-	4.032	.6	467.	0.8512	1.8
-	4.489	.9	779.	0.799	4.05
-	4.641	1.	893.	0.784	5.00
-	5.401	1.5	1563.	0.722	11.25
-	6.162	2.0	2377.	0.675	20.0
-	6.922	2.5	3338.	0.639	31.2
-	7.683	3.0	4448.	0.610	45.0
-	8.443	3.5	5701.	0.585	61.2

US = $3.120 + 1.521 \text{ UP}$, SIG.US = 0.056 KM/SEC.
FOR UP BETWEEN 0.33 AND 3.5 KM/SEC

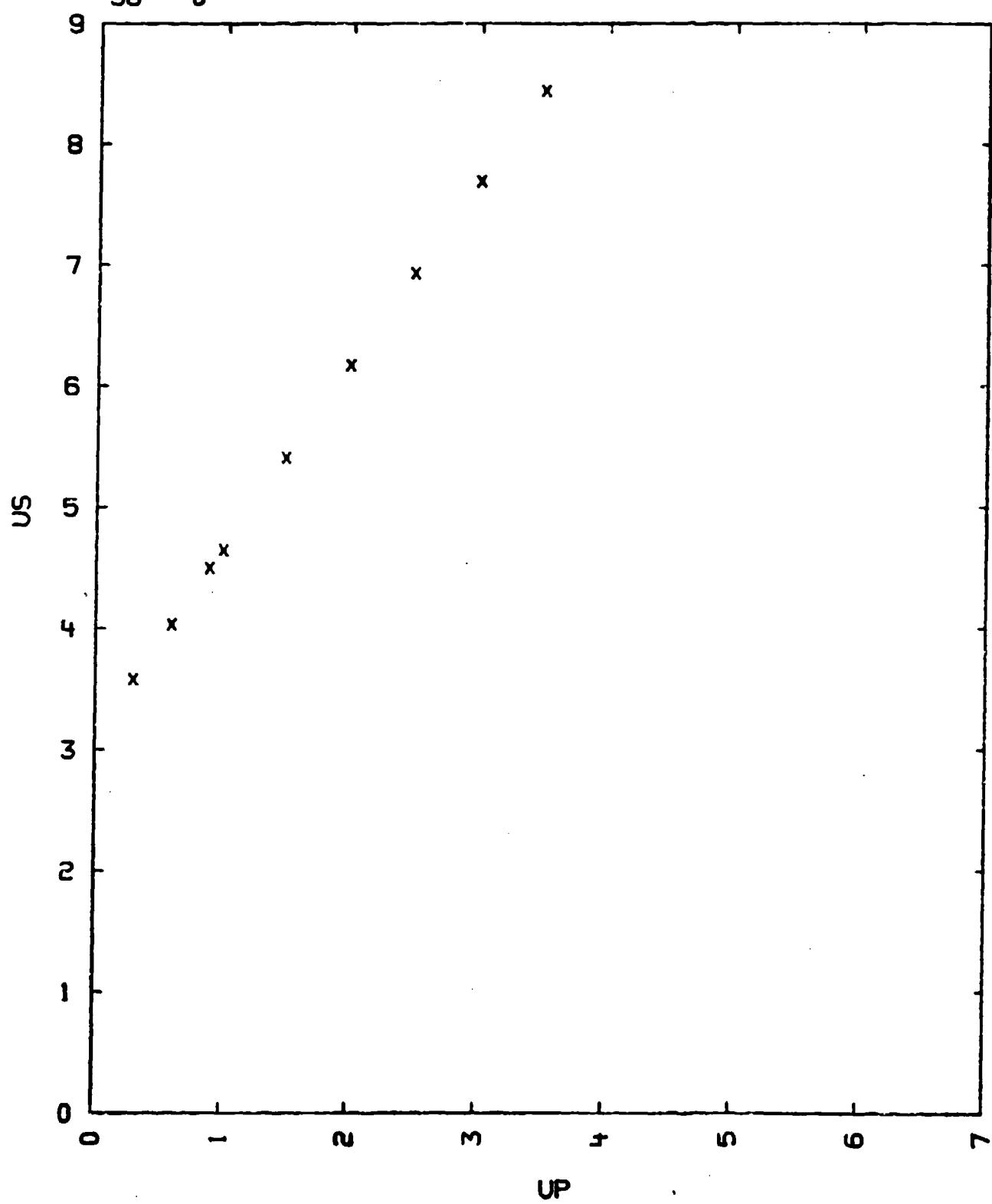
COMMENTS:

- 1) SOURCE: COMPILER
- 1A) THE FIT WAS MADE TO ALL DATA OF 38---1,2,3 AND 4.
- 2) ALL POINTS WERE WEIGHTED EQUALLY
- 3) VOI WAS CALCULATED FROM THE CUBIC UNIT CELL CONSTANT $A = 4.07825$ ANGSTROM AT 25 DEG. C.
HYCKOFF, CRYSTAL STRUCTURES, VOL. I (INTERSCIENCE PUBL. 1963) 2ND ED.
- 3) MORE RECENT SOUND VELOCITY DATA THAN THOSE LISTED ABOVE FROM PAGE 38---1 YIELD A VALUE OF $C_0 = 2.99 \text{ KM/SEC}$:
W. B. DANIELS AND C. S. SMITH, PHYS. REV., VOL. III, P. 713 (1938).

TABLE I

GOLD SUMMARY

38---0



38---1
GOLD

AU 99.9 PER CENT OR GREATER

$V_0 = 0.0520 \text{ CC/G.}$ $C_L = 3.24 \text{ KM/SEC.}$ $C_0 = 2.93 \text{ KM/SEC.}$
 $V_{01} = 0.0516 \text{ CC/G.}$ $C_S = 1.20 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

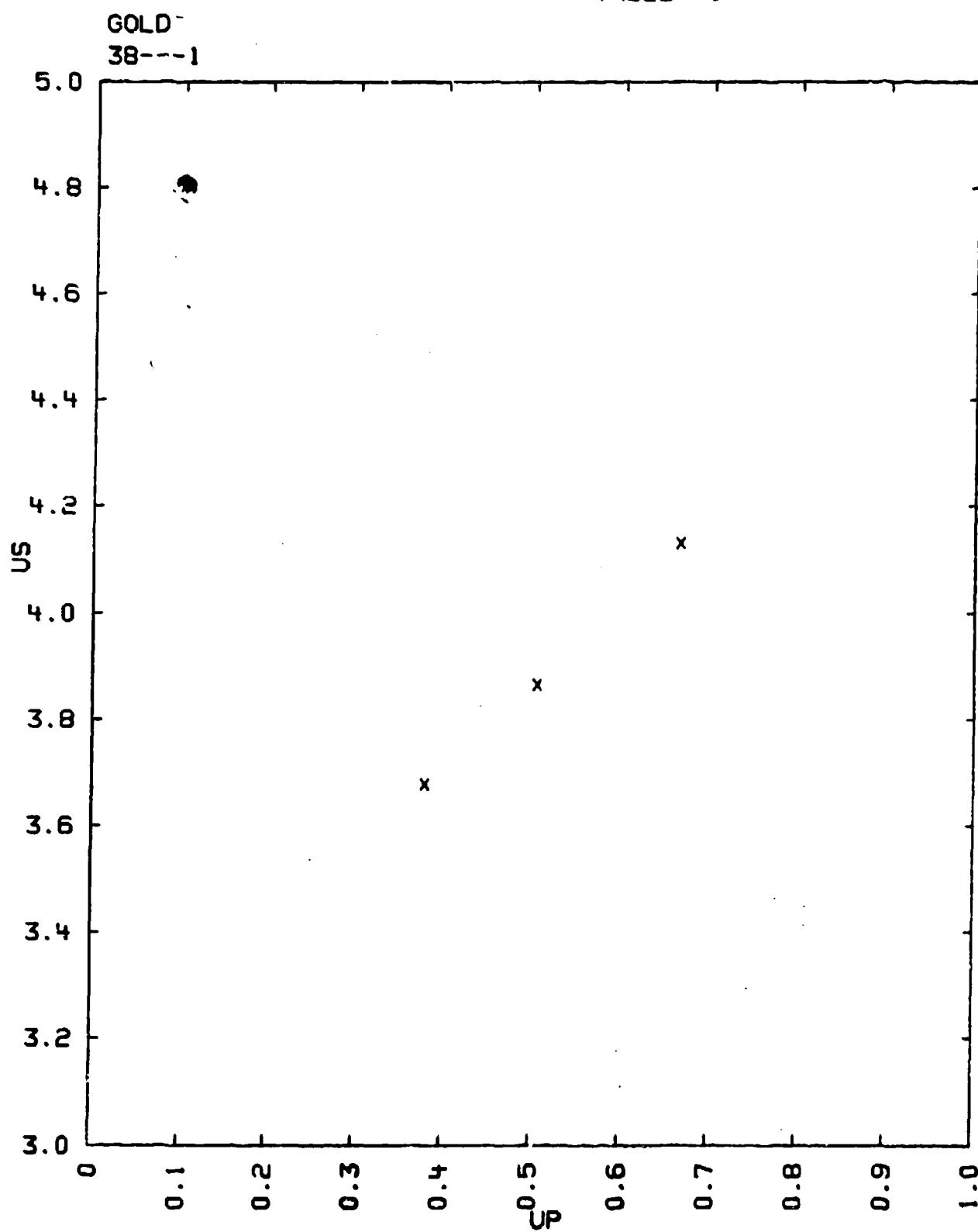
RHO0	US	UFS	UP	P	V/V0
19.24	3.679	0.771	0.380	269.0	0.8967
-	3.864	1.012	0.505	375.4	0.8693
-	4.130	1.375	0.666	529.2	0.8389

$US = 3.074 + 1.580 UP \text{ KM/SEC}$ SIGMA US = 0.3 PERCENT

COMMENTS:

- 1) SOURCE: WALSH, J.M., RICE, M.H., MCQUEEN, R.G. AND YARGER, F.L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS.
- 4) CL AND CS VALUES WERE OBTAINED FROM L. BERGMAN, DER ULTRASCHALL,
(S. HIRZEL VERLAG, STUTTGART, GERMANY, 1954) 6TH ED., P.650
- 5) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE

TABLE I



3B---2
GOLD

AU

$$V_0 = 0.0520 \text{ CC/G.}$$

$$V_{01} = 0.05183 \text{ CC/G.}$$

$$C_0 = 3.05 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURES IN KILOBARS AND DENSITY IN G/CC

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
------	----	----	---	------	-------------------------------

19.24	5.25	1.37	1387	0.739	1002
-	5.21	1.41	1416	0.730	1025
-	5.80	1.73	1932	0.702	1384
-	5.78	1.74	1931	0.700	1387
-	5.78	1.74	1936	0.699	1398
-	5.79	1.74	1942	0.699	1399

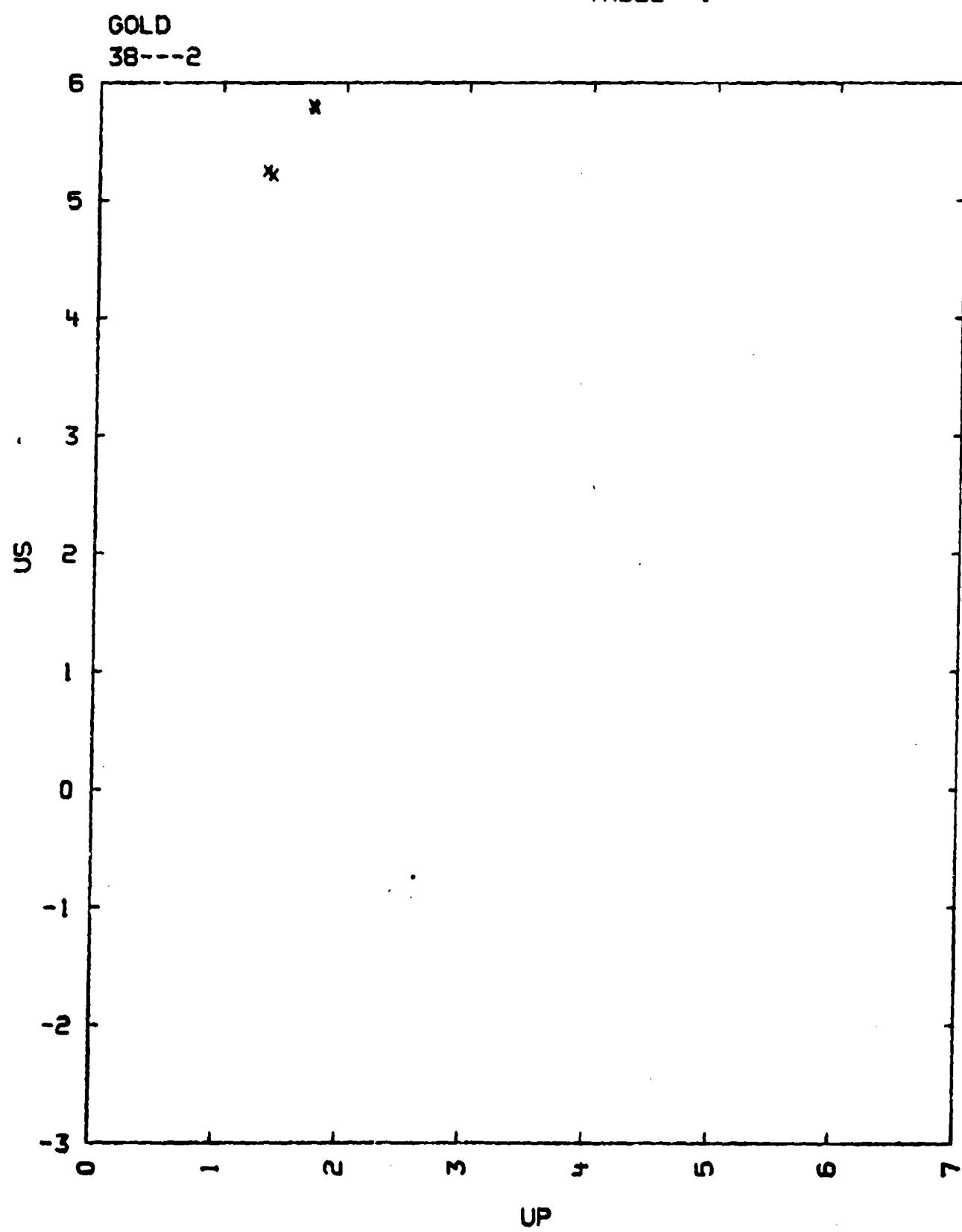
$$US = 3.075 + 1.560 UP \text{ KM/SEC.}$$

$$\text{SIGMA US} = 0.30 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 4) V01 WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW HILL BOOK CO.)
2ND ED.

TABLE I



38---3
GOLD

AU

$$V_0 = 0.0518 \text{ CC/G}$$

$$V_{01} = 0.0518 \text{ CC/G}$$

$$C_B = 3.05 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

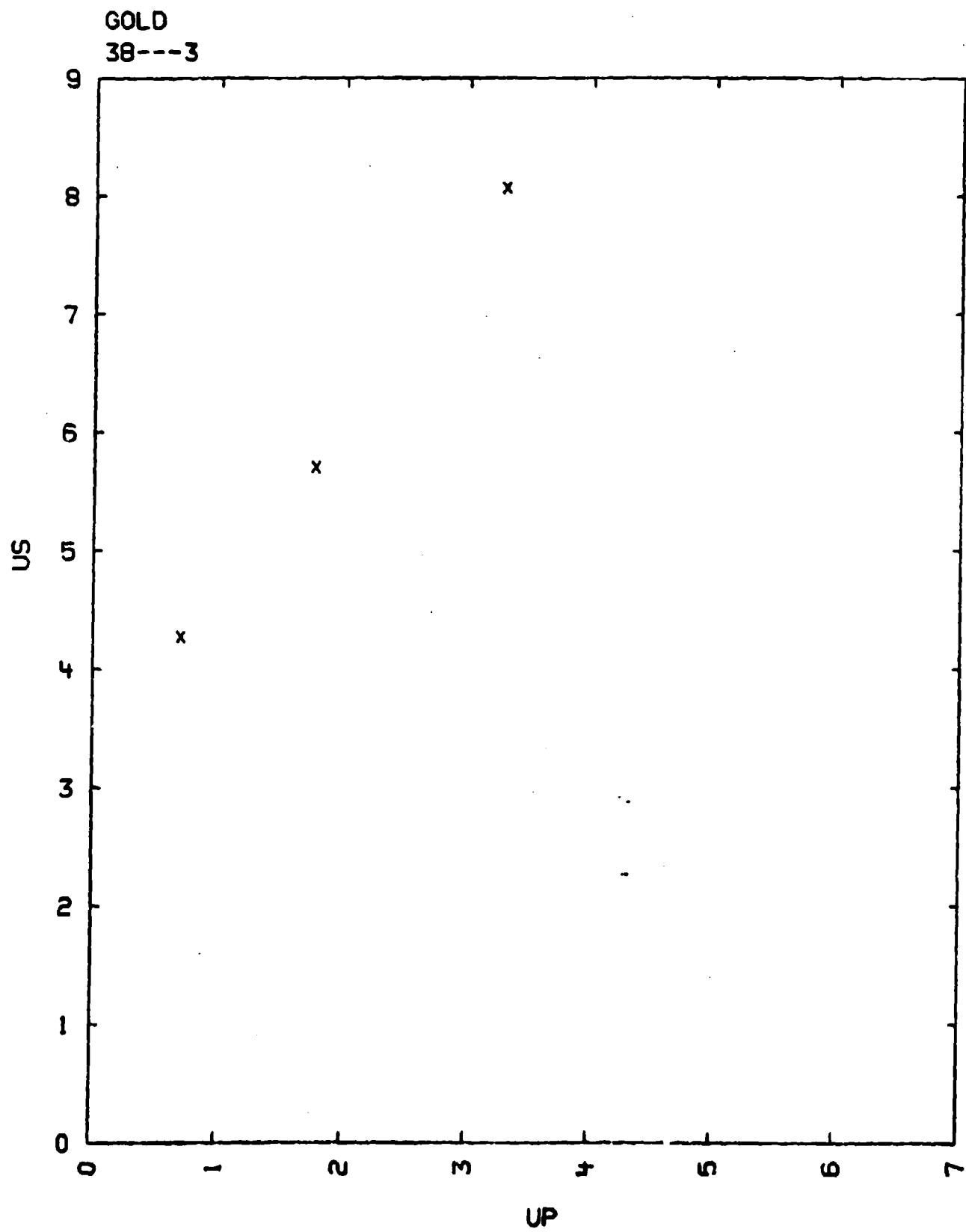
RHO0	US	UP	P	V/V0
19.30	4.27	0.71	590	0.833
-	5.70	1.78	1950	0.690
-	8.06	3.30	5130	0.592

$$US = 3.15 + 1.47UP \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: AL'TSHULER, L. V., KRUPNIKOV, K. K. AND BRAZNIK, M. I.
SOVIET PHYS.-JETP, VOL. 34, P. 614 (1958)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL Fe.
- 3) CB IS CALCULATED USING THE VOLUME COMPRESSION COEFFICIENT TAKEN FROM
P. W. BRIDGMAN, THE PHYSICS OF HIGH PRESSURE. (G. BELL AND SONS LTD.
LONDON 1958)
- 4) VO1 WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK
(MCGRAW-HILL BOOK CO. 1963) 2ND ED.
- 5) THE SPECIMEN WITH A THICKNESS BETWEEN 6-8MM. WAS ATTACHED TO A IRON
PLATE OF COMPARABLE THICKNESS 6-9MM.

TABLE I



38---4
GOLD

AU	99.99	PERCENT
PD	0.0014	-
AG	0.0010	-
CU	0.00017	-
FE	0.0005	-
SI	0.0053	-
MG	0.00038	-

$$V_0 = 0.05178 \text{ CC/G}$$

$$V_{01} = 0.05185 \text{ CC/G.}$$

$$C_0 = 2.99 \text{ KM/SEC}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC., VELOCITIES IN KM/SEC., AND PRESSURE IN KILOBARS.

TABLE

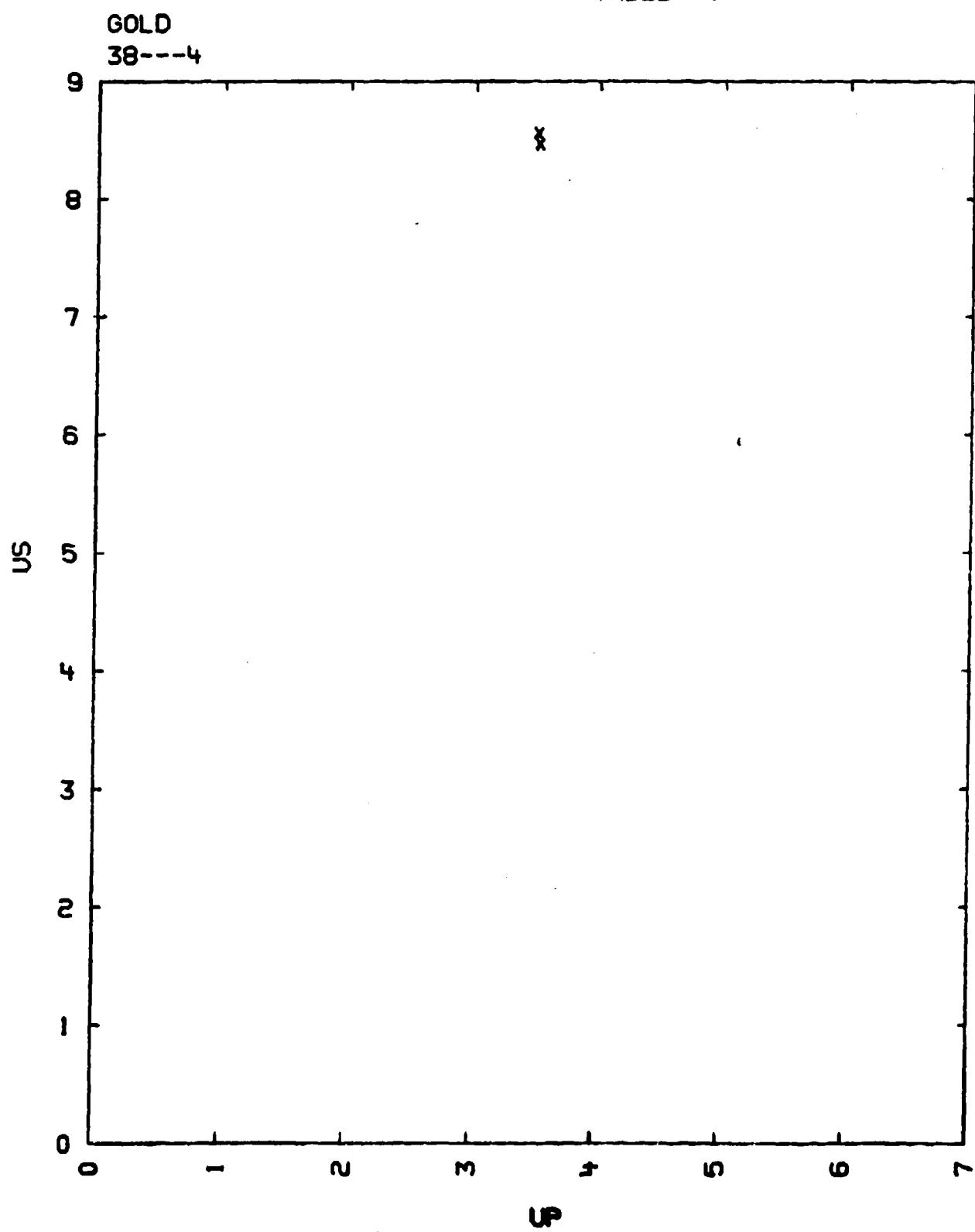
RHO0	US	UP	P	V/V0
19.32	8.46	3.52	5760	0.5854
-	0.55	3.51	5790	0.5895

US *

COMMENTS:

- 1) SOURCE: JONES, A. H., ISBELL, W. M. AND MAIDEN C. J.
JOURNAL OF APPLIED PHYSICS, VOL. 37, P. 3493 (1966)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE A
STANDARD MATERIAL FANSTEEL 77. SEE MATERIAL 53--39--36---1.
- 3) THE DRIVER PLATES WERE ACCELERATED BY MEANS OF A LIGHT GAS GUN.
- 4) THE DRIVER PLATE VELOCITY AND SPATIAL ORIENTATION WAS MEASURED USING TWO PULSE X-RAY STATIONS PRIOR TO IMPACT. THE X-RAY EXPOSURE TIMES WERE 30 NSEC.. THE INTERVAL TIME BETWEEN X-RAY PULSES HAS MEASURED ON A 100-MC/SEC COUNTER.
- 5) FOR BOTH THE DRIVER PLATE IMPACT VELOCITY AND THE SAMPLE SHOCK VELOCITY THE MEASURED EXPERIMENTAL ERROR IS WITHIN 1 PERCENT.
- 6) V01 WAS TAKEN FROM THE AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCGRATH-HILL BOOK CO. N.Y., 1963) 2ND ED.
- 7) C0 TAKEN FROM CHANG,Y.A. AND HIMMEL,L. J. APPL. PHYS. V.37 3587 1966

TABLE I



39---1

NICKEL

NI 99.7 PERCENT
 CO 0.1 PERCENT
 MG 0.05 PERCENT
 SI 0.05 PERCENT
 MN 0.05 PERCENT
 FE 0.05 PERCENT

$$V_0 = 0.1129 \text{ CC/G.} \quad C_L = 5.63 \text{ KM/SEC.} \quad C_D = 4.47 \text{ KM/SEC.}$$

$$V_{01} = 0.1123 \text{ CC/G.} \quad C_S = 2.96 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
 PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

RHO0	US	UFS	UP	P	V1/V0
8.86	5.417	0.981	0.490	235.0	0.9095
-	5.853	1.350	0.678	339.4	0.8801
-	5.620	1.390	0.687	341.8	0.8778
-	6.031	1.955	0.957	511.0	0.8413
-	5.969	1.981	0.982	519.0	0.8355
-	5.952	1.835	0.887	467.4	0.8510

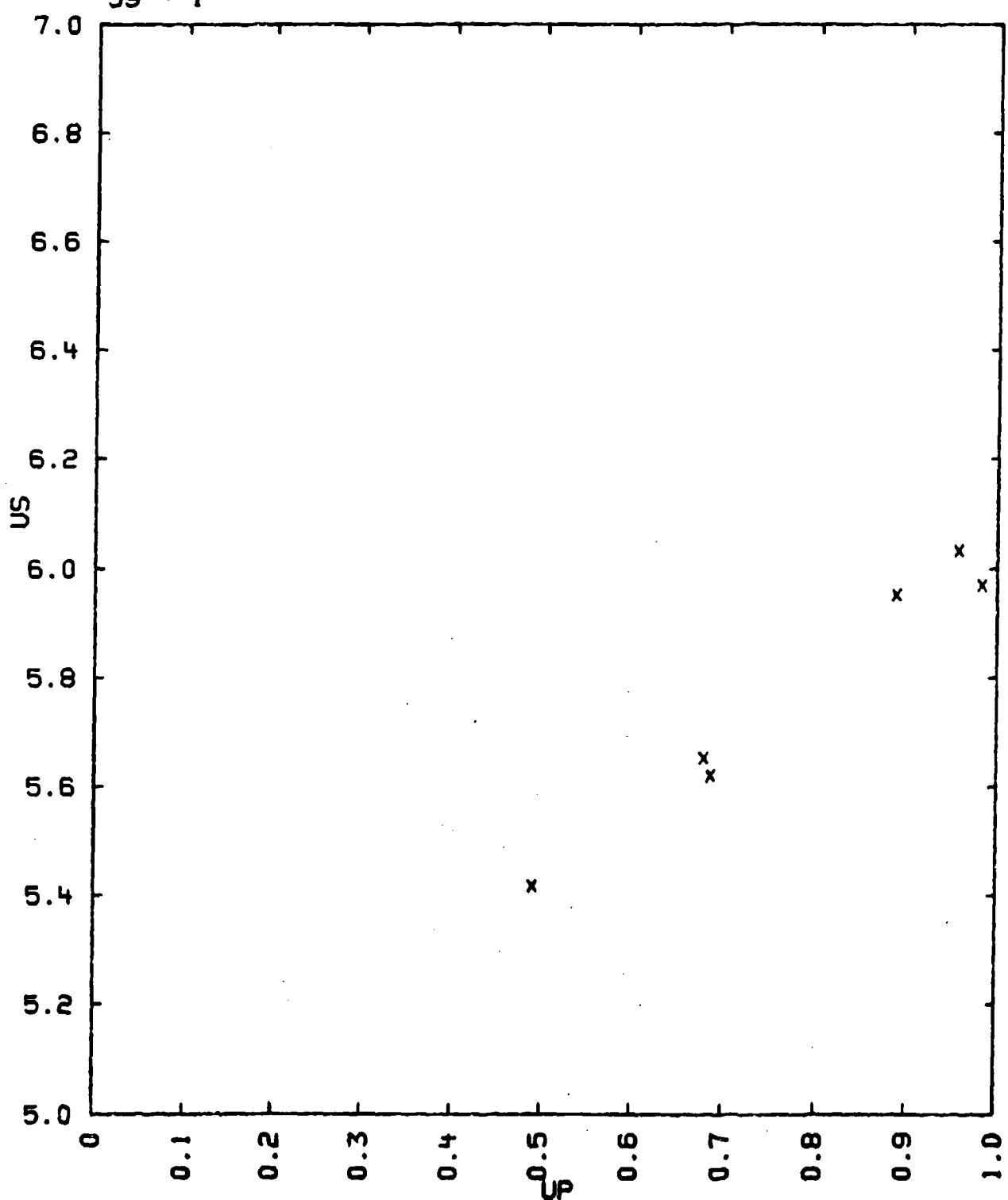
$$US = 4.796 + 1.254 UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.8 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
 PHYS. REV., VOL. 108, P. 189 FF. (1957)
 LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE &
 DATA REDUCTION TECHNIQUE &
 STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
 FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
 COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS
- 5) CL AND CS WERE OBTAINED FROM L. BERGMAN, DER ULTRASCHALL,
 (S. HIRZEL VERLAG, STUTTGART, GERMANY, 1954) 6TH ED., P. 650

TABLE I

NICKEL
39---1



39---2
NICKEL

NI

$$V_0 = 0.1129 \text{ CC/G.}$$

$$V_{01} = 0.1123 \text{ CC/G.}$$

$$C_0 = 4.63 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
8.86	8.93	1.64	1009	0.784	931
-	6.99	1.64	1014	0.786	931
-	7.11	1.62	1022	0.772	931
-	7.78	2.15	1478	0.724	1359
-	7.78	2.17	1491	0.721	1373
-	7.80	2.16	1490	0.723	1368

$$US = 4.646 + 1.445 UP \text{ KM/SEC.}$$

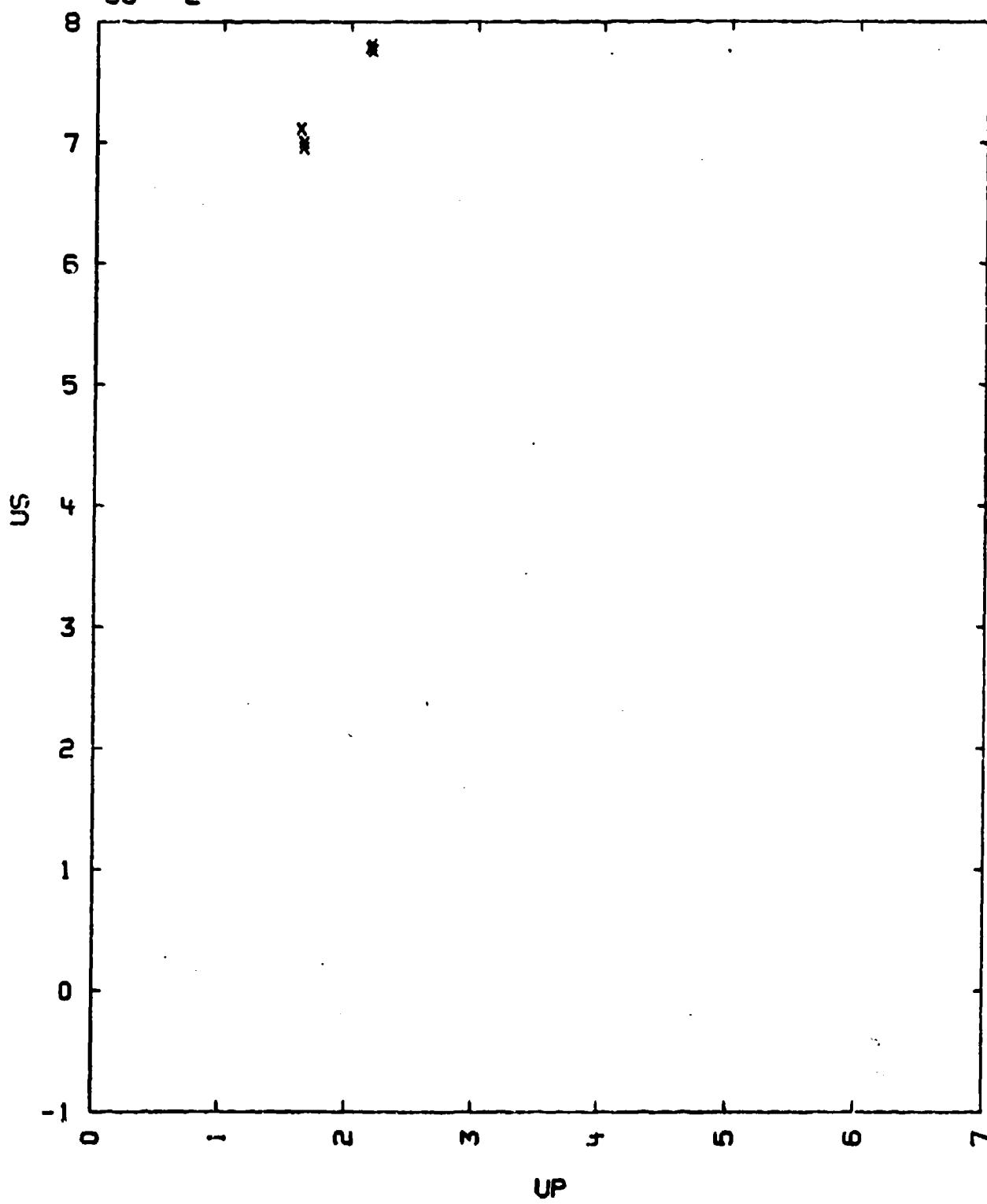
$$\Sigma \text{SIGMA US} = 0.48 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R. G. AND MARSH, S. P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-B, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 38-33
- 4) V01 WAS TAKEN FROM THE AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCGRATH-HILL BOOK CO. N.Y., 1963) 2ND ED.

TABLE I

NICKEL
39---2



39---3
NICKEL

NI

$$V_0 = 0.1129 \text{ CC/G.}$$

$$V_{01} = 0.1122 \text{ CC/G.}$$

$$C_0 = 4.573 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL USED AS STRIKER (PROJECTILE PLATE). U(ST) = STRIKER VELOCITY BEFORE IMPACT.

TABLE

SAMPLE					STRIKER	
RHO0	US	UP	P	V/V0	ST	U(ST)
8.85	7.28	1.72	1116	0.7628	AL	5.60
-	8.72	2.63	2033	0.6983	FE	5.60
-	11.20	4.35	4319	0.6120	FE	9.10
-	14.60	7.09	9180	0.5144	FE	14.68

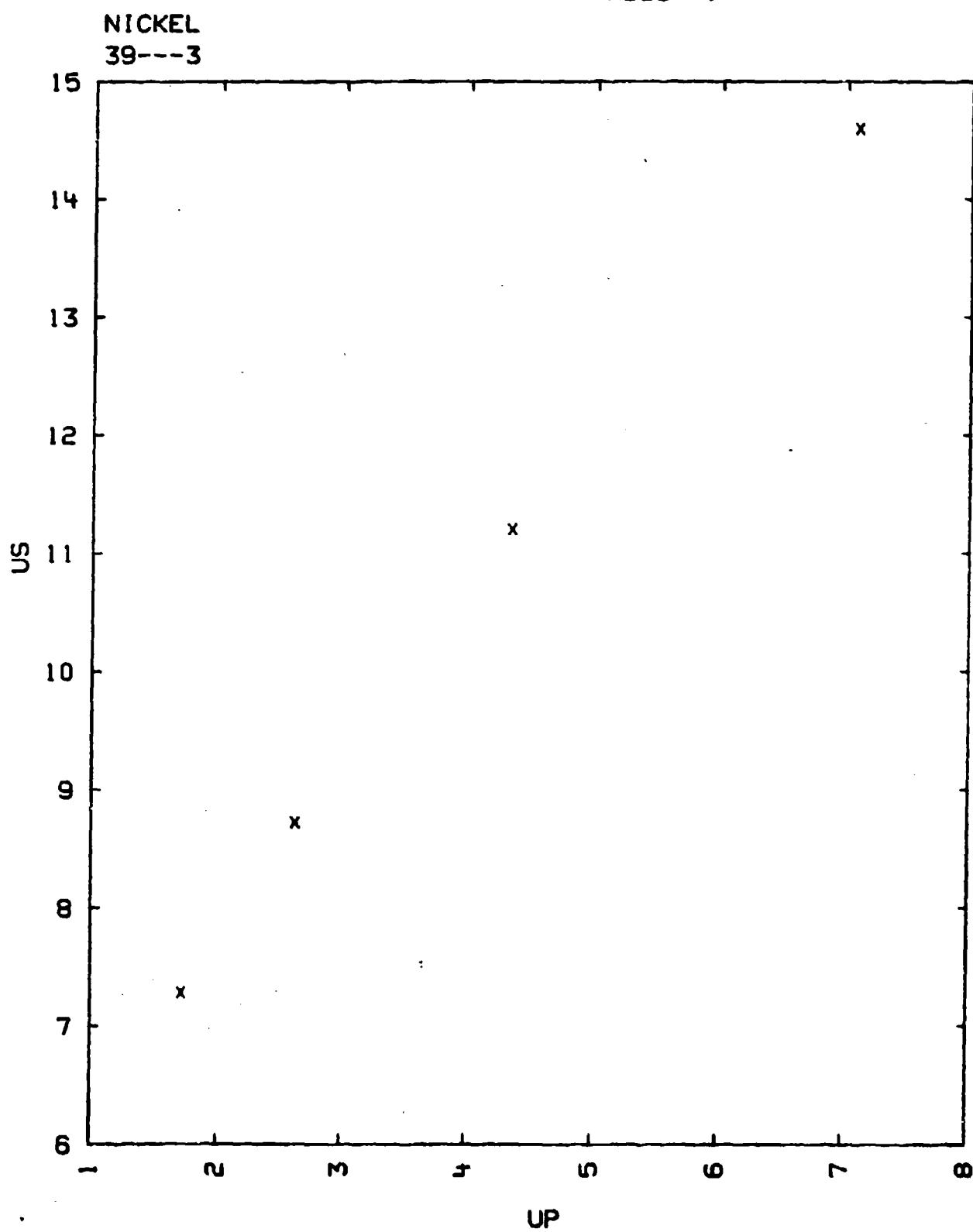
$$US = 4.370 + 1.775 \cdot UP - 0.047 \cdot UP^{0.2} \text{ OR } US = 5.099 + 1.355 \cdot UP \text{ KM/SEC}$$

SIG US = 0.008 SIG US = 0.20

COMMENTS:

- 1) SOURCE: AL'TSHULER, L.V., BAKANOVA, A.A. AND TRUNIN, R.F.
SOVIET PHYS.-JETP, VOL. 15, P. 65 (1962)
J. EXPTL. THEORET. PHYS. (USSR) VOL. 42, P. 91, (1962)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE A AND B
- 3) THE UNCERTAINTIES OF THE VALUES IN THE LAST TABLE ENTRY ARE:
1 PERCENT IN US, OBTAINED FROM 6-8 EXPERIMENTS AND 1.5 PERCENT
IN U(ST) OBTAINED FROM 12 EXPERIMENTS.
- 4) V01 WAS OBTAINED FROM THE CUBIC UNIT CELL WITH $A = 3.52394 \text{ ANGSTROM}$.
AMERICAN INST. OF PHYS. HANDBOOK (MCGRAW-HILL 1963) 2ND ED
- 5) CO IS GIVEN BY: K.M. SCHRAMM, Z. METALKUNDE, VOL. 53, P. 729 (1962)

TABLE I



39---4
NICKEL POROUS

NI 99.5 PERCENT
AVERAGE GRAIN SIZE IS 8 MICRONS

$$V_{01} = 0.112$$

$$V_0 = 1/RH_00 \text{ SEE TABLE}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURES IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL AND THE MATERIAL OF THE STRIKER PLATE. U IS THE STRIKER VELOCITY BEFORE IMPACT.

TABLE

SAMPLE							DRIVER SYSTEM				
RH ₀₀	US	SIG.US	UP	P	V/V ₀	ST	U	US	UP	P	
6.22	9.79	0.26	.048	2908	0.7331	FE	8.64	10.67	4.26	3568	
5.09	9.42	0.18	.051	2469	0.7930	FE	8.64	10.67	4.26	3568	
2.97	9.25	0.15	.060	1839	1.0630	FE	8.64	10.67	4.26	3568	
8.90	14.87	0.20	.072	9560	0.5139	FE	15.45				
5.09	15.55	0.27	.087	6870	0.7722	FE	15.45				
2.97	15.58	0.27	.101	4670	1.0540	FE	15.45				

US =

COMMENTS:

- 1) SOURCE: KORMER, S.B., FUNTIKOV, A.I., URLIN, V.D.
AND KOLESNIKOVA, A.N.
SOVIET PHYS.-JETP, VOL. 15, P. 477 (1962)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
- 3) US IS THE AVERAGE DETERMINATION TAKEN FROM 4-8 EXPERIMENTS, EACH
CONSISTING OF 4-6 READINGS.
- 4) VOI WAS OBTAINED FROM THE HANDBOOK OF CHEMISTRY AND PHYSICS (THE
CHEMICAL RUBBER PUBLISHING CO. 1962-1963) 44TH ED.

TABLE I

NICKEL POROUS
39---4

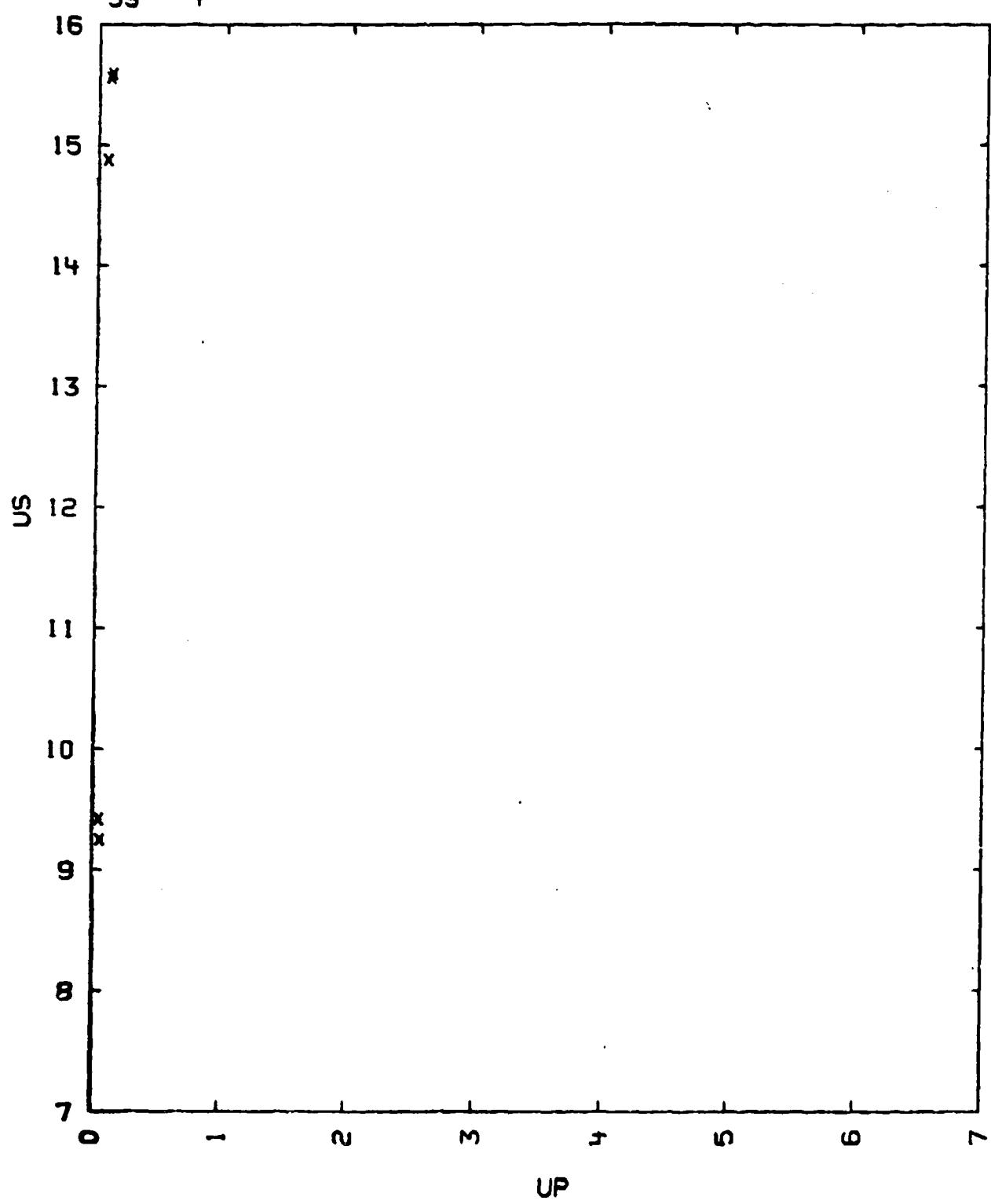
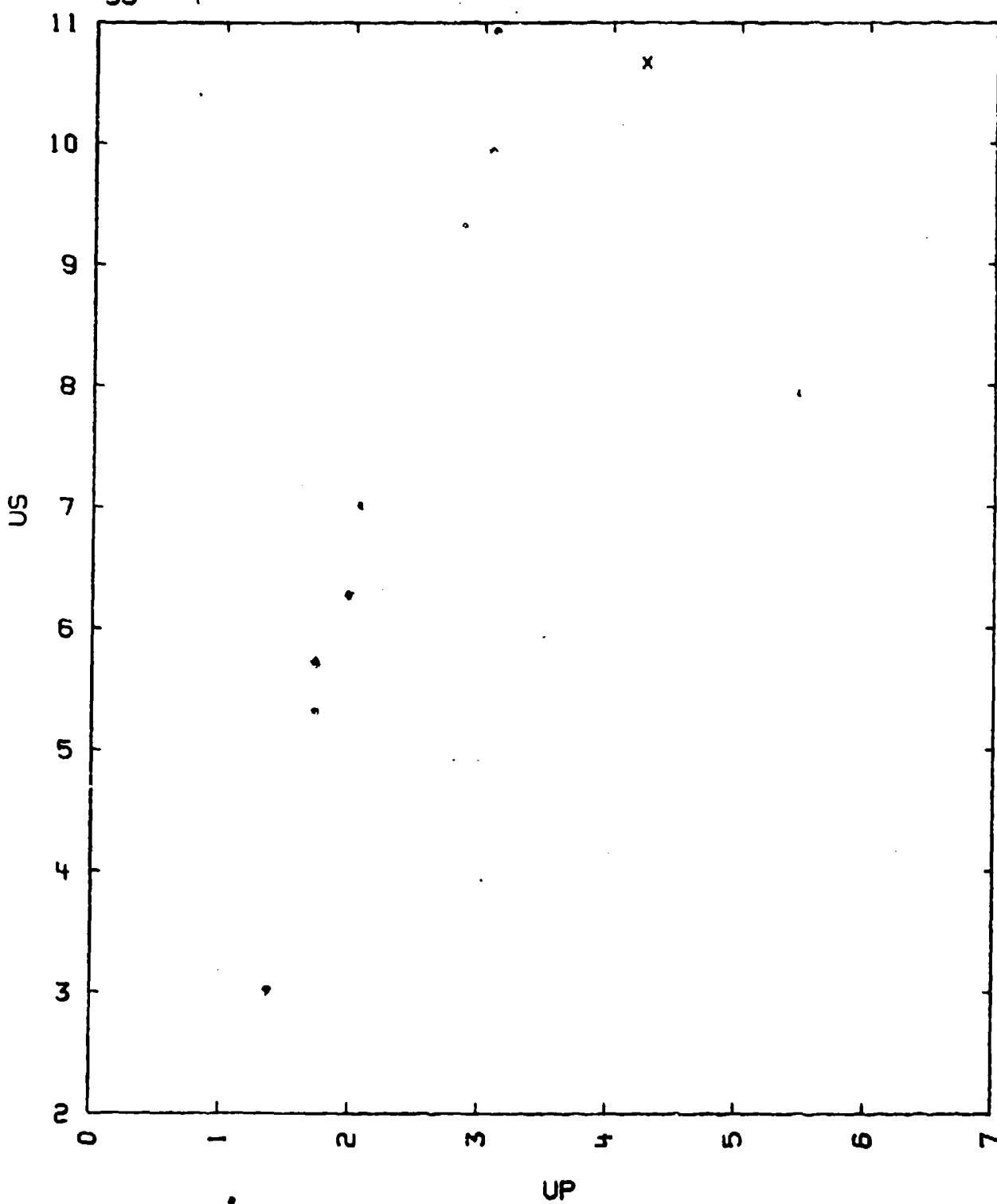


TABLE I

NICKEL POROUS

39---4



39---5
NICKEL

NI

V0 = 0.1123
V01 = 0.1122

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-STANDARD-
RHO0	US	UP	P	V/V0	UFS
8.905	5.682	0.935	473	0.844	3.040
-	5.834	0.910	472	0.835	3.010
-	5.675	0.898	455	0.842	2.940
-	5.643	0.910	461	0.839	2.970
-	5.598	0.835	417	0.853	2.740
-	5.549	0.815	403.5	0.851	2.675
-	5.586	0.845	419	0.849	2.760
-	5.624	0.840	421	0.851	2.760
-	5.668	0.823	416	0.855	2.725
-	5.668	0.810	410	0.857	2.695
-	5.624	0.815	409	0.855	2.695
-	5.518	0.805	396	0.854	2.635
-	5.488	0.733	358.5	0.866	2.430
-	5.488	0.733	358.5	0.866	2.430
-	5.298	0.780	367	0.850	2.510
-	5.230	0.785	366	0.853	2.510
-	5.411	0.715	345	0.868	2.365
-	5.506	0.715	343	0.870	2.355
-	5.524	0.725	356	0.869	2.415
-	5.393	0.725	348	0.866	2.385
-	5.387	0.657	316	0.878	2.200
-	5.422	0.642	313	0.882	2.175
-	5.549	0.635	313	0.886	2.165
-	5.512	0.635	323	0.880	2.225
-	5.593	0.610	303	0.891	2.100
-	5.330	0.620	295	0.884	2.085
-	5.175	0.635	292	0.877	2.080
-	5.202	0.635	295	0.879	2.100
-	5.000	0.529	235.5	0.894	1.750
-	4.970	0.534	236.5	0.893	1.760
-	5.170	0.520	241	0.899	1.785
-	5.112	0.525	240	0.897	1.785
-	4.931	0.405	178	0.918	1.375
-	5.097	0.401	182	0.921	1.390
-	5.092	0.406	183.5	0.920	1.400
-	4.496	0.370	148	0.918	1.205
-	4.634	0.363	150	0.922	1.205
-	4.617	0.380	149	0.922	1.200
-	4.570	0.365	148	0.920	1.200

NICKEL

	RHO0	US	UP	P	V/V0	UFS
-	4.804	0.413	189.5	0.910	1.345	
-	4.873	0.409	170	0.912	1.345	
-	4.960	0.403	178	0.919	1.375	
-	4.936	0.402	177	0.919	1.365	
-	4.840	0.375	161.5	0.923	1.275	
-	4.892	0.370	161.5	0.924	1.265	
-	4.713	0.348	146.5	0.926	1.175	
-	4.638	0.354	146	0.924	1.175	

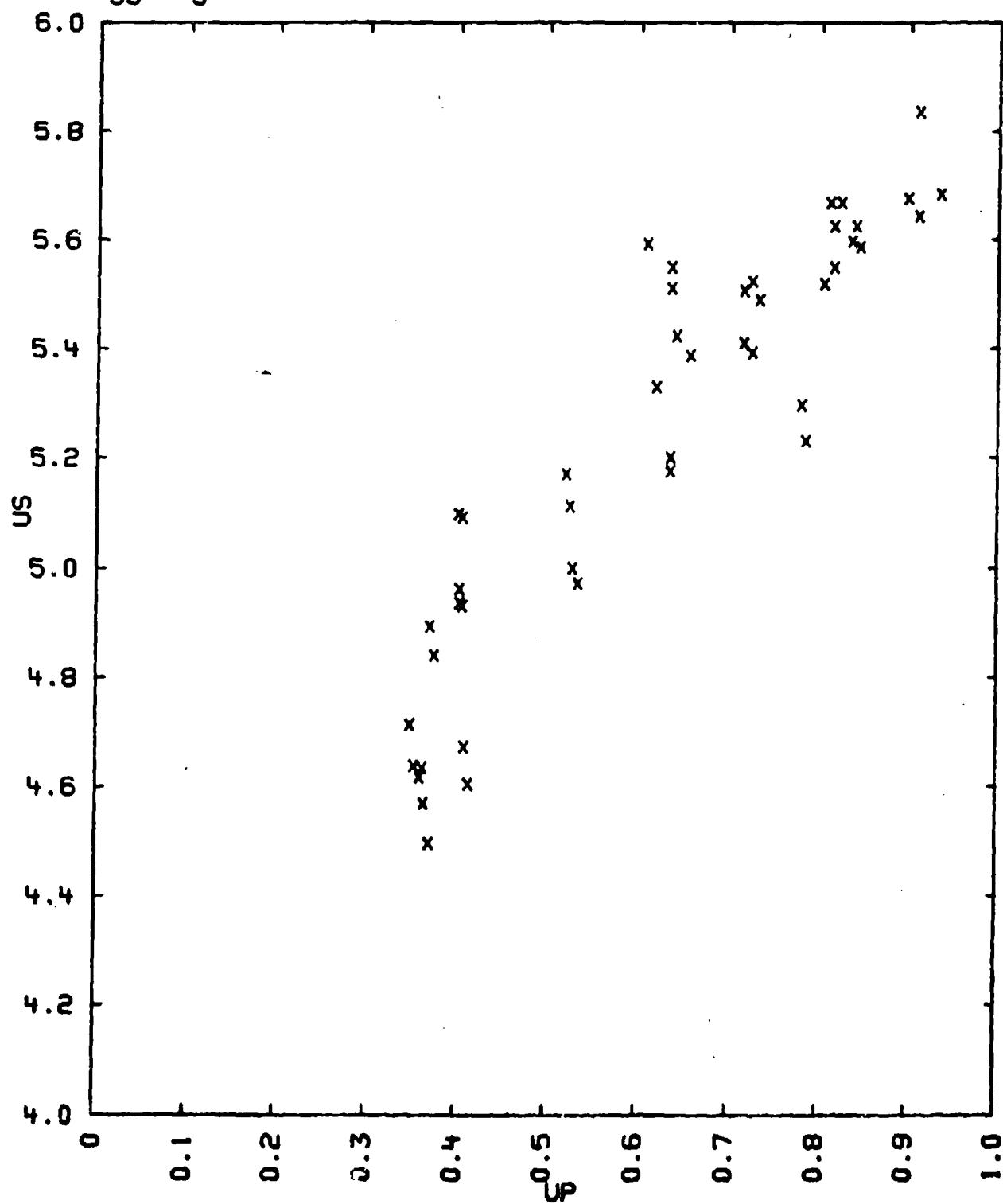
$$US = 4.119 + 1.806 UP \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: BERGER J. AND FAUQIGNON C.
PRIVATE COMMUNICATION (1964), B.P. NO. 7, SEVRAN, FRANCE
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL, ALUMINUM ALMg ALLOY
- 3) VOI WAS OBTAINED FROM THE CUBIC UNIT CELL WITH $A = 3.52394$ ANGSTROMS.
AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO. N.Y.,
1963) 2ND ED.
- 4) SAMPLE DIMENSIONS WERE: 2.0 CM DIAMETER
0.5 CM THICKNESS

TABLE I

NICKEL
39---5



39---6
NICKEL

NI	99.47	WT. PERCENT
C	0.11	-
MN	0.28	-
FE	0.09	-
S	0.005	-
SI	0.02	-
CU	0.02	-

$$V_0 = 0.1128 \text{ CC/G} \quad CL = 5.76 \text{ KM/SEC} \quad C_0 = 4.53 \text{ KM/SEC}$$

$$CS = 3.08 \text{ KM/SEC}$$

THE TABLE LISTS RHOO IN G/CC. VELOCITIES IN KM/SEC AND P IN Kbars. NI= NICKEL, FS= FANSTEEL AND WF= WEIGHTING FACTOR.

TABLE

SAMPLE					IMPACTOR			
RHO0	US	UP	P	V/V0	MAT	U	WF	
8.884	6.548	1.331	772.	.7987	NI	2.662	3	
8.884	7.407	1.919	1280.	.7409	NI	3.839	3	
8.884	10.070	3.592	3208.	.6433	NI	7.184	3	
8.884	11.572	4.581	4689.	.6041	FS	7.934	1	
8.884	11.607	4.616	4749.	.6023	FS	7.993	1	

$$US = 4.456 + 1.555 \cdot UP \text{ KM/SEC}$$

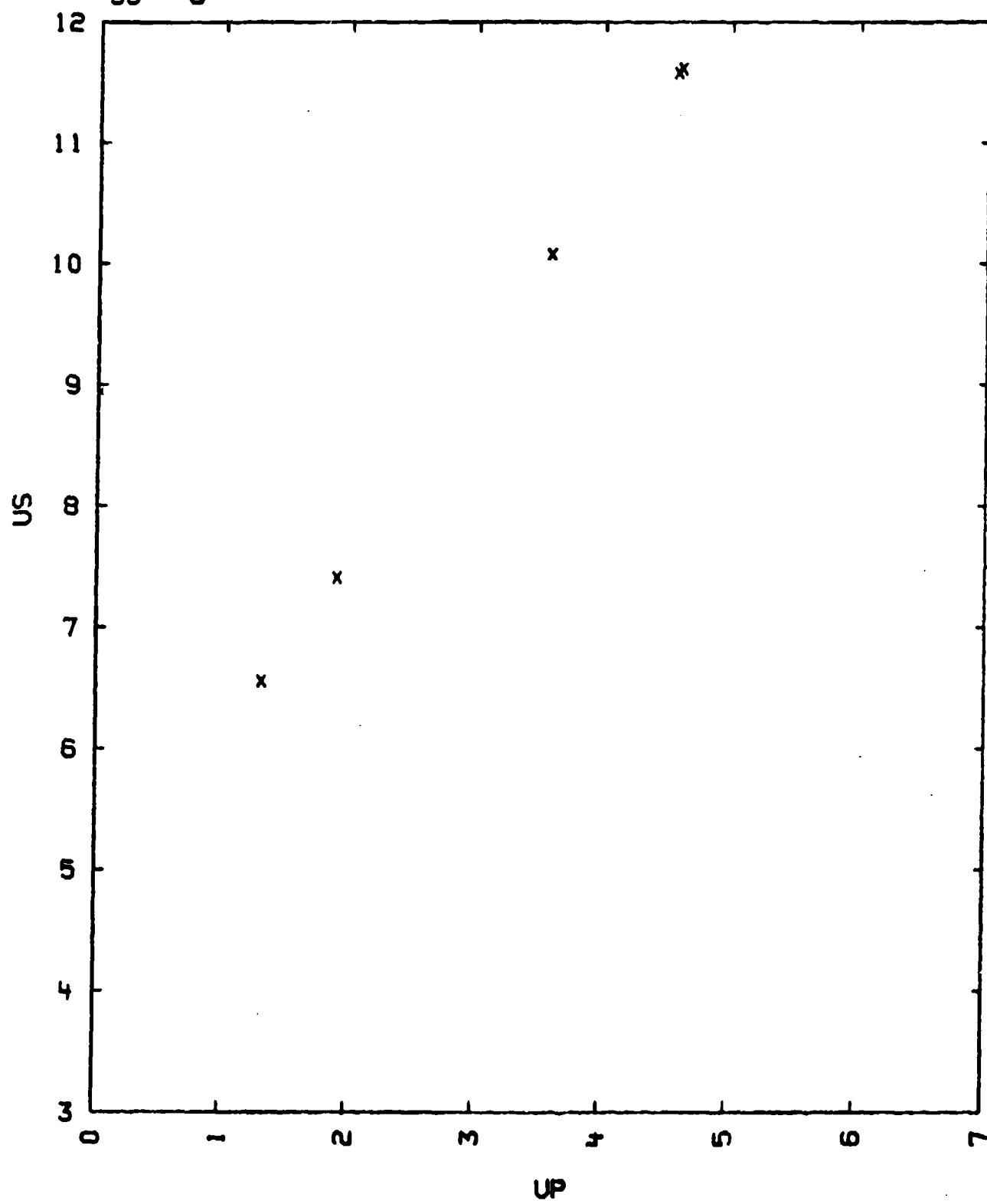
$$\Sigma 10 \cdot US = 0.012 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: ISBELL W.M., SHIPMAN F.H. AND JONES A.H.
HUGONIOT EQUATION OF STATE OF ELEVEN MATERIALS TO FIVE MBARS
MATERIALS SCIENCE LABORATORY REPORT: MSL-68-13
- 2) EXPERIMENTAL TECHNIQUE: A
DATA REDUCTION METHOD : A
- 3) NOMINAL UNCERTAINTIES ARE: $(\Sigma 10 \cdot US)/US = .005$ AND $(\Sigma 10 \cdot U)/U = .0005$
- 4) ALSO GIVEN: POISSONS RATIO 0.30 , YIELD STRENGTH 5.69 KBAR
TENSILE - 6.14 KBAR

TABLE I

NICKEL
39---6



40---1
COBALT

CO 99.45 PERCENT
NI 0.5
OTHER 0.05 -

$$V_0 = 0.1133 \text{ CC/G}$$

$$V_{01} = 0.1138 \text{ CC/G}$$

IN THE TABLE BELOW DENSITY IS GIVEN IN G/CC VELOCITY IN KM/SEC. AND PRESSURE IN KILOBARS.

TABLE

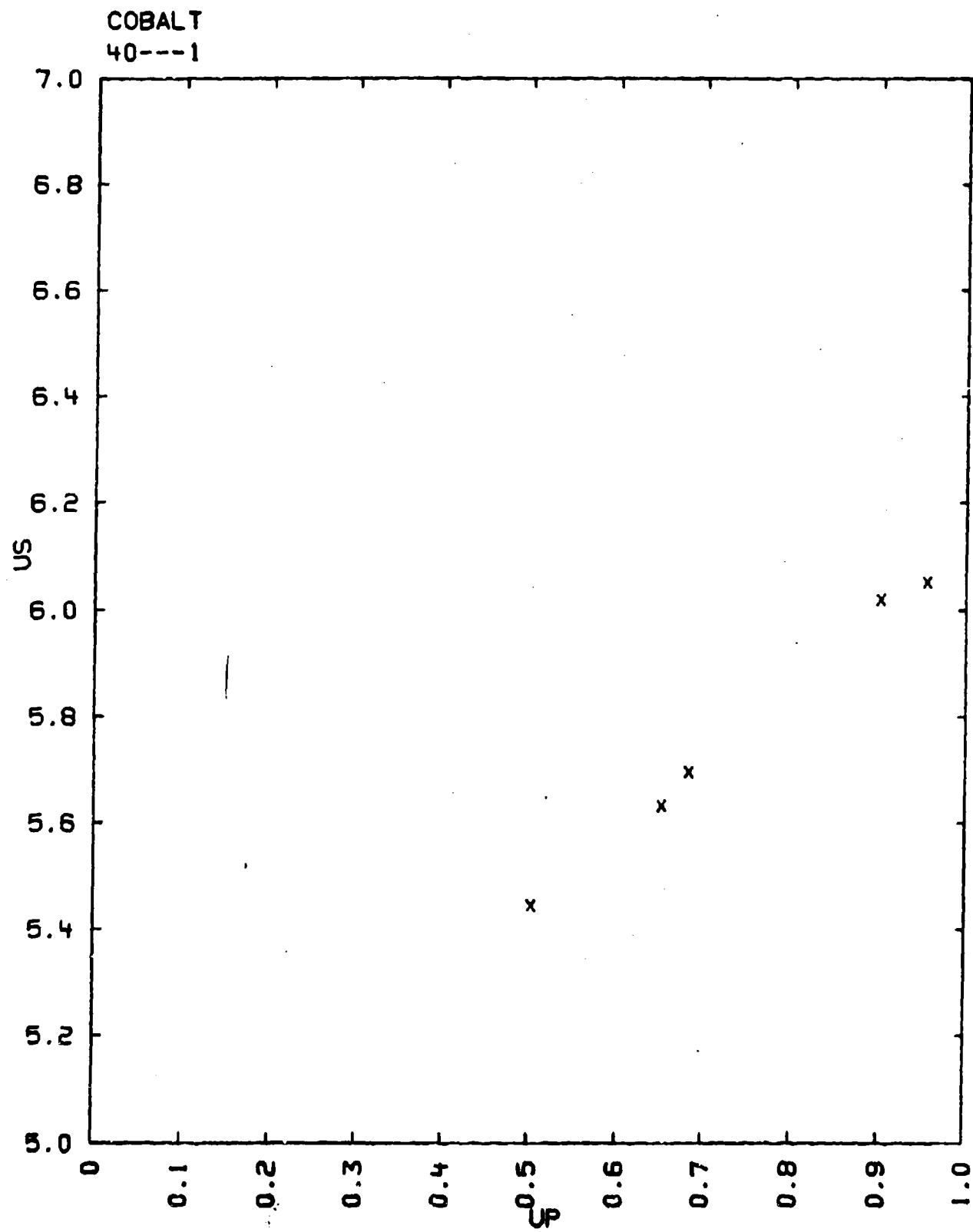
RHO0	US	UFS	UP	P	V/V0
8.82	3.445	1.016	0.502	241.1	0.9078
-	5.606	1.327	0.683	343.2	0.8801
-	5.632	1.334	0.683	324.4	0.8841
-	6.019		0.901	478.1	0.8503
-	6.052	1.890	0.955	509.8	0.8422

$$US = 4.740 + 1.392UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.020$$

COMMENTS:

- 1) SOURCE: WALSH, J.M., RICE, M.H., MCQUEEN, R.G. AND YARGER, F.L.
PHYS. REV., VOL. 109, P. 189 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
- 3) STANDARD MATERIAL: 24ST ALUMINUM
- 4) THE IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS
- 5) V01 WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYS. HANDBOOK
(MCGRAW-HILL BOOK CO. N.Y., 1963) 2ND ED.

TABLE I



40--2
COBALT

CO

$$V_0 = 0.113 \text{ CC/G.}$$

$$V_{01} = 0.1138 \text{ CC/G.}$$

$$C_B = 4.63 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC..
PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
8.82	7.15	1.79	1121	0.750	1038
-	7.15	1.80	1137	0.748	1050
-	7.12	1.83	1148	0.743	1064
-	7.50	2.06	1362	0.726	1267
-	7.45	2.07	1358	0.723	1267
-	7.43	2.07	1357	0.721	1267
-	7.81	2.30	1584	0.706	1478
-	7.79	2.30	1581	0.705	1478
-	7.77	2.30	1577	0.703	1478
-	7.88	2.31	1608	0.707	1497
-	7.83	2.32	1603	0.703	1497

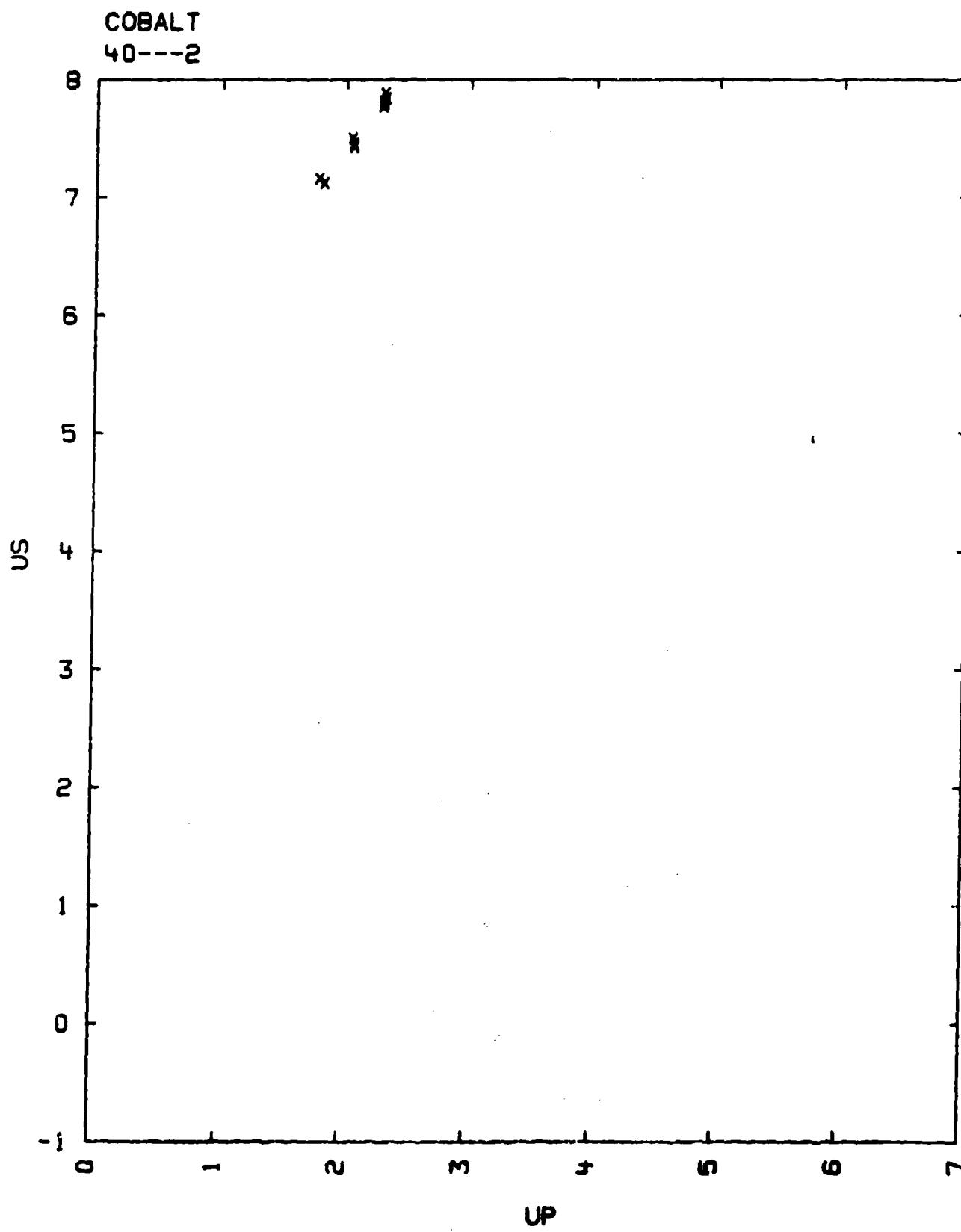
$$US = 4.748 + 1.330 UP \text{ KM/SEC.}$$

$$\text{SIGMA US} = 0.46 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R. G. AND MARSH, S. P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 38-33
- 4) V01 WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW HILL BOOK CO.)
2ND ED.

TABLE I



41---0
IRON SUMMARY

FE 99.99 TO 99.5 PERCENT BY WT.

 $V_0 = 0.1272 - 0.1277 \text{ CC/G}$ $CL = 5.960 \text{ KM/SEC}$ $CO = 4.64 \text{ KM/SEC}$.
 $V_{01} = 0.1269$ CC/G $CS = 3.240 \text{ KM/SEC}$

THE TABLE LISTS THE HUGONIOT POINTS CALCULATED FROM THE FITS GIVEN BELOW.
 UNITS ARE G/CC, KM/SEC, KBAR AND KBAR.G/CC FOR THE ENERGY DIFFERENCE.
 EL = ELASTIC LIMIT, TP = TRANSITION POINT.

TABLE

FIT	RHO0	US	UP	P	V/V0	E-E0	COMMENTS
1	7.85	6.00	0.015	7.1	0.9975	0.0011	EL 1ST WAVE
1	-	6.00	0.022	10.4	0.9963	0.0024	- - -
2	-	5.000	0.10	40.9	0.9807	0.0534	2ND -
2	-	5.045	0.20	80.8	0.9610	0.207	- -
2	-	5.091	0.30	121.	0.9417	0.459	- -
2	-	5.101	0.323	131.	0.9373	0.533	TP -
3	-	3.337	.34	135.	0.9320	0.617	3RD -
3	-	3.565	.40	152.	0.9150	0.910	- -
3	-	4.323	.60	224.	0.8724	1.908	- -
3	-	5.081	.80	321.	0.843	3.077	- -
4	-	5.404	1.0	425.	0.815	5.03	2ND -
4	-	7.106	2.0	1116.	0.718	20.0	1ST WAVE
4	-	8.741	3.0	2058.	0.657	45.0	- -
4	-	10.310	4.0	3237.	0.612	80.0	- -
4	-	13.248	6.0	6240.	0.547	180.	- -
4	-	15.920	8.0	9997.	0.497	320.	- -

 $US = 6.0 \text{ KM/SEC} + OR - 0.1 \text{ KM/SEC}$ (FOR FIT 1)
 FOR UP BETWEEN .022 KM/SEC.

 $US = 4.895 + 0.454 \cdot UP,$ SIG.US = .05 KM/SEC (FOR FIT 2)
 FOR UP BETWEEN .1 AND .33 KM/SEC.

 $US = 2.049 + 3.79 \cdot UP,$ SIG.US = .09 KM/SEC (FOR FIT 3)
 FOR UP BETWEEN .34 AND .79 KM/SEC.

 $US = 3.635 + 1.802 \cdot UP - .0333 \cdot UP^2,$ (FOR FIT 4)
 SIG.US = .13 KM/SEC
 FOR UP BETWEEN 1.0 AND 7.7 KM/SEC

COMMENTS

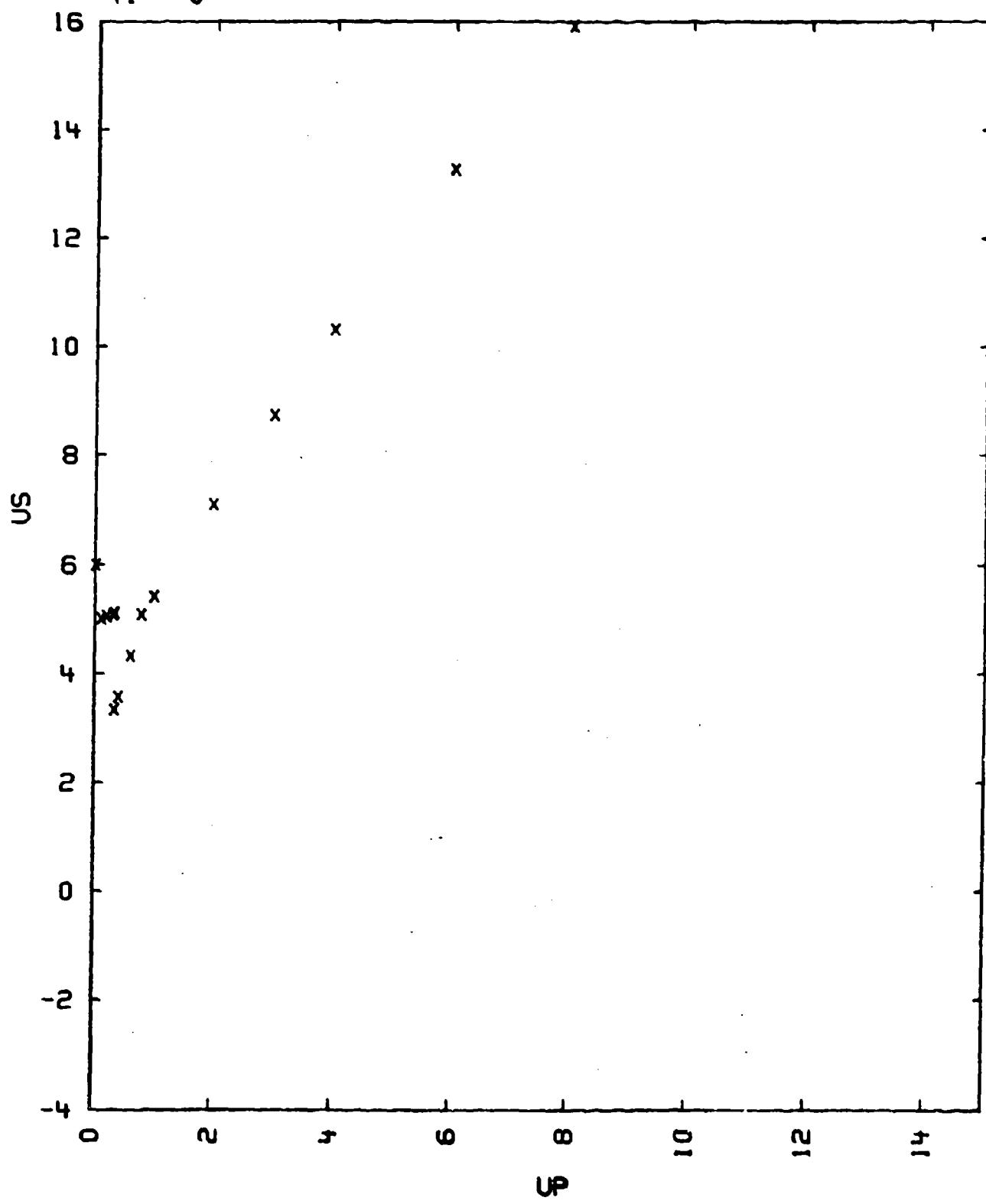
- 1) SOURCE: COMPILER
- 1A) DATA FOR THESE FITS WERE TAKEN FROM 41---1,2,3,4,5,7,8,10,11,12 AND 14
- 2) 41---1 SHOWS THAT THE ELASTIC WAVE STRENGTH DEPENDS ON THICKNESS.

- THE TWO VALUES LISTED WERE OBTAINED ON 30 AND 6 MM. SAMPLES RESPECTIVELY. THE 2ND AND HIGHEST ELASTIC LIMIT WAS USED TO CALCULATE P. V/V₀ AND E-E₀ FOR 2ND WAVES IN THE MULTIPLE WAVE REGION OF THE HUGONIOT. 6.0 KM/SEC WAS TAKEN AS THE BEST AVERAGE WAVE VELOCITY.
- 3) THE 41---1 SOURCE ALSO SHOWED THAT THE 2ND WAVE WAS BROAD, ESPECIALLY FOR SMALL AMPLITUDES. POINTS WITH UP LESS THAN .08 WERE THEREFORE, ELIMINATED. FIT 2 WAS MADE FROM DATA OF 41---1,5,10,11,12 AND 14.
 - 4) THE FINAL PORTION OF THE HUGONIOT CURVE ABOVE THE TRANSITION (HERE TAKEN AS 131 KBAR) IS GIVEN IN TWO PARTS. DATA OF 41---14 SHOW A HIGHER SLOPE AT LOW PRESSURES IN THE MULTIPLE WAVE REGION. IN THIS REGION, MINSHALL'S DATA SHOW 10-PERCENT LOWER SHOCK VELOCITIES. THE DATA OF MINSHALL (41---5,10,11 AND 12) SUGGEST THAT THE TRANSITION WAVE STRENGTH DEPENDS ON SAMPLE THICKNESS.
 - 5) FIT 4 WAS OBTAINED FROM 41---3,4 AND 7 WITH A WEIGHT OF 2 AND 41---2,9 WITH A WEIGHT OF 1.
 - 6) C₀ WAS TAKEN FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK (MCGRAW HILL BOOK CO., N. Y., 1963) 2ND ED. C. A. ROTTER AND C. S. SMITH MEASURED C₀ = 4.606 KM/SEC. (J. PHYS. CHEM. SOLIDS, V. 27, P. 267 (1966)).

TABLE I

IRON SUMMARY

41---0



41---
IRON ARMCO

FE

$$V_0 = .1274 \text{ CC/G} \quad C_L = 3.960 \text{ KM/SEC} \quad C_0 = 4.64 \text{ KM/SEC.}$$

$$V_{01} = 0.1269 \text{ CC/G} \quad C_S = 3.240 \text{ KM/SEC}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN MM/MICRO-SEC, PRESSURE IN KILOBARS AND SAMPLE THICKNESS (LS) IN MM.

TABLE

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0	LS
7.83	6.18	.015	7.28	.9976	5.08	.1935	78.4	.9625	50.8
-	6.18	.015	7.28	.9976	4.88	.0848	34.0	.9833	-
-	6.18	.015	7.28	.9976	4.91	.1080	43.1	.9786	-
-	6.18	.017	8.25	.9972	4.95	.096	39.3	.9812	25.4
-	6.18	.017	8.25	.9972	4.69	.060	24.1	.9880	-
-	6.18	.017	8.25	.9972	5.05	.186	75.2	.9637	-
-	6.18	.0195	9.46	.9968	5.08	.1805	73.6	.9651	12.7
-	6.18	.0195	9.46	.9968	4.65	.0770	30.4	.9844	-
-	6.18	.0195	9.46	.9968	4.17	.0310	13.2	.9946	-
-	6.18	.022	10.67	.9964	4.65	.079	31.5	.9841	6.35
-	6.18	.022	10.67	.9964	5.06	.192	75.9	.9639	-
-	6.18	.022	10.67	.9964	4.98	.130	52.9	.9747	-
-	6.18	.022	10.67	.9964	4.98	.192	75.7	.9622	-

US =

COMMENTS:

- 1) SOURCE: TAYLOR, JOHN W. AND RICE, M.H.
J. APPL. PHYS., VOL. 34, P. 364 (1963)
- 2) EXPERIMENTAL TECHNIQUE G
DATA REDUCTION TECHNIQUE A
THE SHOCK WAS DRIVEN INTO THE SAMPLE BY AN ARMCO PLATE, BACKED BY AN ALUMINUM PROJECTILE, WHICH WAS FIRED FROM A 6.5 INCH GUN.
- 3) A CAPACITOR TECHNIQUE MEASURED SHOCK ARRIVAL AND FREE SURFACE VELOCITY. PINS WERE USED TO MEASURE PROJECTILE VELOCITY.
- 4) CL AND CS WERE OBTAINED FROM THE AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO. N.Y., 1963) 2ND ED.
- 5) PROJECTILE VELOCITY WAS EQUAL TO THE FREE SURFACE VELOCITY TO WELL WITHIN THE ACCURACY OF THE EXPERIMENT (2 PERCENT).
- 6) NOTE THAT US2 = V01(P2-P1)/(V1-V2))^(1/2)

TABLE I

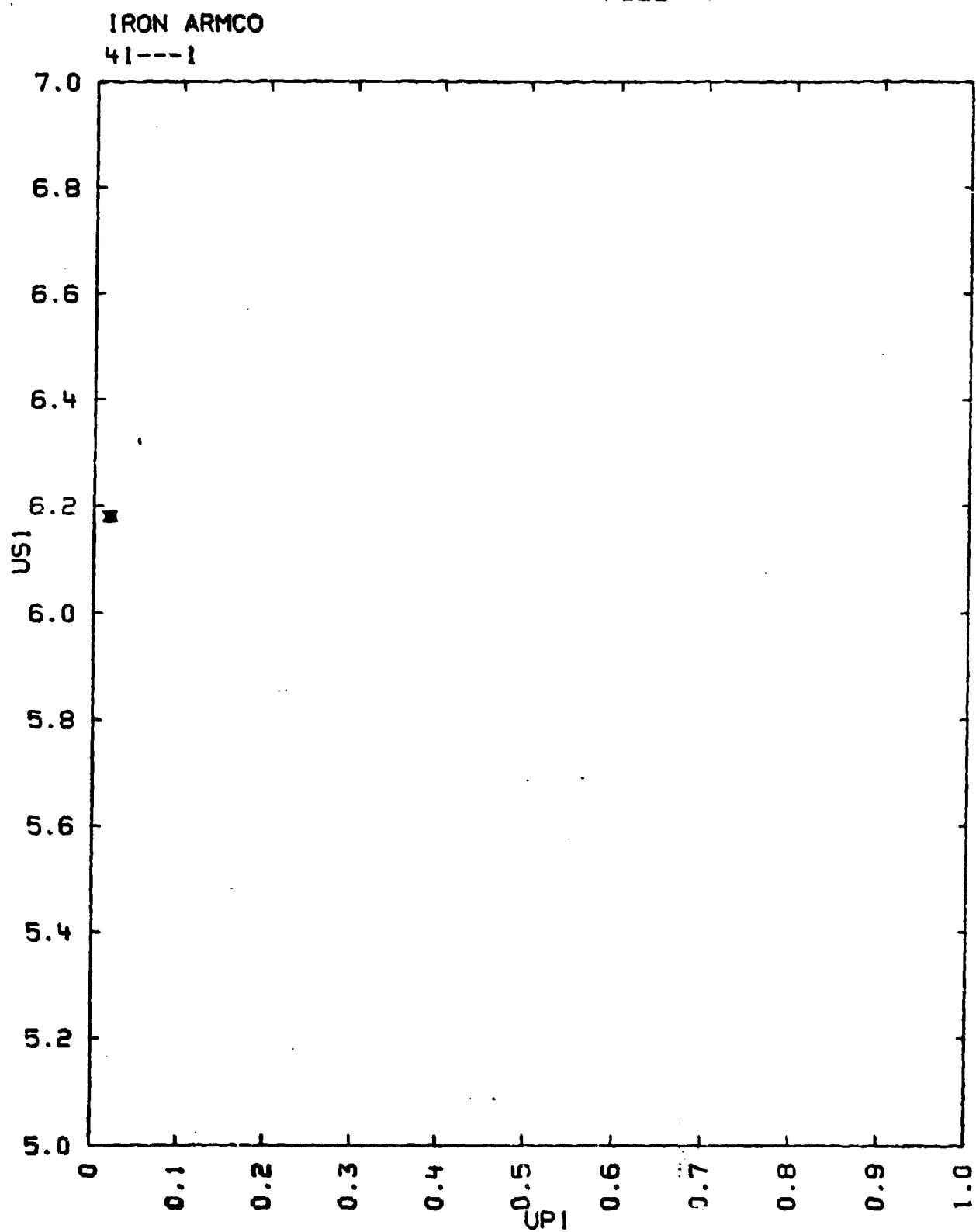
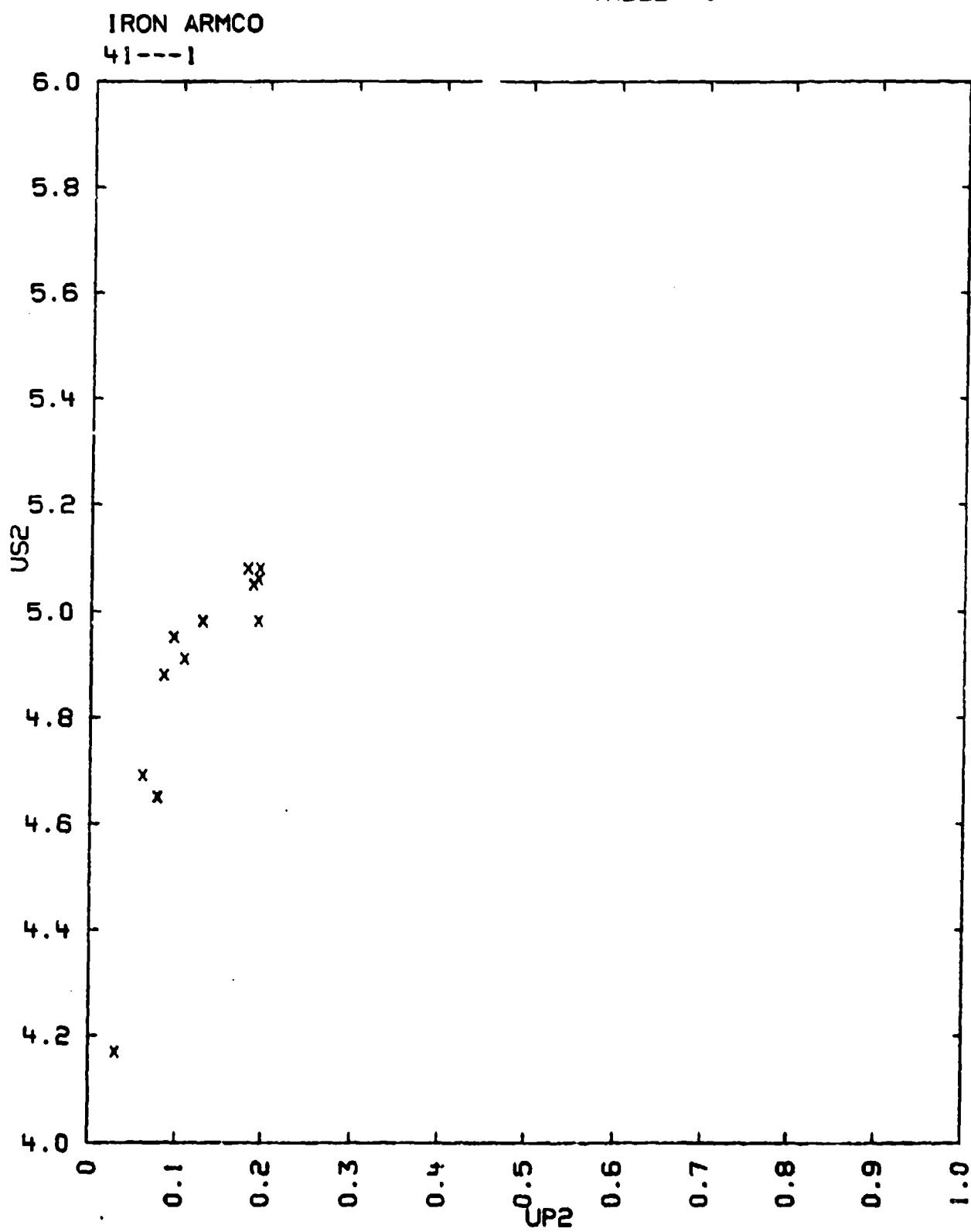


TABLE I



41---2
STEEL MILD. EN3

IRON	FE	98.75 PERCENT
CARBON	C	.23 PERCENT
MANGANESE	MN	1.00 PERCENT

$$V_0 = 0.1275 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC., VELOCITIES IN MM/MICRO-SEC., AND PRESSURE IN KILOBARS.

TABLE

RHO0	US	UP	P	V/V0
7.84	7.06	1.97	1090	.721
-	7.18	1.99	1120	.723
-	6.99	2.05	1120	.707
-	8.08	2.63	1670	.675
-	7.90	2.75	1700	.652
-	8.11	2.70	1720	.667
-	8.07	2.75	1740	.659
-	10.13	3.88	3090	.617
-	10.66	4.09	3420	.616
-	10.90	4.32	3700	.604

$$US = 3.67 + 1.645 UP \text{ KM/SEC.} \quad SIG.US = 0.18 \text{ KM/SEC}$$

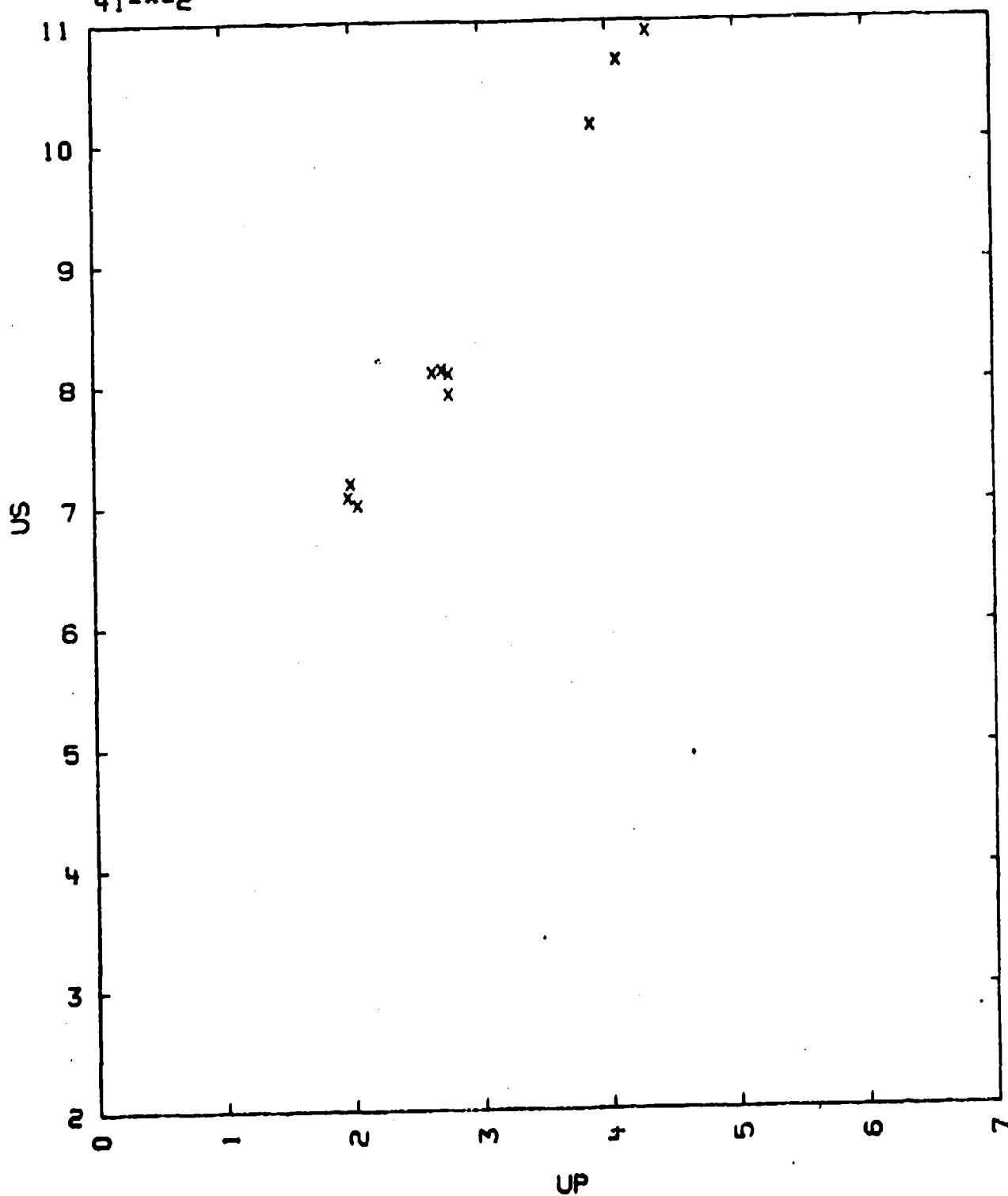
COMMENTS:

- 1) SOURCE: SKIDMORE, I.C. AND MORRIS, E.
THERMODYNAMICS OF NUCLEAR MATERIALS, P. 173 FF. (1962)
INTERN. AT. ENERGY AGENCY, VIENNA
ATOMIC WEAPONS RESEARCH ESTABLISHMENT, ALDERMASTON, ENGLAND
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE A
IN THE EXPERIMENTS, EXPLOSIVELY ACCELERATED PLATES OF THE ABOVE MATERIAL WERE IMPACTED WITH TARGET PLATES OF THE SAME MATERIAL WHICH CONTAINED THE SAMPLES.
- 3) THE VELOCITY OF THE FLYING PLATE AND THE SHOCK AND SURFACE VELOCITY OF THE TARGET PLATE WERE MEASURED AS WELL AS THE SAMPLE SURFACE AND SHOCK VELOCITIES.
- 4) DATA SCATTER WAS ABOUT 0.03 MICROSEC.
- 5) CORRECTIONS WERE MADE FOR FLYING PLATE CURVATURE OF UP TO 1 MICROSEC.
- 6) THE HIGHER PRESSURES WERE OBTAINED BY A SPHERICALLY CONVERGING SYSTEM.
- 7) ALL PELLETS WERE SURROUNDED BY LEAD TO REDUCE LATERAL RAREFACTION.

TABLE I

STEEL MILD, EN3

41---2



4!---3

IRON

FE 99.8 PER CENT OR GREATER

$$V_0 = 0.1278 \text{ CC/G.} \quad C_L = 5.85 \text{ KM/SEC.} \quad C_0 = 4.51 \text{ KM/SEC.}$$

$$V_{01} = 0.1271 \text{ CC/G.} \quad C_S = 3.23 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

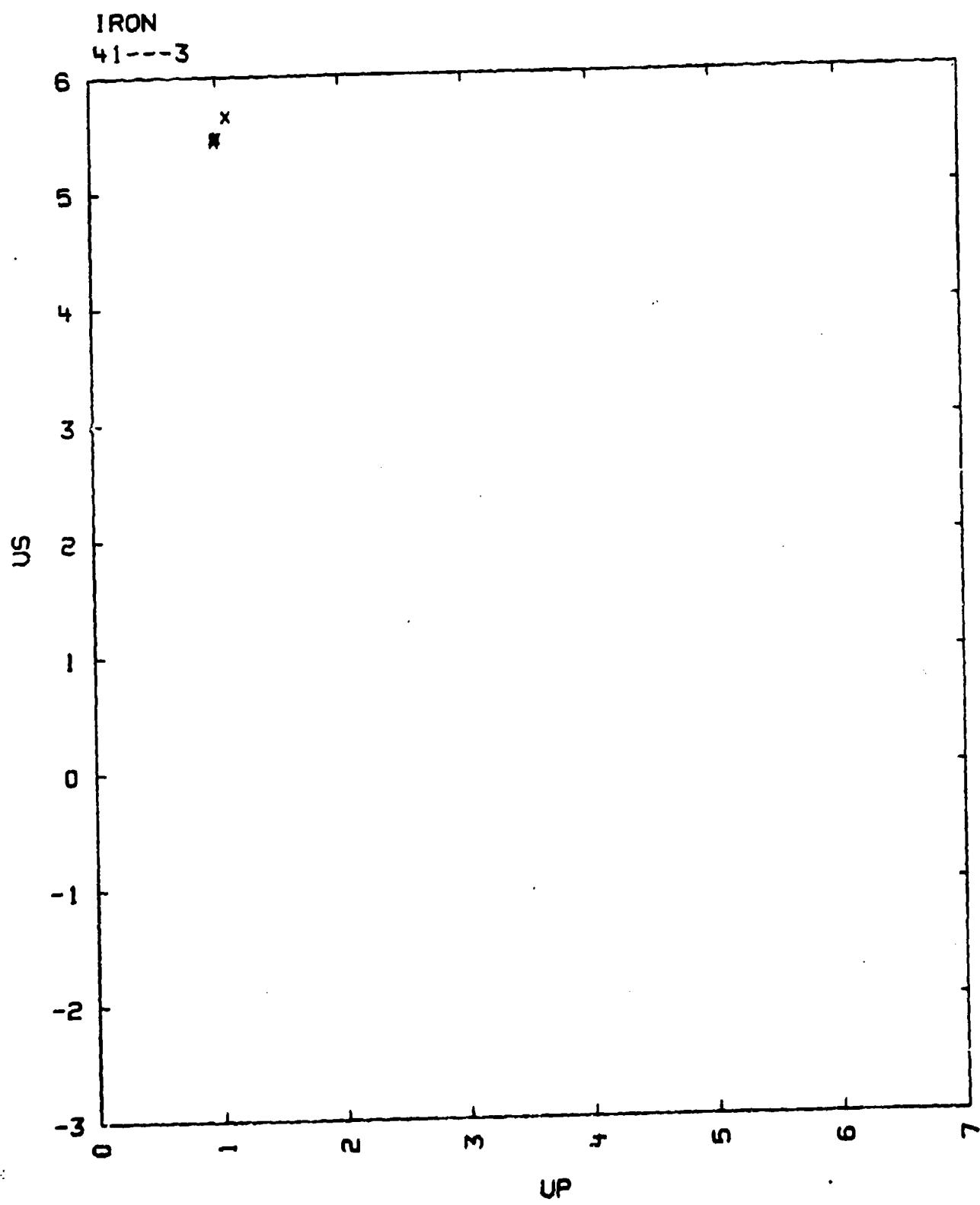
RHO0	US	UFS	UP	P	V/V0
7.84	5.652	2.153	1.085	480.8	0.8080
-	5.474	2.037	1.013	434.7	0.8149
-	5.458	1.988	0.993	424.9	0.8181
-	5.438	2.005	0.984	423.8	0.8172

$$US = 3.205 + 2.252 UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.3 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS.
- 5) CL AND CS VALUES WERE OBTAINED FROM L. BERGMAN, DER ULTRASCHALL
S. HIRZEL VERLAG, STUTTGART, 1954, 6TH ED., P. 650
- 6) V01 WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW HILL BOOK CO.)
2ND ED.

TABLE I



41---4

IRON

FE

$V_0 = 0.1272 \text{ CC/G}$ $C_L = 5.95 \text{ KM/SEC.}$ $C_D = 4.627 \text{ KM/SEC.}$
 $V_{01} = 0.1271 \text{ CC/G.}$ $C_S = 3.240 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC.
PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

RH00	SAMPLE					STANDARD		
	US	UP	P	V/V0		US	P	MAT
7.86	5.57	1.09	477	0.804		339	BR	
7.86	6.54	1.64	843	0.749		847	-	
7.86	6.57	1.66	857	0.749		860	-	
7.86	6.65	1.74	911	0.739		917	-	
7.86	6.71	1.79	943	0.733		950	-	
7.86	6.63	1.86	968	0.720		988	-	
7.86	6.89	1.89	1024	0.726		1030	-	
7.86	6.95	1.89	1033	0.728		1034	-	
7.86	7.42	2.17	1267	0.707		1267	-	
7.86	7.42	2.19	1276	0.705		1276	-	
7.86	7.66	2.32	1397	0.697		1397	-	
7.86	7.58	2.34	1393	0.692		1393	-	
7.86	8.00	2.57	1618	0.679		1618	-	
7.86	8.22	2.68	1728	0.675		1728	-	
7.86	8.20	2.68	1730	0.673		1730	-	
7.86	5.33	1.03	430	0.807	5.40		CU	
7.86	5.49	1.04	451	0.810	5.44		-	
7.86	5.57	1.10	481	0.803	5.52		-	
7.86	5.60	1.13	499	0.797	5.57		-	
7.86	5.57	1.14	498	0.796	5.57		-	
7.86	6.29	1.45	718	0.769	6.05		-	
7.86	6.84	1.81	973	0.735	6.57		-	
7.86	7.40	2.12	1232	0.714	7.03		-	
7.86	7.36	2.13	1231	0.711	7.04		-	
7.87	7.67	2.30	1383	0.701	7.29		-	
7.87	7.57	2.31	1375	0.693	7.29		-	
7.87	7.73	2.33	1418	0.699	7.34		-	
7.87	7.62	2.34	1405	0.693	7.34		-	
7.86	8.14	2.60	1663	0.681	7.73		-	
7.86	8.16	2.64	1693	0.677	7.78		-	
7.85	8.68	2.89	1987	0.667	8.17		-	

US = 3.768 + 1.655 UP KM/SEC.

SIGMA US = 0.101 PERCENT

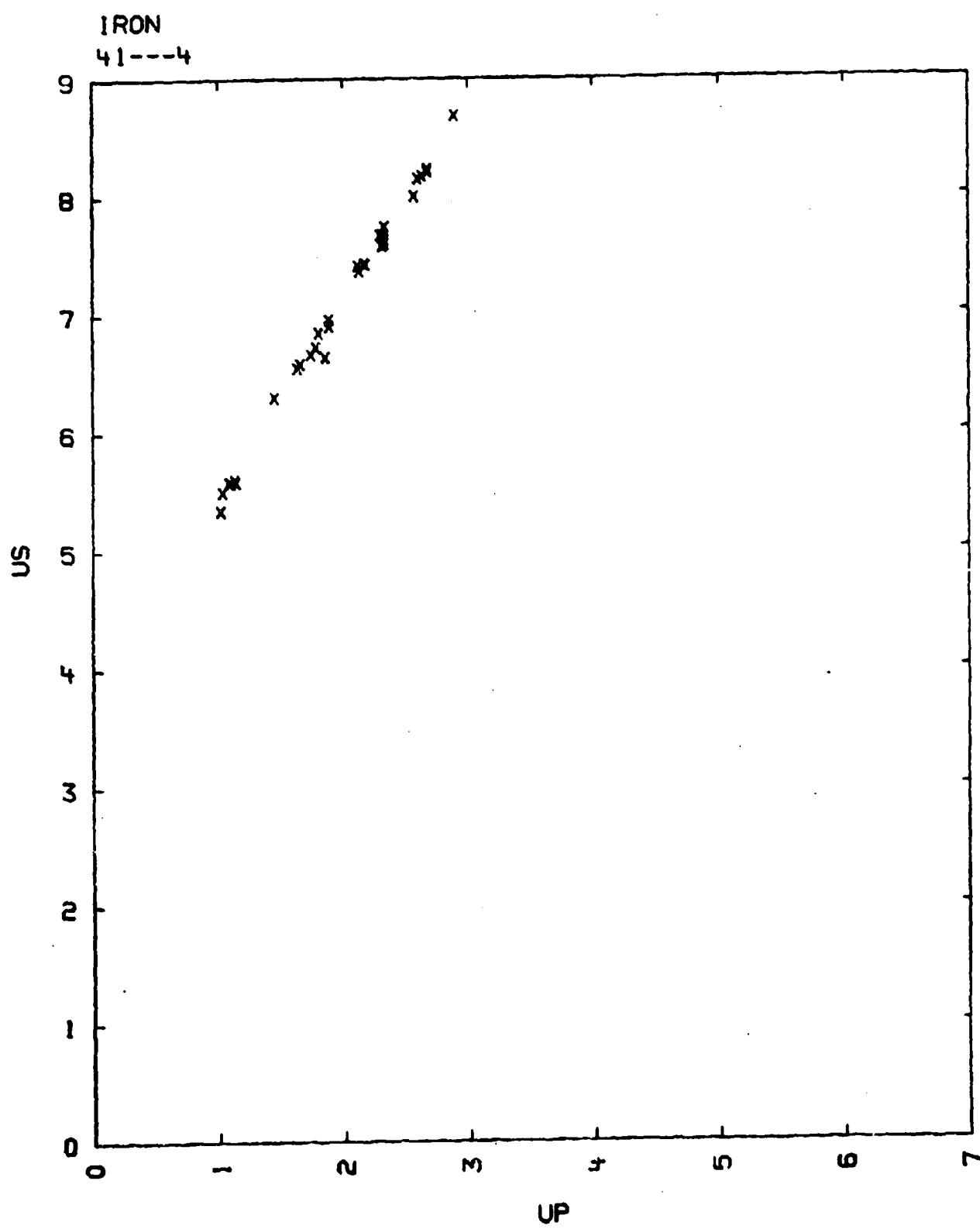
COMMENTS:

- 1) SOURCE: MCQUEEN, R. G. AND MARSH, S. P.
J. APPL. PHYS., VOL. 31, P. 1293 (1960)

ALSO PRIVATE COMMUNICATION
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-B, LOS ALAMOS, N. MEX.

- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL AS INDICATED IN THE TABLE.
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 4) CL AND CS VALUES WERE OBTAINED FROM THE AMERICAN INSTITUTE OF
PHYSICS HANDBOOK, 2ND ED., 1957
- 5) VOI WAS OBTAINED FROM THE SAME SOURCE

TABLE I



41---5
IRON, ARMCO

FE 99.99 PERCENT

$$V_0 = 0.1274 \text{ CC/G} \quad CL = 5.97 \text{ KM/SEC}$$

$$V_{01} = 0.1269 \text{ CC/G}$$

TABLE I LISTS LONGITUDINAL SOUND VELOCITY AS WELL AS SHOCK AND PARTICLE VELOCITIES OF THE MULTIPLE WAVE SYSTEM IN KM/SEC. D IS THE SAMPLE THICKNESS IN MM AND THE EXPERIMENT OR LINE NUMBER (NO) FOLLOWS THE NUMBERING OF THE SOURCE. FOR THE PRESSURES AND V/V₀ SEE TABLE II
 * UNCERTAIN POINT, SEE COMMENTS

TABLE I

NO	CL	US1	UP1	US2	UP2	US3	UP3	D
6	5.97	6.04	0.0143	5.09	0.277			133.63
7	-	-		5.04	0.313			56.93
8	-	-		5.05	0.324			38.63
9	-	-		5.09	0.327			24.74
10	-	-		5.12	0.335			25.41
11	-	-		5.13	0.325			25.46
24-25	5.92	5.99	0.0170					24.78
26-28	-	-	-	5.01	0.302			25.34
29-31	5.81	5.74	0.0291					25.49
29-31	-	5.33	0.0332*					-
32-34				4.995	0.339*			25.48
35	5.90	5.91	0.0151					25.30
36	5.89	5.92	0.0254					25.23
1A		6.04	0.0142	5.09	0.2635			133.63
2A				5.04	0.3125			57.2
3A		-	-	5.05	0.324	3.13	0.402	38.63
4A				5.06	0.321		0.420	31.8
5A		-	-	5.05	0.327	3.41	0.465	24.74
6A					0.32		0.50	19.11
7A		-	-	-	0.330	3.61	0.5345	13.25
8A		-	-	-	0.3405	3.71	0.5665	7.21

US3 = 1.76 + 3.46·UP KM/SEC FOR UP FROM 0.4 TO 0.6 KM/SEC
 SIGMA US3 = 0.033 KM/SEC

TABLE II LISTS DENSITY IN G/CC AND PRESSURE IN KBARS. THE SIGNIFICANCE OF THE THERMAL HISTORY (THIST) IS GIVEN IN THE COMMENTS. AS IN TABLE I THE LINE NUMBERING FOLLOWS THAT OF THE SOURCE.

* UNCERTAIN POINT, SEE COMMENTS

TABLE II

NO	RHO0	P1	V1/V0	P2	V2/V0	P3	V3/V0	THIST
6	7.851	6.8	0.9979	112.1	0.9460			ANN
7	-	-	-	124.9	0.9382			-
8	-	-	-	129.7	0.9366			-
9	-	-	-	131.8	0.9358			-
10	-	-	-	135.8	0.9350			-
11	-	-	-	132.0	0.9374			-
24-25	7.854	8.0	0.9974					AR
26-28	-	-	-	127.6	0.9362			-
29-31	7.840	13.1	0.9949					CR
29-31	-	14.8	0.9941					-
32-34	-			134.9	0.9330			-
35	7.844	7.0	0.9977					CRHT
36	7.842	11.8	0.9959					CRHT
1A	7.851	6.8	0.9977	106	0.9487			
3A	-	-	-	130	0.9363	147	0.909	
5A	-	-	-	131	0.9356	165	0.892	
7A	-	-	-	132	0.9351	188	0.877	
8A	-	-	-	135	0.9331	200	0.871	

COMMENTS:

- 1) SOURCE: MINSHALL, F. S.
MET. SOC. CONF., VOL. 9, P. 249 (1951)
ALSO BANCROFT, D., PETERSON, E. L. AND MINSHALL, S.
J. APPL. PHYS., VOL. 27, P. 291 (1956)
- 2) EXPERIMENTAL TECHNIQUE A. EXCEPT NO. 24-25, 29-31, 35 AND 36, WHERE
TECHNIQUE G WAS USED.
- DATA REDUCTION METHOD D WITH $2U_P = U_{FS}$
- 3) THE FIRST WAVE (SUBSCRIPT 1) IS INTERPRETED AS THE ELASTIC WAVE AND
WAS ASSUMED TO BE IDENTICAL TO THE ONE MEASURED IN 1A FOR LINE 1A-8A
AND EQUAL TO THE ONE MEASURED IN LINE 6 FOR LINE 6-11. EVEN THOUGH
IN LINE 10, THE INITIAL SAMPLE TEMPERATURE WAS -51 DEG. CENTIGRADE
AND IN LINE 11, +56.5 DEG. CENTIGRADE.
THE ELASTIC WAVE IN EXPERIMENTS 29-31 SEEMED TO BE SPLIT INTO TWO
WAVES, BOTH RUNNING SLOWER THAN THE LONGITUDINAL SOUND VELOCITY. THIS
DOUBLE WAVE STRUCTURE HAS BEEN USED IN THE ANALYSIS OF LINE 32-34
- 4) THE THERMAL HISTORY SYMBOLS ARE DEFINED AS FOLLOWS:
ANN = ANNEALED
AR = AS RECEIVED
CR = COLD ROLLED
HT = HEAT TREATED: EXP. 35 HELD AT 950 DEG. C. FOR 2 HRS
EXP. 36 HELD AT 600 DEG. C. FOR 2 HRS
- 5) THE ATTENUATION OF THE 3RD SHOCK WAS TAKEN INTO ACCOUNT IN EXP. 1A-8A
- 6) HERE A REFERS TO THE RESULTS OF THE EARLIER REFERENCE. P AND V/V0
WERE NOT CALCULATED FOR EXP. 6A BECAUSE OF THE LOWER ACCURACY OF THIS
SIMPLER EXPERIMENT, AND FOR 2A AND 4A BECAUSE THESE WERE DESIGNED
PRIMARILY TO MEASURE U_{FS} , FOR USE IN LINES 3 5 7 AND 8.
- 7) ALL VALUES OF P2 EXCEPT THOSE OF THE 133 MM SAMPLES, NO 6 AND 1A WERE
FOLLOWED BY A THIRD WAVE AND THEREFORE TRANSITION WAVES. A PLOT OF

- P2 VS. I/D SUGGESTS THAT THE TRANSITION COULD BE AS LOW AS 123
PLUS OR MINUS 3 KBARS
- 8) VOI WAS OBTAINED FROM WYCKOFF, CRYSTAL STRUCTURES, VOL. I (JOHN
WILEY AND SONS, NEW YORK, 1963)

TABLE I

IRON, ARMCO
41---5

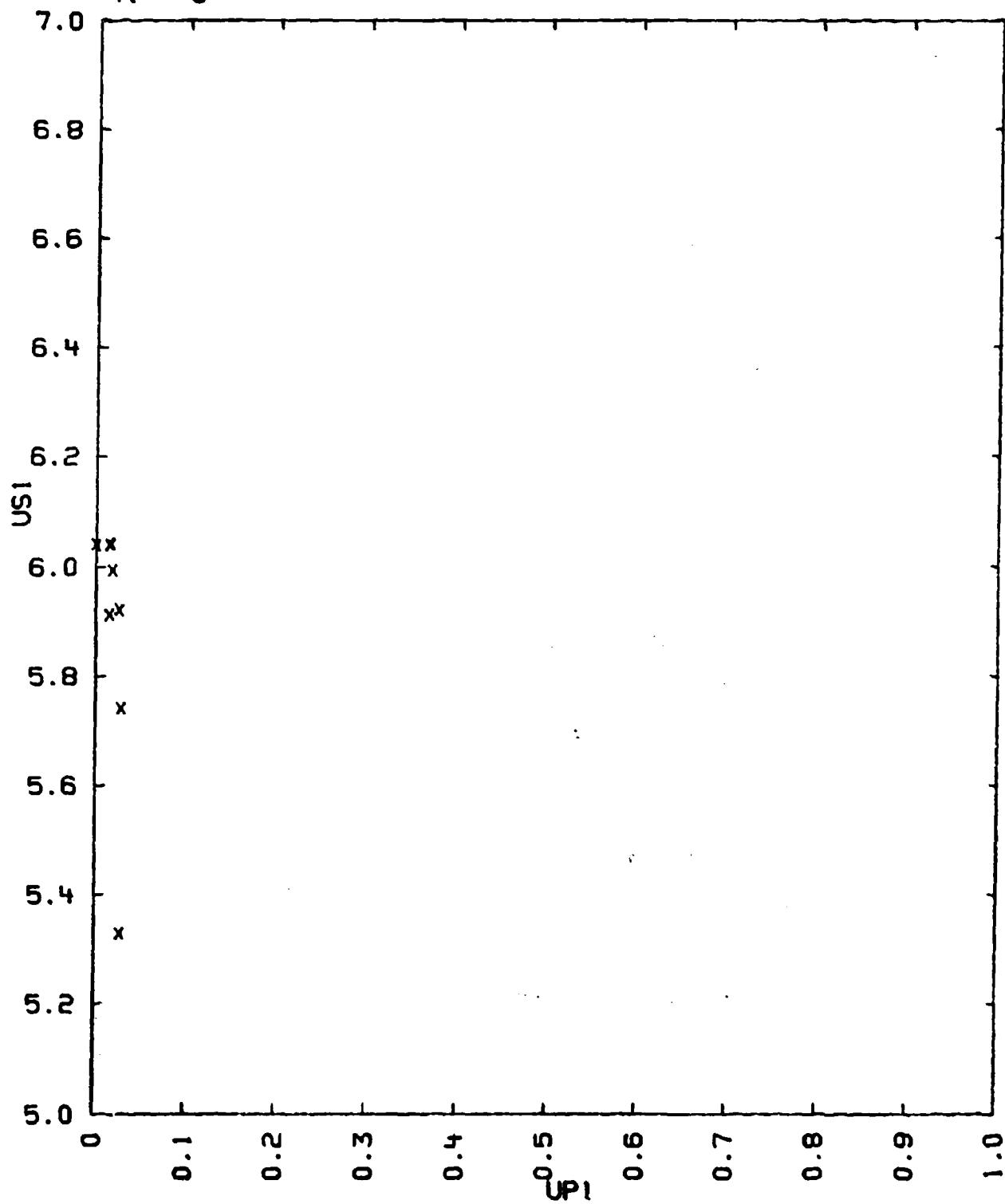


TABLE I

IRON, ARMCO
41---5

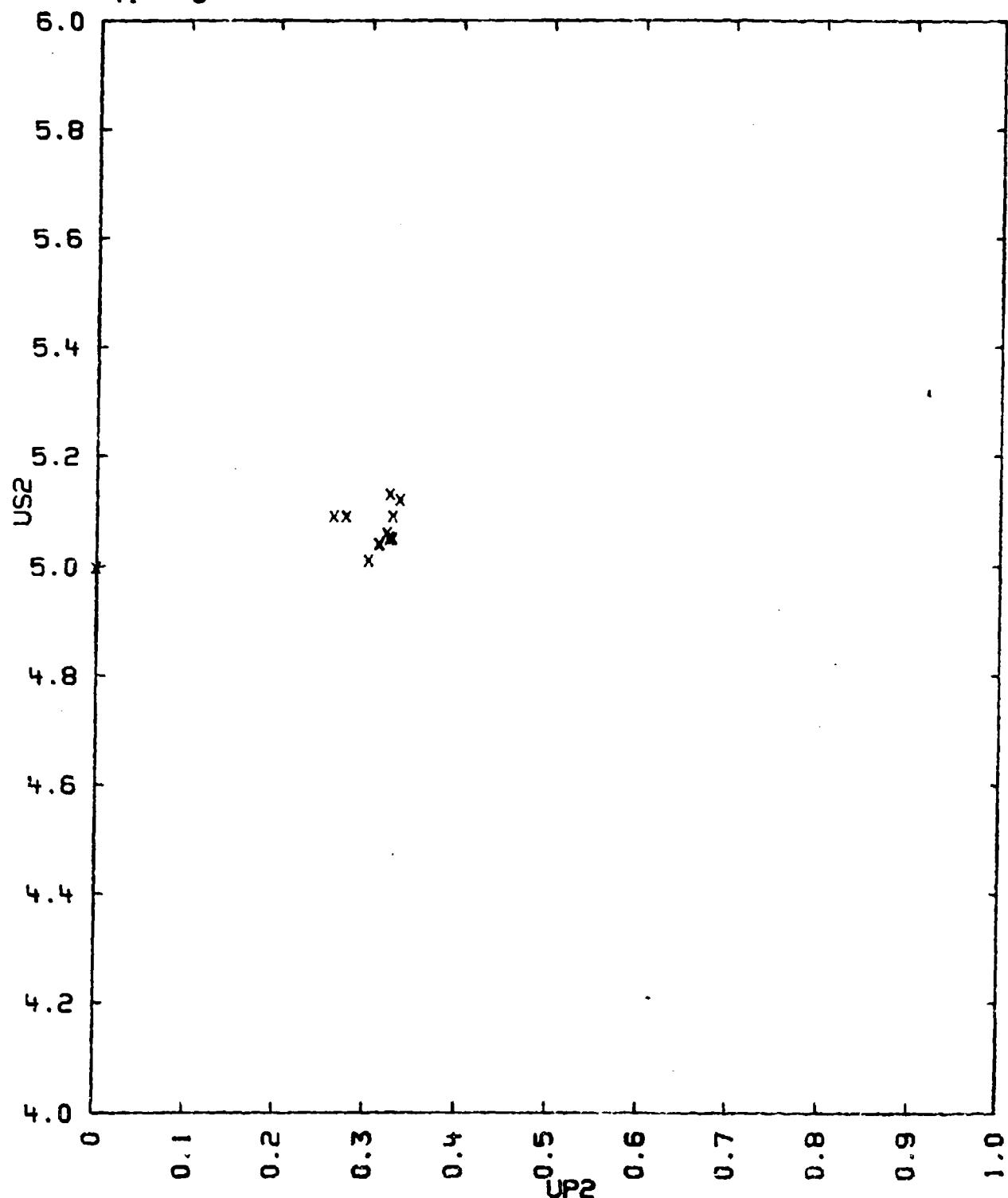
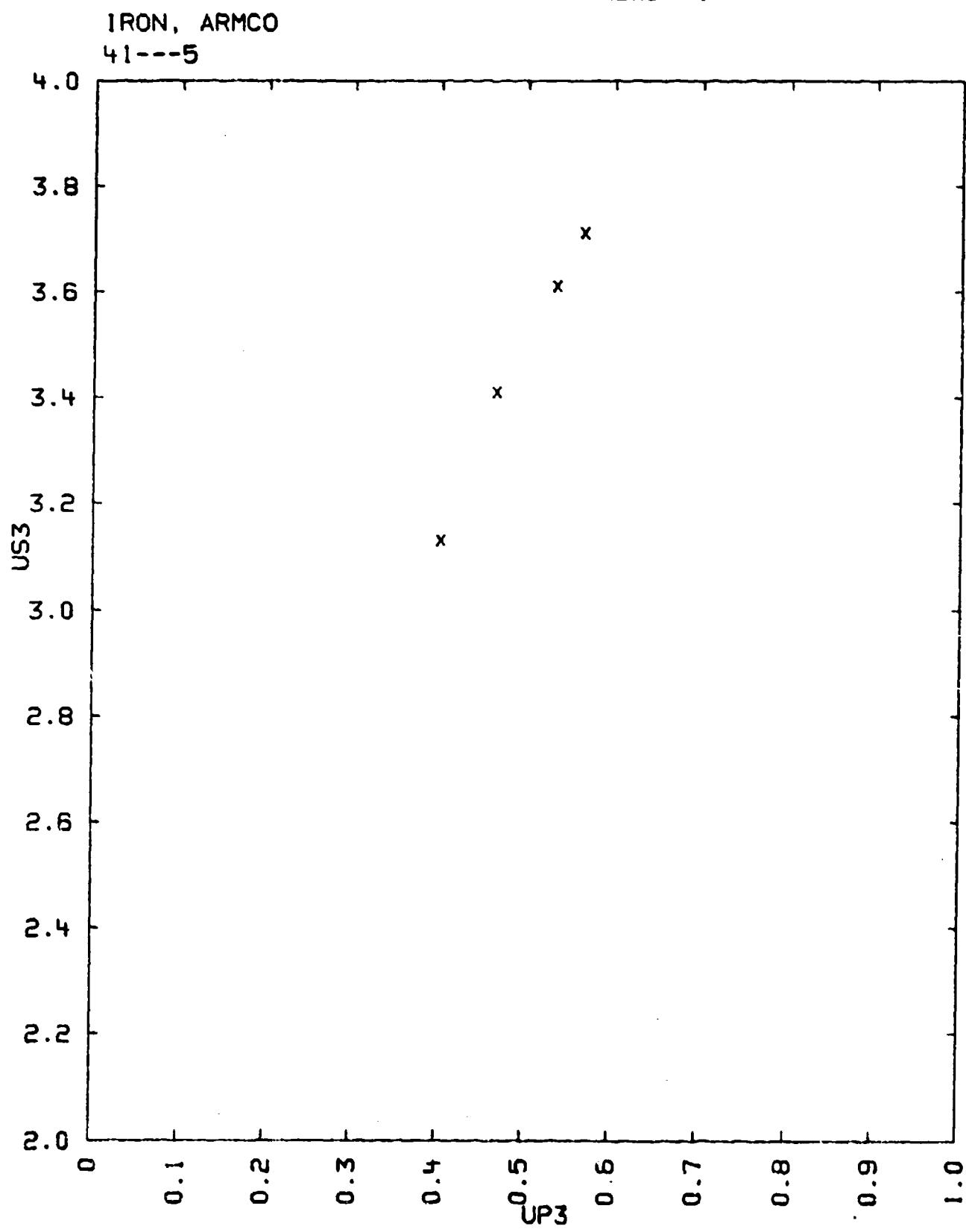


TABLE I



41---6

IRON

FE

$$V_0 = 0.1274 \text{ CC/G.}$$

$$V_{01} = 0.1272 \text{ CC/G.}$$

$$C_B = 4.65 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	U	P	V/V0
7.85	5.30	0.97	400	0.8170
-	5.39	1.00	422	0.8143
-	5.54	1.14	500	0.7943
-	7.27	2.26	1290	0.6892
-	7.54	2.38	1410	0.6645
-	8.89	3.25	2270	0.6345
-	9.36	3.56	2620	0.6196
-	9.98	3.83	3000	0.6161
-	10.45	4.20	3440	0.5981
-	10.67	4.32	3620	0.5952
-	11.10	4.59	4000	0.5865
-	11.32	4.83	4290	0.5734
-	12.00	5.17	4870	0.5691

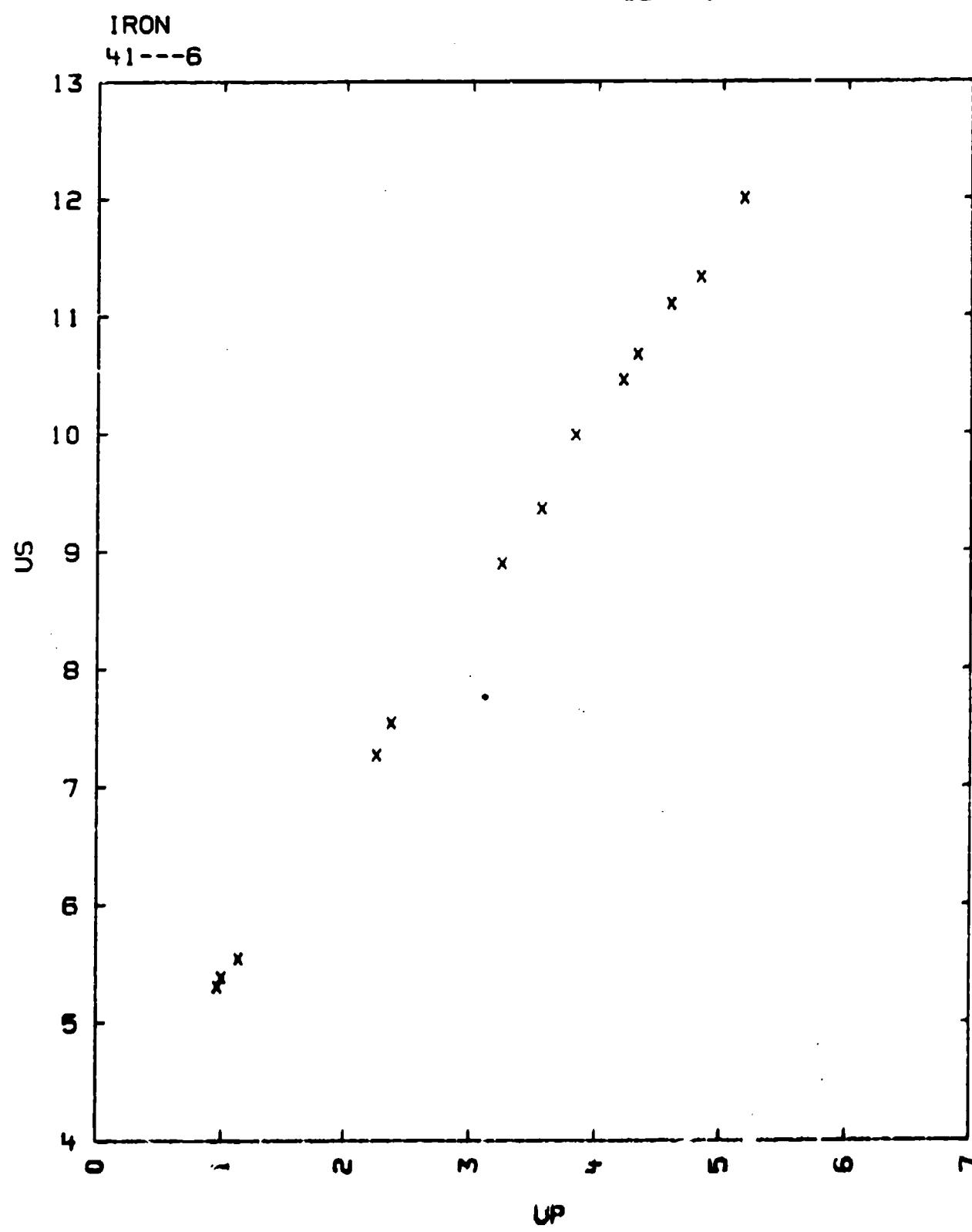
$$US = 3.74 + 1.594 UP, \quad SIG.US = 0.066 \text{ KM/SEC.}$$

FOR UP = 0.97 TO 5.17 KM/SEC.

COMMENTS:

- 1) SOURCE: AL'TSHULER, L. V., KRUPNIKOV, K. K., LEDENEV, B. N., ZHUCHIKHIN, V. I. AND BRAZHNICK M. I. SOVIET PHYS.-JETP, VOL. 34, P. 605 (1958)
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B, FOR US RANGING TO 7.54 KM/SEC.
AND TECHNIQUE A, FOR US = 8.54 AND ABOVE 8.0 KM/SEC.
- 3) V01 WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK (MCGRATH-HILL BOOK CO. N.Y., 1963) 2ND ED.
- 4) EACH VALUE OF US AND UP IS THE AVERAGE VALUE TAKEN FROM THREE OR FOUR EXPERIMENTS. THE RELATIVE SIGMA IS 1.0 PERCENT.
- 5) THE PRESSURES HERE MEASURED WITH 1.5 - 2.0 PERCENT ACCURACY.
- 6) CB IS CALCULATED USING THE VOLUME COMPRESSION COEFFICIENT TAKEN FROM P. H. BRIDGEMAN, THE PHYSICS OF HIGH PRESSURE, (G. BELL AND SONS LTD. LONDON 1958)

TABLE I



41--7

IRON

FE

$$V_0 = 0.1274 \quad C_L = 5.95 \text{ KM/SEC.} \quad C_0 = 4.627 \text{ KM/SEC.}$$

$$V_{01} = 0.1271 \text{ CC/G} \quad C_S = 3.240 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC ST DESIGNATES THE STANDARD MATERIEL USED AS STRIKER. U(ST) IS THE VELOCITY OF STRIKER BEFORE IMPACT.

TABLE

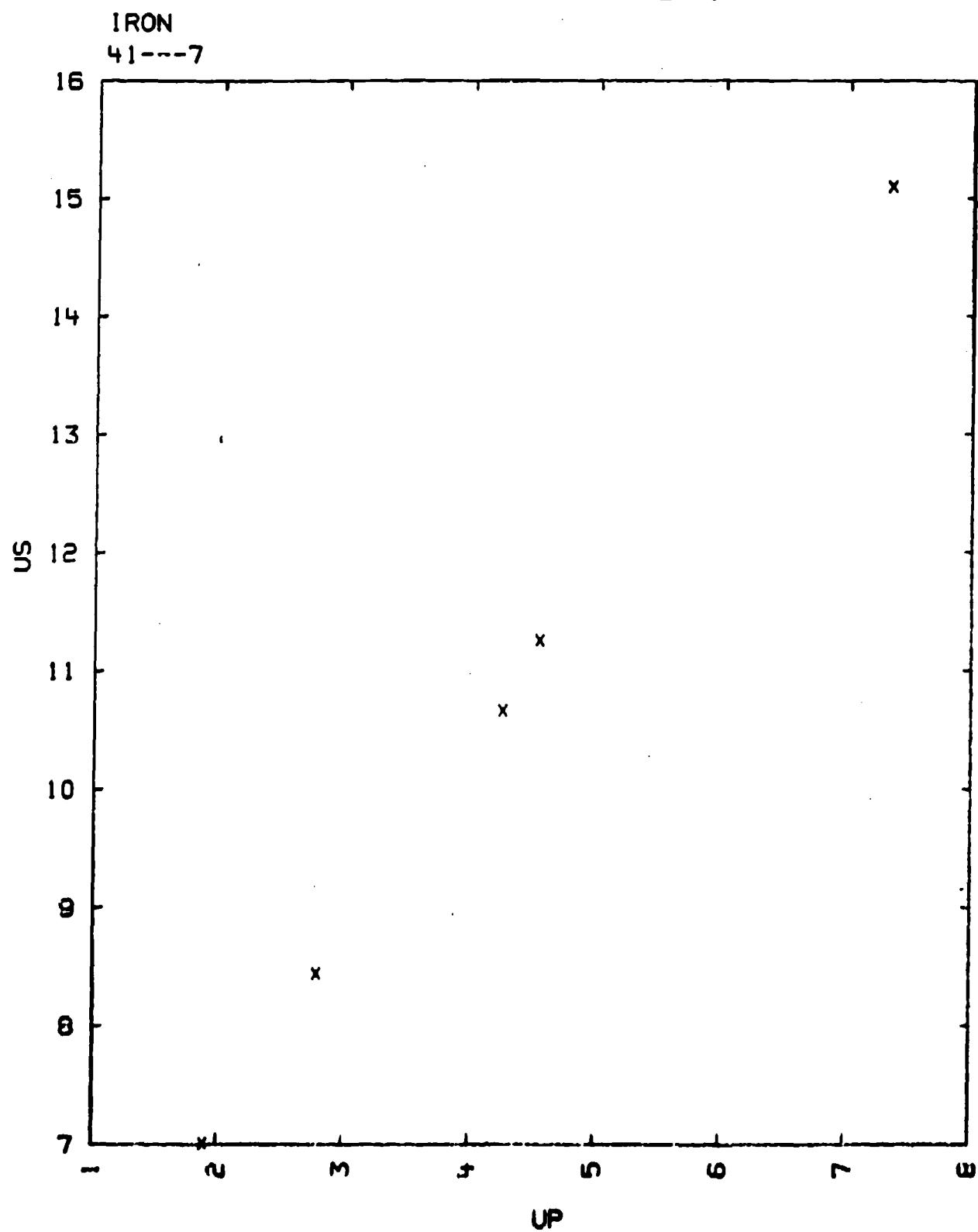
RHO0	US	UP	P	V/V0	ST	U(ST)
7.85	7.01	1.90	1045	0.7289	AL	5.60
-	8.44	2.80	1855	0.6684	FE	5.60
-	10.67	4.26	3568	0.6010	FE	8.64
-	11.26	4.55	4022	0.5959	FE	9.10
-	15.10	7.34	8700	0.5136	FF	14.68

US = $4.29 + 1.49 \text{ UP KM/SEC.}$ FOR UP BETWEEN 1.90 TO 7.34 KM/SEC.
 $\Sigma \text{SIGMA US} = 0.15 \text{ KM/SEC.}$

COMMENTS:

- 1) SOURCE: AL'TSHULER, L.V., KORMER, S.B., BAKANOVA, A.A.
AND TRUNIN, R.F.
SOVIET PHYS.-JETP, VOL. 11, P. 573 (1960)
AL'TSHULER ET AL
SOVIET PHYS.-JETP, VOL. 15, P. 65 (1962).
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE A
- 3) THE PRESSURES WERE PRODUCED BY AN EXPLOSIVELY ACCELERATED ALUMINUM OR STEEL PLATE AS INDICATED IN COLUMN 6.
- 4) FOR PRESSURES TO 4022 KILOBARS THE RELATIVE SIGMA FOR US AND UP DID NOT EXCEED 1.0 PERCENT. EACH POINT IS THE AVERAGE OF 4 TO 6 EXPERIMENTS.
- 5) THE U UNCERTAINTY OF THE VALUES IN THE LAST TABLE ENTRY ARE:
1 PERCENT IN US OBTAINED FROM 5 TO 8 EXPERIMENTS AND 1.5 PERCENT IN U(ST) OBTAINED FROM 12 EXPERIMENTS.
- 6) CL CS AND V01 WERE OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK (MCRAW-HILL BOOK CO. N.Y., 1963) 2ND ED.

TABLE I



41---8

IRON

FE

$$V_0 = 0.1274 \text{ CC/G}$$

$$V_{01} = 0.1272 \text{ CC/G.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

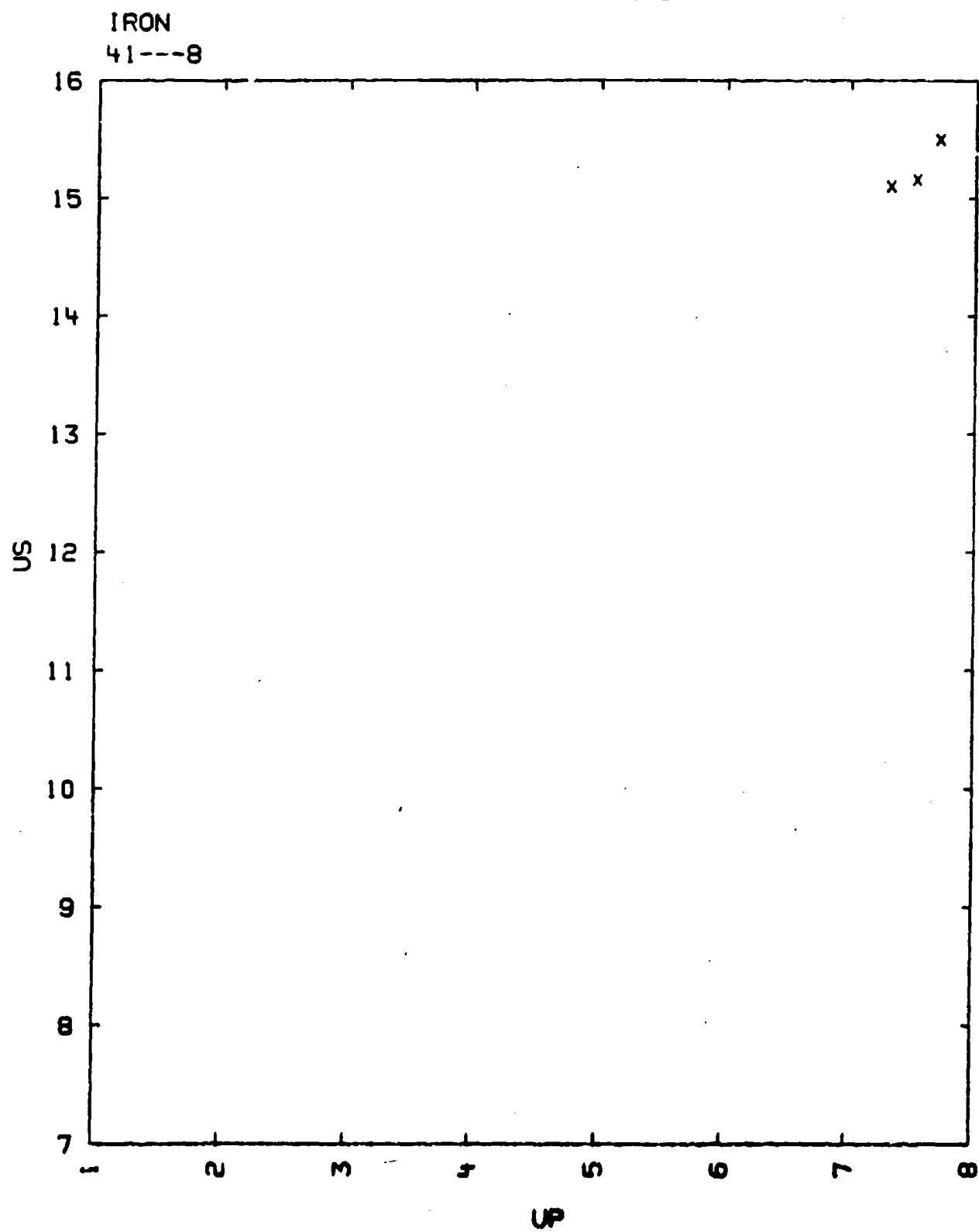
RHO0	US	UP	P	V/V0
7.85	15.50	7.71	8380	0.5025
-	15.15	7.52	8940	0.5038
-	15.10	7.32	8700	0.5139

US = $7.59 + 1.02 \text{ UP KM/SEC.}$ FOR UP BETWEEN 7.71 TO 7.32 KM/SEC.
 $\Sigma \text{US} = 0.12 \text{ KM/SEC.}$

COMMENTS:

- 1) SOURCE: KRUPNIKOV, K.K., BAKANOVA, A.A., BRAZHNICK, M.I.
AND TRUNIN, R.F.
SOVIET PHYS.-JETP, VOL. 8, P. 205 (1963)
- 2) EXPERIMENTAL TECHNIQUE A (UNCERTAIN, NOT DISCUSSED)
DATA REDUCTION TECHNIQUE A FOR LAST ENTRY, THE REST IS NOT DISCUSSED
- 3) V01 WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK
(MCGRAW-HILL BOOK CO. 1963) 2ND ED.

TABLE I



41--9
IRON

FE

$$V_0 = 0.128 \text{ CC/G.}$$

$$V_{01} = 0.1272 \text{ CC/G.}$$

THE TABLES BELOW GIVE THE VELOCITIES OF RELAXATION WAVES, C, AT VARIOUS PRESSURES. THE HUGONIOT STATE THE RELAXATION WAVE TRAVELS THROUGH IS GIVEN BY US, UP, P, OR V/V0. DENSITY IS GIVEN IN G/CC, VELOCITY IN KM/SEC AND PRESSURE IN KILOBARS. IN TABLE II, ST DESIGNATES THE MATERIAL OF THE STRIKER PLATE, U(ST) IS THE VELOCITY BEFORE IMPACT AND US(ST) IS THE SHOCK VELOCITY IN THE PLATE AFTER IMPACT.

TABLE I

SOUND VELOCITIES
PERPENDICULAR TO SHOCK DIRECTION

RHO0	US	UP	P	V/V0	C
7.81	5.34	0.98	409	0.816	7.15
US =					

TABLE II

SOUND VELOCITY
PARALLEL TO SHOCK DIRECTION

---HUGONIOT STATE----- --DRIVER SYSTEM--

RHO0	US	UP	P	V/V0	C	ST	U(ST)	US(ST)
7.81	7.01	1.91	1045	0.7278		AL	5.60	10.31
-	5.76	1.20	542	0.7918	6.70	AL	5.60	10.31
-	5.34	0.98	415	0.8143	6.12	AL	5.60	10.31
-	8.57	2.88	1914	0.6858		FE	5.71	8.54
-	8.21	2.62	1680	0.6793	8.49	FE	5.71	8.54
-	7.38	2.12	1223	0.7117	7.85	FE	5.71	8.54
-	10.7	4.15	3478	0.6081	9.98	FE		
-	9.86	3.70	2849	0.6250	9.53	FE		

US =

COMMENTS:

- 1) SOURCE: AL'TSHULER, L. V., KORMER, S. B., BRAZNIK, M. I., VLADIMIROV, L. A. AND FUNTIKOV, A. I.
SOVIET PHYS.-JETP, VOL. 11, P. 768 (1960)
2) EXPERIMENTAL TECHNIQUE A WAS USED IN TABLE II FOR THE FIRST THREE

ENTRIES. FOR ALL OTHER ENTRIES EXPERIMENTAL TECHNIQUE B WAS USED.
THE HUGONIOT STATE WAS DETERMINED FROM ONE MEASURED PARAMETER
AND THE KNOWN EQUATION OF STATE. AL'TSHULER ET AL., SOVIET PHYS-JETP,
VOL. 11, P. 573 (1960).

- 3) THE SOUND VELOCITY IN TABLE I WAS OBTAINED FROM THE SHAPE OF THE FREE SURFACE AFTER IT HAD MOVED SOME DISTANCE. IN TABLE II THE SOUND VELOCITY WAS DETERMINED FROM THE ATTENUATION IN THE MEASURED SHOCK OR FREE SURFACE VELOCITY PRODUCED BY THE RAREFACTION WAVE ORIGINATING AT THE REAR SURFACE OF THE STRIKER PLATE.
- 4) VOI WAS OBTAINED FROM THE HANDBOOK OF CHEMISTRY AND PHYSICS (THE CHEMICAL RUBBER PUBLISHING CO. CLEVELAND, OHIO, 1962-1963) 44TH ED.
- 5) THE MEASURED SOUND VELOCITY IS HIGHER THAN THE CALCULATED SOUND VELOCITY FOR A PURELY NONELASTIC SAMPLE.

TABLE I

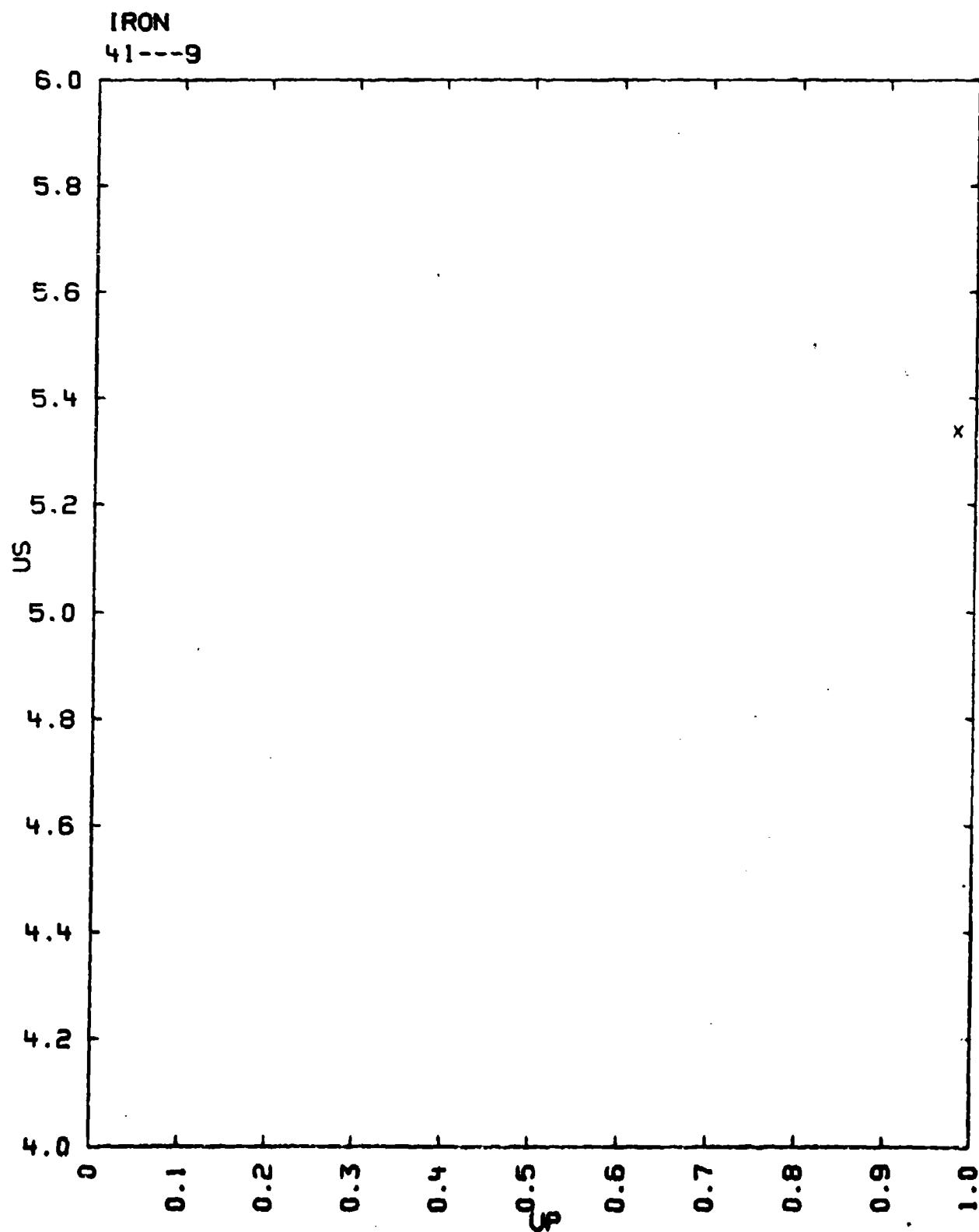
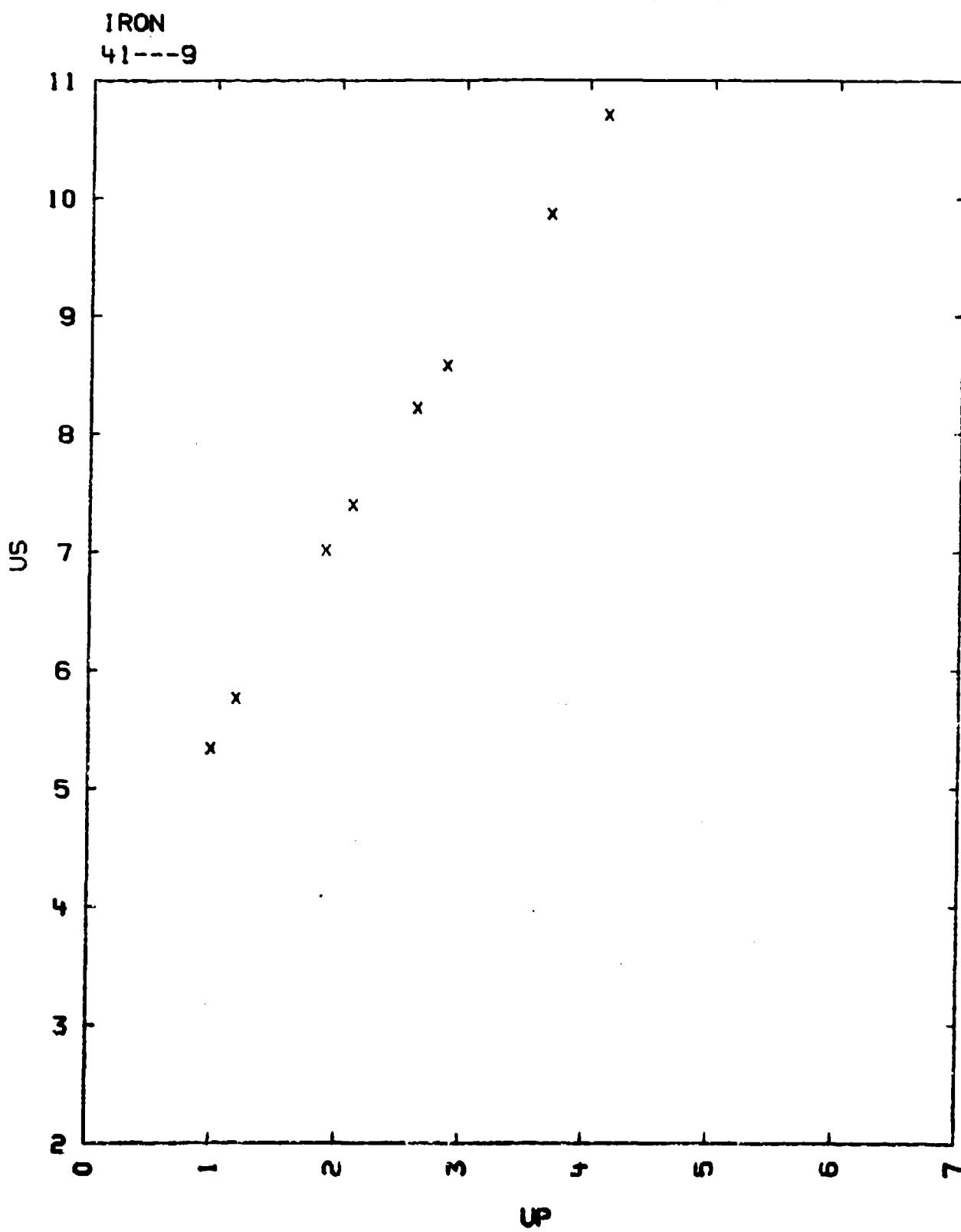


TABLE II



41---10
STEEL, 1020 ALLOY (IRON)

		WEIGHT PERCENT
C	0.18-0.23	
MN	0.30-0.60	-
P	0.040	MAXIMUM
S	0.050	"
FE	REMAINDER	-

$$V_0 = 0.1276 \text{ CC/G} \quad C_L = 5.93 \text{ KM/SEC}$$

TABLE I LISTS LONGITUDINAL SOUND VELOCITY, SHOCK VELOCITY AND PARTICLE VELOCITY IN KM/SEC. THIS DOUBLE WAVE SYSTEM CORRESPONDS TO THE PRESSURES AND COMPRESSIONS IN TABLE II. THE LINE NUMBERS ARE THOSE OF THE SOURCE. D IS THE SAMPLE THICKNESS IN MM.

TABLE I

NO	CL	US1	UP1	US2	UP2	D
1	5.94	6.02	0.0240	5.08	0.1900	127.01
2	-	5.93	0.0256	5.09	0.2700	127.01
3	-	6.03	0.0239	5.11	0.1700	127.00
4	5.91	6.08	0.0264	5.12	0.245	127.01
12	5.93	5.95	0.0330	5.09	0.318	51.01
13	-	5.95	0.0270	5.09	0.317	50.90
14	5.91	5.98	0.0264	5.08	0.2228	126.98
15	-	-	-	5.05	0.342	26.86

US -

TABLE II LISTS THE DENSITY IN G/CC AND PRESSURE IN KBARS. THE SIGNIFICANCE OF THE THERMAL HISTORY (THIST) IS GIVEN IN THE COMMENTS. AS IN TABLE I THE LINE NUMBERING FOLLOWS THAT OF THE SOURCE.

TABLE II

NO	RHO0	P1	V1/V0	P2	V2/V0	THIST
1	7.037	11.3	0.9981	77.4	0.9832	AR
2	-	11.9	0.9953	109.5	0.9475	-
3	-	11.3	0.9981	69.8	0.9870	-
4	7.036	12.6	0.9960	100.1	0.9529	-
12	7.037	15.4	0.9945	129.2	0.9381	-
13	-	12.6	0.9953	128.3	0.9381	ANN
14	7.039	12.4	0.9955	90.5	0.9571	-
15	-	-	-	137.1	0.9477	-

COMMENTS

UOG/14/77

- 1) SOURCE: MINSHALL, F.S.
MET. SOC. CONF., VOL. 9, P. 249 (1961)
J. APPL. PHYS., VOL. 26, P. 463 (1955)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION METHOD D, WITH 2UP = UFS
- 3) THE THIST SYMBOLS ARE DEFINED AS FOLLOWS
AR - AS RECEIVED
ANN - ANNEALED
- 4) IN TABLE II ONLY VALUES OF P2 ABOVE 128 KBARS ARE TRANSITION POINTS

TABLE I

STEEL, 1020 ALLOY (IRON)

41---10

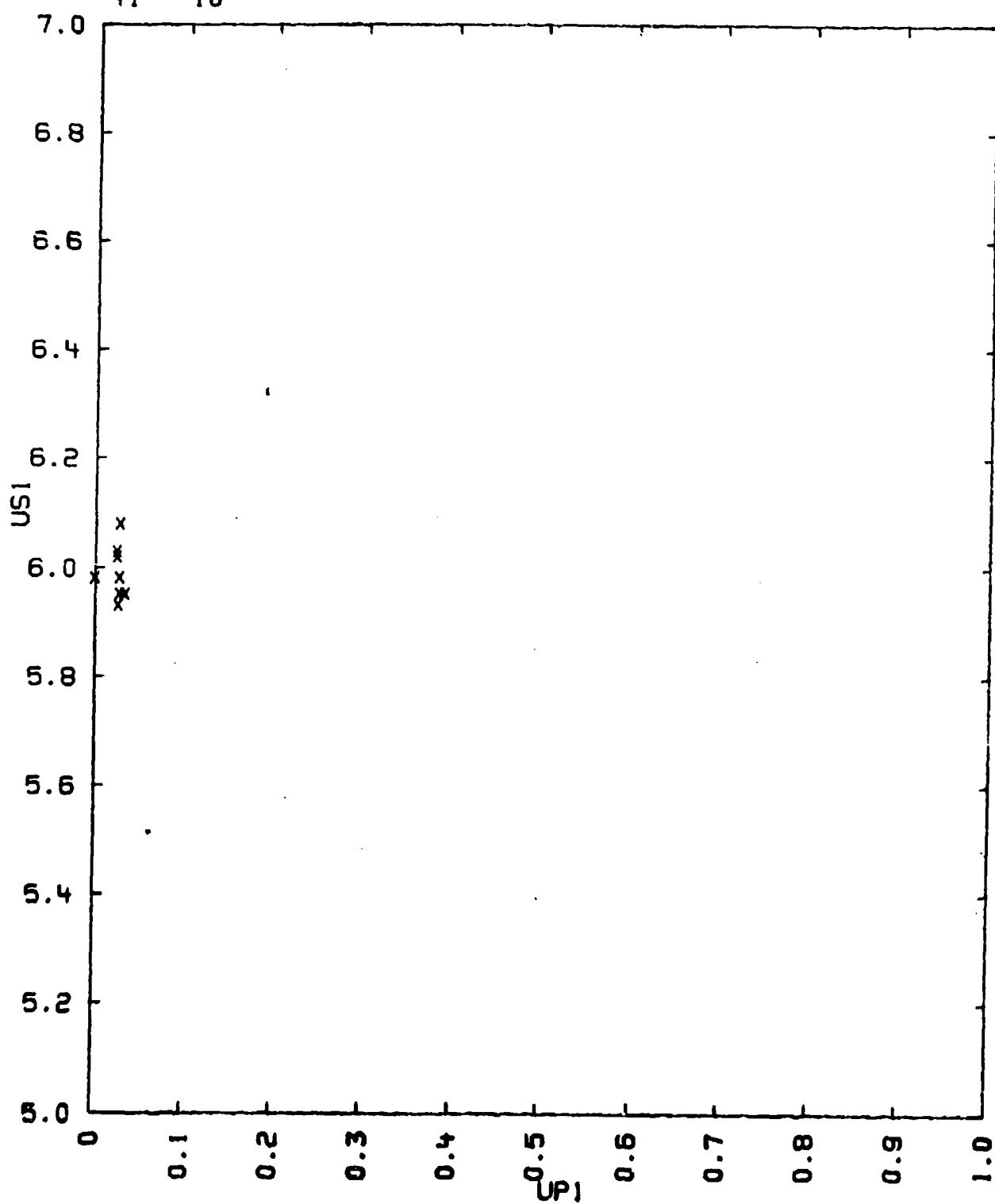
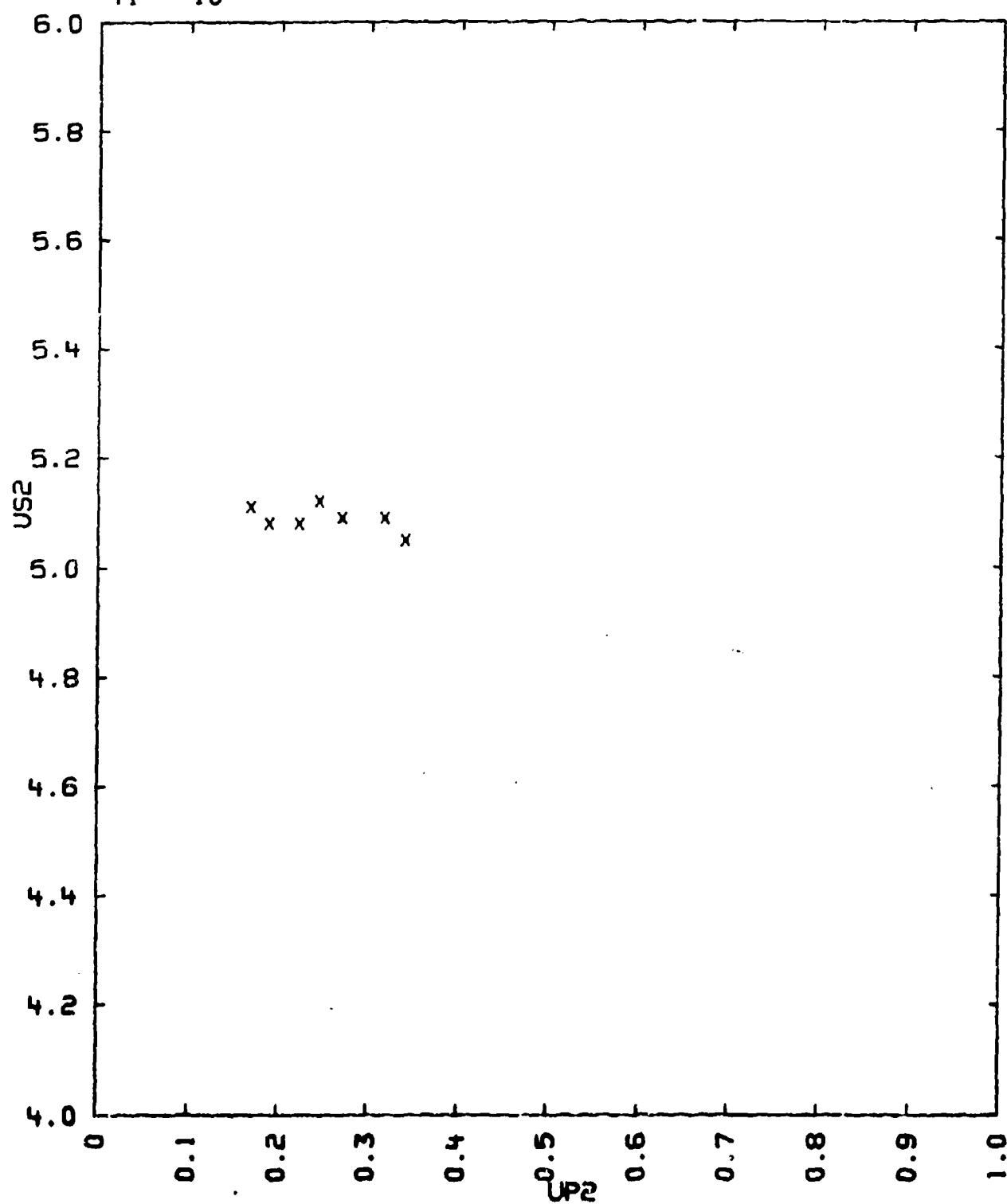


TABLE I

STEEL, 1020 ALLOY (IRON)

41---10



41---11
STEEL, 1040 ALLOY (IRON)

C	0.37-0.44	WEIGHT PERCENT
MN	0.60-0.90	-
P	0.040 MAX.	-
S	0.050	-
FE	REMAINDER	-

$$V_0 = 0.1277 \text{ CC/G} \quad CL = 5.92 \text{ KM/SEC}$$

TABLE I LISTS LONGITUDINAL SOUND VELOCITY, SHOCK AND PARTICLE VELOCITY IN KM/SEC. THIS DOUBLE WAVE SYSTEM CORRESPONDS TO THE PRESSURES AND COMPRESSIONS IN TABLE II. THE LINE NUMBERING FOLLOWS THE SOURCE. D IS THE SAMPLE THICKNESS IN MM.

TABLE I

NO	CL	US1	UP1	US2	UP2	D
5	5.92	5.97	0.0124			127.02
-	-	5.54	0.0248	5.18	0.2500	- SEE COMMENTS
16	5.93	5.99	0.0294	5.10	0.2240	126.94
17	-	-		5.06	0.3364	26.84
20	5.89	5.96	0.0268			127.26
-	-	5.51	0.0326	5.11	0.2724	-

US =

TABLE II

NO	RHO0	P1	V1/V0	P2	V2/V0	THIST
5	7.818	5.8	0.9976			AR
-	-	11.2	0.9960	102.4	0.9522	-
16	7.833	13.8	0.9964	91.5	0.9572	ANN
17	-	-	-	135.9	0.9345	-
20	7.834	12.5	0.9957			AR
-	-	15.0	0.9941	111.0	0.9471	-

COMMENTS

- 1) SOURCE: MINSHALL, F.S.
MET. SOC. CONF., VOL. 9, P. 249 (1961)
J. APPL. PHYS., VOL. 26, P. 463 (1955)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION METHOD D, WITH $U_{UP} = U_{FS}$.
- 3) THE THIST SYMBOLS ARE DEFINED AS

AR = AS RECEIVED
ANN = ANNEALED

- 4) IN THE ABOVE TABLE ALL VALUES OF P2 ARE BELOW THE TRANSITION EXCEPT THE ONE FOR EXPERIMENT 17 WHICH INDICATES A RELATIVELY HIGH TRANSITION POINT
- 5) THE FIRST WAVE IN EXPERIMENT 5 AND 20 APPEARED TO BE SPLIT INTO TWO PRESSURE STEPS. P2 WAS THEREFORE CALCULATED AS THE THIRD WAVE IN A THREE WAVE SYSTEM. NOTE THAT THIS DID NOT OCCUR IN THE ANNEALED SAMPLES

TABLE I

STEEL, 1040 ALLOY (IRON)

41--11

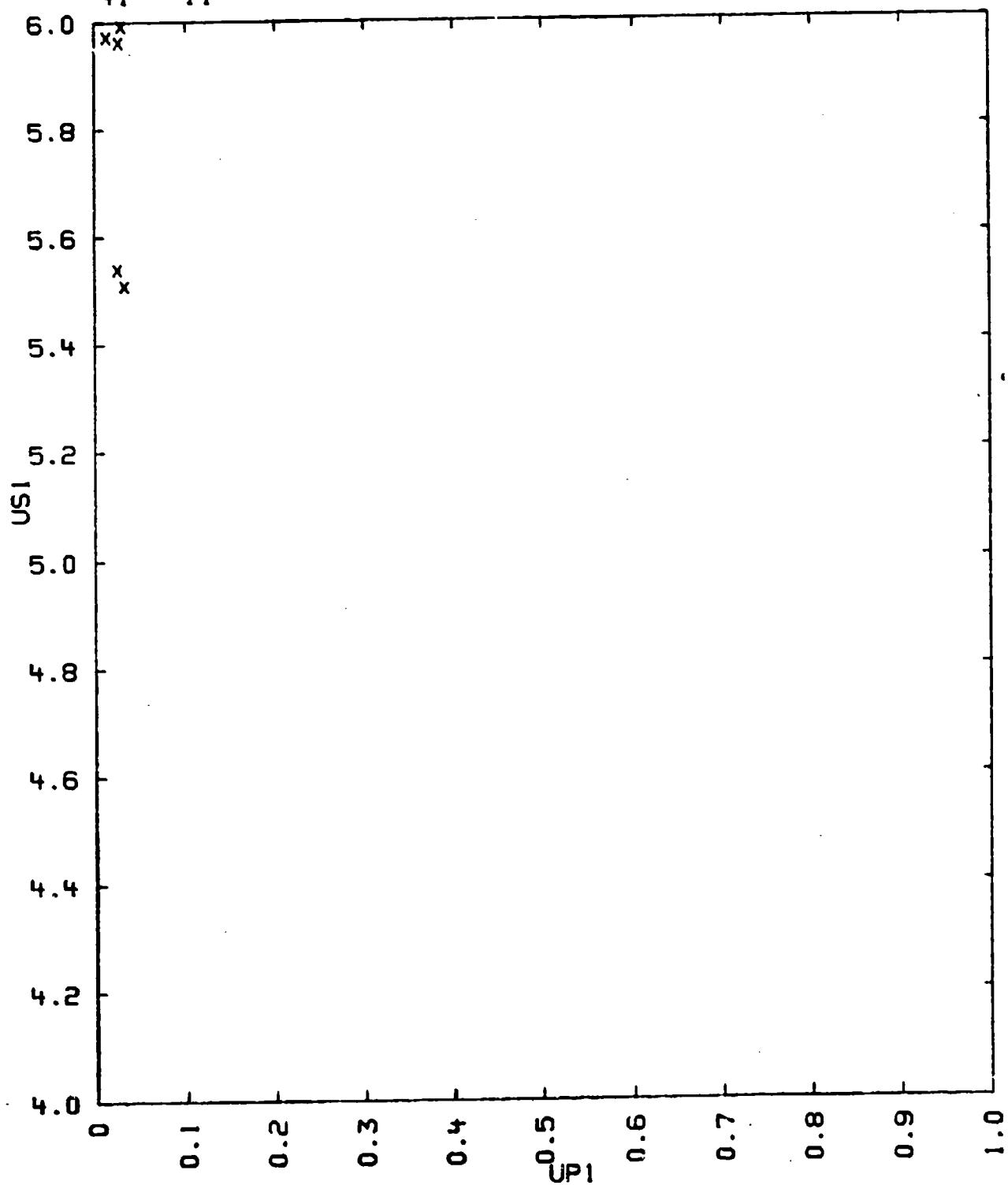
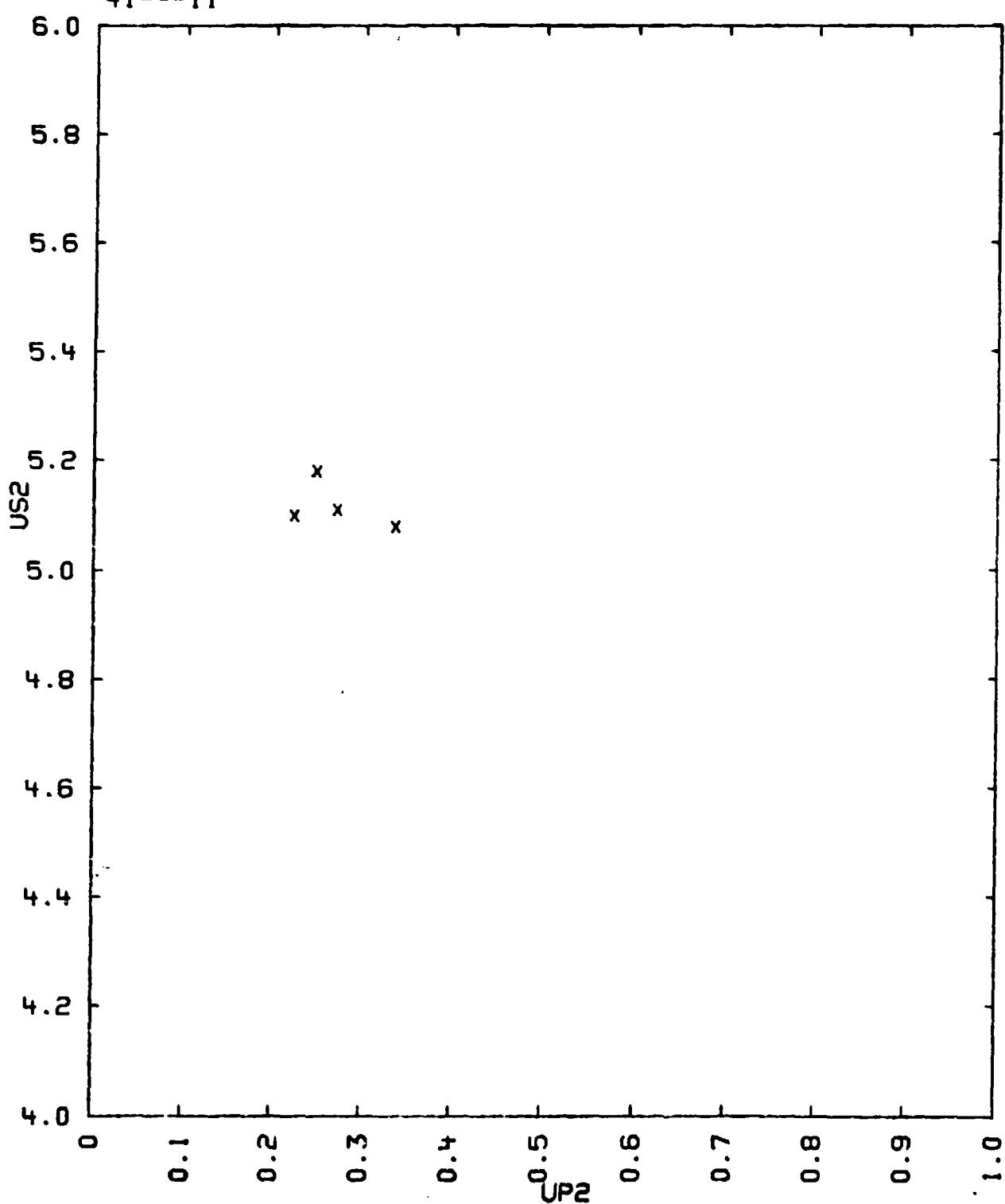


TABLE I

STEEL, 1040 ALLOY (IRON)

41---11



41---12
STEEL, 1055 ALLOY (IRON)

C	0.30-0.60	WEIGHT PERCENT
MN	0.60-0.90	-
P	0.040 MAX.	-
S	0.050	-
FE	REMAINDER	

$$V_0 = 0.1278$$

TABLE I LISTS LONGITUDINAL SOUND VELOCITY, SHOCK AND PARTICLE VELOCITY IN KM/SEC. THIS DOUBLE WAVE SYSTEM CORRESPONDS TO THE PRESSURES AND COMPRESSIONS IN TABLE II. THE LINE NUMBERING FOLLOWS THE SOURCE. D IS THE SAMPLE THICKNESS IN MM.

TABLE I

NO	CL	US1	UP1	US2	UP2	D
18		5.98	0.0310	5.13	0.2305	127.08
19		-	-	5.11	0.3465	26.82
21	5.90	5.99	0.0294	5.13	0.2249	127.24
22	5.93	6.01	0.0302			127.04
23	5.90	6.00	0.0319			124.64

US =

TABLE II

NO	RHO0	P1	V1/V0	P2	V2/V0	THIST
18	7.824	14.5	0.9952	94.5	0.9561	ANN
19	-	-	-	140.5	0.9326	ANN
21	7.825	13.3	0.9953	92.1	0.9570	AR
22	-	14.2	-			ANN
23	-	15.0	0.9946			AR

COMMENTS

- 1) SOURCE: MINSHALL, F.S.
MET. SOC. CONF., VOL. 9, P. 249 (1961)
- 2) EXPERIMENTAL TECHNIQUE A FOR LINE 18, 19 AND 21
G FOR LINE 22 AND 23
DATA REDUCTION METHOD D WITH $2U_P = U_F S$
- 3) THE THIST SYMBOLS ARE DEFINED AS FOLLOWS
 - AR = AS RECEIVED
 - ANN = ANNEALED
- 4) ONLY THE 140.5 KBAR POINT QUALIFIES AS A TRANSITION POINT

TABLE I

STEEL, 1055 ALLOY (IRON)

41---12

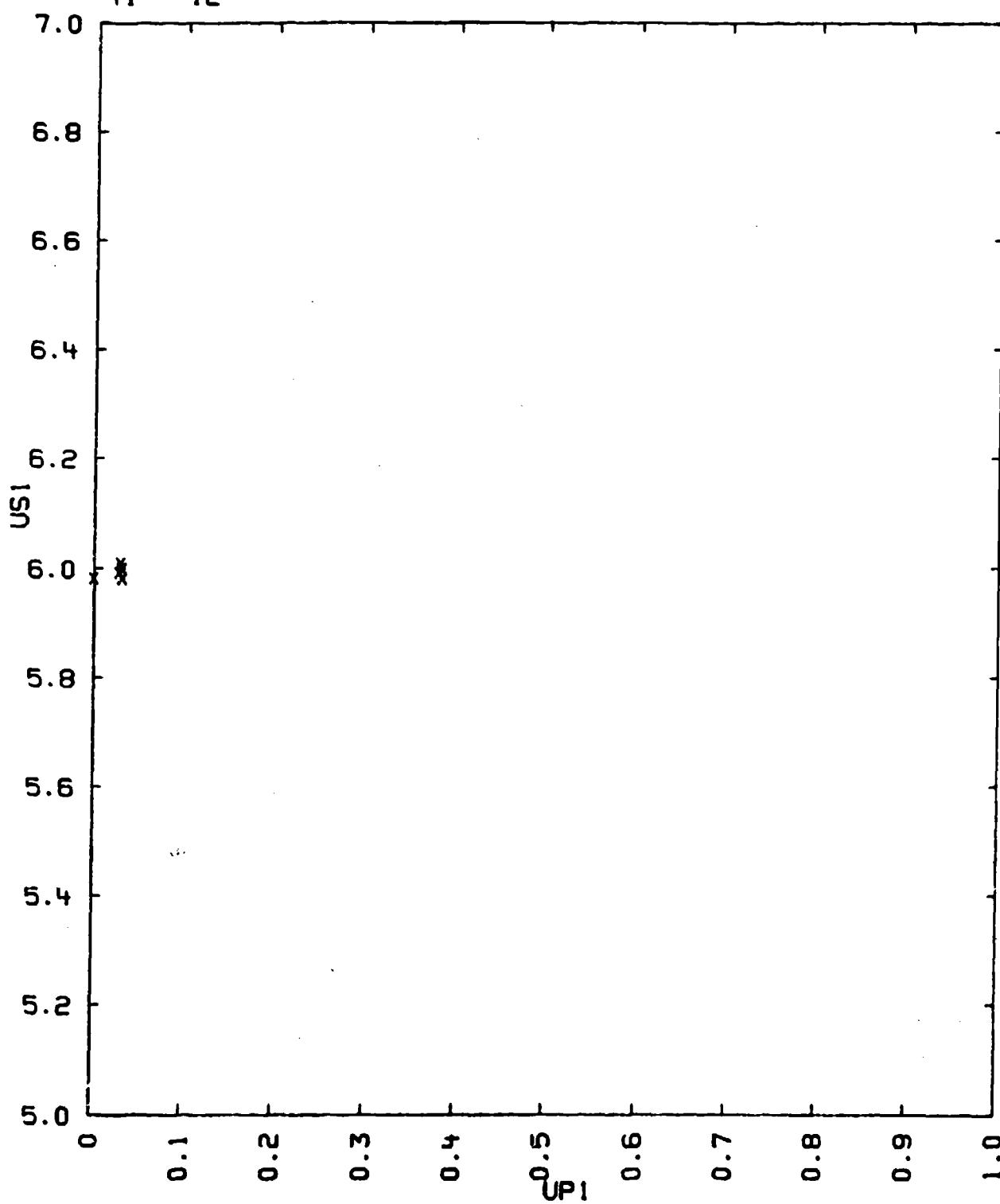
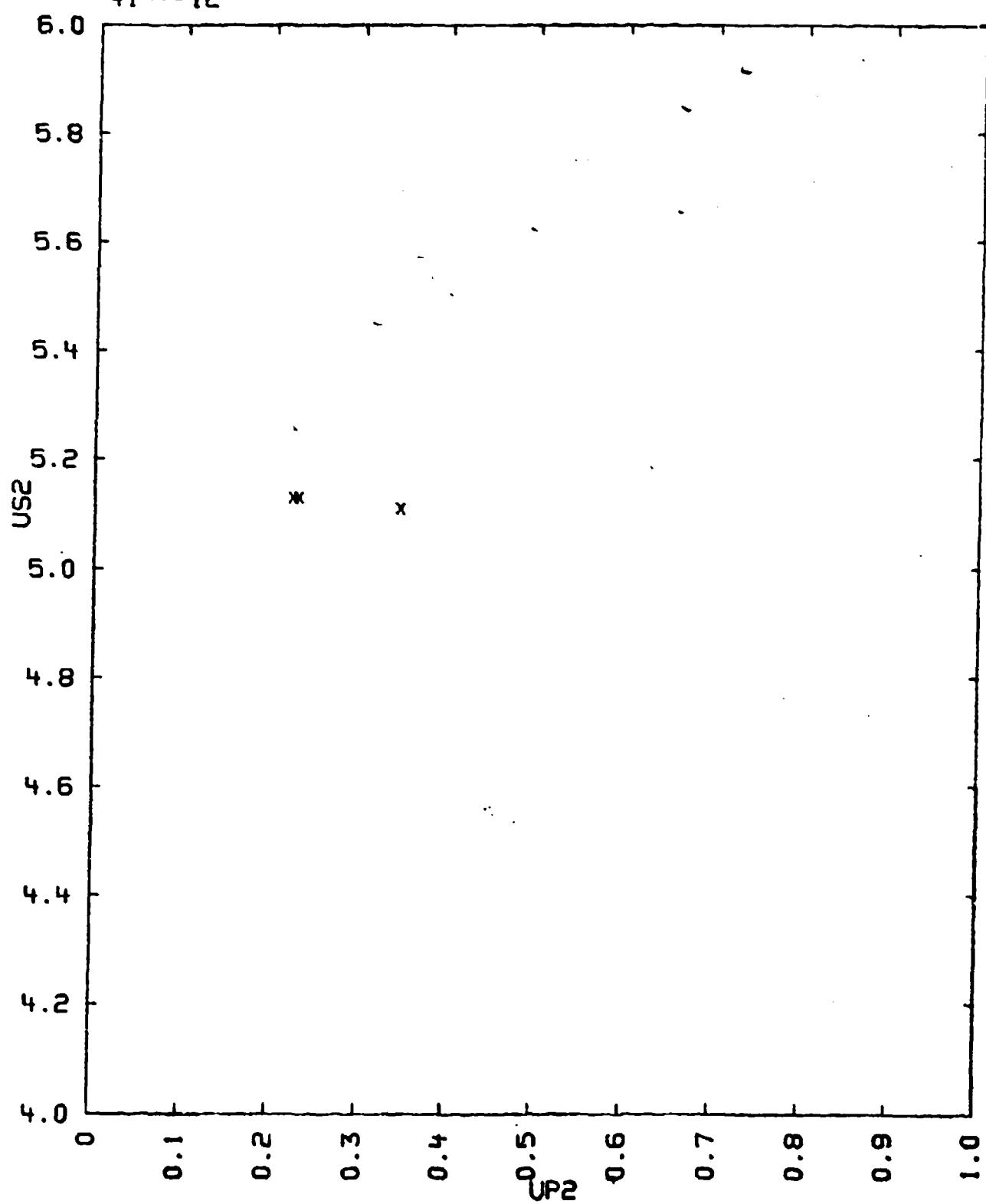


TABLE I

STEEL, 1055 ALLOY (IRON)

41---12



41---13
IRON, ARMCO

FE 99.99 PERCENT

TO = VARIABLE
VOL = 0.1270 CC/G AT 25 DEG. C

THE TABLE LISTS THE TEMPERATURES IN DEG. K. AND THE PRESSURES IN KBARS OF TWO IRON PHASE LINES.

PHASE TRANSFORMATION POINTS

TABLE

TO	P
78	150
201	140
248	134
298	128
298	131
360	121
421	124
481	124
540	124
543	121
600	113
660	113
723	115
773	111
783	115
793	105
813	93
833	87
853	87
875	59
975	48
1023	39
1073	32
1158	19

COMMENTS:

- 1) SOURCE: JOHNSON, P. C., STEIN, B. A. AND DAVIS, R. S.
J. APPL. PHYS., VOL. 33, P. 557 (1962)
 - 2) EXPERIMENTAL TECHNIQUE: MICROSCOPIC EXAMINATION OF RECOVERED- SHOCKED SAMPLES AND DETERMINATION OF THE BOUNDARY OF THE TRANSFORMED PHASE. THE PRESSURE ATTENUATION CURVE HAS ALSO DETERMINED IN THIS WORK. TEMPERATURES LISTED ARE THOSE MEASURED BEFORE SHOCK COMPRESSION.
- DATA REDUCTION METHOD: COMPARISON OF LOCATION OF THE BOUNDARY BETWEEN

TRANSFORMED AND UNTRANSFORMED MATERIAL WITH
THE PRESSURE ATTENUATION CURVE YIELDED THE
ABOVE P-T LINE AFTER NORMALIZING TO YIELD THE
132 KBAR AT 300 DEG K POINT OF MINSHALL.

- 3) A SHARP BREAK IN THE PHASE LINE AT ABOUT 770 DEG. K. CORRELATES WELL
WITH A DRASIC CHANGE IN THE MICROSTRUCTURE OF THE TRANSFORMED REGION
- 4) UNCERTAINTY IN THE PRESSURE DETERMINATION APPEARS TO BE THE DETERMI-
NING FACTOR. MAXIMUM DEVIATION 7 KBAR

NOMINAL - 2-3 KBAR

41--14
IRON (MILD STEEL)

C = FROM 0.40 TO 0.10 PERCENT
 S = LESS THAN 0.06 PERCENT
 P = LESS THAN 0.06 PERCENT
 FE = REMAINDER

$$V_0 = 0.1278 \text{ CC/G.}$$

$$V_{01} = 0.1270 \text{ CC/G.}$$

IN TABLE I BELOW, DENSITY IS GIVEN G/CC. AND VELOCITIES IN KM/SEC.,
 ID IS IDENTIFICATION BETWEEN TABLES. 1, 2, AND 3 DESIGNATE FIRST,
 SECOND, AND THIRD SHOCKS.

IN TABLE II, PRESSURE IS GIVEN IN KILOBARS.

TABLE I

ID	RHO0	US1	UP1	US2	UP2	US3	UP3
1	7.84	5.85	0.027	5.13	0.285		
2	-	-	-	-	0.29		
3	-	-	-	-	0.295		
4	-	-	-	-	0.30		
5	-	-	-	-	0.31		
6	-	-	-	-	0.315		
7	-	-	-	-	0.32		
8	-	-	-	-	0.325		
9	-	-	-	-	0.32	*	
10	-	-	-	-	-	3.25	0.34
11	-	-	-	-	-	3.43	0.345
12	-	-	-	-	-	3.24	0.355
13	-	-	-	-	-	3.30	0.36
14	-	-	-	-	-	3.60	0.365
15	-	-	-	-	-	3.36	0.37
16	-	-	-	-	-	3.60	0.38
17	-	-	-	-	-	3.47	0.38
18	-	-	-	-	-	3.57	0.40
19	-	-	-	-	-	3.77	0.42
20	-	-	-	-	-	3.75	0.425
21	-	-	-	-	-	3.58	0.43
22	-	-	-	-	-	3.76	0.48
23	-	-	-	-	-	3.89	0.485
24	-	-	-	-	-	3.87	0.49
25	-	-	-	-	-	3.87	0.51
26	-	-	-	-	-	4.16	0.52
27	-	-	-	-	-	4.13	0.51
28	-	-	-	-	-	4.00	0.52
29	-	-	-	-	-	4.12	0.525
30	-	-	-	-	-	4.26	0.58
31	-	-	-	-	-	4.27	0.585
32	-	-	-	-	-	4.24	0.59
33	-	-	-	-	-	4.42	0.62
34	-	-	-	-	-	4.54	0.67
35	-	-	-	-	-	4.85	0.70
36	-	-	-	-	-	4.75	0.71
37	-	-	-	-	-	4.70	0.74

IRON (MILD STEEL)

ID	RHO0	US1	UP1	US2	UP2	US3	UP3
38	-	-	-	-	-	4.90	0.76
39	-	-	-	-	-	5.00	0.77
40	-	-	-	-	-	4.98	0.78
41	-	-	-	-	-	5.05	0.785

US3 = 2.05 +3.79 UP KM/SEC.

TABLE II

ID	P1	V1/V0	P2	V2/V0	P3	V3/V0
1	12	0.9950	115	0.9444		
2	-	-	117	0.9435		
3	-	-	119	0.9425		
4	-	-	121	0.9415		
5	-	-	125	0.9395		
6	-	-	127	0.9385		
7	-	-	129	0.9376		
8	-	-	131	0.9366		
9	-	-	129	0.9376		
10					134	0.9312
11	-	-	-	-	136	0.9301
12	-	-	-	-	138	0.9263
13	-	-	-	-	139	0.9250
14	-	-	-	-	142	0.9247
15	-	-	-	-	142	0.9222
16	-	-	-	-	146	0.9204
17	-	-	-	-	145	0.9197
18	-	-	-	-	151	0.9145
19	-	-	-	-	159	0.9087
20	-	-	-	-	160	0.9089
21	-	-	-	-	160	0.9059
22	-	-	-	-	176	0.8940
23	-	-	-	-	179	0.8943
24	-	-	-	-	180	0.8927
25	-	-	-	-	187	0.8875
26	-	-	-	-	194	0.8888
27	-	-	-	-	191	0.8908
28	-	-	-	-	192	0.8866
29	-	-	-	-	195	0.8870
30	-	-	-	-	216	0.8757
31	-	-	-	-	218	0.8747
32	-	-	-	-	219	0.8730
33	-	-	-	-	233	0.8689
34	-	-	-	-	253	0.8598
35	-	-	-	-	273	0.8589
36	-	-	-	-	274	0.8551
37	-	-	-	-	284	0.8477
38	-	-	-	-	298	0.8475
39	-	-	-	-	305	0.8474
40	-	-	-	-	309	0.8450

COMMENTS

- 1) SOURCE: LETHABY AND SKIDMORE
PRIVATE COMMUNICATION (1965)
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE D, ASSUMING ZUP = UFS.
- 3) VOF WAS CALCULATED FOR PURE IRON USING A LATTICE CONSTANT OF 2.86645
ANGSTROMS AT 20 DEG. CENTIGRADE.
A.C.A. MONOGRAPH NUMBER 5 (AMERICAN CRYSTALLOGRAPHIC ASSOCIATION,
POLYCRYSTAL BOOK SERVICE 1963) 2ND. ED.
4. • THIS REPRESENTS THE ALPHA TO EPSILON TRANSITION POINT.
THIS IS THE AVERAGE VALUE OBTAINED FROM 21 EXPERIMENTS.

TABLE I

IRON (MILD STEEL)

41---14

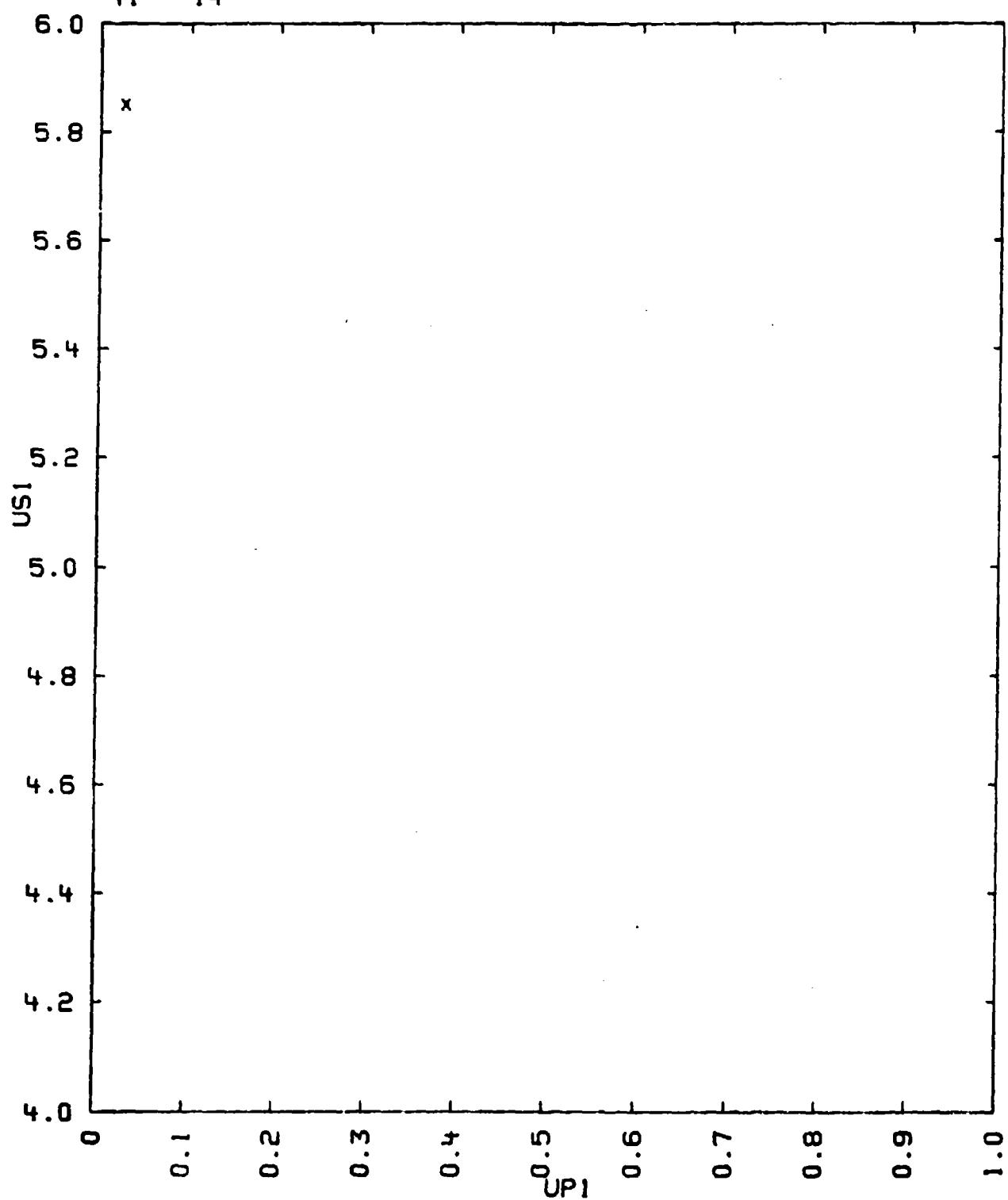


TABLE I

IRON (MILD STEEL)

41---14

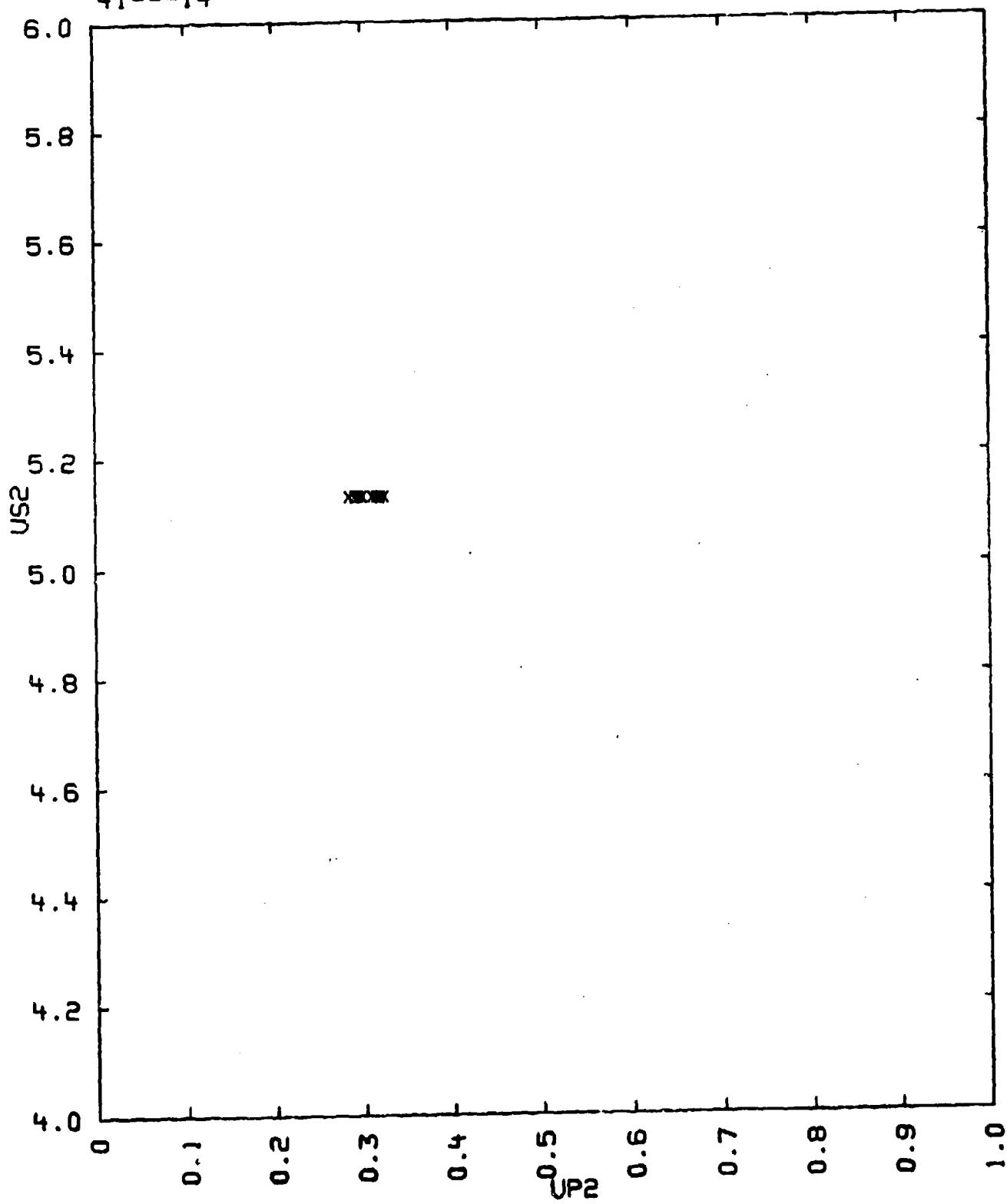
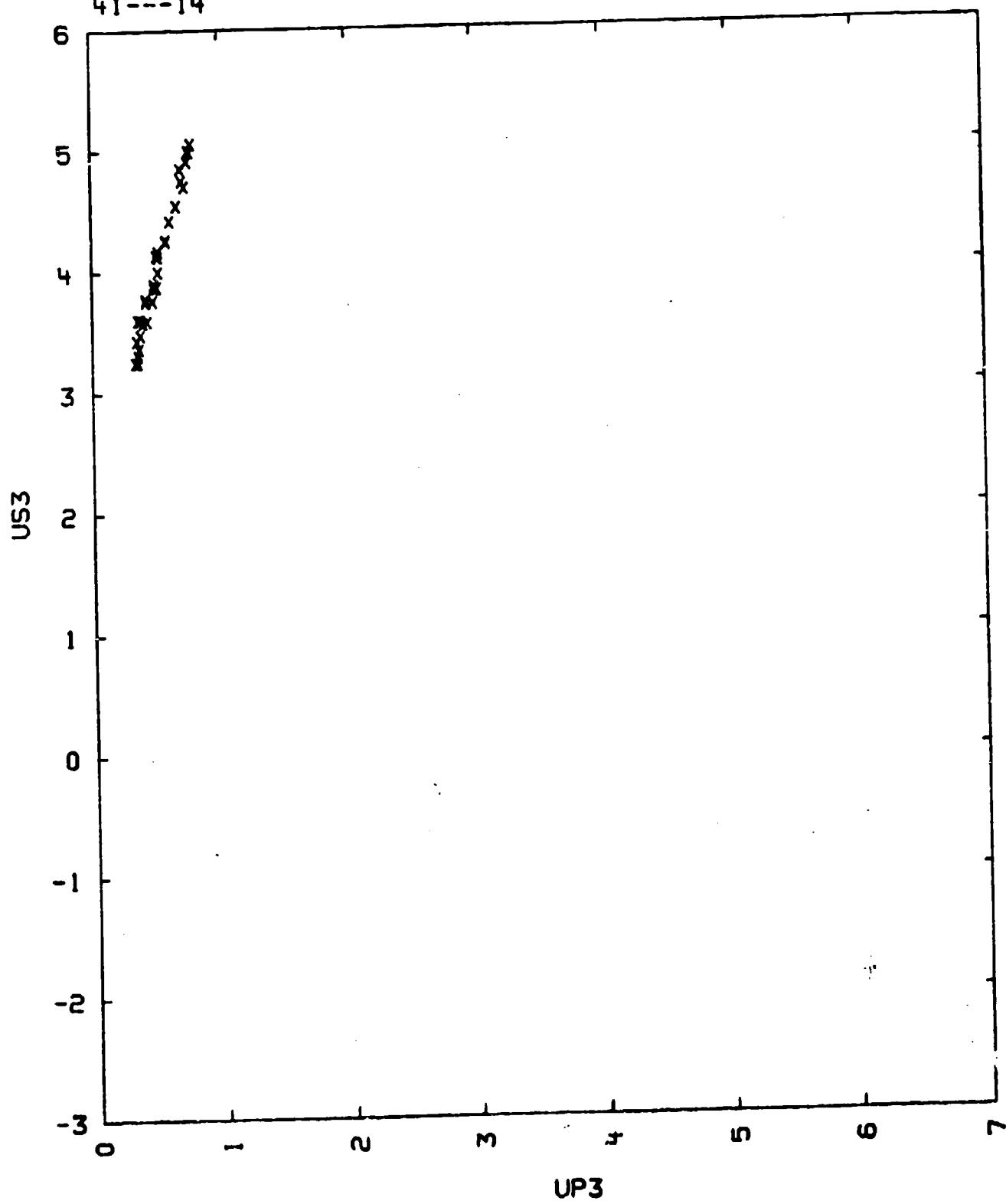


TABLE I

IRON (MILD STEEL)

41---14



41---15

IRON

FE 95.6 WT. PERCENT
 SI 4.0 WT. PERCENT
 W 0.3-0.4 WT. PERCENT

$$V_0 = 0.1308 \text{ CC/G.}$$

$$V_{01} = 0.1311 \text{ CC/G.}$$

IN THE TABLE BELOW, THE DENSITY IS GIVEN IN G/CC, VELOCITIES ARE IN KM/SEC AND PRESSURE IN KILOBARS.

TABLE

RHO0	US	UP	P	V/V0
7.646	5.684	1.074	467.	0.811
-	6.247	1.382	660.	0.779
-	6.474	1.469	727.	0.773
-	6.983	1.861	994.	0.734
-	6.996	1.902	1017.	0.728
-	7.177	1.997	1096.	0.722
-	8.056	2.525	1585.	0.687
-	8.541	2.922	1908.	0.658
-	9.742	3.568	2657.	0.634

$$US = 4.072 + 1.563 \cdot UP \text{ KM/SEC.} \quad \text{SIGMA US} = 0.073 \text{ KM/SEC.}$$

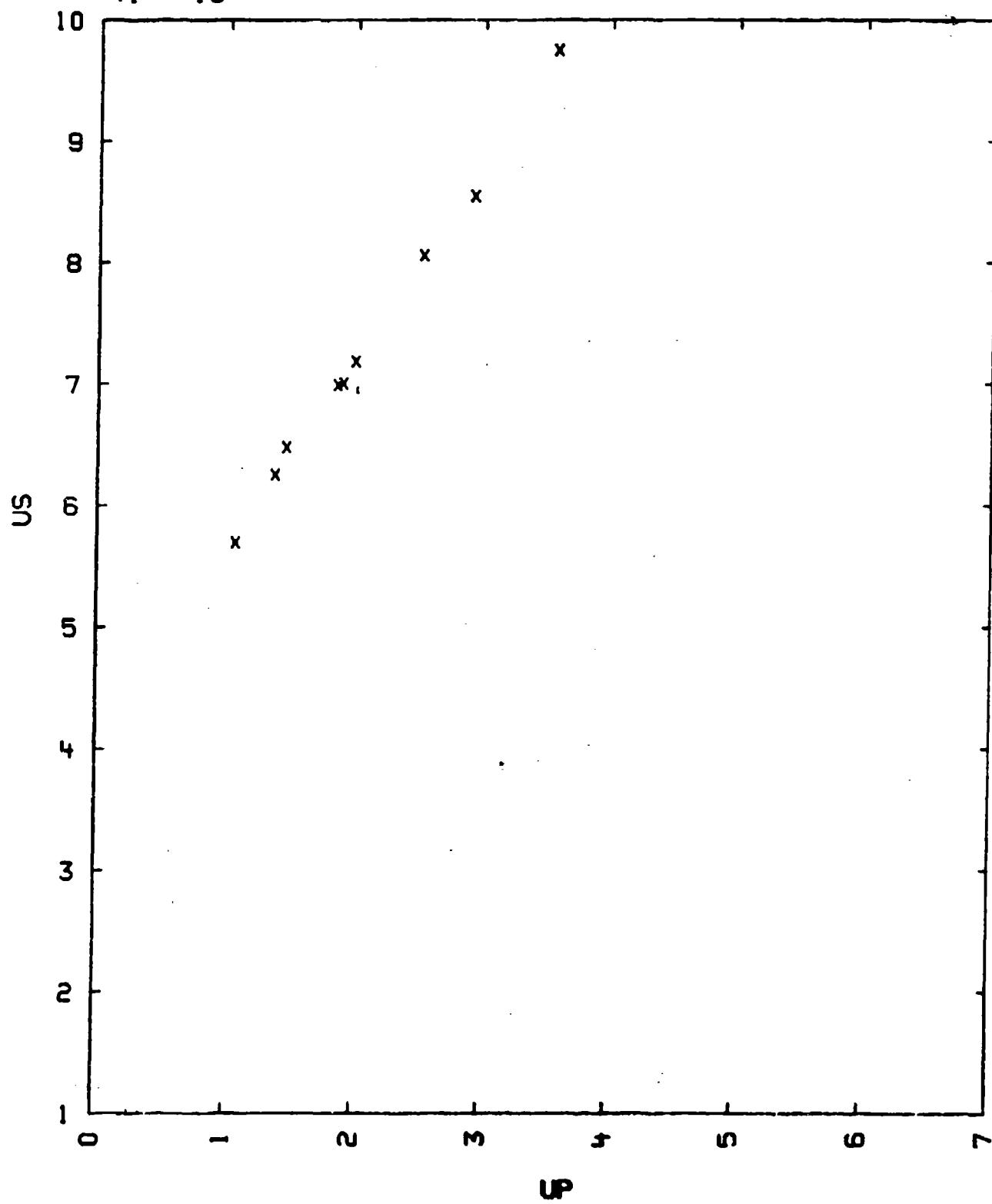
COMMENTS:

- 1) SOURCE: BALCHAN, A. S. AND COWAN, G. R.
JOURNAL OF GEOPHYSICAL RESEARCH, VOL. 71, P. 3577. (1966)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS NICKEL
- 3) THE STANDARD HUGONIOT WAS OBTAINED FROM THE FOLLOWING EQUATION:

$$US = 4.612 + 1.523 \cdot UP - 0.0148 \cdot UP^{0.2} \text{ KM/SEC}$$
 THIS EQUATION IS A COMPOSITE FIT OF DATA OBTAINED FROM WALSH, J. H.,
 ET AL., MATERIAL 39---1, MCQUEEN, R. G. AND MARSH, S. P., MATERIAL
 39---2 AND AL TSHULER, L. V., ET AL., MATERIAL 39---3.
- 4) THE INITIAL DENSITY (RHO0) IS A AVERAGE DENSITY.
- 5) A SHOCK TILT CORRECTION HAS BEEN APPLIED TO THE MEASURED SHOCK VELOCITIES
FOR BOTH THE DRIVER PLATE AND SAMPLE.
- 6) V01 WAS CALCULATED WITH A BODY CENTERED LATTICE PARAMETER OF 2.8820
+OR- 5 ANGSTROM: W. B. PEARSON, HANDBOOK OF LATTICE SPACINGS AND
STRUCTURES OF METALS, VOL. 4 (PERGAMON PRESS, NEW-YORK, 1958).
0.001 ANGSTROM WAS ADDED TO THE LATTICE PARAMETER OF FE-SI AS A
CORRECTION FOR THE W CONTENT.

TABLE I

IRON
41---15



41---16
IRON POROUS

	POROUS	SOLID	MOLE PERCENT
CU	0.001	0.12	-
SI	0.13	0.01	-
CA CR NI MN CO	0.03	0.06	-
C	0.08-0.10	-	-
O	0.7	0.055	-
FE	REST	REST	-

PARTICLE SIZE 5 - 20 MICRONS

$$V_0 = 0.1280 - 0.188 \text{ CC/G}$$

$$V_{01} = 0.1270 \text{ CC/G}$$

THE TABLES LIST DENSITY IN G/CC. VELOCITIES IN KM/SEC AND PRESSURE IN KBAR. SEE COMMENT 7

TABLE I

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0
7.81	5.74	0.026	11.5	0.996	5.18	0.309	126.2	0.941
7.81	5.74	0.028	12.7	0.995	4.90	0.152	60.0	0.970
7.81	5.5	0.02	9.1	0.996	4.56	0.096	35.6	0.980

US =

TABLE II

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0
5.58	3.30	0.012	2.2	0.9964	1.78	0.304	51.1	0.718
5.58	3.01	0.012	1.9	0.9962	1.37	0.336	26.5	0.758
5.58	3.49	0.02	3.	0.996	1.80	0.523	53.6	0.712
5.58	3.36	0.021	3.9	0.9938	0.84	0.22	13.	0.75
5.58	2.81	0.018	2.6	0.9932	0.67	0.14	7.	0.81
5.63	3.11	0.02	3.	0.995	0.77	0.20	11.	0.74
5.42	3.30	0.016	2.8	0.9953	1.21	0.342	24.	0.72
5.65	2.83	0.011	1.8	0.9960	0.70	0.17	8.	0.77
5.50	3.03	0.012	2.1	0.9939	0.83	0.25	13.	0.72
5.33	2.73	0.014	2.0	0.9950	1.60	0.534	46.2	0.669

$$US2 = 0.217 + 2.92 \cdot UP \text{ KM/SEC}$$

$$SIG US = 0.1 \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: SCHMIDT, D.N. AND LINDE, R.K.
AIR FORCE WEAPONS LAB. REP.: CONTRACT NO. AF 29(601)-7235
SRI PROJECT: PGU-5127
STANFORD RES. INST., MENLO PARK, CALIFORNIA, U. S. A.
- 2) EXPERIMENTAL TECHNIQUE: 11 AND 12

DATA REDUCTION TECHNIQUE: C. WAVE PROFILES WERE BROAD BUT WERE APPROXIMATED BY A STEP FUNCTION MOVING AT THE AVERAGE WAVE VELOCITY LISTED

- 3) THE FOOT OF THE ELASTIC WAVE IS BETWEEN 1 AND 10 PERCENT FASTER THAN THE AVERAGE VELOCITY LISTED. THE PROFILE AND AVERAGE VELOCITY OF THE PLASTIC (CRUSHING) WAVE WAS INFERRED FROM THE STRESS-TIME PROFILE AT THE GAUGE-SAMPLE INTERFACE AND THE REVERBERATION TIME OF AN ELASTIC RELEASE WAVE REFLECTING OFF THE CRUSHING WAVE.
- 4) THE POROUS SAMPLES THAT YIELDED THE LAST TWO ENTRIES WERE HEATED TO 1100 DEG. F IN A HYDROGEN ATMOSPHERE. THE SHOCKWAVES THAT PASSED THROUGH THESE SAMPLES ALSO PRODUCED HIGHER COMPRESSIONS, INDICATING THE EFFECT OF ORGANIC AND METAL OXIDE IMPURITIES.
- 5) ALL POROUS SAMPLES WERE COMPACTED AND SINTERED AT 1600 DEG. F.
- 6) VOI WAS OBTAINED FROM A CUBIC CELL CONSTANT OF 2.8665 ANGSTROM WYCKOFF, CRYSTAL STRUCTURES, VOL. 1 (INTERSCIENCE PUBLISHERS, NEW YORK, 1963) 2ND. ED.
- 7) NOTE THAT THE SECOND WAVE PARAMETERS REPRESENT A CRUSHING CURVE. P IS AN EFFECTIVE LONGITUDINAL STRESS OBTAINED BY ASSUMING THAT THE SAMPLE IS A HOMOGENEOUS MEDIUM.

TABLE I

IRON POROUS
41---16

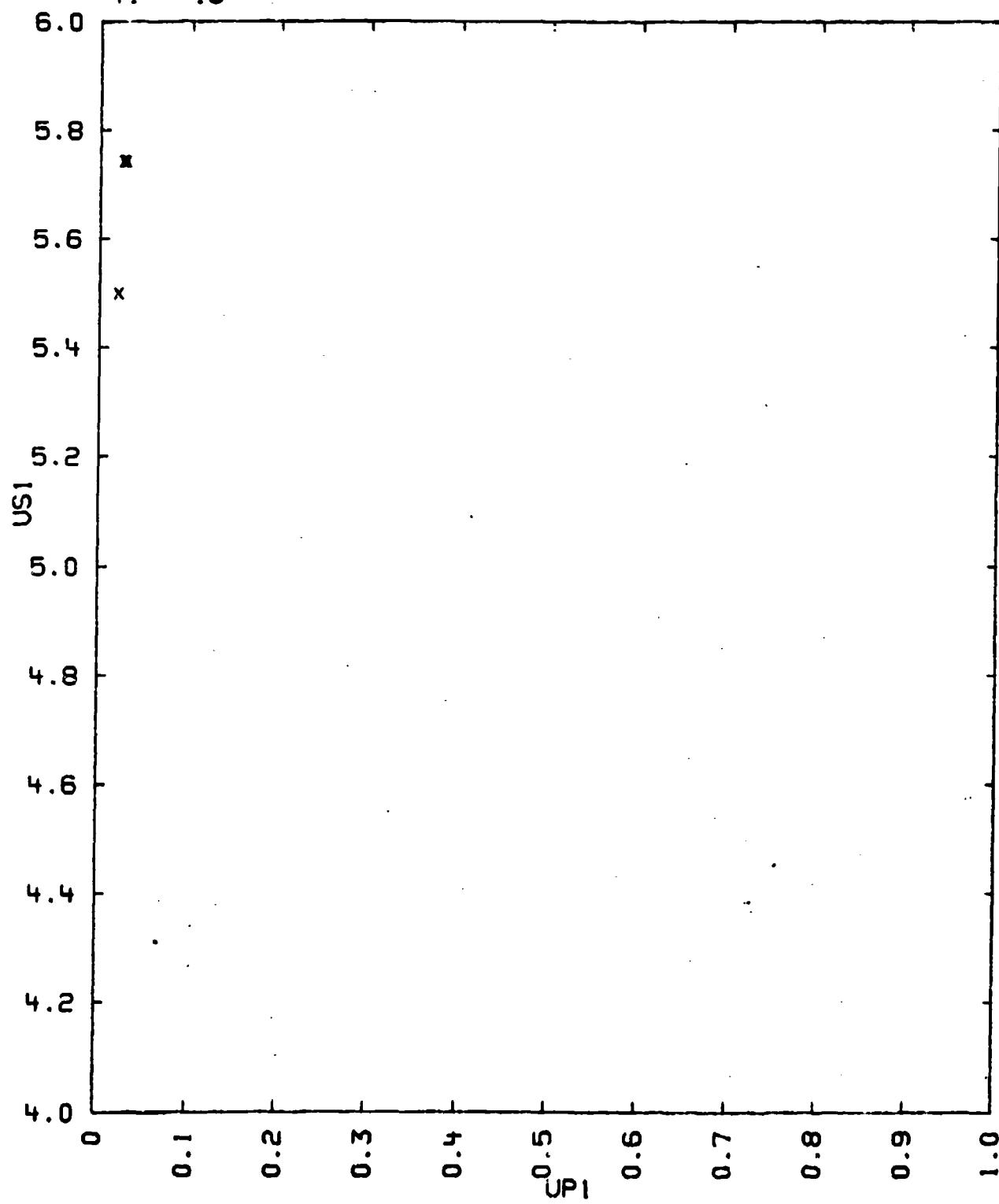


TABLE I

IRON POROUS

41---16

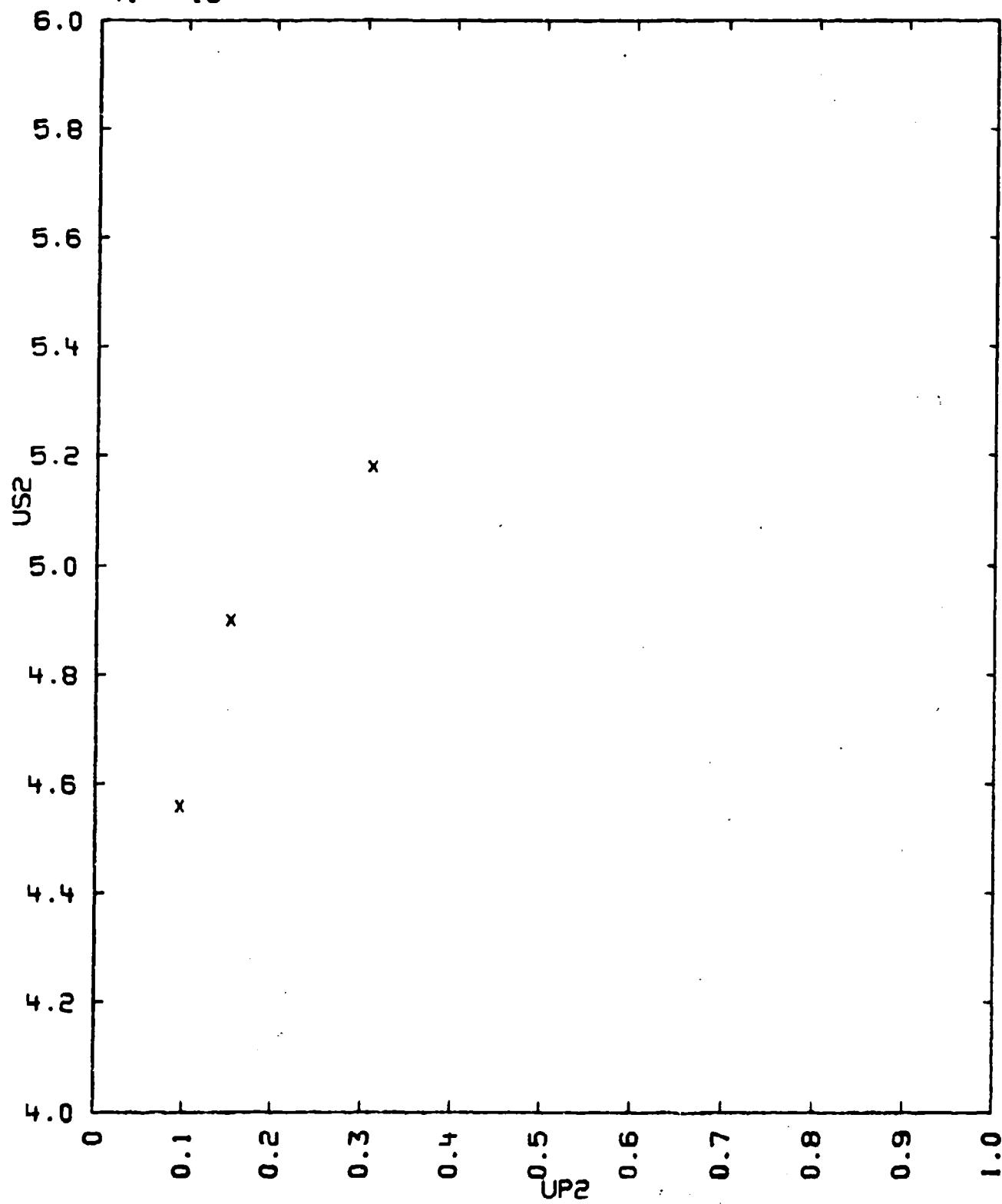


TABLE II

IRON POROUS
41---16

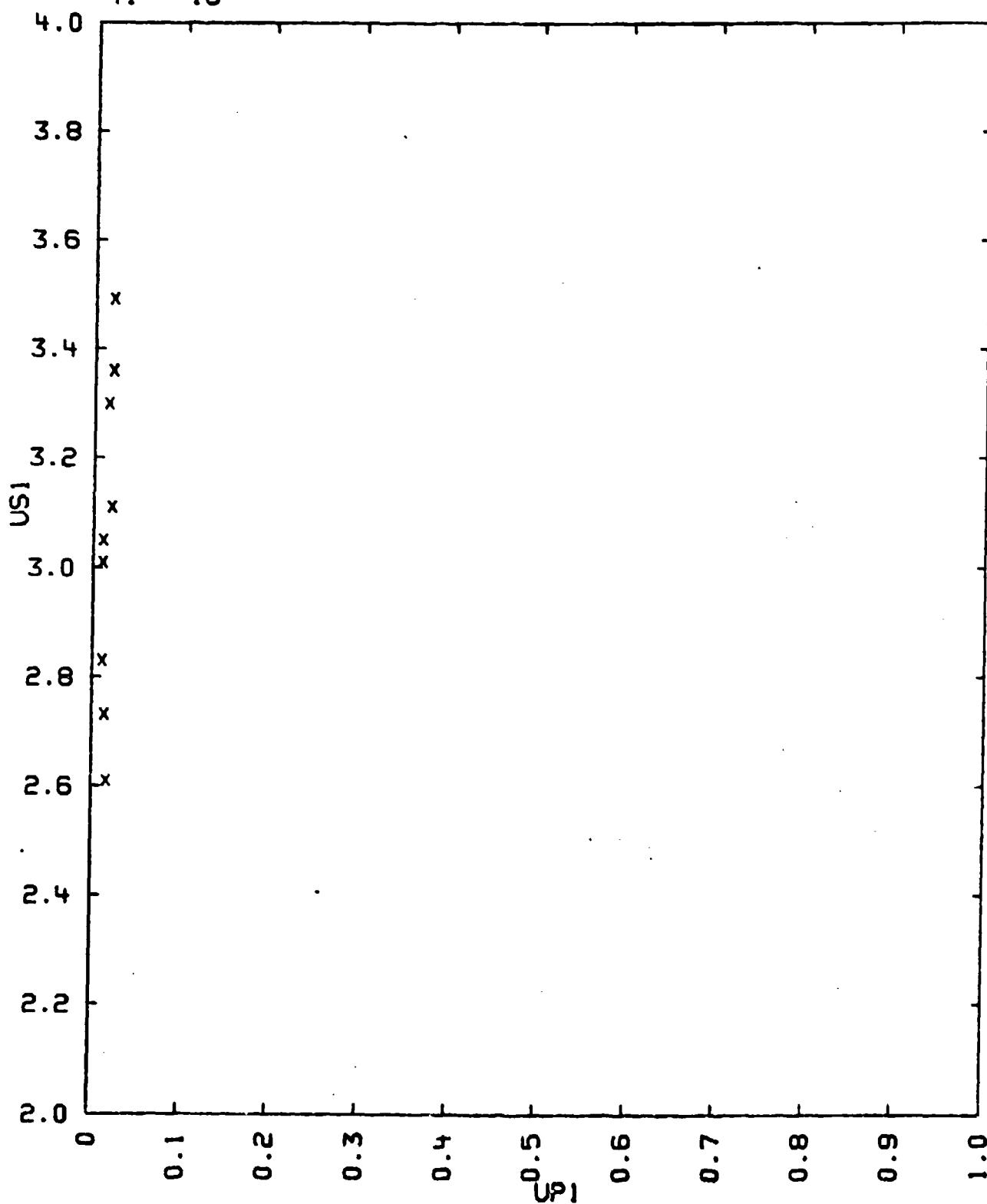
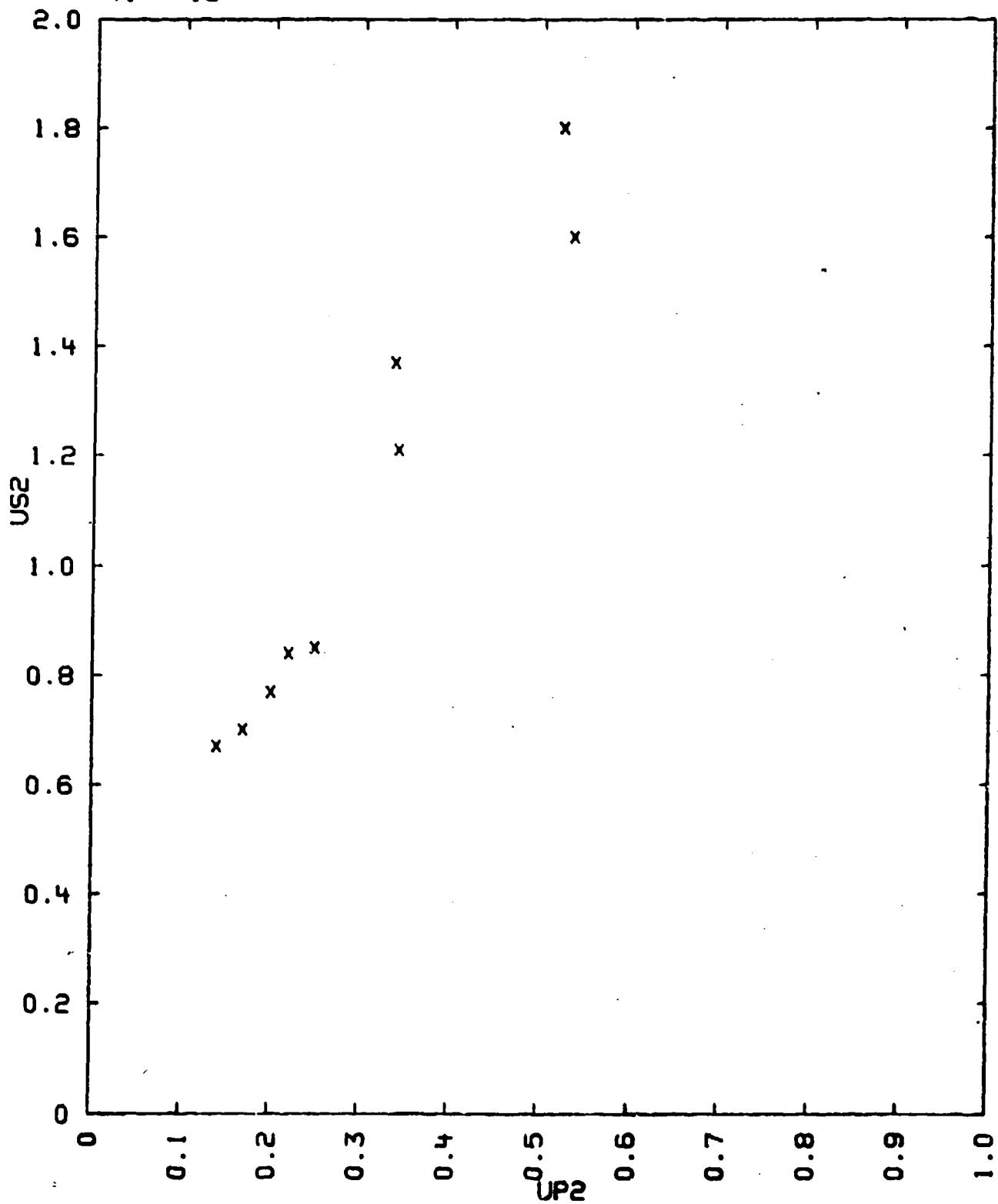


TABLE II

IRON POROUS
41---16

41---17

IRON

FE

$$V_0 = 0.1271 - 0.3192 \text{ CC/G} \quad C_L = 5.94 \text{ KM/SEC} \quad C_0 = 4.59 \text{ KM/SEC}$$

$$V_{0I} = 0.1269 \text{ CC/G} \quad C_S = 3.26 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE I

SAMPLE					STANDARD
RHO0	US	UP	P	V/V0	MATERIAL
7.850	5.33	0.95	397.	0.8218	IRON
7.843	5.46	1.00	428.	0.8168	IRON
7.850	5.56	1.06	463.	0.8094	IRON
7.847	6.20	1.43	696.	0.7694	IRON
7.850	6.24	1.44	705.	0.7692	IRON
7.850	6.58	1.65	852.	0.7492	IRON
7.850	6.66	1.73	904.	0.7402	IRON
7.847	7.07	1.96	1087.	0.7228	IRON
7.850	7.08	1.98	1100.	0.7203	IRON
7.831	7.56	2.27	1344.	0.6997	IRON
7.847	7.81	2.44	1495.	0.6876	IRON
7.832	7.90	2.46	1522.	0.6886	IRON
7.809	8.09	2.61	1649.	0.6774	IRON
7.850	8.01	2.62	1647.	0.6729	IRON
7.847	8.24	2.72	1759.	0.6699	IRON
7.850	8.23	2.72	1757.	0.6695	IRON
7.850	8.32	2.73	1783.	0.6719	IRON
7.850	8.26	2.73	1770.	0.6695	IRON
7.868	8.31	2.82	1844.	0.6606	IRON
7.840	8.49	2.89	1924.	0.6596	IRON
7.842	8.81	3.06	2114.	0.6527	IRON
7.850	8.94	3.12	2190.	0.6510	IRON
7.850	8.99	3.18	2244.	0.6483	IRON
7.850	9.10	3.19	2279.	0.6495	IRON
7.860	9.24	3.35	2433.	0.6374	IRON
7.860	9.28	3.36	2451.	0.6379	IRON
7.855	9.51	3.59	2682.	0.6225	IRON

US = $3.574 + 1.920 \cdot UP - 0.068 \cdot UP^2$ KM/SEC HUGONIOT FIT MADE BY THE SOURCE

COMPARISON LEAST SQUARE FITS:

US = A0 + A1 * UP + A2 * UP^2 KM/SEC

A0 = 3.718 KM/SEC SIG A0 = 0.082 KM/SEC

A1 = 1.781 SIG A1 = 0.079

A2 = -0.040 SEC/KM SIG A2 = 0.018 SEC/KM

SIG US = 0.049 KM/SEC

TABLE II

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
7.861	4.84	0.57	217.	0.8822	2024 AL	6.48
7.840	5.39	0.97	410.	0.8200	2024 AL	7.23
7.840	5.37	0.97	408.	0.8194	2024 AL	7.23
7.840	5.41	0.99	420.	0.8170	2024 AL	7.27
7.840	5.58	1.06	464.	0.8100	2024 AL	7.41
7.865	5.52	1.06	460.	0.8080	2024 AL	7.41
7.857	5.48	1.07	461.	0.8047	2024 AL	7.41
7.861	5.62	1.10	486.	0.8043	2024 AL	7.49
7.862	5.97	1.30	610.	0.7822	2024 AL	7.88
7.861	6.24	1.46	716.	0.7660	2024 AL	8.19
7.861	6.93	1.85	1008.	0.7330	2024 AL	8.95
7.862	7.15	1.99	1119.	0.7217	2024 AL	9.1
7.861	7.47	2.07	1216.	0.7229	2024 AL	9.1
7.861	7.45	2.15	1259.	0.7114	2024 AL	9.54
7.861	7.80	2.34	1435.	0.7000	2024 AL	9.1
7.861	7.75	2.35	1432.	0.6968	2024 AL	9.1
7.861	8.17	2.58	1657.	0.6842	2024 AL	10.1
7.861	8.11	2.59	1651.	0.6806	2024 AL	10.36

$$US = A_0 + A_1 \cdot UP + A_2 \cdot UP^2 \text{ KM/SEC}$$

$$A_0 = 3.764 \text{ KM/SEC} \quad \text{SIG } A_0 = 0.132 \text{ KM/SEC}$$

$$A_1 = 1.682 \quad \text{SIG } A_1 = 0.18$$

$$A_2 = 0.0115 \text{ SEC/KM} \quad \text{SIG } A_2 = 0.054 \text{ SEC/KM}$$

$$\text{SIG US} = 0.067 \text{ KM/SEC}$$

TABLE III

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
7.861	5.67	1.18	526.	0.7919	921-T AL	7.48
7.861	5.66	1.18	525.	0.7915	921-T AL	7.48
7.861	6.35	1.54	769.	0.7575	921-T AL	8.21
7.861	6.33	1.54	766.	0.7567	921-T AL	8.21
7.861	6.86	1.86	1003.	0.7289	921-T AL	8.84
7.861	6.84	1.86	1000.	0.7281	921-T AL	8.84
7.861	7.58	2.26	1347.	0.7018	921-T AL	9.65
7.861	7.55	2.26	1341.	0.7007	921-T AL	9.65
7.861	8.17	2.56	1644.	0.6867	921-T AL	10.26
7.861	8.09	2.57	1634.	0.6823	921-T AL	10.26

$$US = A_0 + A_1 \cdot UP + A_2 \cdot UP^2 \text{ KM/SEC}$$

$$A_0 = 3.651 \text{ KM/SEC} \quad \text{SIG } A_0 = 0.20 \text{ KM/SEC}$$

$$A_1 = 1.697 \quad \text{SIG } A_1 = 0.22$$

$$A_2 = 0.018 \quad \text{SIG } A_2 = 0.058 \text{ SEC/KM}$$

$$\text{SIG US} = 0.037 \text{ KM/SEC}$$

TABLE IV

REPRODUCED FROM
BEST AVAILABLE COPY

-----SAMPLE-----

-----STANDARD-----

RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
7.864	5.25	0.99	409.	0.8114	CU	5.33
7.856	5.41	1.01	429.	0.8133	CU	5.37
7.859	5.49	1.06	457.	0.8069	CU	5.44
7.861	5.48	1.07	461.	0.8047	CU	5.45
7.864	5.52	1.10	478.	0.8007	CU	5.49
7.864	5.49	1.10	475.	0.7996	CU	5.49
7.861	5.95	1.30	608.	0.7815	CU	5.80
7.856	6.20	1.41	687.	0.7726	CU	5.97
7.861	6.58	1.70	879.	0.7416	CU	6.37
7.856	6.74	1.77	937.	0.7374	CU	6.48
7.861	6.74	1.78	943.	0.7359	CU	6.49
7.857	7.29	2.07	1186.	0.7160	CU	6.93
7.856	7.26	2.08	1186.	0.7135	CU	6.94
7.861	7.49	2.23	1313.	0.7023	CU	7.15
7.867	7.56	2.25	1338.	0.7024	CU	7.19
7.867	7.46	2.26	1326.	0.6971	CU	7.19
7.867	7.62	2.29	1367.	0.7008	CU	7.24
7.867	7.51	2.30	1359.	0.6937	CU	7.24
7.859	8.03	2.55	1609.	0.6824	CU	7.62
7.861	8.05	2.59	1639.	0.6783	CU	7.67
7.852	8.55	2.83	1900.	0.6690	CU	8.05

$$US = A_0 + A_1 \cdot UP + A_2 \cdot UP^2 \text{ KM/SEC}$$

$$A_0 = 3.454 \text{ KM/SEC} \quad SIG A_0 = 0.138 \text{ KM/SEC}$$

$$A_1 = 1.971 \quad SIG A_1 = 0.167$$

$$A_2 = -0.070 \text{ SEC/KM} \quad SIG A_2 = 0.046 \text{ SEC/KM}$$

$$SIG US = 0.058 \text{ KM/SEC}$$

TABLE V

-----SAMPLE-----

-----STANDARD-----

RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
7.882	5.22	0.90	370.	0.8276	TU-3 MO	3.70
7.882	5.17	0.91	371.	0.8240	TU-3 MO	3.70
7.882	5.34	0.98	412.	0.8165	TU-3 MO	3.81
7.861	6.18	1.39	675.	0.7751	TU-3 MO	4.34
7.861	6.17	1.39	674.	0.7747	TU-3 MO	4.34
7.861	6.23	1.42	698.	0.7729	TU-3 MO	4.38
7.861	6.40	1.54	775.	0.7594	TU-3 MO	4.53
7.861	6.39	1.54	774.	0.7590	TU-3 MO	4.53
7.861	7.74	2.37	1442.	0.6938	TU-3 MO	5.59
7.861	7.74	2.37	1442.	0.6938	TU-3 MO	5.59
7.861	7.74	2.37	1442.	0.6938	TU-3 MO	5.59
7.861	8.08	2.49	1582.	0.6918	TU-3 MO	5.76
7.861	8.66	2.97	2022.	0.6570	TU-3 MO	6.35

$$US = A_0 + A_1 \cdot UP + A_2 \cdot UP^2$$

$$A_0 = 3.263 \text{ KM/SEC} \quad SIG A_0 = 0.131 \text{ KM/SEC}$$

$$A_1 = 2.298 \quad SIG A_1 = 0.156$$

$$A_2 = -0.163 \text{ SEC/KM} \quad SIG A_2 = 0.042 \text{ SEC/KM}$$

$$SIG US = 0.054 \text{ KM/SEC}$$

TABLE VI

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
7.002	3.21	0.57	128.	0.8224	CU	4.60
6.992	3.26	0.70	160.	0.7853	CU	4.75
7.001	3.37	0.70	185.	0.7823	CU	4.76
7.061	3.47	0.79	194.	0.7723	CU	4.87
6.965	3.57	0.85	211.	0.7619	CU	4.85
6.960	4.42	1.15	354.	0.7398	CU	5.38
6.945	5.29	1.43	525.	0.7297	CU	5.79
6.940	5.14	1.45	517.	0.7178	CU	5.80
6.988	5.42	1.57	595.	0.7103	CU	5.98
6.954	5.51	1.65	632.	0.7005	CU	6.08
7.025	6.14	1.77	763.	0.7117	CU	6.31
6.968	5.90	1.84	756.	0.6881	CU	6.35
6.986	5.92	1.86	769.	0.6858	CU	6.39
7.070	6.24	1.95	860.	0.6875	CU	6.55
6.913	6.29	2.01	874.	0.6804	CU	5.60
6.963	6.30	2.05	899.	0.6746	CU	6.65
6.979	6.43	2.09	938.	0.6750	CU	6.72
6.933	6.94	2.44	1174.	0.6484	CU	7.19
6.958	7.60	2.80	1481.	0.6316	CU	7.71
7.046	7.93	3.01	1682.	0.6204	CU	8.02
6.945	8.07	3.05	1709.	0.6221	CU	8.08
6.910	7.99	3.13	1728.	0.6083	CU	8.15

US =

TABLE VII

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
5.973	2.24	0.54	72.	0.7589	CU	4.48
5.993	2.44	0.85	95.	0.7336	CU	4.60
5.884	2.74	0.79	127.	0.7117	CU	4.76
5.913	3.03	0.94	168.	0.6898	CU	4.93
6.054	3.62	1.12	245.	0.6906	CU	5.20
6.041	4.02	1.25	304.	0.6891	CU	5.38
6.000	4.68	1.56	438.	0.6667	CU	5.79
6.050	4.97	1.70	511.	0.6579	CU	5.98
6.104	5.50	2.00	871.	0.6364	CU	6.39
5.920	5.73	2.19	743.	0.6178	CU	6.60
5.919	5.91	2.28	798.	0.6142	CU	6.72
5.820	6.62	2.83	1013.	0.6027	CU	7.19
6.080	7.22	2.98	1308.	0.5873	CU	7.71
6.055	7.59	3.33	1520.	0.5607	CU	8.15

US =

TABLE VIII

SAMPLE					STANDARD	
RHO	US	UP	P	V/V0	MATERIAL	US(ST)
4.808	1.77	0.59	50.	0.6667	CU	4.48
4.629	1.93	0.72	64.	0.6269	CU	4.60
4.697	2.35	0.86	85.	0.6340	CU	4.76
4.761	2.48	0.96	113.	0.6129	CU	4.87
4.532	2.44	1.06	117.	0.5656	CU	4.95
4.662	2.76	1.13	145.	0.5903	CU	5.05
4.630	4.23	1.74	342.	0.5857	CU	5.79
4.744	2.93	1.26	175.	0.5700	CU	5.20
4.651	3.20	1.43	213.	0.5531	CU	5.38
4.720	3.99	1.77	333.	0.556	CU	5.80
4.680	4.29	1.91	383.	0.5548	CU	5.98
4.827	4.57	1.96	432.	0.5711	CU	6.08
4.729	5.05	2.19	523.	0.5663	CU	6.35
4.908	5.08	2.19	546.	0.5689	CU	6.39
4.901	5.27	2.38	615.	0.5484	CU	6.60
4.734	5.33	2.44	616.	0.5422	CU	6.65
4.619	5.43	2.51	630.	0.5378	CU	6.72
4.880	6.07	2.85	844.	0.5305	CU	7.19
4.770	7.16	3.53	1206.	0.5070	CU	8.02
4.958	7.35	3.58	1305.	0.5129	CU	8.15

US =

TABLE IX

SAMPLE					STANDARD	
RHO	US	UP	P	V/V0	MATERIAL	US(ST)
3.313	1.41	0.64	30.	0.5461	CU	4.48
3.222	1.50	1.94	45.	0.3733	CU	4.73
3.472	2.14	1.05	78.	0.5093	CU	4.87
3.651	1.97	1.07	77.	0.4569	CU	4.88
3.388	1.97	1.18	79.	0.4010	CU	4.98
3.390	2.30	1.24	96.	0.4609	CU	5.05
3.323	2.83	1.65	155.	0.4170	CU	5.45
6.538	3.46	1.95	441.	0.4364	CU	5.80
3.219	3.59	1.96	227.	0.4540	CU	5.78
3.351	3.49	2.01	235.	0.4241	CU	5.85
3.716	4.05	2.16	325.	0.4667	CU	6.08
3.484	4.65	2.37	385.	0.4903	CU	6.31
3.310	4.45	2.40	365.	0.4427	CU	6.37
3.133	4.24	2.51	333.	0.4080	CU	6.35
3.260	4.73	2.70	416.	0.4292	CU	6.60
3.317	4.87	2.73	441.	0.4394	CU	6.65
3.260	5.62	3.16	579.	0.4377	CU	7.13
3.323	5.63	3.16	591.	0.4387	CU	7.15
3.486	6.34	3.47	767.	0.4527	CU	7.57
3.424	6.29	3.65	796.	0.4197	CU	7.71
3.443	6.82	3.95	929.	0.4208	CU	8.07

IRON

RHO0	US	UP	P	V/V0	MATERIAL US(ST)
US -					

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.H., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES.
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: A (TABLE I), B (THE REST)
- 3) V01 WAS CALCULATED FROM WYCKOFF, CRYSTAL STRUCTURES (JOHN WILEY AND SONS, N.Y., 1963) V. I
- 4) TABLE I IS THE HUGONIOT OBTAINED FROM SYMMETRIC IMPACT. TABLES II THROUGH V CHECK THE INTERNAL CONSISTENCY OF THE OTHER STANDARD HUGONIOTS OBTAINED WITH SYMMETRIC IMPACTS.
- 5) THE GRUNEISEN GAMMA OF 1.69 (=GAM) WITH THE ASSUMPTION THAT RHO0.GAM = CONSTANT REPRESENTS THE BEHAVIOR OF THE POROUS SAMPLES. ALSO GIVEN ARE (DV/DT)/V = 3.51E+5 PER DEG. AND CP = 0.448 JOULES PER G. DEG.
- 6) HUGONIOT ELASTIC LIMIT 15. - 14. KBAR (FINE - COARSE GRAIN; HARD)
HUGONIOT ELASTIC LIMIT 11. - 9. KBAR (- - - - ; SOFT)
- 7) THE 921-T AL PLATE STOCK USED TO OBTAIN THE CROSS CHECK DATA IN TABLE III IS SLIGHTLY POROUS

TABLE I

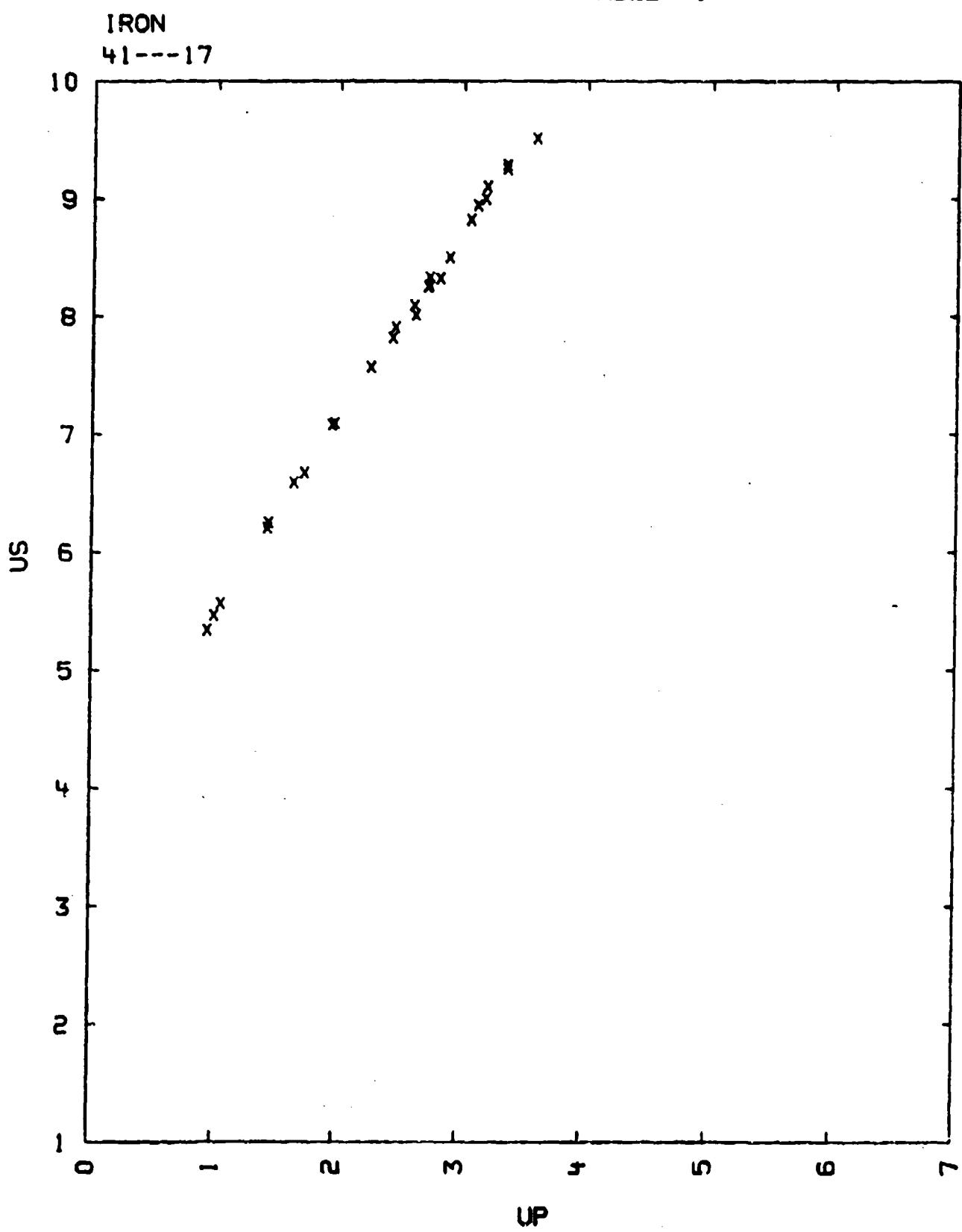


TABLE II

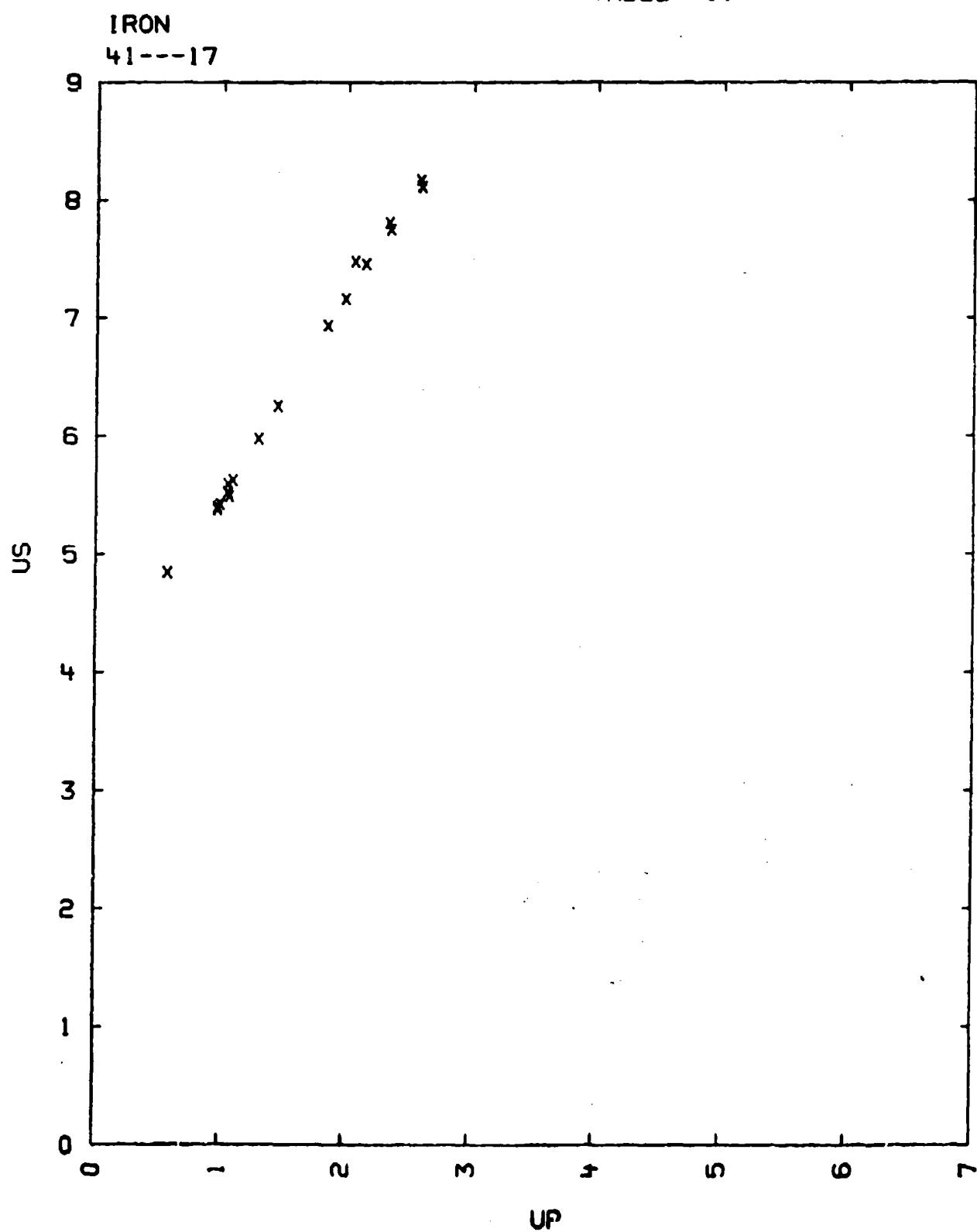


TABLE III

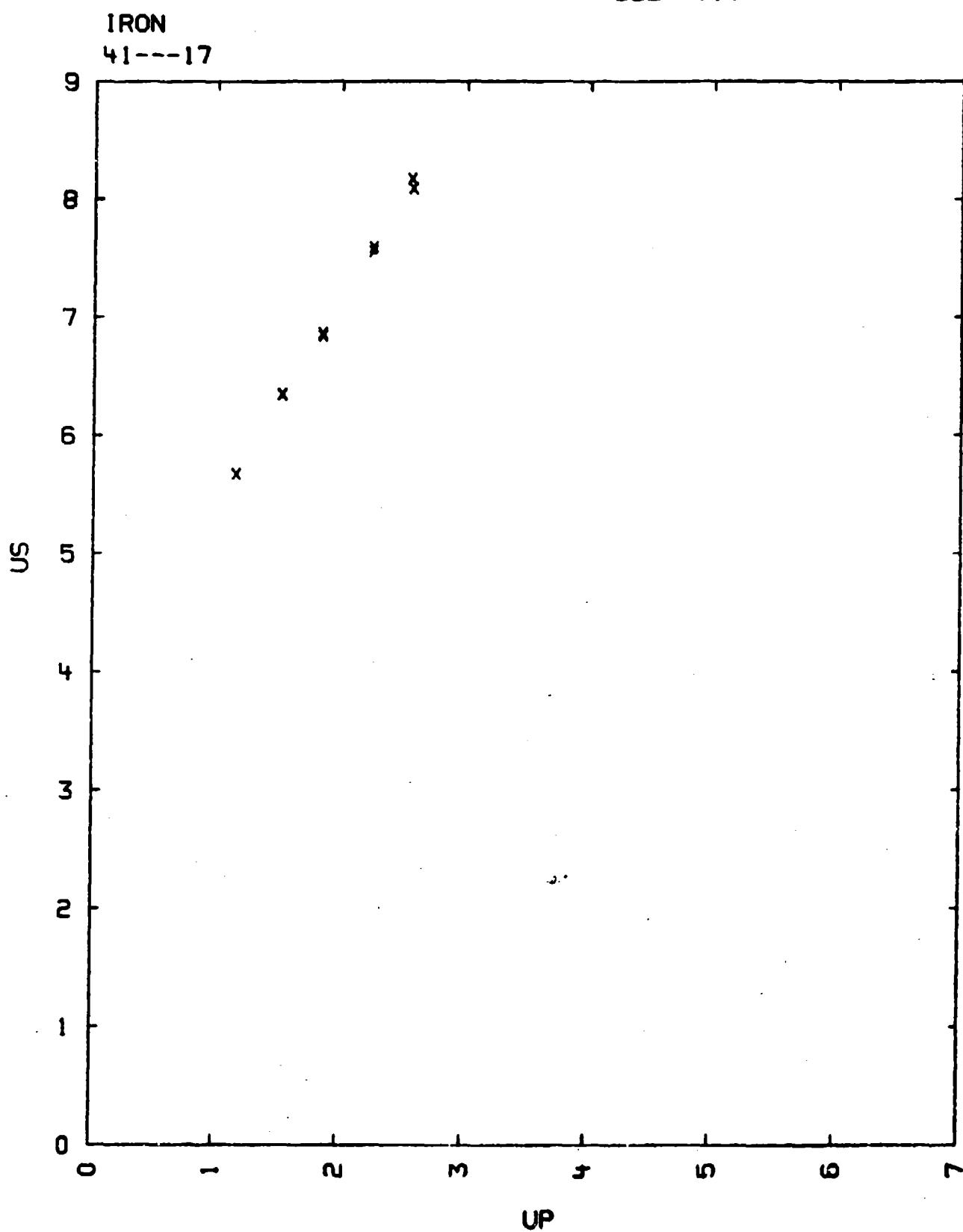


TABLE IV

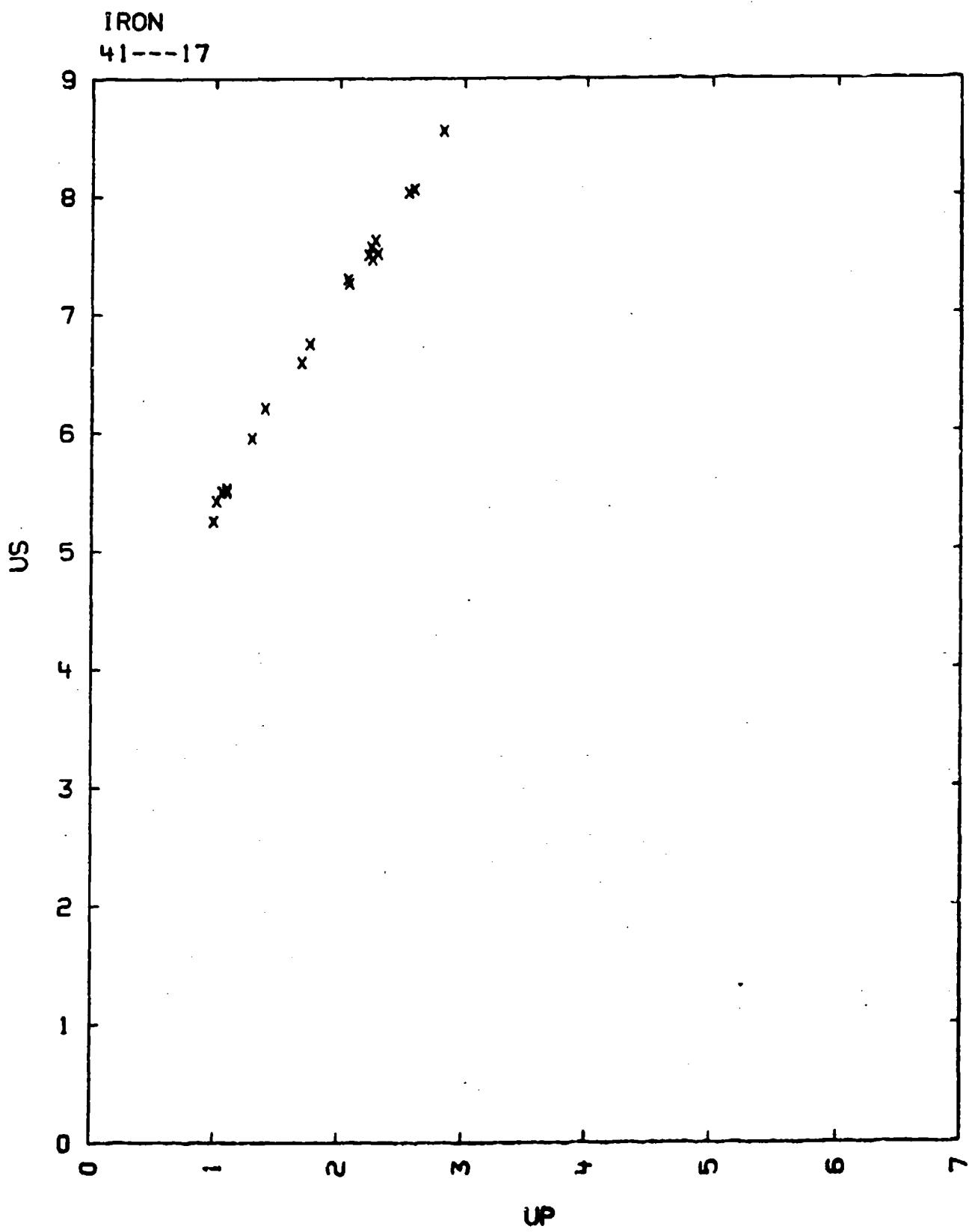


TABLE V

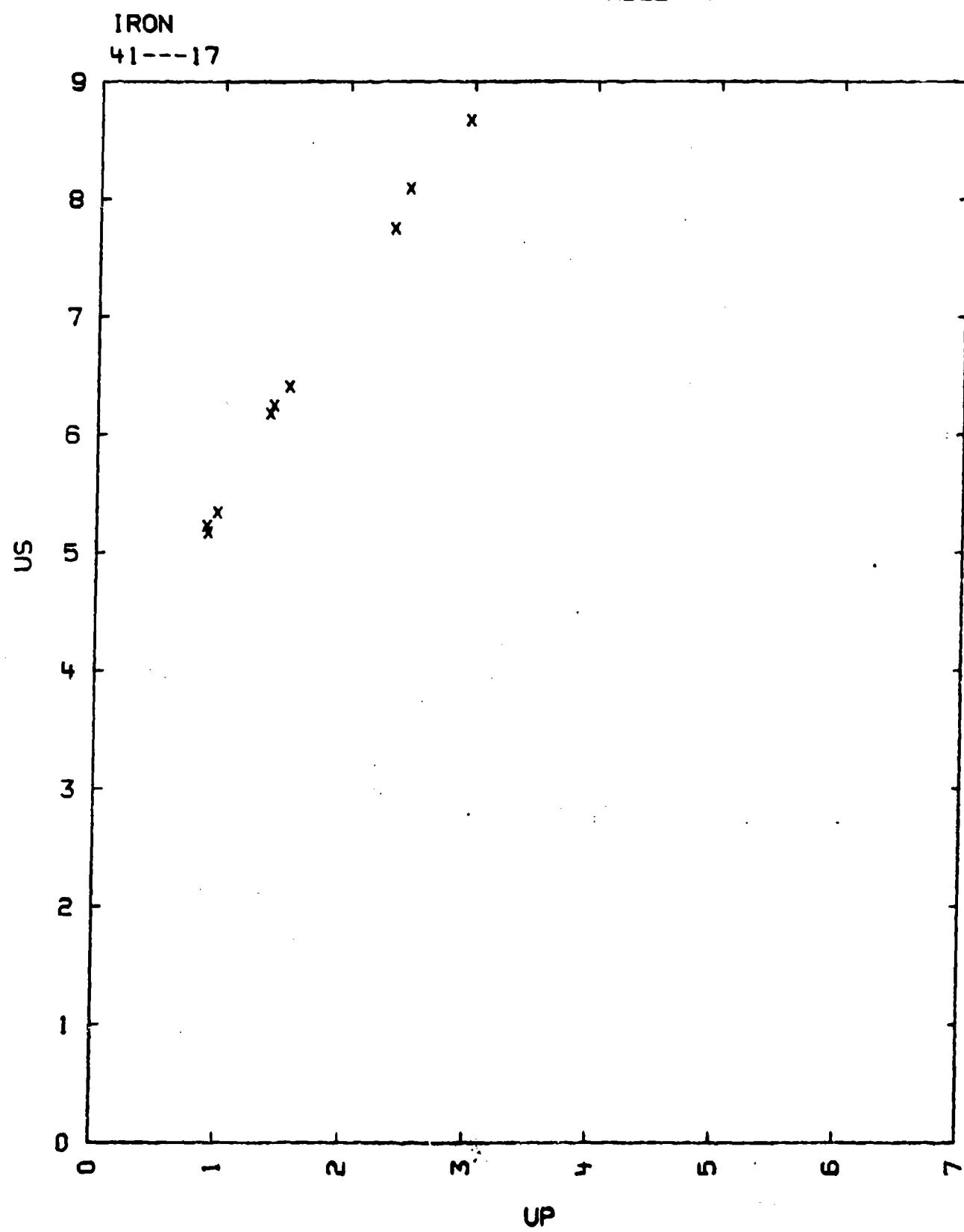


TABLE VI

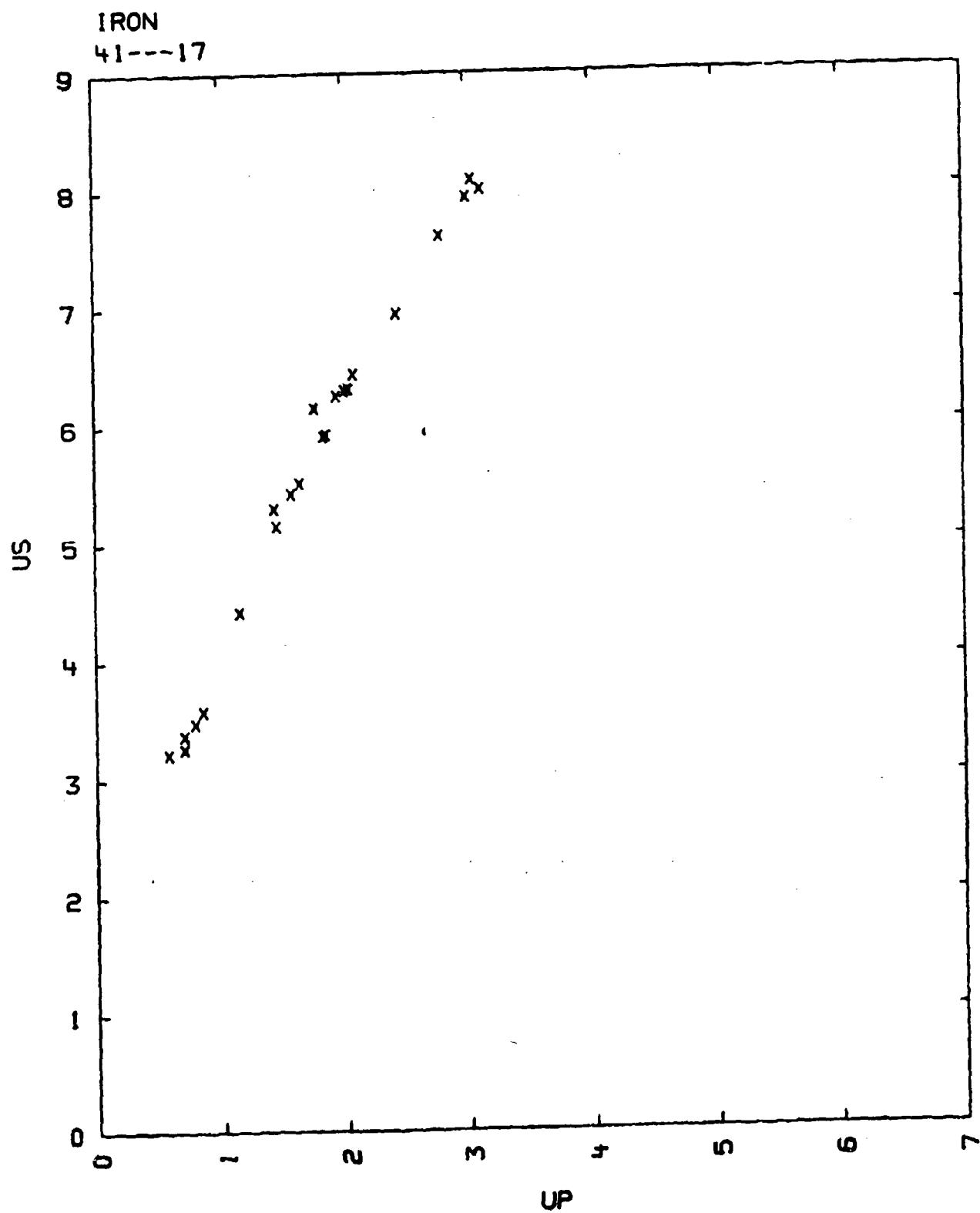


TABLE VII

IRON
41---17

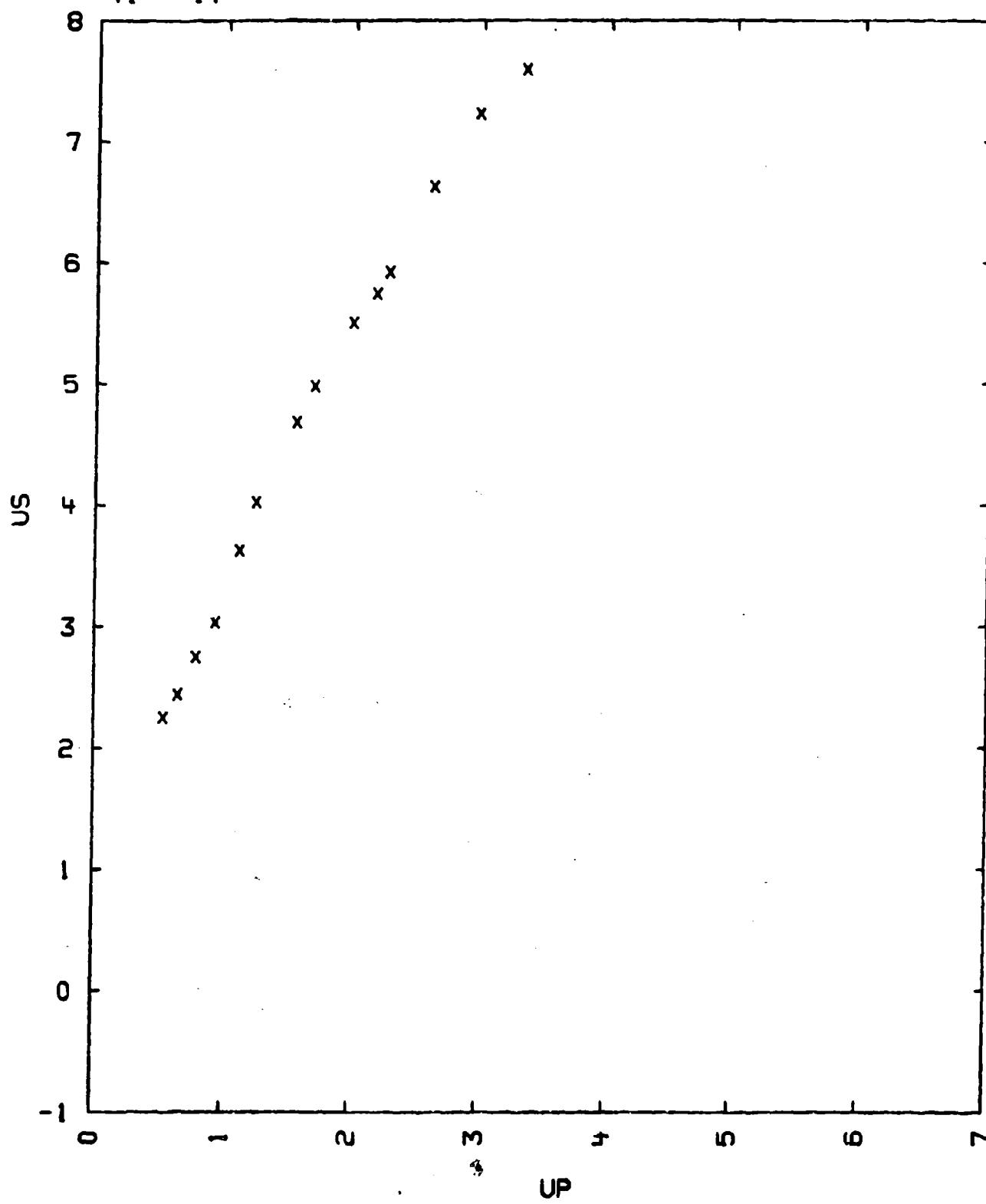


TABLE VIII

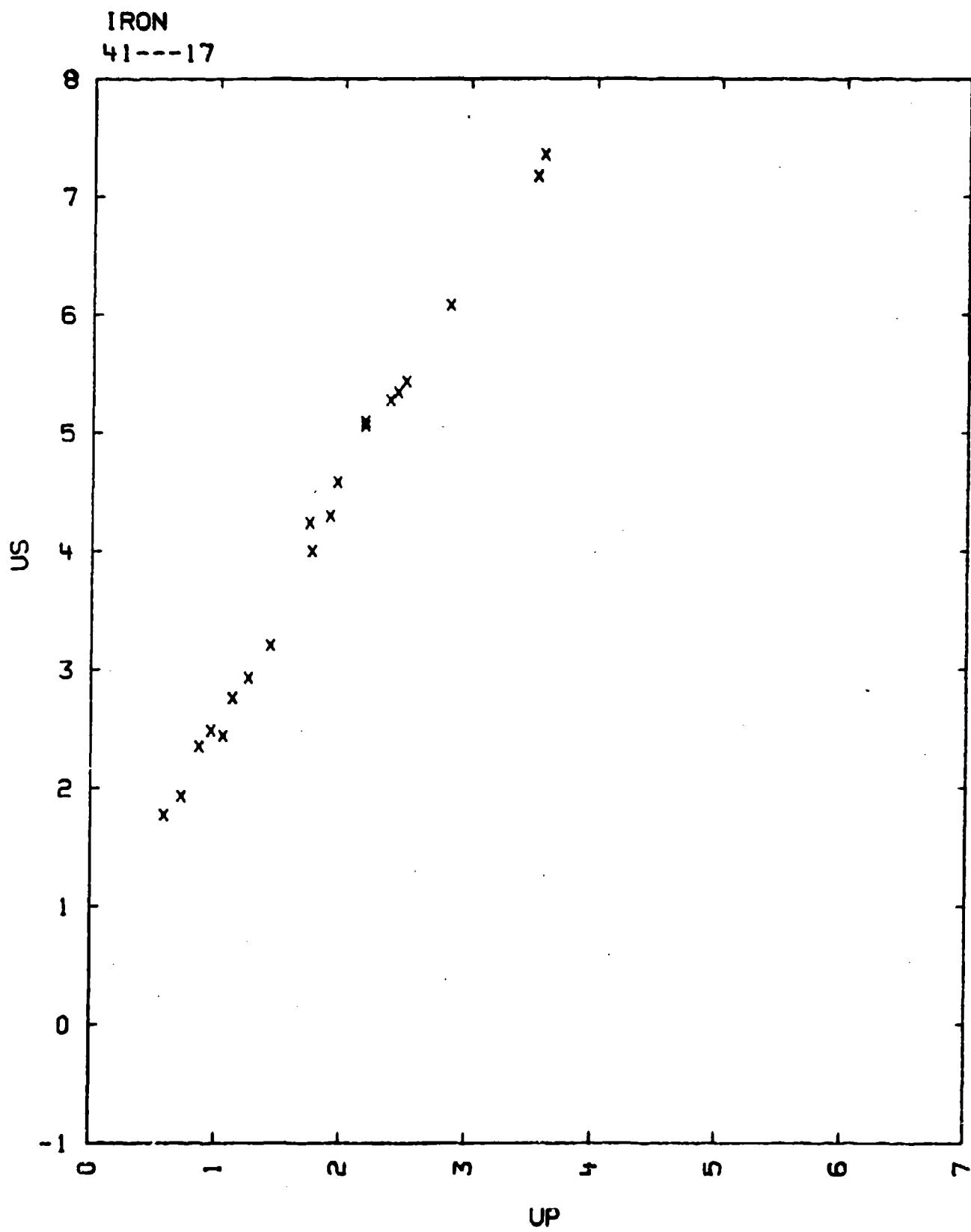
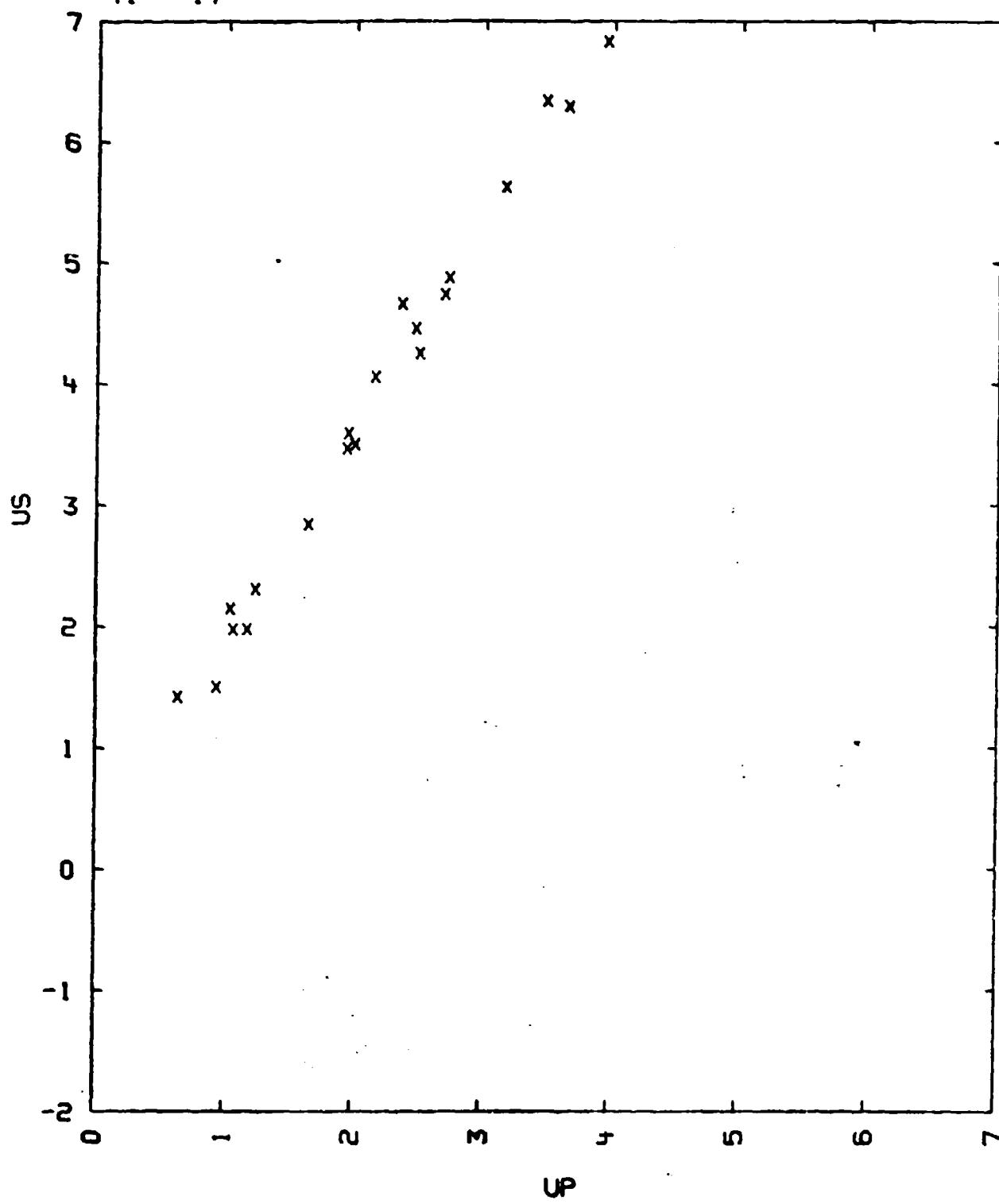


TABLE IX

IRON
41---17



42---1
PALLADIUM

P0 99.8 PER CENT OR GREATER

$$V_0 = 0.08368 \text{ CC/G.}$$

$$V_{01} = 0.08163 \text{ CC/G.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

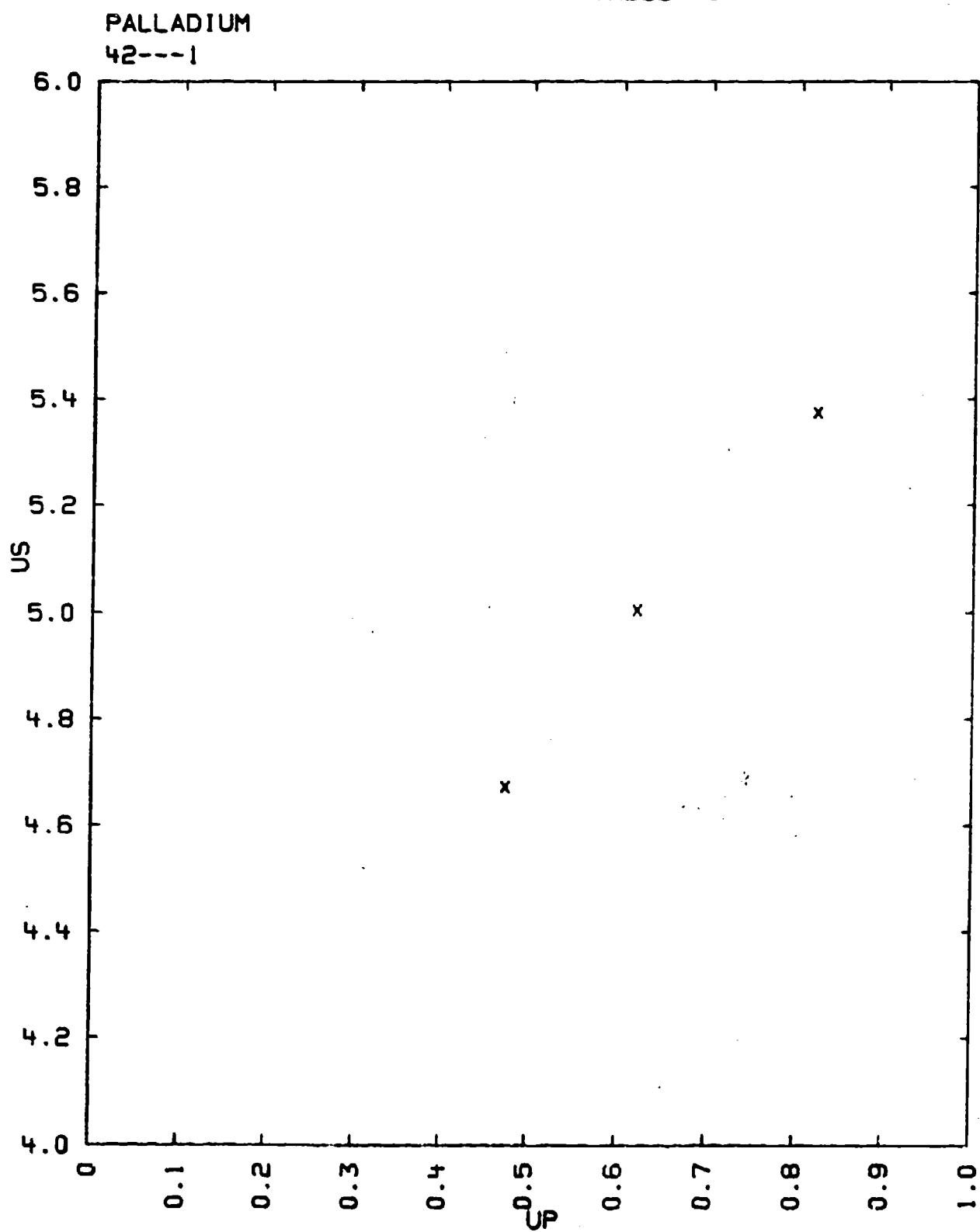
RHO0	US	UP	P	V/V0
11.95	4.673	0.4728	262.5	0.8988
-	5.004	0.6200	372	0.8761
-	5.374	0.8219	531	0.8471

$$US = 3.742 + 1.998 UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.6 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J.M., RICE, M.H., MCQUEEN, R.G. AND YARGER, F.L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 3 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS

TABLE I



42---2
PALLADIUM

P0 99.5 PERCENT OR GREATER

$V_0 = 0.0833 \text{ CC/G.}$ $C_L = 4.57 \text{ KM/SEC.}$ $C_D = 3.90 \text{ KM/SEC.}$
 $V_{01} = 0.08307 \text{ CC/G.}$ $C_S = 2.06 \text{ KM/SEC.}$ $C_B = 3.80 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITIES IN G/CC.

TABLE

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	RHO0	US
12.00	4.737	0.467	265.5	0.9014	8.932	4.764
-	4.992	0.619	370.8	0.8760	8.934	5.024
-	5.396	0.856	554.3	0.8414	-	5.426
12.01	5.522	0.927	614.8	0.8321	-	5.548
12.00	6.271	1.435	1079.9	0.7712	-	6.387
-	6.305	1.435	1085.7	0.7724	-	6.393
-	6.304	1.440	1089.3	0.7715	-	6.400
-	6.981	1.885	1579.1	0.7300	-	7.133
-	7.294	2.108	1845.1	0.7110	-	7.494
-	7.398	2.196	1949.5	0.7032	-	7.632
-	7.691	2.361	2179.0	0.6930	-	7.910

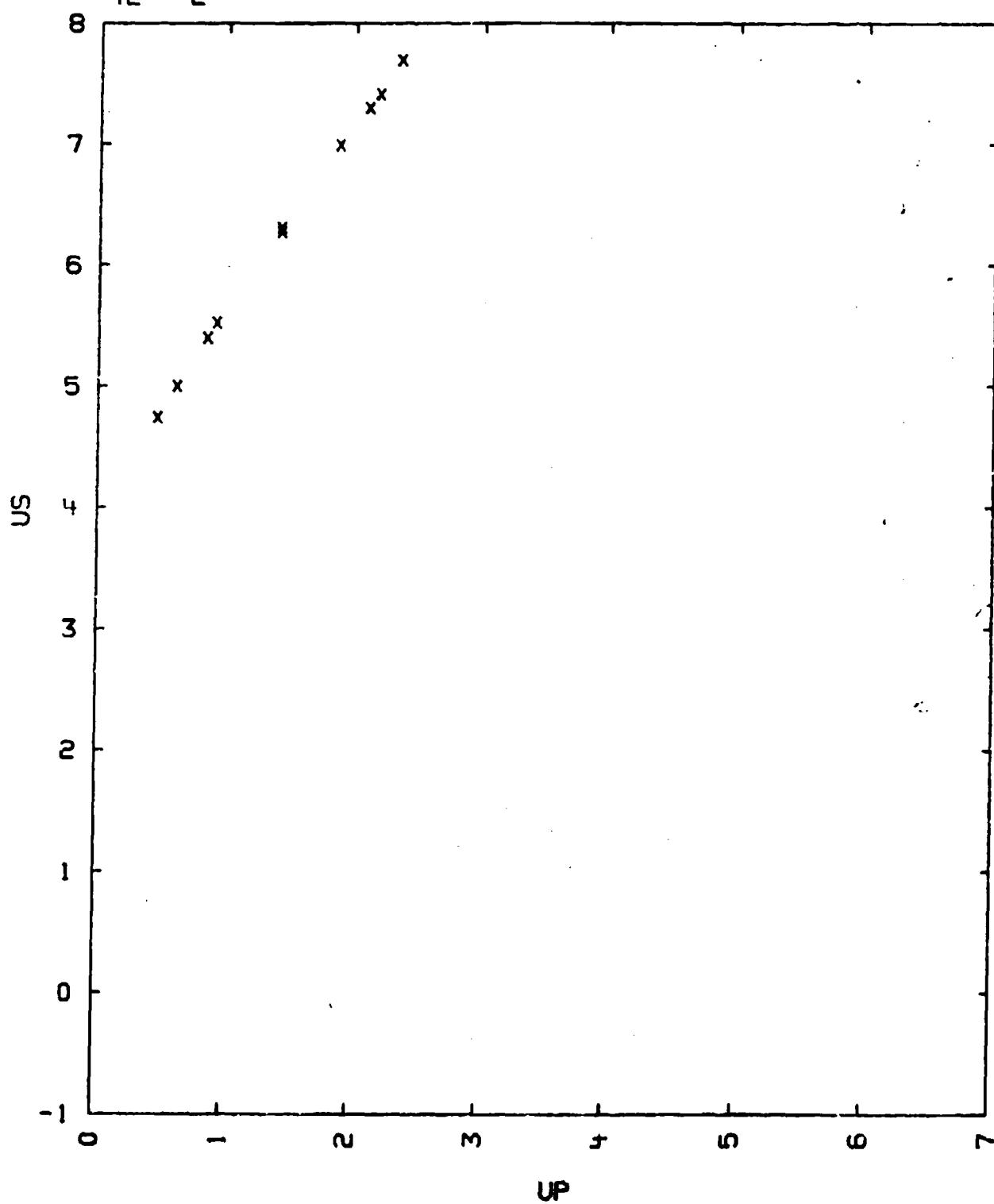
$US = 4.063 + 1.538 UP,$ $SIG.US = 0.03 \text{ KM/SEC.}$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
REPORT NO. GMX-6-566, PP. 51-62 (1964)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS COPPER
- 3) THE COPPER MATERIAL LINEAR EQUATION OF STATE IS REPRESENTED BY
 $US = 3.958 + 1.497 UP \text{ KM/SEC.}$
THE PARTIAL $(\partial E / \partial P)_V = 0.0562 \text{ CUBIC CM.}$
- 4) V_{01} WAS TAKEN FROM A.I.P. HANDBOOK (MCGRAW HILL BOOK CO.)
2ND ED.

TABLE I

PALLADIUM
42---2



42---3
PALLADIUM

PD

$$V_0 = 0.832 \text{ CC/G}$$
$$V_{01} = 0.831 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC., VELOCITIES IN KM/SEC.
AND PRESSURE IN KILOBARS.

TABLE

RHO0	US	UP	P	V/V0
12.02	6.08	1.44	1050	0.784
-	6.22	1.43	1086	0.770
-	6.71	1.81	1460	0.730

$$US = 4.02 + 1.49 UP \text{ KM/SEC}$$

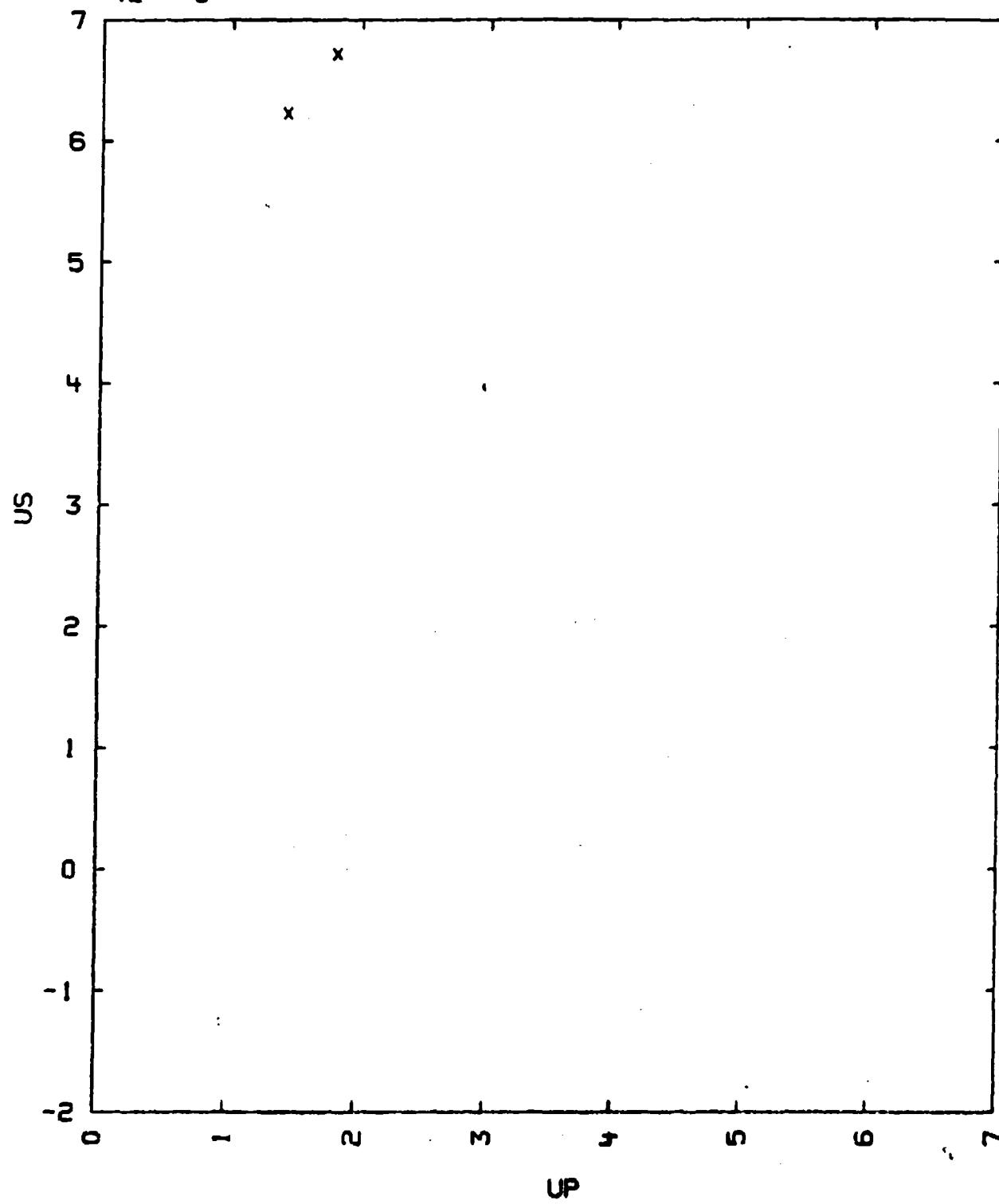
$$\text{SIGMA US} = 0.110 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: LUNDBERG, LENNART.
PRIVATE COMMUNICATION
RESEARCH INSTITUTE OF NATIONAL DEFENSE, STOCKHOLM, SWEDEN
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF VO1 HAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES
(DOVER PUBLICATIONS, INC., NEW YORK, N. Y., 1963).

TABLE I

PALLADIUM
42---3



42--4
PALLADIUM

PO

$V_0 = 0.08333 \text{ CC/G}$ $C_L = 4.57 \text{ KM/SEC}$ $C_0 = 3.90 \text{ KM/SEC}$
 $V_{01} = 0.08328 \text{ CC/G}$ $C_S = 2.06 \text{ KM/SEC}$ $C_B = 3.80 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
12.000	4.67	0.44	247.	0.9058	CU	4.70
11.880	4.82	0.45	247.	0.9026	2024 AL	6.48
12.000	4.92	0.59	348.	0.8801	CU	4.95
11.950	4.94	0.60	354.	0.8785	2024 AL	6.86
12.020	5.31	0.80	511.	0.8493	2024 AL	7.35
12.000	5.32	0.82	523.	0.8459	CU	5.35
12.010	5.44	0.90	588.	0.8346	CU	5.47
12.000	6.18	1.40	1038.	0.7735	CU	6.29
12.000	6.21	1.40	1043.	0.7748	CU	6.30
12.000	6.21	1.40	1043.	0.7746	CU	6.31
12.000	6.88	1.84	1519.	0.7326	CU	7.03
12.000	7.19	2.07	1786.	0.7121	CU	7.39
12.000	7.29	2.15	1881.	0.7051	CU	7.52
12.000	7.58	2.32	2110.	0.6939	CU	7.80

$$US = 3.948 + 1.588 \cdot UP \text{ KM/SEC}$$

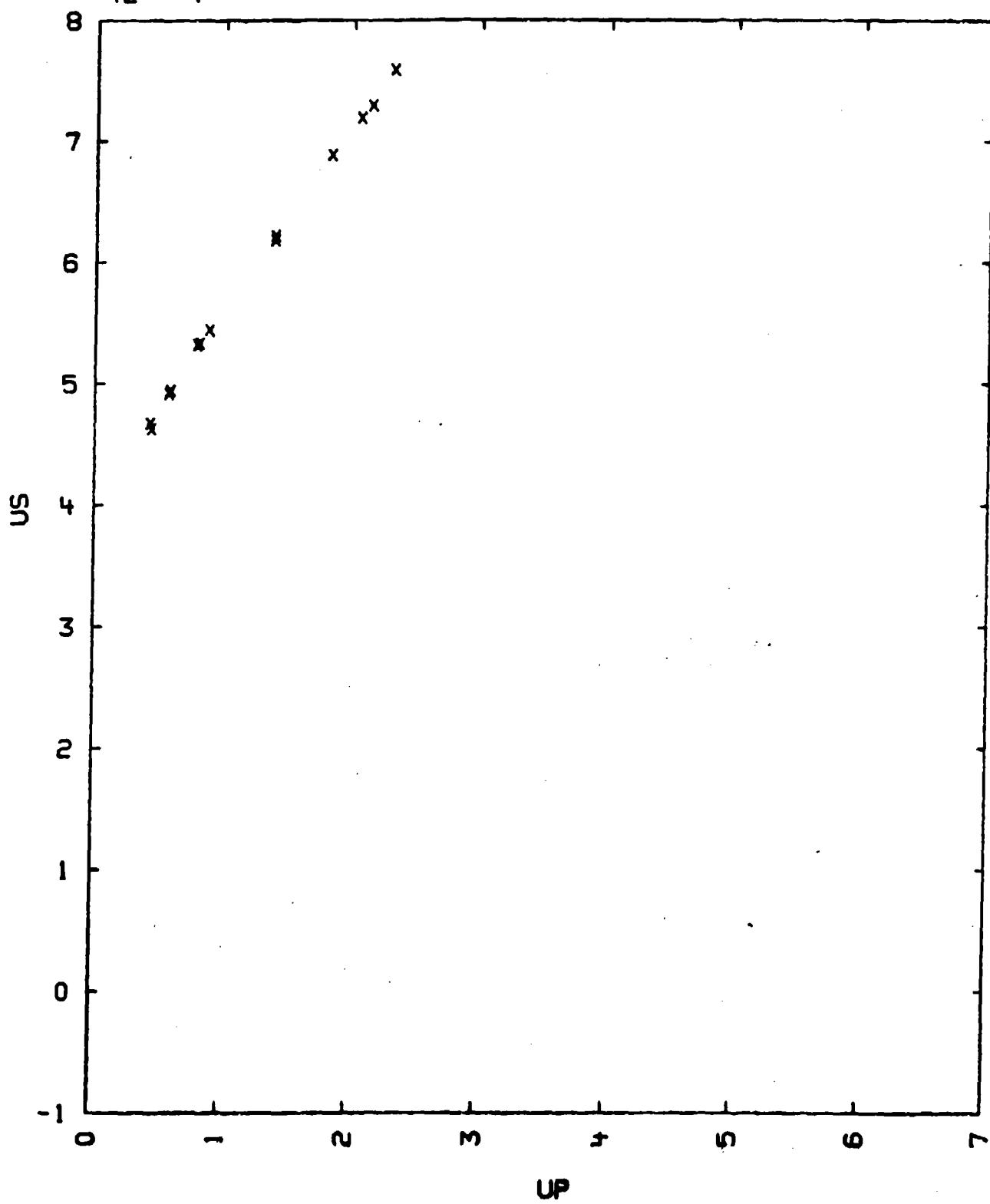
$$\text{SIG US} = 0.041 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.H., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B (STANDARD BASE PLATE AS SHOWN)
- 3) V_{01} IS THE 25 DEG. C. VALUE FROM WYCKOFF CRYSTAL STRUCTURES (JOHN
WILEY AND SONS, N.Y., 1963) VOL. 1
- 4) $V(DP/DE) = 2.260$

TABLE I

PALLADIUM
42---4



43---
RHODIUM

RH

$$V_0 = 0.08045 \text{ CC/G}$$

$$V_{01} = 0.08048 \text{ CC/G}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO	US	UP	P	V/V0	MATERIAL	US(ST)	
12.420	5.40	0.37	248.	0.9315	2024 AL	5.68	
12.430	5.38	0.43	286.	0.9198	CU	4.74	
12.430	5.47	0.54	367.	0.9013	CU	4.94	
12.420	5.79	0.71	511.	0.8774	2024 AL	6.03	
12.430	5.86	0.80	583.	0.8635	CU	5.39	
12.430	5.88	0.84	614.	0.8571	CU	5.46	
12.430	6.26	1.06	825.	0.8307	CU	5.84	
12.430	7.03	1.57	1372.	0.7767	CU	6.70	
12.430	7.11	1.61	1423.	0.7738	CU	6.78	
12.430	7.08	1.68	1478.	0.7627	CU	6.87	
12.430	7.39	1.84	1690.	0.7510	CU	7.15	
12.430	7.47	2.00	1857.	0.7323	CU	7.39	
12.430	7.80	2.19	2132.	0.7192	CU	7.71	

$$US = 4.807 + 1.376 \cdot UP \text{ KM/SEC}$$

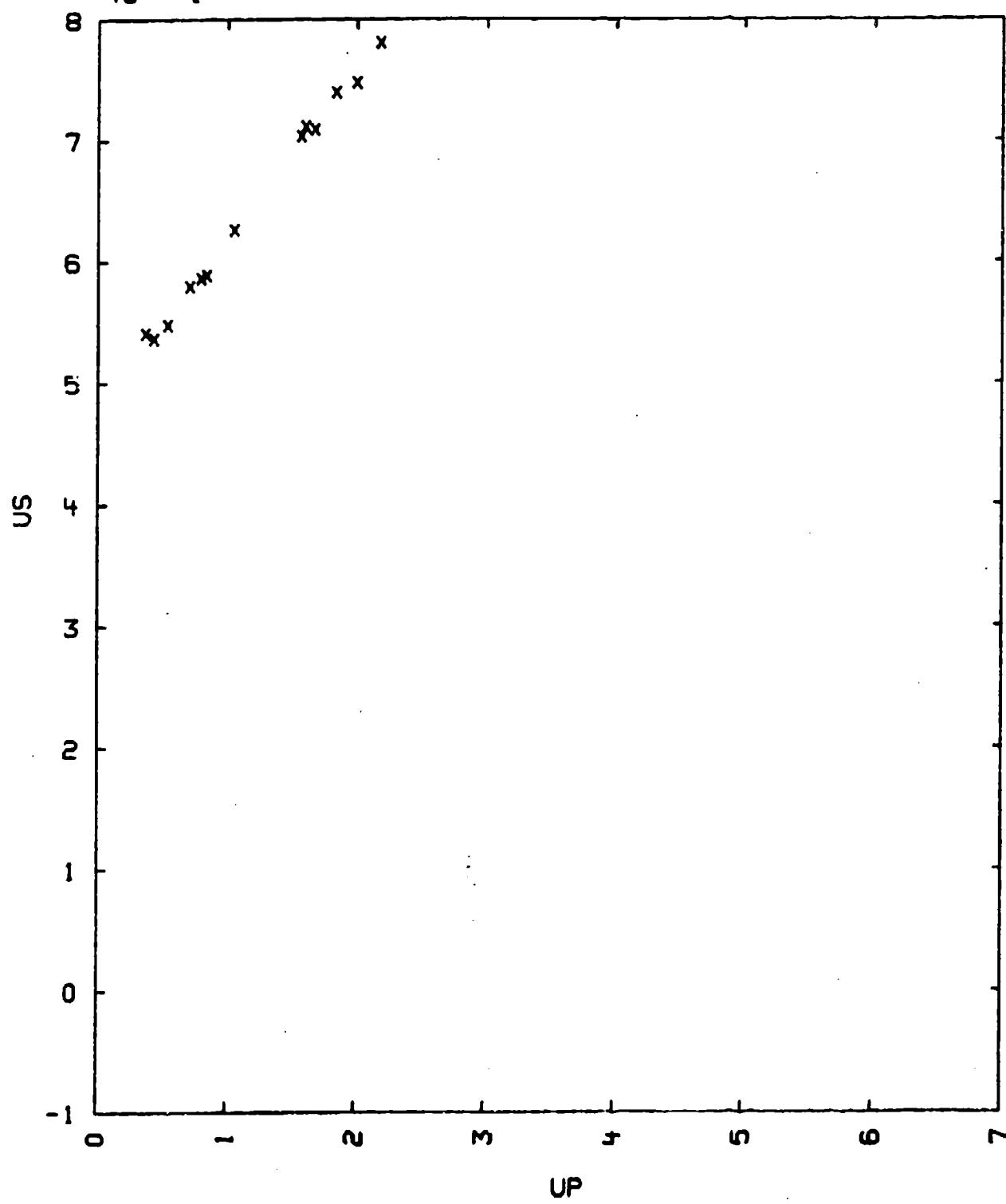
$$SIG US = 0.060 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES.
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B (STANDARD BASEPLATE AS SHOWN)
- 3) V01 IS THE 25 DEG. C VALUE FROM HYCKOFF, CRYSTAL STRUCTURES (JOHN WILEY AND SONS, N.Y., 1963) VOL. 1
- 4) $V(DP/DE) = 1.00$

TABLE I

RHODIUM
43---1



45---1
PLATINUM

PT 99.9 PER CENT OR GREATER

$V_0 = 0.04679 \text{ CC/G.}$
 $V_{01} = 0.04651 \text{ CC/G.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

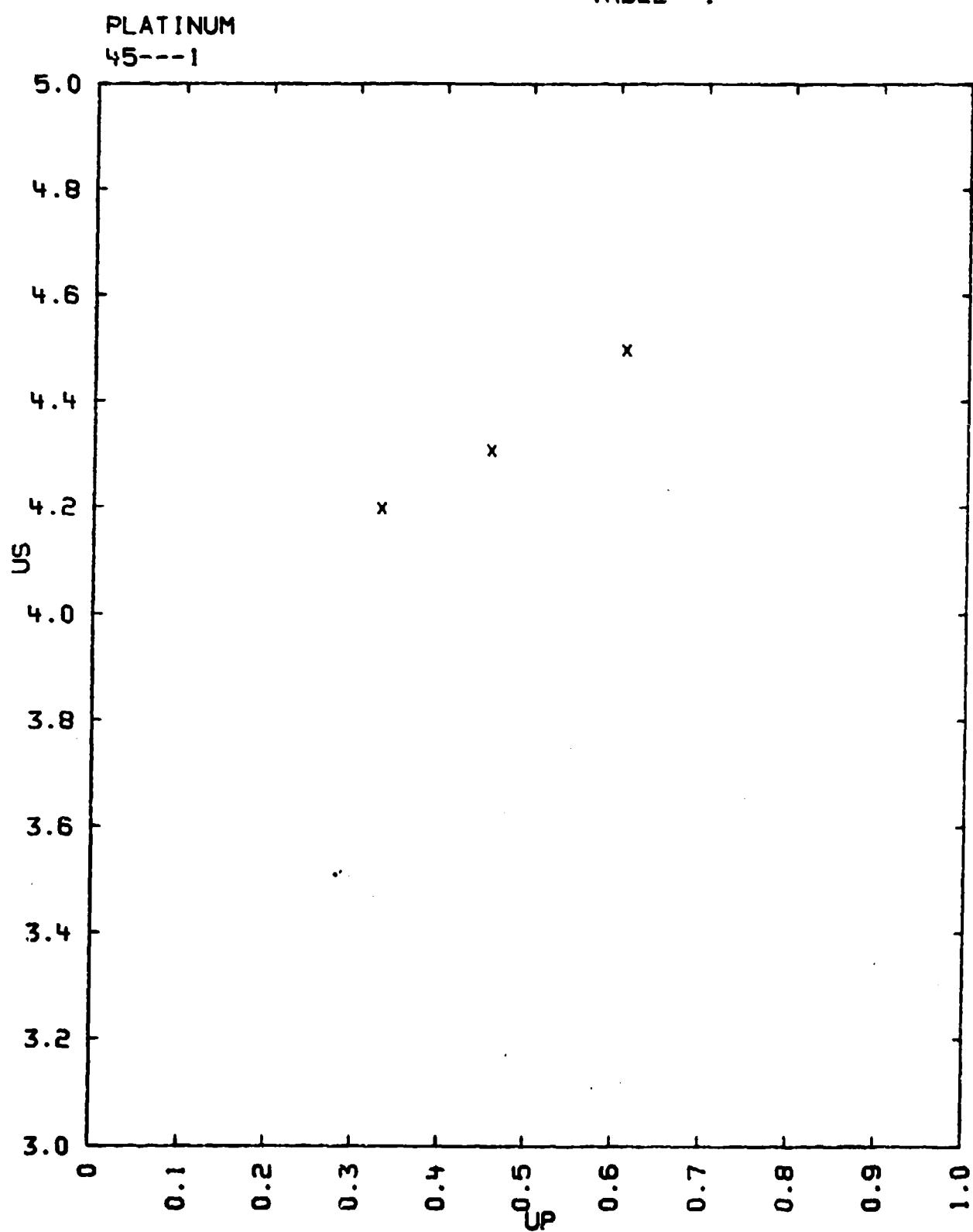
RHO0	US	UP	P	V/V0
21.37	4.199	0.329	295	0.9238
-	4.306	0.4550	416.5	0.8943
-	4.495	0.6102	586	0.8642

$US = 3.841 + 1.059 UP \text{ KM/SEC}$ SIGMA US = 0.5 PERCENT

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 3 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS.

TABLE I



45--2
PLATINUM

PT 99.5 PERCENT OR GREATER

$V_01 = 0.0465 - CS = 1.79 - CB = 3.41 -$
 $V_01 = 0.0465 \text{ CCG. } CS = 1.79 \text{ KM/SEC. } CB = 3.41 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC.,
PRESSURE IN KILOBARS AND DENSITIES IN G/CC.

TABLE

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	RHO0	US
21.43	4.179	0.360	322.4	0.9139	8.932	4.764
-	4.365	0.483	451.8	0.8894	8.934	5.024
-	4.402	0.488	460.3	0.8891	-	5.040
-	4.682	0.675	677.3	0.8558	-	5.426
-	4.778	0.738	755.7	0.8455	-	5.556
-	5.420	1.150	1335.7	0.7878	-	6.387
-	5.465	1.152	1349.2	0.7892	-	6.400
-	5.442	1.180	1376.1	0.7832	-	6.443
-	6.014	1.526	1968.7	0.7463	-	7.133
-	6.316	1.700	2301.0	0.7308	-	7.482
-	6.367	1.787	2438.3	0.7193	-	7.632
-	6.548	1.937	2718.1	0.7042	-	7.910

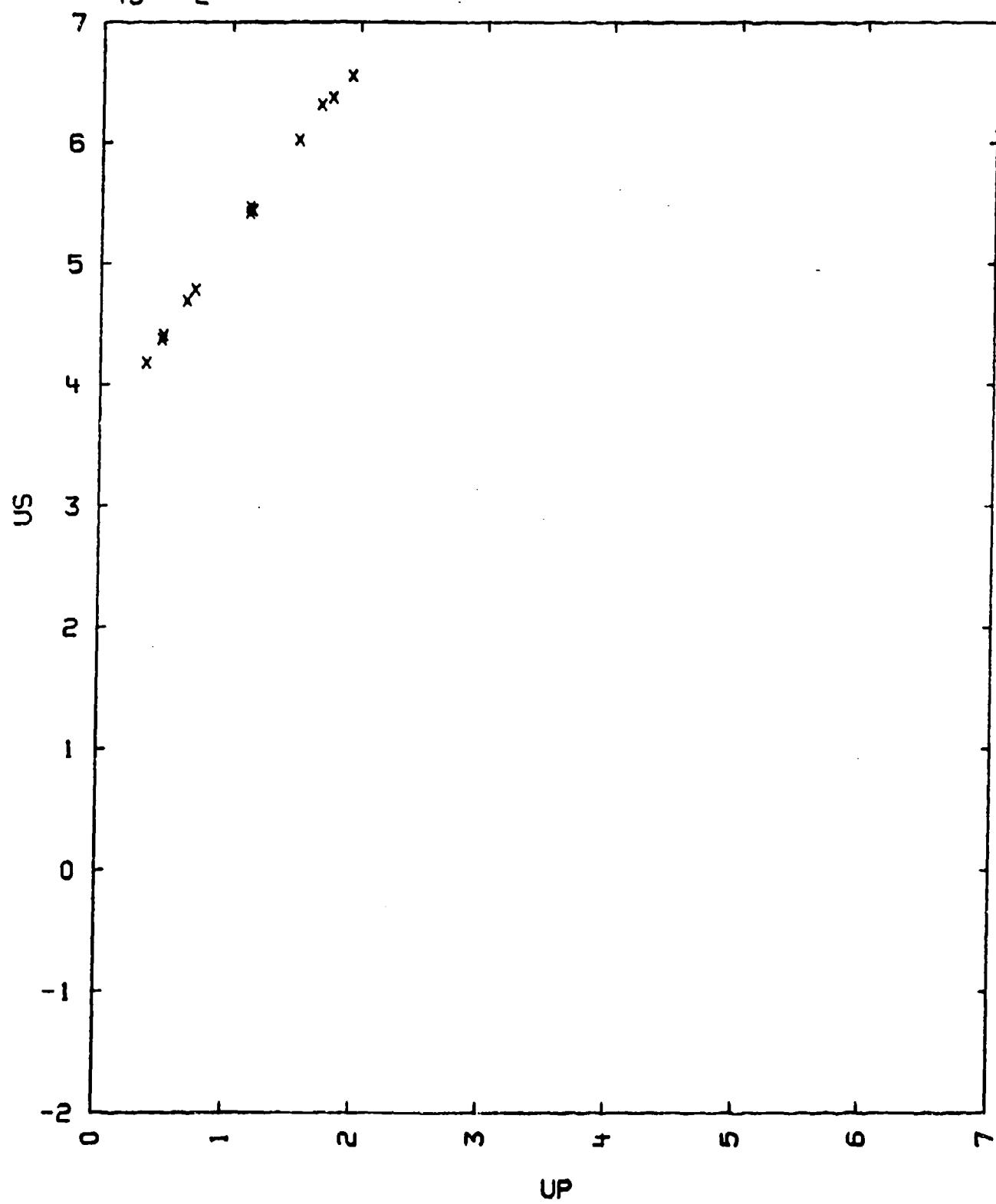
$$US = 3.646 + 1.535 UP, \quad S10.US = 0.037 \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
REPORT NO. GMX-B-566, PP. 51-62 (1964)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS COPPER
- 3) THE COPPER HUGONIOT IS REPRESENTED BY
US = 3.858 + 1.497 UP KM/SEC.
THE PARTIAL (DE/DP)V = 0.0562 CUBIC CM.
- 4) V01 WAS TAKEN FROM THE AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCRAH-HILL BOOK CO. N.Y., 1963) 2ND ED.

TABLE I

PLATINUM
45---2



45---3
PLATINUM

PT 99.97 PER CENT OR GREATER

$V_0 = 0.04664 \text{ CC/G}$
 $V_{01} = 0.04662 \text{ CC/G}$

THE TABLE LISTS DENSITY IN G/CC, VELOCITIES IN KM/SEC AND PRESSURE IN KBARS.

TABLE

RHO0	US	UP	P	V1/V0
21.46	6.67	2.04	2920.	0.694
21.44	6.73	2.14	3080.	0.682
21.45	6.96	2.26	3380.	0.675
21.44	6.91	2.27	3360.	0.671
21.44	7.23	2.38	3880.	0.671
21.45	7.13	2.48	3790.	0.652
21.45	7.17	2.49	3830.	0.653
21.43	7.39	2.52	4000.	0.658
21.46	7.45	2.60	4160.	0.651
21.44	7.49	2.66	4270.	0.645
21.44	7.64	2.70	4420.	0.647
21.47	7.58	2.73	4440.	0.639
21.44	7.62	2.74	4470.	0.641
21.44	7.83	2.88	4840.	0.631
21.44	7.77	2.90	4830.	0.627
21.45	7.90	2.92	4950.	0.630
21.44	8.25	3.04	5380.	0.632
21.45	8.21	3.08	5420.	0.625
21.45	8.25	3.18	5620.	0.615
21.44	8.18	3.18	5580.	0.611
21.44	8.37	3.19	5730.	0.619
21.44	8.46	3.25	5890.	0.616
21.45	8.59	3.28	6040.	0.618
21.45	8.59	3.37	6210.	0.608
21.45	8.87	3.46	6590.	0.609
21.42	8.60	3.48	6410.	0.596
21.45	8.75	3.49	6550.	0.601
21.42	8.91	3.49	6660.	0.608
21.42	8.88	3.49	6640.	0.607
21.44	8.67	3.52	6540.	0.594
21.43	8.87	3.59	6820.	0.596

$US = 3.53 + 1.50 \cdot UP \text{ KM/SEC}$
 $SIG US = 0.092 \text{ KM/SEC}$

COMMENTS:

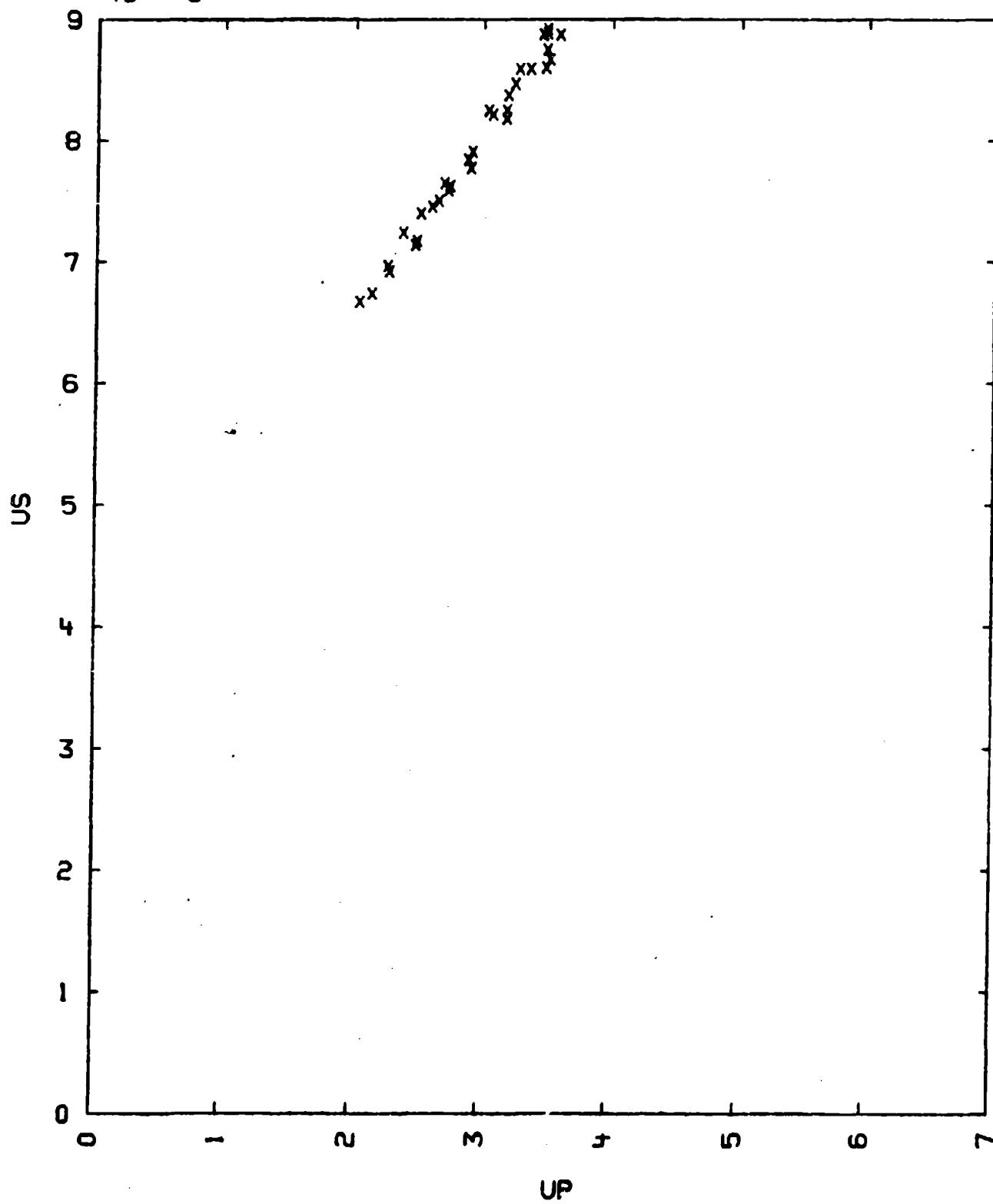
1) SOURCE: MORGAN,J.A.

HIGH TEMP HIGH PRES. VOL 6, P. 195 (1974)

- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE A
- 3) SYMMETRIC IMPACT USING ACCELERATING RESERVOIR LIGHT GAS GUN
- 4) THE PRECISION IN UP=1/2 IMPACT VELOCITY IS 0.2 PERCENT AND IN US
IS 1. PERCENT
- 5) THE SAMPLES WERE CAST ROLLED AND PEENED TO THE LISTED DENSITY

TABLE I

PLATINUM
45---3



45---4
PLATINUM

PT

$$V_0 = 0.04666 \text{ CC/G} \quad C_L = 4.08 \text{ KM/SEC} \quad C_0 = 3.51 \text{ KM/SEC}$$

$$V_{01} = 0.04660 \quad - \quad C_S = 1.79 \quad - \quad C_B = 3.64 \quad -$$

THE TABLE LISTS VELOCITIES IN KM/SEC, PRESSURE IN KBARS AND DENSITY IN G/CC. STANDARD MATERIAL DESCRIBES THE IMPACTOR

TABLE

-----SAMPLE-----					----STANDARD----	
RHO0	US	UP	P	V/V0	MATERIAL	
21.370	4.15	0.32	284.	0.9229	2024 AL	6.48
21.430	4.12	0.34	300.	0.9173	CU	4.70
21.370	4.25	0.44	400.	0.8965	2024 AL	6.86
21.430	4.30	0.46	424.	0.8930	CU	4.93
21.430	4.34	0.47	437.	0.8917	CU	4.97
21.370	4.44	0.60	569.	0.8649	2024 AL	7.32
21.430	4.61	0.65	642.	0.8590	CU	5.35
21.430	4.71	0.71	717.	0.8493	CU	5.48
21.430	5.34	1.12	1282.	0.7903	CU	6.29
21.430	5.39	1.12	1294.	0.7922	CU	6.31
21.430	5.38	1.15	1321.	0.7854	CU	6.35
21.430	5.93	1.49	1893.	0.7487	CU	7.03
21.430	6.22	1.66	2213.	0.7331	CU	7.37
21.430	6.28	1.75	2355.	0.7213	CU	7.52
21.430	6.45	1.90	2626.	0.7054	CU	7.80

$$US = 3.598 + 1.544 \cdot UP$$

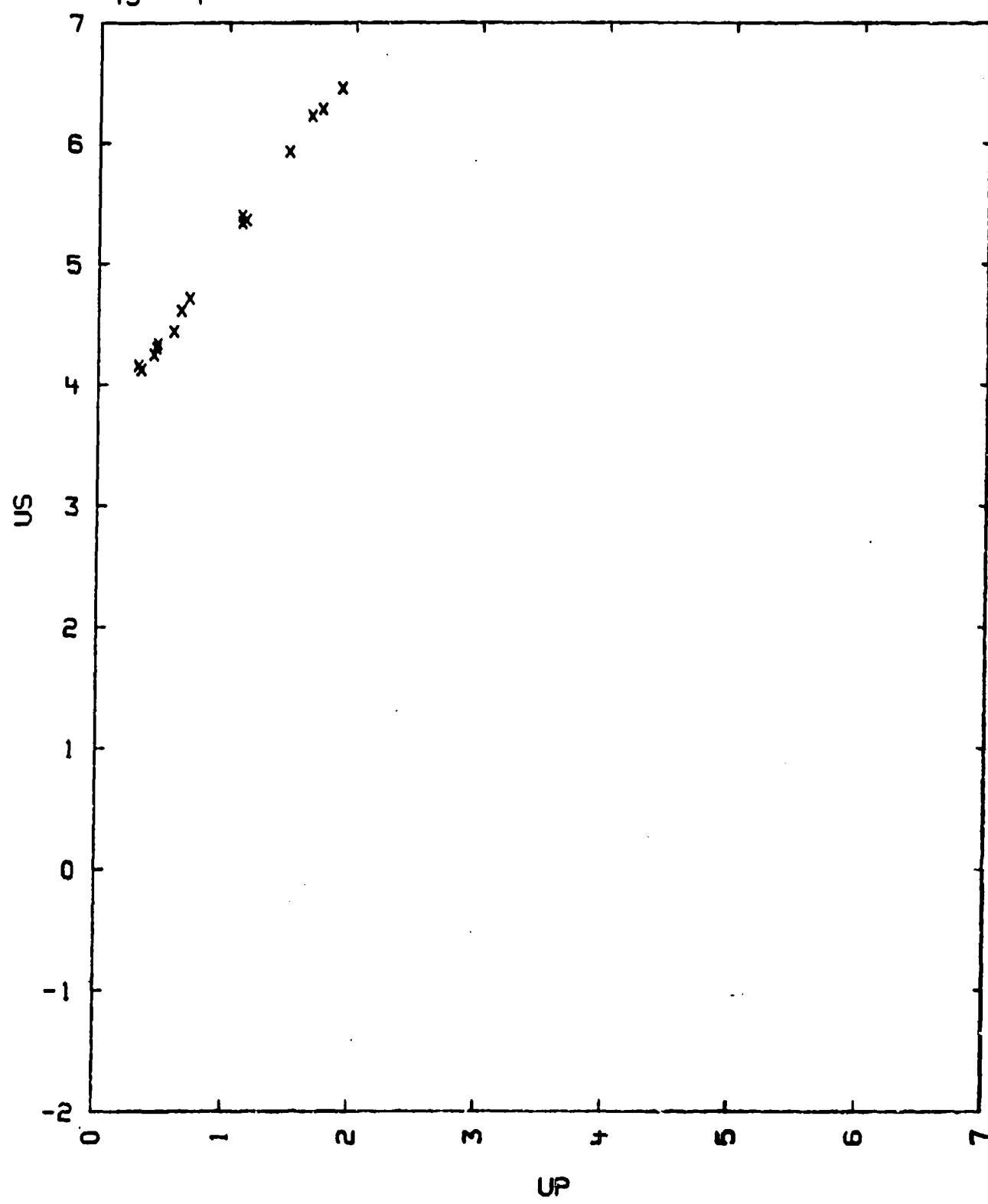
$$\Sigma \sigma US = 0.046 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.H., FRITZ, J.M., AND CARTER, H.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B (STANDARD BASEPLATE AS SHOWN)
- 3) V01 IS THE 25 DEG. C VALUE FROM HYCKOFF, CRYSTAL STRUCTURES (JOHN
WILEY AND SONS, N.Y., 1963) VOL. 1
- 4) V(DP/DE)=2.4

TABLE I

PLATINUM
45---4



46---1
IRIDIUM

IR

$$V_0 = 0.4452 \text{ CC/G}$$

$$V_{01} = 0.4434 \text{ CC/G}$$

THE TABLE LISTS VELOCITIES IN KM/SEC, PRESSURE IN KBARS AND DENSITY IN G/CC. STANDARD MATERIAL DESCRIBES THE IMPACTOR

TABLE

-----SAMPLE-----					----STANDARD----	
RHO0	US	UP	P	V/V0	MATERIAL	
22.420	4.18	0.19	178.	0.9545	CU	4.41
22.500	4.90	0.67	739.	0.8633	CU	5.46
22.420	5.44	0.93	1134.	0.8290	CU	6.04
22.500	5.90	1.36	1805.	0.7693	CU	6.87
22.500	6.26	1.63	2296.	0.7396	CU	7.39
22.500	6.73	1.93	2923.	0.7132	CU	7.99

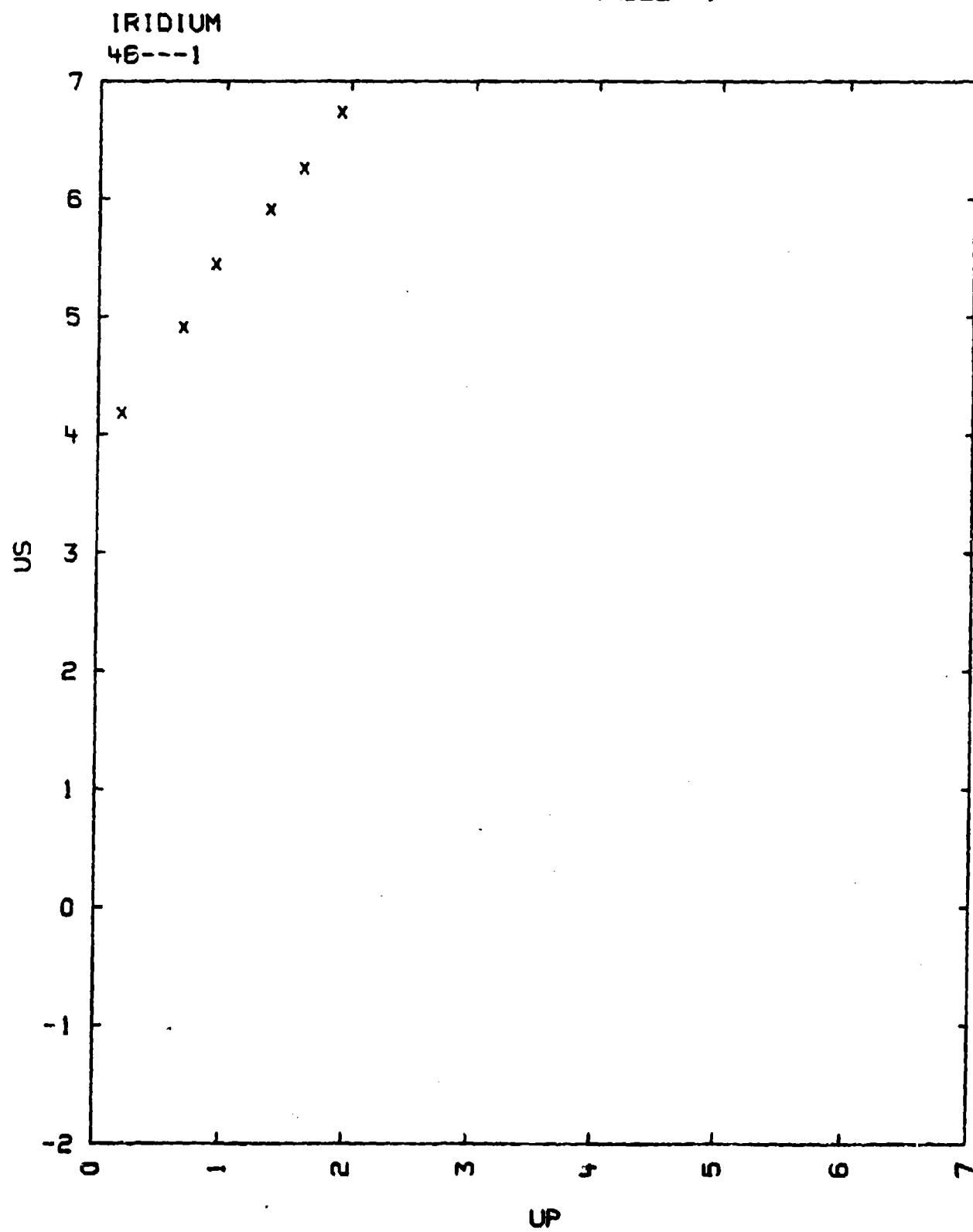
$$US = 3.916 + 1.457 \cdot UP$$

$$\text{SIGMA US} = 0.081 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, H.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B (STANDARD BASEPLATE AS SHOWN)
- 3) V01 IS THE 25 DEG. C VALUE FROM HYCKOFF, CRYSTAL STRUCTURES (JOHN
WILEY AND SONS, N.Y., 1963) VOL. I
- 4) V(DP/DE)=1.97

TABLE I



50---1
RHENIUM

RE

$$V_0 = 0.04764 \text{ CC/G} \quad C_L = 5.33 \text{ KM/SEC} \quad C_0 = 4.16 \text{ KM/SEC}$$

$$V_{01} = 0.04759 \text{ CC/G} \quad C_S = 2.89 \text{ KM/SEC}$$

IN THE TABLES BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOGBAR
AND DENSITY IN G/CC.

TABLE I

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
20.990	4.69	0.33	325.	0.9296	CU	4.72
20.960	4.74	0.43	427.	0.9093	CU	4.94
20.980	5.04	0.66	698.	0.8690	CU	5.42
20.990	5.19	0.69	752.	0.8671	CU	5.49
20.990	5.51	0.96	1110.	0.8258	CU	6.04
20.990	5.71	1.13	1354.	0.8021	CU	6.36

$$US = 4.184 + 1.367 \cdot UP \text{ KM/SEC}$$

$$\text{SIGMA US} = 0.050 \text{ KM/SEC}$$

$$V_0 = 0.04871 \text{ CC/G}$$

$$C_0 = 4.10 \text{ KM/SEC}$$

TABLE II

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
20.530	4.15	0.21	179.	0.9494	CU	4.41
20.530	4.08	0.37	310.	0.9093	CU	4.74
20.530	4.29	0.48	423.	0.8881	CU	4.96
20.530	4.77	0.68	606.	0.8374	CU	5.39
20.530	4.79	0.72	708.	0.8497	CU	5.48
20.530	5.74	1.35	1581.	0.7648	CU	6.70
20.530	5.87	1.38	1663.	0.7649	CU	6.78
20.530	5.81	1.44	1718.	0.7522	CU	6.87
20.530	6.04	1.58	1959.	0.7384	CU	7.15
20.530	6.15	1.72	2172.	0.7203	CU	7.39
20.530	6.69	2.03	2788.	0.6986	CU	7.99

$$US = 3.653 + 1.596 \cdot UP - 0.053 \cdot UP^2 \text{ KM/SEC}$$

$$\text{SIG US} = 0.11 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.H., FRITZ, J.M.,
AND CARTER, H.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES.
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B (STANDARD BASE PLATE AS SHOWN)
- 3) VOL IS THE 25 DEG. C. VALUE FROM WYCKOFF, CRYSTAL STRUCTURES (JOHN
WILEY AND SONS, N.Y., 1963) VOL. 1
- 4) V(DP/DE) = 2.440

TABLE I

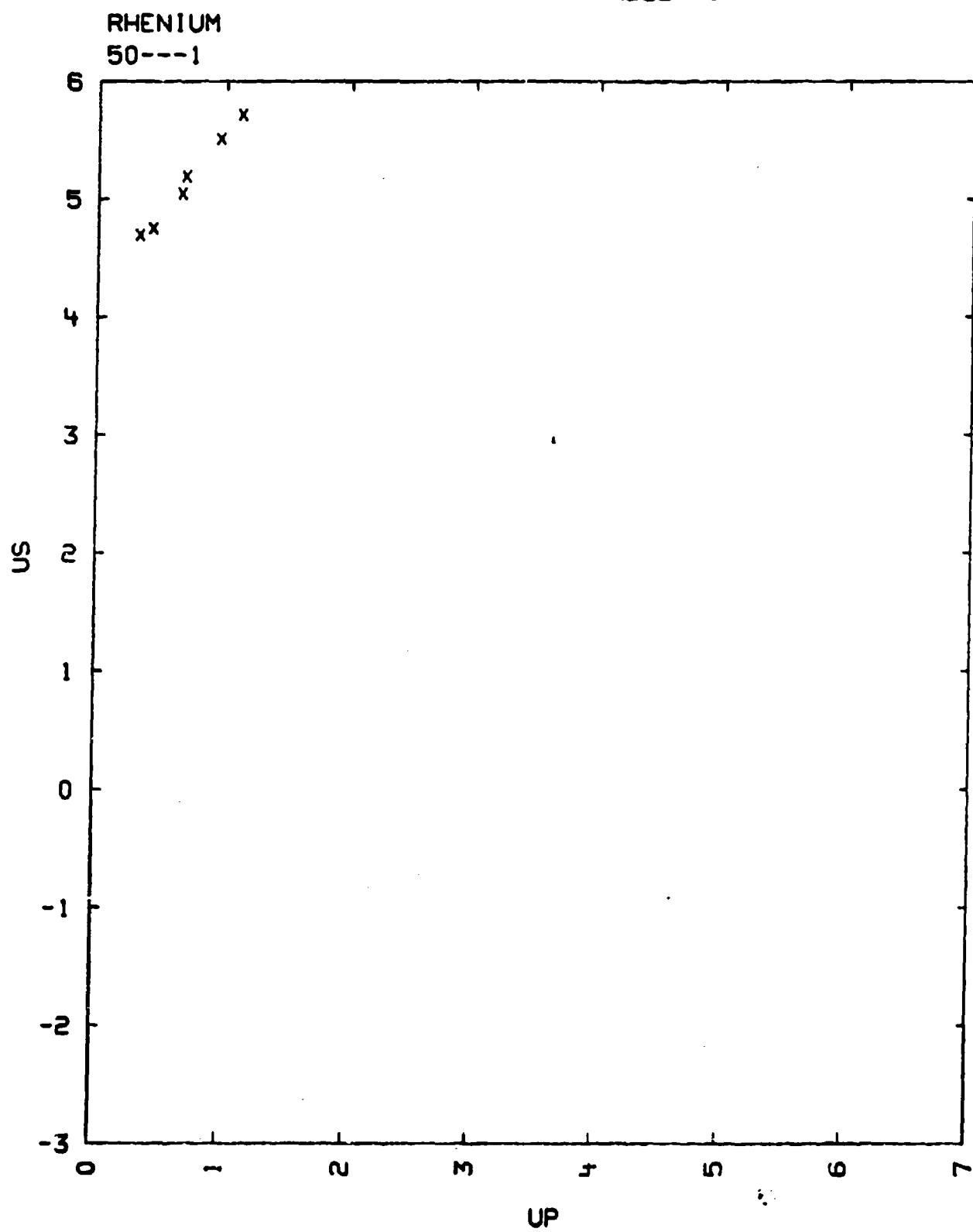
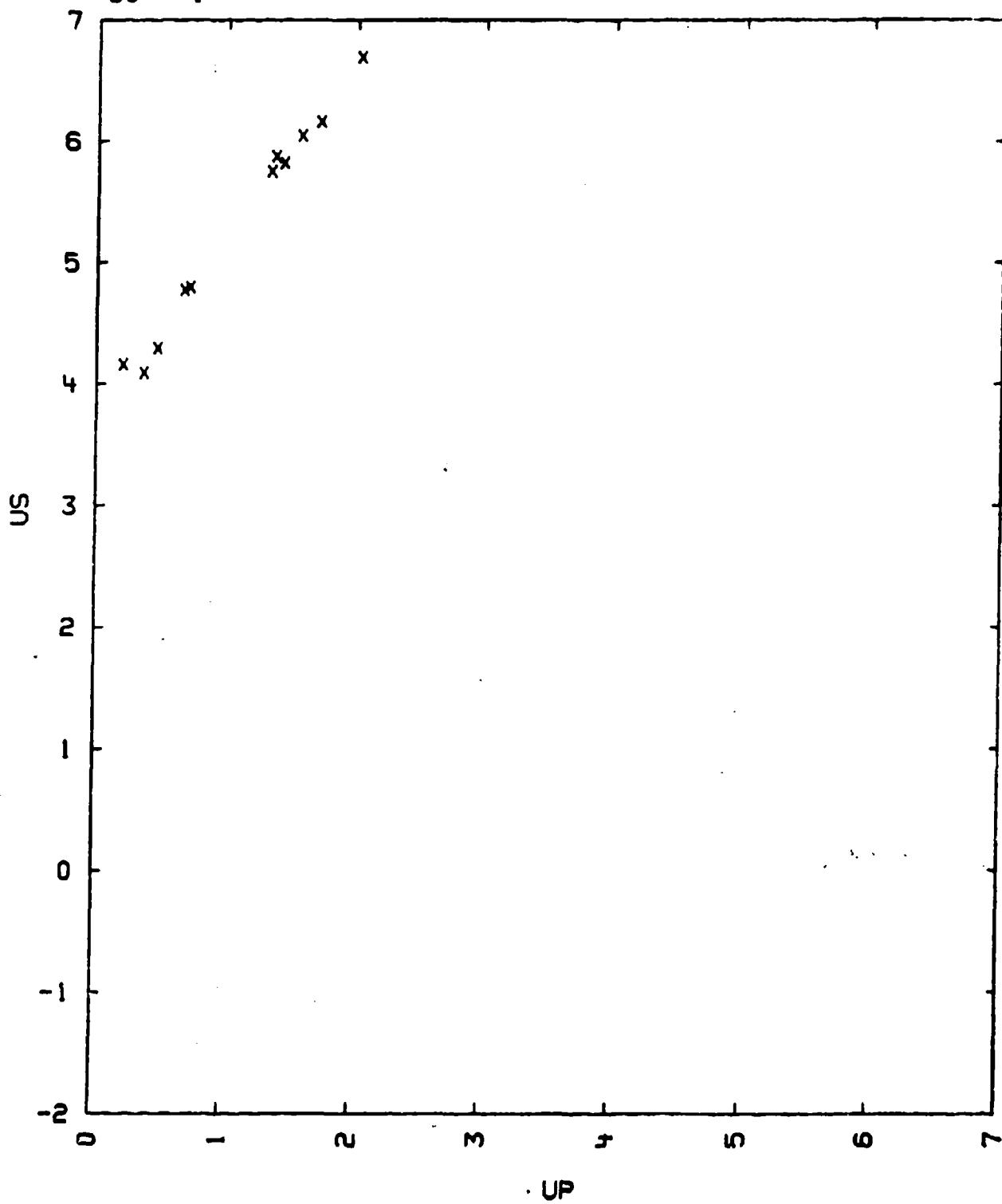


TABLE II

RHENIUM
50---1



51---1
CHROMIUM

CR 99.8 PER CENT OR GREATER

$$V_0 = 0.1403 \text{ CC/G.}$$

$$V_{01} = 0.1385 \text{ CC/G.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

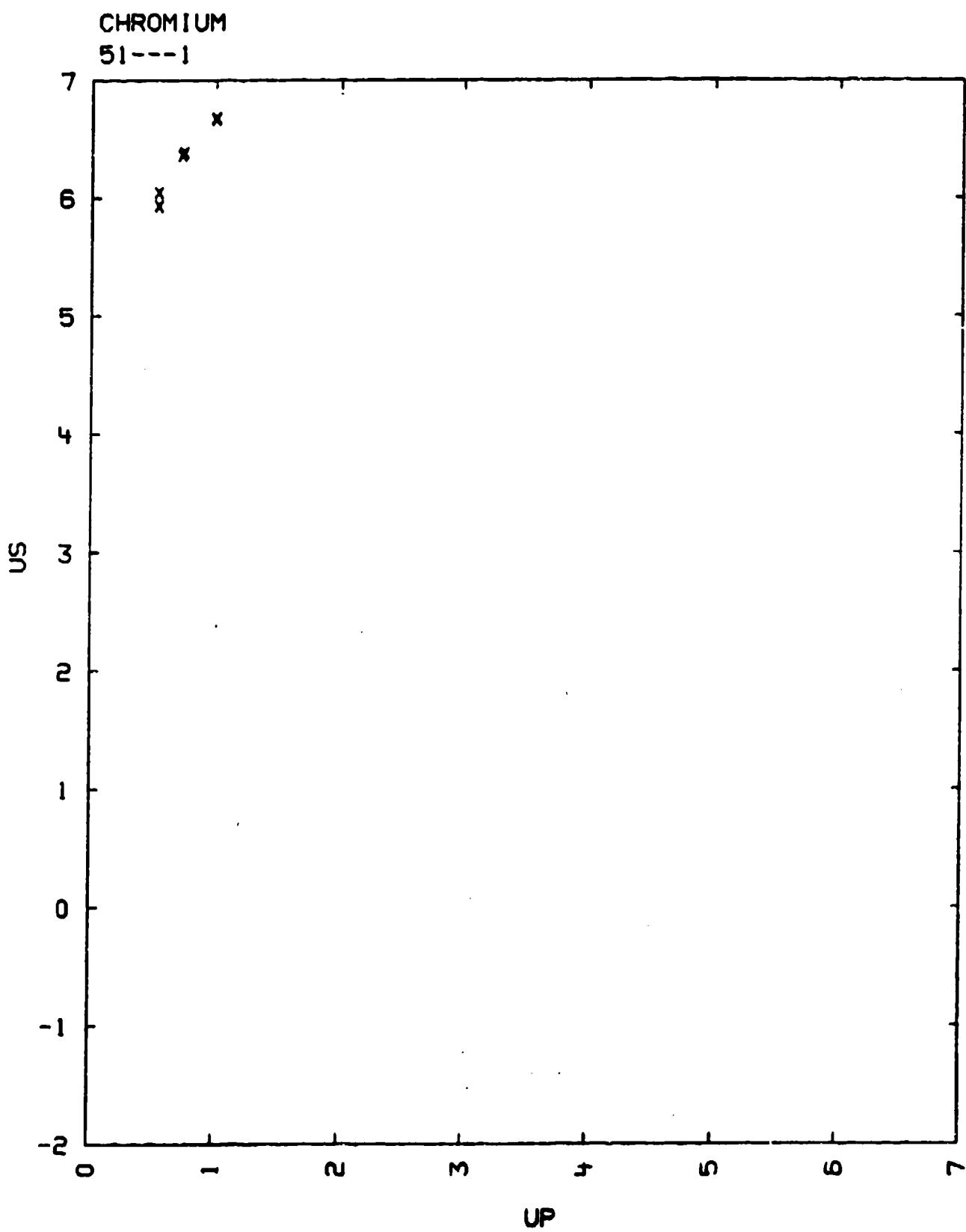
RHO0	US	UP	P	V/V0
7.13	6.043	0.5448	234.5	0.9098
-	5.923	0.5395	233	0.9089
-	6.381	0.7436	338	0.8835
-	6.370	0.7448	338	0.8831
-	6.355	0.7407	336	0.8834
-	6.357	0.7403	336	0.8835
-	6.660	1.007	478	0.8488
-	6.674	1.008	478	0.8490

$$US = 5.249 + 1.445 UP \text{ KM/SEC} \quad \text{SIGMA US} = 1.0 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 189 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS
- 4) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.

TABLE I



51---2
CHROMIUM

CR

$V_0 = 0.141 \text{ CC/G.}$
 $V_{01} = 0.139 \text{ CC/G.}$

$C_0 = 5.15 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
7.10	7.63	1.71	924	0.777	914
-	7.59	1.71	922	0.775	914
-	8.44	2.25	1347	0.734	1363
-	8.57	2.27	1382	0.735	1369
-	8.63	2.25	1379	0.739	1358

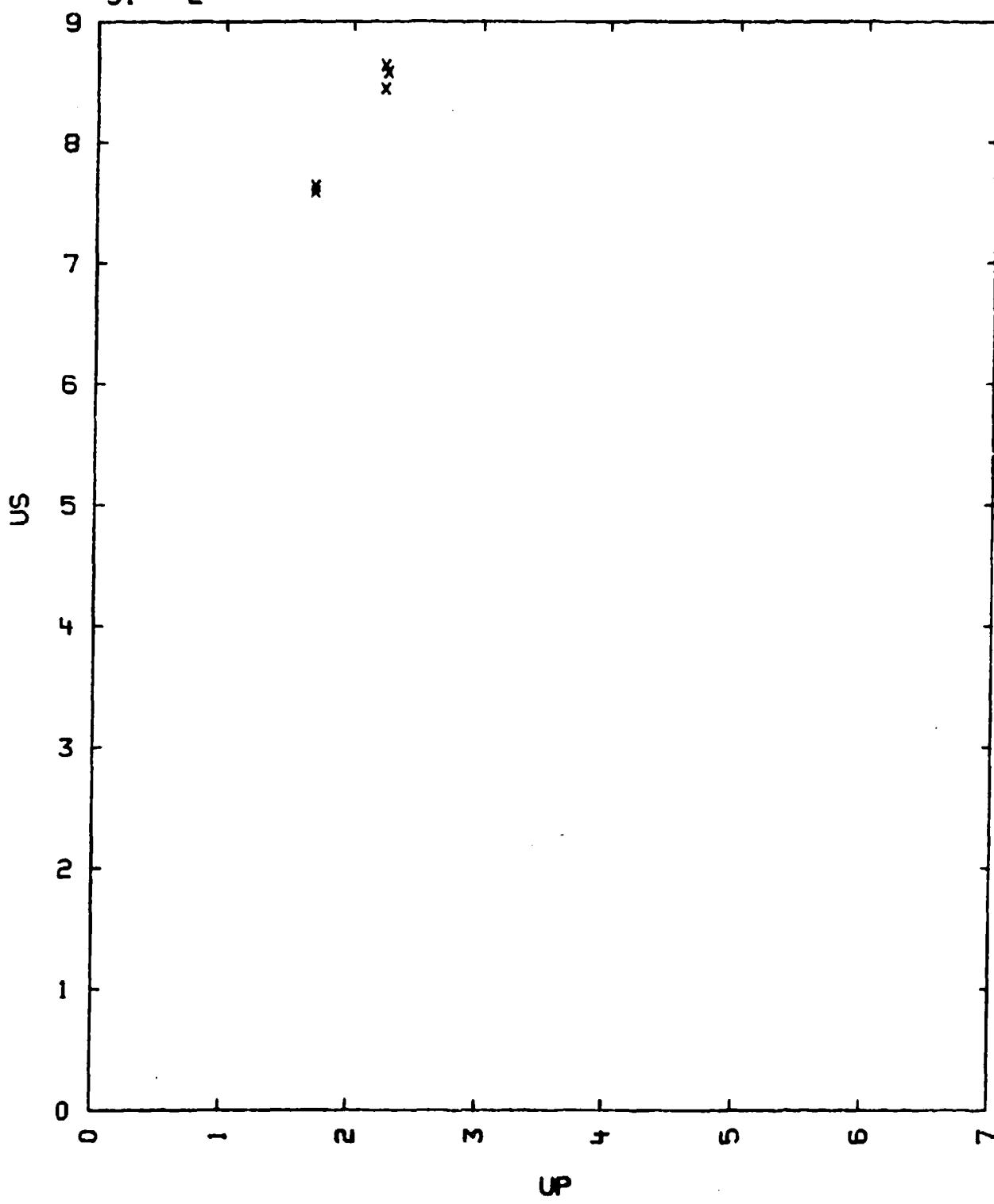
$US = 5.217 + 1.465 UP \text{ KM/SEC.}$ SIGMA US = 0.63 PERCENT

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 4) V01 WAS TAKEN FROM THE AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO. N.Y., 1963) 2ND ED.

TABLE I

CHROMIUM
51---2



51---3
CHROMIUM

CR

$$V_0 = 0.1402 - 0.1408 \text{ CC/G} \quad C_L = 6.60 \text{ KM/SEC} \quad C_0 = 4.32 \text{ KM/SEC}$$

$$V_{01} = 0.1389 \quad C_S = 4.32 \quad C_B = 5.15$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(IST)
7.100	5.15	0.	0.	1.0000	C0	0.
7.130	5.85	0.52	217.	0.9111	2024 AL	6.43
7.130	5.97	0.52	221.	0.9129	2024 AL	6.46
7.130	6.28	0.72	322.	0.8854	2024 AL	6.85
7.130	6.28	0.72	322.	0.8854	2024 AL	6.85
7.130	6.30	0.72	323.	0.8857	2024 AL	6.86
7.130	6.29	0.72	323.	0.8855	2024 AL	6.86
7.130	6.58	0.98	460.	0.8511	2024 AL	7.36
7.130	6.59	0.98	460.	0.8513	2024 AL	7.36
7.123	7.20	1.44	739.	0.8000	CU	6.04
7.100	7.54	1.65	883.	0.7812	BRASS	6.17
7.100	7.50	1.66	884.	0.7787	BRASS	6.18
7.100	8.34	2.19	1297.	0.7374	BRASS	6.93
7.100	8.53	2.19	1326.	0.7433	BRASS	6.96
7.100	8.47	2.21	1329.	0.7391	BRASS	6.97

$$US = 5.173 + 1.473 \cdot UP \text{ KM/SEC}$$

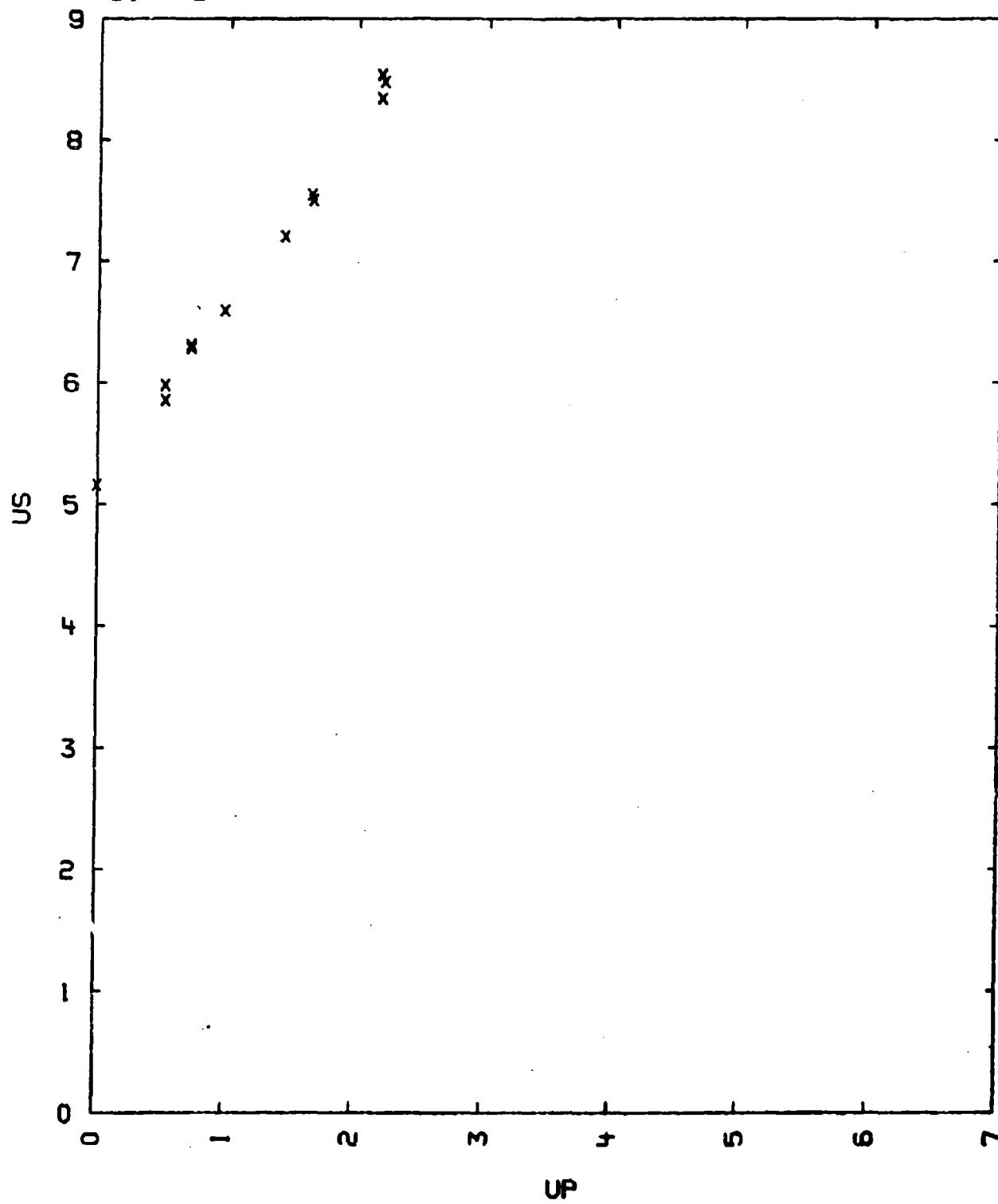
$$\text{SIGMA US} = 0.074 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES.
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B
- 3) V01 FROM WYCKOFF CRYSTAL STRUCTURES (JOHN WILEY AND SONS, N.Y., 1963) VOL. 1
- 4) V(DP/DE) = 1.19
- 5) HUGONIOT ELASTIC LIMIT 18. KBAR

TABLE I

CHROMIUM
51---3



52---0
MOLYBDENUM SUMMARY

MO

$$V_0 = 0.0980 - 0.0987 \text{ CC/G}$$

$$V_{01} = 0.09785 \text{ CC/G}$$

$$C_B = 5.19 \text{ KM/SEC}$$

THE TABLE LISTS THE HUGONIOT POINTS CALCULATED FROM THE FIT GIVEN BELOW.
UNITS ARE: G/CC, KM/SEC, KBAR AND KBAR.CC/G FOR THE ENERGY DIFFERENCE.

TABLE

RHO0	US	UP	P	V/V0	E-E0
10.20	5.645	.4	230.	0.9291	0.8
	6.021	.7	430	0.8837	2.45
	6.398	1.	652	0.844	5.0
	7.025	1.5	1075	0.786	11.25
	7.653	2.	1561	0.739	20.0
	8.908	3.	2726	0.663	45.0
	10.163	4.	4146	0.606	80.0
	10.790	4.5	4953	0.583	101.

$$US = 5.143 + 1.255 UP, \text{ SIG.US} = 0.06 \text{ KM/SEC}$$

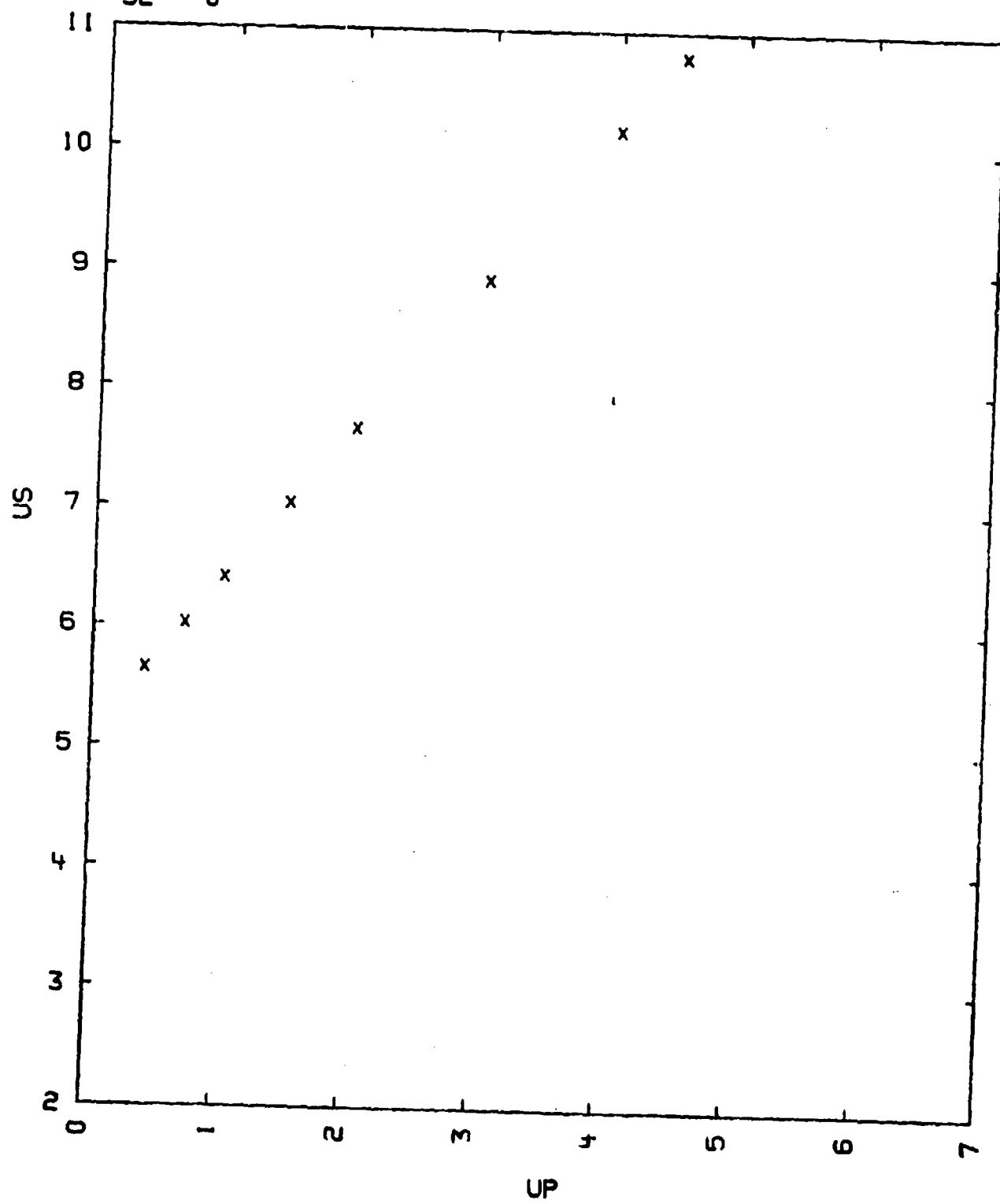
FOR UP BETWEEN .4 AND 4.3 KM/SEC.

COMMENTS:

- 1) SOURCE: COMPILER
- 1A) DATA FROM 52---1,2 AND 3 WERE USED TO MAKE THE FIT
- 2) V01 HAS OBTAINED FROM THE CUBIC LATTICE CONSTANT A = 3.1473 ANGSTROM AT 25 DEG. C. LISTED BY: HYCKOFF, CRYSTAL STRUCTURES, VOL 1, (INTERSCIENCE PUBL., N. Y. 1963) 2ND. ED.
- 3) CB HAS TAKEN FROM 52---2.

MOLYBDENUM SUMMARY
52---0

TABLE I



52---1
MOLYBDENUM

Mo 99.8 PER CENT OR GREATER

$V_0 = 0.0980 \text{ CC/G.}$
 $V_{01} = 0.0980 \text{ CC/G.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

RHO0	US	UFS	UP	P	V/V0
10.20	5.699	0.874	0.437	254.0	0.9233
-	5.647	0.888	0.444	255.2	0.9214
-	5.955	1.176	0.591	358.0	0.9008
-	5.861	1.200	0.606	362.3	0.8966
-	6.210	1.724	0.850	538.4	0.8631
-	6.124	1.636	0.792	494.7	0.8707

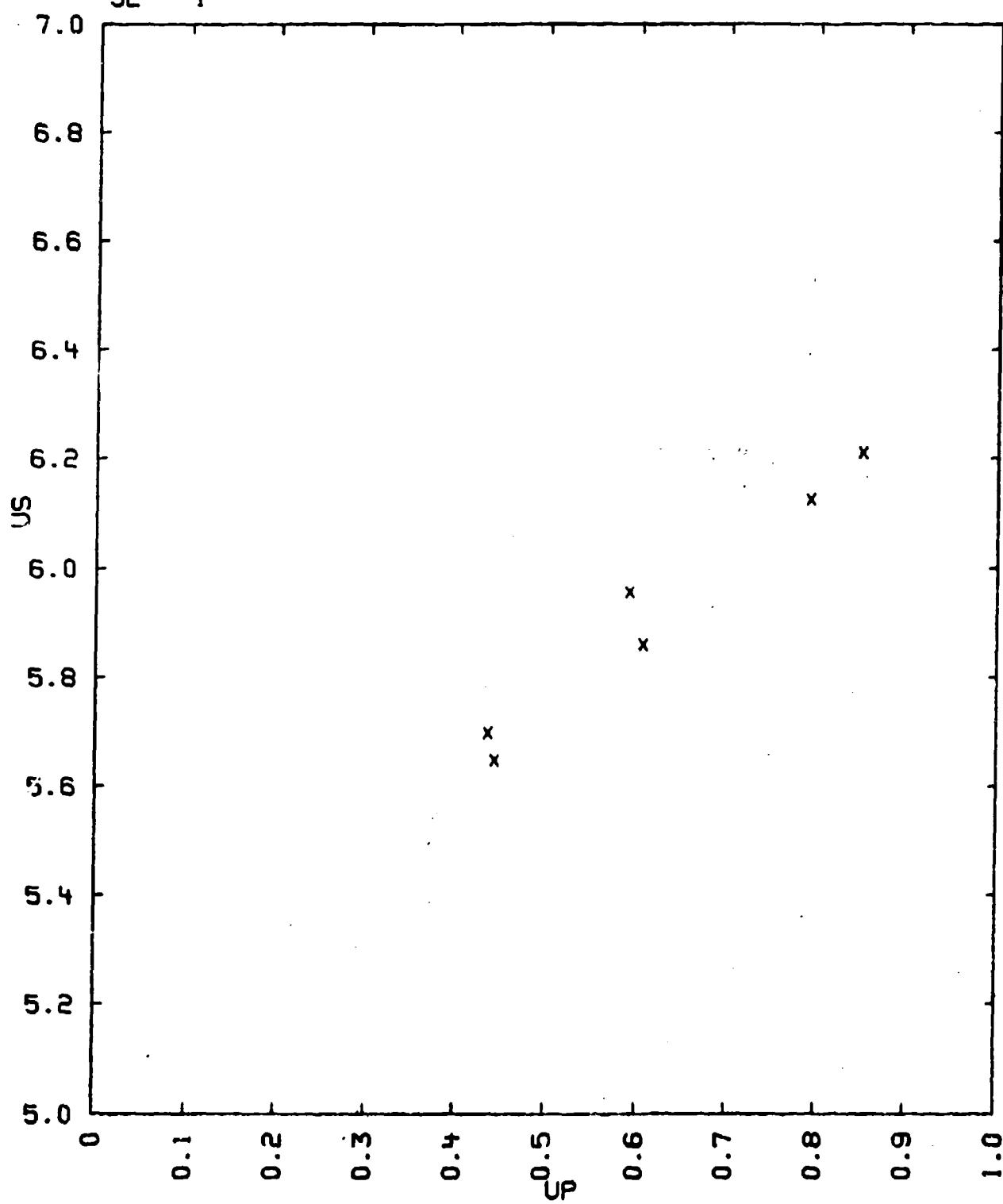
$$US = 5.120 + 1.285 UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.08 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS.

TABLE I

MOLYBDENUM
52---1



52---2
MOLYBDENUM

MO

$V_0 = 0.09804 \text{ CC/G.}$
 $V_{01} = 0.09804 \text{ CC/G.}$

$C_B = 5.19 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
10.20	7.29	1.69	1256	0.768	1064
-	7.20	1.70	1245	0.764	1064
-	7.29	1.68	1250	0.770	1058
-	7.65	2.06	1604	0.731	1380
-	7.71	2.06	1618	0.733	1387
-	7.75	2.07	1633	0.733	1399

$US = 5.157 + 1.238 UP \text{ KM/SEC.}$ SIGMA US = 0.41 PERCENT

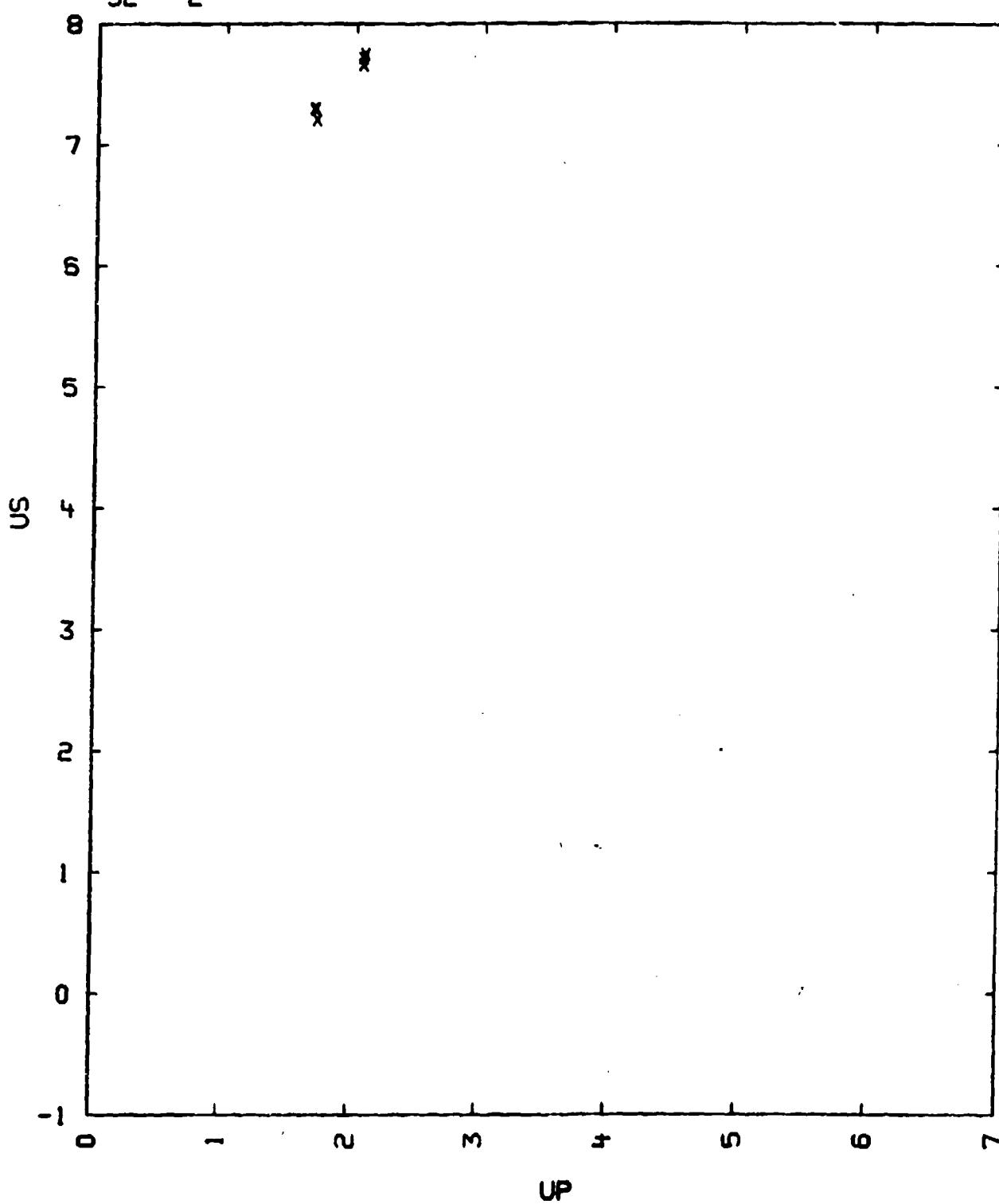
COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE &
DATA REDUCTION TECHNIQUE &
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 4) V_{01} WAS TAKEN FROM THE AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO. N.Y., 1963) 2ND ED.

TABLE I

MOLYBDENUM

52---2



52---3
MOLYBDENUM

MO

$$V_0 = 0.0980 \text{ CC/G}$$

$$V_{01} = 0.09804 \text{ CC/G.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL USED AS STRIKER PLATE. U(ST) IS THE VELOCITY BEFORE IMPACT.

TABLE

RHO0	US	UP	P	V/V0	ST	U(ST)
10.2	7.21	1.60	1176	0.7782	AL	5.600
-	8.39	2.52	2154	0.6998	FE	5.600
-	10.16	3.92	4060	0.5903	FE	8.540
-	10.35	4.24	4477	0.6143	FE	9.100

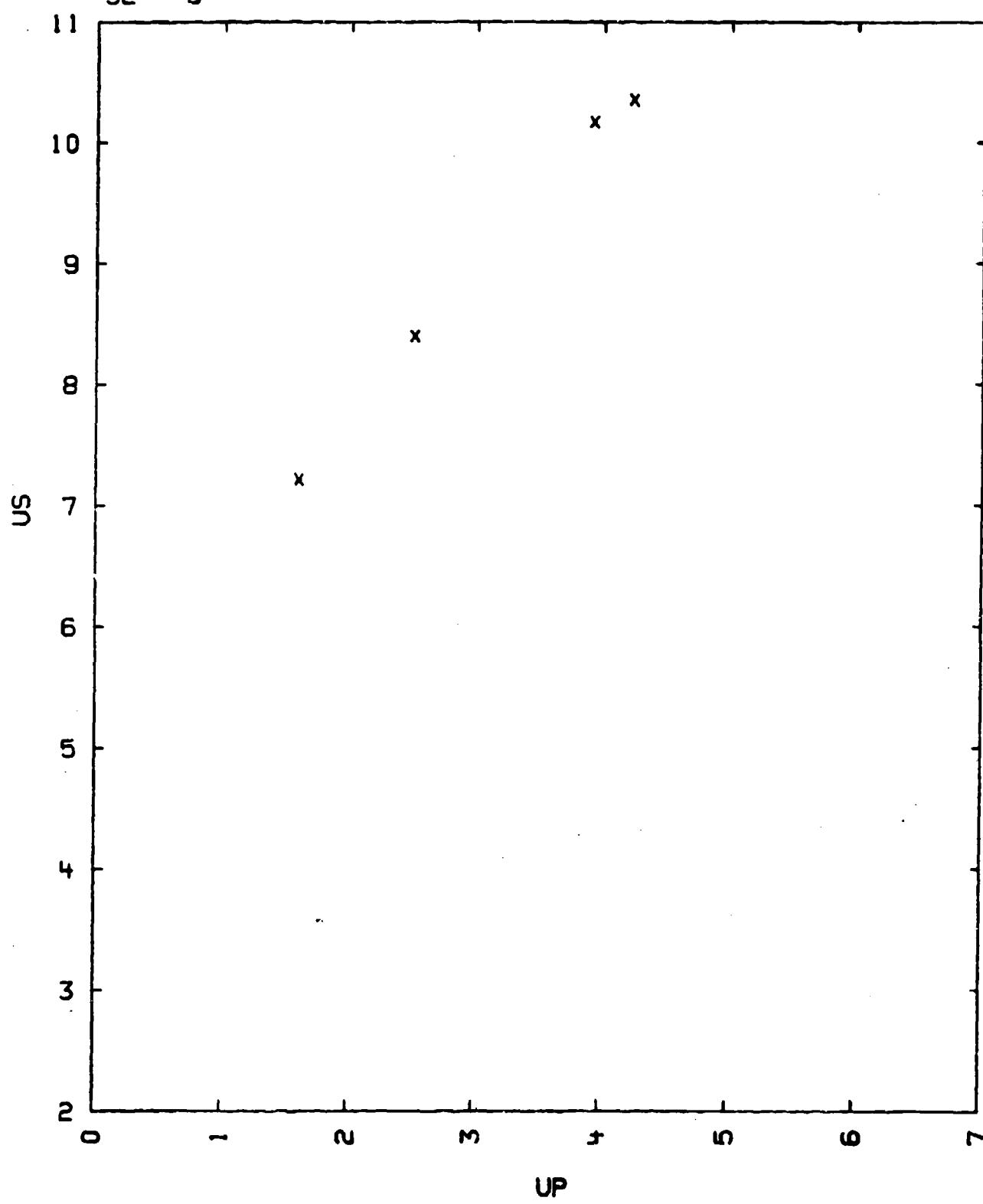
US = $5.30 + 1.21 \text{ UP KM/SEC.}$ FOR UP BETWEEN 1.60 TO 4.24 KM/SEC.
 SIGMA US = 0.10 KM/SEC.

COMMENTS:

- 1) SOURCE: KRUPNIKOV, K. K., BAKANOVA, A. A., BRAZNIK, M. I. AND TRUNIN, R. F.
 SOVIET PHYS.-JETP, VOL. 8, P. 205 (1963)
- 2) EXPERIMENTAL TECHNIQUE A
 DATA REDUCTION TECHNIQUE B
- 3) THE SAMPLES WERE POSITIONED ON PLATES OF AL OR FE AS INDICATED IN COLUMN 6 OF THE TABLE.
- 4) THE SPECIMENS WERE 3 TO 4 MM THICK AND 10 TO 15 MM IN DIAMETER.
- 5) THE PRESSURES WERE PRODUCED BY AN EXPLOSIVELY ACCELERATED ALUMINUM OR STEEL PLATE AS INDICATED IN COLUMN 6.
- 6) V01 WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK (MCGRAH-HILL BOOK CO. N.Y., 1963) 2ND ED.
- 7) ALSO LISTED ARE: RHO0 AT 0 DEG. K. = 10.24 G/CC
 ELECTRONIC HEAT CAPACITY COEFFICIENT
 225 ERG/G DEG⁻²
 GRUNEISEN CONSTANT = 1.56

TABLE I

MOLYBDENUM
52---3



52---4
MOLYBDENUM

MO

$V_0 = 0.0987 \text{ CC/G}$
 $V_{01} = 0.09752 \text{ CC/G}$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC,
AND PRESSURE IN KILOBARS.

TABLE

SAMPLE						BASIC PLATE
RHO0	US	UFS	UP	P	V/V0	PRESSURE
10.13	5.85		0.79	469	0.865	289
10.13	5.39	0.935	0.50	270	0.908	166

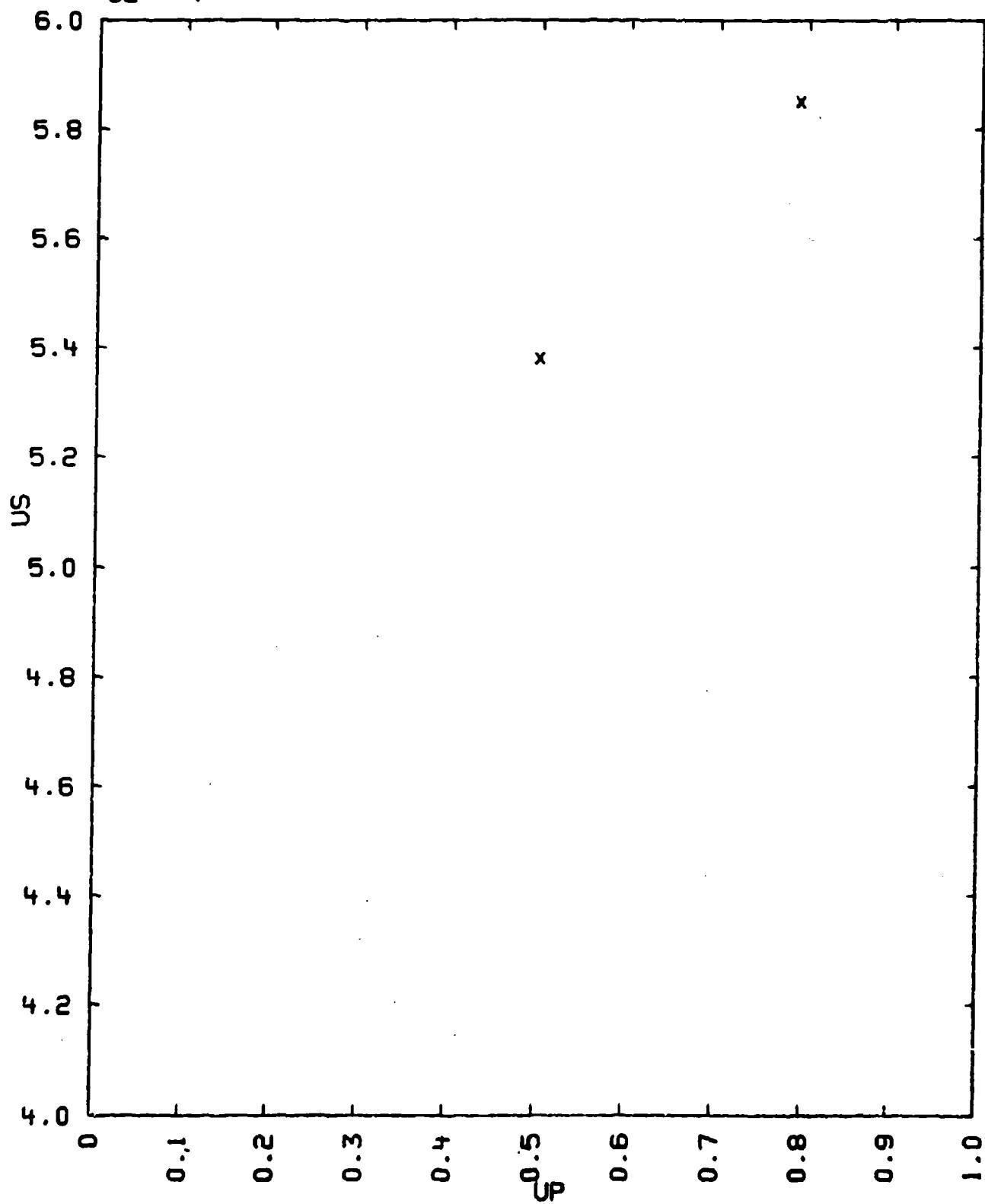
US =

COMMENTS:

- 1) SOURCE: COMPILER
L.R.L. EQUATION OF STATE FILE
LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B. (ALUMINUM STANDARD BASE PLATE)
DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF V_{01} WAS OBTAINED FROM A LATTICE CONSTANT OF 3.14737 A
A.C.A MONOGRAPH NO. 5, CRYSTAL DATA DETERMINATIVE TABLES
(AMERICAN CRYSTALLOGRAPHIC ASSOCIATION, 1963) 2ND EDITION.

TABLE I

MOLYBDENUM
52---4



52---5
MOLYBDENUM

MO

V0 = 0.979 CC/G
V01 = 0.9785 CC/GCL = 6.45 KM/SEC
CS = 3.47 -
CO = 5.04 KM/SEC
CB = 5.19 -IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
10.200	5.19	0.	0.	1.0000	CB	0.
10.200	5.04	0.	0.	1.0000	CO	0.
10.200	5.63	0.42	241.	0.9254	2024 AL	6.44
10.200	5.58	0.43	245.	0.9229	2024 AL	6.45
10.220	5.66	0.45	260.	0.9205	CU	4.72
10.210	5.67	0.46	266.	0.9189	CU	4.74
10.200	5.88	0.57	342.	0.9031	2024 AL	6.81
10.200	5.79	0.59	348.	0.8981	2024 AL	6.83
10.210	5.88	0.59	354.	0.8997	CU	4.96
10.200	6.05	0.77	475.	0.8727	2024 AL	7.25
10.200	6.14	0.83	520.	0.8648	2024 AL	7.38
10.210	6.22	0.86	546.	0.8617	CU	5.39
10.220	6.23	0.87	554.	0.8604	CU	5.41
10.210	6.23	0.87	553.	0.8604	CU	5.41
10.210	6.24	0.89	567.	0.8574	CU	5.44
10.210	6.70	1.26	862.	0.8119	CU	6.04
10.210	6.90	1.48	1043.	0.7855	CU	6.36
10.210	7.05	1.56	1123.	0.7787	CU	6.49
10.200	7.20	1.63	1197.	0.7736	BRASS	6.45
10.200	7.21	1.64	1206.	0.7725	BRASS	6.46
10.200	7.11	1.65	1197.	0.7679	BRASS	6.46
10.210	7.25	1.69	1251.	0.7689	CU	6.70
10.210	7.28	1.73	1301.	0.7596	CU	6.78
10.210	7.57	1.99	1538.	0.7371	CU	7.15
10.200	7.56	2.00	1542.	0.7354	BRASS	7.00
10.200	7.62	2.00	1554.	0.7375	BRASS	7.01
10.200	7.66	2.01	1570.	0.7378	BRASS	7.02
10.210	8.01	2.39	1955.	0.7016	CU	7.75
10.210	8.05	2.44	2005.	0.6969	CU	7.82
10.210	8.22	2.49	2090.	0.6971	CU	7.91

US = 5.124 + 1.233*UP KM/SEC

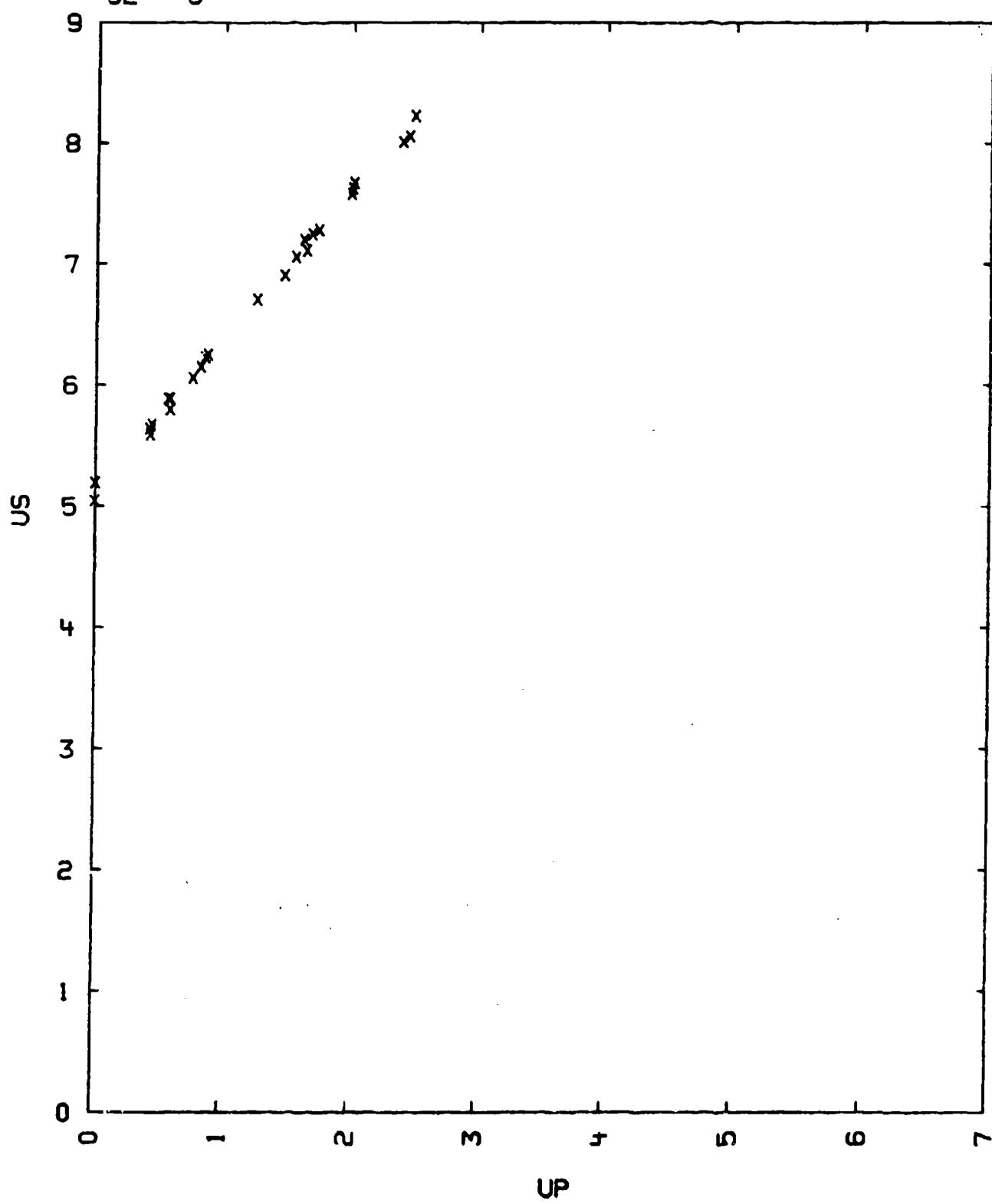
SIGMA US = 0.047 KM/SEC

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M.
AND CARTER, W.J.
HIGH VELOCITY IMPACT PHENOMENA, ED. KINSLOW (ACADEMIC PRESS
NEW YORK, 1970) CHAPTER VII; EOS OF SOLIDS FROM
SHOCK WAVE STUDIES
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B
- 3) VOI FROM WYCKOFF CRYSTAL STRUCTURES (JOHN WILEY AND SONS, N.Y.,
1963) VOL. 1.
- 4) $V(DP/DE) = 1.52$

TABLE I

MOLYBDENUM
52---5



53---1
TUNGSTEN

H

$$V_0 = 0.0521 \text{ CC/G.}$$

$$V_{01} = 0.0518 \text{ CC/G.}$$

$$C_D = 4.05 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN STANDARD
19.17	4.56	0.45	10	0.901	218
-	4.55	0.45	384	0.901	218
-	4.78	0.64	587	0.886	320
-	4.82	0.64	590	0.886	320
-	5.47	1.17	1225	0.786	853
-	5.49	1.17	1227	0.786	853
-	6.21	1.73	2061	0.721	1430
-	6.19	1.73	2054	0.721	1427
-	6.24	1.73	2074	0.723	1426

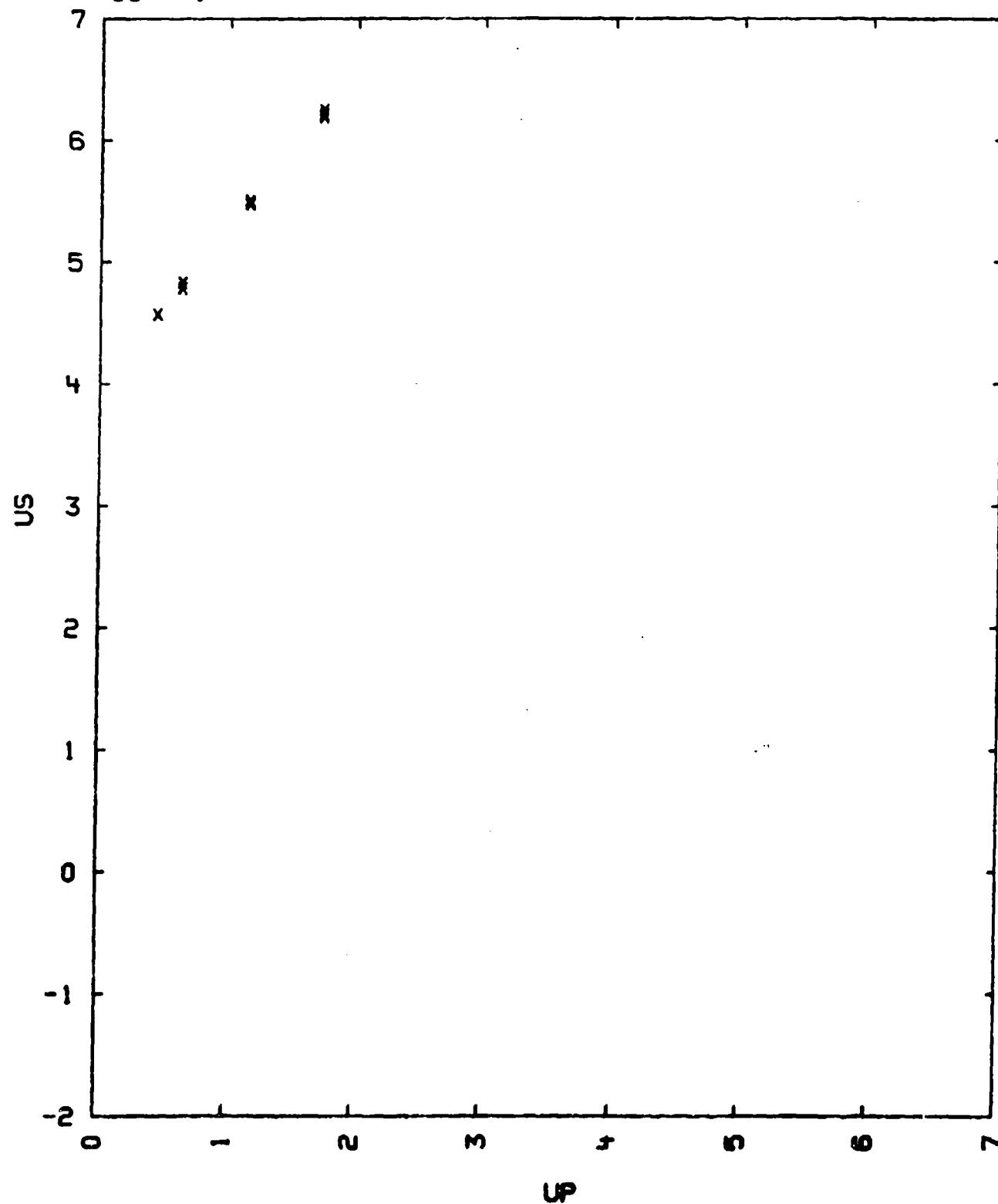
$$US = 4.005 + 1.268 \cdot UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.32 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-5, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE SECTION 36-33--2
- 4) FOR P(STANDARD)=320 AND 218 THE STANDARD WAS AL. ABOVE THAT BRASS

TABLE I

TUNGSTEN
53---1



53---2
TUNGSTEN POROUS

W 99.8 PERCENT BY WEIGHT

MO TRACE

FE -

P -

CU -

O -

$$V_0 = 0.0532 - 0.222 \text{ CC/G}$$

$$V_{01} = 0.05182 \text{ CC/G}$$

IN THE TABLES BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURES IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL.
P(ST) IS THE PRESSURE IN THE STANDARD PLATE.

TABLE

RHO0	US	UP	P	V/V0	ST	P(ST)
18.79	8.11	3.26	5010	0.5977	FE	3585
11.38	8.02	4.12	3760	0.4859	FE	3585
9.17	7.80	4.53	3240	0.4193	FE	3585
6.47	7.95	5.09	2620	0.3604	FE	3585
4.50	8.11	3.03	2030	0.3064	FE	3585
18.79	6.01	1.74	1980	0.7112	FE	1390
10.99	5.01	2.39	1320	0.5230	FE	1390
9.00	4.96	2.60	1160	0.4757	FE	1390
6.54	4.78	2.93	910	0.3880	FE	1390
10.99	2.56	1.12	315	0.5624	AL	258
6.32	2.29	1.51	218	0.3406	AL	258
4.85	2.29	1.67	185	0.2684	AL	258

US =

COMMENTS:

- 1) SOURCE: KRUPNIKOV, K. K., BRAZNIK, M. I. AND KRUPNIKOVA V. P.
SOVIET PHYS.-ETP, VOL. 15, P. 470 (1962)
J. EXPTL. THEORET. PHYS. (U.S.S.R.) VOL. 47, P. 1202 (1964)
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
- 3) CO, THE VELOCITY OF THE RAREFACTION WAVE MOVING PARALLEL TO THE SHOCK FRONT, WAS MEASURED ALSO. THESE RESULTS ARE GIVEN BELOW WITH THE MEASURED SHOCK VELOCITY AND PRESSURE.

TABLE

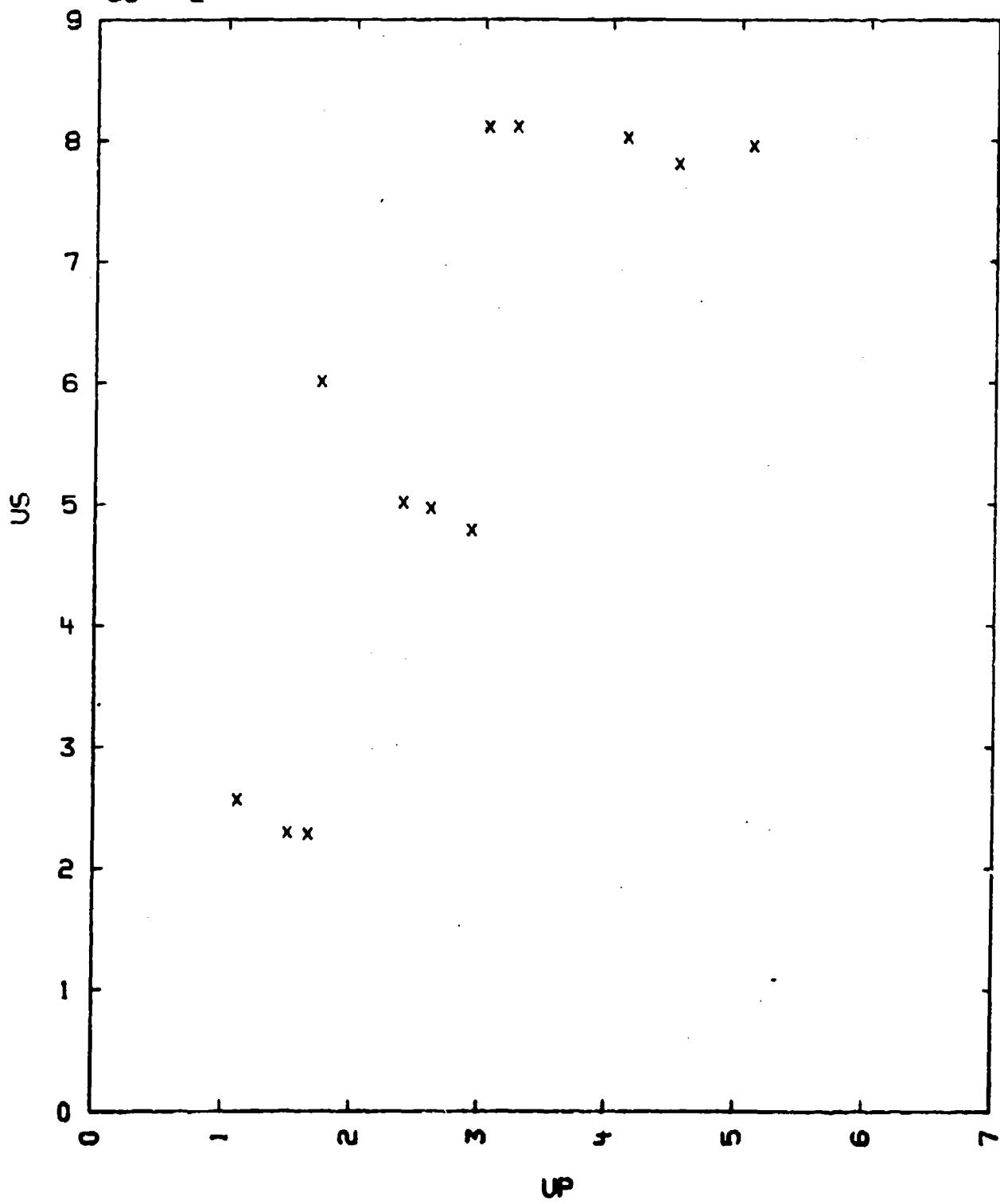
RHO0	US	P	CO	ST	P(ST)
10.79	7.64	3860	7.19	FE	3860
	5.04	1310	5.61	FE	1310

2.52 310 6.1 AL 258

- 4) THE MEASURED VALUE FOR CO AT THE LOWEST PRESSURE IS 6.1 KM/SEC. THIS IS REPRESENTATIVE OF THE VELOCITY OF THE ELASTIC WAVE. THE CALCULATED PLASTIC WAVE VELOCITY $C_D = 4.79$ KM/SEC.
- 5) THE EXPERIMENTAL TECHNIQUE USED IN THE DETERMINATION OF C_D WAS DESCRIBED BY AL'TSHULER, KORMER, BRAZNIK, VLADIMIROV, SPERANSKAYA, AND FUNTIKOV, SOVIET PHYS.-JETP, VOL. 38, P. 1061 (1950)
- 6) VOI WAS OBTAINED FROM THE CRYSTAL DATA DETERMINATIVE TABLES, AMERICAN CRYSTALLOGRAPHIC ASSOCIATION, 2ND. ED. (1963)

TABLE I

TUNGSTEN POROUS
53---2



53---3
TUNGSTEN

W 99.95 PERCENT
FE 0.005 -
C 0.005 -
O2 0.005 -
H2 0.001 -
N2 0.005 -

$$V_0 = 0.05216 \text{ CC/G}$$

$$V_{01} = 0.05192 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN G/CC, VELOCITIES IN KM/SEC., AND PRESSURE IN KILOBARS.

TABLE

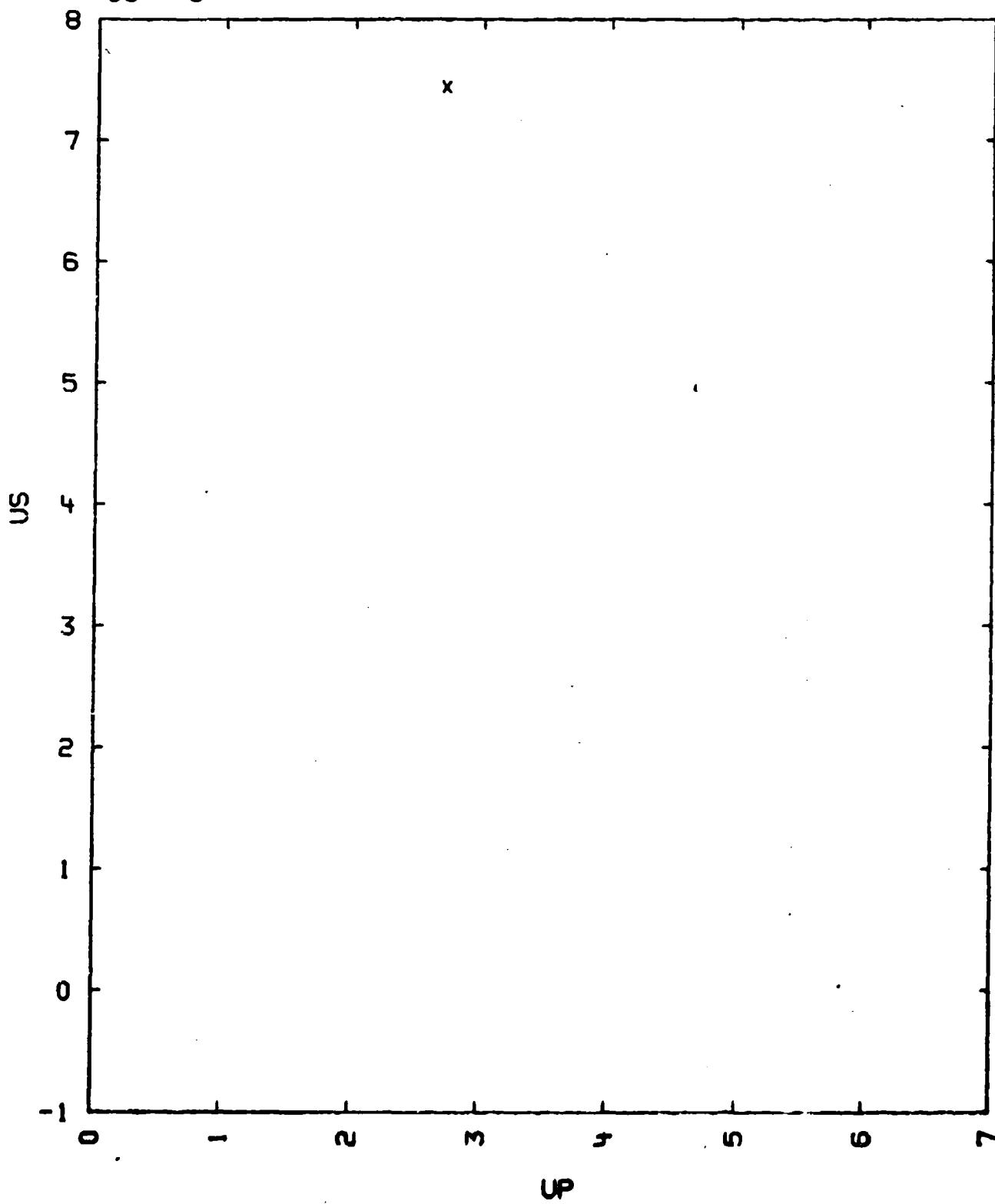
RHO0	US	UP	P	V/V0
19.17	7.44	2.71	3960	0.6358
US	-			

COMMENTS:

- 1) SOURCE: JONES, A. H., ISBELL, W. M. AND MAIDEN C. J.
JOURNAL OF APPLIED PHYSICS, VOL. 37, P. 3493 (1966).
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE A
STANDARD MATERIAL FANSTEEL 77, SEE MATERIAL 53--39--36--1.
- 3) THE DRIVER PLATE WAS ACCELERATED BY MEANS OF A LIGHT GAS GUN.
- 4) THE DRIVER PLATE VELOCITY AND SPATIAL ORIENTATION WERE MEASURED WITH
TWO PULSED X-RAY STATIONS JUST BEFORE IMPACT.
- 5) FOR BOTH THE DRIVER PLATE IMPACT VELOCITY AND THE SAMPLE SHOCK
VELOCITY THE MEASURED EXPERIMENTAL ERROR IS WITHIN 1 PERCENT.
- 6) V01 WAS TAKEN FROM THE CRYSTAL DATA DETERMINATIVE TABLES, AMERICAN
CRYSTALLOGRAPHIC ASSOCIATION, 2ND. ED. (1963).

TABLE I

TUNGSTEN
53---3



53---4
TUNGSTEN

	FOAM	COM. ACTED	
NI FE MO	0.09	0.01	MOLE PERCENT
C	0.01	-	-
O	0.37	0.0004	-
W	REST	REST	
PARTICLE SIZE	5. TO 20. MICRONS		

$$V_0 = 0.05208 - 0.0709 \text{ CC/G}$$

$$V_{01} = 0.05192 \text{ CC/G}$$

THE TABLE LISTS DENSITY IN G/CC., VELOCITIES IN KM/SEC AND PRESSURES IN KBAR. SEE COMMENT 6

TABLE

RHO0	US1	UP1	P1	V1/V0	US2	UP2	P2	V2/V0
19.2	4.84	0.039	36.0	0.992	4.41	0.338	289.5	0.923
-	4.77	0.039	36.1	0.992	4.15	0.121	101.0	0.971
-	4.87	0.039	36.3	0.992	4.24	0.129	109.8	0.971
-	4.91	0.037	35.2	0.992	4.25	0.213	178.9	0.950
-	4.83	0.038	35.5	0.992	4.27	0.077	67.4	0.982
14.3	4.10	0.019	11.2	0.995	1.68	0.421	107.2	0.753
14.2	3.74	0.018	9.8	0.995	1.88	0.506	139.3	0.734
14.2	3.51	0.020	9.8	0.994	1.63	0.419	101.9	0.746
14.4	3.71	0.019	10.0	0.995	1.15	0.153	31.9	0.877
14.1	3.73	0.018	9.6	0.995	1.33	0.323	66.0	0.765
14.5	3.61	0.017	8.8	0.995	1.51	0.354	82.2	0.770
14.2	3.74	0.016	8.3	0.996	1.93	0.505	141.9	0.740

$$US = 0.732 + 2.23 \cdot UP \text{ KM/SEC.}$$

$$\text{SIG US} = 0.07 \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: SCHMIDT, D.N. AND LINDE, R.K.
AIR FORCE WEAPONS LAB. REP. AFWL-TR-68-33 (1968)
CONTRACT NO. AF 29(601)-7238
SRI PROJECT: POU-6127
STANFORD RES. INST., MENLO PARK, CALIFORNIA, U. S. A.
SEE ALSO J APPL. PHYS. 43 3387 (1972) FOR A DISCUSSION ON
TUNGSTEN, COPPER AND POLYURETHANE.
- 2) EXPERIMENTAL TECHNIQUE: 1)
DATA REDUCTION TECHNIQUE: C. WAVE PROFILES WERE BROAD BUT WERE
APPROXIMATED BY A STEP FUNCTION MOVING AT
THE AVERAGE WAVE VELOCITY LISTED
- 3) THE FOOT OF THE ELASTIC WAVE IS BETWEEN 1 AND 10 PERCENT FASTER THAN
THE AVERAGE VELOCITY LISTED. THE PROFILE AND AVERAGE VELOCITY OF THE
PLASTIC (CRUSHING) WAVE WAS INFERRED FROM THE STRESS-TIME PROFILE AT
THE GAUGE-SAMPLE INTERFACE AND THE REVERBERATION TIME OF AN ELASTIC
RELEASE WAVE REFLECTING OFF THE CRUSHING WAVE.

- 4) ALL SAMPLES WERE COMPACTED AND SINTERED AT 2000 DEG. F.
THE HIGH DENSITY SAMPLES WERE HOT ROLLED IN ADDITION.
- 5) VOI WAS TAKEN FROM CRYSTAL DATA DETERMINATIVE TABLES, AMERICAN CRYSTALLOGRAPHIC ASSOCIATION, 2ND. ED. (1963)
- 6) NOTE THAT THE SECOND WAVE PARAMETERS IN THE TABLE REPRESENT A CRUSHING CURVE. P IS AN EFFECTIVE LONGITUDINAL STRESS OBTAINED BY CONSIDERING THE SAMPLE AS A HOMOGENEOUS ISOTROPIC MEDIUM.
- 7) THE ELASTIC WAVE PARAMETERS OF THE FIRST TABLE ENTRY WERE ASSUMED.

TABLE I

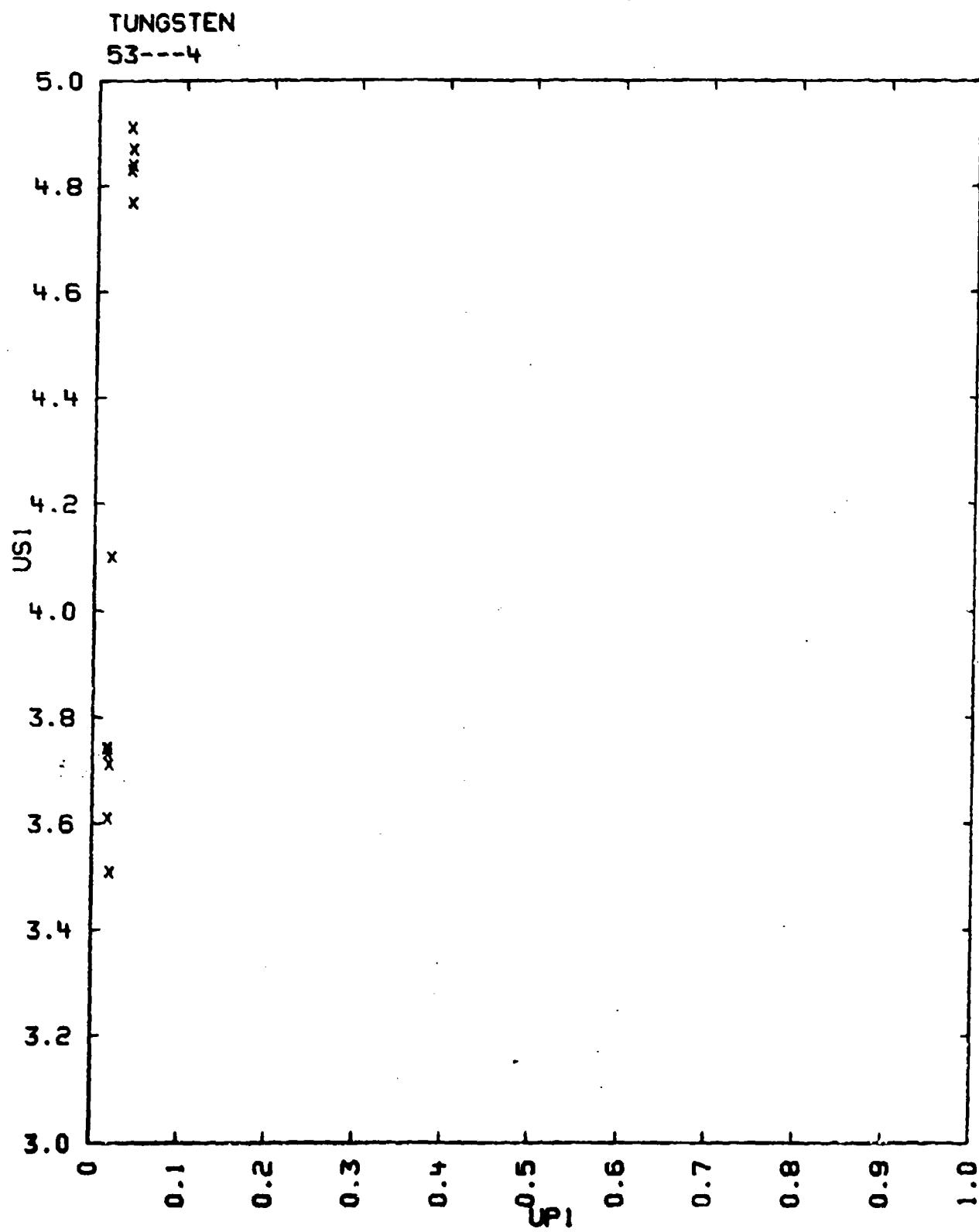
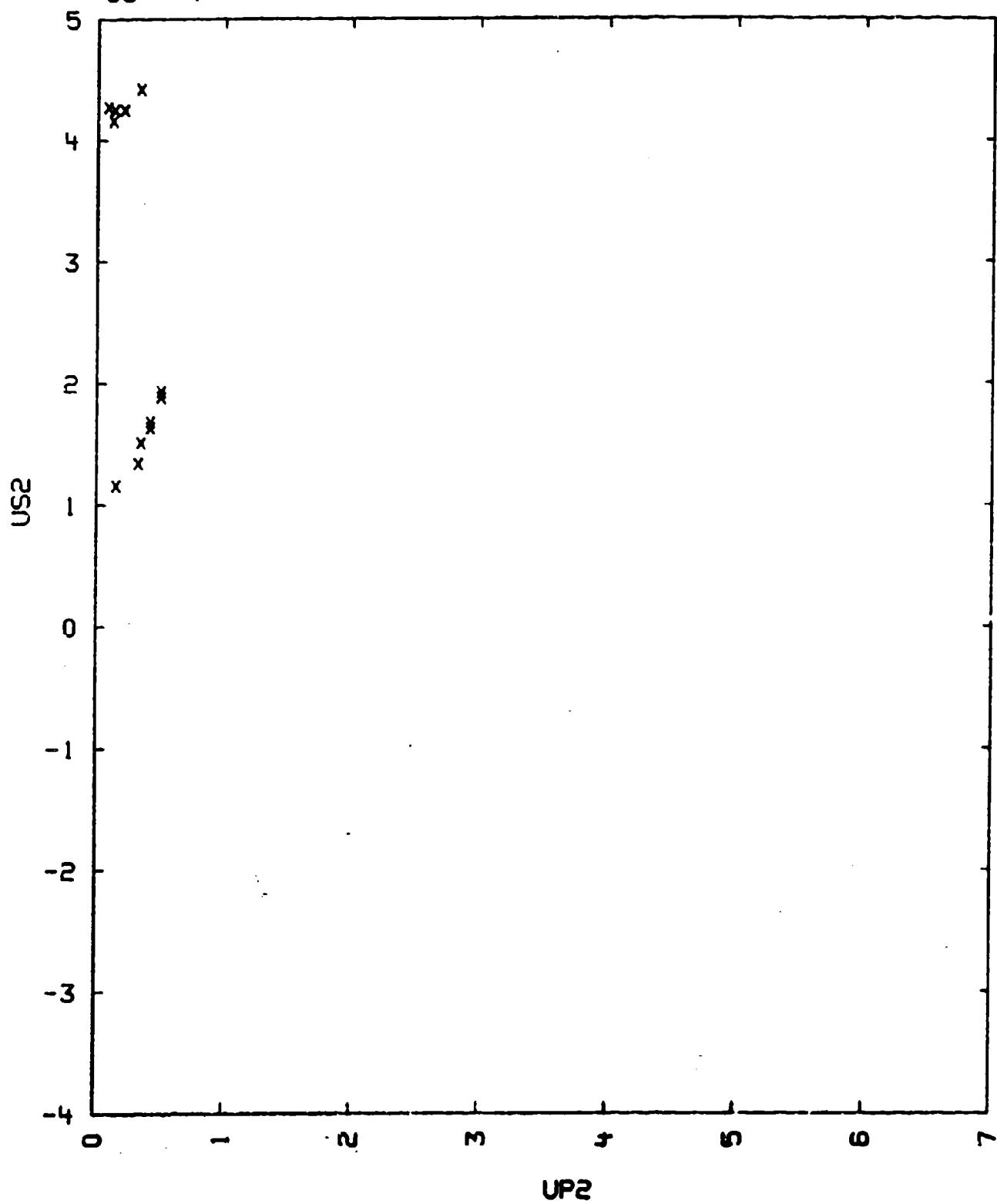


TABLE I

TUNGSTEN
53---4



53---5
TUNGSTEN

W

$$V_0 = 0.0521 \text{ CC/G} \quad C_L = 5.22 \text{ KM/SEC} \quad C_D = 4.02 \text{ KM/SEC}$$

$$V_{01} = 0.05192 \text{ CC/G} \quad C_S = 2.89 \text{ KM/SEC} \quad C_B = 4.05 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
19.250	4.43	0.32	273.	0.9278	CU	4.64
19.250	4.45	0.35	308.	0.9191	CU	4.72
19.250	4.52	0.36	313.	0.9204	CU	4.74
19.270	4.52	0.37	322.	0.9181	CU	4.75
19.170	4.51	0.44	380.	0.9024	2024 AL	6.80
19.170	4.50	0.44	380.	0.9022	2024 AL	6.80
19.250	4.65	0.47	421.	0.8989	CU	4.96
19.170	4.77	0.62	567.	0.8700	2024 AL	7.34
19.170	4.73	0.62	562.	0.8689	2024 AL	7.34
19.256	4.92	0.69	654.	0.8598	CU	5.39
19.250	4.92	0.70	663.	0.8577	CU	5.41
19.250	4.92	0.70	663.	0.8577	CU	5.41
19.250	4.95	0.72	686.	0.8545	CU	5.44
19.170	5.42	1.13	1174.	0.7915	BRASS	6.07
19.170	5.40	1.13	1170.	0.7907	BRASS	6.07
19.250	5.80	1.38	1541.	0.7621	CU	6.70
19.250	5.84	1.43	1608.	0.7551	CU	6.78
19.250	6.05	1.64	1910.	0.7289	CU	7.15
19.250	6.05	1.64	1910.	0.7289	CU	7.15
19.170	6.12	1.68	1971.	0.7255	BRASS	7.08
19.170	6.14	1.69	1989.	0.7248	BRASS	7.09
19.170	6.17	1.69	1999.	0.7261	BRASS	7.10
19.250	6.42	1.98	2447.	0.6916	CU	7.75
19.250	6.45	2.02	2508.	0.6868	CU	7.82
19.250	6.56	2.06	2601.	0.6860	CU	7.91
19.250	6.65	2.07	2650.	0.6887	CU	7.94

$$US = 4.029 + 1.237 \cdot UP \text{ KM/SEC}$$

$$\Sigma US = 0.044 \text{ KM/SEC}$$

COMMENTS:

1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M.,

AND CARTER, W.J.

THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC

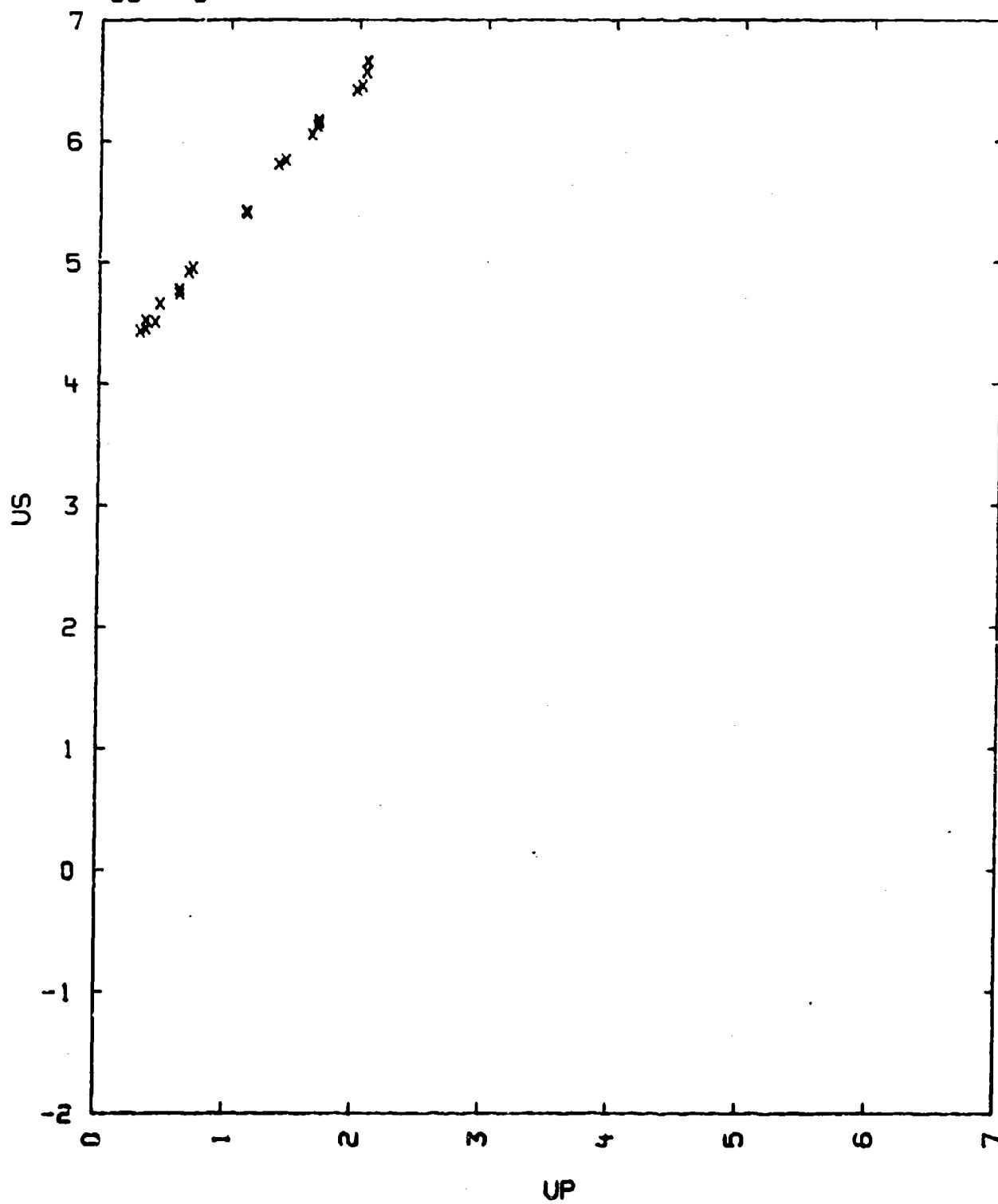
PRESS, NEW YORK, 1970) CHAPTER VII

- 2) EXPERIMENTAL TECHNIQUE: B
- DATA REDUCTION TECHNIQUE: B (STANDARD BASEPLATE AS SHOWN)
- 3) THE H.E.L. = 25-32 KBAR FOR A 5 MM THICK SAMPLE.
- 4) SAMPLES WERE ARC MELTED.
- 5) VOL FROM WYCKOFF, CRYSTAL STRUCTURES (JOHN WILEY AND SONS, N.Y.
1963) VOL. I

TABLE I

TUNGSTEN

53---5



54---1
VANADIUM

V

$$V_0 = 0.164 \text{ CC/G}$$

$$V_{01} = 0.1660 \text{ CC/G}$$

$$C_0 = 5.18 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.
PRESSURES IN KILOBARS AND DENSITY IN G/CC

TABLE

SAMPLE					STANDARD		
RHO0	US	UP	P	V/Y0	P	MTRL.	
6.10	5.78	0.58	204	0.900	149	AL	
-	5.73	0.58	203	0.898	149	-	
-	6.16	0.80	301	0.870	218	-	
-	6.07	0.81	301	0.866	218	-	
-	6.08	0.81	302	0.866	218	-	
-	6.05	0.82	301	0.865	220	-	
-	6.08	0.81	301	0.866	220	-	
-	6.49	1.12	441	0.828	320	-	
-	6.50	1.11	441	0.829	320	-	
-	6.46	1.12	441	0.827	320	-	
-	7.29	1.86	825	0.746	914	BRASS	
-	7.28	1.86	825	0.745	914	-	
-	7.32	1.85	828	0.747	914	-	
-	7.34	1.85	829	0.748	914	-	
-	8.20	2.59	1244	0.697	1378	-	
-	8.17	2.49	1241	0.695	1378	-	

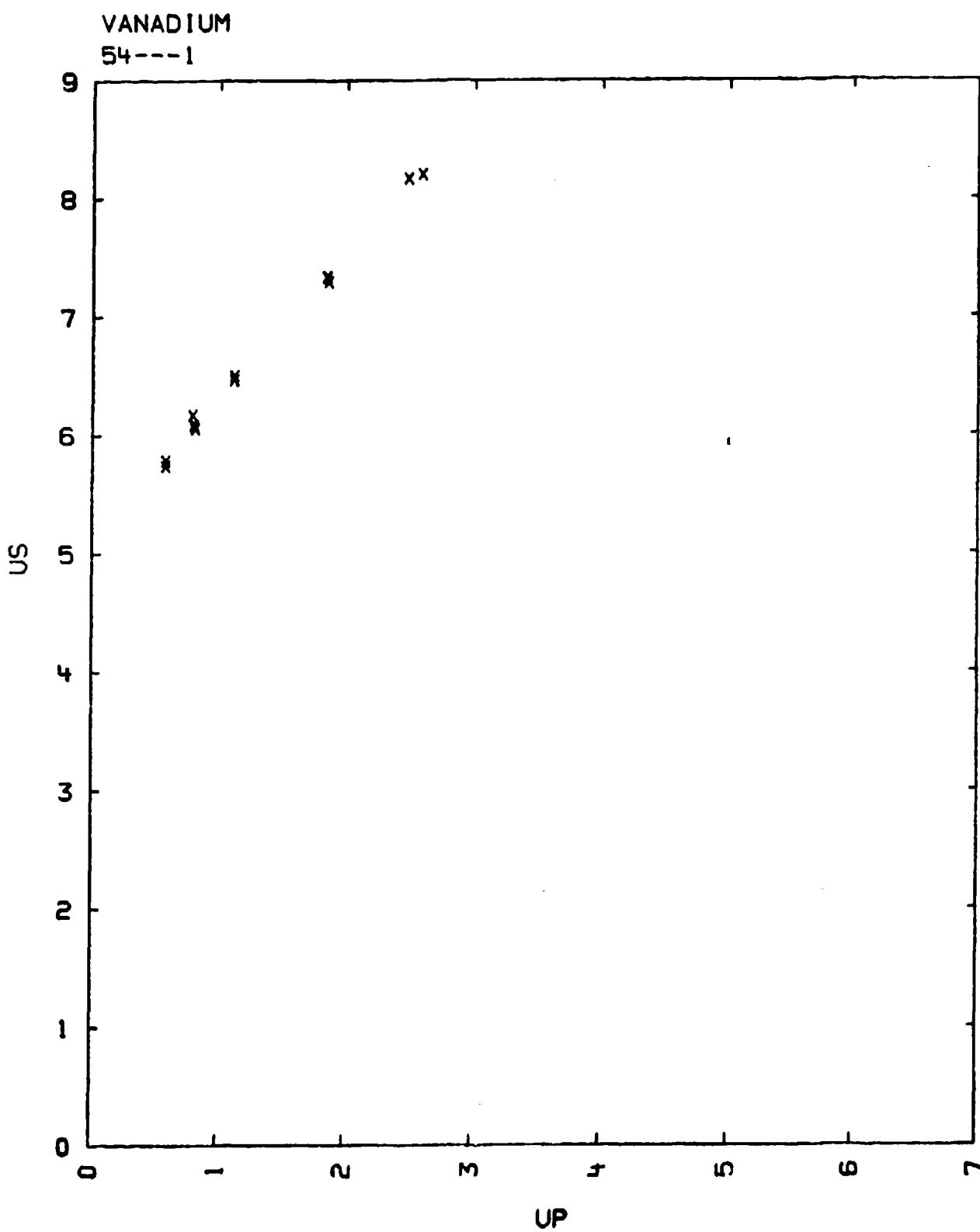
$$U_S = 5.108 + 1.210 \text{ UP KM/SEC.}$$

SIGMA US = 0.49 PER CENT

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31: P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, OMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 4) V01 WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW HILL BOOK CO.)
2ND ED.

TABLE I



54---2
VANADIUM

V

$\nu_0 = 0.154 \text{ CC/G}$
 $\nu_{01} = 0.16346 \text{ CC/G}$

$c_0 = 5.14 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

----- SAMPLE -----					----- STANDARD -----		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
6.08	11.33	4.92	3389.	0.566	IRON	7.85	11.21
-	8.94	3.37	1832.	0.623	IRON	7.85	8.83
-	7.53	2.12	970.	0.718	ALUMINUM	2.71	9.17
-	6.42	1.08	422.	0.832	ALUMINUM	2.71	7.34
-	5.86	0.46	164.	0.922	ALUMINUM	2.71	6.21

$US = 5.34 + 1.031 \cdot UP$ FOR UP BELOW 3.14 KM/SEC

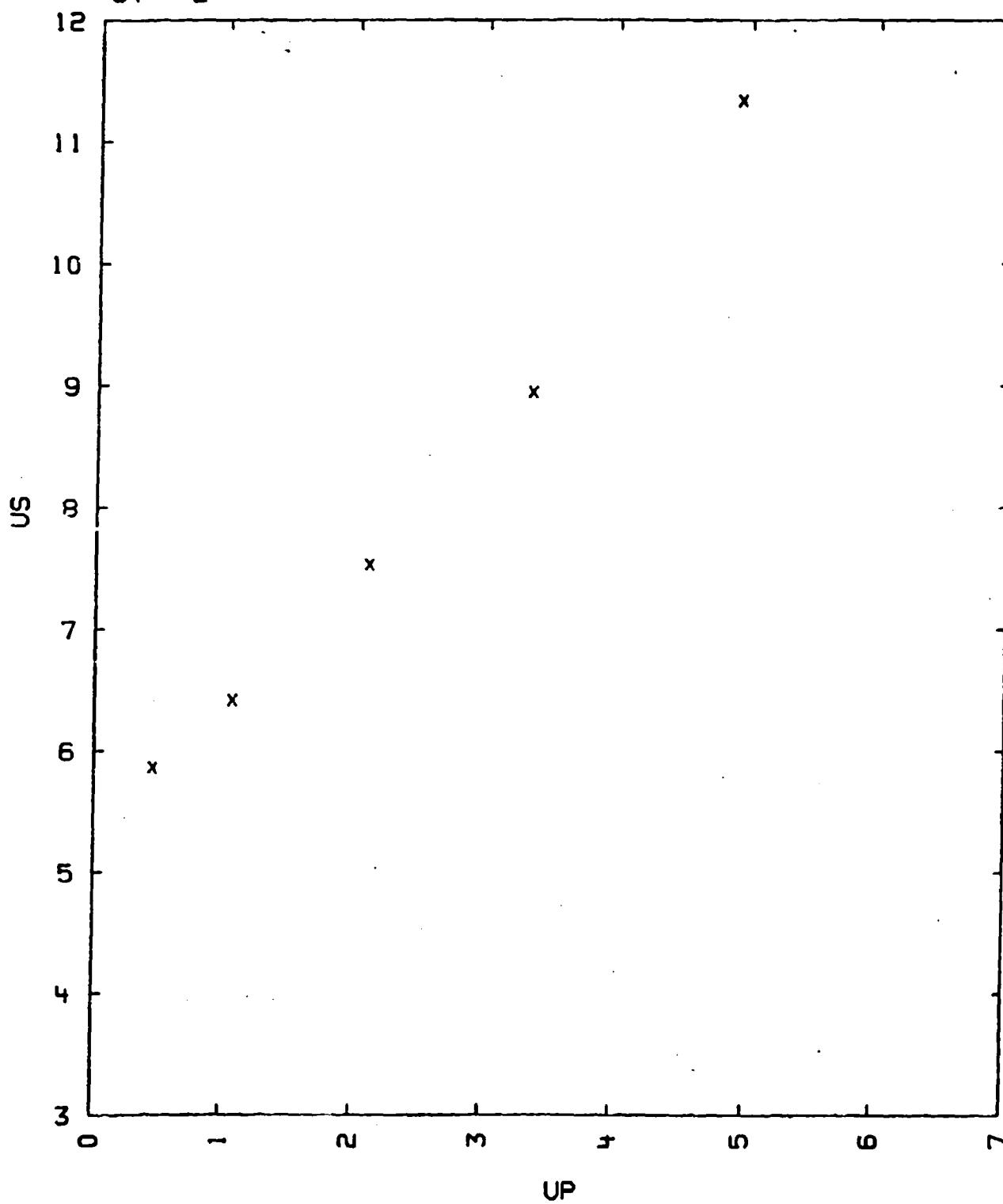
$US = 3.74 + 1.541 \cdot UP$ FOR UP BETWEEN 3.14 AND 4.92 KM/SEC

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P. ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES. THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE 1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM $US = 5.25 + 1.39 \cdot UP$ FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
IRON $US = 3.85 + 1.615 \cdot UP$ FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF ν_{01} WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANT OF THE CUBIC BODY-CENTERED STRUCTURE.
 $A = 3.0240 \text{ ANGSTROMS (25 DEG. C)}$
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 16.
- 6) THE AUTHORS OBTAINED THE VALUE OF c_0 FROM HANDBOOK OF RARE METALS. (MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT $UP = 3.14 \text{ KM/SEC}$ IS ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I

VANADIUM
54---2



54---3
VANADIUM

V

$$V_0 = 0.1639 \text{ CC/G}$$

$$V_{01} = 0.1635 \text{ CC/G}$$

$$CB = 5.18 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)	
6.100	5.71	0.56	195.	0.9019	2024 AL	6.39	
6.100	5.66	0.56	193.	0.9011	2024 AL	6.39	
6.100	6.08	0.78	289.	0.8717	2024 AL	6.81	
6.100	6.00	0.79	289.	0.8683	2024 AL	6.81	
6.100	6.00	0.79	289.	0.8683	2024 AL	6.81	
6.100	6.00	0.79	289.	0.8683	2024 AL	6.81	
6.100	5.98	0.79	288.	0.8679	2024 AL	6.81	
6.100	6.42	1.09	427.	0.8302	2024 AL	7.35	
6.100	6.41	1.09	426.	0.8300	2024 AL	7.35	
6.100	6.38	1.09	424.	0.8292	2024 AL	7.35	
6.100	7.25	1.80	796.	0.7517	BRASS	6.18	
6.100	7.23	1.80	794.	0.7510	BRASS	6.18	
6.100	7.20	1.80	791.	0.7500	BRASS	6.18	
6.100	7.19	1.80	789.	0.7497	BRASS	6.18	
6.100	8.07	2.43	1196.	0.6989	BRASS	7.01	
6.100	8.10	2.52	1245.	0.6889	BRASS	7.12	

$$US = 5.077 + 1.201 \cdot UP \text{ KM/SEC}$$

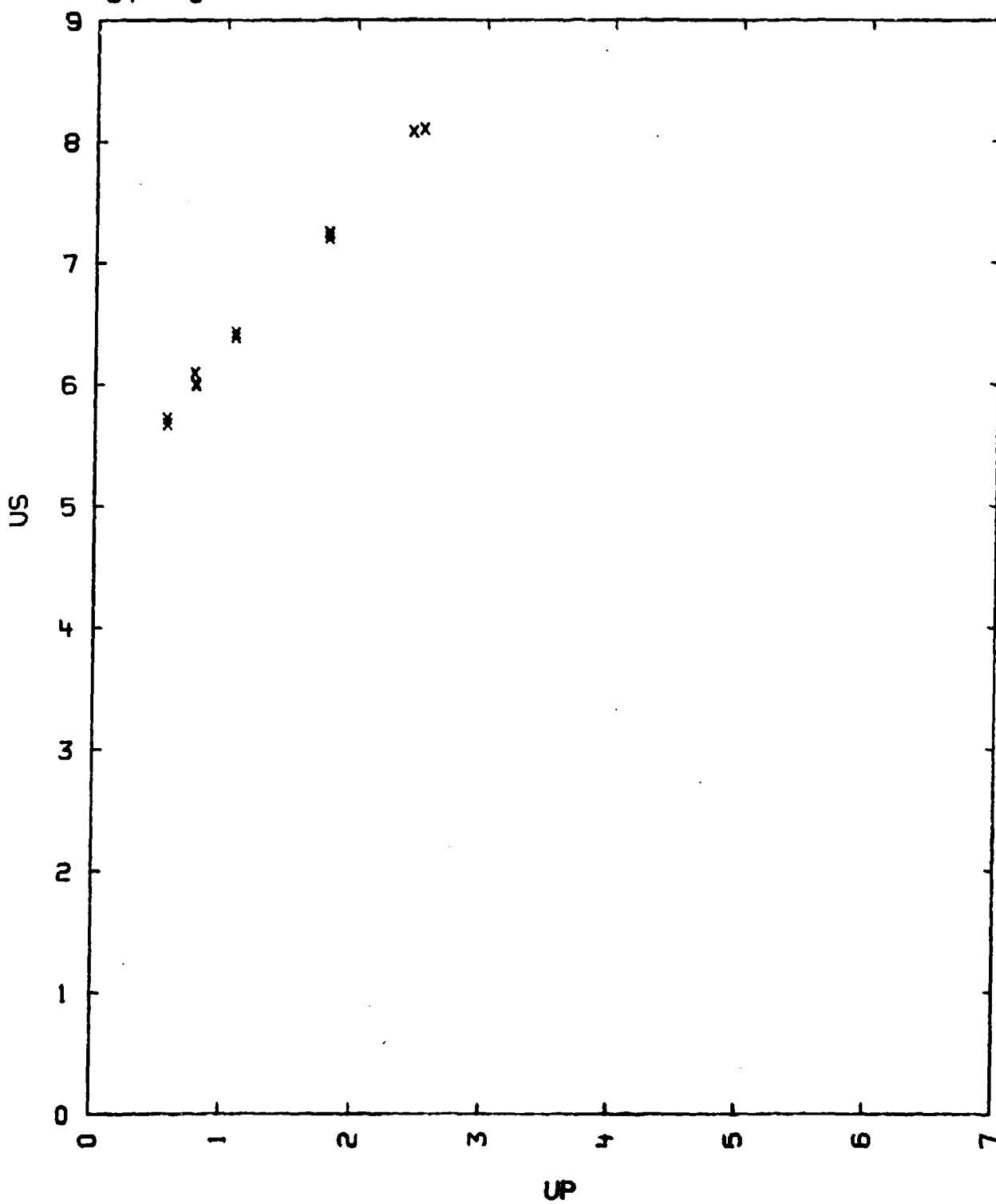
$$\text{SIGMA US} = 0.042 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES.
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B
- 3) V01 FROM WYCKOFF, CRYSTAL STRUCTURES (JOHN WILEY AND SONS, N.Y.,
1963) VOL. I
- 4) V(DP/DE) = 1.29

TABLE I

VANADIUM
54---3



55---1
NIOBIUM

NB---99.8 PER CENT OR GREATER

$$V_0 = 0.1162 \text{ CC/G.}$$

$$V_{01} = 0.1167 \text{ CC/G.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

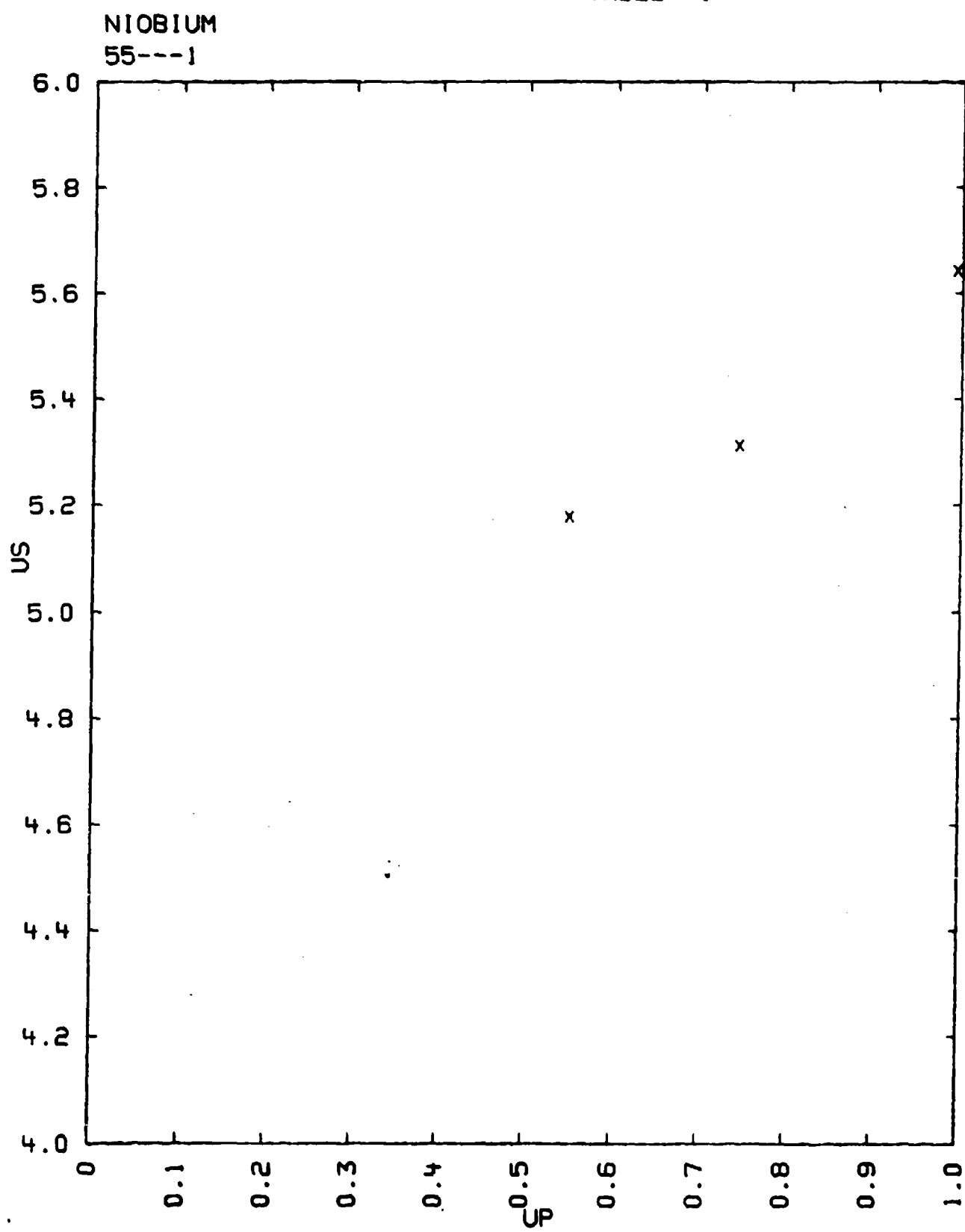
RHO0	US	UP	P	V/V0
8.604	5.177	0.5489	244.5	0.8940
-	5.311	0.7434	341	0.8606
-	5.642	0.9929	482	0.8240

$$US = 4.569 + 1.060 UP \text{ KM/SEC} \quad \text{SIGMA US} = 1.1 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PERCENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS.

TABLE I



55---2
NIOBIUM

NB 99.5 PERCENT OR GREATER

$V_0 = 0.1165 \text{ CC/G.}$ $C_L = 5.02 \text{ KM/SEC.}$ $C_0 = 4.39 \text{ KM/SEC.}$
 $V_{01} = 0.1162 \text{ CC/G.}$ $C_S = 2.11 \text{ KM/SEC.}$ $C_B = 4.43 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC.,
PRESSURE IN KILOBARS AND DENSITIES IN G/CC.

TABLE

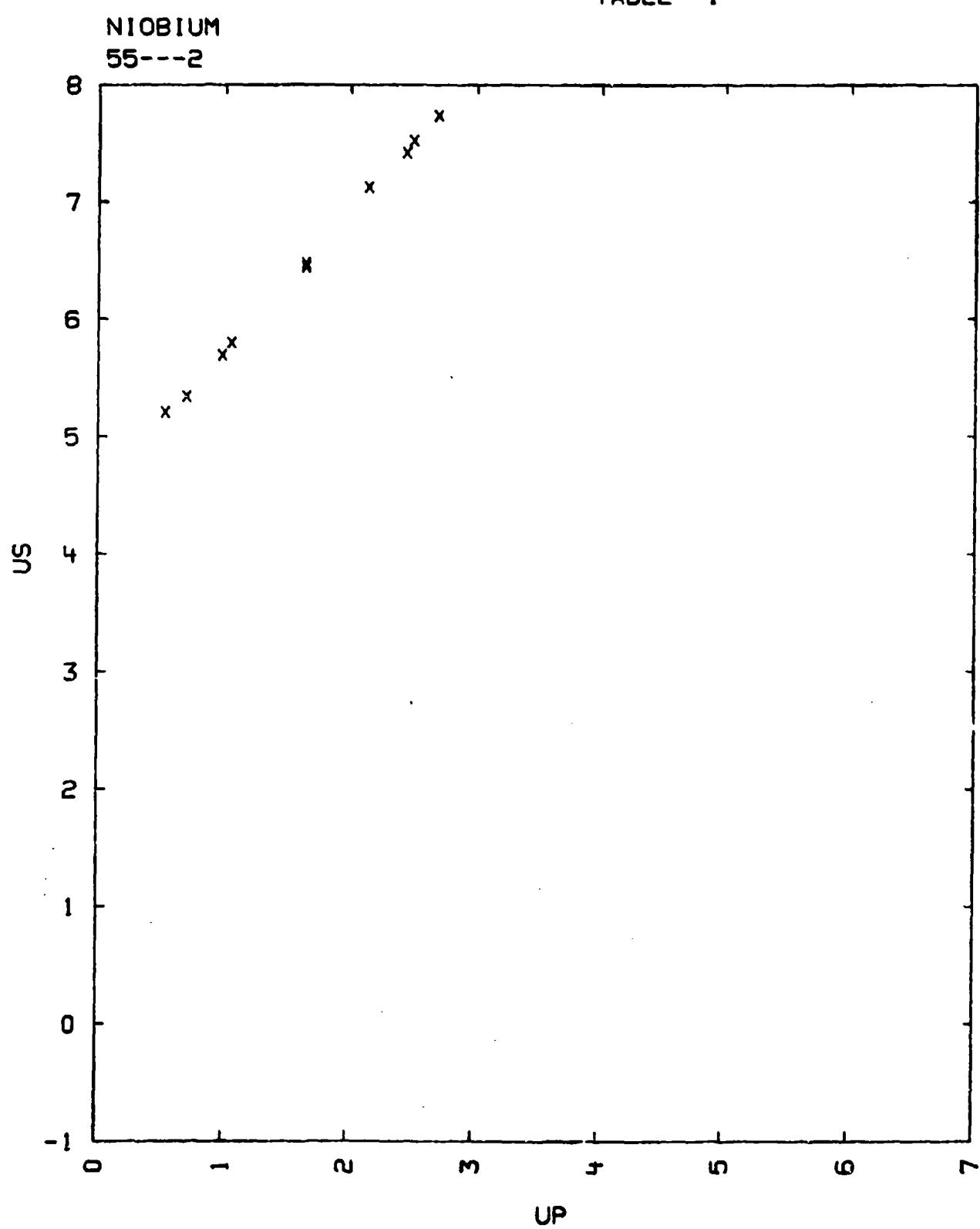
SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	RHO0	US
8.583	5.195	0.527	235.0	0.8986	8.932	4.764
-	5.335	0.706	323.4	0.8677	8.934	5.024
-	5.687	0.978	477.4	0.8280	-	5.426
-	5.793	1.061	527.5	0.8168	-	5.548
-	6.449	1.643	909.5	0.7452	-	6.387
-	6.432	1.650	910.6	0.7435	-	6.393
-	6.471	1.651	916.4	0.7449	-	6.400
-	7.114	2.158	1317.8	0.6966	-	7.133
-	7.417	2.450	1557.5	0.6697	-	7.494
-	7.518	2.508	1618.7	0.6664	-	7.632
-	7.734	2.705	1795.8	0.6503	-	7.910

$$US = 4.464 + 1.216 UP \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
REPORT NO. GMX-6-566, PP. 51-62 (1964)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS 2024 ALUMINUM
- 3) THE ALUMINUM HUGONIOT IS REPRESENTED BY
 $US = 5.355 + 1.345 UP \text{ KM/SEC}$
THE PARTIAL $(\partial E / \partial P)V = 0.162 \text{ CUBIC CM}$
- 4) VOI WAS TAKEN FROM THE AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCGRAH-HILL BOOK CO. N.Y., 1963) 2ND ED.

TABLE I



55---3
NIOBIUM

NB

$V_0 = 0.117 \text{ CC/G}$
 $V_{01} = 0.11652 \text{ CC/G}$

 $C_0 = 4.48 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
8.58	10.28	4.50	3970.	0.562	IRON	7.85	11.21
-	8.26	3.06	2169.	0.630	IRON	7.85	8.83
-	6.64	1.87	1065.	0.718	ALUMINUM	2.71	9.17
-	6.31	1.73	937.	0.726	COPPER	8.93	6.45
-	5.53	0.95	450.	0.829	ALUMINUM	2.71	7.34
-	5.17	0.40	177.	0.923	ALUMINUM	2.71	6.21

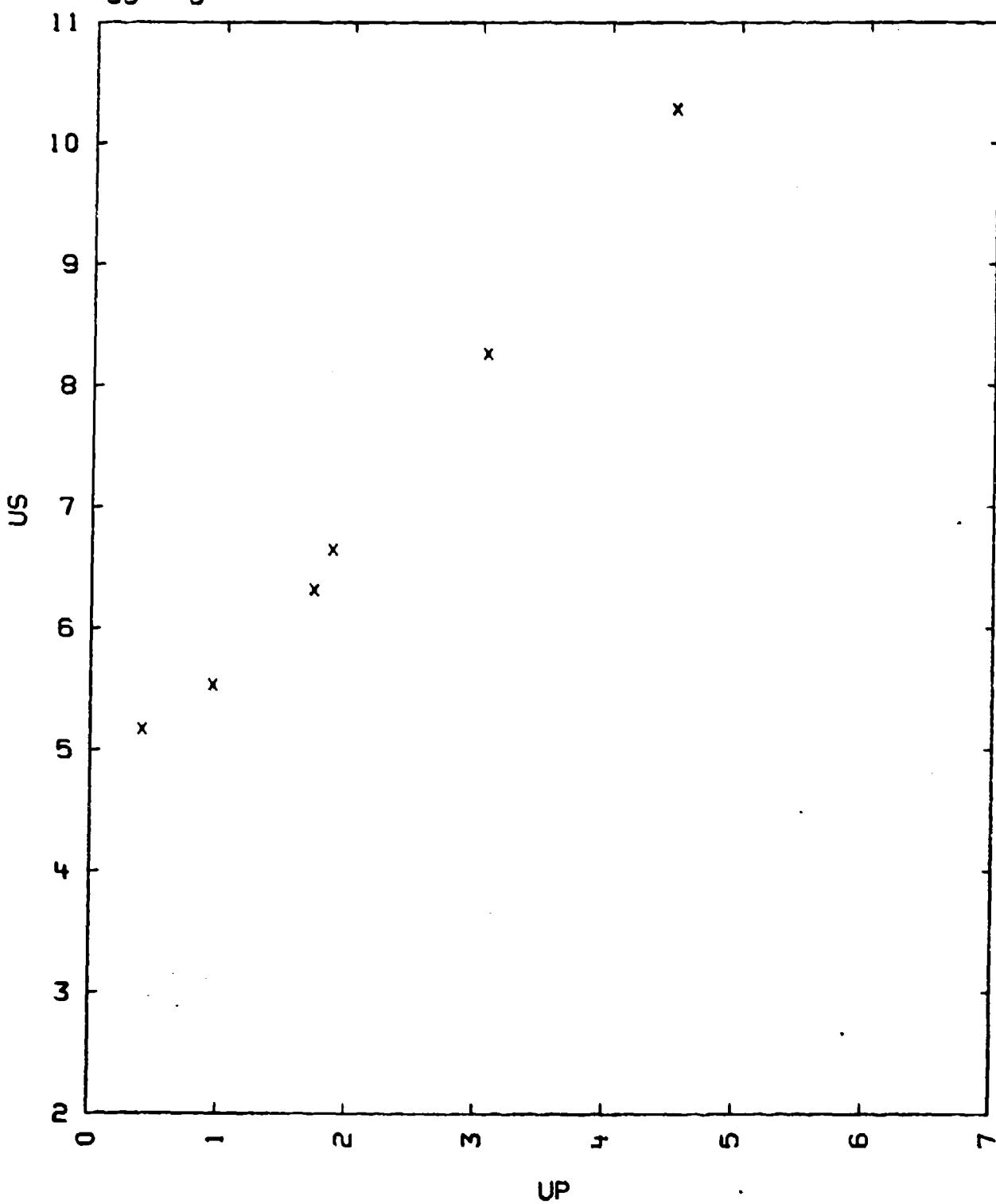
 $US = 4.70 + 0.920 \cdot UP \text{ KM/SEC}$ FOR UP BELOW 1.62 KM/SEC $US = 3.91 + 1.409 \cdot UP \text{ KM/SEC}$ FOR UP BETWEEN 1.62 AND 4.50 KM/SEC

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P.
ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES.
THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE
1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = $5.25 + 1.39 \cdot UP$ FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
COPPER US = $3.95 + 1.50 \cdot UP$ FOR UP BETWEEN 0.0 AND 4.0 KM/SEC
IRON US = $3.85 + 1.615 \cdot UP$ FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF V_{01} WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANT
OF THE CUBIC BODY-CENTERED STRUCTURE,
 $A = 3.3004 \text{ ANGSTROMS}$ (IB DEG. C)
NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 16.
- 6) THE AUTHORS OBTAINED THE VALUE OF C_0 FROM HANDBOOK OF RARE METALS,
(MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO
SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT $UP = 1.62 \text{ KM/SEC}$ IS
ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I

NIOBIUM
55---3



55--4
NIOBUM

NB

$V_0 = 0.1166 \text{ CC/G}$ $C_L = 5.03 \text{ KM/SEC}$ $C_0 = 4.39 \text{ KM/SEC}$
 $V_{01} = 0.1165 \text{ CC/G}$ $C_S = 2.11 \text{ KM/SEC}$ $C_B = 4.43 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(IST)
8.583	5.12	0.50	220.	0.9023	CU	4.70
8.604	5.11	0.53	233.	0.8963	2024 AL	6.49
8.586	5.26	0.67	303.	0.8726	CU	4.98
8.604	5.25	0.72	325.	0.8629	2024 AL	6.86
8.584	5.60	0.94	452.	0.8321	CU	5.35
8.604	5.57	0.97	465.	0.8259	2024 AL	7.36
8.583	5.71	1.02	500.	0.8214	CU	5.47
8.584	6.36	1.60	874.	0.7484	CU	6.29
8.580	6.34	1.61	876.	0.7461	CU	6.30
8.578	6.38	1.61	881.	0.7476	CU	6.31
8.584	7.01	2.11	1270.	0.6990	CU	7.03
8.571	7.31	2.36	1479.	0.6772	CU	7.39
8.585	7.41	2.46	1565.	0.6680	CU	7.52
8.584	7.62	2.65	1733.	0.6522	CU	7.80

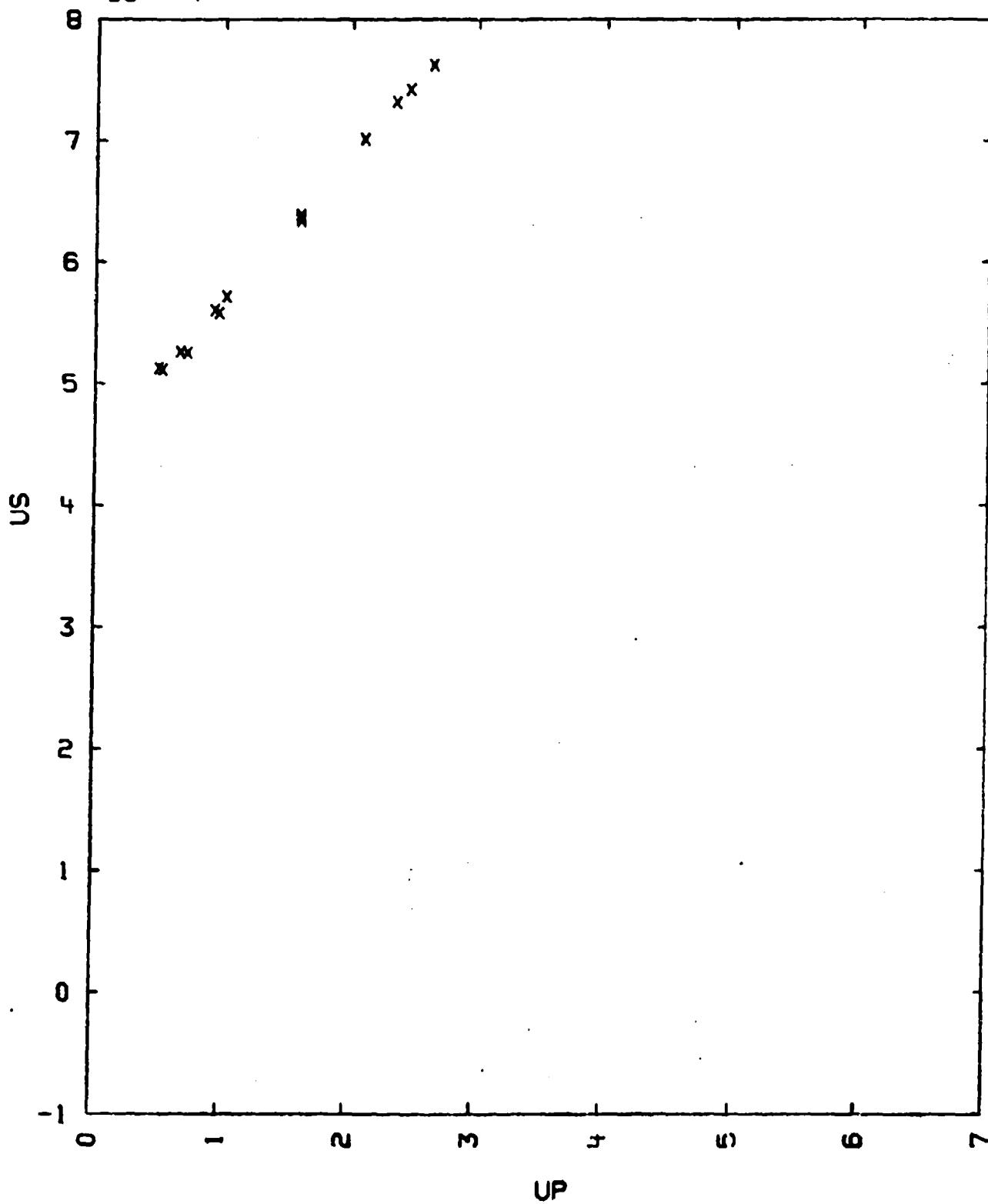
$US = 4.438 + 1.207 \cdot UP \text{ KM/SEC}$
 $\Sigma \text{SIGMA US} = 0.037 \text{ KM/SEC}$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES.
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B
- 3) V_{01} IS FROM WYCKOFF CRYSTALL STRUCTURES (JOHN WILEY AND SONS, N.Y., 1963) V. I
- 4) $V(DP/DE) = 1.47$
- 5) HUGONIOT ELASTIC LIMIT 20.7 KBAR

TABLE I

NIOBIUM
55---4



56---1
TANTALUM

TA 99.8 PER CENT OR GREATER

$V_0 = 0.06075 \text{ CC/G.}$ $C_L = 3.35 \text{ KM/SEC.}$
 $V_{01} = 0.05848 \text{ CC/G.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0
16.46	3.811	0.4327	271.5	0.8865
-	4.010	0.5800	383	0.8554
-	4.323	0.7685	547	0.8222

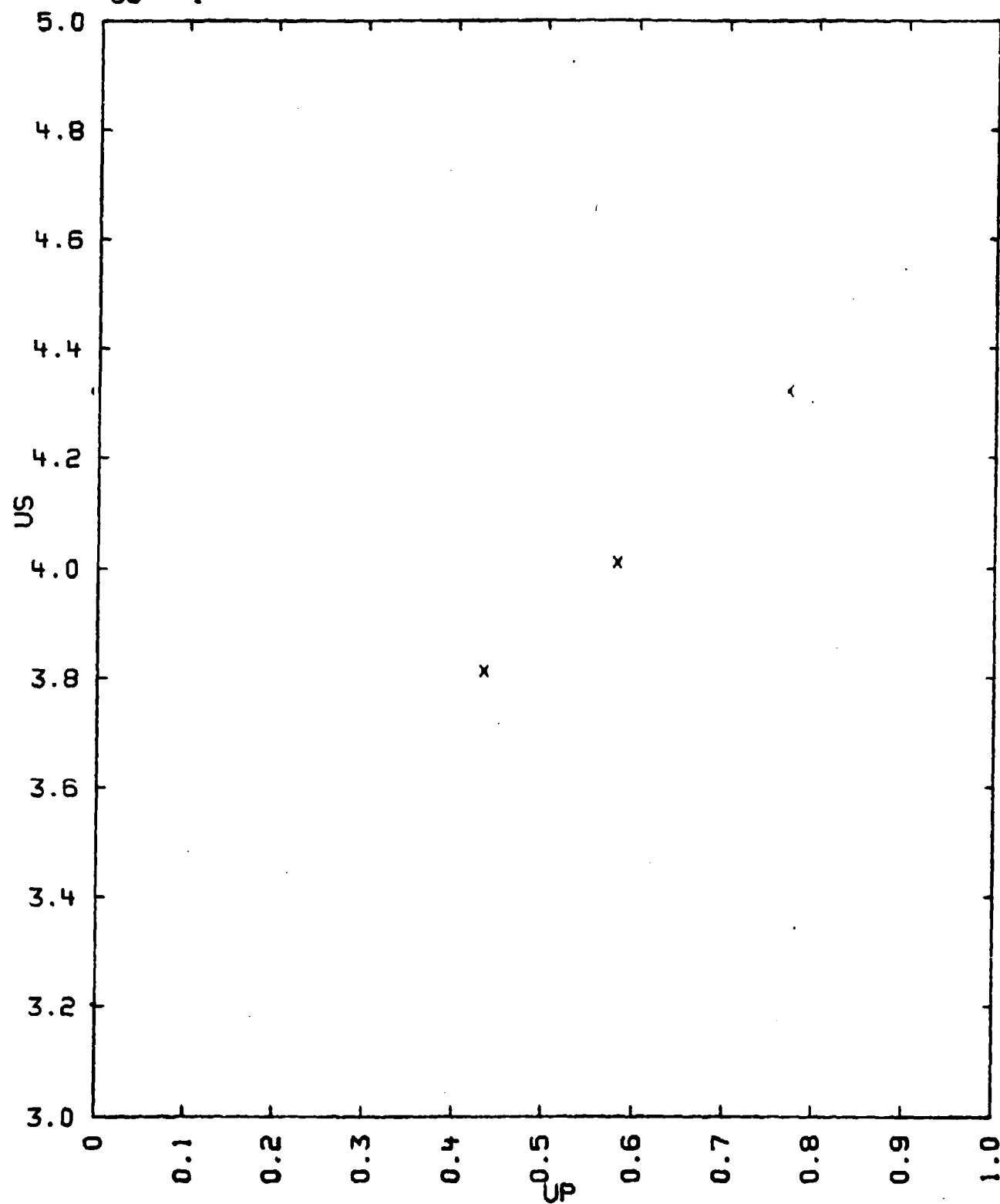
$$US = 3.139 + 1.531 UP \quad SIGMA = 0.5 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 3 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS
- 5) CL OBTAINED FROM L. BERGMAN, DER ULTRASCHALL.
(S. HIRZEL VERLAG, STUTTGART, GERMANY, 1954) 6TH ED., P. 650

TABLE I

TANTALUM
56---1



56--2
TANTALUM

TA 99.5 PERCENT OR GREATER

$V_0 = 0.0600 \text{ g/cc}$. $C_L = 4.15 \text{ km/sec}$. $C_0 = 3.39 \text{ km/sec}$.
 $V_{01} = 0.0585 \text{ g/cc}$. $C_S = 2.07 \text{ km/sec}$. $C_B = 3.41 \text{ km/sec}$.

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	RHO0	US
16.66	3.981	0.429	284.5	0.8922	8.932	4.784
-	4.134	0.576	396.2	0.8607	8.934	5.024
-	4.395	0.804	588.3	0.8171	-	5.426
-	4.495	0.872	653.0	0.8060	-	5.548
-	5.092	1.358	1152.0	0.7333	-	6.393
-	5.061	1.359	1145.2	0.7315	-	6.387
-	5.621	1.790	1676.3	0.6816	-	7.133
-	5.889	2.003	1962.8	0.6599	-	7.494
-	5.967	2.087	2074.7	0.6502	-	7.632
-	6.103	2.262	2298.5	0.6294	-	7.910

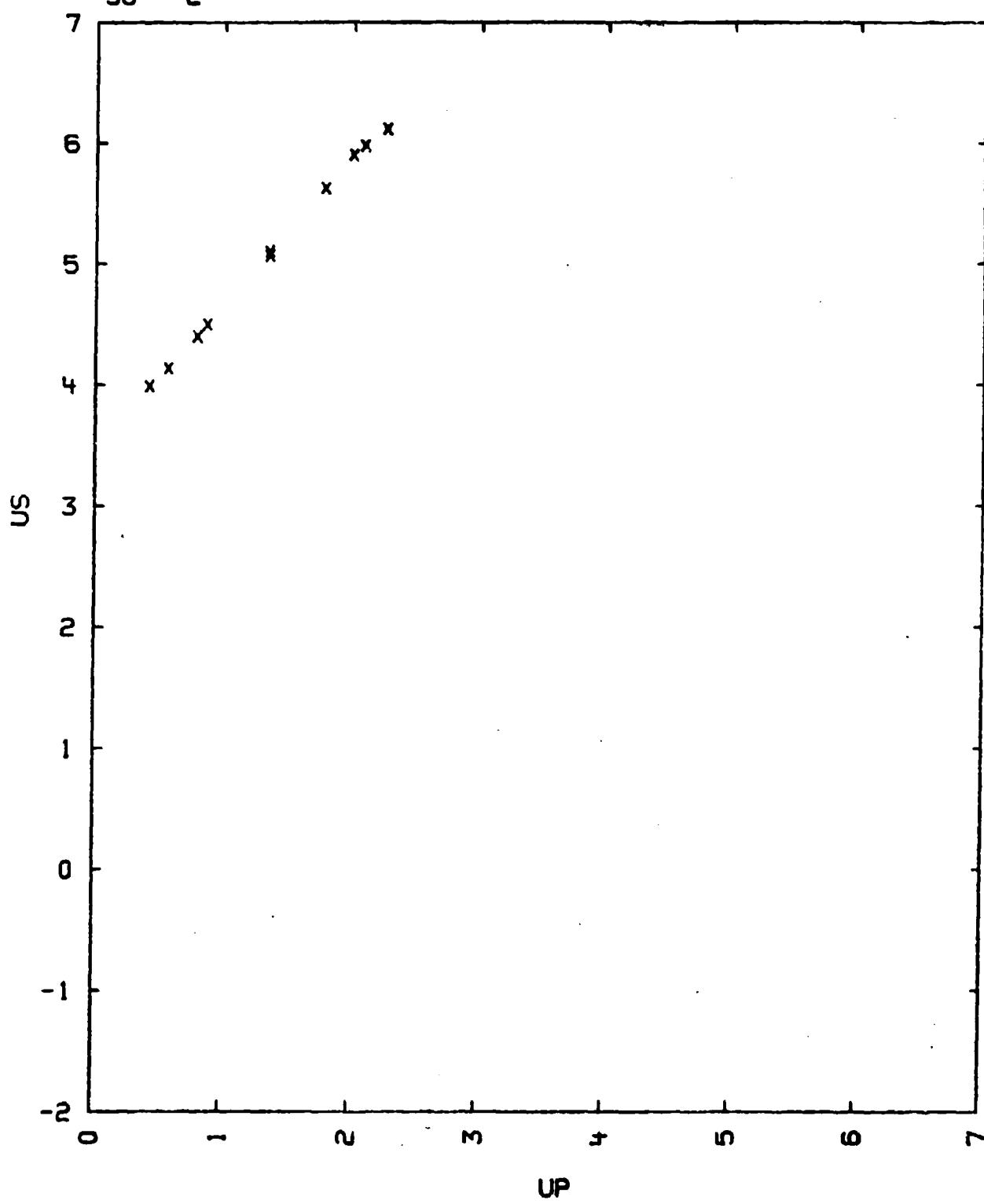
$$US = 3.453 + 1.197 UP, \text{ SIG.US} = 0.029 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
REPORT NO. GMX-6-566, PP. 51-62 (1964)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS COPPER
- 3) COPPER HUGONIOT IS REPRESENTED BY:
 $US = 3.958 + 1.497 UP \text{ KM/SEC.}$
THE PARTIAL $(\partial E / \partial P)V = 0.0562 \text{ CUBIC CM.}$
- 4) V_{01} WAS TAKEN FROM THE AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO. N.Y., 1963) 2ND ED.
- 5) A HUGONIOT ELASTIC LIMIT OF 17 KBAR. HAS BEEN MEASURED BY
R. HASLEY AND K.G. HOGE, CHEM. TECH. NOTE 292, LAWRENCE LIVERMORE LAB
LIVERMORE CAL.

TABLE I

TANTALUM
56---2



56---3
TANTALUM

TA

$$V_0 = 0.0602 \text{ g/cc.}$$

$$V_{01} = 0.05994 \text{ g/cc}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL USED AS STRIKER PLATE. U(ST) IS THE VELOCITY BEFORE IMPACT.

TABLE

RHO0	US	UP	P	V/V0	ST	U(ST)
16.6	3.24	1.45	1245	0.7231	AL	5.60
-	6.45	2.28	2409	0.6464	FE	5.60
-	8.36	3.74	5121	0.5525	FE	9.10

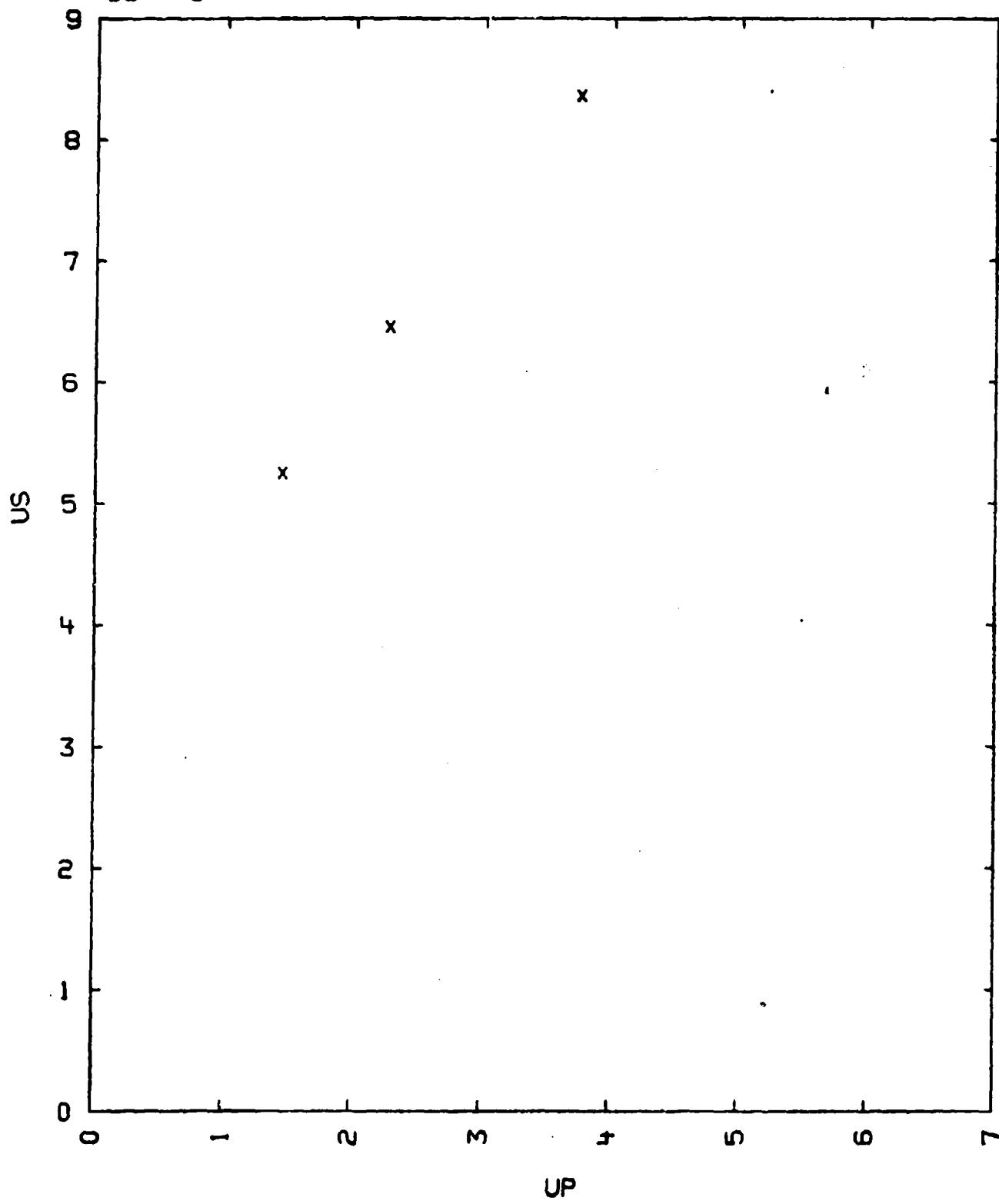
US = $3.31 + 1.36 \cdot UP$ KM/SEC. FOR UP BETWEEN 1.45 TO 3.74 KM/SEC.
 SIGMA US = 0.06 KM/SEC.

COMMENTS:

- 1) SOURCE: KRUPNIKOV, K. K., BAKANOVA, A. A., BRAZHIK, M. I. AND TRUNIN, R. F.
 SOVIET PHYS.-DOKLADY, VOL. 8, P. 205 (1963)
- 2) EXPERIMENTAL TECHNIQUE A
 DATA REDUCTION TECHNIQUE B
- 3) THE SAMPLES WERE POSITIONED ON PLATES OF AL OR FE AS INDICATED IN COLUMN 6.
- 4) THE SPECIMENS WERE 3 TO 4 MM THICK AND 10 TO 15 MM IN DIAMETER.
- 5) THE PRESSURES WERE PRODUCED BY AN EXPLOSIVELY ACCELERATED ALUMINUM OR STEEL PLATE AS INDICATED IN COLUMN 6.
- 6) V01 WAS OBTAINED FROM A CUBIC LATTICE CONSTANT OF 3.3026 ANGSTROM AMERICAN INSTITUTE OF PHYSICS HANDBOOK (MCGRAW-HILL BOOK CO., N.Y., 1963) 2ND ED.
- 7) ALSO LISTED ARE: RHO0 AT 0 DEG. K. 16.18 G/CC
 ELECTRONIC HEAT CAPACITY COEFFICIENT
 $325 \text{ ERGS/G DEG}^{+2}$
 GRUNEISEN CONSTANT = 1.68

TABLE I

TANTALUM
56---3



56---4
TANTALUM

TA

$V_0 = 0.06006 \text{ G/CC}$ $C_L = 4.15 \text{ KM/SEC}$ $C_D = 3.39 \text{ KM/CC}$
 $V_{01} = 0.06012$ - $C_S = 2.07$ - $C_B = 3.41$ -

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
16.660	3.92	0.40	261.	0.8980	CU	4.70
16.640	4.07	0.55	372.	0.8649	CU	4.95
16.650	4.33	0.77	555.	0.8222	CU	5.35
16.660	4.43	0.84	620.	0.8104	CU	5.47
16.650	4.99	1.32	1097.	0.7355	CU	6.29
16.660	5.02	1.32	1104.	0.7371	CU	6.30
16.660	5.54	1.75	1615.	0.6841	CU	7.03
16.640	5.80	1.96	1892.	0.6621	CU	7.39
16.660	5.88	2.04	1998.	0.6531	CU	7.52
16.650	6.01	2.22	2221.	0.6306	CU	7.80

$$\text{US} = 3.414 + 1.200 \cdot \text{UP KM/SEC}$$

$$\text{SIGMA US} = 0.06 \text{ KM/SEC}$$

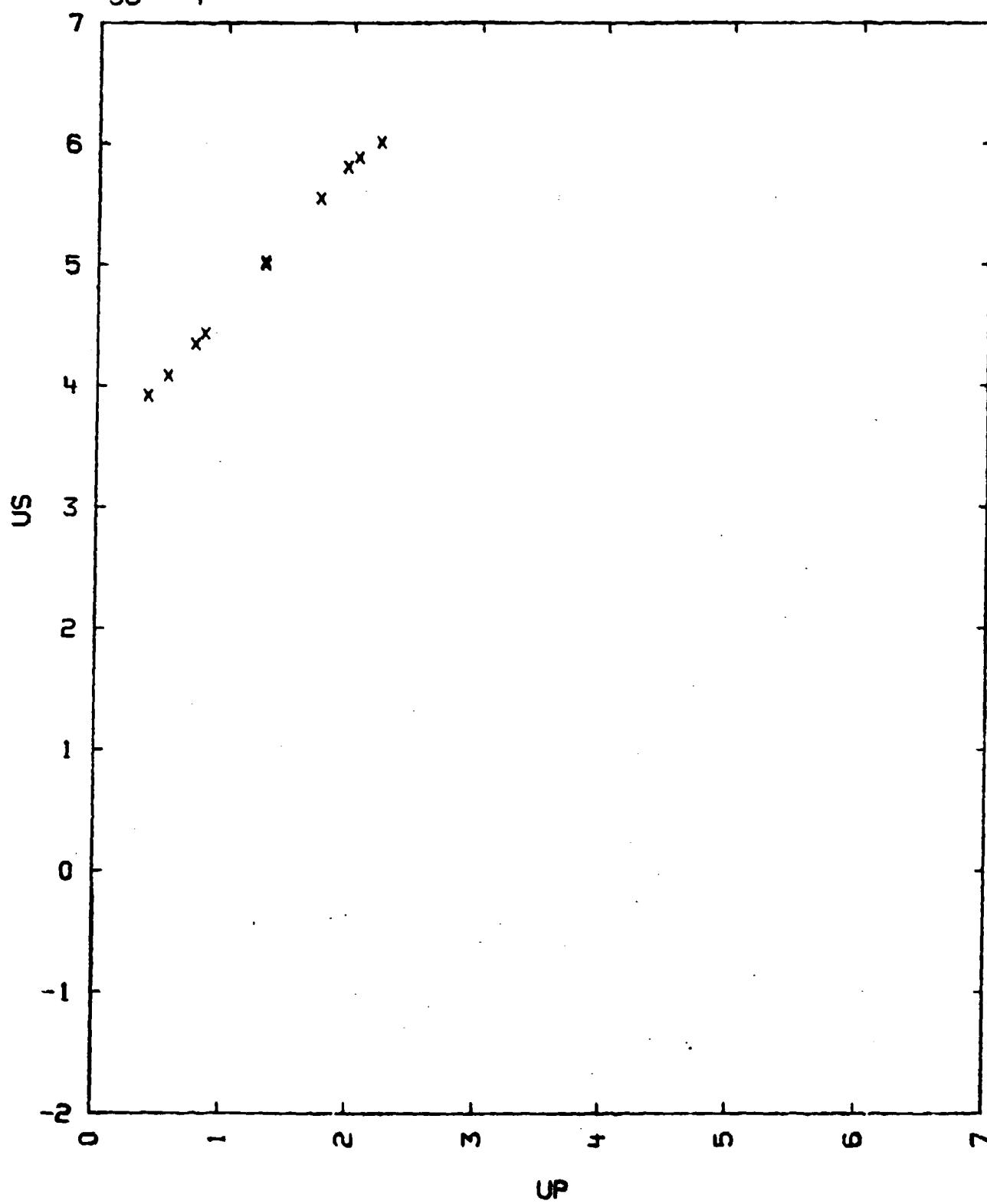
COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M.,
AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION METHOD : B
- 3) HUGONIOT ELASTIC LIMIT = 8 KBAR
- 4) VOI FROM WYCKOFF CRYSTAL STRUCTURES (JOHN WILEY AND SONS, N.Y. 1963)
VOL. I.
- 5) HUGONIOT ELASTIC LIMIT 18.7 KBAR

TABLE I

TANTALUM

56---4



57---0
TITANIUM SUMMARY

T1

V0 = 0.2217 - 0.2222 CC/G
V0I = 0.2183 CC/G

CB = 4.84 KM/SEC

THE TABLE LISTS HUGONIOT POINTS CALCULATED FROM THE FIT GIVEN BELOW.
UNITS ARE: G/CC, KM/SEC, KBAR AND KBAR.CC/G FOR THE ENERGY DIFFERENCE.

TABLE

RHO0	US	UP	P	V/V0	E-E0
4.51	5.382	0.6	146.	0.8885	1.8
-	5.841	1.0	263.	0.8288	5.0
-	6.897	2.0	630.	0.714	20.0
-	8.133	3.0	1100.	0.631	45.0
-	9.279	4.0	1674.	0.569	80.0
-	10.425	5.0	2351.	0.520	125.
-	11.571	6.0	3131.	0.481	180.

US = 4.695 + 1.146 UP, SIG.US = 0.10 KM/SEC

FOR UP BETWEEN .68 AND 5.6 KM/SEC.

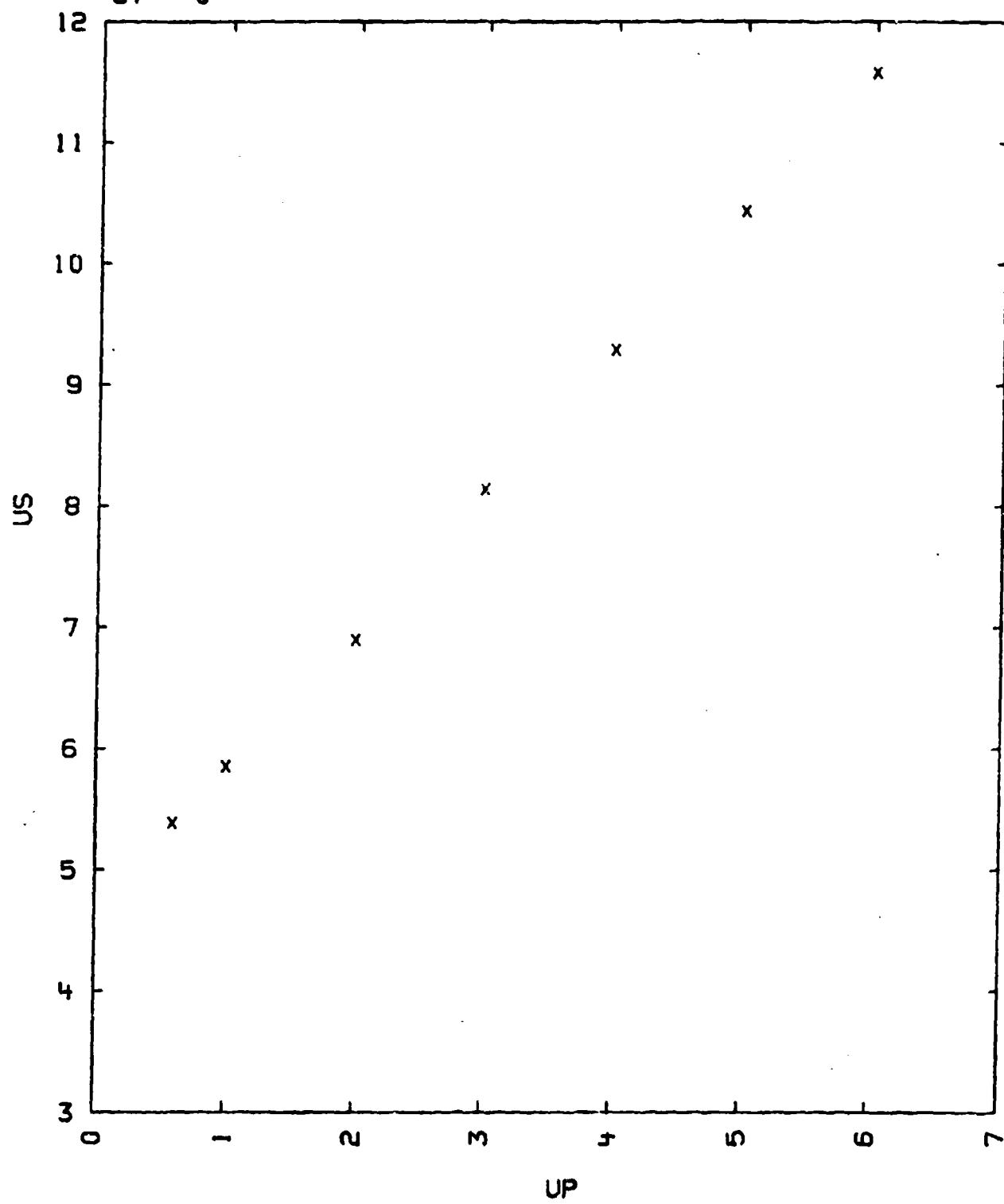
COMMENTS:

- 1) SOURCE: COMPILER
- 1A) DATA FROM 57---1,2 AND 3 WERE USED FOR THIS FIT.
- 2) V0I AND CB WERE TAKEN FROM 57---2.

TABLE I

TITANIUM SUMMARY

57---0



57---1
TITANIUM

TI 99.75 PER CENT
 AL 0.05 PER CENT
 SI 0.05 PER CENT
 CR 0.05 PER CENT
 MN 0.05 PER CENT
 SN 0.05 PER CENT

$$V_0 = 0.2217 \text{ CC/G.}$$

$$V_{01} = 0.2183 \text{ CC/G.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
 PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

RHO0	US	UFS	UP	P	V/V0
4.51	6.329	2.723	1.370	390.8	0.7835
-	5.790	1.926	0.980	255.7	0.8307
-	5.501	1.437	0.723	179.3	0.8686
-	5.469	1.364	0.684	168.3	0.8749

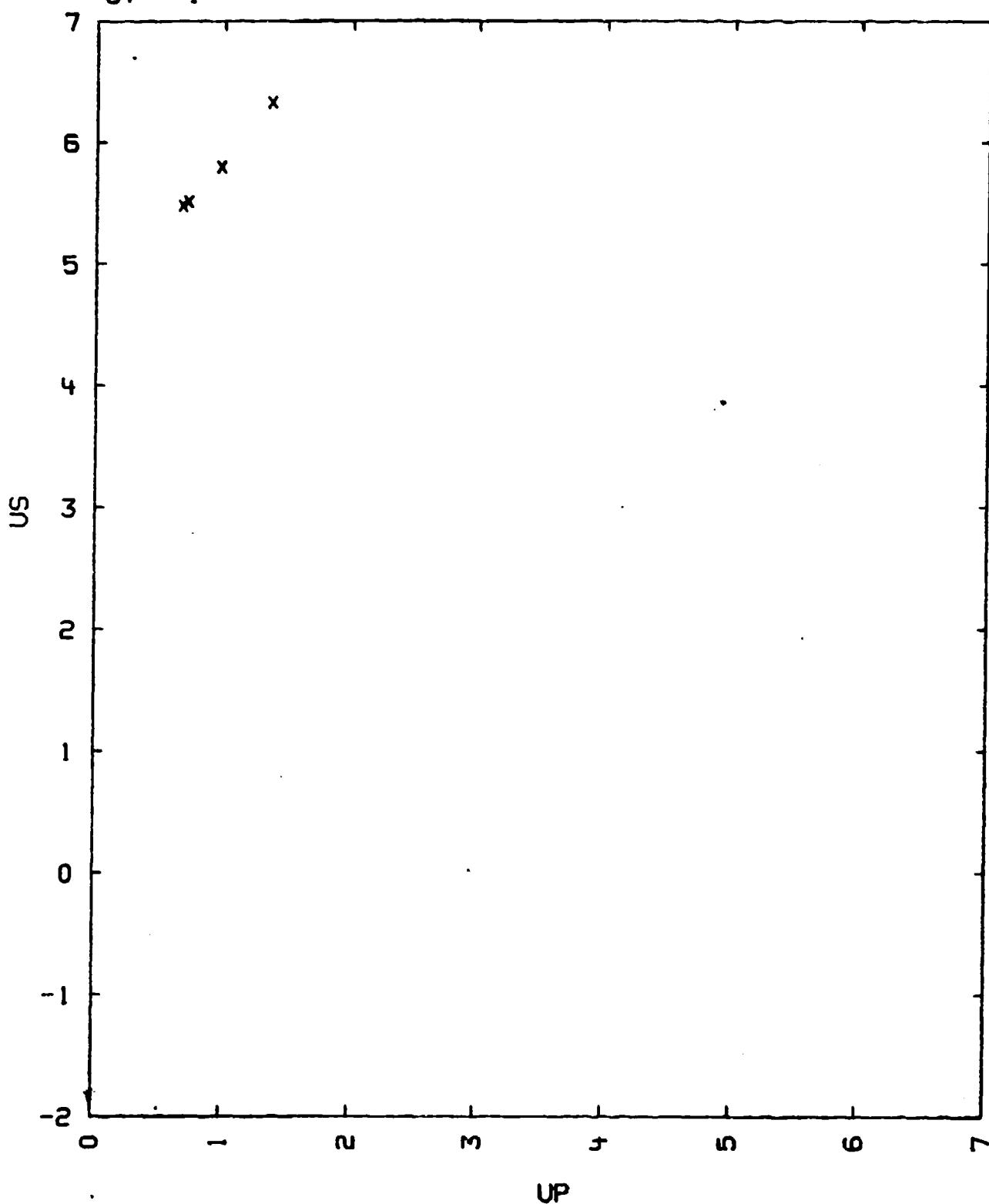
$$US = 4.590 + 1.259 UP \text{ KM/SEC} \quad \text{SIGMA US} = 0.5 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J.M., RICE, M.H., MCQUEEN, R.G. AND YARGER, F.L.
 PHYS. REV., VOL. 108, P. 169 FF. (1957)
 LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
 DATA REDUCTION TECHNIQUE B
 STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
 FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
 COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS

TABLE I

TITANIUM
57---1



57---2
TITANIUM

TI

$$V_0 = 0.2217 \text{ CC/G.}$$

$$V_{01} = 0.2183 \text{ CC/G.}$$

$$C_0 = 4.84 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE IN BRASS STANDARD
4.51	7.35	2.30	762	0.687	1043
-	7.40	2.29	764	0.691	1040
-	7.34	2.29	758	0.689	1038
-	7.94	2.97	1063	0.626	1497
-	7.92	2.97	1060	0.625	1497

$$US = 4.779 + 1.089 UP \text{ KM/SEC.}$$

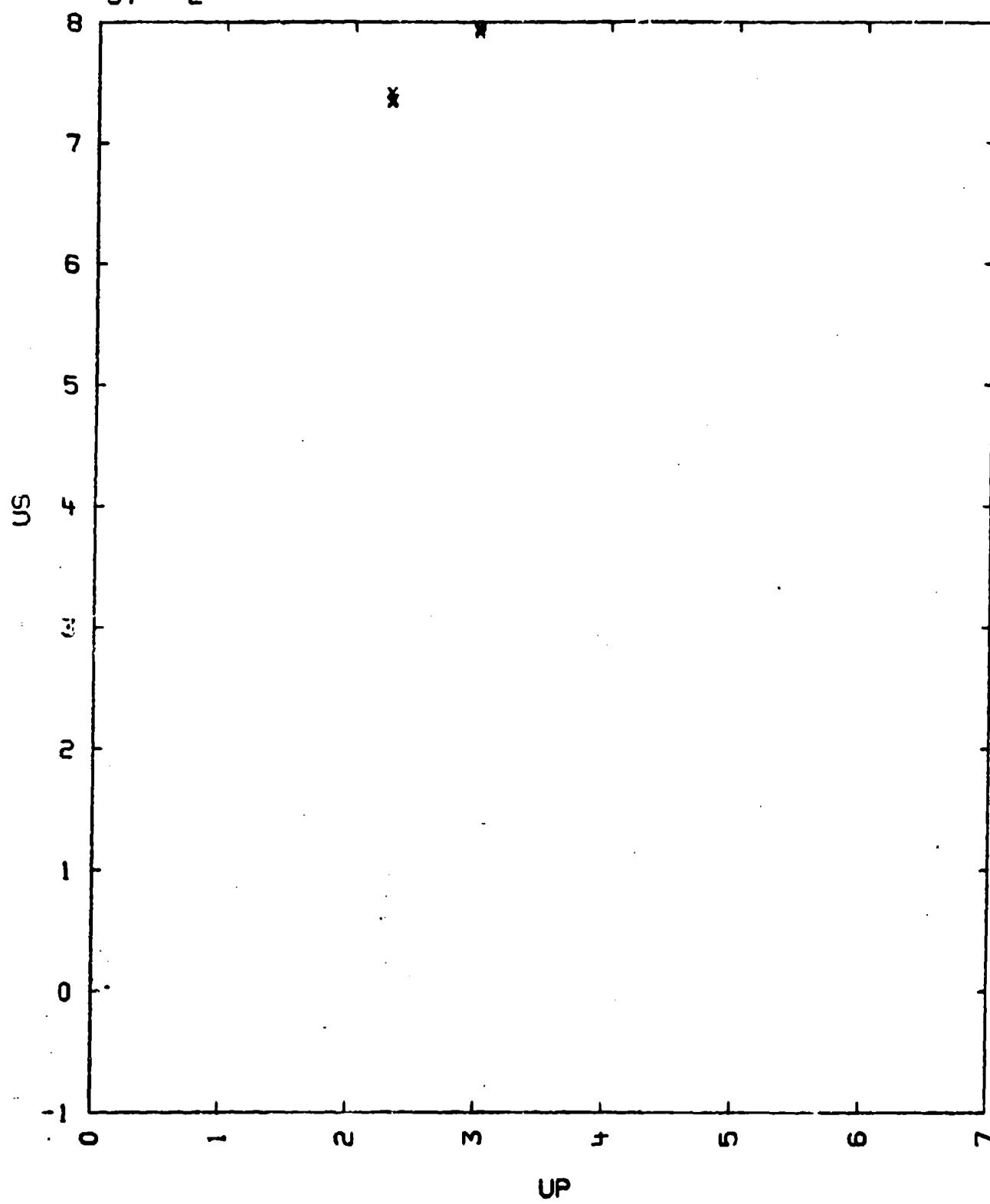
$$\text{SIGMA US} = 0.74 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 4) V01 WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW HILL BOOK CO.)
2ND ED.

TABLE I

TITANIUM
57---2



57---3
TITANIUM

TI

$$V_0 = 0.222 \text{ CC/G}$$

$$V_{01} = 0.2183 \text{ CC/G}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC. ST DESIGNATES THE STANDARD MATERIAL USED AS STRIKER PLATE. U(ST) IS THE VELOCITY BEFORE IMPACT.

TABLE

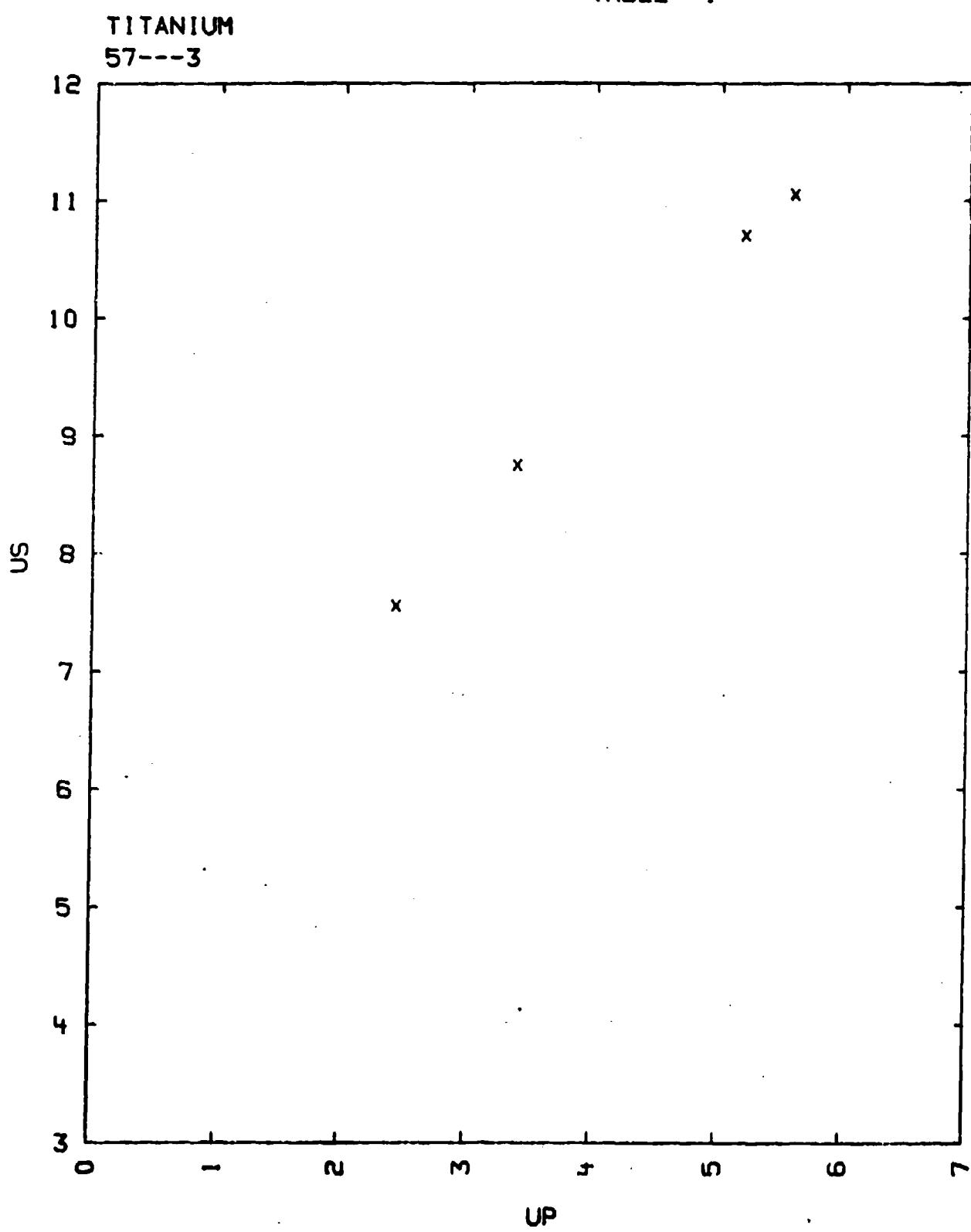
RHO0	US	UP	P	V/V0	ST	U(ST)
4.50	7.55	2.44	828	0.6768	AL	5.60
-	8.74	3.39	1331	0.6124	FE	5.60
-	10.70	5.19	2500	0.5155	FE	8.640
-	11.05	5.58	2775	4.4950	FE	9.10

US = $4.89 + 1.11 \text{ UP KM/SEC.}$ FOR UP BETWEEN 2.44 TO 5.59 KM/SEC.
 $\Sigma \text{ US} = 0.08 \text{ KM/SEC.}$

COMMENTS:

- 1) SOURCE: KRUPNIKOV, K. K., BAKANOVA, A. A., BRAZHNICK, M. I. AND TRUNIN, R. F.
 SOVIET PHYS.-JETP, VOL. 8, P. 205 (1963)
- 2) EXPERIMENTAL TECHNIQUE A
 DATA REDUCTION TECHNIQUE B
- 3) THE SAMPLES WERE POSITIONED ON PLATES OF AL OR FE AS INDICATED IN COLUMN 6.
- 4) THE SPECIMENS WERE 3 TO 4 MM THICK AND 10 TO 15 MM IN DIAMETER.
- 5) THE PRESSURES WERE PRODUCED BY AN EXPLOSIVELY ACCELERATED ALUMINUM OR STEEL PLATE AS INDICATED IN COLUMN 6.
- 6) V01 WAS OBTAINED FROM THE AMERICAN INSTITUTE OF PHYSICS HANDBOOK, (MCGRAW-HILL BOOK CO. 1957) 2ND ED.
- 7) ALSO LISTED ARE: RHO0 AT 0 DEG. K. = 4.524 G/CC
 ELECTRONIC HEAT CAPACITY COEFFICIENT
 $710 \text{ EROS}/0 \text{ DEG}^{+2}$
 GRUNEISEN CONSTANT = 1.14

TABLE I



57---4
TITANIUM

TI

$V_0 = 0.2203-0.2217 \text{ CC/G}$
 $V_{01} = 0.2220 \text{ CC/G}$

 $C_B = 5.22 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
4.517	5.54	0.40	100.	0.9278	2024 AL	5.98
4.517	5.58	0.52	131.	0.9068	2024 AL	6.18
4.517	5.71	0.55	142.	0.9037	CU	4.62
4.539	5.68	0.57	147.	0.8996	2024 AL	6.26
4.517	5.71	0.63	162.	0.8897	CU	4.71
4.539	5.75	0.71	185.	0.8765	2024 AL	6.48
4.539	5.73	0.71	185.	0.8761	2024 AL	6.48
4.517	5.69	0.75	193.	0.8682	2024 AL	6.53
4.517	5.73	0.78	202.	0.8639	CU	4.88
4.539	5.95	1.00	270.	0.8319	2024 AL	6.91
4.539	5.89	1.01	270.	0.8285	2024 AL	6.91
4.517	5.87	1.03	273.	0.8245	921-T AL	6.76
4.517	6.09	1.16	319.	0.8095	CU	5.33
4.539	6.29	1.31	374.	0.7917	2024 AL	7.38
4.539	6.23	1.32	373.	0.7881	2024 AL	7.38
4.517	6.36	1.44	414.	0.7736	921-T AL	7.41
4.539	6.45	1.49	436.	0.7690	2024 AL	7.64
4.539	6.44	1.49	436.	0.7686	2024 AL	7.64
4.539	6.53	1.56	482.	0.7611	2024 AL	7.74
4.539	6.50	1.56	460.	0.7600	2024 AL	7.74
4.539	6.68	1.66	503.	0.7515	2024 AL	7.90
4.539	6.65	1.66	501.	0.7504	2024 AL	7.90
4.539	6.69	1.71	395.	0.8251	2024 AL	7.97
4.539	6.68	1.71	518.	0.7440	2024 AL	7.97
4.517	6.99	1.94	613.	0.7225	921-T AL	8.20
4.539	7.14	2.17	703.	0.5961	2024 AL	8.64
4.539	7.12	2.17	701.	0.8952	2024 AL	8.84
4.510	7.31	2.22	732.	0.6983	BRASS	6.42
4.510	7.25	2.23	729.	0.6924	BRASS	6.42
4.510	7.26	2.23	730.	0.6928	BRASS	6.43
4.517	7.27	2.28	749.	0.6864	921-T AL	8.72
4.510	7.51	2.50	881.	0.6538	2024 AL	9.24
4.539	7.77	2.75	970.	0.6461	2024 AL	9.50
4.539	7.68	2.76	952.	0.6408	2024 AL	9.50
4.517	7.83	2.85	1008.	0.6360	CU	7.35
4.510	7.84	2.89	1022.	0.6314	BRASS	7.18
4.510	7.82	2.89	1019.	0.6304	BRASS	7.18
4.539	8.19	3.15	1171.	0.6154	2024 AL	10.08

TITANIUM

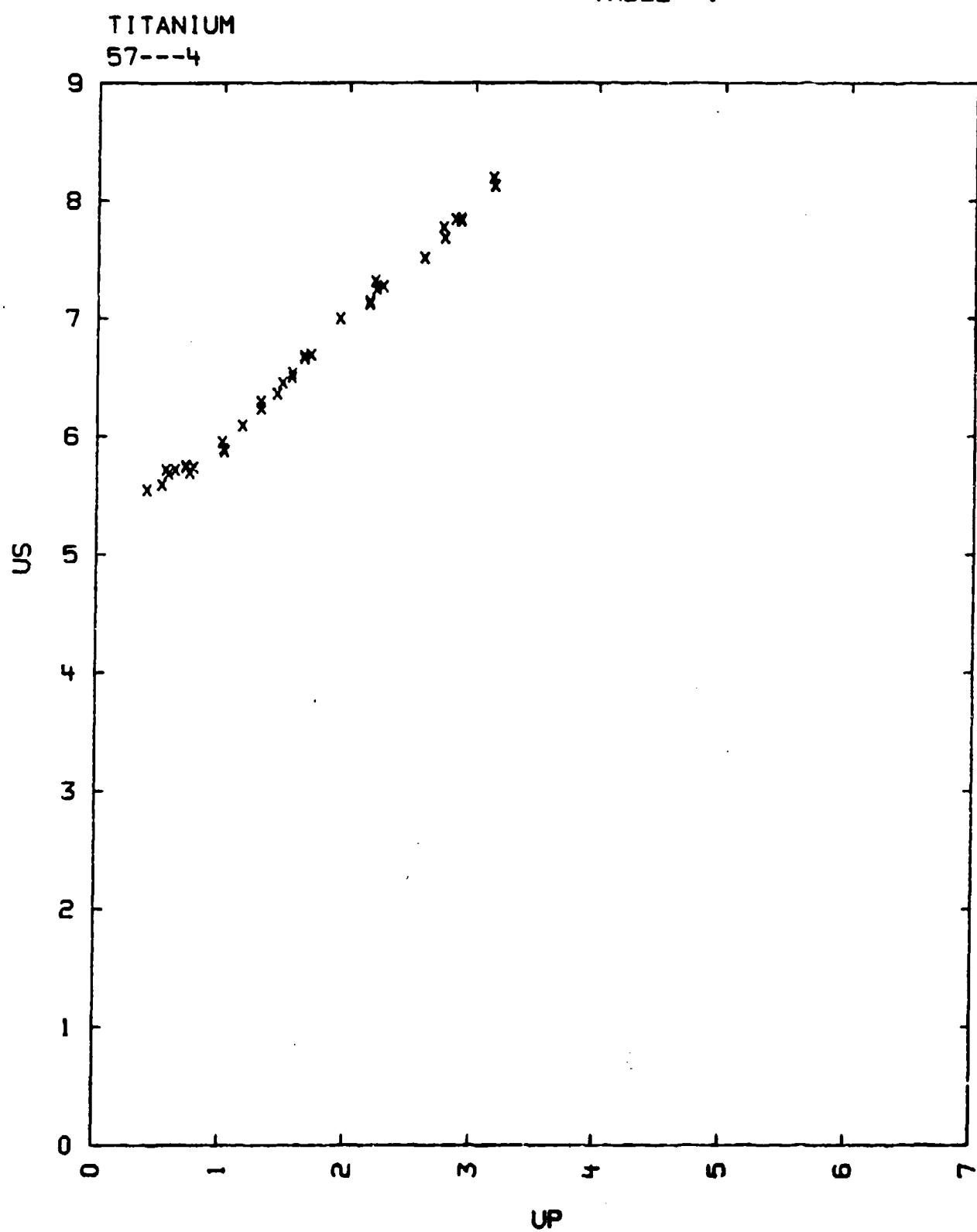
RHO0	US	UP	P	V/V0	MATERIAL US(ST)
4.539	8.12	3.16	1165.	0.6108	2024 AL 10.08

US = $5.220 + 0.767 \cdot UP$ KM/SEC KM/SEC FOR UP LESS THAN 0.72 AND USING CO
 SIG US = 0.038 KM/SEC
 US = $4.889 + 1.045 \cdot UP$ KM/SEC
 SIG US = 0.068 KM/SEC

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES.
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B
- 3) VOI FROM WYCKOFF, CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, N.Y. (1963) V. 1
- 4) SAMPLES WERE STRONGLY ANISOTROPIC, SUGGESTING THAT THE ISOTROPY ASSUMPTION IN CALCULATING CO IS POOR.
- 5) TRANSITION AT 175 KBAR PROBABLY FCC-BCC
- 6) HUGONIOT ELASTIC LIMIT 18.5 KBAR

TABLE I



57---5
TITANIUM

TI	99.6	WT. PERCENT
FE	0.25	-
C	0.02	-
N	0.01	-
O	0.115-0.123	-

$$V_0 = 0.2218 \text{ CC/G} \quad C_L = 6.118 \text{ KM/SEC} \quad C_D = 4.83 \text{ KM/SEC}$$

$$V_{01} = 0.2220 \quad - \quad C_S = 3.246 \quad -$$

THE TABLE LISTS RHO0 IN G/CC, VELOCITIES IN KM/SEC AND P IN KBARS.
TI = TITANIUM, FS = FANSTEEL AND WF = WEIGHTING FACTOR.

TABLE

SAMPLE				IMPACTOR			
RHO0	US	UP	P	V/V0	MAT	U	WF
4.508	6.321	1.422	405.	.7750	TI	2.845	1
4.508	7.604	2.621	899.	.6553	TI	5.242	1
4.508	10.939	5.534	2728.	.4939	FS	7.849	1

$$US = 4.695 + 1.146 \cdot UP \text{ KM/SEC}$$

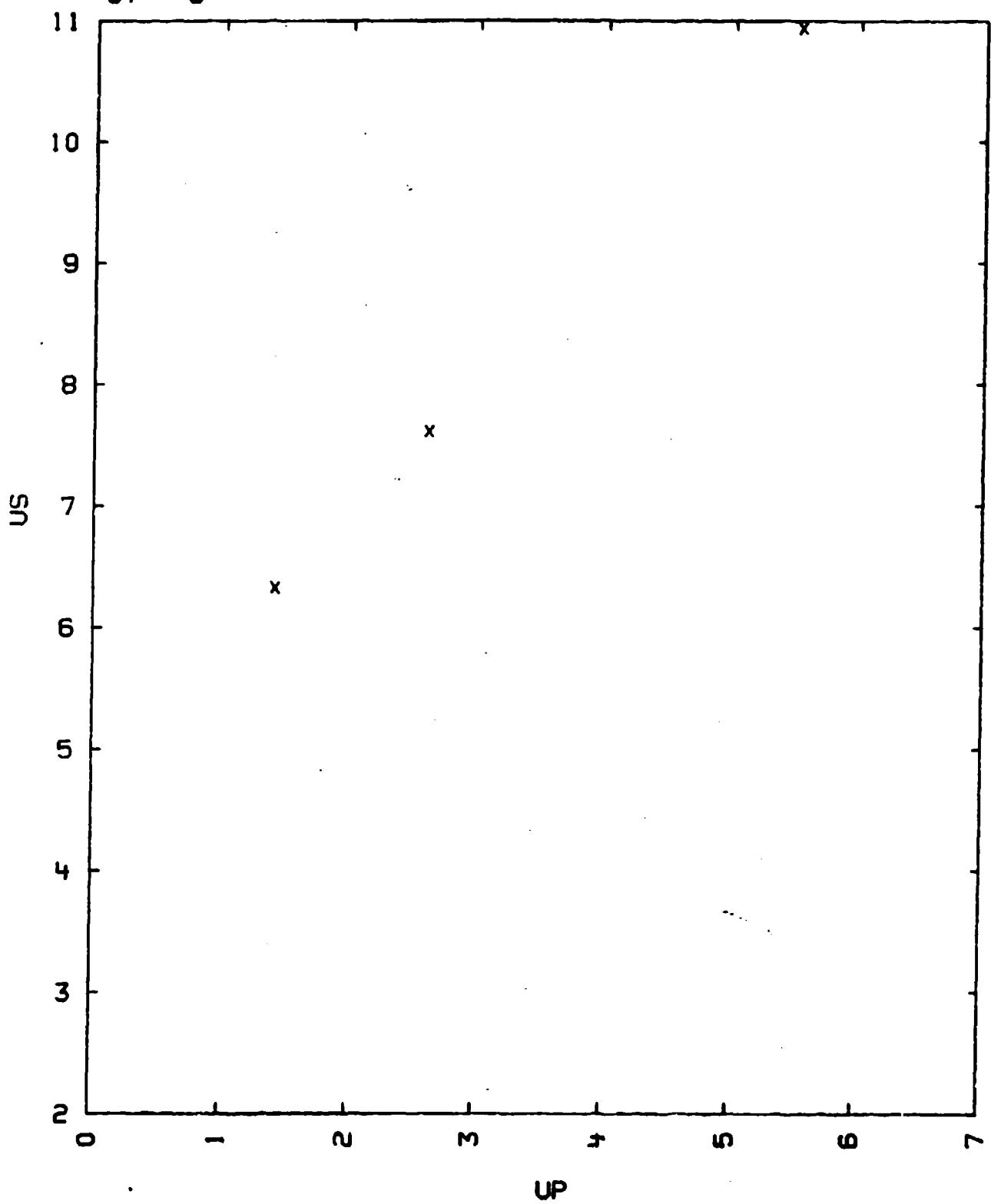
$$SIG.US = 0.045 \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: ISBELL W.M., SHIPMAN F.H. AND JONES A.H.
HUGONIOT EQUATION OF STATE OF ELEVEN MATERIALS TO FIVE KBARS
MATERIALS SCIENCE LABORATORY REPORT: MSL-68-13
- 2) EXPERIMENTAL TECHNIQUE: A
DATA REDUCTION METHOD : A
- 3) NOMINAL UNCERTAINTIES ARE: $(SIG.US)/US = .005$ AND $(SIG.U)/U = .0005$
- 4) ALSO LISTED ARE: POISSONS RATIO= 0.304, YIELD STRENGTH = 3.4-3.8 KBAR
TENSILE = 5.2-5.3 KBAR
- 5) V01 OBTAINED FROM WYCKOFF CRYSTAL STRUCTURES
(JOHN WILEY AND SONS, N.Y. 1963) VOL. 1

TABLE I

TITANIUM
57---5



58--1
ZIRCONIUM

ZR 99.8 PER CENT OR GREATER

$$V_0 = 0.1540 \text{ CC/G.}$$

$$V_{01} = 0.1545 \text{ CC/G.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	VIVO
6.49	4.494	0.7117	207.5	0.8416
-	4.674	0.9563	290	0.7954
-	4.920	1.275	407	0.7408

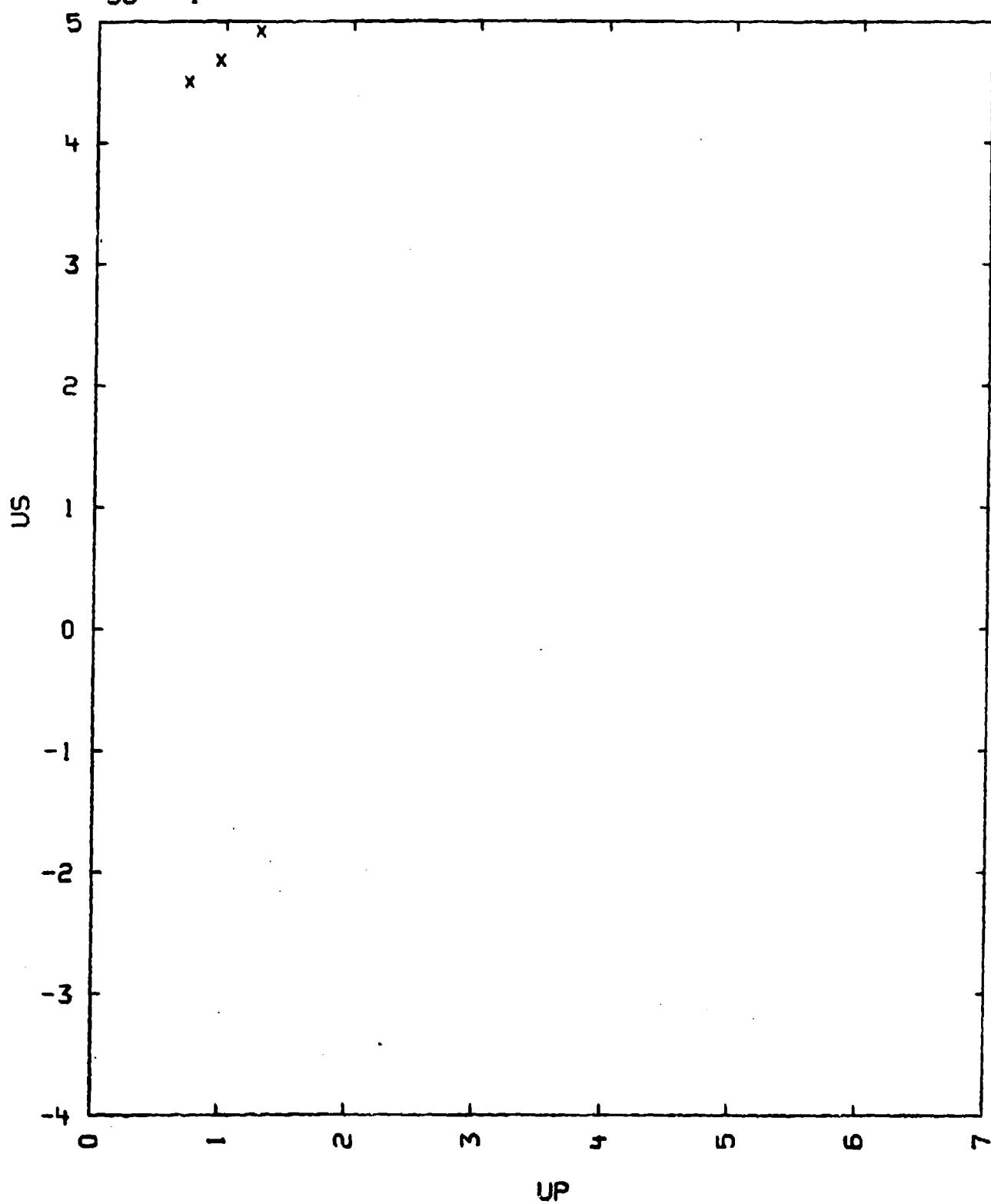
$$US = 3.953 + 0.7570 \text{ UP KM/SEC} \quad \text{SIGMA US} = 0.1 \text{ PERCENT}$$

COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. C. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 3 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS.

TABLE I

ZIRCONIUM
58---1



58---2
ZIRCONIUM

ZR 99.5 PERCENT OR GREATER

$V_0 = 0.1537 \text{ CC/G.}$ $C_L = 4.76 \text{ KM/SEC.}$ $C_0 = 3.88 \text{ KM/SEC.}$
 $V_{01} = 0.1542 \text{ CC/G.}$ $C_S = 2.39 \text{ KM/SEC.}$ $C_B = 3.74 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITIES IN G/CC.

TABLE

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	RHO0	US
6.506	4.440	0.634	183.1	0.8572	8.932	4.764
6.506	4.609	0.841	252.2	0.8175	8.934	5.024
6.509	4.561	0.850	257.9	0.8176	-	5.040
6.503	4.804	1.168	364.9	0.7569	-	5.426
6.503	4.938	1.267	407.0	0.7434	-	5.554
6.505	5.720	1.911	711.1	0.6659	-	6.387
6.512	5.758	1.917	718.8	0.6671	-	6.400
6.505	5.753	1.956	732.0	0.6600	-	6.443
6.506	6.467	2.476	1041.8	0.6171	-	7.133
6.506	6.867	2.732	1220.6	0.6022	-	7.482
6.505	7.211	3.065	1437.7	0.5750	-	7.910

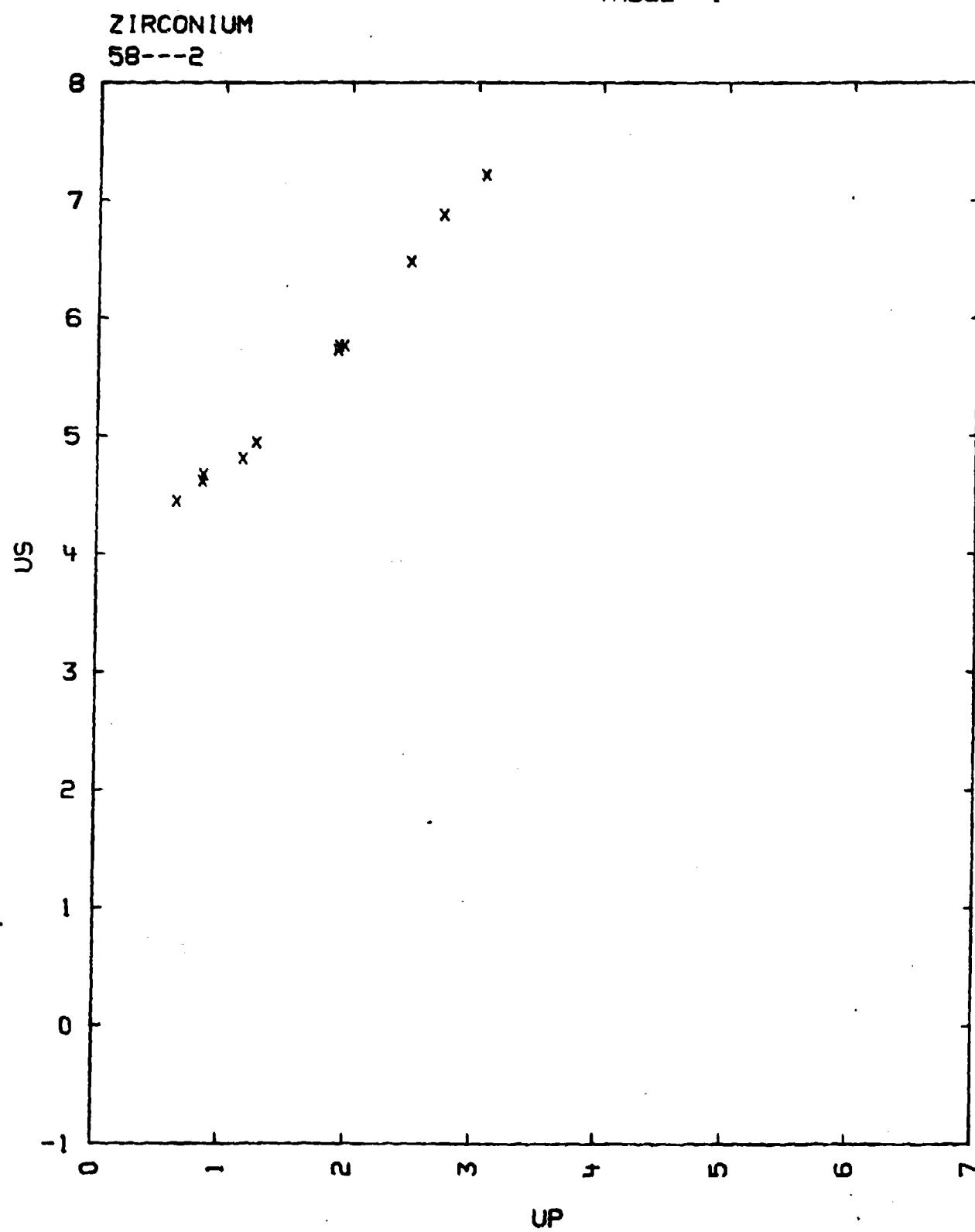
$US = 3.812 + 0.9767 UP \text{ KM/SEC.}$ FOR BELOW 257.9 KILOBARS

$US = 3.287 + 1.287 UP \text{ KM/SEC.}$ FOR 364.9 KILOBARS AND ABOVE

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
REPORT NO.: GMX-6-566, PP. 51-62 (1964)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS COPPER
- 3) THE COPPER HUGONIOT IS REPRESENTED BY
 $US = 3.958 + 1.497 UP \text{ KM/SEC.}$
THE PARTIAL $(\partial E / \partial P)V = 0.0562 \text{ CUBIC CM.}$
- 4) A TRANSITION APPEARS BETWEEN 257.9 AND 364.9 KILOBARS
- 5) V_{01} WAS TAKEN FROM THE AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCGRAH-HILL BOOK CO. N.Y., 1963) 2ND ED.

TABLE I



58---3
ZIRCONIUM

ZR

$V_0 = 0.1536 - 0.1541 \text{ CC/G}$ $C_L = 4.77 \text{ KM/SEC}$ $C_0 = 3.89 \text{ KM/SEC}$
 $V_{01} = 0.1537 \text{ CC/G}$ $C_S = 2.39 \text{ KM/SEC}$ $C_B = 3.74 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
6.510	4.34	0.53	150.	0.8779	CU	4.61
6.510	4.28	0.54	150.	0.8738	CU	4.62
6.506	4.38	0.60	171.	0.8630	CU	4.70
6.510	4.38	0.61	174.	0.8607	CU	4.71
6.488	4.44	0.68	196.	0.8468	2024 AL	6.49
6.510	4.51	0.74	217.	0.8359	CU	4.87
6.510	4.52	0.74	218.	0.8363	CU	4.88
6.506	4.54	0.80	236.	0.8238	CU	4.95
6.509	4.59	0.81	242.	0.8235	CU	4.97
6.488	4.62	0.93	279.	0.7987	2024 AL	6.88
6.510	4.73	1.11	342.	0.7653	CU	5.33
6.503	4.73	1.13	348.	0.7611	CU	5.35
6.510	4.86	1.21	383.	0.7510	CU	5.45
6.505	4.87	1.22	386.	0.7495	CU	5.47
6.488	4.86	1.24	391.	0.7449	2024 AL	7.38
6.505	5.64	1.86	682.	0.6702	CU	6.29
6.512	5.67	1.87	690.	0.6702	CU	6.31
6.505	5.67	1.91	704.	0.6631	CU	6.35
6.509	5.97	2.10	816.	0.6482	CU	6.62
6.510	6.11	2.23	887.	0.6350	CU	6.77
6.506	6.37	2.42	1003.	0.6201	CU	7.03
6.510	6.52	2.57	1091.	0.6058	CU	7.22
6.510	6.73	2.66	1165.	0.6048	CU	7.35
6.506	6.77	2.68	1180.	0.6041	CU	7.37
6.505	7.11	3.01	1392.	0.5767	CU	7.80

$US = 3.757 + 1.018 \cdot UP \text{ KM/SEC FOR UP LESS THAN } 0.8 \text{ KM/SEC}$

SIGMA US = 0.023 KM/SEC

$US = 3.411 + 1.222 \cdot UP \text{ KM/SEC FOR UP ABOVE } 1.0 \text{ KM/SEC}$

SIGMA US = 0.07 KM/SEC

COMMENTS:

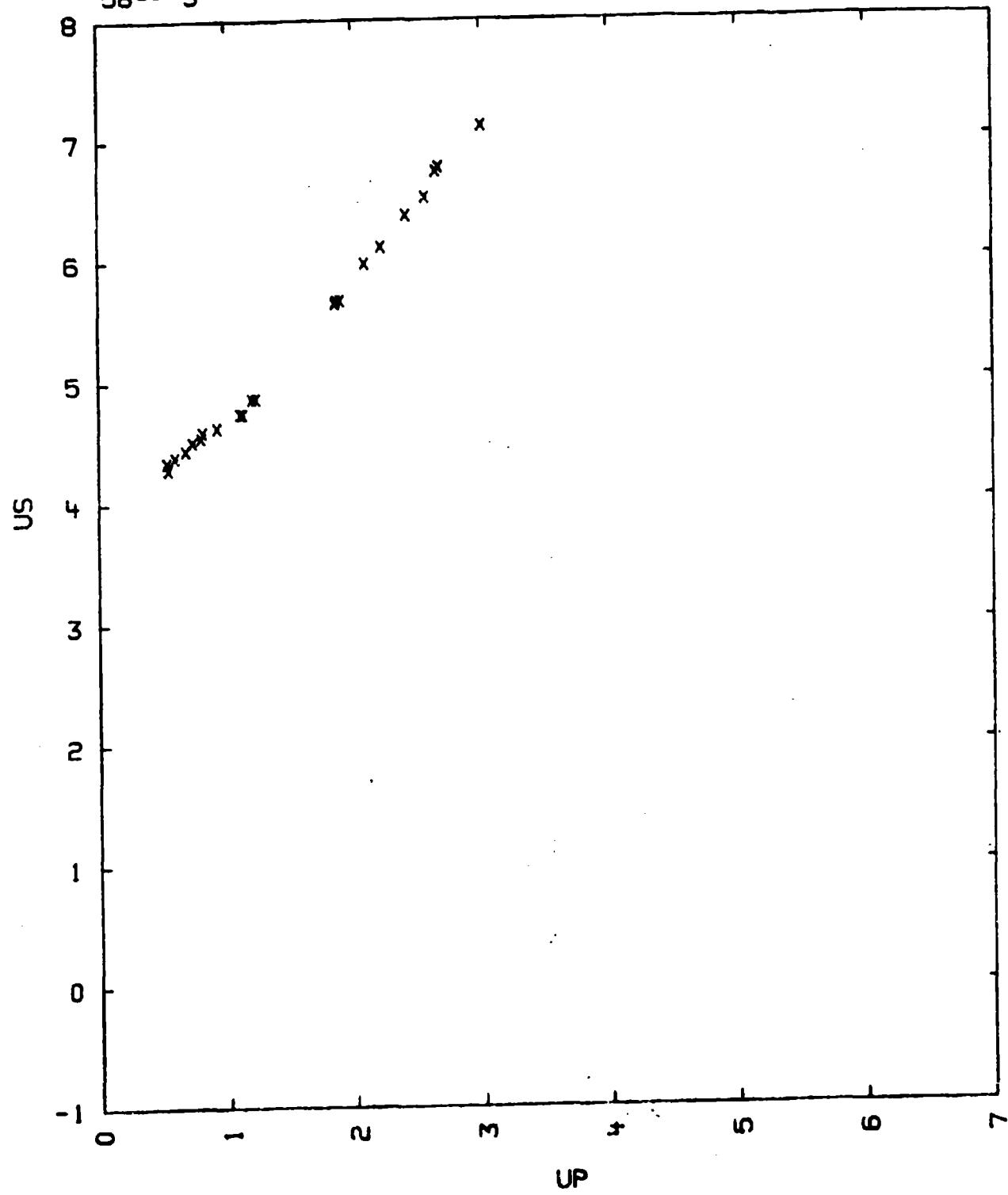
- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M.,
AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES.

HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII

- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B
- 3) VOI FROM WYCKOFF, CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, N.Y.
(1963) V. 1
- 4) TRANSITION AT 260 KBAR PROBABLY FCC-BCC
- 5) $V(DP/DE) = 1.09$

TABLE I

ZIRCONIUM
5B---3



38--4
ZIRCONIUM

ZR

$$V_0 = 0.154 \text{ CC/G}$$

$$V_{01} = 0.1537 \text{ CC/G}$$

$$C_0 = 3.70$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
6.51	9.87	5.04	3238.	0.489	IRON	7.85	11.21
-	7.58	3.47	1735.	0.548	IRON	7.85	8.83
-	6.21	2.26	913.	0.636	ALUMINUM	2.71	9.17
-	5.77	2.08	782.	0.639			
-	4.78	1.22	380.	0.745	ALUMINUM	2.71	7.34
-	4.28	0.54	150.	0.874	ALUMINUM	2.71	6.21

$$US = 3.79 + 0.877 \cdot UP \text{ KM/SEC FOR UP BELOW 1.58 KM/SEC}$$

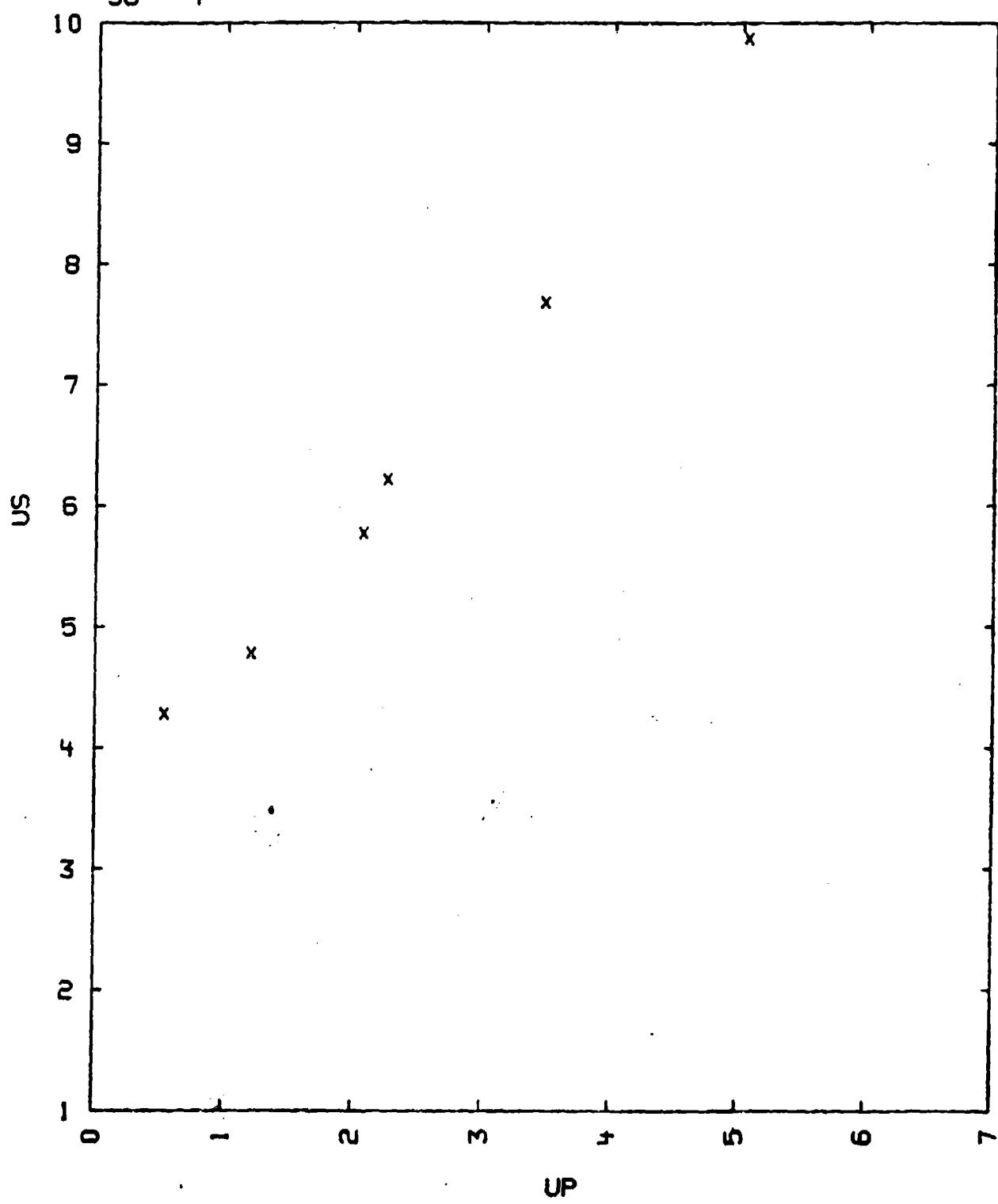
$$US = 3.05 + 1.344 \cdot UP \text{ KM/SEC FOR UP BETWEEN 1.58 AND 5.04 KM/SEC}$$

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P. ZHURNAL EKSP- TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES. THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE 1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED HERE AS FOLLOWS
ALUMINUM US = 5.23 + 1.39 · UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
IRON US = 3.85 + 1.615 · UP FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF V01 WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS OF THE HEXAGONAL CLOSE-PACKED STRUCTURE.
 $A = 3.232 \text{ ANGSTROMS}, C = 5.147 \text{ ANGSTROMS (23 DEG. C)}$
WYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 11.
- 6) THE AUTHORS OBTAINED THE VALUE OF C0 FROM HANDBOOK OF RARE METALS. (MIR PUBLISHERS, USSR, 1983) OR FROM C. KITTEL, INTRODUCTION TO SOLID STATE PHYSICS (FIZMATOIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT UP = 1.58 KM/SEC IS ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I

ZIRCONIUM
58---4



59---1
HAFNIUM

HF 97 WT PERCENT
ZR 3 WT PERCENT

$V_0 = 0.07792 \text{ CC/G}$ $C_L = 3.87 \text{ KM/SEC}$ $C_0 = 3.03 \text{ KM/SEC}$
 $V_{01} = 0.075515 \text{ CC/G}$ $C_S = 2.08 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC. PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE							-- STANDARD --	
RHO0	US	UFS	UP	P	V/V0	MATERIAL	P	
12.843	3.35	0.64	0.37	158	0.891	BRASS	148	
12.849	3.73	2.33	1.23	590	0.870	BRASS	597	
12.846	5.51	4.76	2.36	1667	0.573	BRASS	1543	
12.830	3.68	1.51	0.70	332	0.809	ALUMINUM	221	
12.824	4.26	2.91	1.35	738	0.683	ALUMINUM	472	
12.828	4.76	3.61	1.78	1082	0.627	ALUMINUM	688	
12.822	3.88	1.91	0.99	491	0.746	BRASS	462	

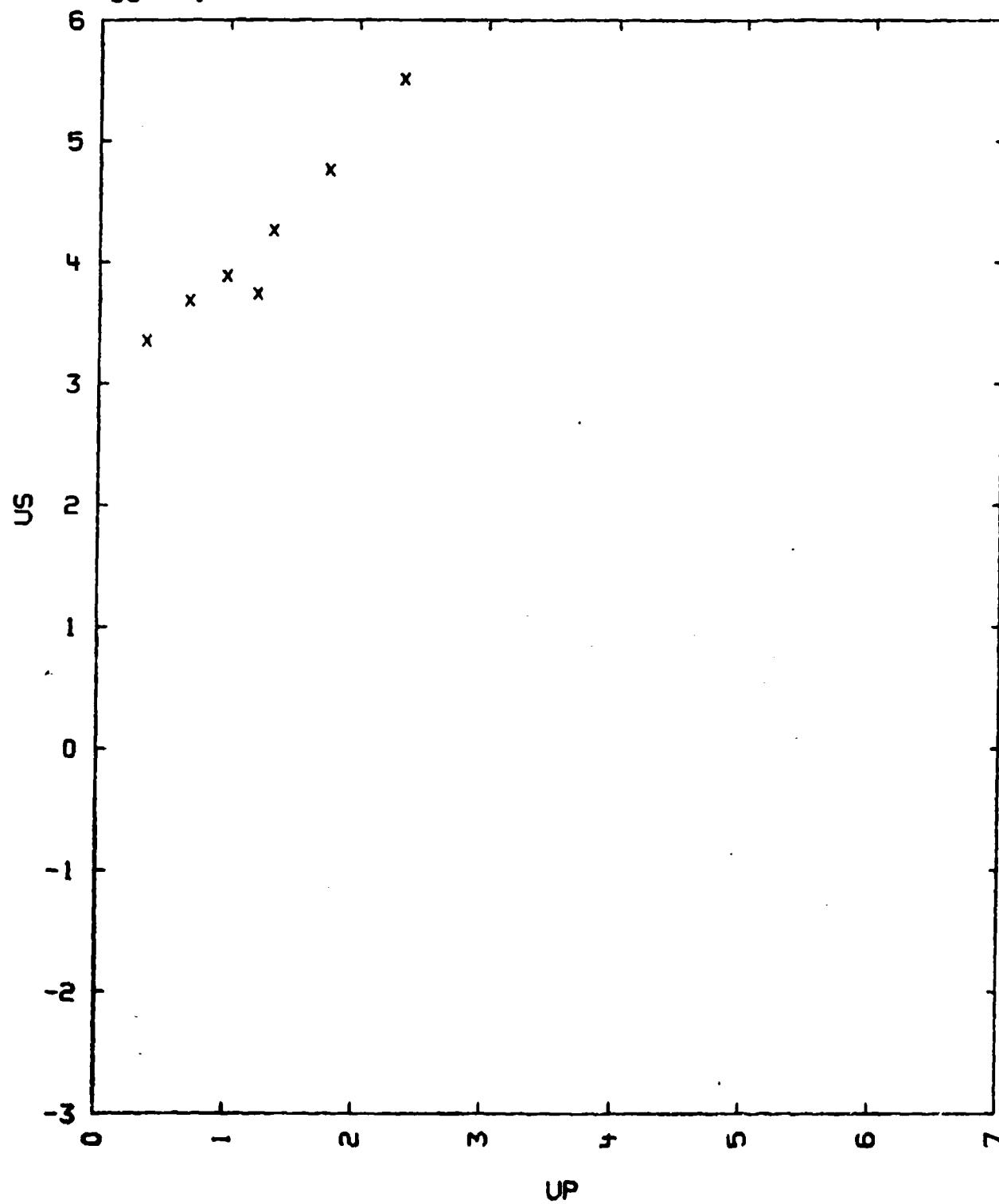
US =

COMMENTS:

- 1) SOURCE: GUST, W. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
BOX 808, LIVERMORE, CALIFORNIA. 94550
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B.
- 3) THIS IS PRELIMINARY DATA AND FURTHER WORK IS IN PROGRESS.
TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE,
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF V_{01} WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL CLOSE-PACKED STRUCTURE.
 $A = 3.1987 \text{ ANGSTROMS}$, $C = 5.0578 \text{ ANGSTROMS}$ (28 DEG. C.)
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. I, P. II.
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA. 94550

TABLE I

HAFNIUM
59---1



59---2
HAFNIUM

HF

$$\begin{array}{lll} V_0 = 0.0775 \text{ CC/G} & C_L = 3.88 \text{ KM/SEC} & C_0 = 2.98 \text{ KM/SEC} \\ V_{01} = 0.07551 \text{ CC/G} & C_S = 2.12 \text{ KM/SEC} & C_B = 2.91 \text{ KM/SEC} \end{array}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
12.870	3.34	0.32	138.	0.9042	CU	4.44
12.900	3.43	0.44	195.	0.8717	CU	4.62
12.890	3.50	0.50	226.	0.8571	CU	4.71
12.900	3.53	0.50	228.	0.8584	CU	4.72
12.890	3.62	0.61	285.	0.8315	CU	4.88
12.880	3.86	0.92	457.	0.7617	CU	5.33
12.900	3.90	0.98	493.	0.7487	CU	5.42
12.900	3.90	1.00	503.	0.7438	CU	5.45
12.890	3.85	1.02	506.	0.7331	CU	5.47
12.900	3.96	1.02	521.	0.7424	CU	5.49
12.890	4.25	1.40	787.	0.6706	CU	6.04
12.830	4.48	1.55	887.	0.6525	CU	6.27
12.890	4.51	1.61	936.	0.6430	CU	6.36
12.910	4.65	1.89	1015.	0.8368	CU	6.49
12.900	4.74	1.77	1082.	0.8266	CU	6.62
12.840	4.88	1.88	1178.	0.6148	CU	6.77
12.920	5.21	2.17	1481.	0.5835	CU	7.22
12.890	5.44	2.23	1584.	0.5901	CU	7.35
12.910	5.42	2.25	1574.	0.5849	CU	7.37
12.890	5.52	2.42	1722.	0.5616	CU	7.61

$$US = 2.954 + 1.121 \cdot UP \text{ KM/SEC BELOW UP} = 0.8$$

$$\text{SIGMA US} = 0.021 \text{ KM/SEC}$$

$$US = 2.421 + 1.311 \cdot UP \text{ KM/SEC FOR UP ABOVE 182 KM/SEC}$$

$$\text{SIGMA US} = 0.051 \text{ KM/SEC}$$

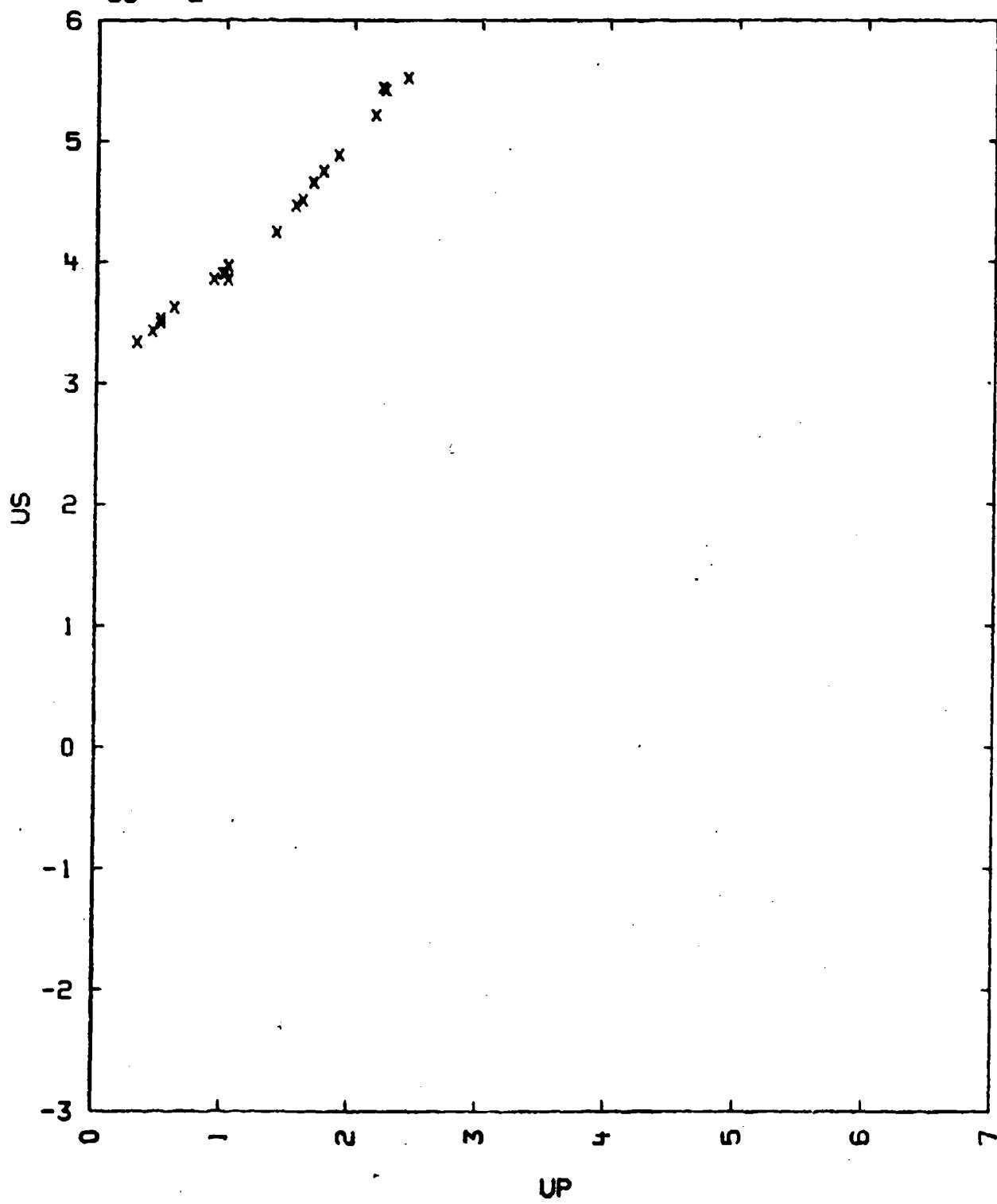
COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.H., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B
- 3) VOI FROM WYCKOFF, CRYSTAL STRUCTURES (JOHN WILEY & SONS, N.Y., 1963)

- V.I
4) TRANSITION AT 400 KBAR PROBABLY FCC-BCC
5) $V(DP/DE) = 0.98$

TABLE I

HAFNIUM
59---2



60---
SCANDIUM

SC

$V_0 = 0.3320 \text{ CC/G}$ $C_L = 5.69 \text{ KM/SEC}$ $C_0 = 4.28 \text{ KM/SEC}$
 $V_{01} = 0.33494 \text{ CC/G}$ $C_S = 3.25 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

SAMPLE						-- STANDARD --	
RHO0	US	UFS	UP	P	V/V0	MATERIAL	P
3.004	5.03	0.96	0.58	87	0.885	BRASS	147
3.010	5.99	3.10	1.68	301	0.720	BRASS	565
3.022	5.65	2.49	1.29	221	0.771	ALUMINUM	238
3.001	7.34	6.67	3.36	744	0.544	BRASS	1480
3.035	6.69	5.06	2.50	508	0.628	ALUMINUM	556
2.996	4.92	0.98	0.54	79	0.891	ALUMINUM	85

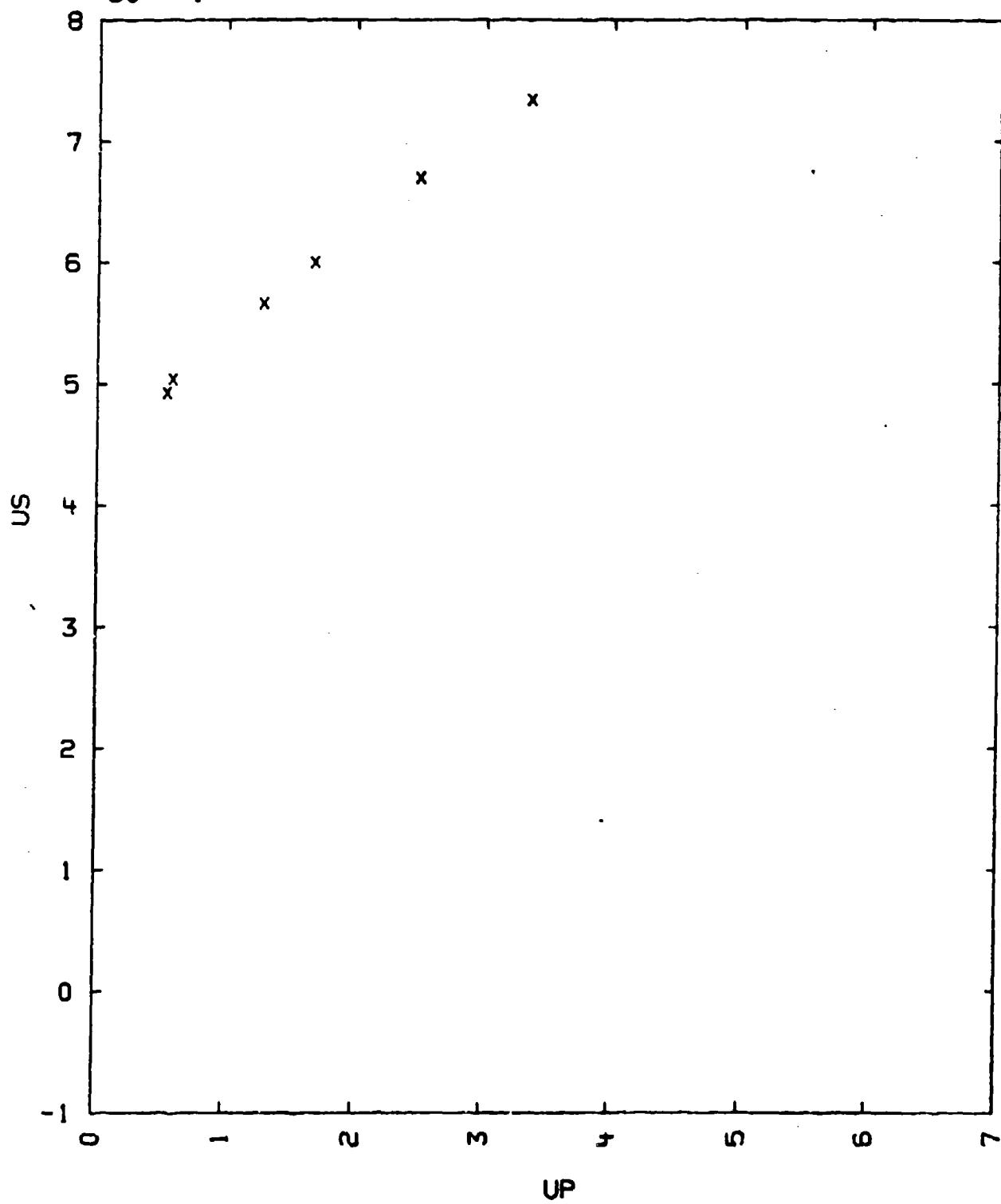
US *

COMMENTS:

- 1) SOURCE: GUST, W. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
BOX 808, LIVERMORE, CALIFORNIA. 94550
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THIS IS PRELIMINARY DATA AND FURTHER WORK IS IN PROGRESS.
TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE,
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF V_{01} WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL CLOSE-PACKED STRUCTURE.
 $A = 3.3090 \text{ ANGSTROMS}$, $C = 5.2733 \text{ ANGSTROMS}$ (CA. 20 DEG. C.)
WYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. 1, P. II.
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA. 94550

TABLE I

SCANDIUM
60---1



60---2
SCANDIUM

SC

$$V_0 = 0.324 \text{ CC/G}$$

$$V_{01} = 0.33494 \text{ CC/G}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
3.09	10.94	6.04	2042.	0.448	IRON	7.83	11.21
-	8.42	4.16	1082.	0.506	IRON	7.83	8.83
-	7.31	2.93	662.	0.599	ALUMINUM	2.71	9.17
-	6.09	1.56	294.	0.744	ALUMINUM	2.71	7.34
-	5.63	0.69	120.	0.877	ALUMINUM	2.71	6.21

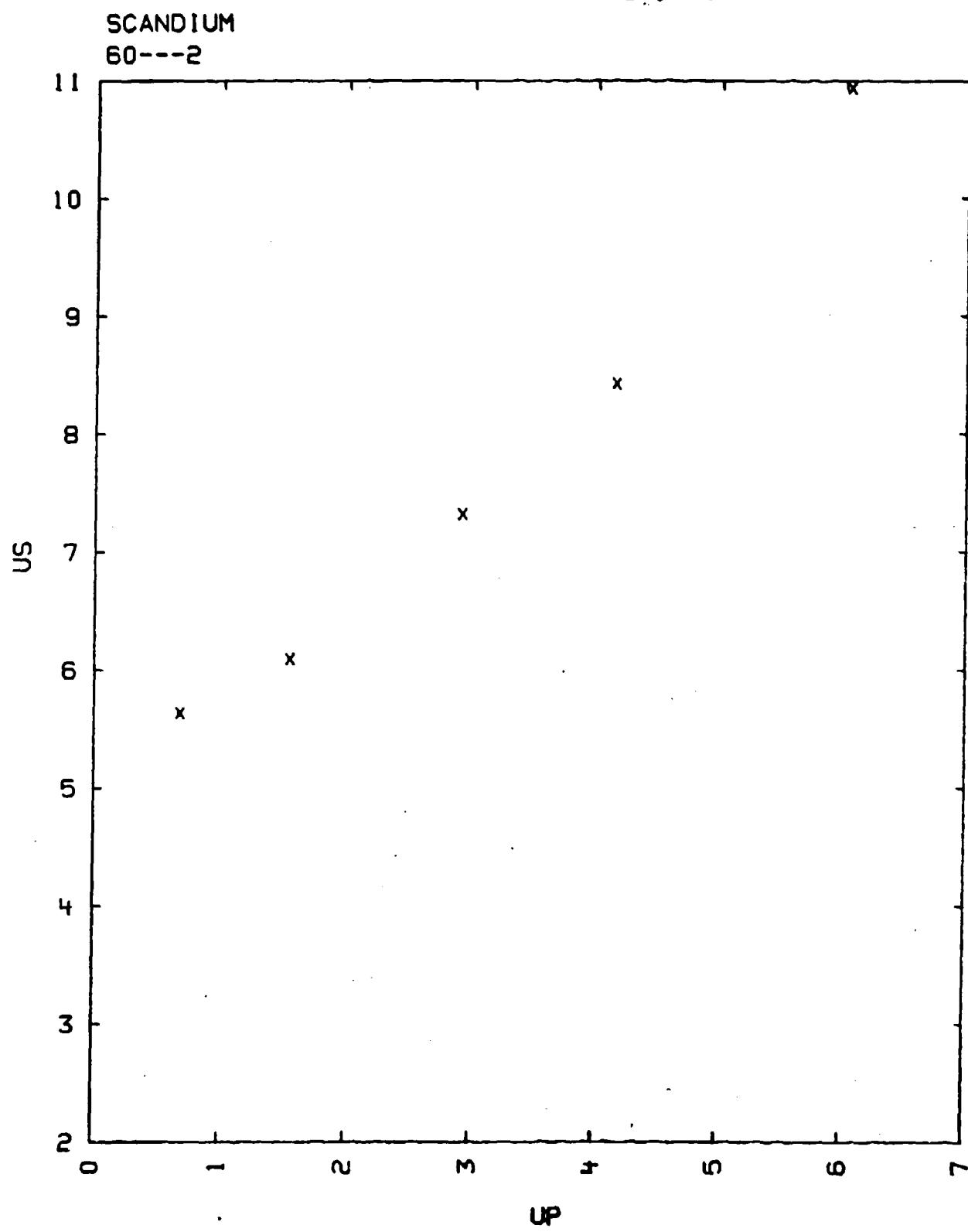
$$US = 5.05 + 0.746 \cdot UP \text{ KM/SEC FOR UP BELOW } 3.72 \text{ KM/SEC}$$

$$US = 2.83 + 1.342 \cdot UP \text{ KM/SEC FOR UP BETWEEN } 3.72 \text{ AND } 6.04 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P.
ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES.
THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE
1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = 5.25 + 1.39 · UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
IRON US = 3.85 + 1.615 · UP FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF V01 WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL CLOSE-PACKED STRUCTURE,
A = 3.3090 ANGSTROMS, C = 5.2733 ANGSTROMS (CA. 20 DEG. C)
WYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 11.
- 6) THE DISCONTINUITY OF THE US VS. UP PLOT AT UP = 3.72 KM/SEC IS
ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I



61---1
YTTRIUM

Y

$$\begin{array}{lll} V_0 = 0.2216 \text{ CC/G} & C_L = 4.24 \text{ KM/SEC} & C_U = 3.32 \text{ KM/SEC} \\ V_{01} = 0.22364 \text{ CC/G} & C_S = 2.28 \text{ KM/SEC} & \end{array}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

----- SAMPLE -----						-- STANDARD --	
RHO0	US	UFS	UP	P	V/V0	MATERIAL	P
4.513	4.16	2.13	1.24	234	0.705	ALUMINUM	238
4.513	4.52	3.03	1.62	329	0.640	BRASS	565
4.512	3.65	0.98	0.56	93	0.847	BRASS	147
4.512	6.37	6.21	3.14	905	0.508	BRASS	1480
4.511	5.26	4.72	2.35	556	0.552	ALUMINUM	556
4.514	4.89	3.77	1.92	423	0.607	ALUMINUM	423

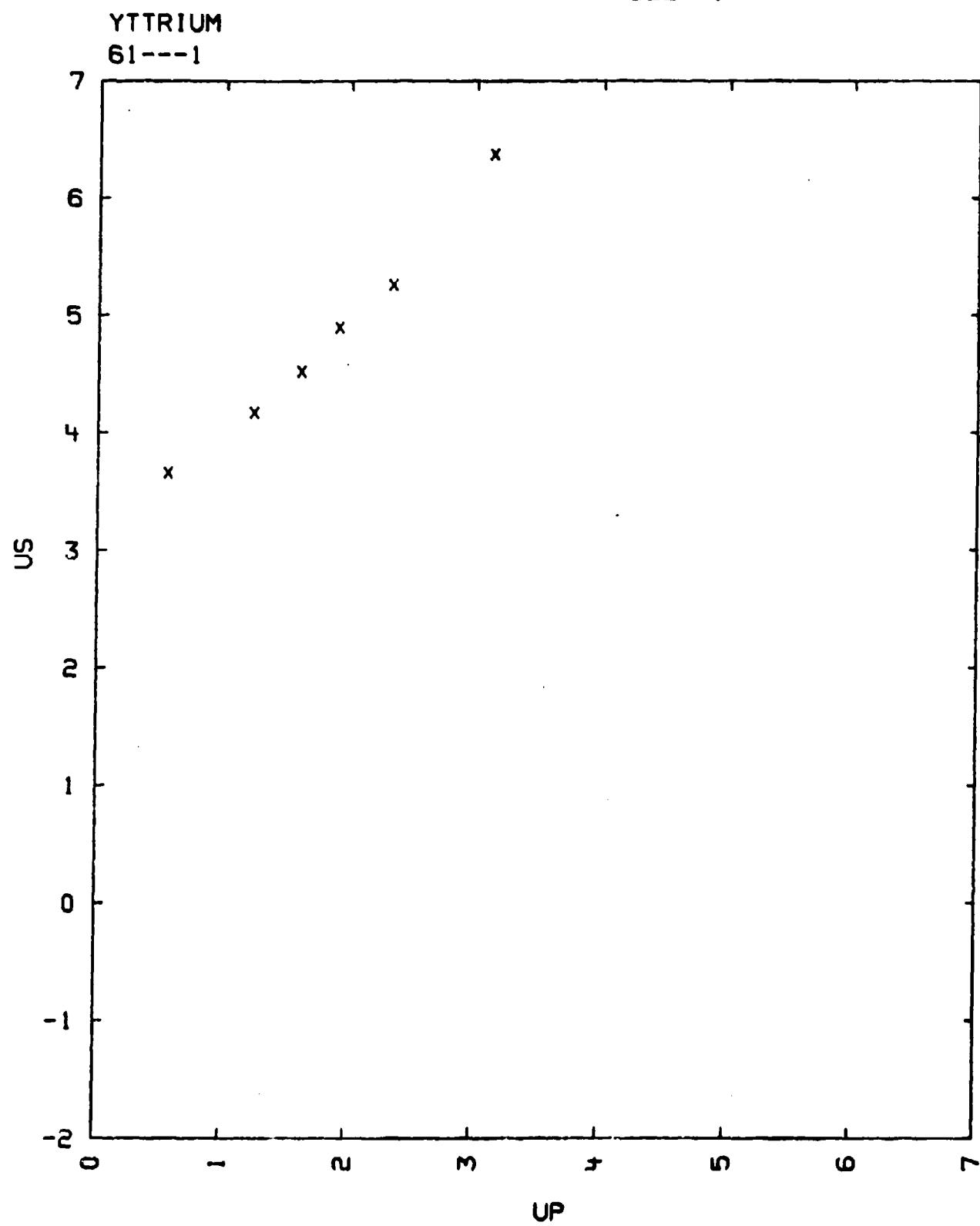
US =

COMMENTS:

- 1) SOURCE: GUST, W. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
BOX 808, LIVERMORE, CALIFORNIA, 94550
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THIS IS PRELIMINARY DATA AND FURTHER WORK IS IN PROGRESS.
TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE,
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF V_{01} WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL CLOSE-PACKED STRUCTURE,
 $A = 3.6474 \text{ ANGSTROMS}$, $C = 5.7308 \text{ ANGSTROMS}$
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. 1, P. 11.
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA, 94550
- 6) THE LONGITUDINAL AND SHEAR VELOCITIES HAVE ALSO BEEN MEASURED BY
SMITH, J. F., CARLSON, C. E. AND SPEDDING, F. H., ELASTIC PROPERTIES
OF YTTRIUM AND ELEVEN OF THE RARE EARTH ELEMENTS, JOURNAL OF METALS
(TRANSACTIONS AIME) OCT. 1957 P. 1212. THE MEASURED VALUES ARE:
 $C_L = 4.280 \text{ PLUS OR MINUS } 0.021 \text{ KM/SEC}$

CS = 2.420 PLUS OR MINUS 0.018 KM/SEC.

TABLE I



61---2
YTTRIUM

Y

$$V_0 = 0.223 \text{ CC/G}$$

$$V_{01} = 0.22364 \text{ CC/G}$$

$$C_0 = 3.23 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
4.49	10.13	5.59	2542.	0.448	IRON	7.85	11.21
-	7.45	3.92	1311.	0.474	IRON	7.85	8.83
-	5.79	2.71	705.	0.532			
-	5.82	2.77	724.	0.524	ALUMINUM	2.71	9.17
-	5.36	2.37	570.	0.558			
-	4.46	1.52	304.	0.659	ALUMINUM	2.71	7.34
-	3.85	0.69	119.	0.821	ALUMINUM	2.71	6.21

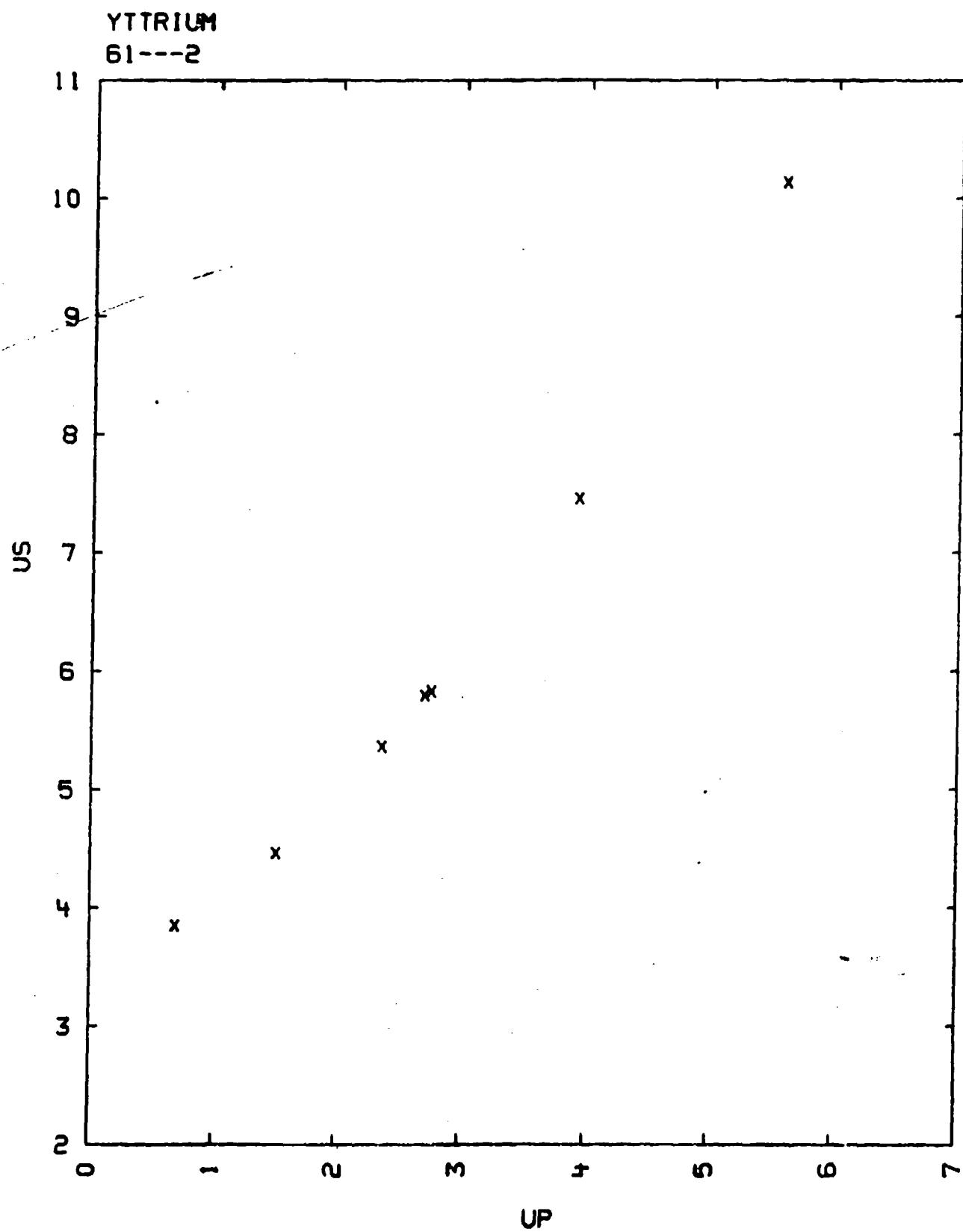
$$US = 3.33 + 0.753 \cdot UP \text{ KM/SEC FOR UP BELOW 2.09 KM/SEC}$$

$$US = 1.86 + 1.456 \cdot UP \text{ KM/SEC FOR UP BETWEEN 2.09 AND 5.59 KM/SEC}$$

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P. ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES. THE UNCERTAINTY IN THE SHOCK VELOCITIES HAS ESTIMATED TO BE 1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED HERE AS FOLLOWS
ALUMINUM US = 5.29 + 1.39 · UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
IRON US = 3.85 + 1.615 · UP FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF V_{01} WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS OF THE HEXAGONAL CLOSE-PACKED STRUCTURE,
 $A = 3.6474 \text{ ANGSTROMS}, C = 5.7306 \text{ ANGSTROMS (CA. } 20 \text{ DEG. C)}$
WYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 11.
- 6) THE AUTHORS OBTAINED THE VALUE OF C_0 FROM HANDBOOK OF RARE METALS. (MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT $UP = 2.09 \text{ KM/SEC}$ IS ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I



62---1
LUTETIUM

LU

$V_0 = 0.1026 \text{ CC/O}$
 $V_{01} = 0.10153 \text{ CC/O}$

$C_0 = 2.13 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

----- SAMPLE -----					----- STANDARD -----		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
9.74	8.56	4.60	3835.	0.463	IRON	7.85	11.21
-	4.53	2.16	953.	0.523	ALUMINUM	2.71	9.17
-	3.38	1.17	385.	0.654	ALUMINUM	2.71	7.34
-	2.76	0.54	145.	0.805	ALUMINUM	2.71	6.21

$US = 2.20 + 1.00 \cdot UP \text{ KM/SEC}$ FOR UP BELOW 1.88 KM/SEC

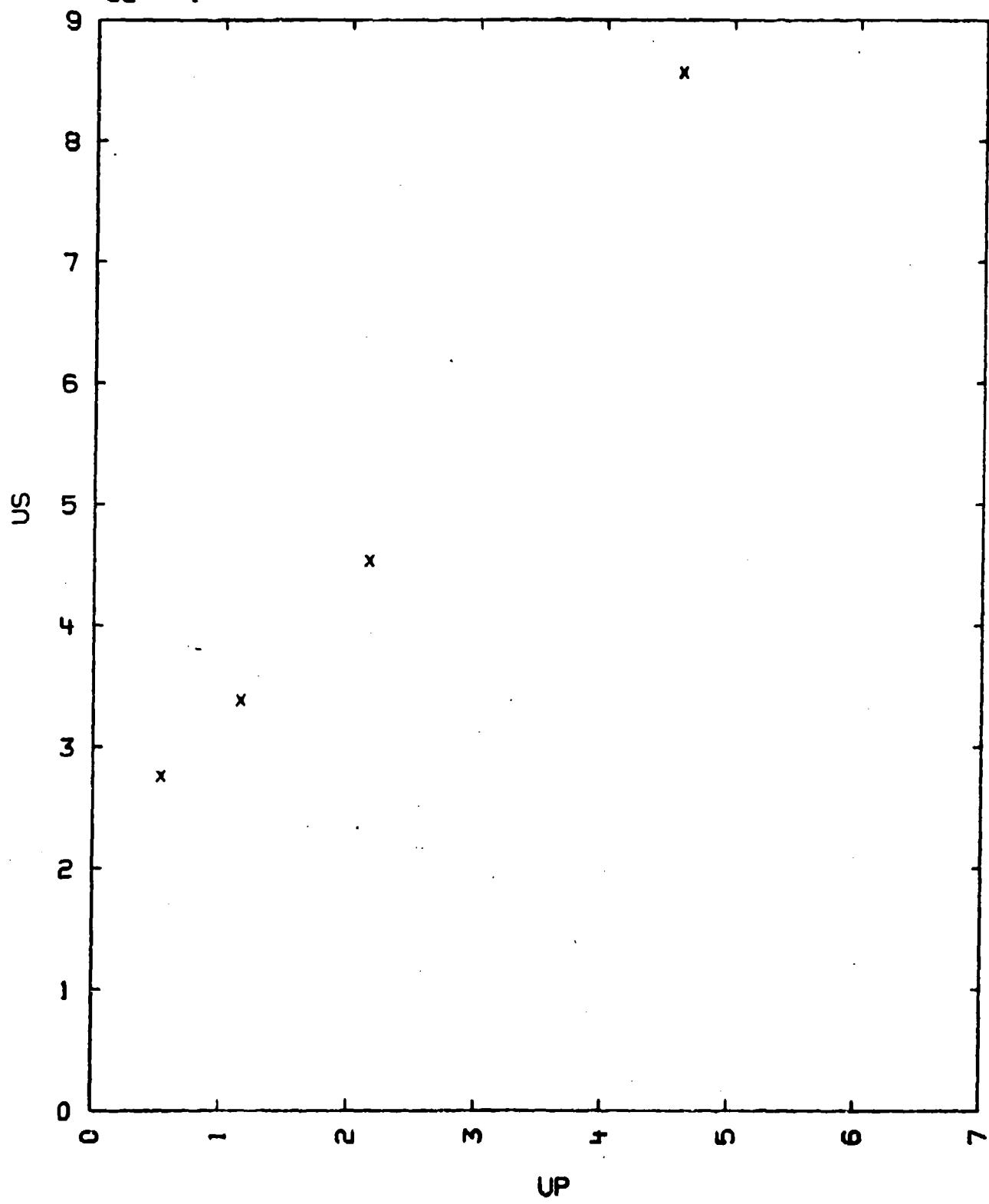
$US = 0.98 + 1.65 \cdot UP \text{ KM/SEC}$ FOR UP BETWEEN 1.88 AND 4.60 KM/SEC

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P.
ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES.
THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE
1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM $US = 5.25 + 1.39 \cdot UP$ FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
IRON $US = 3.85 + 1.615 \cdot UP$ FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF V_{01} WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL CLOSE-PACKED STRUCTURE.
 $A = 3.5031 \text{ ANGSTROMS}, C = 5.5509 \text{ ANGSTROMS}$ (CA. 20 DEG. C)
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 11-
- 6) THE AUTHORS OBTAINED THE VALUE OF C_0 FROM HANDBOOK OF RARE METALS,
(MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO
SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT $UP = 1.88 \text{ KM/SEC}$ IS
ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I

LUTETIUM
62---1



63---1
YTTERBIUM

YB

$$V_0 = 0.1430 \text{ CC/G} \quad C_L = 1.88 \text{ KM/SEC} \quad C_0 = 1.46 \text{ KM/SEC}$$

$$V_{01} = 0.14368 \text{ CC/G} \quad C_S = 1.02 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE						-- STANDARD --	
RHO0	US	UFS	UP	P	V/V0	MATERIAL	P
7.020	1.86	1.02	0.48	62	0.745	BRASS	118
7.021	4.89	5.98	2.99	1024	0.389	BRASS	1479
6.978	2.57	2.40	1.20	214	0.533	ALUMINUM	228
6.980	2.94	3.04	1.50	307	0.490	BRASS	506
6.977	3.90	4.40	2.18	593	0.440	ALUMINUM	537

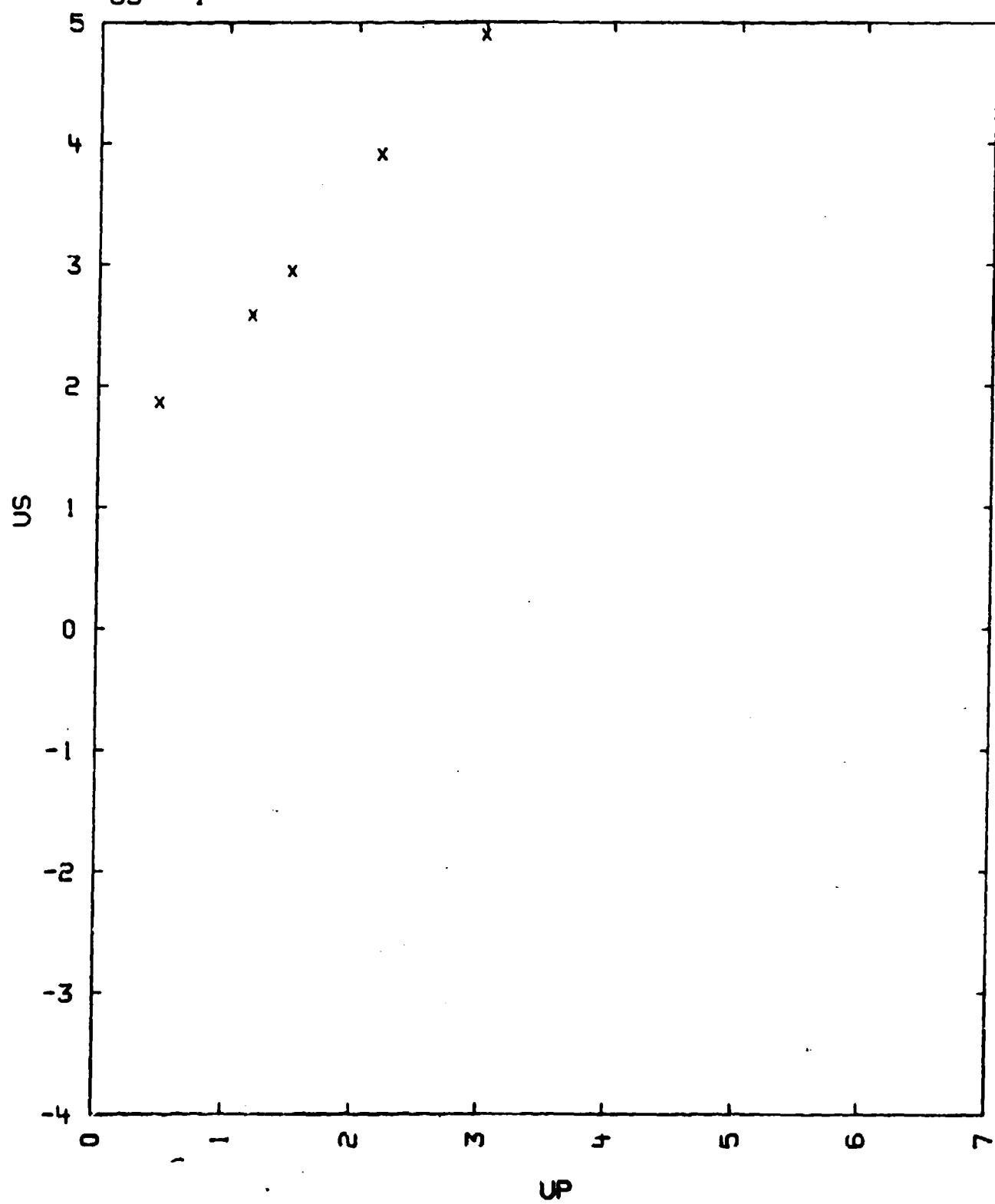
US *

COMMENTS:

- 1) SOURCE: GUST, W. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
BOX 808, LIVERMORE, CALIFORNIA. 94550
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THIS IS PRELIMINARY DATA AND FURTHER WORK IS IN PROGRESS.
TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE,
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF VO1 WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANT
OF THE CUBIC CLOSE-PACKED STRUCTURE.
A = 5.4862 ANGSTROMS
HYCKOFF, R. W. O., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. 1, P. 10.
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA. 94550
- 6) THE LONGITUDINAL AND SHEAR VELOCITIES HAVE ALSO BEEN MEASURED BY
SMITH, J. F., CARLSON, C. E. AND SPEDDING, F. H., ELASTIC PROPERTIES
OF YTTRIUM AND ELEVEN OF THE RARE EARTH ELEMENTS, JOURNAL OF METALS
(TRANSACTIONS AIME) OCT. 1957 P. 1212. THE MEASURED VALUES ARE:
CL = 1.820 PLUS OR MINUS 0.012 KM/SEC
CS = 0.999 PLUS OR MINUS 0.001 KM/SEC.

TABLE I

YTTERBIUM
63---1



65---1
ERBIUM

ER

VO = 0.110 CC/G
VOI = 0.11033 CC/G

CO = 2.26 KM/SEC

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC. PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO0	US	UP	P	V/VO	MATERIAL	RHO0(ST)	US(ST)
9.05	8.22	4.79	3563.	0.417	IRON	7.85	11.21
-	6.01	3.37	1833.	0.439	IRON	7.85	8.83
-	4.56	2.23	920.	0.511	ALUMINUM	2.71	9.17
-	4.20	1.96	745.	0.533	COPPER	8.93	6.45
-	3.44	1.21	377.	0.648	ALUMINUM	2.71	7.34
-	2.84	0.56	144.	0.803	ALUMINUM	2.71	6.21

US = 2.20 + 1.033*UP KM/SEC FOR UP BELOW 2.71 KM/SEC

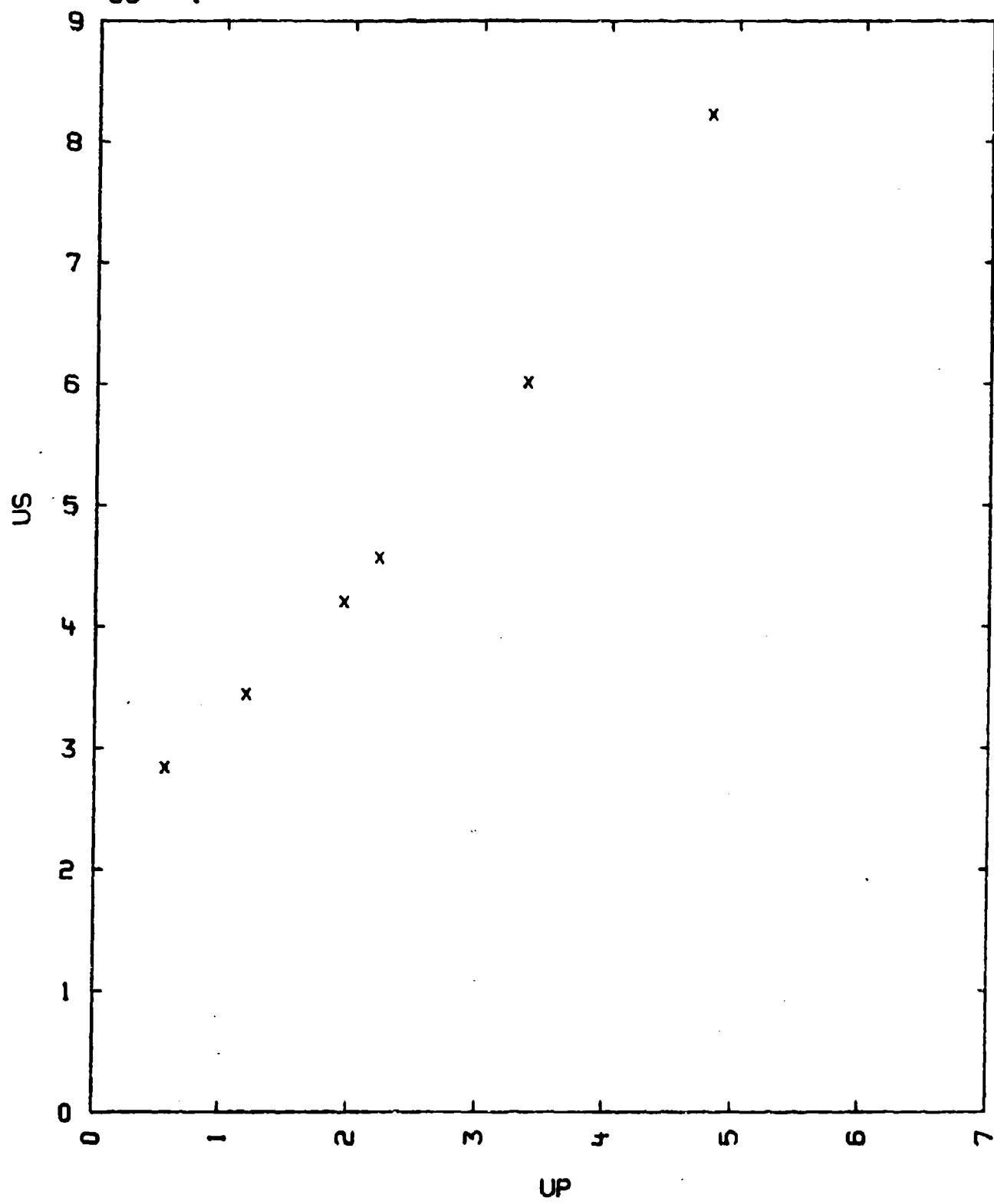
US = 0.77 + 1.560*UP KM/SEC FOR UP BETWEEN 2.71 AND 4.79 KM/SEC

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P.
ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES.
THE UNCERTAINTY IN THE SHOCK VELOCITIES HAS ESTIMATED TO BE
1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = 5.25 + 1.39*UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
COPPER US = 3.95 + 1.50*UP FOR UP BETWEEN 0.0 AND 4.0 KM/SEC
IRON US = 3.85 + 1.815*UP FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF VOI WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL CLOSE-PACKED STRUCTURE.
A = 3.5588 ANGSTROMS, C = 5.5874 ANGSTROMS (CA. 20 DEG. C)
WYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 11.
- 6) THE AUTHORS OBTAINED THE VALUE OF CO FROM HANDBOOK OF RARE METALS.
(MIR PUBLISHERS, USSR, 1983) OR FROM C. KITTEL, INTRODUCTION TO
SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT UP = 2.71 KM/SEC IS
ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I

ERBIUM
65---1



67---1
DYSPROSIUM

DY

$V_0 = 0.1168 \text{ CC/G}$ $C_L = 2.95 \text{ KM/SEC}$ $C_D = 2.19 \text{ KM/SEC}$
 $V_{01} = 0.11683 \text{ CC/G}$ $C_S = 1.72 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS,
AND DENSITY IN G/CC.

TABLE

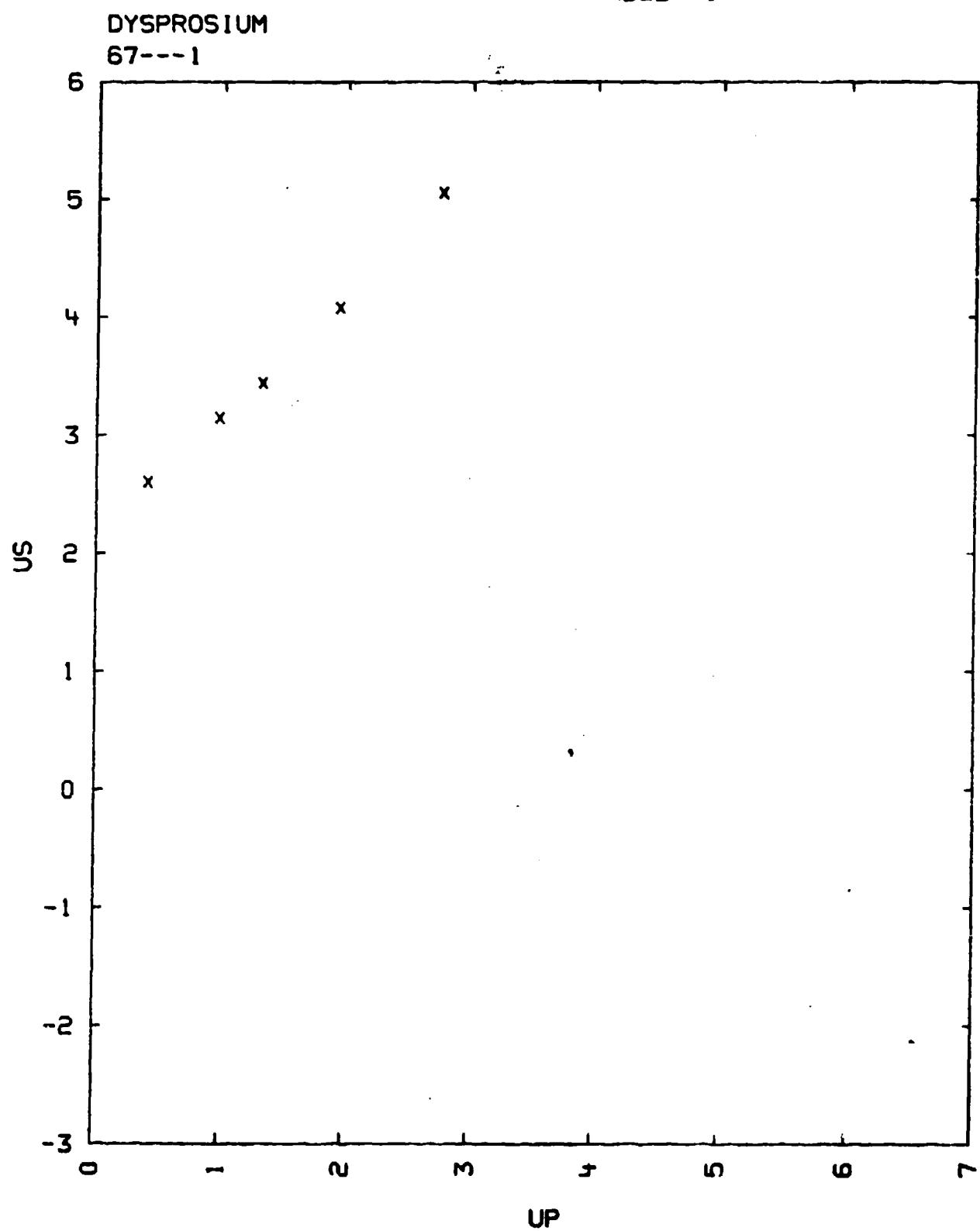
SAMPLE						-- STANDARD --	
RHO0	US	UFS	UP	P	V/V0	MATERIAL	P
8.563	2.60	0.85	0.41	92	0.841	BRASS	118
8.557	5.05	5.66	2.77	1195	0.453	BRASS	1479
8.556	3.14	2.05	0.98	264	0.688	ALUMINUM	228
8.560	3.44	2.71	1.33	391	0.615	BRASS	506
8.559	4.07	3.91	1.95	679	0.522	ALUMINUM	537

US *

COMMENTS:

- 1) SOURCE: GUST, W. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
BOX 808, LIVERMORE, CALIFORNIA, 94550
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THIS IS PRELIMINARY DATA AND FURTHER WORK IS IN PROGRESS.
TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF V_{01} WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL CLOSE-PACKED STRUCTURE,
 $A = 3.5903 \text{ ANGSTROMS}$, $C = 5.6475 \text{ ANGSTROMS}$ (CA. 20 DEG. C.)
WYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. 1, P. 11.
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA, 94550
- 6) THE LONGITUDINAL AND SHEAR VELOCITIES HAVE ALSO BEEN MEASURED BY
SMITH, J.F., CARLSON, C. E. AND SPEEDING, F. H., ELASTIC PROPERTIES
OF YTTRIUM AND ELEVEN OF THE RARE EARTH ELEMENTS, JOURNAL OF METALS
(TRANSACTIONS AIME) OCT. 1957 P. 1212. THE MEASURED VALUES ARE:
 $C_L = 2.958 \text{ PLUS OR MINUS } 0.008 \text{ KM/SEC}$
 $C_S = 1.724 \text{ PLUS OR MINUS } 0.009 \text{ KM/SEC}$

TABLE I



67---2
DYSPROSIUM

DY

$$V_0 = 0.121 \text{ CC/G}$$

$$V_{01} = 0.11683 \text{ CC/G}$$

$$C_0 = 2.19 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

----- SAMPLE -----					----- STANDARD -----		
RHO0	US	UP	P	V/V0	MATERIAL	RHO(ST)	US(ST)
8.52	8.22	4.89	3425.	0.405	IRON	7.85	11.21
-	6.20	3.40	1796.	0.452	IRON	7.85	8.83
-	4.56	2.30	893.	0.496	ALUMINUM	2.71	9.17
-	4.23	1.99	717.	0.530	COPPER	8.93	6.45
-	3.57	1.47	447.	0.588			
-	3.38	1.26	362.8	0.627	ALUMINUM	2.71	7.34
-	2.85	0.57	138.4	0.800	ALUMINUM	2.71	6.21

$$US = 2.37 + 0.81 \cdot UP \text{ KM/SEC FOR UP BELOW 1.70 KM/SEC}$$

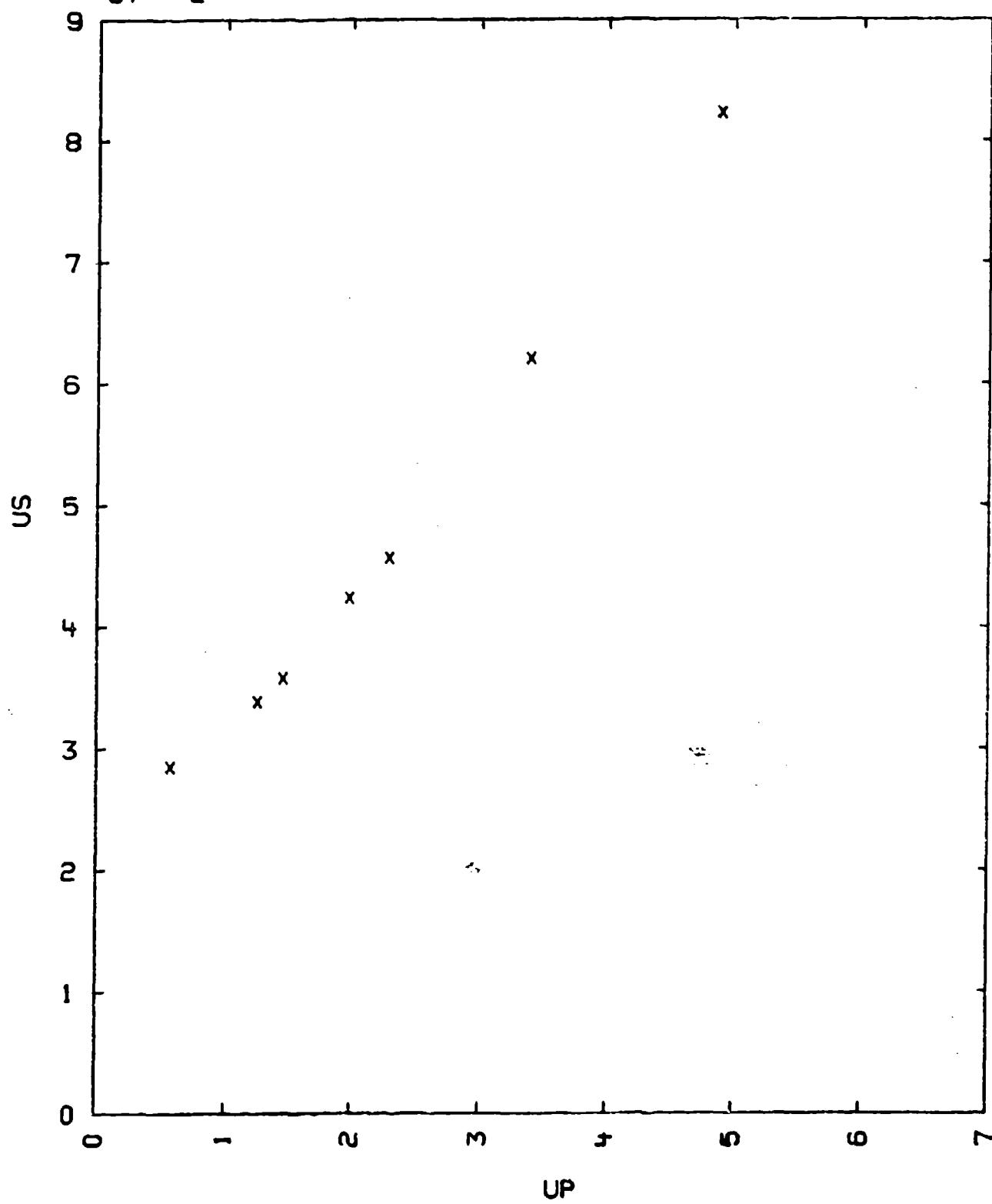
$$US = 1.34 + 1.417 \cdot UP \text{ KM/SEC FOR UP BETWEEN 1.70 AND 4.89 KM/SEC}$$

COMMENTS:

- 1) SOURCE: ALTMULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P. ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES. THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE 1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = 5.25 + 1.39 · UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
COPPER US = 3.95 + 1.50 · UP FOR UP BETWEEN 0.0 AND 4.0 KM/SEC
IRON US = 3.65 + 1.615 · UP FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF VOI WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS OF THE HEXAGONAL CLOSE-PACKED STRUCTURE.
A = 3.5903 ANGSTROMS, C = 3.6475 ANGSTROMS (CA. 20 DEG. C)
WYCKOFF, R. W. B., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 11.
- 6) THE AUTHORS OBTAINED THE VALUE OF CO FROM HANDBOOK OF RARE METALS, (MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT UP = 1.70 KM/SEC IS ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I

DYSPROSIUM
67---2



69---1
GADOLINIUM

GD

$$V_0 = 0.1264 \text{ CC/G} \quad C_L = 2.94 \text{ KM/SEC} \quad C_0 = 2.21 \text{ KM/SEC}$$

$$V_{01} = 0.12678 \text{ CC/G} \quad C_S = 1.68 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE						-- STANDARD --	
RHO0	US	UFS	UP	P	V/V0	MATERIAL	P
7.937	2.67	0.93	0.36	100	0.819	BRASS	139
7.933	3.60	2.82	1.42	405	0.605	BRASS	547
7.941	5.26	5.69	2.76	1151	0.476	BRASS	1433
7.892	3.15	2.06	1.05	261	0.665	ALUMINUM	230
7.891	4.31	3.99	1.94	657	0.551	ALUMINUM	534
7.893	2.86	1.49	0.67	151	0.765	ALUMINUM	135

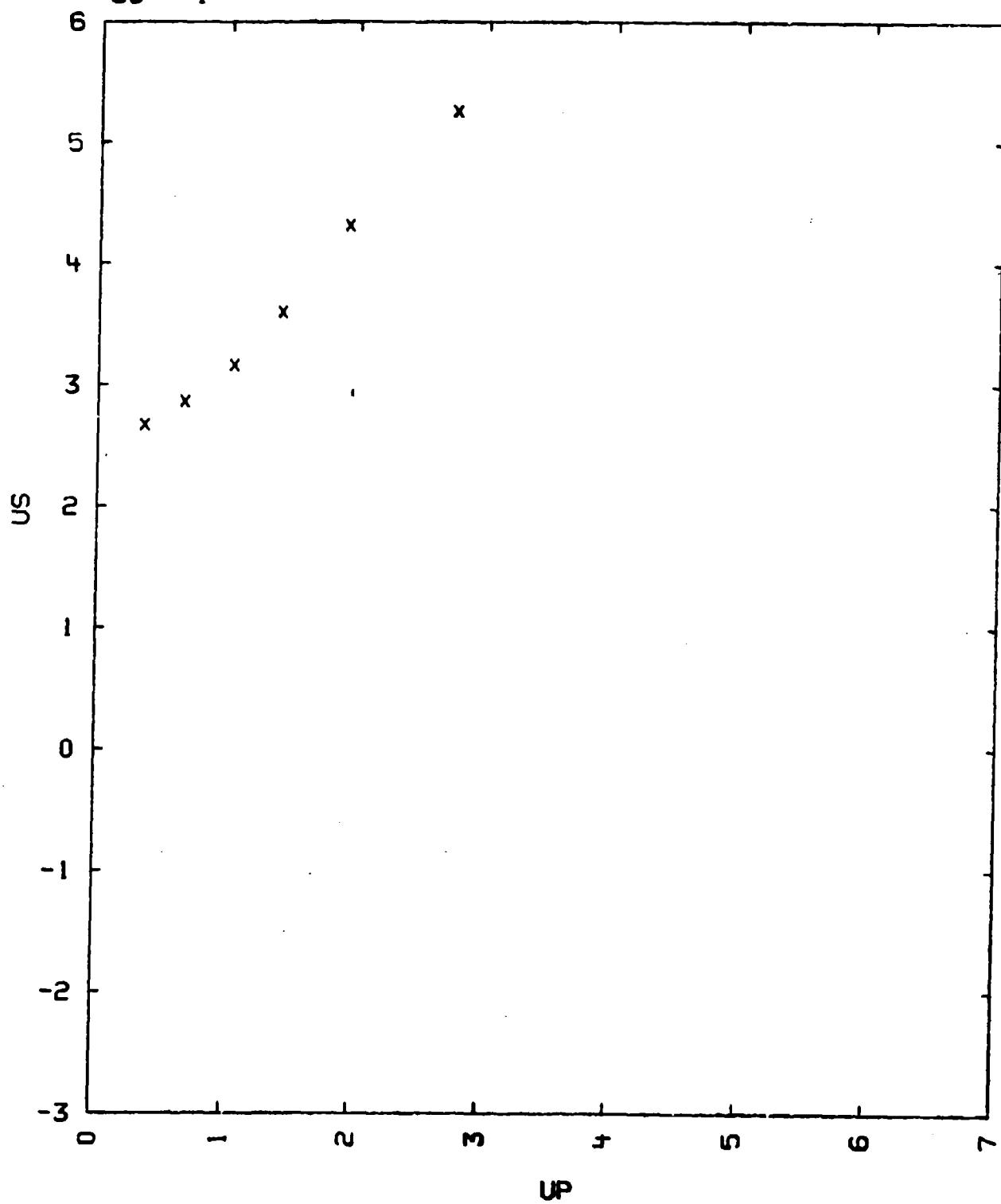
US =

COMMENTS:

- 1) SOURCE: GUST, W. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
BOX 808, LIVERMORE, CALIFORNIA. 94550
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THIS IS PRELIMINARY DATA AND FURTHER WORK IS IN PROGRESS.
TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE,
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF V01 WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL CLOSE-PACKED STRUCTURE.
 $A = 3.8360 \text{ ANGSTROMS}$, $C = 5.7826 \text{ ANGSTROMS}$ (CA. 20 DEG. C.)
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. 1
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA. 94550
- 6) THE LONGITUDINAL AND SHEAR VELOCITIES HAVE ALSO BEEN MEASURED BY
SMITH, J. F., CARLSON, C. E. AND SPEDDING, F. H., ELASTIC PROPERTIES
OF YTTRIUM AND ELEVEN OF THE RARE EARTH ELEMENTS, JOURNAL OF METALS
(TRANSACTIONS AIME) OCT. 1957 P. 1212. THE MEASURED VALUES ARE:
 $C_L = 2.951 \pm 0.003 \text{ KM/SEC}$
 $C_S = 1.682 \pm 0.002 \text{ KM/SEC}$.

TABLE I

GADOLINIUM
69---1



69---2
GADOLINIUM

00

$V_0 = 0.128 \text{ CC/G}$
 $V_{01} = 0.1268 \text{ CC/G}$

$C_0 = 2.25 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
7.93	8.29	5.00	3287.	0.397	IRON	7.85	11.21
-	6.26	3.48	1728.	0.444	IRON	7.85	8.83
-	4.79	2.33	885.	0.514	ALUMINUM	2.71	9.17
-	3.51	1.28	356.	0.635	ALUMINUM	2.71	7.34
-	2.83	0.60	135.	0.788	ALUMINUM	2.71	6.21

$US = 2.25 + 0.98 \cdot UP \text{ KM/SEC}$ FOR UP BELOW 1.74 KM/SEC

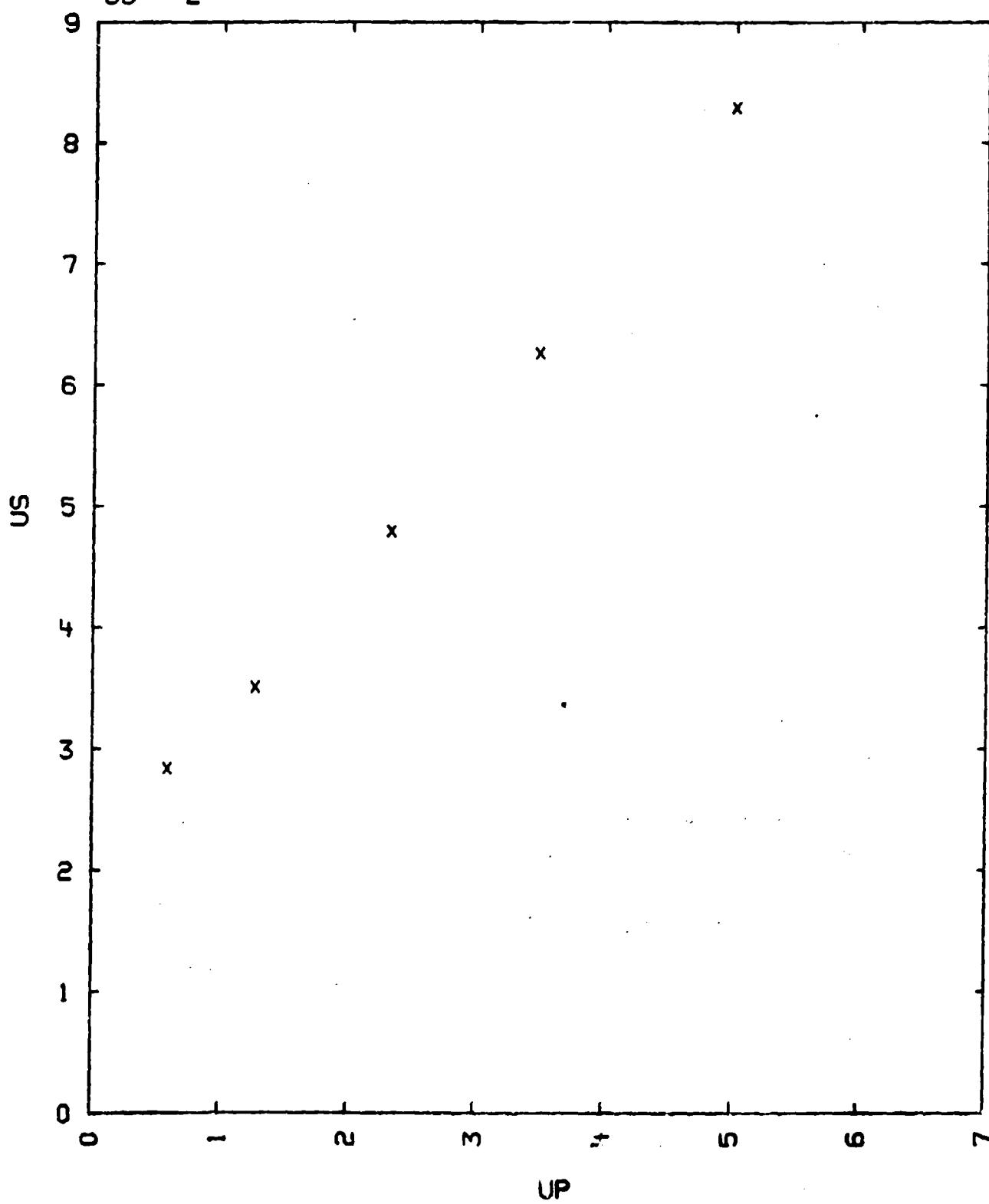
$US = 1.650 + 1.325 \cdot UP \text{ KM/SEC}$ FOR UP BETWEEN 1.74 AND 5.00 KM/SEC

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P.
ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES.
THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE
1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = $3.25 + 1.39 \cdot UP$ FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
IRON US = $3.85 + 1.615 \cdot UP$ FOR UP BETWEEN 1.0 AND 4.8 KM/SEC
- 5) THE VALUE OF V_{01} WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL CLOSE-PACKED STRUCTURE.
 $A = 3.636 \text{ ANGSTROMS}, C = 5.7826 \text{ ANGSTROMS (CA. } 20 \text{ DEG. C)}$
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 11.
- 6) THE AUTHORS OBTAINED THE VALUE OF C_0 FROM HANDBOOK OF RARE METALS.
(MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO
SOLID STATE PHYSICS (FIZMATOIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT $UP = 1.74 \text{ KM/SEC}$ IS
ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I

GADOLINIUM
69---2



70---I
EUROPIUM

EU

$$V_0 = 0.1878 \text{ CC/G} \quad C_L = 2.26 \text{ KM/SEC} \quad C_0 = 1.79 \text{ KM/SEC}$$

$$V_{OI} = 0.1936 \text{ CC/G} \quad C_S = 1.19 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE						-- STANDARD --	
RHO0	US	UFS	UP	P	V/V0	MATERIAL	P
5.402	2.09	1.01	0.54	61	0.741	BRASS	131
5.253	3.07	3.30	1.70	274	0.446	BRASS	553
5.400	5.14	6.57	3.26	903	0.366	BRASS	1530
5.294	2.70	2.69	1.30	186	0.518	ALUMINUM	221
5.301	4.56	5.77	2.79	673	0.388	ALUMINUM	688
5.301	2.40		0.97	123	0.597	ALUMINUM	149
5.302	3.86	4.38	2.21	452	0.428	ALUMINUM	472

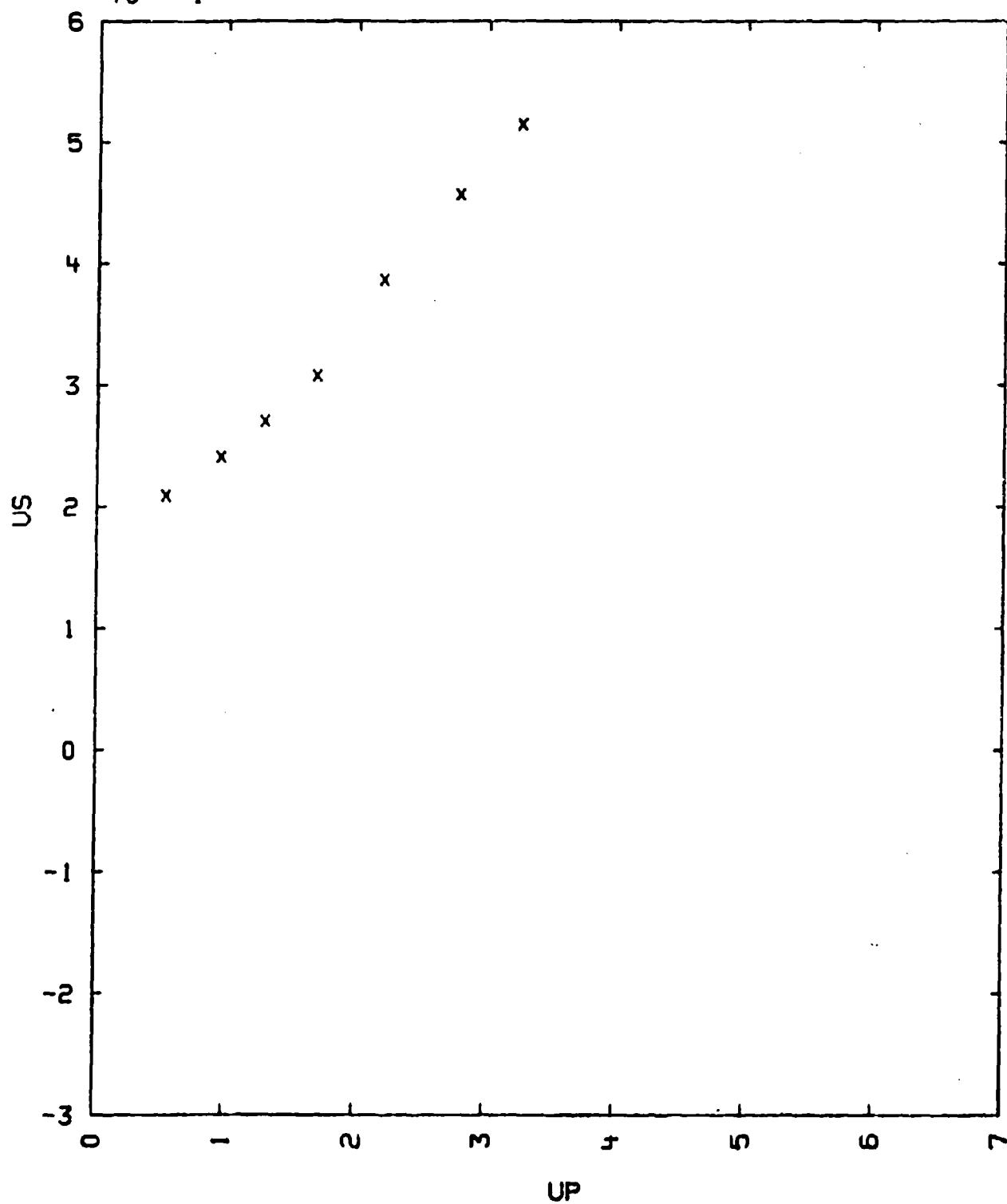
US

COMMENTS:

- 1) SOURCE: GUST, W. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
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- 2) EXPERIMENTAL TECHNIQUE B.
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TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE,
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF VOI WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANT
OF THE CUBIC BODY CENTERED STRUCTURE.
 $A = 4.808 \text{ ANGSTROMS}$
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. 1, P. 18.
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA, 94550

TABLE I

EUROPIUM
70---1



71---1
SAMARIUM

SM

$$V_0 = 0.1338 \text{ CC/G} \quad C_L = 2.92 \text{ KM/SEC} \quad C_0 = 2.23 \text{ KM/SEC}$$

$$V_{01} = 0.1327 \text{ CC/G} \quad C_S = 1.63 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE							-- STANDARD --	
RHO0	US	UFS	UP	P	V/V0	MATERIAL	P	
7.452	2.72	0.94	0.50	101	0.819	BRASS	149	
7.505	3.61	2.93	1.53	414	0.577	BRASS	583	
7.451	5.39	6.11	2.96	1188	0.450	BRASS	1570	
7.475	3.14	2.12	1.09	254	0.854	ALUMINUM	233	
7.477	4.30	4.23	2.04	654	0.526	ALUMINUM	542	
7.478	4.83	5.11	2.47	891	0.489	ALUMINUM	730	

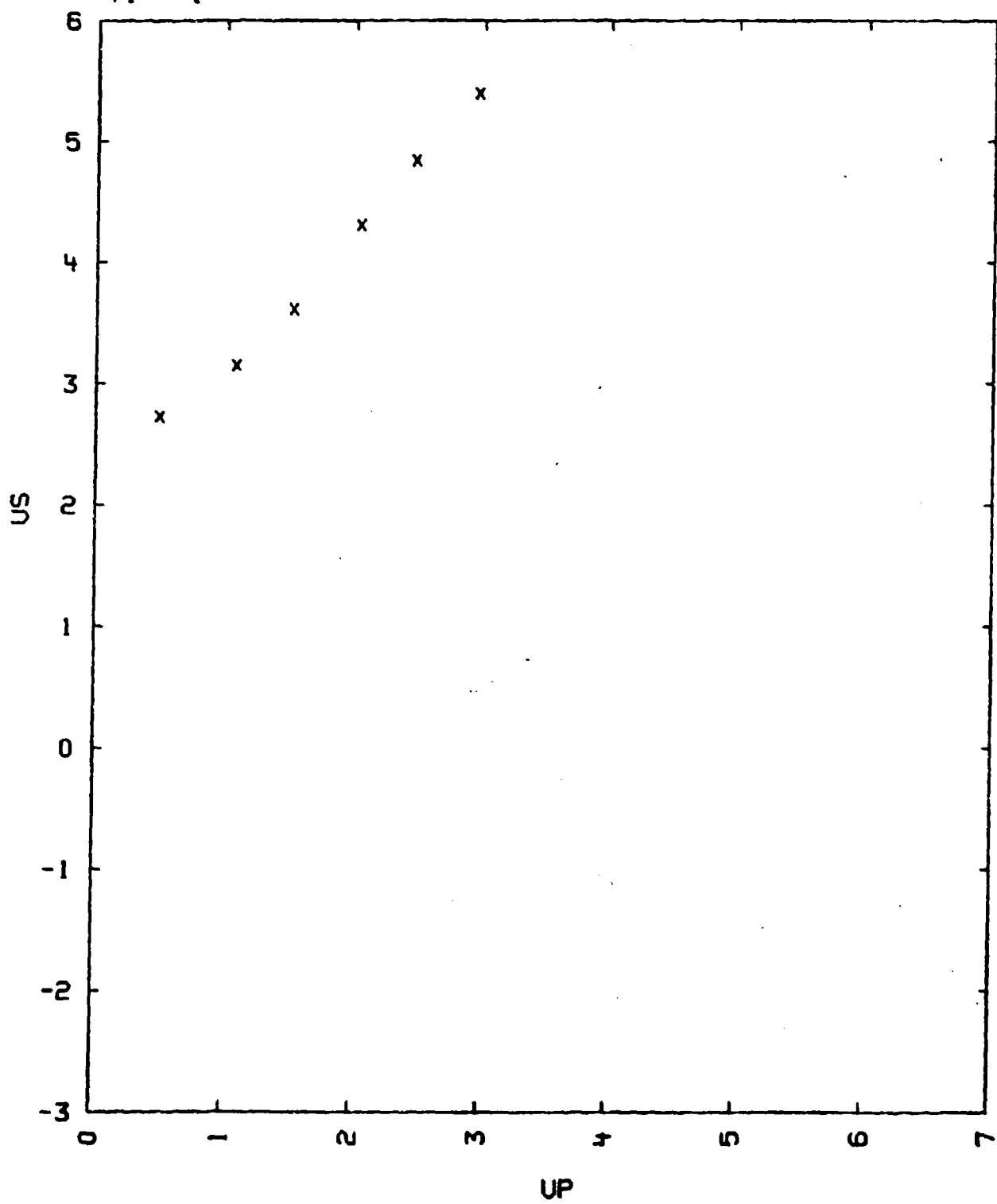
US *

COMMENTS:

- 1) SOURCE: DUST, W. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
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- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THIS IS PRELIMINARY DATA AND FURTHER WORK IS IN PROGRESS.
TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE,
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF V01 WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF A HEXAGONAL UNIT CELL CONTAINING NINE ATOMS.
 $A = 3.621 \text{ ANGSTROMS}$, $C = 26.25 \text{ ANGSTROMS}$
WYCKOFF, R. W. O., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. I, P. 24.
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA. 94550
- 6) THE LONGITUDINAL AND SHEAR VELOCITIES HAVE ALSO BEEN MEASURED BY
SMITH, J. F., CARLSON, C. E. AND SPEDDING, F. H., ELASTIC PROPERTIES
OF YTTRIUM AND ELEVEN OF THE RARE EARTH ELEMENTS, JOURNAL OF METALS
(TRANSACTIONS AIME) OCT. 1957 P. 1212. THE MEASURED VALUES ARE:
 $C_L = 2.702 \pm 0.004 \text{ KM/SEC}$
 $C_S = 1.292 \pm 0.017 \text{ KM/SEC}$.

TABLE I

SAMARIUM
71---1



71---2
SAMARIUM

SM

$$V_0 = 0.133 \text{ CC/G}$$

$$V_{01} = 0.1327 \text{ CC/G}$$

$$C_0 = 2.26 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

----- SAMPLE -----					----- STANDARD -----		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
7.50	8.78	5.00	3293.	0.430	IRON	7.85	11.21
-	6.49	3.49	1699	0.462	IRON	7.85	8.83
-	4.65	2.43	847.	0.477	ALUMINUM	2.71	9.17
-	4.28	2.09	671.	0.512	COPPER	8.93	6.45
-	3.42	1.34	344.	0.608	ALUMINUM	2.71	7.34
-	2.71	0.63	128.	0.767	ALUMINUM	2.71	6.21

$$US = 2.19 + 0.88 \cdot UP \text{ KM/SEC}$$

$$US = 0.84 + 1.600 \cdot UP \text{ KM/SEC}$$

FOR UP BELOW 1.87 KM/SEC

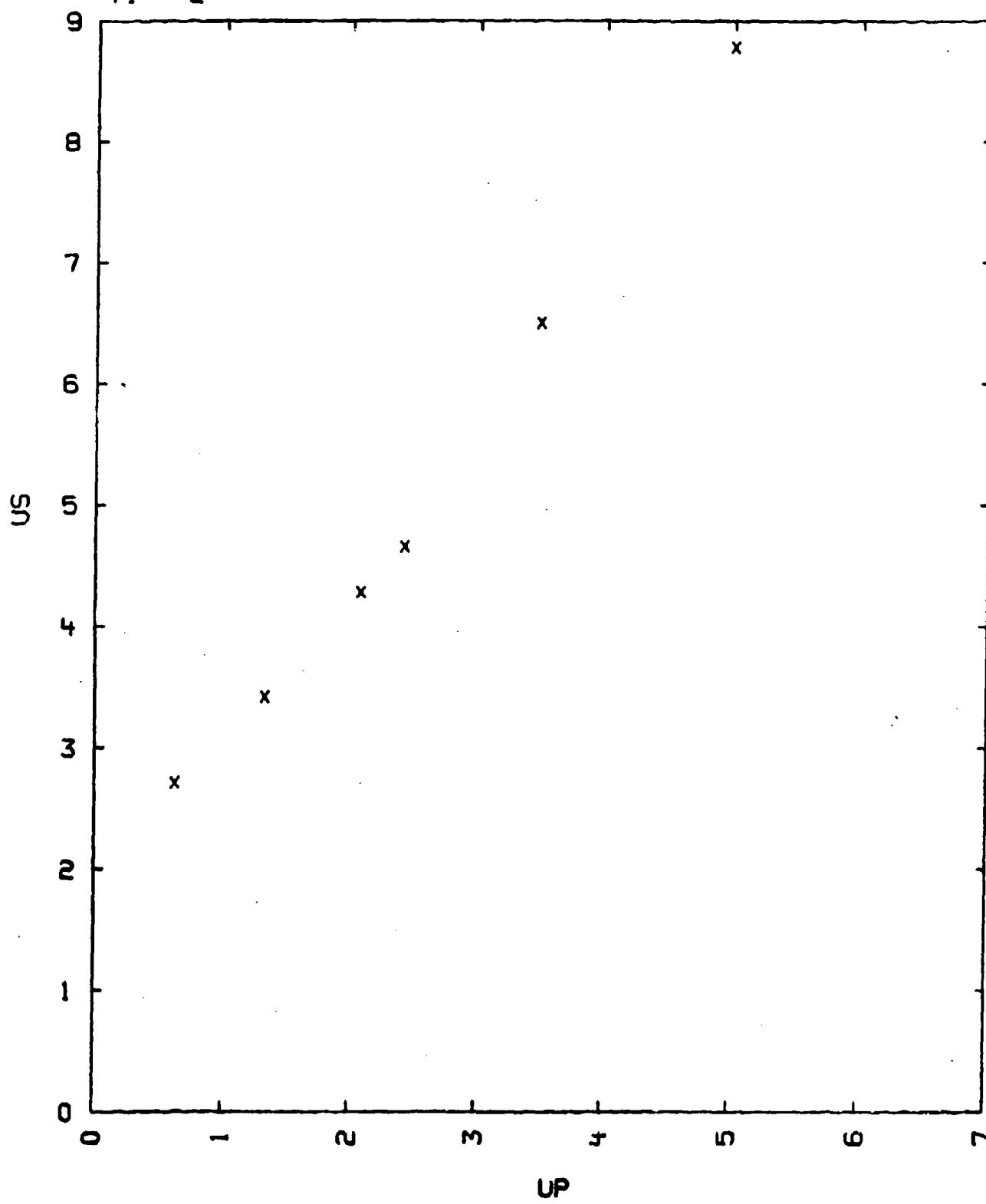
FOR UP BETWEEN 1.87 AND 5.00 KM/SEC

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P. ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES. THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE 1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = 5.25 + 1.39 · UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
COPPER US = 3.95 + 1.50 · UP FOR UP BETWEEN 0.0 AND 4.0 KM/SEC
IRON US = 3.85 + 1.615 · UP FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF VOI WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS OF A HEXAGONAL UNIT CELL CONTAINING 9 ATOMS
A = 3.621 ANGSTROMS, C = 26.25 ANGSTROMS
WYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 24.
- 6) THE AUTHORS OBTAINED THE VALUE OF CO FROM HANDBOOK OF RARE METALS, (MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT UP = 1.87 KM/SEC IS ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I

SAMARIUM
71---2



73---1
NEODYMIUM

ND

$$V_0 = 0.1427 \text{ CC}/\text{G} \quad C_L = 2.82 \text{ KM}/\text{SEC} \quad C_0 = 2.18 \text{ KM}/\text{SEC}$$

$$V_{01} = 0.1427 \text{ CC}/\text{G} \quad C_S = 1.55 \text{ KM}/\text{SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

----- SAMPLE -----							-- STANDARD --	
RHO0	US	UFS	UP	P	V/V0	MATERIAL	P	
7.012	2.50	1.00	0.51	89	0.799	BRASS	139	
7.008	3.54	2.93	1.48	369	0.581	BRASS	547	
7.007	5.38	5.90	2.86	1075	0.469	BRASS	1433	
7.010	3.13	2.21	1.11	243	0.644	ALUMINUM	230	
7.002	4.32	4.31	2.03	618	0.526	ALUMINUM	534	
7.005	2.77	1.58	0.73	141	0.738	ALUMINUM	135	

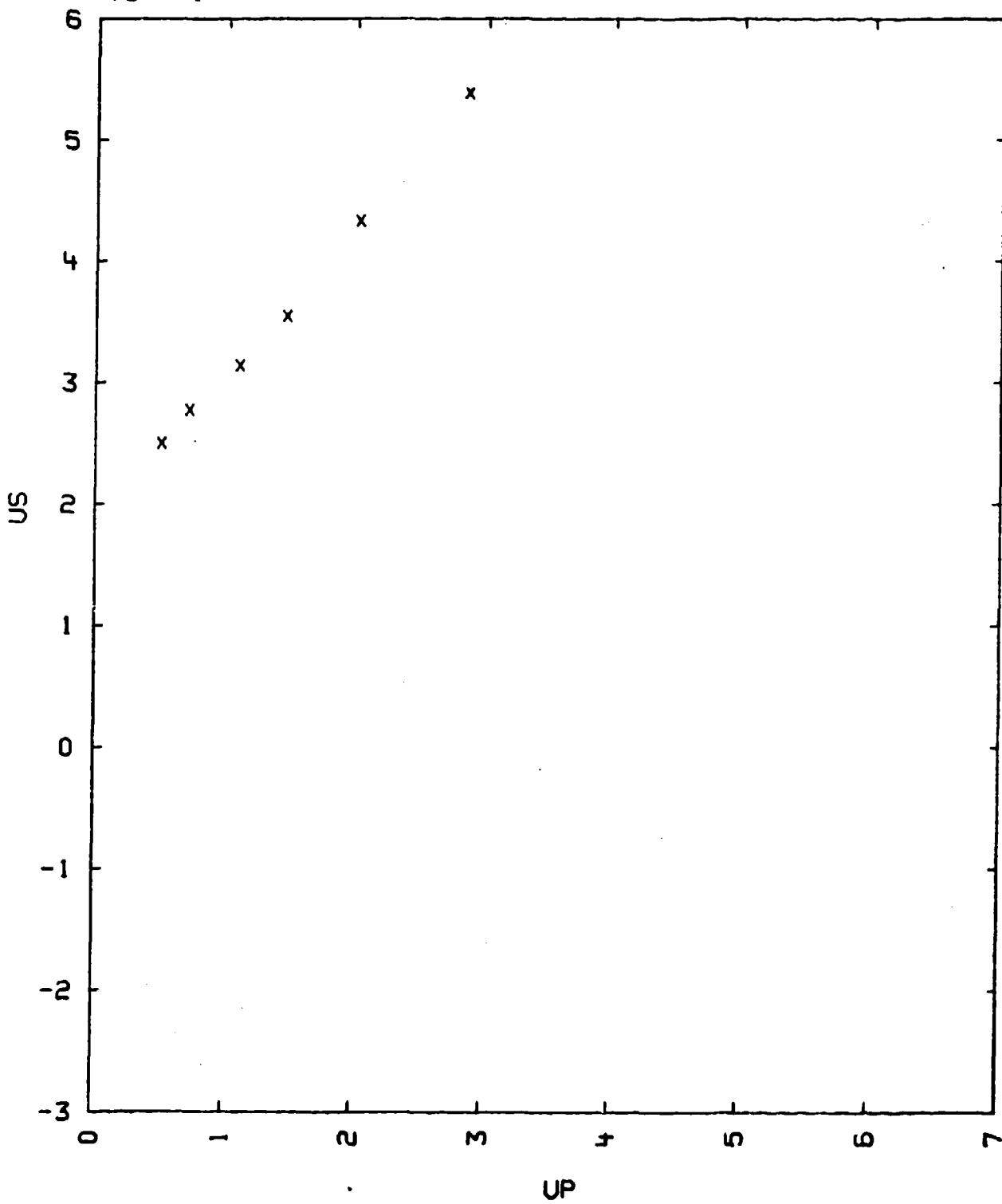
US =

COMMENTS:

- 1) SOURCE: GUST, W. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
BOX 808, LIVERMORE, CALIFORNIA. 94550
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THIS IS PRELIMINARY DATA AND FURTHER WORK IS IN PROGRESS.
TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE,
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF V01 WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL CLOSE-PACKED STRUCTURE.
 $A = 3.857 \text{ ANGSTROMS}$, $C = 5.902 \text{ ANGSTROMS}$
HYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. I, P. 11.
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA. 94550
- 6) THE LONGITUDINAL AND SHEAR VELOCITIES HAVE ALSO BEEN MEASURED BY
SMITH, J. F., CARLSON, C. E. AND SPEDDING, F. H., ELASTIC PROPERTIES
OF YTTRIUM AND ELEVEN OF THE RARE EARTH ELEMENTS, JOURNAL OF METALS
(TRANSACTIONS AIME) OCT. 1957 P. 1212. THE MEASURED VALUES ARE:
 $C_L = 2.718 \text{ PLUS OR MINUS } 0.006 \text{ KM}/\text{SEC}$
 $C_S = 1.438 \text{ PLUS OR MINUS } 0.013 \text{ KM}/\text{SEC}$.

TABLE I

NEODYMIUM
73---1



73---2
NEODYMIUM

ND

$$V_0 = 0.14 \text{ CC/G}$$

$$V_{01} = 0.1427 \text{ CC/G}$$

$$C_0 = 2.15 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

----- SAMPLE -----					----- STANDARD -----		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
7.0	8.89	5.09	3164.	0.427	IRON	7.85	11.21
-	6.62	3.54	1640.	0.465	IRON	7.85	8.83
-	4.76	2.48	826.	0.479	ALUMINUM	2.71	9.17
-	3.48	1.38	336.	0.604	ALUMINUM	2.71	7.34
-	2.75	0.66	127.	0.760	ALUMINUM	2.71	6.21

$$US = 2.08 + 1.015 \cdot UP \text{ KM/SEC}$$

$$US = 0.90 + 1.585 \cdot UP \text{ KM/SEC}$$

FOR UP BELOW 2.07 KM/SEC

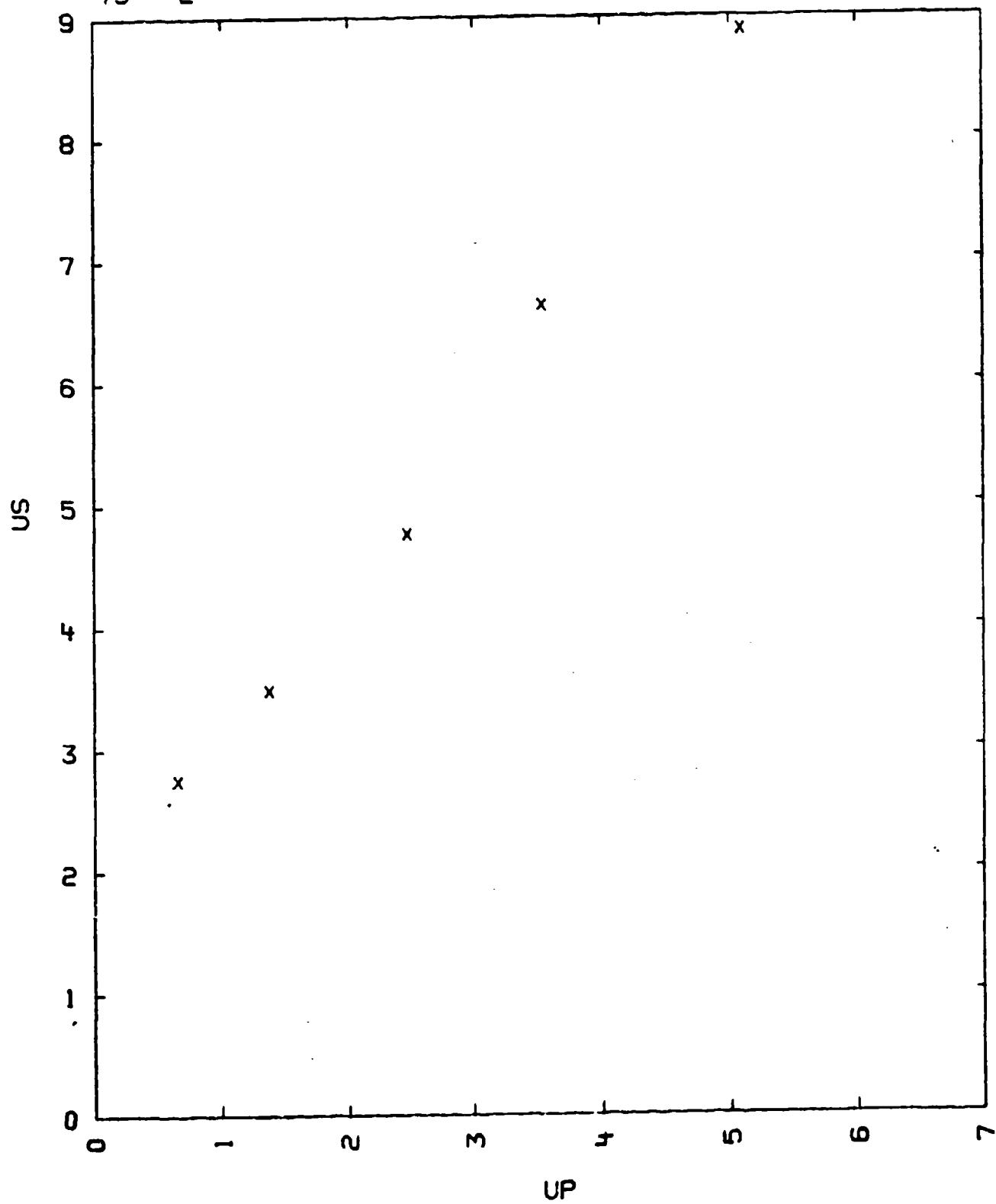
FOR UP BETWEEN 2.07 AND 5.09 KM/SEC

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. O. ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES. THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE 1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = 5.25 + 1.39 · UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
IRON US = 3.85 + 1.615 · UP FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF VO1 WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS OF THE HEXAGONAL CLOSE-PACKED STRUCTURE.
 $A = 3.657 \text{ ANGSTROMS}$, $C = 5.902 \text{ ANGSTROMS}$
WYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 11.
- 6) THE AUTHORS OBTAINED THE VALUE OF CO FROM HANDBOOK OF RARE METALS, (MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT UP = 2.07 KM/SEC IS ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I

NEODYMIUM
73---2



74---1
PRASEODYMIUM

PR

$$V_0 = 0.1480 \text{ CC/G} \quad C_L = 2.72 \text{ KM/SEC} \quad C_0 = 2.11 \text{ KM/SEC}$$

$$V_{01} = 0.1469 \text{ CC/G} \quad C_S = 1.49 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE						-- STANDARD --	
RHOD	US	UFS	UP	P	V/V0	MATERIAL	P
6.730	2.53	0.97	0.52	90	0.793	BRASS	149
6.754	3.53	3.02	1.59	379	0.550	BRASS	583
6.728	5.70	6.73	3.01	1153	0.473	BRASS	1570
6.776	3.04	2.11	1.16	238	0.830	ALUMINUM	233
6.769	4.43	4.30	2.11	631	0.525	ALUMINUM	542
6.772	5.06	5.04	2.53	867	0.499	ALUMINUM	730

US =

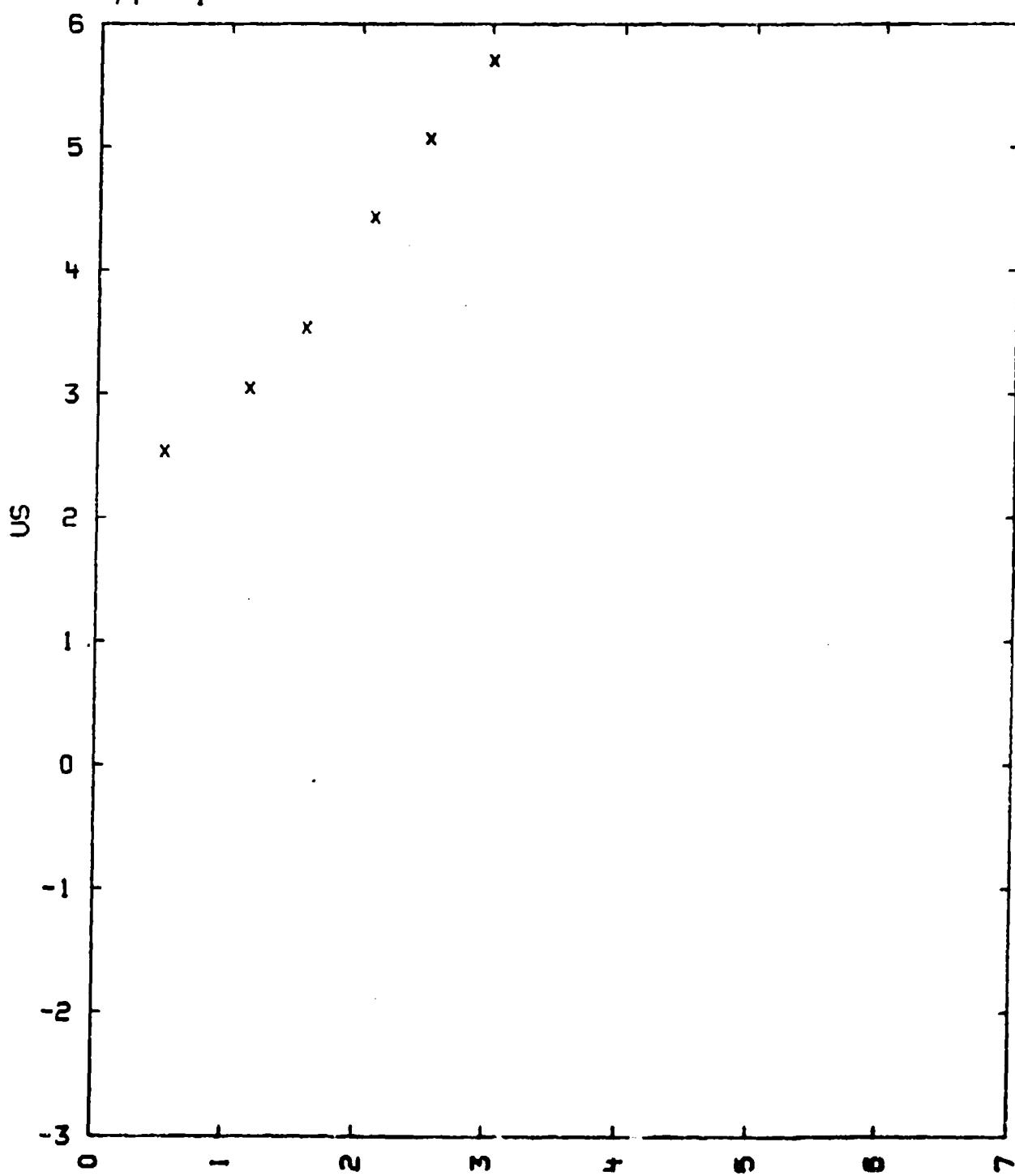
COMMENTS:

- 1) SOURCE: GUST, H. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
BOX 808, LIVERMORE, CALIFORNIA. 94550
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THIS IS PRELIMINARY DATA AND FURTHER WORK IS IN PROGRESS.
TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE,
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF V01 HAS BEEN OBTAINED FROM THE FOLLOWING LATTICE CONSTANT
OF THE CUBIC CLOSE-PACKED STRUCTURE.
 $A = 5.161 \text{ ANGSTROMS}$
WYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. 1, P. 10.
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA. 94550
- 6) THE LONGITUDINAL AND SHEAR VELOCITIES HAVE ALSO BEEN MEASURED BY
SMITH, J. F., CARLSON, C. E. AND SPEDDING, F. H., ELASTIC PROPERTIES
OF YTTRIUM AND ELEVEN OF THE RARE EARTH ELEMENTS, JOURNAL OF METALS
(TRANSACTIONS AIME) OCT. 1957 P. 1212. THE MEASURED VALUES ARE:
 $C_L = 2.660 \text{ PLUS OR MINUS } 0.011 \text{ KM/SEC}$
 $C_S = 1.410 \text{ PLUS OR MINUS } 0.016 \text{ KM/SEC}$

TABLE I

PRASEODYMIUM

74---1



75---1
CERIUM

CE

$$\begin{array}{lll} V_0 = 0.1476 \text{ CC/G} & C_L = 2.35 \text{ KM/SEC} & C_0 = 1.79 \text{ KM/SEC} \\ V_{01} = 0.14773 \text{ CC/G} & C_S = 1.32 \text{ KM/SEC} & \end{array}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

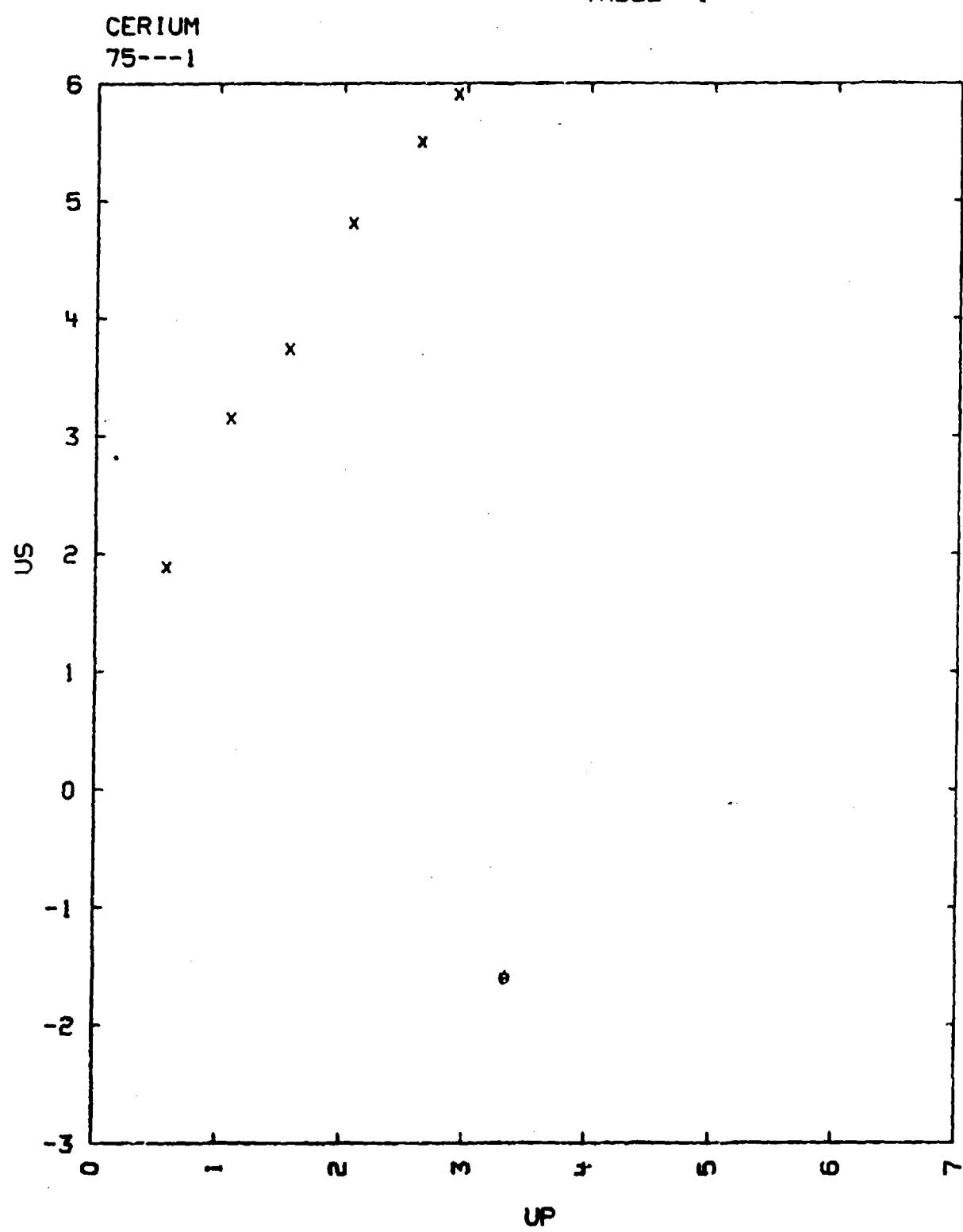
SAMPLE						-- STANDARD --	
RHO0	US	UFS	UP	P	V/V0	MATERIAL	P
6.760	1.88	1.12	0.58	73	0.694	BRASS	148
6.742	3.73	2.90	1.57	395	0.580	BRASS	597
6.753	5.90	6.02	2.93	1166	0.504	BRASS	1543
6.797	3.14	2.18	1.10	235	0.649	ALUMINUM	223
6.798	4.80	4.26	2.08	679	0.556	ALUMINUM	561
6.797	5.49	5.33	2.62	977	0.523	BRASS	1290

US =

COMMENTS:

- 1) SOURCE: GUST, W. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
BOX 808, LIVERMORE, CALIFORNIA. 94550
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THIS IS PRELIMINARY DATA AND FURTHER WORK IS IN PROGRESS.
TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE,
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF V01 WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANT
OF THE CUBIC CLOSE-PACKED STRUCTURE.
 $A = 5.1812 \text{ ANGSTROMS}$
HYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. I P. 10.
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA. 94550
- 6) THE LONGITUDINAL AND SHEAR VELOCITIES HAVE ALSO BEEN MEASURED BY
SMITH, J. F., CARLSON, C. E. AND SPEDDING, F. H., ELASTIC PROPERTIES
OF YTTRIUM AND ELEVEN OF THE RARE EARTH ELEMENTS, JOURNAL OF METALS
(TRANSACTIONS AIME) OCT. 1957 P. 1212. THE MEASURED VALUES ARE:
 $C_L = 2.300 \pm 0.020 \text{ KM/SEC}$
 $C_S = 1.332 \pm 0.011 \text{ KM/SEC}$.

TABLE I



75---2

CERIUM

CE

$$V_0 = 0.148 \text{ CC/G}$$

$$V_{01} = 0.14773 \text{ CC/G}$$

$$C_0 = 1.71 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
6.75	9.08	5.12	3133.	0.436	IRON	7.85	11.21
-	6.93	3.54	1656.	0.489	IRON	7.85	8.83
-	5.17	2.42	845.	0.532	ALUMINUM	2.71	9.17
-	3.56	1.38	332.	0.612	ALUMINUM	2.71	7.34
-	2.37	0.71	113.6	0.700	ALUMINUM	2.71	6.21
-	2.10	0.50	70.8	0.762			

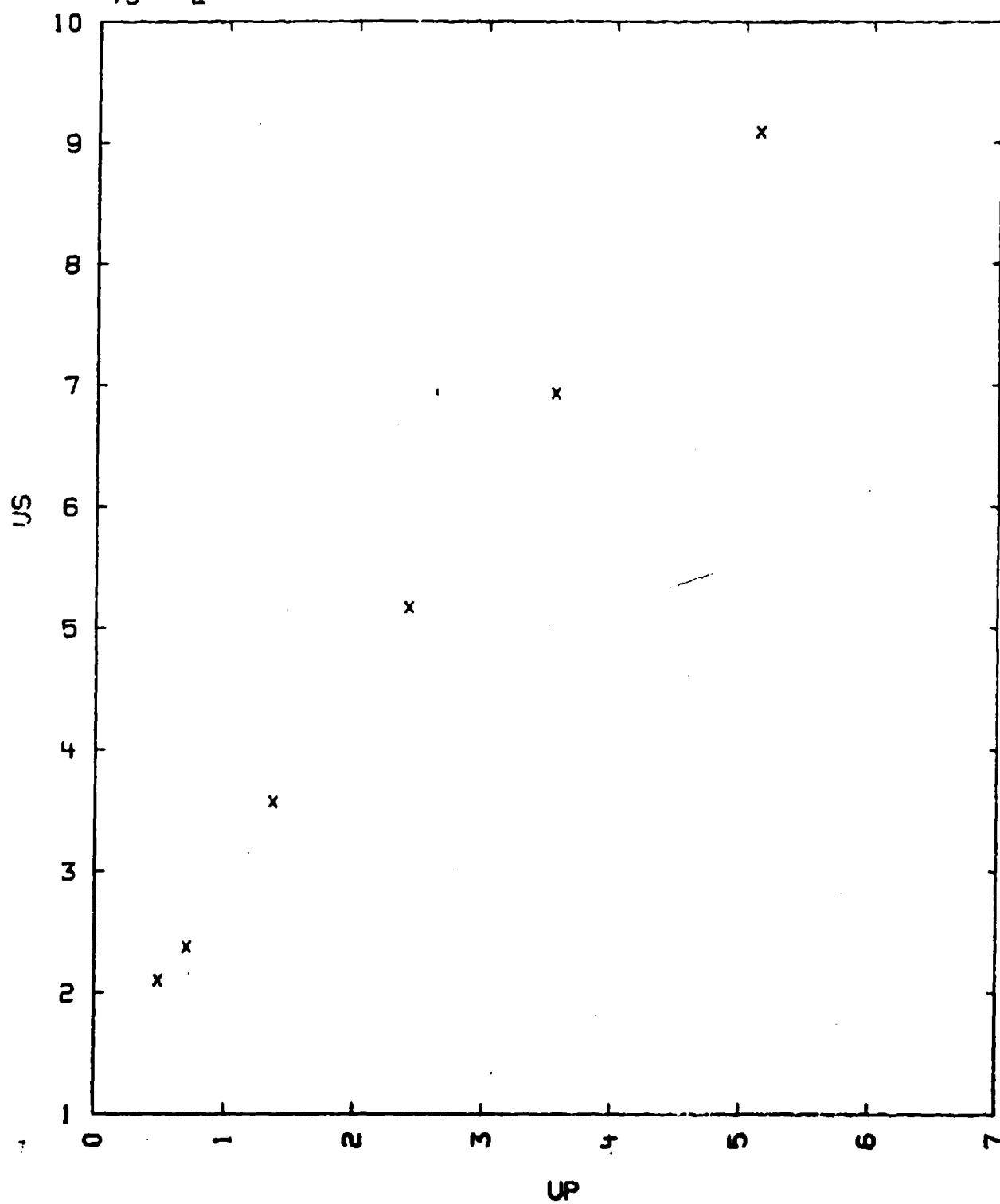
$$US = 1.27 + 1.596 \cdot UP \text{ KM/SEC FOR UP BETWEEN 0.7 AND 5.12 KM/SEC}$$

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P. ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES. THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE 1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = 5.25 + 1.39 · UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
IRON US = 3.85 + 1.815 · UP FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF V01 WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANT OF THE CUBIC CLOSE-PACKED STRUCTURE.
A = 3.1612 ANGSTROMS
WYCKOFF, R. W. B., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 10.
- 6) THE AUTHORS OBTAINED THE VALUE OF CO FROM HANDBOOK OF RARE METALS, (MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).

TABLE I

CERIUM
75---2



76---1
LANTHANUM

LA

$$\begin{array}{lll} V_0 = 0.1634 \text{ CC/G} & C_L = 2.77 \text{ KM/SEC} & C_0 = 2.13 \text{ KM/SEC} \\ V_{01} = 0.160 \text{ CC/G} & C_S = 1.53 \text{ KM/SEC} & \end{array}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

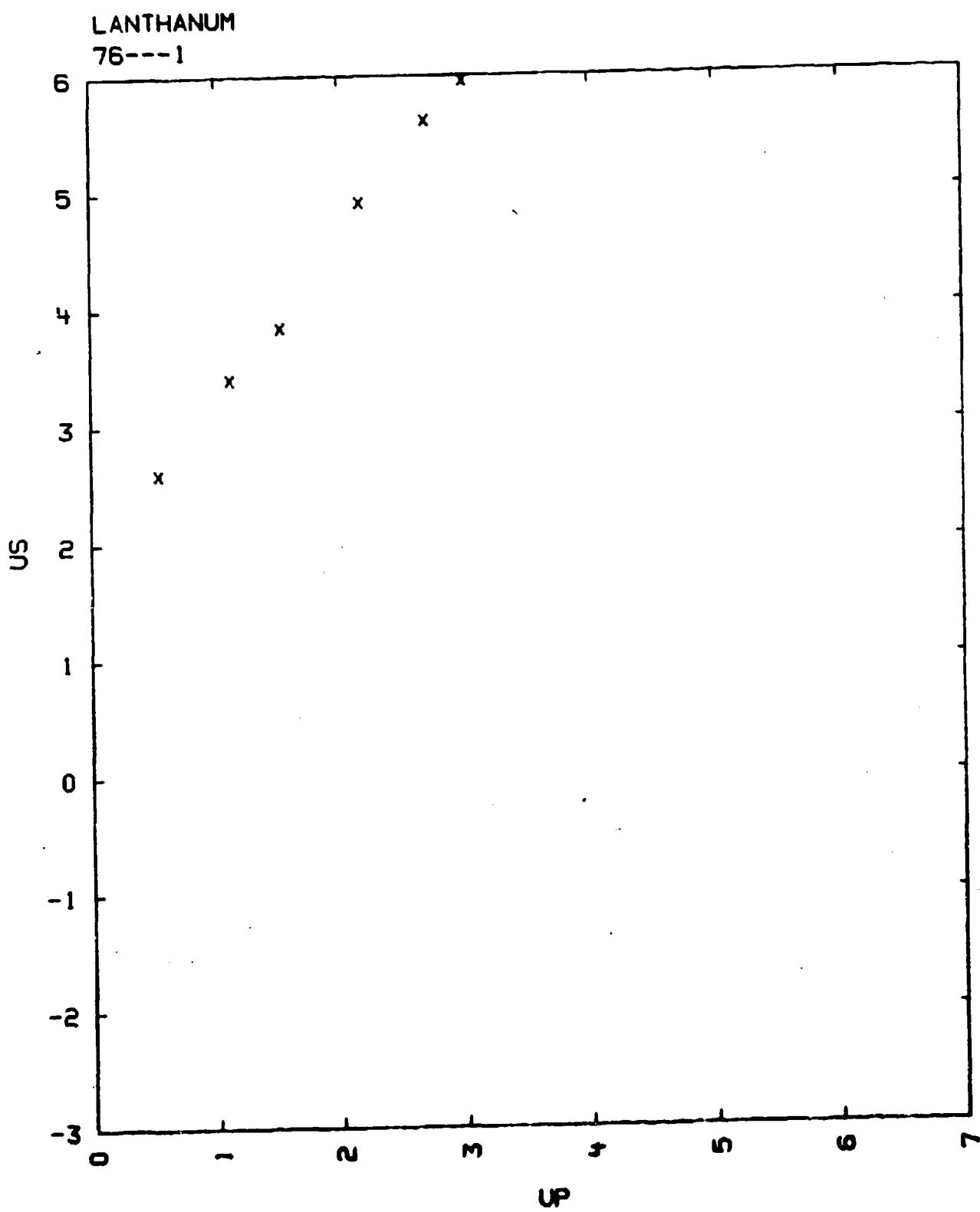
SAMPLE						-- STANDARD --	
RHO0	US	UFS	UP	P	V/V0	MATERIAL	P
6.148	2.59	0.97	0.54	87	0.790	BRASS	131
6.147	3.84	2.98	1.53	361	0.602	BRASS	553
6.135	5.95	6.09	3.00	1093	0.497	BRASS	1530
6.094	3.39	2.27	1.12	232	0.668	ALUMINUM	223
6.091	4.90	4.45	2.17	648	0.557	ALUMINUM	561
6.098	5.61		2.70	923	0.520	BRASS	1290

US =

COMMENTS:

- 1) SOURCE: GUST, W. H.
PRIVATE COMMUNICATION
UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION LABORATORY,
BOX 808, LIVERMORE, CALIFORNIA, 94550
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THIS IS PRELIMINARY DATA AND FURTHER WORK IS IN PROGRESS.
TAKING INTO ACCOUNT THE ACCURACY OF THE EXPERIMENTAL TECHNIQUE,
THE DATA ARE BEST DESCRIBED BY MORE THAN ONE STRAIGHT LINE IN THE
US VS. UP PLANE. THE SIGNIFICANCE OF THIS CHARACTERISTIC IS BEING
INVESTIGATED.
- 4) THE VALUE OF V_{01} WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANT
OF THE CUBIC CLOSE-PACKED STRUCTURE.
 $A = 5.296 \text{ ANGSTROMS}$
HYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SYDNEY, 1963) 2ND EDITION, VOL. 1, P. 10.
- 5) SOUND VELOCITIES WERE MEASURED BY H. L. DUNEGAN (SUPPORT ENGINEERING,
ULTRASONIC GROUP) UNIVERSITY OF CALIFORNIA, LAWRENCE RADIATION
LABORATORY, BOX 808, LIVERMORE, CALIFORNIA, 94550
- 6) THE LONGITUDINAL AND SHEAR VELOCITIES HAVE ALSO BEEN MEASURED BY
SMITH, J. F., CARLSON, C. E. AND SPEDDING, F. H., ELASTIC PROPERTIES
OF YTTRIUM AND ELEVEN OF THE RARE EARTH ELEMENTS, JOURNAL OF METALS
(TRANSACTIONS AIME) OCT. 1957 P. 1212. THE MEASURED VALUES ARE:
 $C_L = 2.751 \text{ PLUS OR MINUS } 0.008 \text{ KM/SEC}$
 $C_S = 1.501 \text{ PLUS OR MINUS } 0.002 \text{ KM/SEC.}$

TABLE I



76---2
LANTHANUM

LA

$$V_0 = 0.163 \text{ CC/G}$$

$$V_{01} = 0.160 \text{ CC/G}$$

$$C_0 = 2.22 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC. PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
6.15	9.09	5.27	2946.	0.420	IRON	7.85	11.21
-	6.95	3.65	1560.	0.475	IRON	7.85	8.83
-	5.35	2.49	820.	0.534	ALUMINUM	2.71	9.17
-	4.08	1.66	417.	0.593			
-	3.70	1.41	321.	0.619	ALUMINUM	2.71	7.34
-	3.26	1.10	221.	0.663			
-	2.79	0.68	117.	0.756	ALUMINUM	2.71	6.21
-	2.67	0.48	79.	0.820			
-	2.34	0.24	34.	0.898			

US = $2.16 + 1.0 \cdot UP$ KM/SEC FOR UP BELOW 1.11 KM/SEC

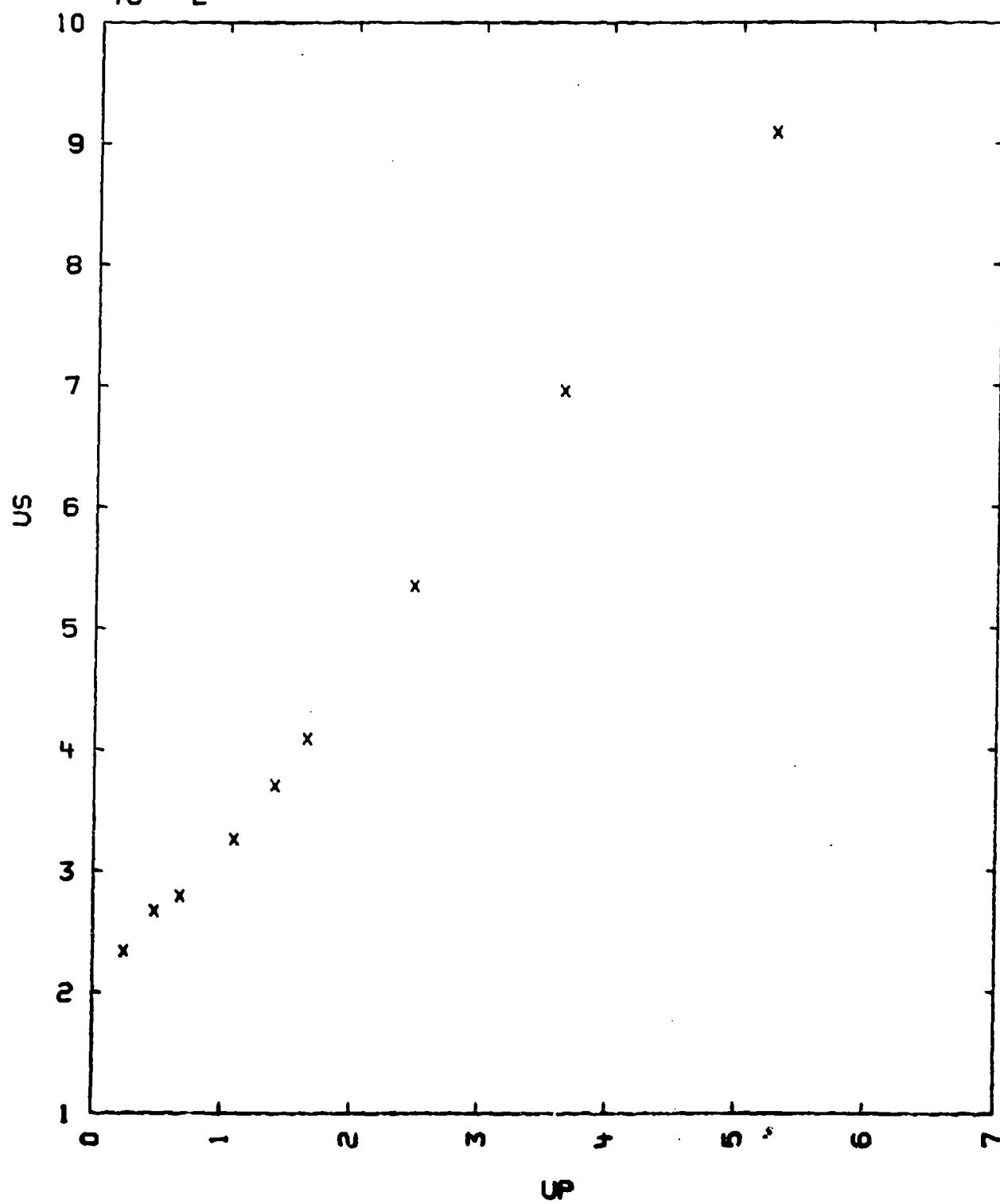
US = $1.65 + 1.46 \cdot UP$ KM/SEC FOR UP BETWEEN 1.11 AND 4.0 KM/SEC

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P.
ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES.
THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE
1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = $3.25 + 1.39 \cdot UP$ FOR UP BETWEEN 0.0 AND 8.0 KM/SEC
IRON US = $3.85 + 1.615 \cdot UP$ FOR UP BETWEEN 1.0 AND 4.8 KM/SEC
- 5) THE VALUE OF V_{01} HAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANT
OF THE CUBIC CLOSE-PACKED STRUCTURE.
 $A = 5.288 \text{ ANGSTROMS}$
WYCKOFF, R. H. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 10.
- 6) THE AUTHORS OBTAINED THE VALUE OF C_0 FROM HANDBOOK OF RARE METALS,
(MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO
SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT UP = 1.11 KM/SEC IS
ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I

LANTHANUM
76--2



88---0
URANIUM SUMMARY

U

$V_0 = 0.0533 - 0.0525 \text{ CC/G}$
 $V_{01} = 0.05249 \text{ CC/G}$

THE TABLE LISTS HUGONIOT POINTS CALCULATED FROM THE FITS GIVEN BELOW.
 UNITS ARE: G/CC, KM/SEC, KBAR AND KBAR.CC/G FOR THE ENERGY DIFFERENCE.
 EL = ELASTIC LIMIT.

TABLE

FIT	RHOO	US	UP	P	V/V0	E-E0	COMMENTS	
							EL	1ST WAVE
	18.00	3.45	0.0025	1.6	0.9993	0.000031	EL	1ST WAVE
1	-	2.610	0.1	49.4	0.9619	0.0475	2ND	-
	-	2.941	0.3	166.	0.8981	0.442	-	-
	-	3.272	0.5	307.	0.8473	1.24	-	-
	-	3.603	0.7	474.	0.806	2.45	1ST	-
	-	3.653	0.73	501.	0.800	2.66	-	-
2	-	3.694	0.80	556.	0.783	3.2	-	-
	-	4.013	1.0	754.	0.751	5.0	-	-
	-	5.606	2.0	2108.	0.643	20.0	-	-
	-	7.199	3.0	4060.	0.583	45.0	-	-
	-	8.792	4.0	6612.	0.545	80.00	-	-
3	9.5	1.828	1.0	173.	0.453	5.0	-	-
	-	3.563	2.0	677.	0.439	20.0	-	-
	-	5.185	3.0	1478.	0.421	45.0	-	-
	-	6.694	4.0	2543.	0.402	80.0	-	-
	-	8.081	5.0	3843.	0.382	125.0	-	-

$US = 2.445 + 1.655 \cdot UP, \quad SIG.US = 0.07 \text{ KM/SEC}$ FIT 1
 FOR UP BETWEEN 0.13 AND .74 KM/SEC.
 $US = 2.420 + 1.593 \cdot UP, \quad SIG.US = 0.08 \text{ KM/SEC}$ FIT 2
 FOR UP BETWEEN 0.8 AND 4. KM/SEC
 $US = 2.483 + 1.579 \cdot UP - 0.262(19.05 - RHOO) +$ FIT 3
 $+ 0.034(19.05 - RHOO) \cdot UP - 0.0059 \cdot (19.05 - RHOO) \cdot UP^{+2},$
 $SIG.US = 0.16 \text{ KM/SEC OVER ALL}$
 $BUT SIG.US = 0.3 \text{ AT } RHOO = 9.5 \text{ G/CC}$
 FOR UP BETWEEN 1 AND 5 AND RHOO FROM 19.05 TO 9.5 G/CC

COMMENTS:

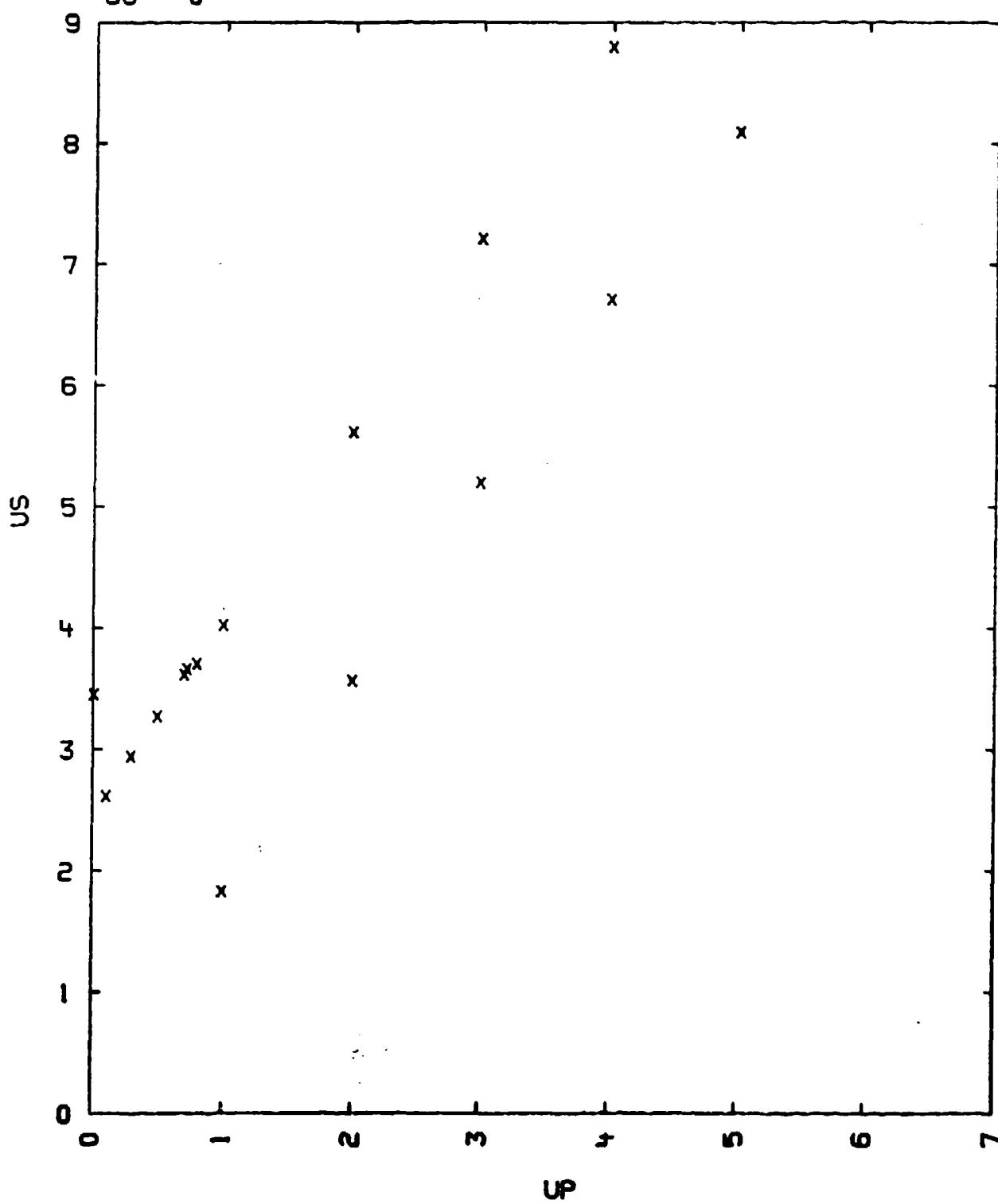
- 1) SOURCE: COMPILER
- 1A) DATA FROM 88---1,2,3,4 AND 5 WERE USED FOR THESE FITS.
- 2) THE ELASTIC LIMIT WAS TAKEN FROM 88---3.
- 3) THE DUAL FIT, 1 AND 2 WERE MADE IN LINE WITH THE TRANSITION SUGGESTED BY DATA OF 88---4.
- 4) THIS TRANSITION HAS BEEN IGNORED IN GENERATING FIT 3 WHICH INCLUDES

POROUS DATA, SINCE THE LATTER ARE INSUFFICIENTLY ACCURATE TO SHOW
THE TRANSITION. MOST OF THE UNCERTAINTY IN THIS FIT IS DUE TO THE
POROUS URANIUM DATA WHICH SHOW A MAXIMUM DEVIATION
 $U_S - U_S(CALC) = 0.7 \text{ KM/SEC}$
AND AN RMS VALUE OF 0.3 KM/SEC.

TABLE I

URANIUM SUMMARY

88---0



88---1
URANIUM POROUS

URANIUM U 96 PERCENT
URANIUM OXIDE U-02 4 PERCENT

$$V_0 = .116 (+OR-) .005 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC., VELOCITIES IN MM/MICRO-SEC., AND PRESSURE IN KILOBARS.

TABLE

RHO0	US	UP	P	V/V0
9.5	2.41	1.34	307	.883
-	2.56	1.41	343	.893
-	2.50	1.40	333	.877
-	4.05	2.23	858	.993
-	4.14	2.20	866	.933
-	4.07	2.38	919	.825
-	4.22	2.36	946	.875
-	4.24	2.41	970	.857
-	5.14	2.80	1370	.905
-	5.00	2.97	1390	.794
-	5.04	2.95	1400	.814
-	5.26	2.90	1450	.891
-	7.94	4.77	3600	.794
-	8.30	4.80	3790	.839
-	7.90	5.20	3900	.676

$$US = 1.869 \cdot UP - 0.047 \cdot UP^{0.2} \text{ KM/SEC}$$

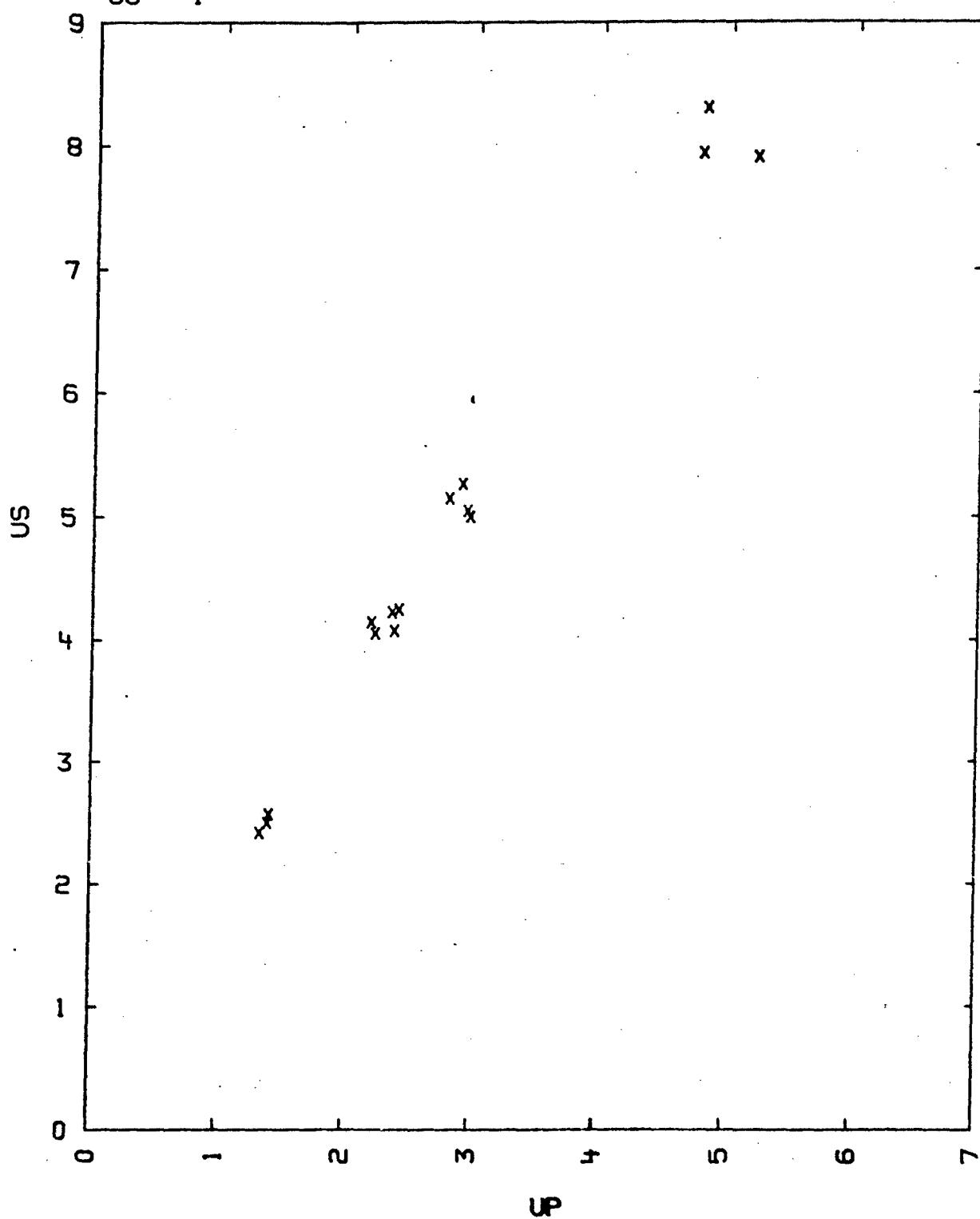
COMMENTS:

- 1) SOURCE: SKIDMORE, I.C. AND MORRIS, E.
THERMODYNAMICS OF NUCLEAR MATERIALS, P. 173 FF. (1962)
INTERN. AT. ENERGY AGENCY, VIENNA
ATOMIC WEAPONS RESEARCH ESTABLISHMENT, ALDERMASTON, ENGLAND
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
THE SHOCK WAS PRODUCED BY AN EXPLOSIVELY ACCELERATED EN3 STEEL PLATE.
THE SHOCK WAS TRANSMITTED THROUGH A STEEL PLATE INTO THE SAMPLE.
- 3) THE VELOCITY OF THE FLYING PLATE AND THE SHOCK AND SURFACE VELOCITY OF THE TARGET PLATE WERE MEASURED AS WELL AS THE SAMPLE SURFACE AND SHOCK VELOCITIES.
- 4) DATA SCATTER WAS ABOUT 0.03 MICROSEC.
- 5) CORRECTIONS WERE MADE FOR FLYING PLATE CURVATURE OF UP TO 1 MICROSEC.
- 6) THE HIGHER PRESSURES WERE OBTAINED BY A SPHERICALLY CONVERGING SYSTEM.
- 7) ALL PELLETS WERE SURROUNDED BY LEAD TO REDUCE LATERAL RAREFACTION.

TABLE I

URANIUM POROUS

88---1



88---2
URANIUM

U

$$V_0 = .05295 \text{ CC/G}$$

$$V_{01} = .05249 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN MM/MICRO-SEC, AND PRESSURE IN KILOBARS.

TABLE

RHO0	US	UP	P	V/V0
18.85	3.35	.53	335	.842
18.9	5.80	2.17	2380	.626
18.9	7.55	3.22	4470	.562
18.9	8.66	3.94	6450	.545

$$US = 2.55 + 1.504UP + 0.0901(UP - 2.50)^{+2} \text{ MM/MICROSEC FOR UP}$$

GREATER THAN 2.50

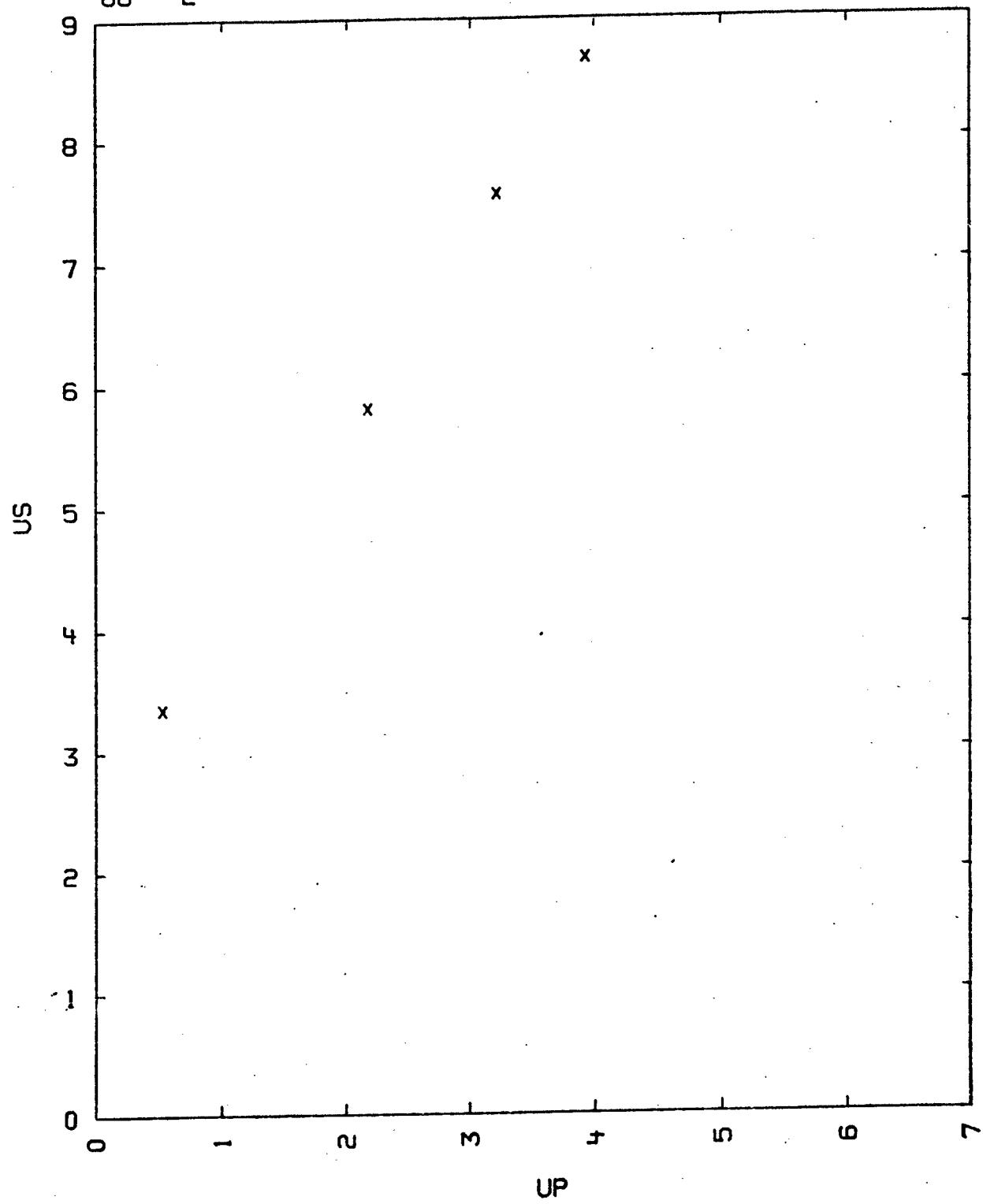
$$US = 2.55 + 1.504UP \text{ MM/MICROSEC FOR UP LESS THAN 2.50}$$

COMMENTS:

- 1) SOURCE: SKIDMORE, I.C. AND MORRIS, E.
THERMODYNAMICS OF NUCLEAR MATERIALS, P. 173 FF. (1962)
INTERN. AT. ENERGY AGENCY, VIENNA
ATOMIC WEAPONS RESEARCH ESTABLISHMENT, ALDERMASTON, ENGLAND
- 2) EXPERIMENTAL TECHNIQUE A
DATA REDUCTION TECHNIQUE B
ALL SAMPLES WERE POSITIONED ON A TARGET PLATE OF STEEL (MTRL. 41---2)
AND THE PRESSURE WAS PRODUCED BY AN EXPLOSIVELY ACCELERATED PLATE OF
STEEL.
- 3) THE VELOCITY OF THE FLYING PLATE AND THE SHOCK AND SURFACE VELOCITY
OF THE TARGET PLATE WERE MEASURED AS WELL AS THE SAMPLE SURFACE AND
SHOCK VELOCITIES.
- 4) DATA SCATTER WAS ABOUT 0.03 MICROSEC
- 5) CORRECTIONS WERE MADE FOR FLYING PLATE CURVATURE OF UP TO 1 MICROSEC.
- 6) THE HIGHER PRESSURES WERE OBTAINED BY A SPHERICALLY CONVERGING
SYSTEM.

TABLE I

URANIUM
88---2



88---3
URANIUM

U

$$V_0 = 0.0531 \text{ CC/G}$$

$$V_{01} = 0.05249 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC
AND PRESSURE IN KILOBARS.

TABLE

RH00	SAMPLE			V/V0	MATERIAL	UFS	PROJECTILE
	US	UP	P				
18.76	3.66	0.520	355	0.858	BRASS	1.365	
18.76	3.62	0.528	356	0.854	BRASS	1.373	
18.76	3.61	0.544	362	0.849	BRASS	1.391	
18.76	3.60	0.760	515	0.789	BRASS		
18.76	3.85	0.837	602	0.783	DURAL	3.415	
18.76	3.94	0.773	570	0.804	DURAL	3.250	
18.76	3.83	0.790	570	0.794	DURAL	3.270	
18.76	3.80	0.755	532	0.801	BRASS	1.905	
18.76	3.77	0.762	525	0.798	BRASS	1.883	
18.76	4.59	1.33	1150	0.710			3.35
18.76	4.59	1.34	1150	0.708			3.36
18.76	4.58	1.33	1140	0.710			3.35
19.00	4.48	1.33	1130	0.703			3.33
19.00	4.53	1.37	1180	0.698			3.43
19.00	4.56	1.35	1170	0.704			3.40
18.76	5.20	1.68	1670	0.677			4.29
18.76	5.24	1.71	1680	0.674			4.34
19.00	5.16	1.69	1660	0.672			4.30
19.00	5.19	1.69	1620	0.674			4.25
19.00	5.18	1.70	1630	0.672			4.28

$$US = 2.77 + 1.39 UP \text{ KM/SEC}$$

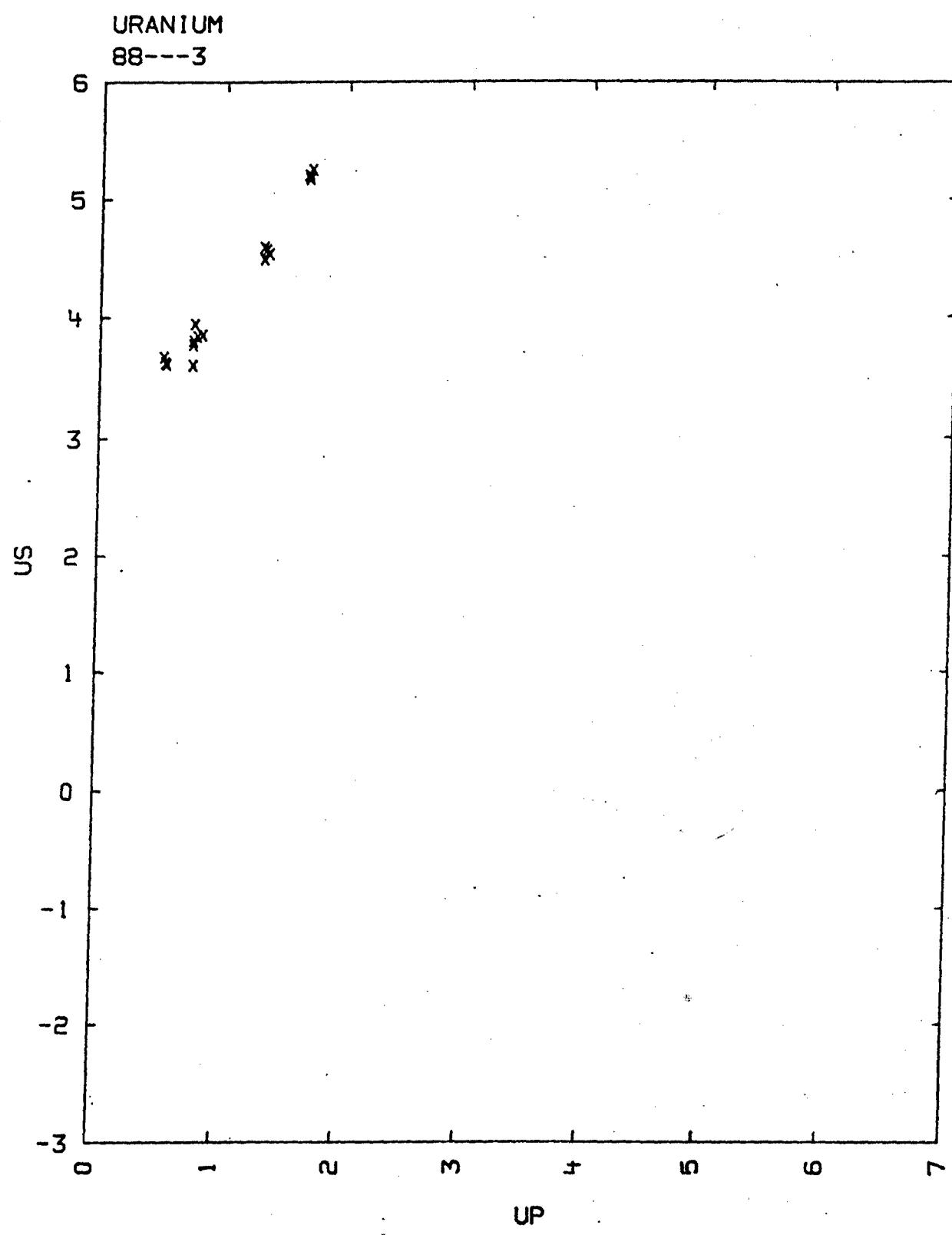
$$\Sigma US = 0.105 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: PERSSON, P.A. AND PERSSON, I.
FOA 2 REPORT A 2299-222, SEPT. 1964,
FORSVARETS FORSKNINGSANSTALT AVDELNING, SWEDEN.
UCRL TRANSLATION 1173(L).
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) IN THE EXPERIMENTS USING FLYING PLATE PROJECTILES BOTH THE BASE PLATE
AND PROJECTILE WERE MADE OF SVENSKA METALLWERKEN ALLOY 1163 WITH
COMPOSITION CU/ZN/PB = 63.0/36.6/0.4, RHO = 8.48 G/CC
CALIBRATION EXPERIMENTS OF THIS ALLOY AGREED TO 1 PERCENT WITH DATA
BY R. G. MCQUEEN AND S. P. MARSH, J. APPL. PHYS., VOL 31 P 1253

- (1960).
- 4) THE ALUMINUM ALLOY USED FOR BASE PLATES HAD A COMPOSITION
AL/CU/MG/MN/SI=93.6/4.5/0.6/0.8/0.7 AND IT WAS ASSUMED WITHOUT
CHECKING TO HAVE THE SAME HUGONIOT AS REPORTED BY M. H. RICE ET AL.
SOLID STATE PHYSICS, VOL. 16, P 1, (1958).
 - 5) THE VALUE OF VOI WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES
(DOVER PUBLICATIONS, INC., NEW YORK, N. Y., 1963).

TABLE I



88---4
URANIUM

U

$$V_0 = 0.5283 - 0.5249 \text{ CC/G}$$

$$V_{01} = 0.5249 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC
AND PRESSURE IN KILOBARS.

TABLE

RHO0	US	UP	P	V/V0
19.05	3.124	0.4525	269	0.855
-	3.131	0.460	274	0.853
-	3.2825	0.4975	312	0.848
-	3.225	0.510	313	0.842
-	3.308	0.545	344	0.835
-	3.317	0.546	345	0.835
-	3.390	0.535	345	0.842
-	3.400	0.585	379	0.828
-	3.3775	0.5925	381	0.824
-	3.408	0.610	396	0.821
-	3.490	0.605	402	0.827
-	3.425	0.6175	403	0.820
-	3.442	0.630	412	0.817
-	3.520	0.645	432	0.817
-	3.480	0.655	433	0.812
-	3.490	0.655	435	0.812
-	3.595	0.6825	466	0.810
-	3.575	0.6875	468	0.808
-	3.570	0.695	472	0.805
-	3.645	0.682	473	0.813
-	3.650	0.720	500	0.8025
-	3.715	0.708	501	0.809
-	3.705	0.7125	502.5	0.808
-	3.675	0.720	504	0.804
-	3.693	0.7325	514	0.802
-	3.660	0.740	515	0.798
-	3.601	0.760	521	0.789
-	3.618	0.762	525	0.789
-	3.627	0.767	530	0.789
-	3.655	0.775	539	0.788
-	3.624	0.7975	551	0.780
-	3.718	0.8125	575	0.781
-	3.682	0.829	581	0.775
-	3.779	0.8625	621	0.771

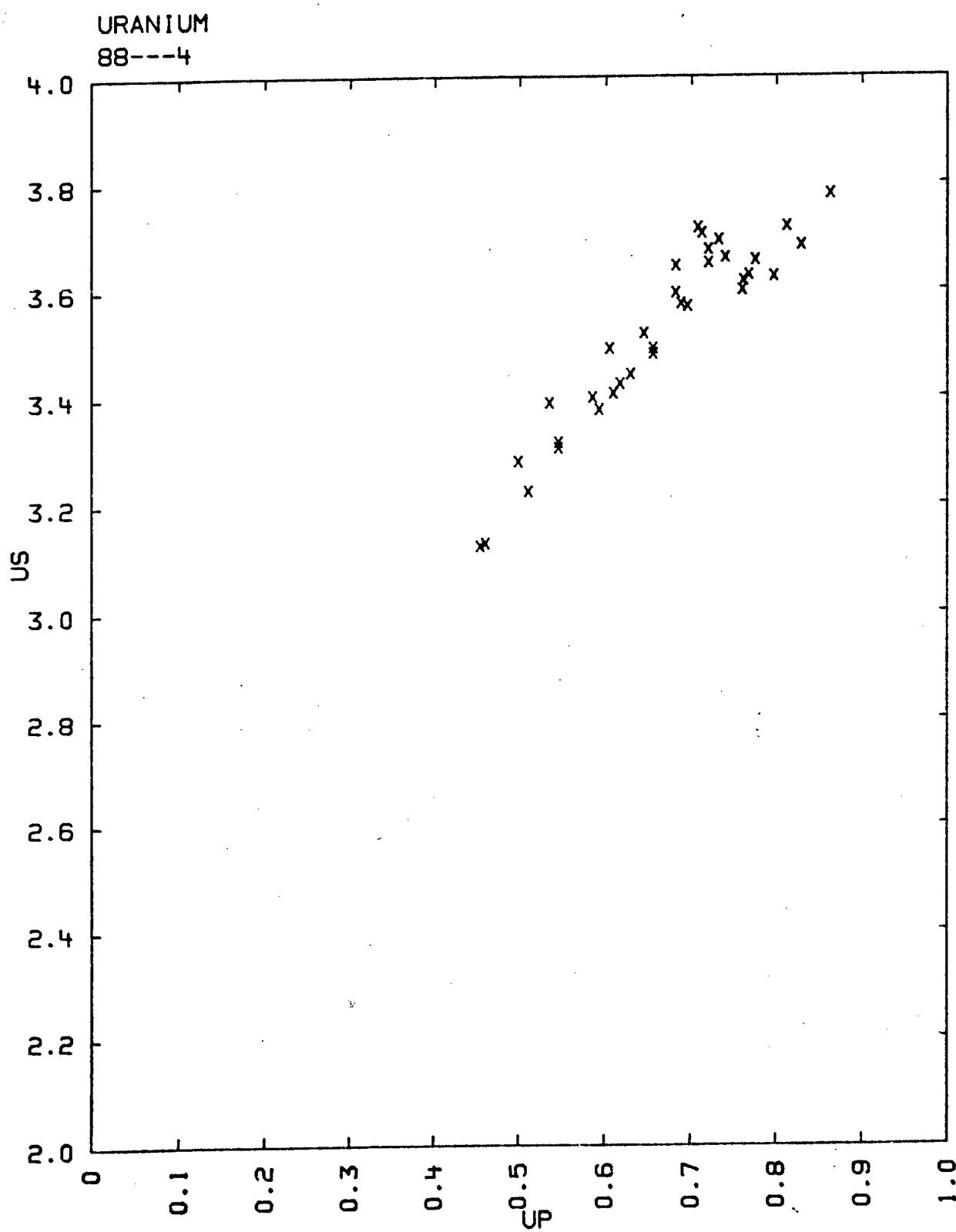
US = 2.317 + 1.85 UP KM/SEC FOR UP BETWEEN 0.0 AND 0.720 KM/SEC.
US = 2.317 + 1.66 UP KM/SEC FOR UP FROM 0.800 KM/SEC

COMMENTS:

- 1) SOURCE: VIARD, J.
LES ONDES DE DETONATION. PP. 383-389
EDITIONS DU CENTRE NATIONAL DE LA RECHERCHE SCIENTIFIQUE
15, QUAI ANATOLE-FRANCE, PARIS (1962)
- 2) EXPERIMENTAL TECHNIQUE B.
DATA REDUCTION TECHNIQUE B.
- 3) THE FREE SURFACE VELOCITY MEASUREMENTS WERE NOT USED BECAUSE OF
RELATIVELY HIGH UNCERTAINTIES IN TIME AND DISTANCE MEASUREMENTS.
 - A) THE UFS FOR URANIUM IS VERY LOW, THUS THE FREE SURFACE VELOCITY
HAS TO BE MEASURED OVER A VERY SHORT DISTANCE.
 - B) THE FREE SURFACE IN URANIUM IS PRECEDED BY A CLOUD OF URANIUM
OF LOW DENSITY THAT PROPAGATES WITH A MUCH HIGHER VELOCITY THAN
THE FREE SURFACE VELOCITY.
- 4) THE INITIAL DENSITY VARIED FROM 18.93-19.05 G/CC.
- 5) A PHASE TRANSITION IS OBSERVED AT 505 KILOBARS.
- 6) THE VALUE OF VOI WAS OBTAINED FROM CRYSTALLOGRAPHIC DATA ON METAL
AND ALLOY STRUCTURES BY A. TAYLOR AND BRENDA J. KAGLE
DOVER PUBLICATIONS, INC. NEW YORK, N. Y. (1963).

U06/14/77

TABLE I



88---5
URANIUM

U

VOL = 0.05294 CC/G.

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC., VELOCITIES IN KM/SEC., AND PRESSURE IN KILOBARS. CL IS THE LONGITUDINAL VELOCITY BEHIND THE SHOCK-FRONT. X DESIGNATES THE SHOCK THICKNESS GIVEN IN MICROSECONDS. U(PRO) DESIGNATES THE PROJECTILE VELOCITY AT IMPACT. A WEAK PRECURSOR PRECEDES THE MAIN WAVE. SEE COMMENT 3

TABLE

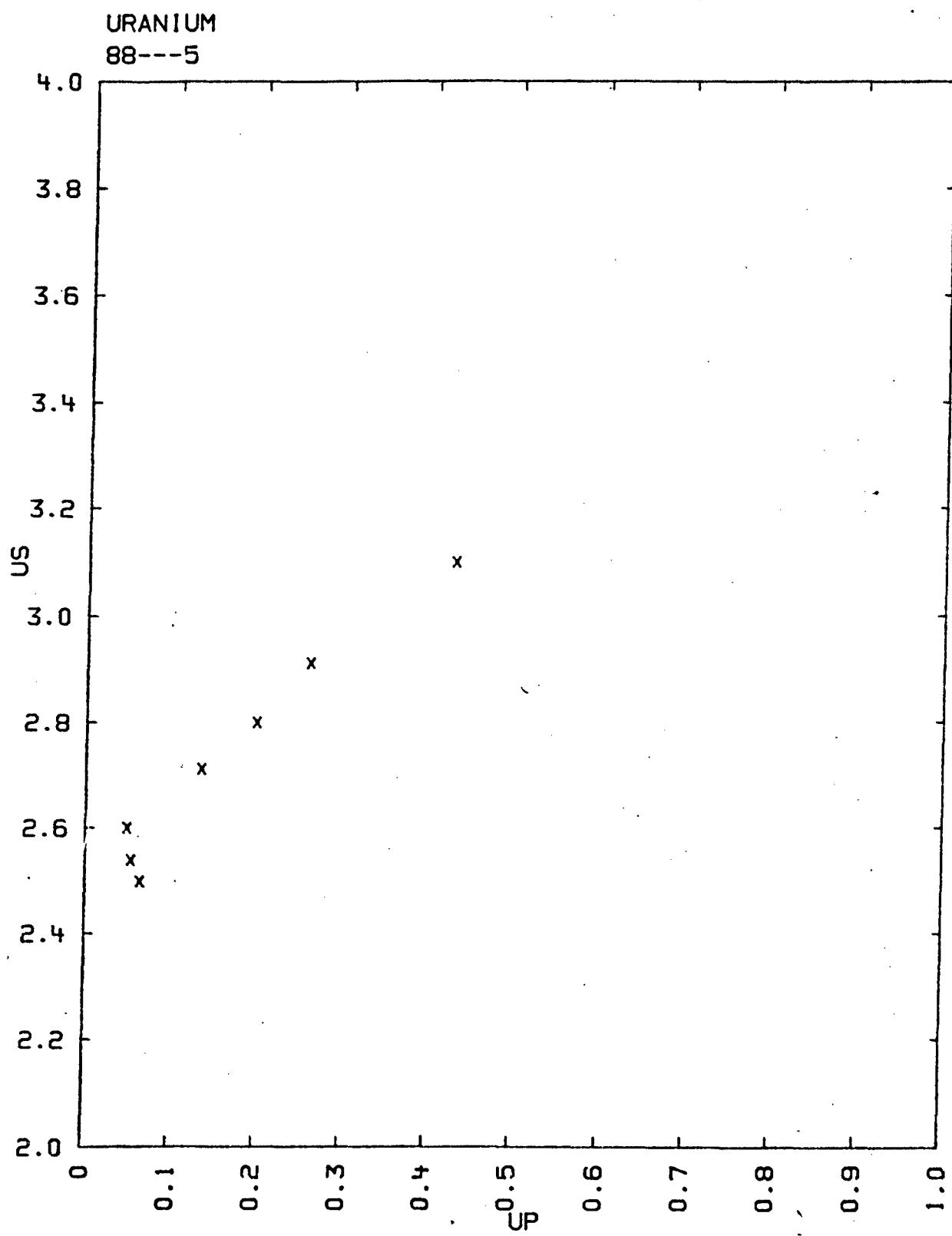
RHO01	US	UFS	UP	P	V/V0	CL	X	U(PRO)
18.89	2.54	0.110	0.055	26	0.978	3.45		0.317
-	2.80	0.395	0.198	105	0.929	3.54		0.390
-	2.60	0.099	0.050	25	0.816	3.48	2.5	0.105
-	2.91	0.520	0.260	143	0.910	3.52	0.1	0.525
-	3.1	0.855	0.428	251	0.862	3.4	0.1	0.849
-	2.50	0.130	0.065	31	0.974	3.43	1.0	0.152
-	2.71	0.270	0.135	60	0.950	3.32	1.4	0.267

US = 2.48 + 1.53 UP KM/SEC. SIGMA US = 0.05 KM/SEC.

COMMENTS:

- 1) SOURCE: TAYLOR, J. W.
LOS ALAMOS SCIENTIFIC LAB. REPORT LADD-5626 (1963)
LOS ALAMOS NEW MEXICO
- 2) EXPERIMENTAL TECHNIQUE G.
DATA REDUCTION TECHNIQUE D. ASSUMING $2U_P = U_{FS}$, WHERE U_{FS} IS THE FINAL FREE SURFACE VELOCITY OF A BROAD WAVE.
- 3) THE AVERAGE ELASTIC WAVE VELOCITY IS 3.45 KM/SEC. AND ITS FREE-SURFACE VELOCITY IS ABOUT 0.005 KM/SEC.. THE AVERAGE PRESSURE IS 1.6 KILOBARS.
- 4) VOL WAS OBTAINED FROM A.C.A. MONOGRAPH NUMBER 5 (AMERICAN CRYSTALLOGRAPHIC ASSOCIATION, POLYCRYSTAL BOOK SERVICE, BROOKLYN, N.Y., 1963) 2ND ED.
- 5) THE PRESSURE WAS ACHIEVED BY USING A GUN TO ACCELERATE A PROJECTILE WHICH WAS IMPACTED INTO A PLATE CONTAINING THE SAMPLE.

TABLE I



88---5
URANIUM

U

V0I = 0.05294 CC/G.

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC., VELOCITIES IN KM/SEC., AND PRESSURE IN KILOBARS. CL IS THE LONGITUDINAL VELOCITY BEHIND THE SHOCK-FRONT. X DESIGNATES THE SHOCK THICKNESS GIVEN IN MICROSECONDS. U(PRO) DESIGNATES THE PROJECTILE VELOCITY AT IMPACT. A WEAK PRECURSOR PRECEDES THE MAIN WAVE. SEE COMMENT 3

TABLE

RHO0I	US	UFS	UP	P	V/V0	CL	X	U(PRO)
18.89	2.54	0.110	0.055	26	0.978	3.45		0.317
-	2.80	0.395	0.198	105	0.929	3.54		0.390
-	2.60	0.099	0.050	25	0.816	3.48	2.5	0.105
-	2.91	0.520	0.260	143	0.910	3.52	0.1	0.525
-	3.1	0.855	0.428	251	0.862	3.4	0.1	0.849
-	2.50	0.130	0.065	31	0.974	3.43	1.0	0.152
-	2.71	0.270	0.135	60	0.950	3.32	1.4	0.267

US = 2.48 + 1.53 UP KM/SEC. SIGMA US = 0.05 KM/SEC.

COMMENTS:

- 1) SOURCE: TAYLOR, J. W.
LOS ALAMOS SCIENTIFIC LAB. REPORT LADC-5626 (1963)
LOS ALAMOS NEW MEXICO
- 2) EXPERIMENTAL TECHNIQUE G.
DATA REDUCTION TECHNIQUE D. ASSUMING 2UP = UFS, WHERE UFS IS THE FINAL FREE SURFACE VELOCITY OF A BROAD WAVE.
- 3) THE AVERAGE ELASTIC WAVE VELOCITY IS 3.45 KM/SEC. AND ITS FREE-SURFACE VELOCITY IS ABOUT 0.005 KM/SEC.. THE AVERAGE PRESSURE IS 1.6 KILOBARS.
- 4) V0I WAS OBTAINED FROM A.C.A. MONOGRAPH NUMBER 5 (AMERICAN CRYSTALLOGRAPHIC ASSOCIATION, POLYCRYSTAL BOOK SERVICE, BROOKLYN, N.Y., 1963) 2ND ED.
- 5) THE PRESSURE WAS ACHIEVED BY USING A GUN TO ACCELERATE A PROJECTILE WHICH WAS IMPACTED INTO A PLATE CONTAINING THE SAMPLE.

U06/14/77

88---6
URANIUM

U 99.8 WT. PERCENT

$V_0 = 0.05277 \text{ CC/G}$ $C_L = 2.97 \text{ KM/SEC}$ $C_0 = 2.63 \text{ KM/SEC}$
 $V_{01} = 0.05250 - \text{ ALPHA U}$ $C_S = 1.20 \text{ KM/SEC}$
 $V_{01} = 0.05329 - \text{ BETHA U}$

THE TABLE LISTS RHOO IN G/CC, VELOCITIES IN KM/SEC AND P IN KBARS. CU= COPPER, FS= FANSTEEL AND WF= WEIGHTING FACTOR.

TABLE

- - - - - SAMPLE - - - - - IMPACTOR - -

RHOO	US	UP	P	V/V0	MAT	U	WF
18.95	4.206	1.077	859.	.7439	FS	2.044	3
18.95	4.902	1.602	1488.	.6732	CU	3.885	3
18.95	4.940	1.613	1510.	.6735	CU	3.918	3
18.95	5.806	2.086	2295.	.6407	FS	4.134	3
18.95	7.596	3.214	4627.	.5769	CU	7.967	1

$$US = 2.443 + 1.582 \cdot UP \text{ KM/SEC}$$

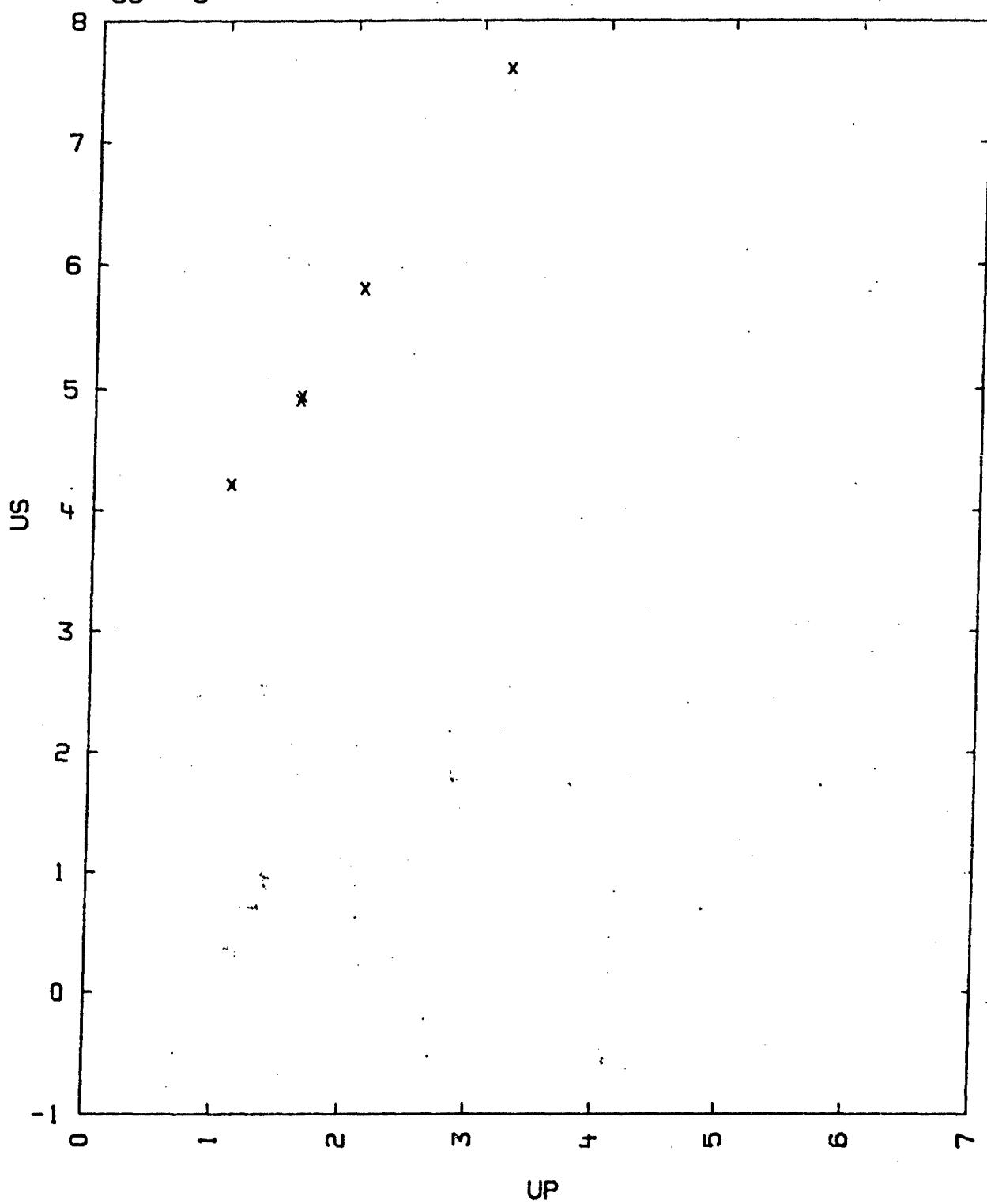
$$\text{SIG.US} = 0.072 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: ISBELL W.M., SHIPMAN F.H. AND JONES A.H.
HUGONIOT EQUATION OF STATE OF ELEVEN MATERIALS TO FIVE MBARS
MATERIALS SCIENCE LABORATORY REPORT: MSL-68-13
- 2) EXPERIMENTAL TECHNIQUE: A
DATA REDUCTION METHOD : A
- 3) NOMINAL UNCERTAINTIES ARE: $(\text{SIG.US})/\text{US} = .005$ AND $(\text{SIG.U})/\text{U} = .0005$
- 4) POISSON RATIO = 0.402; YIELD STRENGTH = 3.25 KBAR
TENSILE - = 8.55 KBAR
- 5) IMPURITIES IN PPM.: AL 9, B 0.3, CD 0.3; CR 3, CU 8, FE 10, MG 2,
MN 15, MO 2, NI 20, PB 1, SI 40, SM 1

TABLE I

URANIUM
88---6



88---7
URANIUMU 97. WT PERCENT
HO 3. -

$$V_0 = 0.05420 - 0.05458 \text{ CC/G} \quad C_L = 3.31 \text{ KM/SEC} \quad C_0 = 2.53 \text{ KM/SEC}$$

$$V_{01} = 0.0543 \quad C_S = 1.85 \text{ KM/SEC}$$

IN THE TABLES BELOW, VELOCITIES ARE GIVEN IN KM/SEC. PRESSURE IN KILOBAR AND DENSITY IN G/CC. STANDARD MATERIAL DESCRIBES THE IMPACTOR.

TABLE I

-----SAMPLE-----					-----STANDARD-----
RHO0	US	UP	P	V/V0	MATERIAL
18.450	3.17	0.41	240.	0.8707	URANIUM
18.450	3.33	0.49	301.	0.8529	URANIUM
18.450	3.67	0.72	488.	0.8038	URANIUM
18.450	3.73	0.75	516.	0.7989	URANIUM
18.450	4.24	1.11	868.	0.7382	URANIUM
18.450	4.54	1.29	1081.	0.7159	URANIUM
18.450	4.53	1.32	1103.	0.7086	URANIUM
18.450	5.19	1.76	1685.	0.6609	URANIUM
18.450	5.31	1.80	1763.	0.6610	URANIUM
18.450	5.36	1.84	1820.	0.6567	URANIUM
18.450	5.83	2.11	2270.	0.6381	URANIUM
18.450	6.06	2.36	2639.	0.6106	URANIUM
18.450	6.27	2.38	2753.	0.6204	URANIUM
18.450	6.51	2.54	3051.	0.6098	URANIUM
18.450	6.76	2.75	3430.	0.5932	URANIUM

US = $2.565 + 1.531 \cdot UP$ KM/SEC HUGONIOT FIT ADOPTED BY THE SOURCE
COMPARISON LEAST SQUARE FITS:

US = $A_0 + A_1 \cdot UP$ KM/SEC $A_0 = 2.553$ KM/SEC SIG $A_0 = 0.029$ KM/SEC $A_1 = 1.532$ - SIG $A_1 = 0.016$ -

SIG US = 0.049 -

TABLE II

-----SAMPLE-----					-----STANDARD-----
RHO0	US	UP	P	V/V0	MATERIAL US(ST)
18.320	2.96	0.24	130.	0.9189	2024 AL 5.98
18.330	3.07	0.31	174.	0.8990	2024 AL 6.18
18.480	3.25	0.41	246.	0.8738	2024 AL 6.45
18.480	3.20	0.42	248.	0.8687	2024 AL 6.45
18.330	3.24	0.43	255.	0.8673	2024 AL 6.48

RHO0	US	UP	P	V/V0	MATERIAL US(ST)
18.450	3.26	0.45	271.	0.8620	2024 AL 6.55
18.450	3.44	0.58	368.	0.8314	2024 AL 6.87
18.480	3.49	0.59	381.	0.8309	2024 AL 6.91
18.480	3.46	0.59	377.	0.8295	2024 AL 6.91
18.450	3.80	0.82	575.	0.7842	2024 AL 7.48
18.480	3.85	0.83	591.	0.7844	2024 AL 7.54
18.480	4.18	1.11	857.	0.7344	2024 AL 8.21
18.480	4.68	1.40	1211.	0.7009	2024 AL 8.95
18.480	5.10	1.55	1461.	0.6961	2024 AL 9.41
18.480	5.10	1.62	1527.	0.6824	2024 AL 9.54
18.450	5.30	1.69	1653.	0.6811	2024 AL 9.74
18.480	5.31	1.75	1717.	0.6704	2024 AL 9.86
18.480	5.64	1.88	1959.	0.6667	2024 AL 10.23
18.480	5.60	1.91	1977.	0.6589	2024 AL 10.28

US = 2.565 + 1.531 * UP KM/SEC

AO = 2.534 KM/SEC SIG AO = 0.026 KM/SEC

AI = 1.603 - SIG AI = 0.023 -

SIG US = 0.058 -

TABLE III

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
18.480	3.05	0.32	180.	0.8951	921-T AL	5.98
18.480	3.50	0.62	401.	0.8229	921-T AL	6.80
18.480	3.87	0.88	629.	0.7726	921-T AL	7.48
18.480	4.29	1.17	928.	0.7273	921-T AL	8.21
18.480	4.66	1.41	1214.	0.6974	921-T AL	8.84
18.480	5.20	1.71	1643.	0.6712	921-T AL	9.65
18.480	5.60	1.95	2018.	0.6518	921-T AL	10.26

US = 2.565 + 1.531 * UP KM/SEC

AO = 2.517 KM/SEC SIG AO = 0.038 KM/SEC

AI = 1.557 - SIG AI = 0.030 -

SIG US = 0.044 -

TABLE IV

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
18.480	2.86	0.19	100.	0.9336	CU	4.27
18.480	3.03	0.26	146.	0.9142	CU	4.41
18.480	3.17	0.39	228.	0.8770	CU	4.64
18.480	3.23	0.41	245.	0.8731	CU	4.68
18.330	3.29	0.47	283.	0.8571	CU	4.78
18.480	3.35	0.53	328.	0.8418	CU	4.87

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URANIUM

RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
18.450	3.42	0.55	347.	0.8392	CU	4.93
18.490	3.79	0.80	561.	0.7889	CU	5.37
18.480	3.80	0.82	576.	0.7842	CU	5.40
18.450	3.77	0.82	570.	0.7825	CU	5.40
18.450	3.78	0.84	586.	0.7778	CU	5.43
18.480	3.82	0.84	593.	0.7801	CU	5.44
18.480	3.83	0.86	609.	0.7755	CU	5.47
18.450	3.91	0.88	635.	0.7749	CU	5.51
18.480	4.25	1.14	895.	0.7318	CU	5.97
18.480	4.55	1.30	1093.	0.7143	CU	6.27
18.450	4.57	1.35	1138.	0.7046	CU	6.35
18.480	4.68	1.42	1228.	0.6966	CU	6.48
18.480	5.10	1.67	1574.	0.6725	CU	6.93
18.480	5.09	1.67	1571.	0.6719	CU	6.94
18.450	5.31	1.80	1763.	0.6610	CU	7.18
18.480	5.48	1.91	1934.	0.6515	CU	7.37
18.450	5.47	1.91	1928.	0.6508	CU	7.38
18.480	5.73	2.00	2118.	0.6510	CU	7.57
18.480	5.75	2.04	2168.	0.6452	CU	7.62
18.480	5.72	2.04	2156.	0.6434	CU	7.61
18.480	5.78	2.07	2211.	0.6419	CU	7.67
18.480	5.95	2.21	2430.	0.6286	CU	7.91
18.480	5.88	2.24	2434.	0.6190	CU	7.94
18.480	6.10	2.29	2581.	0.6246	CU	8.07
18.320	6.12	2.31	2590.	0.6225	CU	8.08

US = 2.565 + 1.531 * UP KM/SEC

AO = 2.552 KM/SEC SIG AO = 0.018 KM/SEC

AI = 1.536 - SIG AI = 0.012 -

SIG US = 0.046 -

TABLE V

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
18.330	3.49	0.57	365.	0.8367	FE	4.93
18.330	3.85	0.82	579.	0.7870	FE	5.49
18.330	4.56	1.31	1095.	0.7127	FE	6.53
18.330	5.04	1.61	1487.	0.6806	FE	7.15
18.320	5.77	2.05	2167.	0.6447	FE	8.05
18.320	6.13	2.31	2594.	0.6232	FE	8.54

US = 2.565 + 1.531 * UP KM/SEC

AO = 2.592 KM/SEC SIG US = 0.032 KM/SEC

AI = 1.532 - SIG AI = 0.020 -

SIG US = 0.031 -

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE A (TABLE I), B (THE REST)
- 3) TABLE I LISTS THE STANDARD HUGONIOT OBTAINED WITH SYMMETRIC IMPACT.
TABLES II THROUGH V CHECK THE INTERNAL CONSISTENCY OF THE OTHER
STANDARDS OBTAINED WITH SYMMETRIC IMPACT.
- 4) ALSO GIVEN ARE $V \cdot DP/DE = 2.03$, $(DV/DT)/V = 3.84E-5$ PER DEG. AND
 $CP = 0.121$ JOULES PER G. DEG.
- 5) VOI WAS CALCULATED FROM THE ALPHA PRIME - ALPHA DOUBLE PRIME STRUC-
TURE M!X OBTAINED BY WATER QUENCHING THE GAMMA PHASE TO YIELD LATTIC
CONSTANTS $A_0 = 2.869$ $B_0 = 5.782$ $C = 4.958$ ANGSTROMS AND GAMMA =
91.12 DEGREES. FOR THE ABOVE AVERAGE MOLECULAR WEIGHT OF
227.90 G/MOL. (3 WT PERCENT Mo). K. TANGRIC AND G.I. WILLIAMS, J.
NUCL. MATS. NO. 4, P. 226, (1961).
- 6) THE 921-T AL PLATE STOCK USED TO OBTAIN THE CROSS CHECK DATA IN
TABLE III IS SLIGHTLY POROUS

TABLE I

URANIUM
88---7

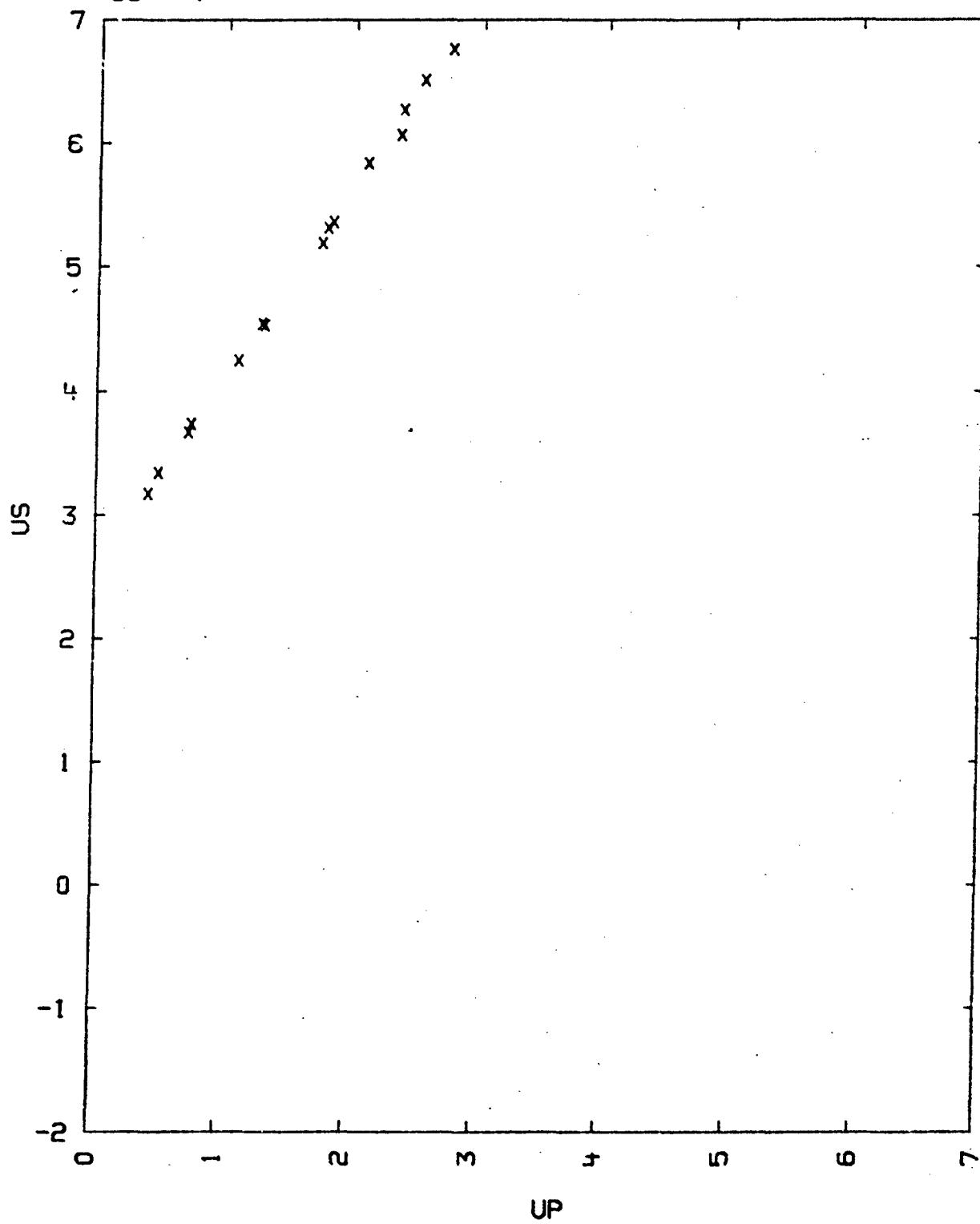


TABLE II

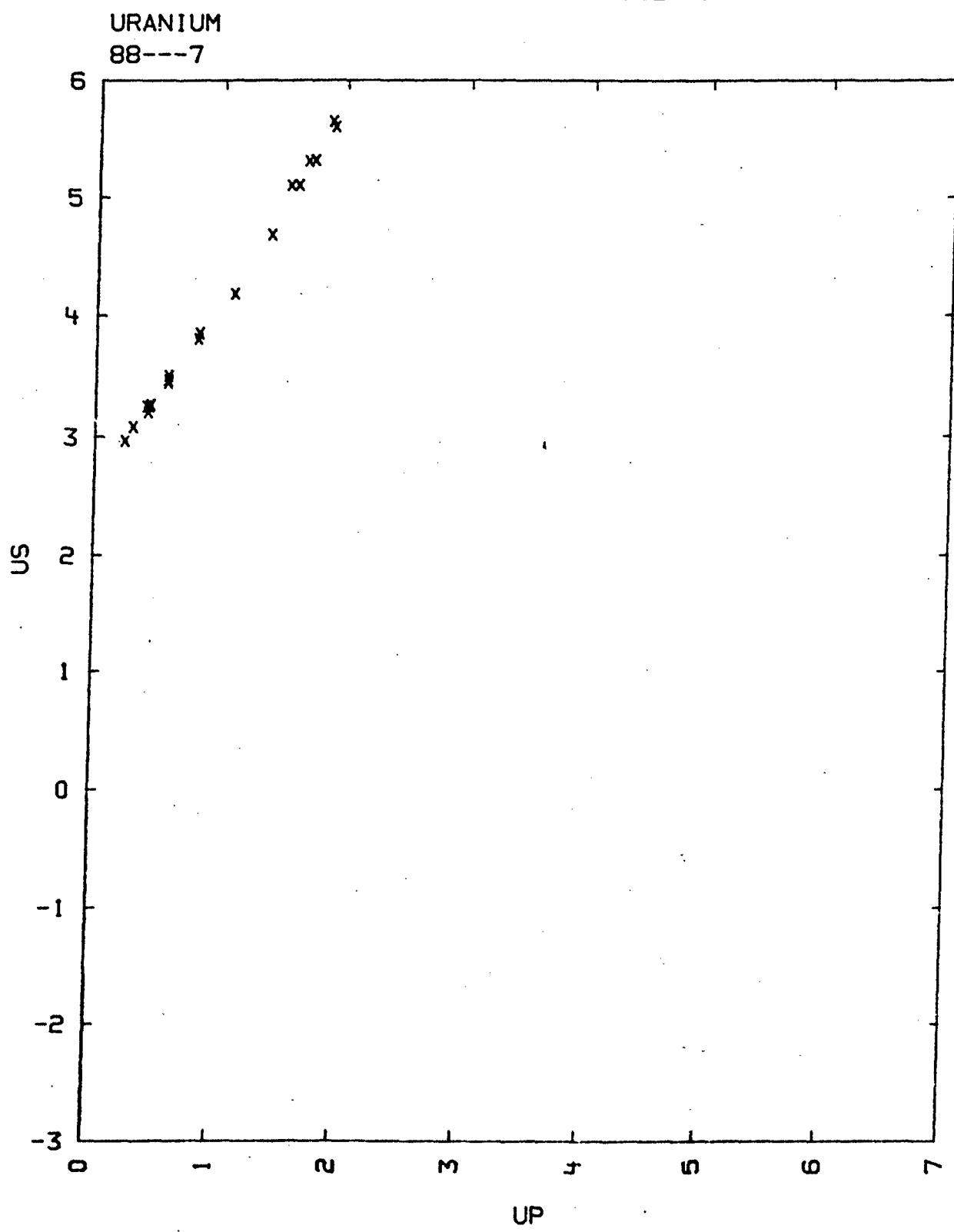


TABLE III

URANIUM
88---7

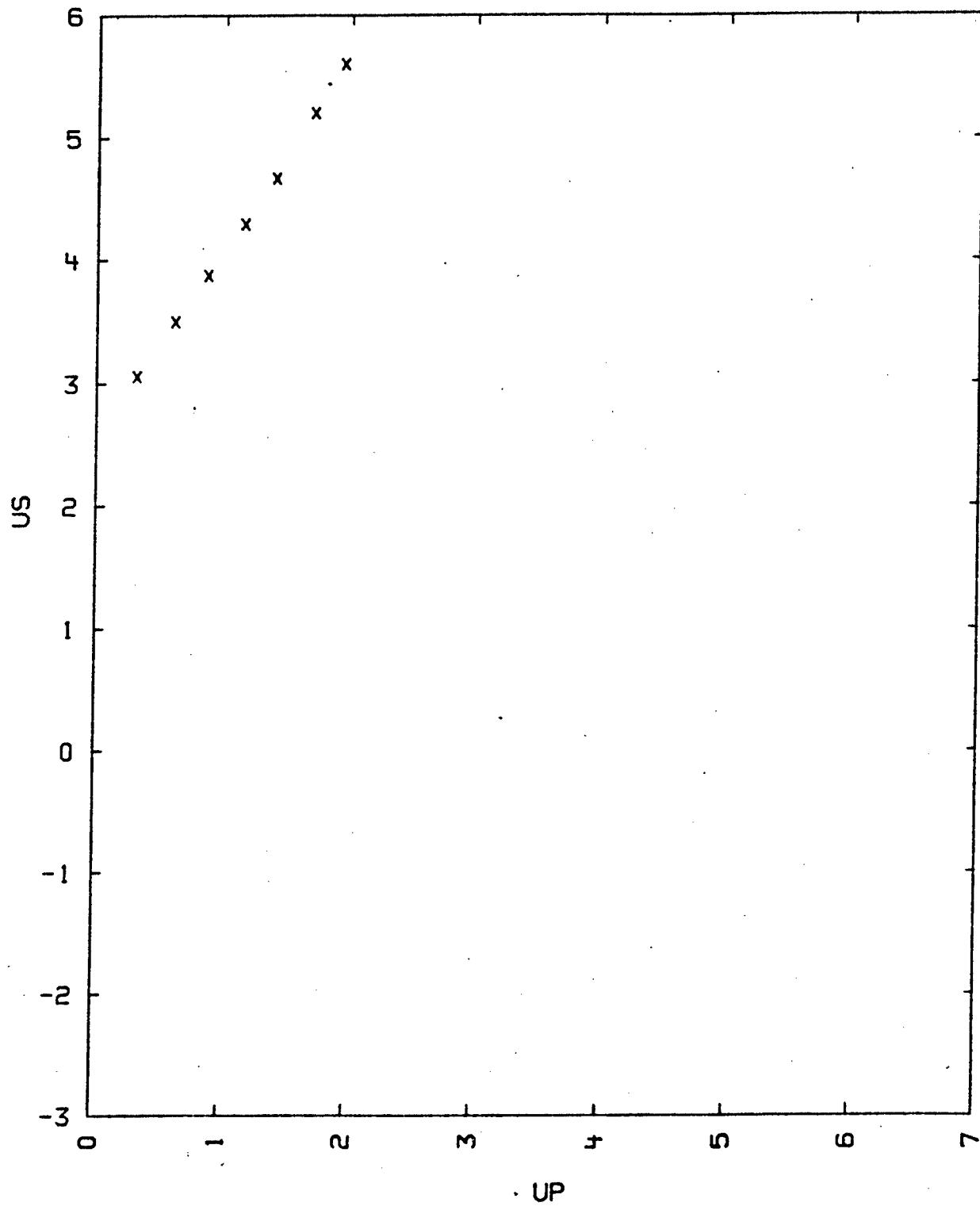


TABLE IV

URANIUM

88---7

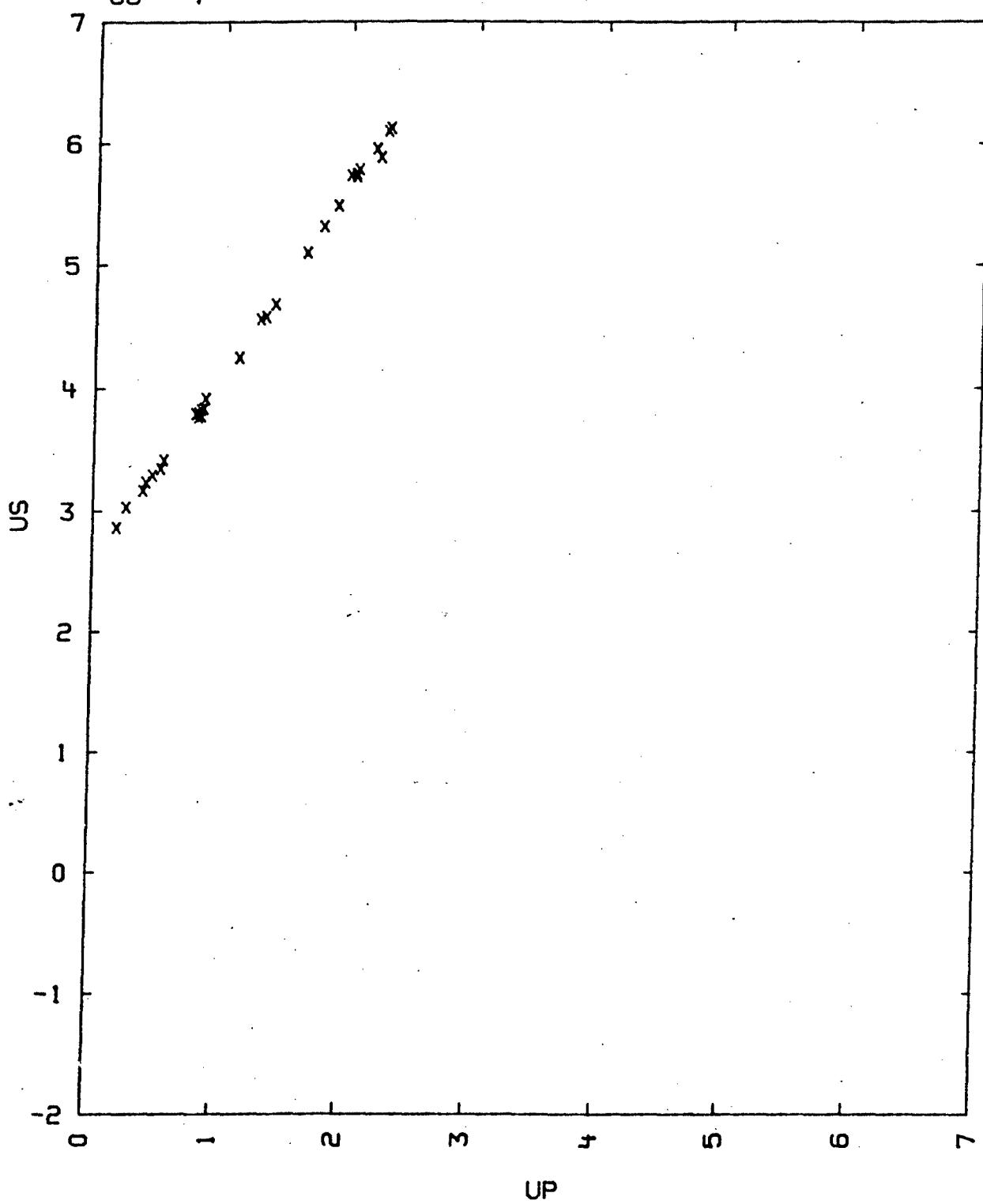
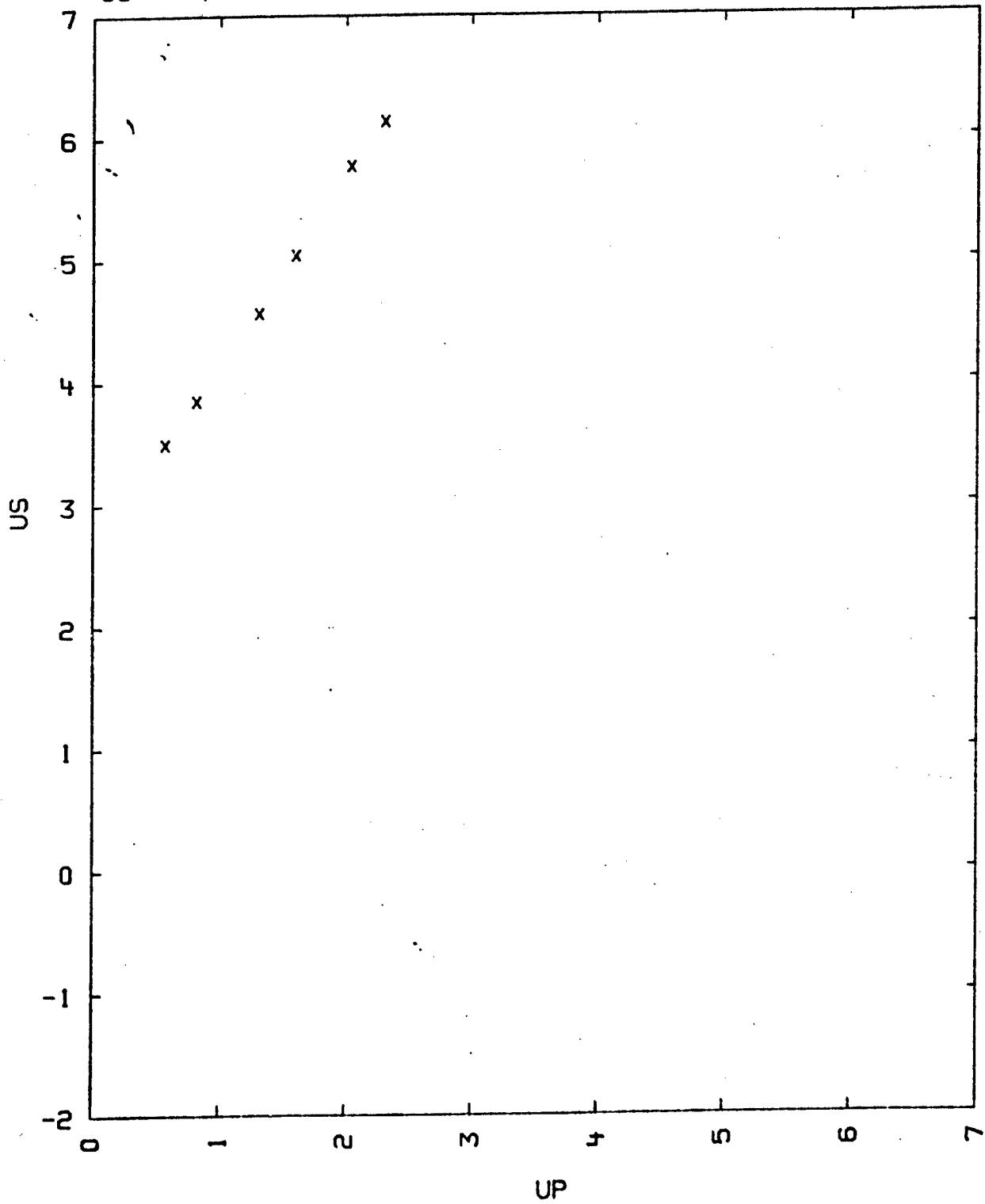


TABLE V

URANIUM
88---7.



90---I
THORIUM

TH 99.8 PER CENT OR GREATER

$V_0 = 0.0856 \text{ CC/G.}$
 $V_{01} = 0.0833 \text{ CC/G.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN MM/MICROSEC.
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

RHO0	US	UFS	UP	P	V/V0
11.68	3.497	2.112	1.043	426.0	0.7017
-	3.192	1.604	0.812	302.7	0.7456
-	2.954	1.246	0.620	213.9	0.7901
-	2.900	1.198	0.571	193.4	0.8031

$US = 2.170 + 1.268 UP \text{ KM/SEC}$ SIGMA US = 0.2 PERCENT

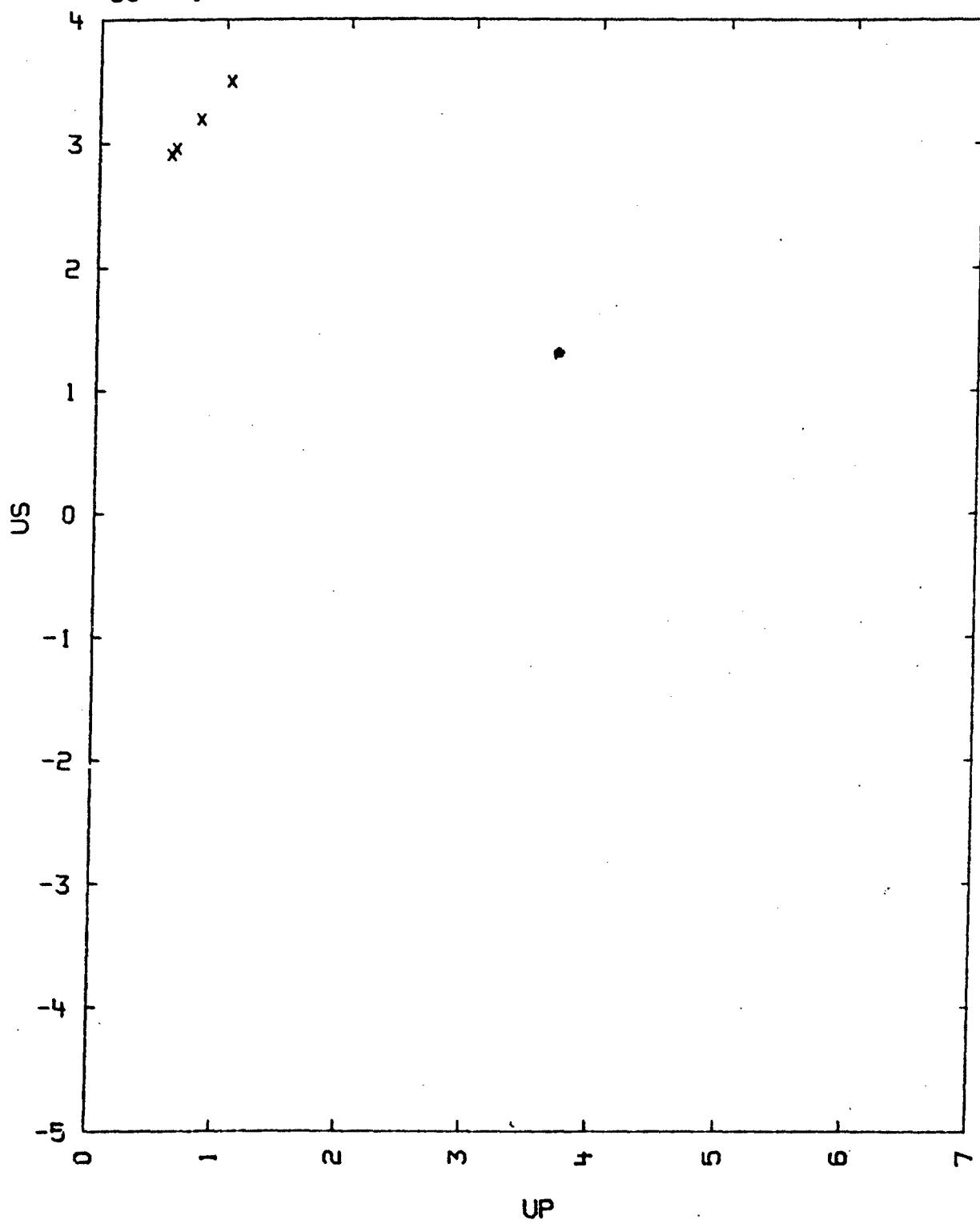
COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PER CENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 4) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS

TABLE I

THORIUM

90---1



90---2
THORIUM

TH

$V_0 = 0.0856 \text{ CC/G.}$
 $V_{01} = 0.0847 \text{ CC/G.}$

 $C_0 = 2.05 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURES IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0	PRESSURE BRASS STANDARD
------	----	----	---	------	----------------------------

11.68	4.51	1.90	1003	0.578	1028
-	4.53	1.94	1026	0.572	1022
-	5.16	2.32	1400	0.550	1403
-	5.11	2.31	1378	0.548	1385
-	5.09	2.33	1384	0.543	1396
-	5.10	2.36	1405	0.538	1414

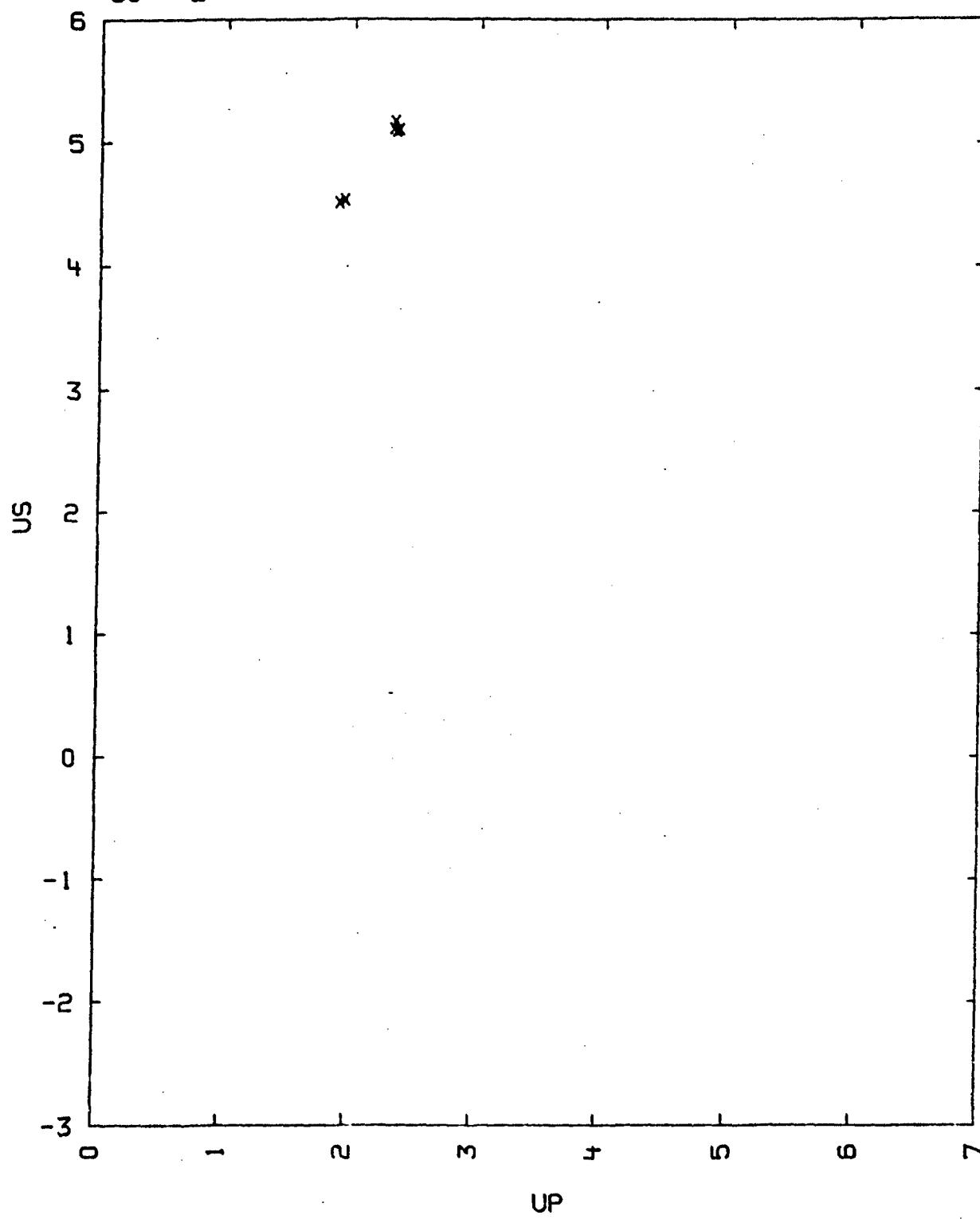
$US = 2.132 + 1.278 UP \text{ KM/SEC.}$ SIGMA US = 0.74 PERCENT

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
J. APPL. PHYS., VOL. 31, P. 1253 (1960)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS BRASS
- 3) FOR THE BRASS STANDARD DATA SEE MATERIAL 36-33
- 4) V_{01} WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW HILL BOOK CO.)
2ND ED.

TABLE I

THORIUM
90---2



92---1
BERYLLIUM

BE 100 PERCENT

$V_0 = .5402 \text{ CC/G}$
 $V_{01} = .540 \text{ CC/G}$

$C_0 = 8.137 \text{ KM/SEC}$
 $C_B = 7.934 \text{ KM/SEC}$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN MM/MICROSEC AND PRESSURE IN KILOBARS.

TABLE

-----BERYLLOUM-----					ALUMINUM STANDARD	
RHO0	US	UP	P	V/V0	RHO0	US
1.851	9.174	.931	158.1	.8985	2.784	6.566
	9.137	.932	157.6	.8980	2.784	6.566
10.146	1.818	341.4	.8208	2.784	7.670	
10.128	1.819	341.0	.8204	2.784	7.670	
10.111	1.821	340.8	.8199	2.784	7.670	
10.758	2.336	465.2	.7829	2.788	8.306	
11.143	2.689	554.6	.7587	2.788	8.735	
11.120	2.691	553.9	.7580	2.788	8.735	
11.530	3.001	640.5	.7397	2.788	9.117	
11.451	3.009	637.8	.7372	2.788	9.117	
12.165	3.620	815.1	.7024	2.784	9.858	
12.160	3.621	815.0	.7022	2.784	9.858	
12.445	3.799	875.1	.6947	2.783	10.08	
12.491	3.816	882.3	.6945	2.780	10.11	
12.535	3.951	916.7	.6848	2.788	10.25	

$US = 8.056 + 1.145 UP$, SIG.US = 0.038 KM/SEC.

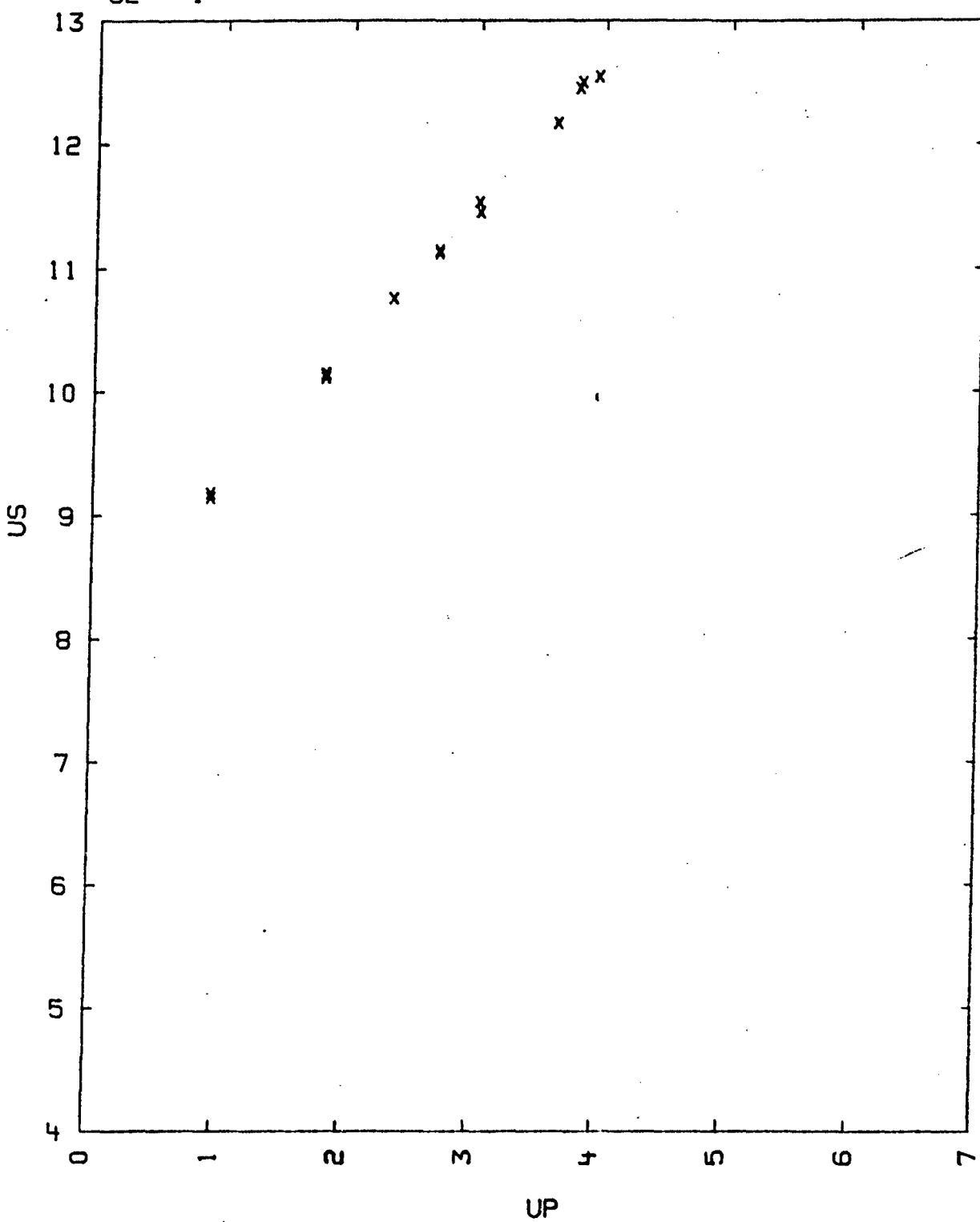
COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
REPORT NO. GMY-6-566, PP. 51-62 (1964)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 2024 ALUMINUM
- 3) THE HUGONIOT ELASTIC LIMIT IS HIGHLY DEPENDENT ON CRYSTAL ORIENTATION
POPE,L.E. AND JOHNSON J.N. J. APPL. PHYS. V.46 P.720 (1975)

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TABLE I

BERYLLIUM
92---1



92---2
BERYLLIUM

BE 99.8 PER CENT OR GREATER

$V_0 = 0.5420 \text{ CC/G.}$ $C_L = 12.89 \text{ KM/SEC.}$ $C_0 = 7.81 \text{ KM/SEC.}$
 $V_{01} = 0.5412 \text{ CC/G.}$ $C_S = 8.880 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS, AND DENSITY IN G/CC.

TABLE

RHO0	US	UFS	UP	P	V/V0
1.845	9.044	1.697	0.847	141.3	0.9063
-	8.934	1.739	0.865	142.6	0.9032
-	9.112	2.358	1.189	199.9	0.8695
-	9.332	2.364	1.221	210.2	0.8692
-	9.851	3.422	1.730	314.4	0.8244
-	9.832	3.235	1.609	291.9	0.8364
-	9.633	3.189	1.592	282.9	0.8347

$$US = 8.069 + 1.023 UP \text{ KM/SEC} \quad \text{SIGMA US} = 1.2 \text{ PERCENT}$$

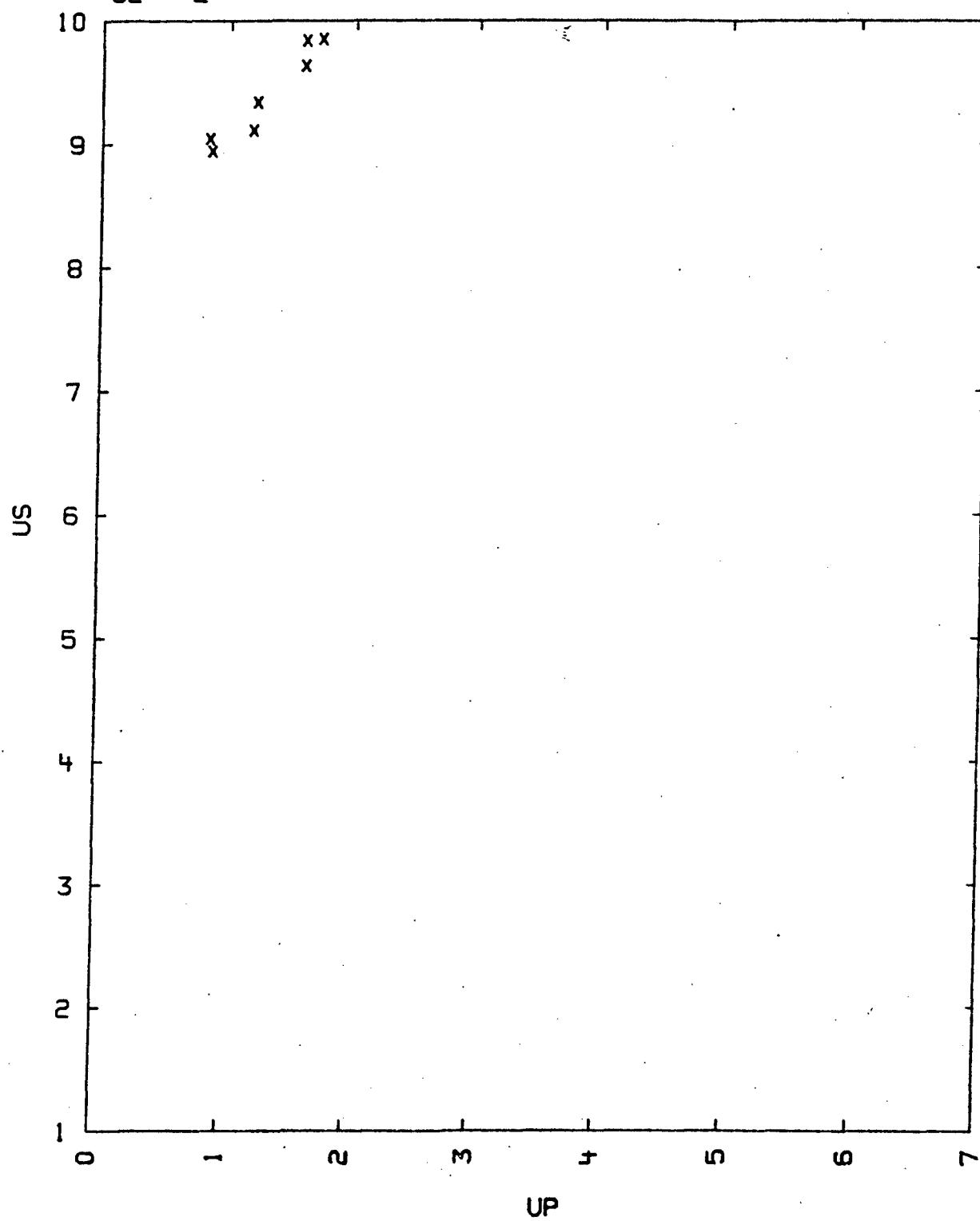
COMMENTS:

- 1) SOURCE: WALSH, J. M., RICE, M. H., MCQUEEN, R. G. AND YARGER, F. L.
PHYS. REV., VOL. 108, P. 169 FF. (1957)
LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 24ST ALUMINUM
- 3) IMPURITIES WERE MEASURED BY SPECTROCHEMICAL ANALYSIS.
- 4) CL AND CS WERE OBTAINED FROM D. GRAY, AMERICAN INSTITUTE OF PHYSICS
HANDBOOK, 1957, 2ND. ED., P. 33-82
- 5) THE PROBABLE ERROR PER DATA POINT IS 0.7 PER CENT IN SHOCK VELOCITY
FOR A GIVEN FREE-SURFACE VELOCITY AND APPROXIMATELY 1 PERCENT IN
COMPRESSION AT A GIVEN PRESSURE.
- 6) V01 WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW HILL BOOK CO.)
2ND ED.

TABLE I

BERYLLIUM

92---2



92---3
BERYLLIUM POROUS

BE

$$V_0 = 0.772 \text{ CC/G}$$

$$V_{OI} = 0.5420 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC AND PRESSURE IN KILOBARS.

TABLE

RHO0	SAMPLE			AL BASE PLATE		
	US	UFS	UP	P	V/V0	PRESSURE
1.274	3.93	1.57	1.40	70	0.644	165
1.31	4.96	2.43	1.68	109	0.661	224
1.295	5.43	3.08	2.11	148	0.612	296
1.31	6.17	3.57	2.24	182	0.637	335

$$US = 0.79 + 2.33 UP \text{ KM/SEC}$$

$$\text{SIGMA US} = 0.302 \text{ KM/SEC}$$

COMMENTS:

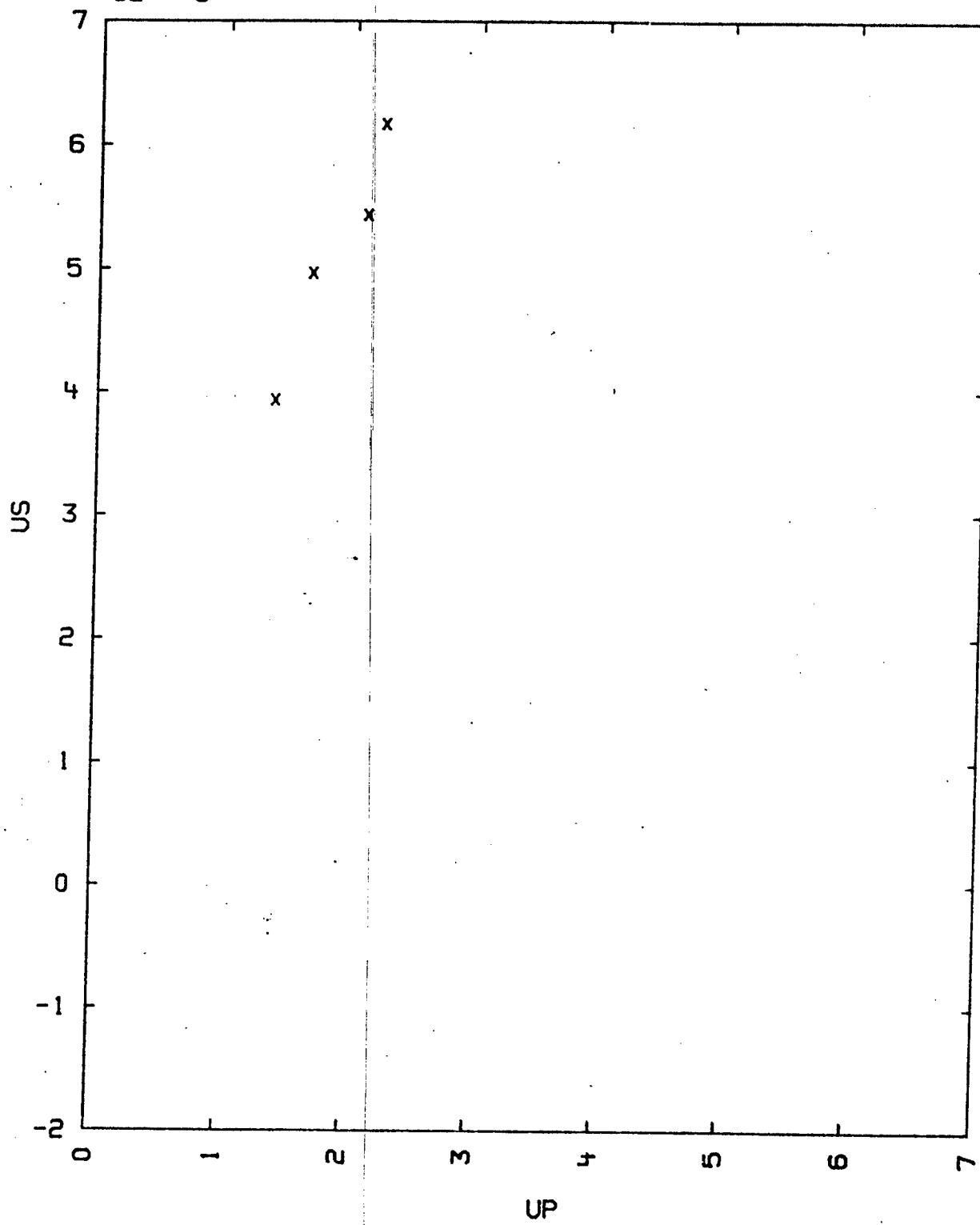
- 1) SOURCE: COMPILER
L.R.L. EQUATION OF STATE FILE
LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B. STANDARD MATERIAL 2024 AL ALLOY.
DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF V0I WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES
(DOVER PUBLICATIONS, INC., NEW YORK, N.Y., 1963).

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TABLE I

BERYLLIUM POROUS

92---3



92---4
BERYLLOUM

TYPE	S-200	I-400	WT. PERCENT
BE	98.18	95.77	
BE-O	2.00	3.96	-
C	0.126	0.18	-
FE	0.163	0.16	-
AL	0.054	0.04	-
MG	0.035	0.02	-
SI	0.080	0.04	-
MN	----	0.01	-
REST	0.04 MAX	0.10	-

S-200 :

V0 = 0.5385 CC/G CL= 12.916 KM/SEC CO= 7.87 KM/SEC
CS= 8.86 KM/SEC

I-400 :

V0 = 0.5316 CC/G CL= 12.83 KM/SEC CO= 7.83 KM/SEC
CS= 8.80 KM/SEC

V0I= 0.5421 CC/G

THE TABLE LISTS RHO0 IN G/CC. VELOCITIES IN KM/SEC AND P IN KBARS.
FS= FANSTEEL, CU= COPPER AND WF= WEIGHTING FACTOR.

TABLE

- - - - - SAMPLE - - - - - IMPACTOR - - -

MAT	RHO0	US	UP	P	V/V0	MAT	U	WF
S-200	1.851	9.965	1.631	302.	.8363	FS	2.205	3
-	1.851	10.266	1.903	363.	.8147	CU	2.695	3
-	1.851	11.779	3.459	757.	.7063	FS	4.330	3
-	1.851	12.218	3.990	905.	.6734	CU	5.596	1
-	1.851	13.034	4.792	1160.	.6324	FS	7.734	3
-	1.851	13.970	5.678	1473.	.5936	CU	7.945	3
-	1.851	14.324	6.139	1633.	.5714	FS	7.734	1
I-400	1.881	10.981	2.549	526.	.7679	FS	3.191	0

US = 8.390 + 0.975*UP KM/SEC

SIG.US = 0.017 KM/SEC

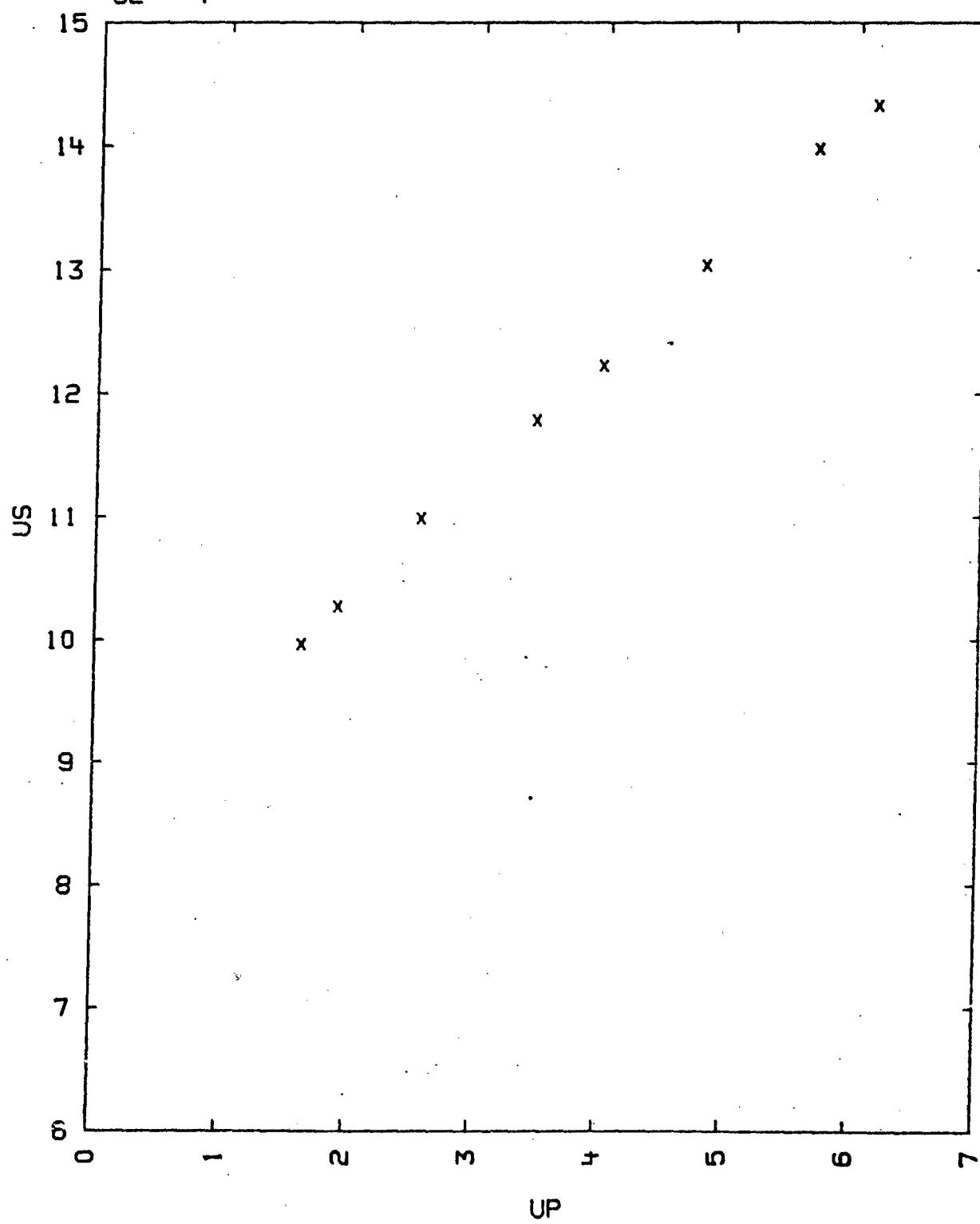
COMMENTS:

- 1) SOURCE: ISBELL W.M., SHIPMAN F.H. AND JONES A.H.
HUGONIOT EQUATION OF STATE OF ELEVEN MATERIALS TO FIVE MBARS.
MATERIALS SCIENCE LABORATORY REPORT: MSL-68-13
- 2) EXPERIMENTAL TECHNIQUE: A
DATA REDUCTION METHOD : A
- 3) NOMINAL UNCERTAINTIES ARE: (SIG.US)/US = .005 AND (SIG.U)/U = .005
- 4) ALSO LISTED: POISSONS RATIO: 0.055; ULTIMATE TENSILE STRENGTH:
3.98 KBAR (S-200); 4.60 KBAR (I-400); YIELD STRENGTH: 2.58 KBAR
(S-200); 3.86 KBAR (I-400)
- 5) V0I OBTAINED FROM WYCKOFF CRYSTAL STRUCTURES
(JOHN WILEY AND SONS, N.Y. 1963) VOL. I

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TABLE I

BERYLLIUM
92---4



92---5
BERYLLIUM

BE COMMENT 5
0 - -

$V_0 = 0.5402 \text{ CC/G}$ $C_L = 13.18 \text{ KM/SEC}$ $C_0 = 8.14 \text{ KM/SEC}$
 $V_{01} = 0.5421 \text{ CC/G}$ $C_S = 8.98 \text{ KM/SEC}$ $C_B = 7.93 \text{ KM/SEC}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
1.850	8.99	0.34	57.	0.9622	CU	4.31
1.850	8.58	0.48	76.	0.9441	2024 AL	5.96
1.850	8.67	0.49	79.	0.9435	2024 AL	5.98
1.850	8.60	0.50	80.	0.9419	2024 AL	5.98
1.850	8.59	0.50	79.	0.9418	2024 AL	5.98
1.850	8.78	0.68	110.	0.9226	2024 AL	6.22
1.851	9.04	0.88	147.	0.9027	2024 AL	6.47
1.851	9.01	0.88	147.	0.9023	2024 AL	6.47
1.850	9.05	0.93	156.	0.8972	2024 AL	6.53
1.850	9.02	0.93	155.	0.8969	2024 AL	6.53
1.850	9.05	0.94	157.	0.8961	2024 AL	6.54
1.850	9.05	1.02	171.	0.8873	921-T AL	6.43
1.850	9.03	1.02	170.	0.8870	921-T AL	6.43
1.850	9.42	1.25	218.	0.8673	2024 AL	6.92
1.850	9.06	1.25	210.	0.8620	2024 AL	6.91
1.850	9.02	1.26	210.	0.8603	2024 AL	6.91
1.850	9.39	1.34	233.	0.8573	921-T AL	6.86
1.850	9.36	1.34	232.	0.8568	921-T AL	6.86
1.850	9.96	1.75	322.	0.8243	2024 AL	7.55
1.850	9.94	1.75	322.	0.8239	2024 AL	7.55
1.851	10.00	1.76	326.	0.8240	2024 AL	7.56
1.851	9.98	1.76	325.	0.8236	2024 AL	7.56
1.851	9.97	1.76	325.	0.8235	2024 AL	7.56
1.850	10.01	1.77	328.	0.8232	921-T AL	7.43
1.850	9.97	1.78	328.	0.8215	921-T AL	7.43
1.850	9.97	1.80	332.	0.8195	2024 AL	7.60
1.850	10.01	1.84	341.	0.8162	921-T AL	7.52
1.850	9.99	1.85	342.	0.8148	921-T AL	7.52
1.850	10.58	2.21	433.	0.7911	2024 AL	8.12
1.851	10.60	2.27	445.	0.7858	2024 AL	8.19
1.851	10.98	2.62	532.	0.7614	2024 AL	8.61
1.851	10.96	2.63	534.	0.7600	2024 AL	8.61
1.853	11.07	2.78	570.	0.7489	921-T AL	8.72
1.850	11.41	2.92	616.	0.7441	2024 AL	8.98
1.851	11.36	2.93	616.	0.7421	2024 AL	8.99
1.851	11.29	2.94	614.	0.7396	2024 AL	8.99
1.855	11.42	3.16	669.	0.7233	2024 AL	9.24

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BERYLLIUM

RHO0	US	UP	P	V/V0	MATERIAL US(IST)
1.851	11.84	3.38	741.	0.7145	921-T AL 9.47
1.851	11.78	3.39	739.	0.7122	921-T AL 9.47
1.835	11.86	3.44	749.	0.7099	921-T AL 9.54
1.851	11.98	3.54	785.	0.7045	2024 AL 9.72
1.851	11.99	3.54	786.	0.7048	2024 AL 9.72
1.851	12.08	3.72	832.	0.6921	921-T AL 9.88
1.851	12.27	3.72	845.	0.6968	2024 AL 9.94
1.851	12.31	3.74	852.	0.6962	2024 AL 9.96
1.851	12.35	3.87	885.	0.6866	2024 AL 10.11

US = 7.998 + 1.124 * UP KM/SEC

SIGMA US = 0.137 KM/SEC

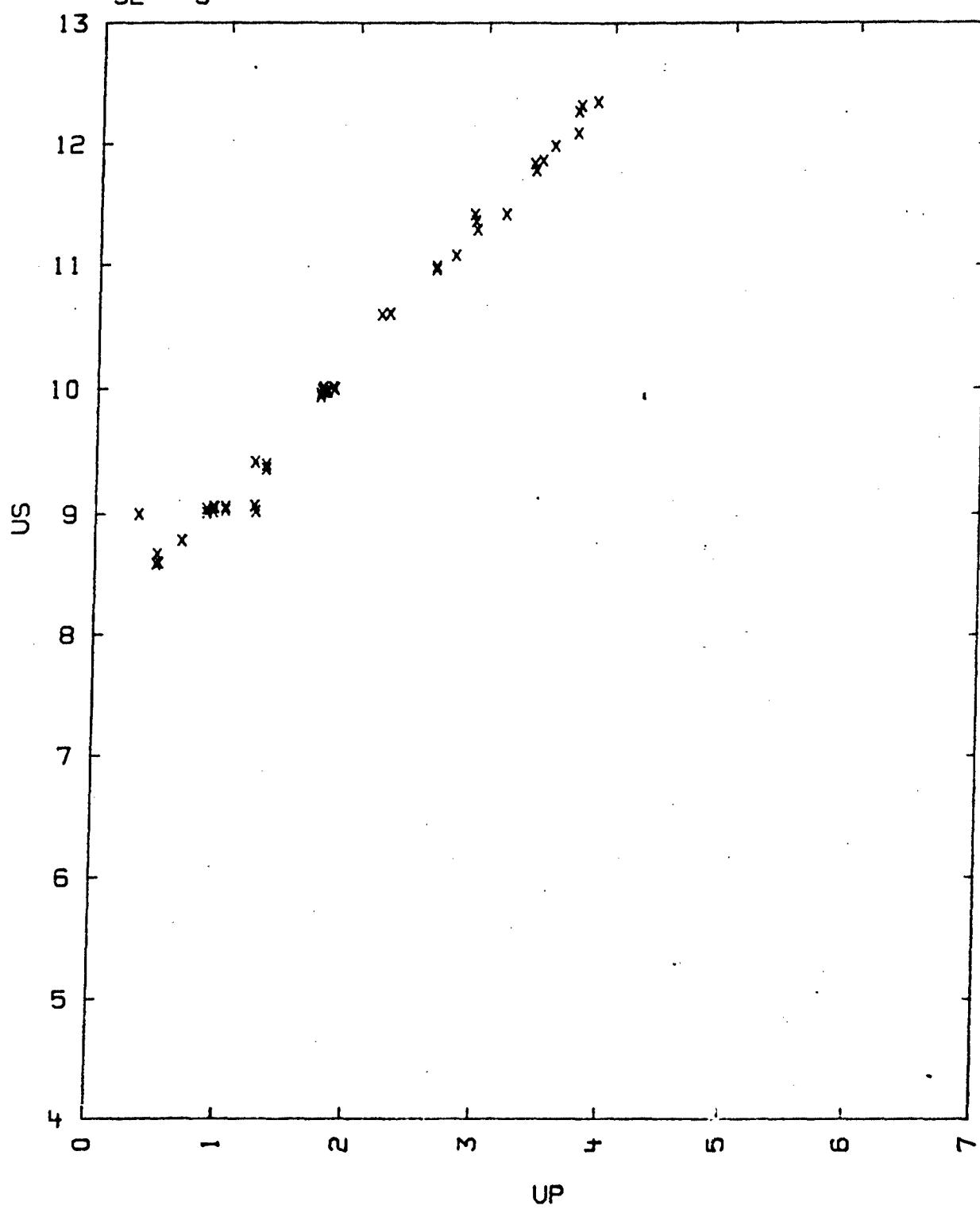
COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B
- 3) VOI FROM WYCKOFF, CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, N.Y.
(1963) VI
- 4) V'(DP/DE) = 1.16
- 5) OXYGEN CONTENT PROBABLY SMALL. SAMPLES WERE SINTERED.
- 6) HUGONIOT ELASTIC LIMIT 40. KBAR (C-AXIS), 4. KBAR. BASAL PLANE

TABLE I

BERYLLIUM

92---5



93---I
MAGNESIUM

THE DOW METAL PRODUCTS COMPANY ALLOY AZ31B

MG 96.0 PERCENT BY WEIGHT
AL 3.0 PERCENT -
ZN 1.0 PERCENT -

$V_0 = 0.563 \text{ CC/G}$ $C_L = 5.60 \text{ KM/SEC}$ $C_0 = 4.46 \text{ KM/SEC}$
 $V_{01} = 0.565 \text{ CC/G}$ $C_S = 2.93 \text{ KM/SEC}$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC., AND PRESSURE IN KILOBARS.

TABLE

SAMPLE						BASE PLATE
RHO0	US	UFS	UP	P	V/V0	P
1.780	7.49	4.87	2.40	320	0.679	439
1.774	6.12	2.54	1.28	138	0.792	194
1.774	7.21	4.75	2.23	285	0.691	398
1.776	8.24	6.31	3.09	453	0.625	620
1.774	5.47	1.47	0.73	71	0.867	98
1.774	6.12	2.35	1.17	127	0.809	176
1.775	8.64	6.81	3.33	511	0.615	693
1.776	8.11	6.08	3.01	434	0.629	592
1.775	6.67	3.39	1.68	200	0.748	275
1.775	5.51	1.49	0.73	71	0.868	100
1.775	6.32		1.43	160	0.774	222
1.775	7.01	3.91	1.99	248	0.716	339
1.779	7.31	4.81	2.30	300	0.685	407
1.778	6.70	3.37	1.77	210	0.736	290
1.774	9.86	9.10	4.27	747	0.567	1017
1.776	7.79	5.16	2.62	361	0.664	493
1.775	5.93	1.95	0.98	103	0.834	143
1.774	6.76	3.35	1.69	202	0.750	277
1.773	7.39	4.56	2.19	287	0.704	391
1.775	7.92	5.47	2.80	394	0.646	537
1.776	6.33	2.74	1.40	157	0.779	218
1.780	5.98	2.07	1.03	110	0.828	151
1.775	5.42	1.26	0.65	62	0.881	85
1.778	9.49	8.29	3.90	658	0.589	886
1.778	6.02	2.16	1.11	119	0.816	163
1.776	6.91	3.66	1.85	226	0.732	312

US = $4.65 + 1.20 \text{ UP KM/SEC}$
SIGMA US = 0.008 KM/SEC

COMMENTS:

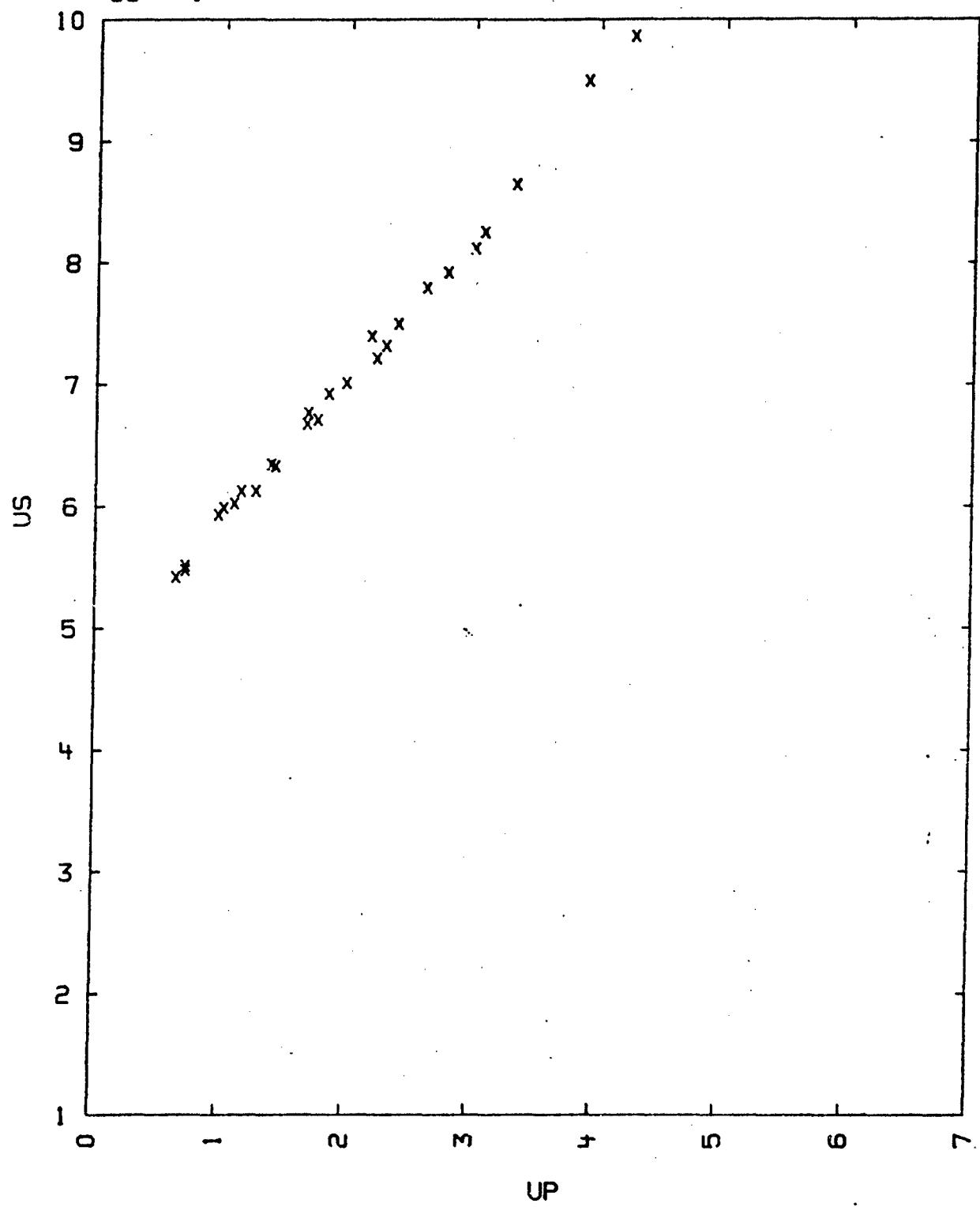
- 1) SOURCE: COMPILER
L.R.L. EQUATION OF STATE FILE
LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B. (ALUMINUM BASE PLATE)
DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF VOI WAS CALCULATED BY ASSUMING ADDITIVITY OF THE VOLUMES OF COMPONENTS. THE VOLUMES OF THE COMPONENTS WERE OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE, CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES (DOVER PUBLICATIONS, INC., NEW YORK, N. Y., 1963).
- 4) SOUND VELOCITIES WERE MEASURED BY H.L. DUNEGAN (SUPPORT ENGINEERING, ULTRASONIC GROUP) LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIF.
- 5) PHYSICAL PROPERTIES:
MELTING POINT 627 DEG C
ELECTRICAL RESISTIVITY 9.2 MICROOHM-CM
FROM DOW METALS PRODUCTS CO. BROCHURE (1962), ON PROPERTIES OF MAGNESIUM MILL PRODUCTS. DOW CHEMICAL COMPANY, MIDLAND, MICH. U.S.A.

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TABLE I

MAGNESIUM

93---1



93---2
MAGNESIUM

MG

$V_0 = 0.578 \text{ CC/G}$
 $V_{01} = 0.575 \text{ CC/G}$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN MM/MICRO-SEC, AND PRESSURE IN KILOBARS.

TABLE

RHO0	US	UP	P	V/V0
1.73	8.34	2.93	420	.638
-	8.61	2.96	440	.657
-	8.42	3.02	440	.651
-	9.22	4.05	645	.562
-	9.18	4.08	648	.555
-	9.64	3.88	650	.600
-	9.54	3.98	660	.586
-	13.43	7.02	1630	.477
-	13.89	7.12	1710	.487
-	13.58	7.32	1720	.461

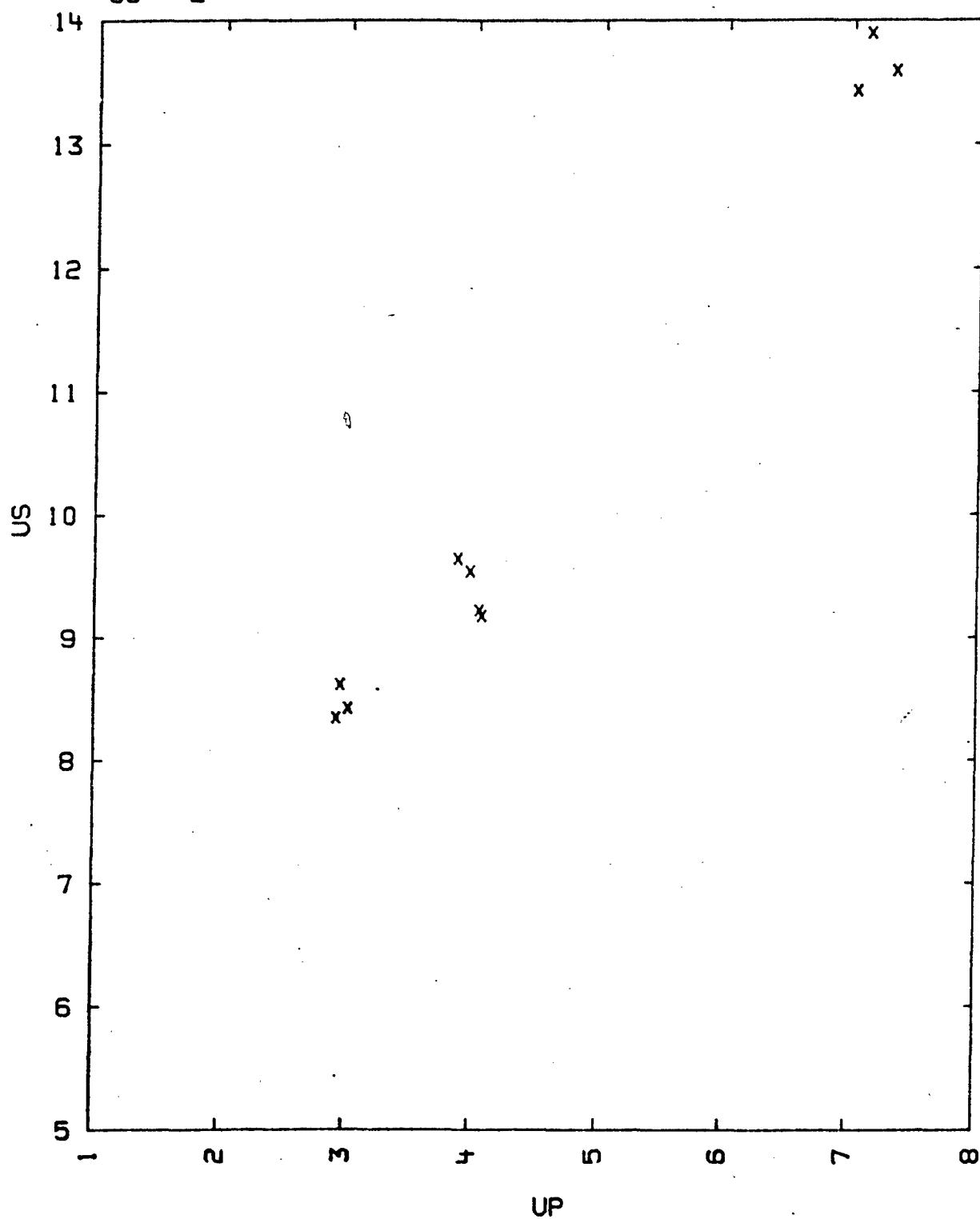
$$US = 4.48 + 1.273 UP \text{ MM/MICROSEC}$$

COMMENTS:

- 1) SOURCE: SKIDMORE, I.C. AND MORRIS, E.
 THERMODYNAMICS OF NUCLEAR MATERIALS, P. 173 FF. (1962)
 INTERN. AT. ENERGY AGENCY, VIENNA
 ATOMIC WEAPONS RESEARCH ESTABLISHMENT, ALDERMASTON, ENGLAND
- 2) EXPERIMENTAL TECHNIQUE A
 DATA REDUCTION TECHNIQUE B
 THE SHOCK WAS PRODUCED BY AN EXPLOSIVELY ACCELERATED EN3 STEEL PLATE.
 THE SHOCK WAS TRANSMITTED THROUGH A STEEL PLATE INTO THE SAMPLE.
- 3) THE VELOCITY OF THE FLYING PLATE AND THE SHOCK AND SURFACE VELOCITY OF THE TARGET PLATE WERE MEASURED AS WELL AS THE SAMPLE SURFACE AND SHOCK VELOCITIES.
- 4) DATA SCATTER WAS ABOUT 0.03 MICROSEC.
- 5) CORRECTIONS WERE MADE FOR FLYING PLATE CURVATURE OF UP TO 1 MICROSEC.
- 6) THE HIGHER PRESSURES WERE OBTAINED BY A SPHERICALLY CONVERGING SYSTEM.
- 7) ALL PELLETS WERE SURROUNDED BY LEAD TO REDUCE LATERAL RAREFACTION.

TABLE I

MAGNESIUM
93---2



93---3
MAGNESIUM POROUS

MG
SPECIFIC SURFACE AREA ABOUT 1150 (SQUARE CM)/G

$V_0 = .77$ AND $.575 \text{ CC/G}$
 $V_{01} = .575 \text{ CC/G}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC. AND PRESSURE IN KILOBARS. $P(ST)$ IS THE PRESSURE IN THE STANDARD.

TABLE

SAMPLE						STANDARD	
RHO0	US	UP	UFS	P	V/V0	P(ST)	
1.3	5.88	2.48	9.2	189.	0.578	189.	
-	5.70	2.28	8.0	169.	0.600	164.	
-	5.10	1.78	7.2	118.	0.651	112.	
-	3.98	1.11	4.8	57.	0.721	56.	
-	2.82	0.69	2.34	25.	0.755	30.	
-	1.58	0.36	0.48	7.	0.771	10.	
1.74	6.6	1.74		200.	0.736		
	-	4.80	0.54		45.	0.887	

US =

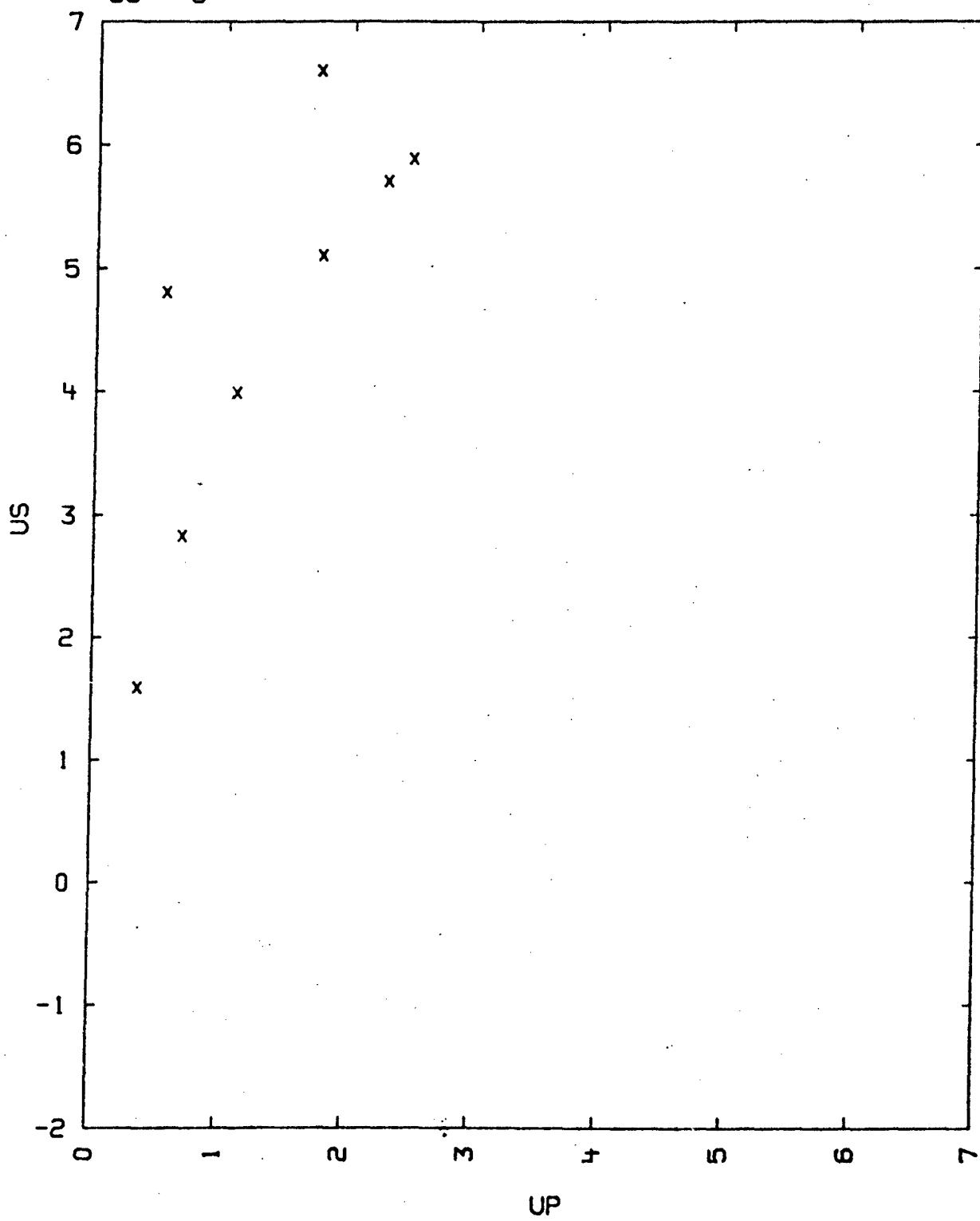
COMMENTS:

- 1) SOURCE: JOHNSON J.O. AND WACKERELE J.
HIGH DYNAMIC PRESSURE SYMPOSIUM, I.U.T.A.M. SEPT. 11-15 1967
PARIS, FRANCE (ALSO PRIVATE COMM.)
- 2) EXPERIMENTAL TECHNIQUE E
DATA REDUCTION TECHNIQUE: B. STANDARD MATERIAL- LUCITE, WITH
 $US = 2.75 + 1.49 \cdot UP$ AND $UP = 0.5 \cdot UFS$.
- 3) ESTIMATED PERCENT UNCERTAINTIES FOR THE POROUS SAMPLES ARE 3.5 IN
 $P(ST)$, 2.6 IN UP, 2.1 IN US, 2.3 IN P AND 3.1 IN V/V0

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TABLE I

MAGNESIUM POROUS
93---3



93---4
MAGNESIUM

MG 99.5 PERCENT OR GREATER

$V_0 = 0.5731 \text{ CC/G.}$ $C_L = 5.74 \text{ KM/SEC.}$ $C_0 = 4.44 \text{ KM/SEC.}$
 $V_{01} = 0.5756 \text{ CC/G.}$ $C_S = 3.15 \text{ KM/SEC.}$ $C_B = 4.45 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC.,
 EDIT OF 24-1---1 SANDSTONE COCONINO
 PRESSURE IN KILOBARS AND DENSITIES IN G/CC.

TABLE

SAMPLE					STANDARD	
RHO0	US	UP	P	V/V0	RHO0	US
1.745	5.747	0.933	93.57	0.8376	2.784	6.353
-	5.733	0.934	93.44	0.8371	2.784	6.353
-	5.960	1.133	117.8	0.8099	2.785	6.571
-	5.949	1.134	117.7	0.8094	2.785	6.571
-	6.409	1.534	171.6	0.7606	2.785	7.001
-	6.361	1.538	170.7	0.7582	2.785	7.001
-	7.080	1.943	240.0	0.7256	2.285	7.485
-	6.987	1.952	238.0	0.7206	2.785	7.485
-	7.279	2.190	278.2	0.6991	2.785	7.754
-	7.231	2.195	277.0	0.6964	2.785	7.754
-	7.379	2.285	293.8	0.6907	2.784	7.857
-	7.374	2.283	293.8	0.6904	2.784	7.857
-	7.559	2.424	319.7	0.6793	2.786	8.018
-	7.635	2.479	330.3	0.6753	2.786	8.082
-	7.556	2.488	328.1	0.6707	2.786	8.082
-	8.364	3.084	450.1	0.6313	2.795	8.770
-	8.342	3.067	449.4	0.6299	2.785	8.770
-	9.414	3.821	627.7	0.5941	2.786	9.635
-	9.281	3.839	621.7	0.5864	2.786	9.635
-	9.991	4.344	757.3	0.5652	2.786	10.23
-	9.883	4.360	751.9	0.5588	2.786	10.23

$$US = 4.507 + 1.255 UP$$

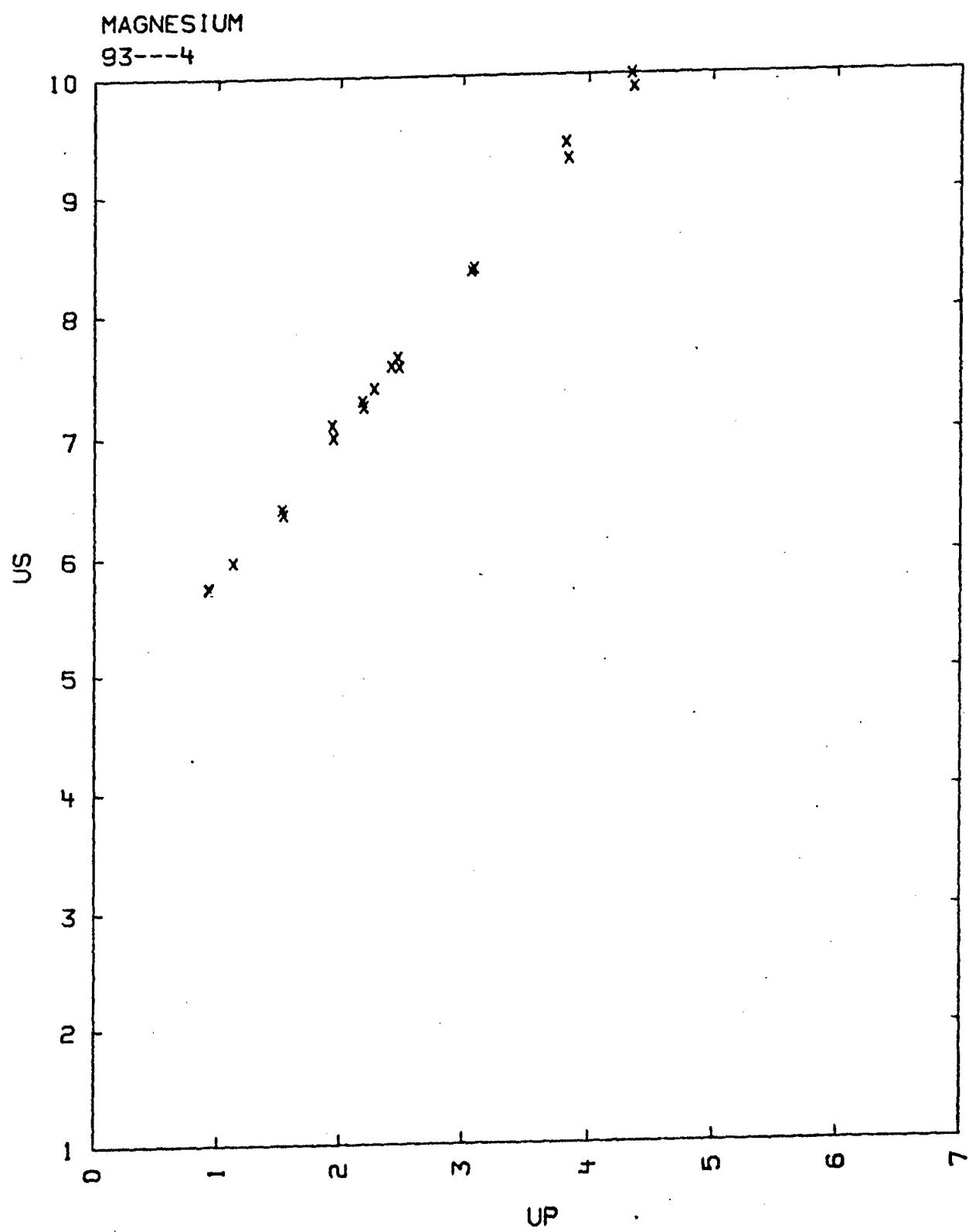
COMMENTS:

- 1) SOURCE: MCQUEEN, R.G. AND MARSH, S.P.
 REPORT NO. GMX-6-566, PP. 51-62 (1964)
 LOS ALAMOS SCIENTIFIC LABORATORY, GMX-6, LOS ALAMOS, N. MEX.
- 2) EXPERIMENTAL TECHNIQUE B
 DATA REDUCTION TECHNIQUE B
 STANDARD MATERIAL IS 2024 ALUMINUM
- 3) THE ALUMINUM HUGONIOT IS REPRESENTED BY
 $US = 5.355 + 1.345 UP \text{ KM/SEC.}$
 THE PARTIAL $(\partial E / \partial P)V = 0.162 \text{ CUBIC CM. PER GRAM.}$
- 4) V_{01} WAS CALCULATED FROM THE LATTICE CONSTANTS

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A = 3.2028 AND C = 5.19983 KX UNITS, OF THE HEXAGONAL UNIT CELL.
AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO., N.Y.,
1963) 2ND ED.

TABLE I



93---5
MAGNESIUM

MG

V0 = 0.571 G/CC.

VOI = 0.5756

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----						-STANDARD-
RHO0	US	UFS	UP	P	V/V0	UFS
1.75	7.042	3.974	1.970	243.	0.720	
-	7.153	3.937	1.920	240	0.732	
-	6.944	3.655	1.825	222	0.737	
-	6.802	3.844	1.910	227	0.719	
-	6.830	3.526	1.760	210	0.742	
-	6.973	3.686	1.835	224	0.732	
-	6.711		1.891	222	0.718	3.080
-	6.811		1.879	224	0.724	3.060
-	6.702		1.880	220.5	0.719	3.060
-	6.803		1.865	222	0.726	3.050
-	6.501		1.772	202	0.727	2.870
-	6.640		1.732	201	0.739	2.830
-	6.527		1.681	192	0.742	2.710
-	6.493		1.716	195	0.736	2.780
-	6.765		1.580	187	0.767	2.600
-	6.631		1.607	186.5	0.756	2.630
-	6.435		1.522	172	0.763	2.480
-	6.501		1.623	185.5	0.750	2.640
-	6.443		1.575	178.5	0.756	2.565
-	6.289		1.645	181.5	0.738	2.650
-	6.369		1.513	168	0.762	2.445
-	6.402		1.508	169	0.764	2.445
-	6.172		1.564	159	0.747	2.495
-	6.195		1.555	168	0.749	2.480
-	6.188		1.509	163	0.756	2.415
-	6.273		1.545	170	0.754	2.495
-	6.443		1.460	165	0.773	2.390
-	6.257		1.514	165	0.758	2.430
-	6.060		1.367	145	0.774	2.200
-	6.150		1.384	149	0.775	2.210
-	6.195		1.328	144	0.786	2.150
-	6.195		1.365	148	0.780	2.200
-	6.134		1.146	123	0.813	1.870
-	5.938		1.169	121.5	0.803	1.880
-	5.720		1.213	121.5	0.788	1.950
-	6.038		1.202	127	0.801	1.930
-	5.307		0.888	82.5	0.833	1.390
-	5.530		0.884	85.5	0.840	1.405
-	5.446		0.840	80	0.846	1.315

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MAGNESIUM

RHO0	US	UFS	UP	P	V/V0	UFS
-	5.399		0.845	79	0.843	1.325
-	5.509		0.793	76.5	0.856	1.260
-	5.509		0.792	76	0.856	1.255
-	5.561		0.773	75	0.861	1.240
-	5.537		0.763	74	0.862	1.210
-	5.561		0.760	74	0.863	1.210
-	5.488		0.765	73.5	0.861	1.220
-	5.324		0.774	72	0.855	1.215
-	5.524		0.752	73	0.864	1.205
-	5.636		0.707	70	0.875	1.135
-	5.364		0.713	67	0.867	1.125
-	5.393		0.660	62	0.878	1.040
-	5.458		0.656	63	0.880	1.060

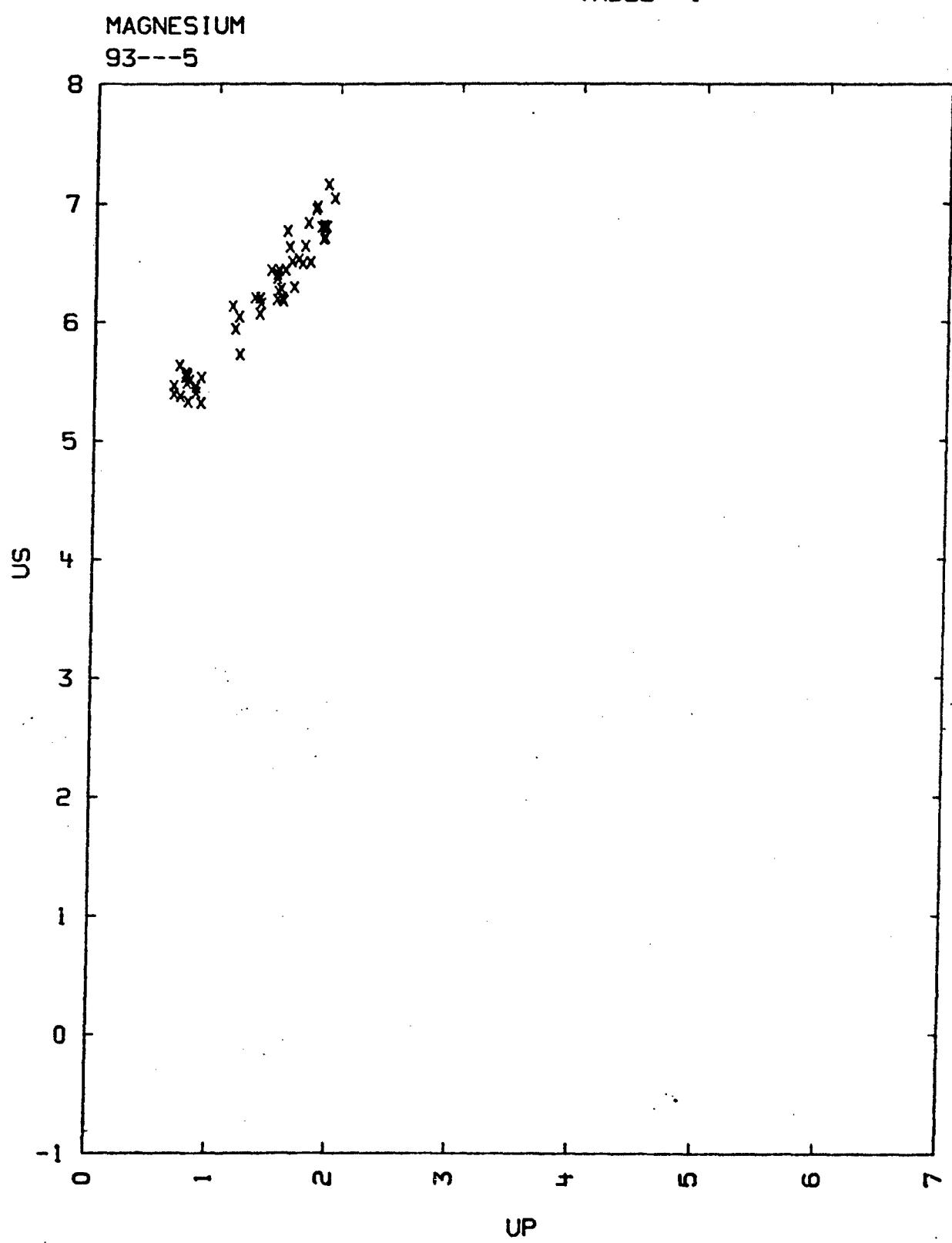
$$US = 4.529 + 1.212 UP \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: BERGER J. AND FAUQUIGNON C.
PRIVATE COMMUNICATION (1964), B.P. NO. 7, SEVRAN, FRANCE
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL ALUMINUM AU4G
- 3) SAMPLE DIMENSIONS FOR THE US MEASUREMENTS: 2.0 CM DIAMETER
0.5 CM THICKNESS
SAMPLE DIMENSIONS FOR THE UFS MEASUREMENTS: 2.0 CM DIAMETER
0.25 CM THICKNESS
- 4) V01 WAS CALCULATED FROM THE CONSTANTS, A = 3.2028 AND C = 5.19983
UNITS, OF THE HEXAGONAL UNIT CELL.
AMERICAN INSTITUTE OF PHYS. HANDBOOK (MCGRAW-HILL BOOK CO., N.Y.
1963) 2ND ED.

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TABLE I



93---6
MAGNESIUM

MG 96 PERCENT BY WEIGHT
ZN 3 PERCENT - -
ZR 1 PERCENT - -

$V_0 = 0.556 \text{ CC/G.}$
 $V_{01} = 0.5756 \text{ CC/G.}$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC., VELOCITIES IN KM/SEC., AND PRESSURE IN KILOBARS.

TABLE

RHO0	US	UP	P	V/V0
1.80	5.24	0.86	81	0.836
-	5.29	0.87	83	0.836
-	5.42	0.86	84	0.841
-	6.55	1.63	192	0.751
-	6.60	1.69	201	0.744
-	6.83	1.67	205	0.756
-	6.55	1.84	217	0.719
-	6.79	2.11	258	0.689
-	6.86	2.11	260	0.692
-	7.10	2.10	268	0.704
-	8.16	2.96	435	0.637
-	8.29	2.93	437	0.647
-	8.96	3.51	566	0.608
-	9.22	3.52	284	0.618
-	9.21	3.82	633	0.585

$US = 4.40 + 1.274 UP \text{ KM/SEC.}$

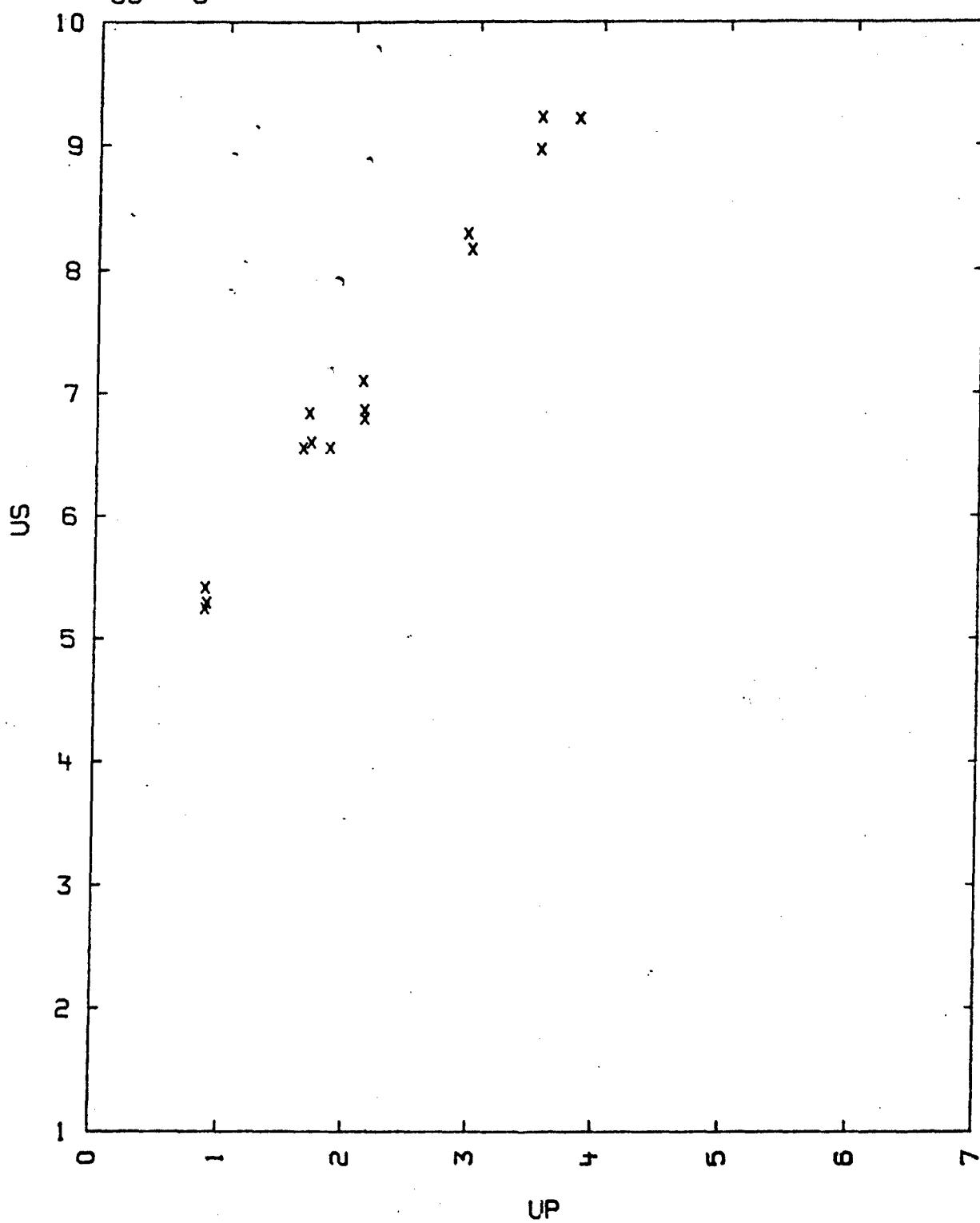
COMMENTS:

- 1) SOURCE: HART AND SKIDMORE
PRIVATE COMMUNICATION (1965)
ATOMIC WEAPONS RESEARCH ESTABLISHMENT, ALDERMASTON,
BERKSHIRE, ENGLAND.
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
STANDARD MATERIAL FE AND BRASS.
- 3) V_{01} WAS CALCULATED FOR PURE MAGNESIUM USING THE LATTICE CONSTANTS
 $A = 3.20280$ AND $C = 5.19983 \text{ KX UNITS}$, OF THE HEXAGONAL UNIT CELL
A.C.A. MONOGRAPH NUMBER 5(AMERICAN CRYSTALLOGRAPHIC ASSOCIATION,
POLYCRYSTAL BOOK SERVICE 1963) 2ND. ED.
- 4) THE ABOVE US-UP FIT WHICH WAS OBTAINED BY USING IDEAL MIXTURE
THEORY ON THE HUGONIOTS OF THE COMPONENTS IN THE P-V PLANE, IS IN
AGREEMENT WITH THE TABULATED DATA.

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TABLE I

MAGNESIUM
93---6



93----7
MAGNESIUM

MG

VO = 0.575 CC/G
VOI = 0.57562 CC/G

CD = 4.39 KM/SEC

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

----- SAMPLE -----					----- STANDARD -----		
RHO0	US	UP	P	V/VO	MATERIAL	RHO0(ST)	US(ST)
1.74	12.24	6.25	1330.	0.489	ALUMINUM	2.71	9.17
-	8.63	3.39	509.	0.607			
-	8.49	3.30	487.	0.611			
-	8.02	2.67	372.	0.667			
-	7.08	2.08	256.	0.706			
-	6.66	1.85	214.4	0.722			
-	5.99	1.12	116.7	0.813			

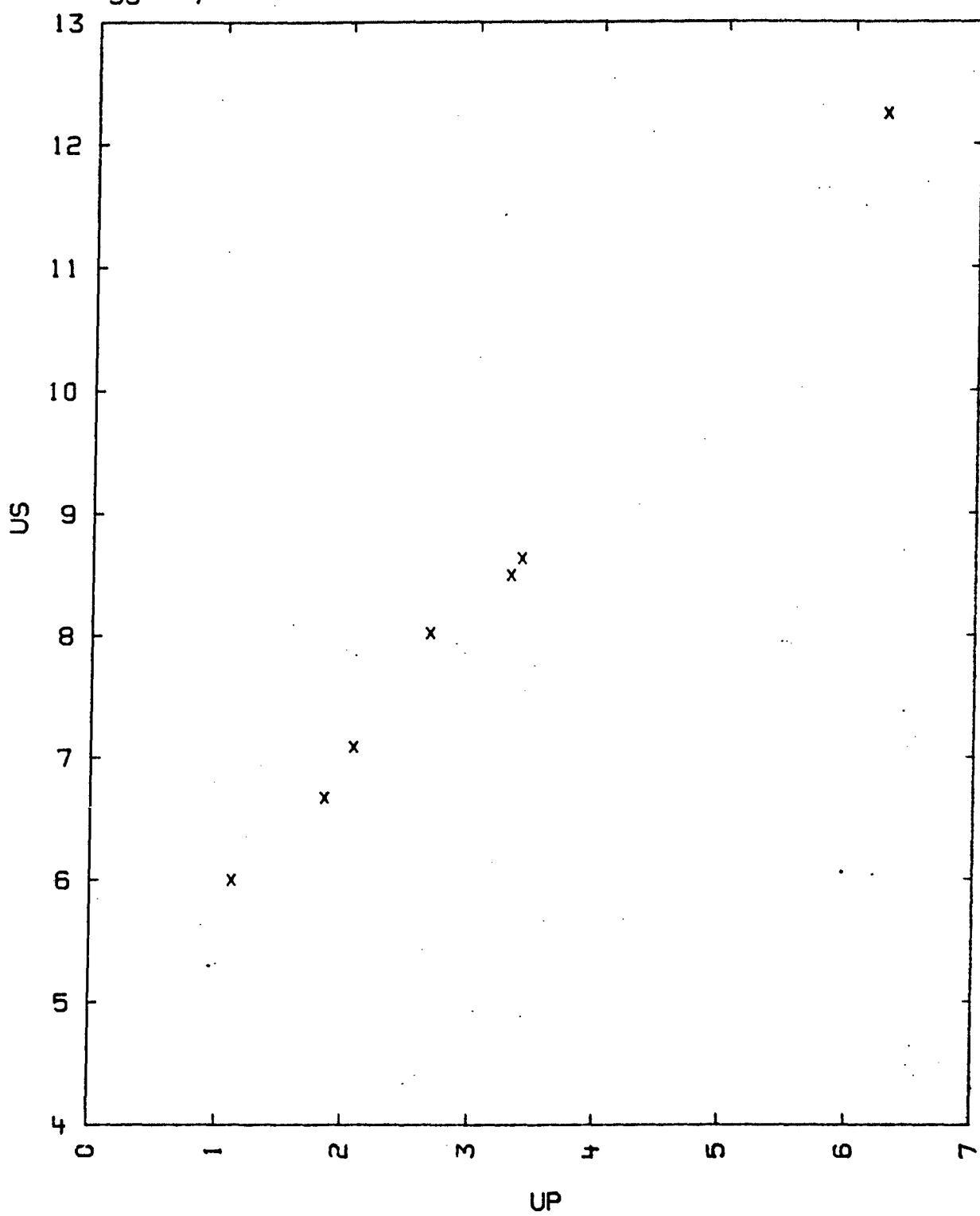
US = 4.41 + 1.27*UP KM/SEC FOR UP BELOW 6.25 KM/SEC

COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P.
ZHURNAL EKSP. TEOREDT FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES.
THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE
1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = 5.25 + 1.39*UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
- 5) THE VALUE OF VOI WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANTS
OF THE HEXAGONAL CLOSE-PACKED STRUCTURE,
A = 3.20927 ANGSTROMS, C = 5.21033 ANGSTROMS (25 DEG. C)
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 11.
- 6) THE AUTHORS OBTAINED THE VALUE OF CD FROM HANDBOOK OF RARE METALS,
(MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO
SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).

TABLE I

MAGNESIUM
93---7



93---B
MAGNESIUMMG
GRAIN SIZE 5. MM MAX

$$V_0 = 0.5731 - 0.5764 \text{ CC/G} \quad C_C = 5.74 \text{ KM/SEC} \quad C_0 = 4.44 \text{ KM/SEC}$$

$$V_{01} = 0.5756 \quad C_C/G \quad C_S = 3.15 \text{ KM/SEC} \quad C_B = 4.45 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC., PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
1.745	5.66	0.88	87.	0.8445	2024 AL	6.26
1.745	5.65	0.88	87.	0.8442	2024 AL	6.26
1.745	5.87	1.08	111.	0.8160	2024 AL	6.48
1.735	5.91	1.08	111.	0.8173	2024 AL	6.48
1.745	5.86	1.08	110.	0.8157	2024 AL	6.48
1.734	6.12	1.20	127.	0.8039	2024 AL	6.61
1.734	6.28	1.34	146.	0.7866	2024 AL	6.77
1.734	6.24	1.37	148.	0.7804	2024 AL	6.80
1.745	6.32	1.47	162.	0.7674	2024 AL	6.91
1.770	6.15	1.48	161.	0.7593	2024 AL	6.91
1.745	6.27	1.48	162.	0.7640	2024 AL	6.91
1.741	6.38	1.58	175.	0.7524	921-T AL	6.86
1.745	6.98	1.88	229.	0.7307	2024 AL	7.38
1.745	6.89	1.89	227.	0.7257	2024 AL	7.38
1.734	7.01	1.93	235.	0.7247	2024 AL	7.43
1.734	6.76	1.93	226.	0.7145	2024 AL	7.41
1.734	6.87	1.95	232.	0.7162	2024 AL	7.43
1.735	7.00	2.03	247.	0.7100	2024 AL	7.53
1.745	7.17	2.12	265.	0.7043	2024 AL	7.64
1.745	7.13	2.13	265.	0.7013	2024 AL	7.64
1.740	7.11	2.14	265.	0.6990	921-T AL	7.52
1.734	7.25	2.15	270.	0.7034	2024 AL	7.67
1.745	7.27	2.21	280.	0.6960	2024 AL	7.74
1.745	7.27	2.21	280.	0.6960	2024 AL	7.74
1.745	7.45	2.35	306.	0.6846	2024 AL	7.90
1.745	7.52	2.41	316.	0.6795	2024 AL	7.97
1.745	7.45	2.42	315.	0.6752	2024 AL	7.97
1.745	8.24	3.01	433.	0.6347	2024 AL	8.64
1.745	8.22	3.01	432.	0.6338	2024 AL	8.64
1.734	8.83	3.36	514.	0.6195	2024 AL	9.06
1.734	8.85	3.36	516.	0.6203	2024 AL	9.06
1.734	8.79	3.37	514.	0.6166	2024 AL	9.06
1.734	9.03	3.50	548.	0.6124	2024 AL	9.23
1.734	8.93	3.51	544.	0.6069	2024 AL	9.22
1.735	9.04	3.72	583.	0.5885	2024 AL	9.45
1.745	9.28	3.73	604.	0.5981	2024 AL	9.50
1.745	9.15	3.75	599.	0.5902	2024 AL	9.50
1.734	9.75	4.02	680.	0.5877	2024 AL	9.84

U06/14/77

MAGNESIUM

RHO0	US	UP	P	V/V0	MATERIAL US(ST)
------	----	----	---	------	-----------------

1.734	9.65	4.03	674.	0.5824	2024 AL 9.83
1.734	9.54	4.05	670.	0.5755	2024 AL 9.83
1.745	9.85	4.25	731.	0.5685	2024 AL 10.08
1.745	9.74	4.27	726.	0.5616	2024 AL 10.08

$$US = 4.492 + 1.263 \cdot UP \text{ KM/SEC}$$

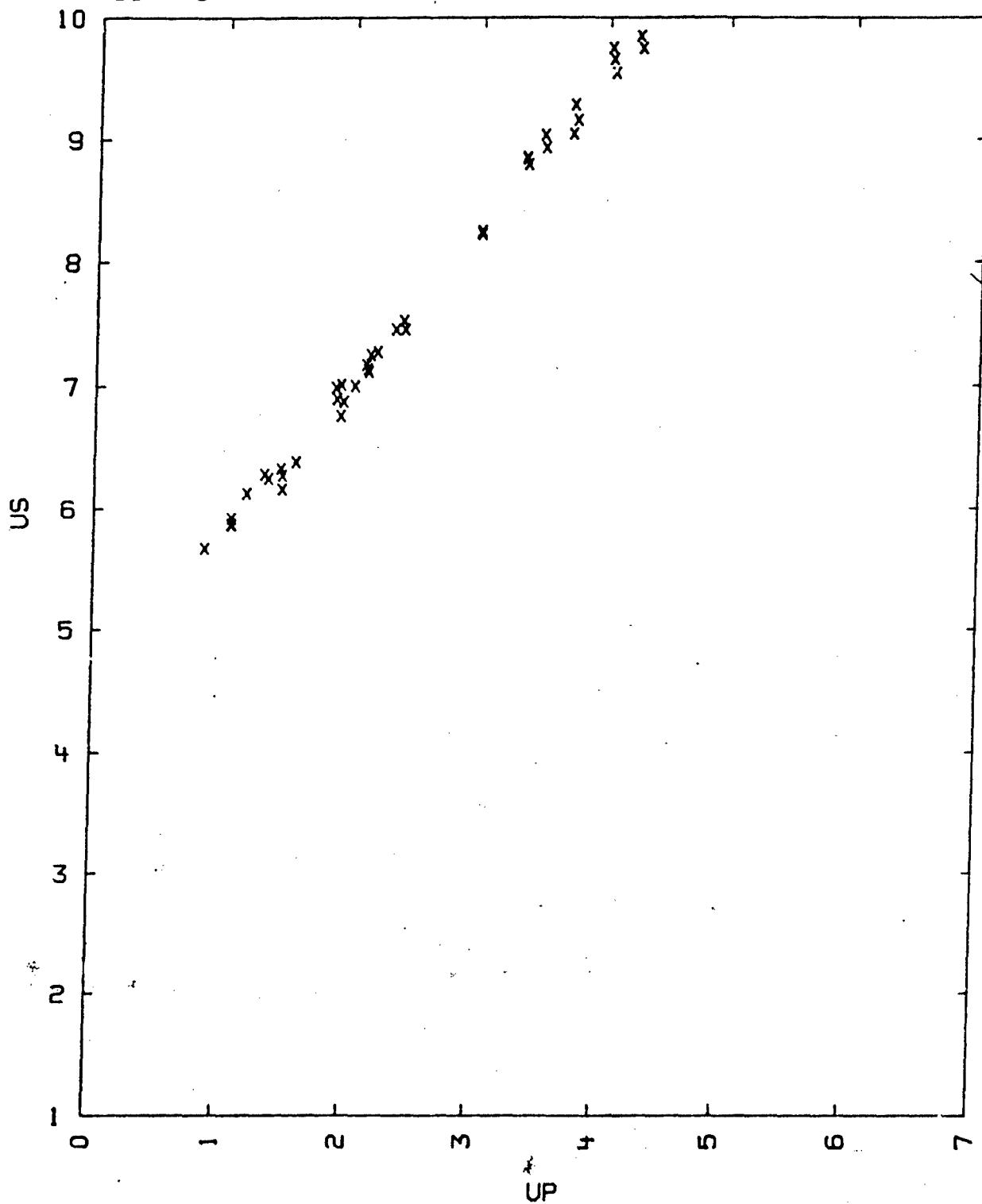
SIGMA US = 0.089 KM/SEC

COMMENTS:

- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
PRESS, NEW YORK, 1970) CHAPTER VII
- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B
- 3) VOI FROM WYCKOFF, CRYSTAL STURCTURES (INTERSCIENCE PUBLISHERS, N.Y.,
1963) V. I
- 4) V(OP/DE) = 1.425

TABLE I

MAGNESIUM
93---8



93---9

MAGNESIUM AZ-31B

MG	96.0 WT PERCENT
AL	3.0 WT PERCENT
ZN	1.0 WT PERCENT

$$V_0 = 0.562 - 0.565 \text{ KM/SEC} \quad C_L = 5.73 \text{ KM/SEC} \quad C_0 = 4.52 \text{ KM/SEC}$$

$$V_{OI} = 0.565 \text{ KM/SEC} \quad C_S = 3.05 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

-----SAMPLE-----					-----STANDARD-----	
RHO0	US	UP	P	V/V0	MATERIAL	US(ST)
1.770	5.53	0.39	38.	0.9295	CU	4.31
1.780	5.30	0.60	57.	0.8868	2024 AL	5.96
1.770	5.35	0.60	57.	0.8879	2024 AL	5.96
1.770	5.33	0.61	58.	0.8856	2024 AL	5.98
1.770	5.36	0.64	61.	0.8806	921-T AL	5.76
1.780	5.52	0.82	81.	0.8514	2024 AL	6.20
1.770	5.60	0.83	82.	0.8518	2024 AL	6.22
1.770	5.95	1.12	118.	0.8118	2024 AL	6.53
1.780	5.91	1.12	118.	0.8105	2024 AL	6.53
1.770	5.93	1.13	119.	0.8094	2024 AL	6.54
1.770	5.96	1.22	129.	0.7953	921-T AL	6.43
1.780	6.28	1.42	159.	0.7739	2024 AL	6.87
1.770	6.39	1.47	166.	0.7700	2024 AL	6.92
1.770	7.00	2.03	252.	0.7100	2024 AL	7.55
1.780	7.04	2.04	256.	0.7102	2024 AL	7.56
1.770	7.14	2.04	258.	0.7143	921-T AL	7.43
1.770	7.09	2.07	260.	0.7080	2024 AL	7.60
1.780	7.12	2.12	269.	0.7022	2024 AL	7.65
1.770	7.66	2.53	343.	0.6697	2024 AL	8.12
1.780	7.70	2.60	356.	0.6623	2024 AL	8.19
1.780	8.23	3.02	442.	0.6330	2024 AL	8.67
1.780	8.54	3.29	500.	0.6148	2024 AL	8.98
1.780	8.72	3.36	522.	0.6147	2024 AL	9.08
1.780	9.13	3.71	603.	0.5936	2024 AL	9.48
1.780	9.63	4.05	694.	0.5794	2024 AL	9.88
1.780	10.00	4.14	737.	0.5860	2024 AL	10.03

$$US = 4.522 + 1.242 \cdot UP \text{ KM/SEC}$$

$$\text{SIGMA US} = 0.128 \text{ KM/SEC}$$

COMMENTS:

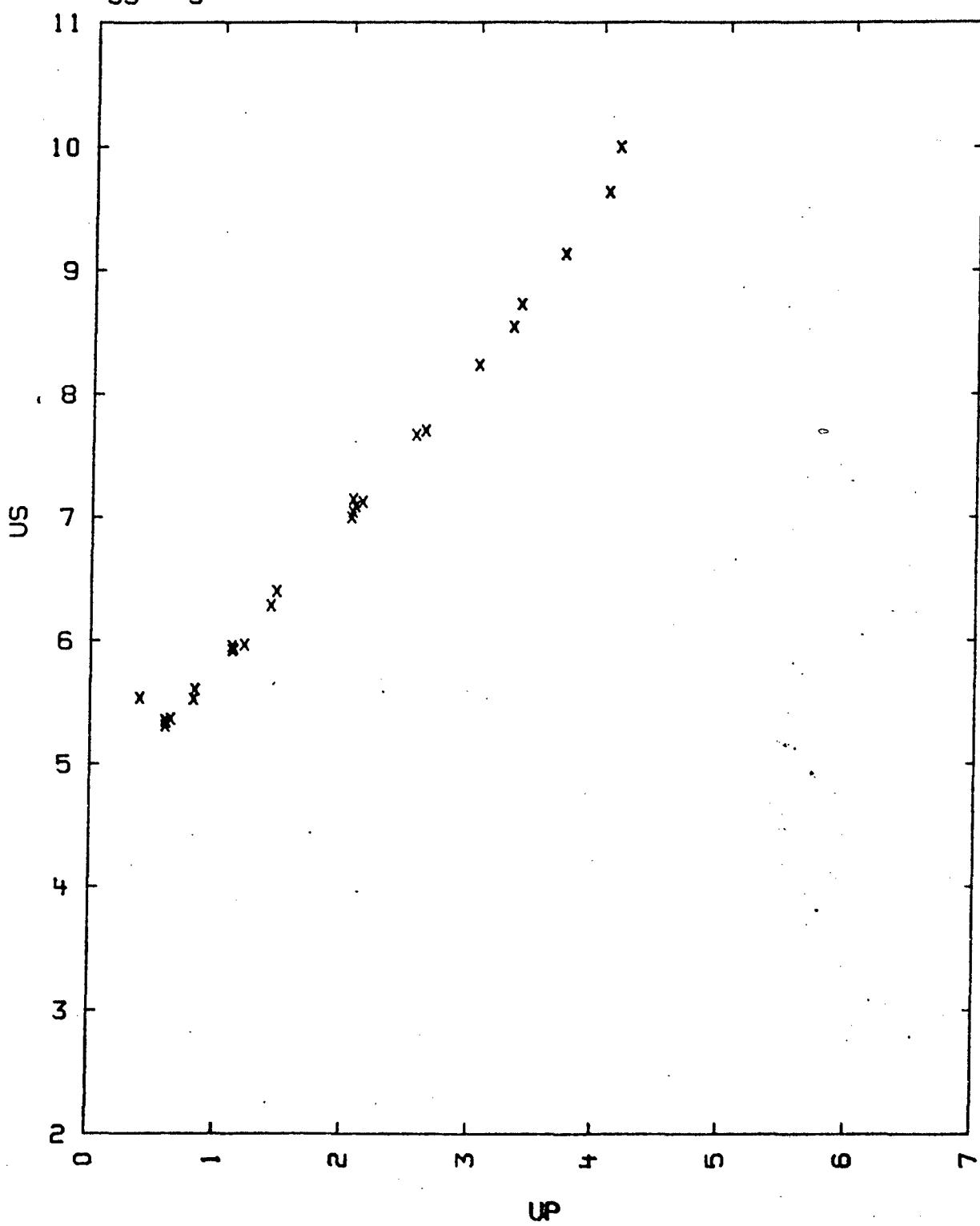
- 1) SOURCE: MCQUEEN, R.G., MARSH, S.P., TAYLOR, J.W., FRITZ, J.M., AND CARTER, W.J.
 THE EQUATION OF STATE OF SOLIDS FROM SHOCK WAVE STUDIES,
 HIGH VELOCITY IMPACT PHENOMENA, KINSLOW (ED.) (ACADEMIC
 PRESS, NEW YORK, 1970) CHAPTER VII

- 2) EXPERIMENTAL TECHNIQUE: B
DATA REDUCTION TECHNIQUE: B
- 3) VOI AS FOR 93---1
- 4) V(DP/DE) = 1.434
- 5) HUGONIOT ELASTIC LIMIT 1.1 KBAR

TABLE I

MAGNESIUM AZ-31B

93---9



93---10
MAGNESIUM, AZ31B ALLOY

MG	96.	WT. PERCENT NOMINAL
AL	3.	- - -
ZN	1.	- - -

$$V_0 = 0.564 \text{ CC/G}$$

$$V_{01} = 0.565 \text{ - }$$

THE TABLE LISTS RHO0 IN G/CC, VELOCITIES IN KM/SEC AND P IN KBARS.
MG= MAGNESIUM, CU= COPPER, FS= FANSTEEL AND WF= WEIGHTING FACTOR.

TABLE

SAMPLE					IMPACTOR			
RHO0	US	UP	P	V/V0	MAT	U	WF	
1.773	6.731	1.769	211.	.7372	MG	3.539	3	
1.773	8.020	2.956	420.	.6314	MG	5.911	3	
1.773	10.884	4.959	957.	.5444	CU	6.631	1	
1.773	12.221	6.388	1384.	.4773	FS	7.795	3	

$$US = 4.551 + 1.209 \cdot UP \text{ KM/SEC}$$

$$\text{SIG.US} = 0.083 \text{ KM/SEC}$$

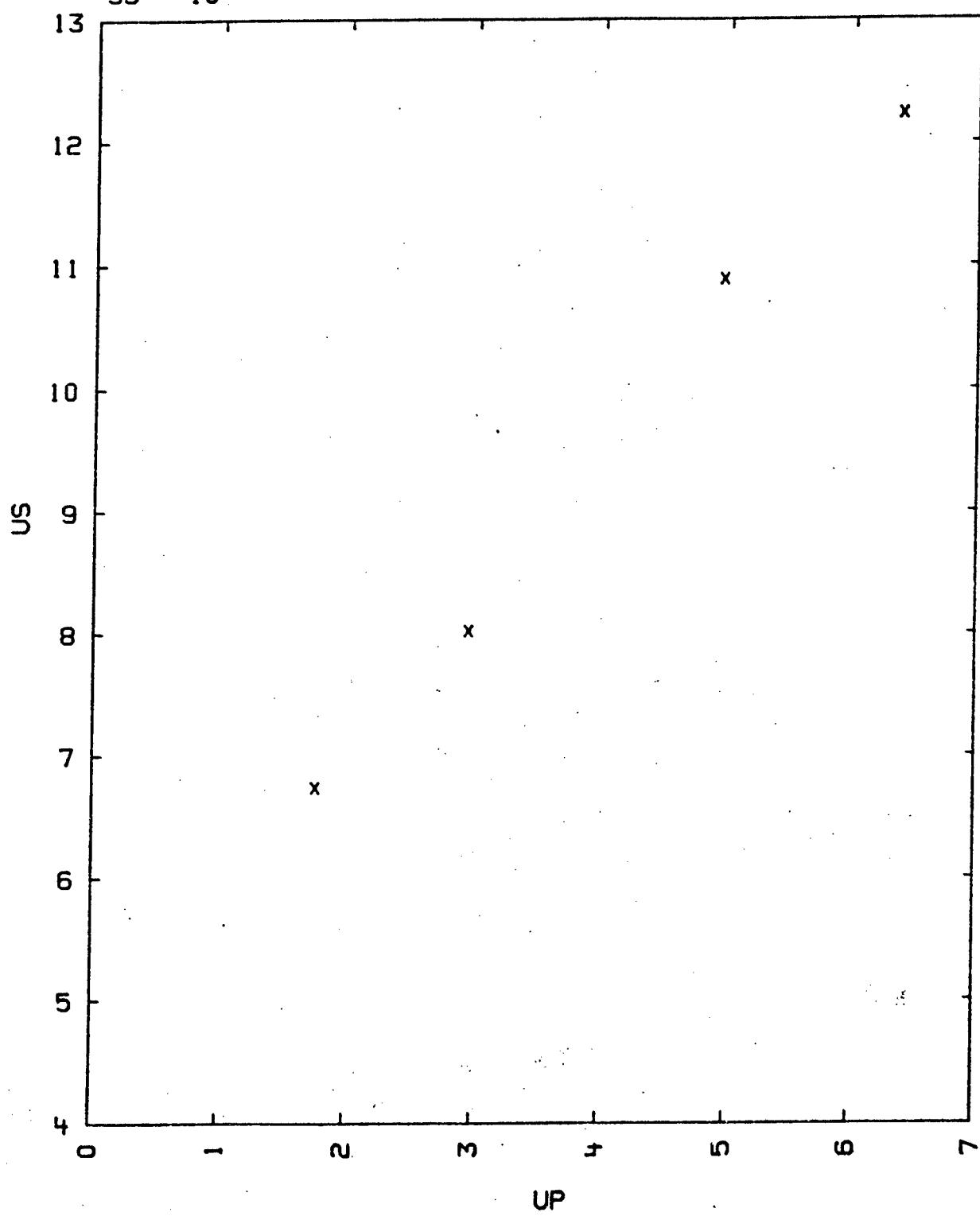
COMMENTS:

- 1) SOURCE: ISBELL W.M., SHIPMAN F.H. AND JONES A.H.
HUGONIOT EQUATION OF STATE OF ELEVEN MATERIALS TO FIVE MBARS.
MATERIALS SCIENCE LABORATORY REPORT: MSL-68-13
- 2) EXPERIMENTAL TECHNIQUE: A
DATA REDUCTION METHOD : A
- 3) NOMINAL UNCERTAINTIES ARE (SIG.US)/US = .005 AND (SIG.U)/U = .005
- 4) V01 WAS TAKEN FROM ENTRY 93---

TABLE I

MAGNESIUM, AZ31B ALLOY

93---10



94---1
CALCIUM

CA

$$V_0 = 0.656 \text{ CC/G}$$

$$V_{01} = 0.6536 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC AND PRESSURE IN KILOBARS.

TABLE

RHO0	SAMPLE			AL BASE PLATE		
	US	UFS	UP	P	V/V0	PRESSURE
1.523	5.61	4.29	2.15	183	0.618	324
1.523	5.47	4.16	2.16	178	0.605	324
1.527	4.64	2.22	1.15	81	0.753	146
1.526	6.73	7.79	3.44	354	0.489	618
1.524	7.09	6.87	3.45	353	0.488	618

$$US = 3.58 + 0.91 UP \text{ KM/SEC}$$

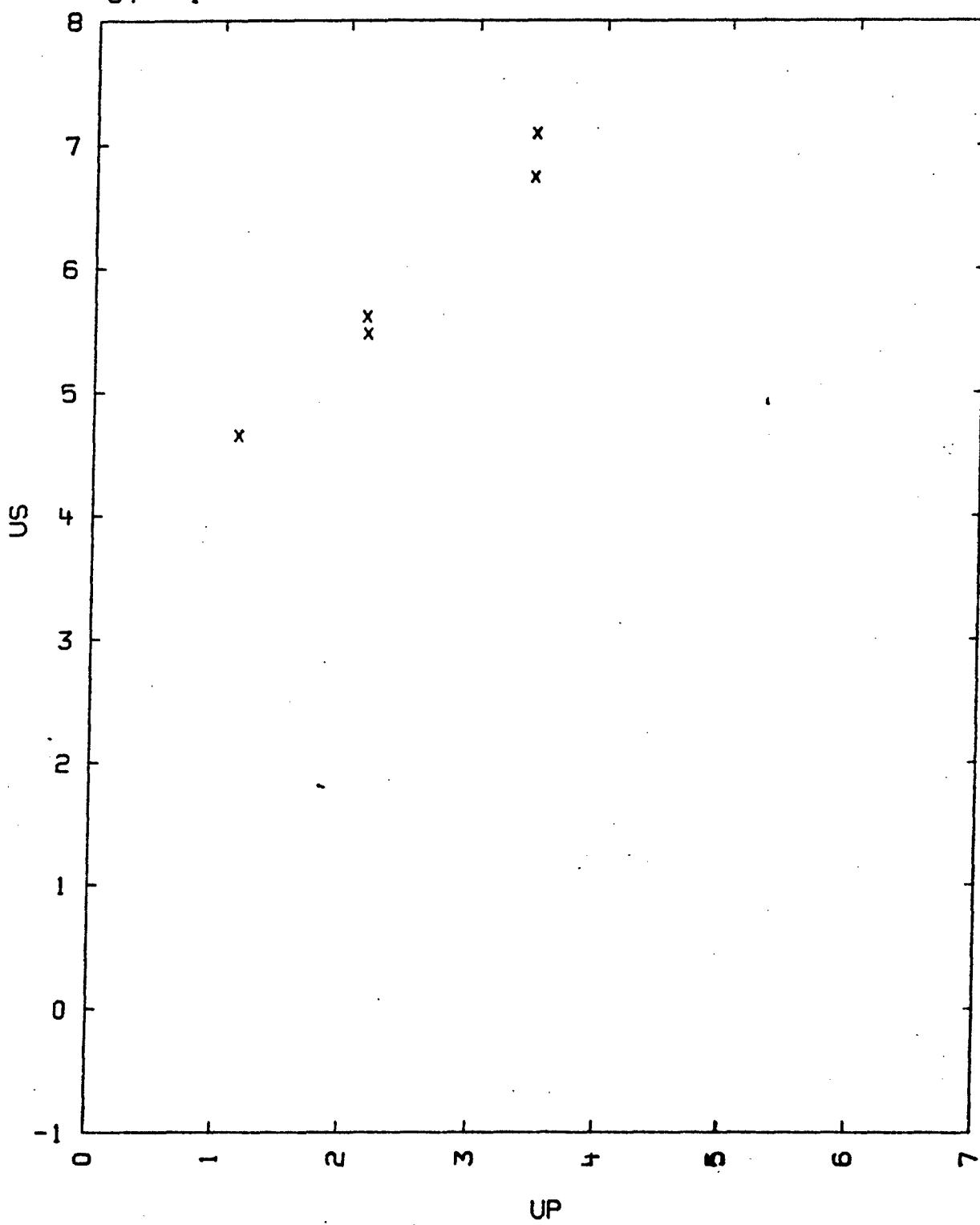
$$\text{SIGMA US} = 0.062 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: COMPILER
L.R.L. EQUATION OF STATE FILE
LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B. STANDARD MATERIAL 2024 AL ALLOY.
DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF VOI WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES
(DOVER PUBLICATIONS, INC., NEW YORK, N.Y., 1963).

TABLE I

CALCIUM
94---1



94---2
CALCIUM

CA

V₀ = 0.658 CC/G
V₀₁ = 0.6513 CC/GC₀ = 3.36 KM/SECIN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO ₀	US	UP	P	V/V ₀	MATERIAL	RHO _{0(ST)}	US _(ST)
1.52	11.13	6.96	1178.	0.375	IRON	7.85	11.21
-	8.42	4.84	619.	0.425	IRON	7.85	8.83
-	7.02	3.77	402.	0.463	ALUMINUM	2.71	9.17
-	6.18	2.89	271.	0.532			
-	5.78	2.46	216.	0.574			
-	5.48	2.04	170.	0.628	ALUMINUM	2.71	7.34
-	4.96	1.59	120.	0.679			
-	4.43	0.96	64.6	0.783	ALUMINUM	2.71	6.21

US = 3.55 + 0.912*UP KM/SEC FOR UP BELOW 3.71 KM/SEC

US = 2.08 + 1.308*UP KM/SEC FOR UP BETWEEN 3.71 AND 6.96 KM/SEC

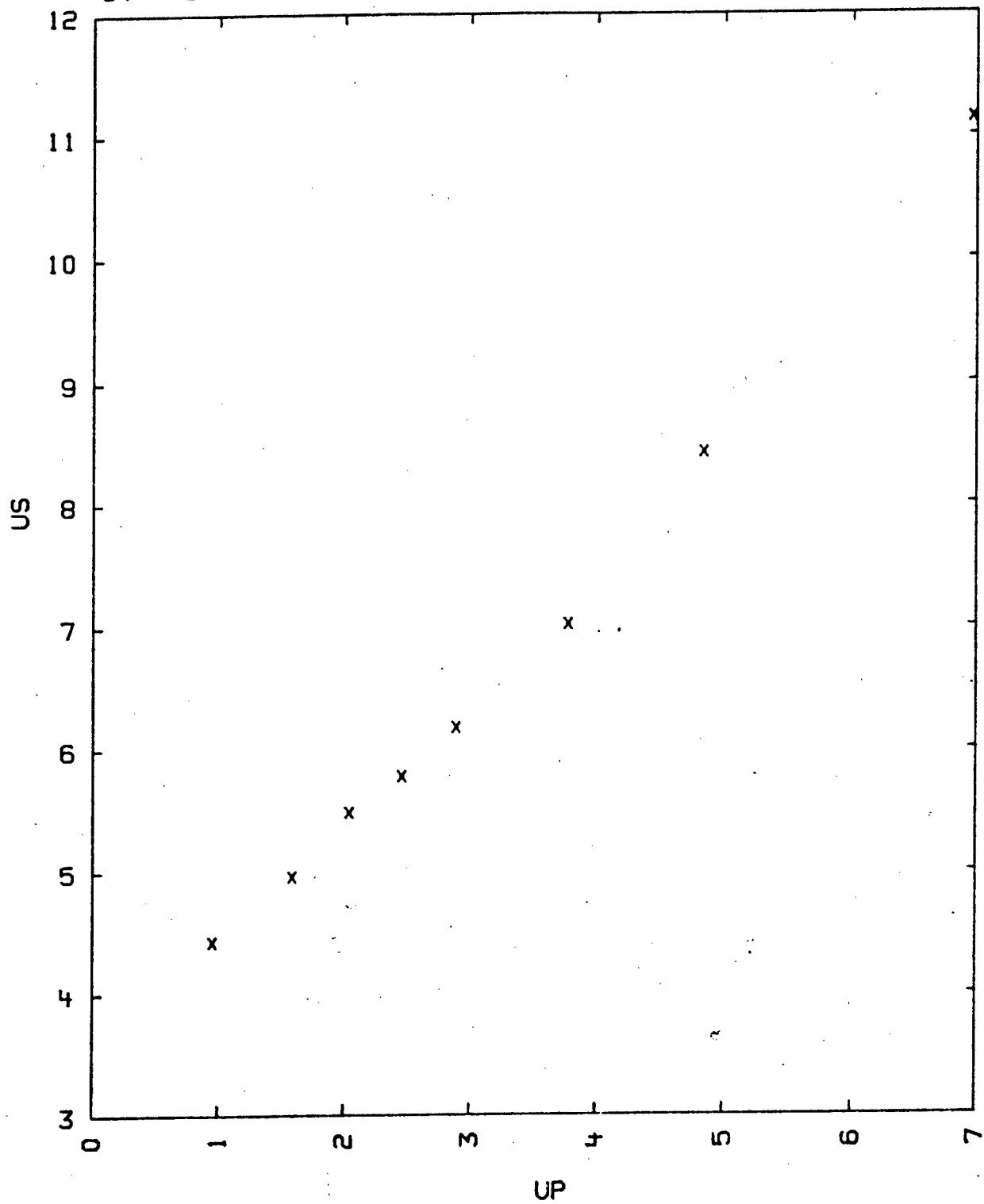
COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P.
ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES.
THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE
1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = 5.25 + 1.39*UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
IRON US = 3.85 + 1.615*UP FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF V₀₁ WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANT
OF THE CUBIC CLOSE-PACKED STRUCTURE.
A = 5.576 ANGSTROMS
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 10.
- 6) THE AUTHORS OBTAINED THE VALUE OF C₀ FROM HANDBOOK OF RARE METALS,
(MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO
SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT UP = 3.71 KM/SEC IS
ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

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TABLE I

CALCIUM
94---2



95---I
STRONTIUM

SR

$$V_0 = 0.385 \text{ CC/G}$$

$$V_{01} = 0.38711 \text{ CC/G}$$

$$C_0 = 2.11 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

SAMPLE					STANDARD		
RHO0	US	UP	P	V/V0	MATERIAL	RHO0(ST)	US(ST)
2.60	10.08	6.40	1677.	0.365	IRON	7.85	11.21
-	7.37	4.49	860.	0.391	IRON	7.85	8.83
-	5.55	3.35	483.	0.396	ALUMINUM	2.71	9.17
-	4.64	2.63	317.	0.433	COPPER	8.93	6.45
-	3.94	1.92	196.7	0.513	ALUMINUM	2.71	7.34
-	2.93	0.93	70.8	0.683	ALUMINUM	2.71	6.21

$$US = 2.10 + 0.94 \cdot UP \text{ KM/SEC}$$

$$US = 0.58 + 1.50 \cdot UP \text{ KM/SEC}$$

FOR UP BELOW 2.71 KM/SEC

FOR UP BETWEEN 2.71 AND 6.40 KM/SEC

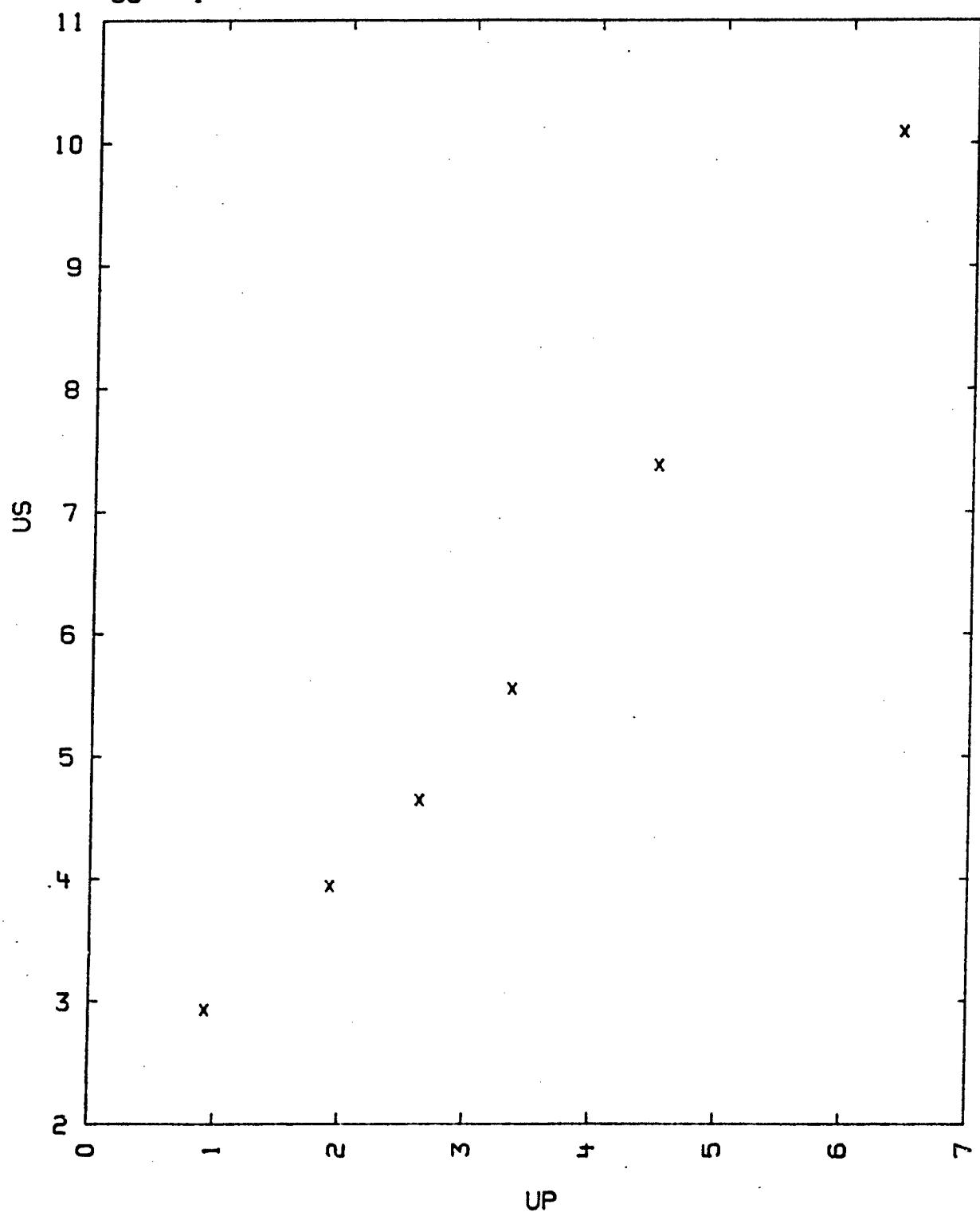
COMMENTS:

- 1) SOURCE: ALTSHULER, L. V., BAKANOVA, A. A., DUDOLADOV, I. P. ZHURNAL EKSP- TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES. THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE 1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = 5.25 + 1.39 · UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
COPPER US = 3.95 + 1.50 · UP FOR UP BETWEEN 0.0 AND 4.0 KM/SEC
IRON US = 3.85 + 1.615 · UP FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF VOI WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANT OF THE CUBIC CLOSE-PACKED STRUCTURE,
A = 6.0847 ANGSTOMS (25 DEG. C)
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS, NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. I, P. 10.
- 6) THE AUTHORS OBTAINED THE VALUE OF CO FROM HANDBOOK OF RARE METALS, (MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).
- 7) THE DISCONTINUITY OF THE US VS. UP PLOT AT UP = 2.71 KM/SEC IS ATTRIBUTED TO AN ELECTRONIC PHASE CHANGE.

TABLE I

STRONTIUM

95---1



96---1
BARIUM

BA

V₀ = 0.275 CC/G
V₀₁ = 0.2782 CC/GC₀ = 1.66 KM/SECIN THE TABLE BELOW, VELOCITIES ARE GIVEN IN KM/SEC, PRESSURE IN KILOBARS
AND DENSITY IN G/CC.

TABLE

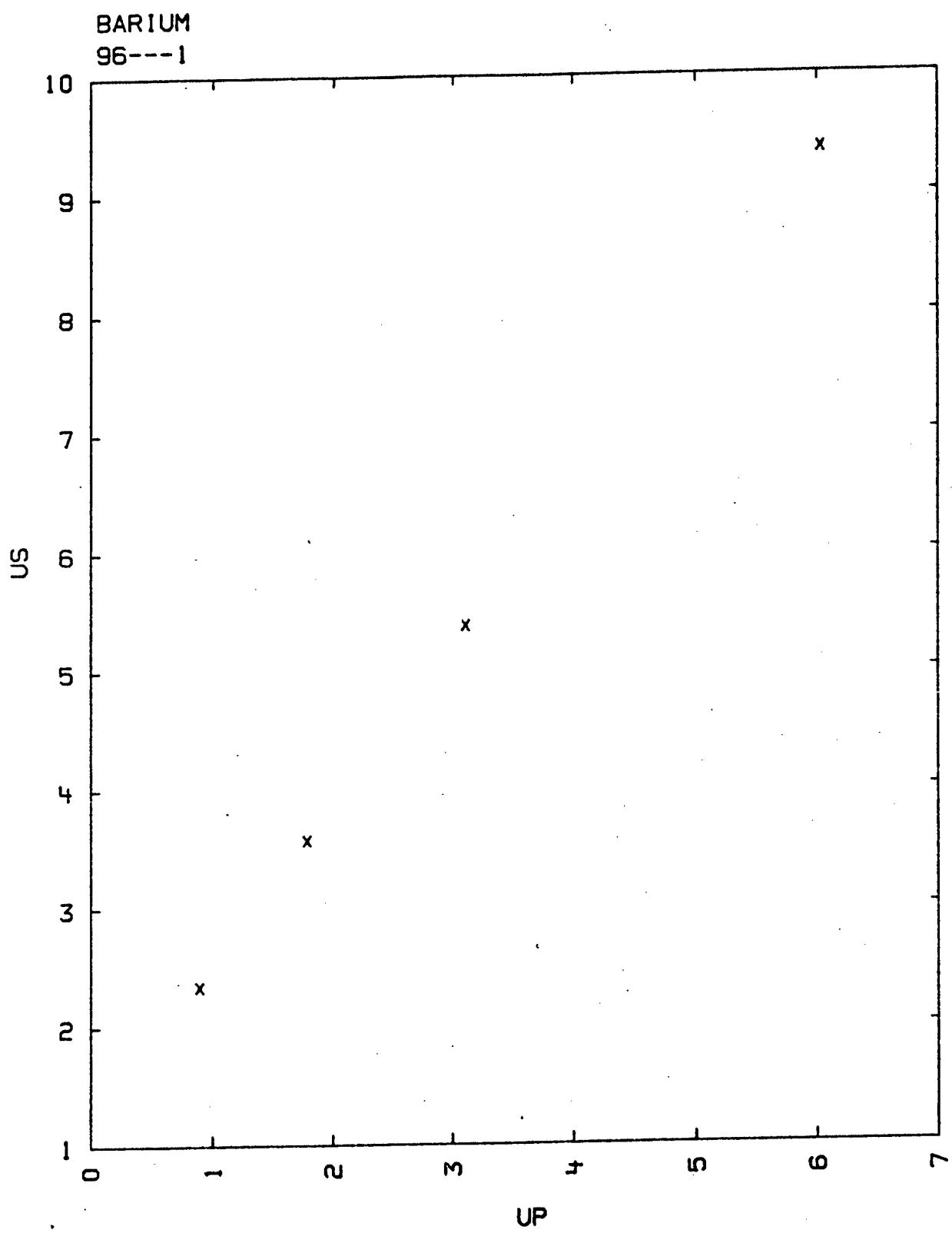
----- SAMPLE -----					----- STANDARD -----		
RHO ₀	US	UP	P	V/V ₀	MATERIAL	RHO _{0(ST)}	US(ST)
3.63	9.36	6.03	2049.	0.356	IRON	7.85	11.21
-	5.37	3.11	606.	0.421	ALUMINUM	2.71	9.17
-	3.56	1.79	231.	0.497	ALUMINUM	2.71	7.34
-	2.34	0.90	76.4	0.615	ALUMINUM	2.71	6.21

US = 1.11 + 1.366*UP KM/SEC FOR UP BETWEEN 0.9 AND 6.03 KM/SEC

COMMENTS:

- 1) SOURCE: ALTSHULER, L.V., BAKANOVA, A. A., DUDOLADOV, I. P., ZHURNAL EKSP. TEORET. FIZIKI V. 53, P. 1967 (1967).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) ONLY SHOCK VELOCITIES WERE MEASURED TO DETERMINE THE STRESS STATES.
THE UNCERTAINTY IN THE SHOCK VELOCITIES WAS ESTIMATED TO BE
1.0 PERCENT.
- 4) HUGONIOTS OF STANDARD MATERIALS USED WERE AS FOLLOWS
ALUMINUM US = 5.25 + 1.39*UP FOR UP BETWEEN 0.0 AND 6.0 KM/SEC
IRON US = 3.85 + 1.615*UP FOR UP BETWEEN 1.0 AND 4.6 KM/SEC
- 5) THE VALUE OF V₀₁ WAS OBTAINED FROM THE FOLLOWING LATTICE CONSTANT
OF THE CUBIC BODY-CENTERED STRUCTURE,
A = 5.025 ANGSTROMS (26 DEG. C)
WYCKOFF, R. W. G., CRYSTAL STRUCTURES (INTERSCIENCE PUBLISHERS,
NEW YORK, LONDON, SIDNEY, 1963) 2ND EDITION, VOL. 1, P. 16.
- 6) THE AUTHORS OBTAINED THE VALUE OF C₀ FROM HANDBOOK OF RARE METALS.
(MIR PUBLISHERS, USSR, 1963) OR FROM C. KITTEL, INTRODUCTION TO
SOLID STATE PHYSICS (FIZMATGIZ, USSR, 1963).

TABLE I



98---1
LITHIUM

LI

V0 = 1.887 CC/G
V0I = 1.873 CC/G

CO = 4.69 KM/SEC.

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0
0.530	6.329	1.425	47.8	0.7748
-	6.382	1.457	49.3	0.7717
-	6.734	1.941	69.3	0.7118
-	7.433	2.554	100.6	0.6564
-	7.449	2.553	100.8	0.6573
-	8.251	3.231	141.3	0.6084
-	8.893	3.679	173.4	0.5863
-	8.929	3.727	176.4	0.5826
-	10.192	4.908	265.1	0.5184
-	10.335	4.887	267.7	0.5271

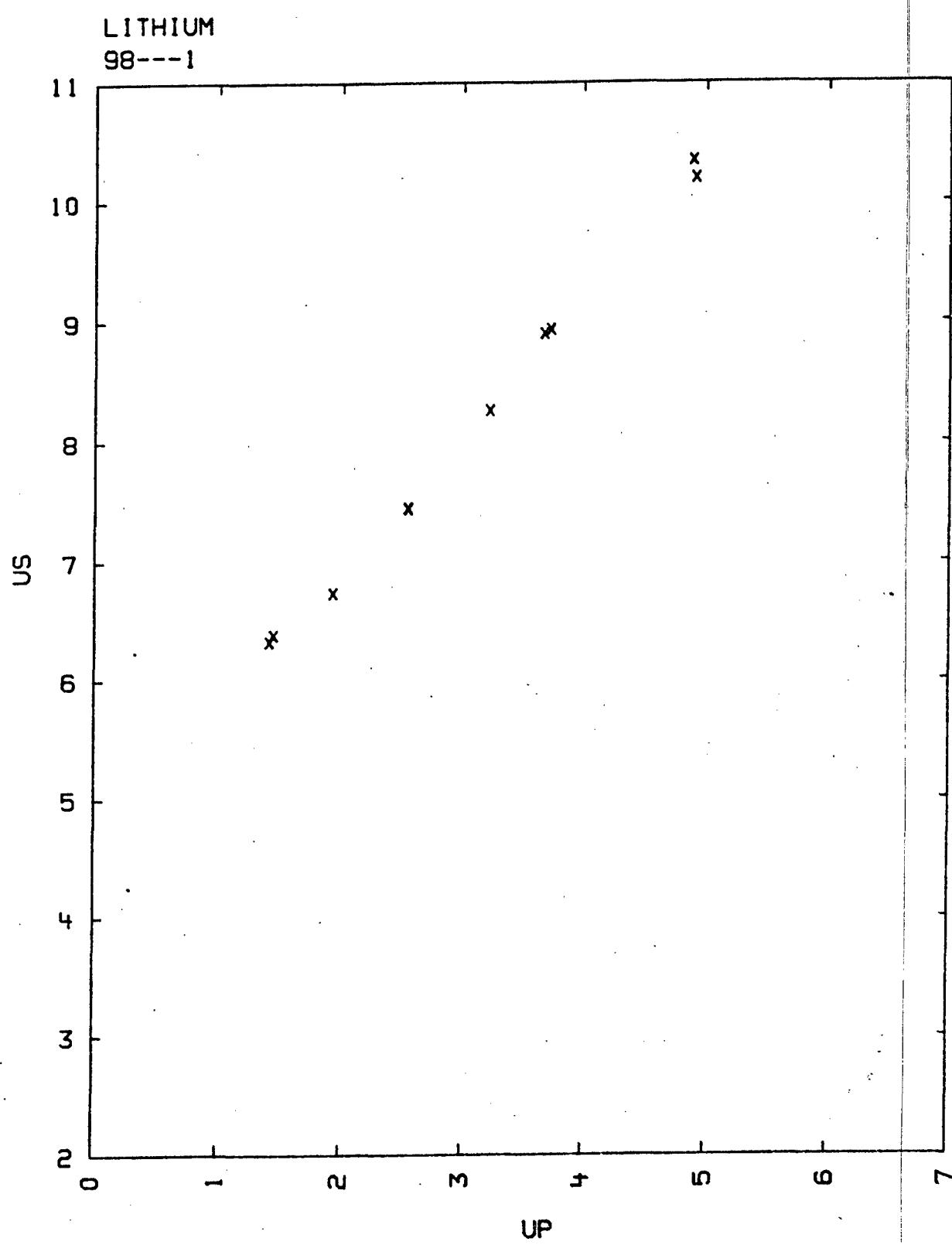
US = 4.589 + 1.154 UP KM/SEC.

COMMENTS:

- 1) SOURCE: RICE, M. H.
J. PHYS. CHEM. SOLIDS. VOL. 26, P. 483 (1965)
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL IS 2024 ALUMINUM
- 3) CO IS CALCULATED BY USING A BULK MODULUS OF 11.49 E 10 DYNE/CM².
C. KITTEL, INTRODUCTION TO SOLID STATE PHYSICS, 10 ED.
JOHN WILEY AND SONS, INC., NEW YORK, 1956.
- 4) V0I WAS TAKEN FROM THE A. I. P. HANDBOOK (MCGRAW-HILL BOOK CO.)
2ND ED.

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TABLE I



98---2
LITHIUM

LI

$$V_0 = 1.69 \text{ CC/G}$$

$$V_{01} = 1.876 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC,
AND PRESSURE IN KILOBARS.

TABLE

SAMPLE				BASE PLATE		
RHO0	US	UP	P	V/V0	MATERIAL	UP
0.53	5.86	1.14	35	0.806	AL	0.69
-	6.43	1.87	64	0.709	AL	1.14
-	7.25	2.59	100	0.641	AL	1.60
-	9.32	4.45	220	0.521	AL	2.82
-	10.75	5.54	316	0.485	PLEXIGLAS	4.51
-	11.64	6.42	396	0.448	PLEXIGLAS	5.21
-	12.10	6.81	437	0.437	PLEXIGLAS	5.53
-	12.43	7.15	471	0.426	PLEXIGLAS	5.80
-	12.40	7.23	475	0.417	PLEXIGLAS	5.85
-	13.14	7.97	555	0.394	PLEXIGLAS	6.44
-	14.39	9.15	698	0.364	PLEXIGLAS	7.38

$$US = 4.51 + 1.09 UP \text{ KM/SEC}$$

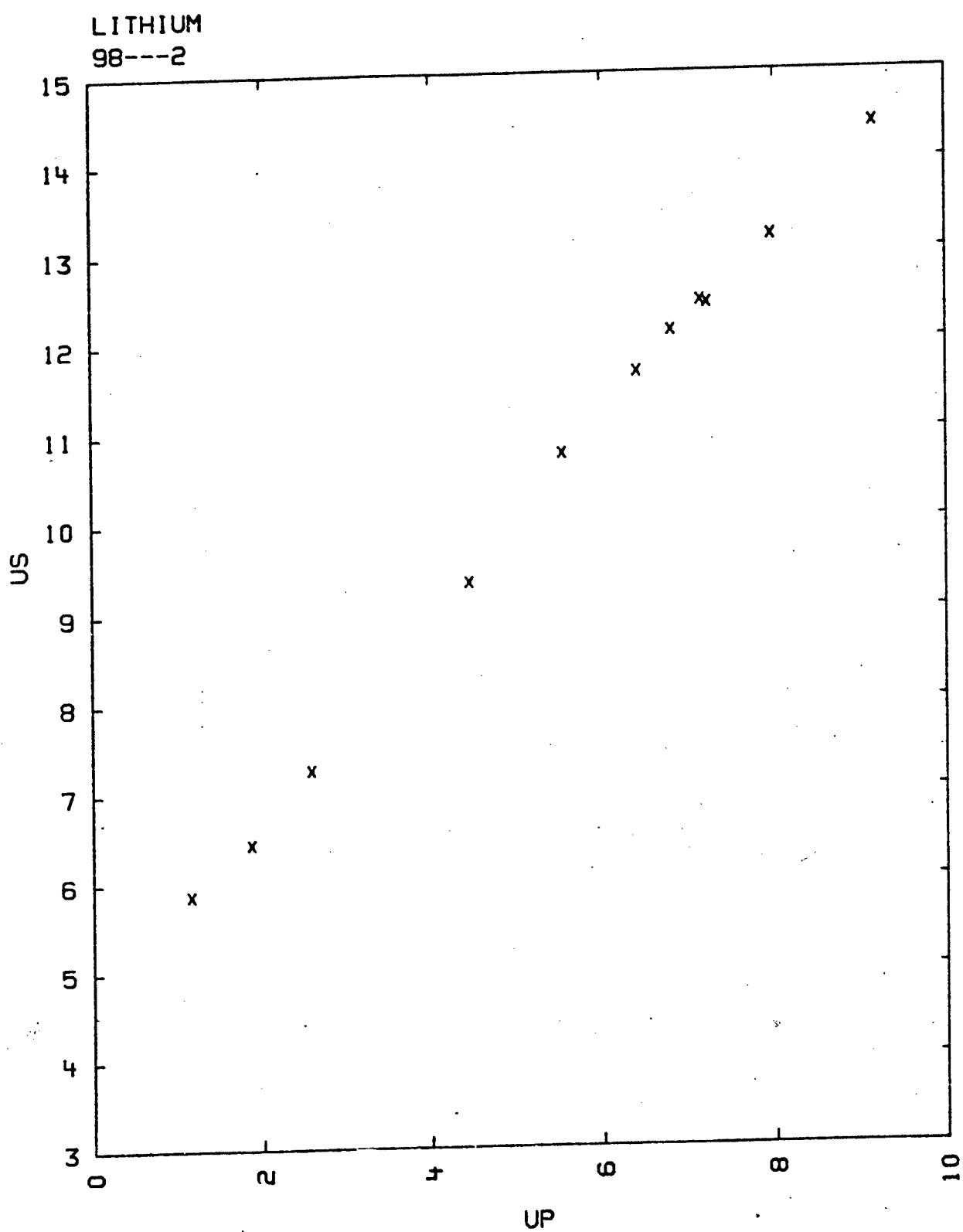
$$\Sigma US = 0.123 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: BAKANOVA, A.A., DUDOLADOV, I.P., TRUNIN, R. F.
FIZIKA TVERDOGO TELA, VOL. 7, NO. 6, PP. 1616 - 1622 (1965).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) THE FOLLOWING HUGONIOT RELATIONSHIPS FOR THE BASE PLATE WERE USED:
AL STANDARD $US = 5.25 + 1.39 UP \text{ KM/SEC}$ $RHO0 = 2.71 \text{ G/CC}$
PLEXIGLAS $US = 3.10 + 1.32 UP \text{ KM/SEC}$ $RHO0 = 1.18 \text{ G/CC}$ (FOR US EQUAL OR GREATER THAN 6.0 KM/SEC).
- 4) OTHER LISTED LI PARAMETERS:
AT 300 DEG. K
RHO0 0.53 G/CC
COMPRESSIBILITY 9.12 PER MEGABAR
GRUNEISEN COEF 0.90
AT 0.0 DEG. K
RHO0 0.544 G/CC
COMPRESSIBILITY 7.9 PER MEGABAR
ELECTRONIC HEAT CAPACITY 2520 ERG/G DEG. SQUARE
- 5) THE VALUE OF V01 WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES
(DOVER PUBLICATIONS, INC., NEW YORK, N.Y., 1963).

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TABLE I



98--3
LITHIUM

Li

$$V_0 = 1.87 \text{ CC/G}$$

$$V_{01} = 1.876 \text{ CC/G}$$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC,
AND PRESSURE IN KILOBARS.

TABLE

RHO0	SAMPLE					V/V0	BASE PLATE
	US	UFS	UP	P	PRESSURE		
0.537	9.71		4.77	250	0.509		773
0.533	7.04	4.28	2.14	80	0.696		257
0.531	6.25	2.59	1.29	43	0.794		140
0.534	11.26	10.53	5.68	342	0.496		1017
0.533	8.45		3.48	157	0.588		493
0.533	8.66	6.69	3.72	172	0.570		537
0.533	9.97	8.65	4.87	259	0.510		796
0.533	9.31	8.10	4.33	215	0.535		670

$$US = 4.66 + 1.11 UP \text{ KM/SEC}$$

$$\text{SIGMA US} = 0.232 \text{ KM/SEC}$$

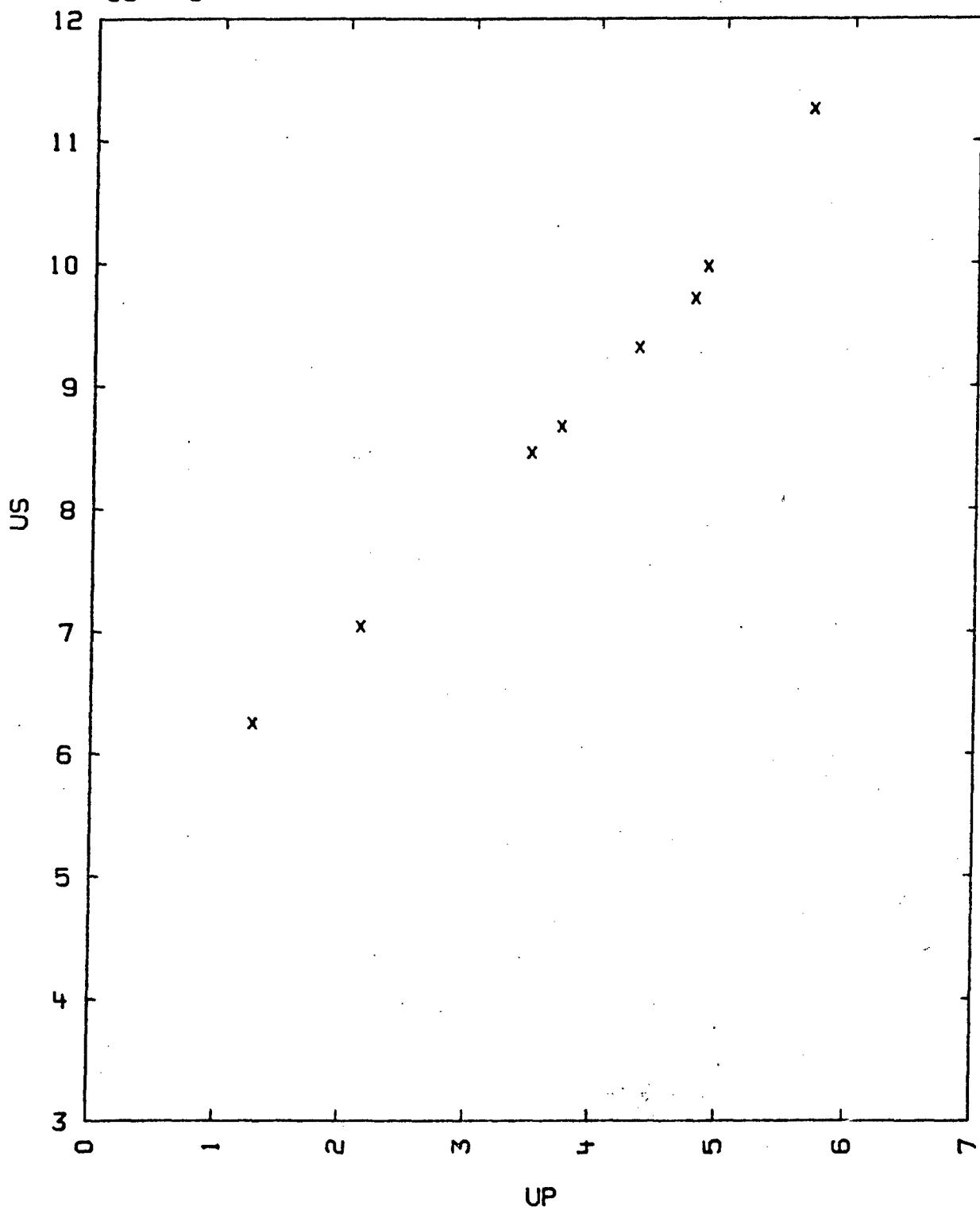
COMMENTS:

- 1) SOURCE: COMPILER
L.R.L. EQUATION OF STATE FILE
LAWRENCE RADIATION LABORATORY, LIVERMORE, CALIFORNIA.
- 2) EXPERIMENTAL TECHNIQUE B (ALUMINUM STANDARD BASE PLATE).
DATA REDUCTION TECHNIQUE B.
- 3) THE VALUE OF V01 WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES
(DOVER PUBLICATIONS, INC., NEW YORK, N.Y., 1963).

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TABLE I

LITHIUM
98---3



99---0
SODIUM SUMMARY

NA

$$V_0 = 0.1033 \text{ CC/G}$$

$$V_{01} = 0.1034 \text{ CC/G}$$

$$C_0 = 2.33 \text{ KM/SEC}$$

THE TABLE LISTS HUGONIOT POINTS CALCULATED FROM THE FIT GIVEN BELOW.
UNITS ARE: G/CC, KM/SEC, KBAR AND KBAR.CC/G FOR THE ENERGY DIFFERENCE.

TABLE

RHO0	US	UP	P	V/V0	E-E0
0.967	3.777	1.0	36.5	0.735	5.00
-	5.037	2.0	97.4	0.603	20.0
-	6.297	3.0	183.	0.523	45.0
-	7.557	4.0	292.	0.471	80.0
-	10.077	6.0	585.	0.404	180.
-	12.597	8.0	974.	0.365	320.

$$US = 2.517 + 1.260 \cdot UP, \text{ SIG. } US = 0.06 \text{ KM/SEC}$$

FOR UP BETWEEN 1 AND 8 KM/SEC.

COMMENTS:

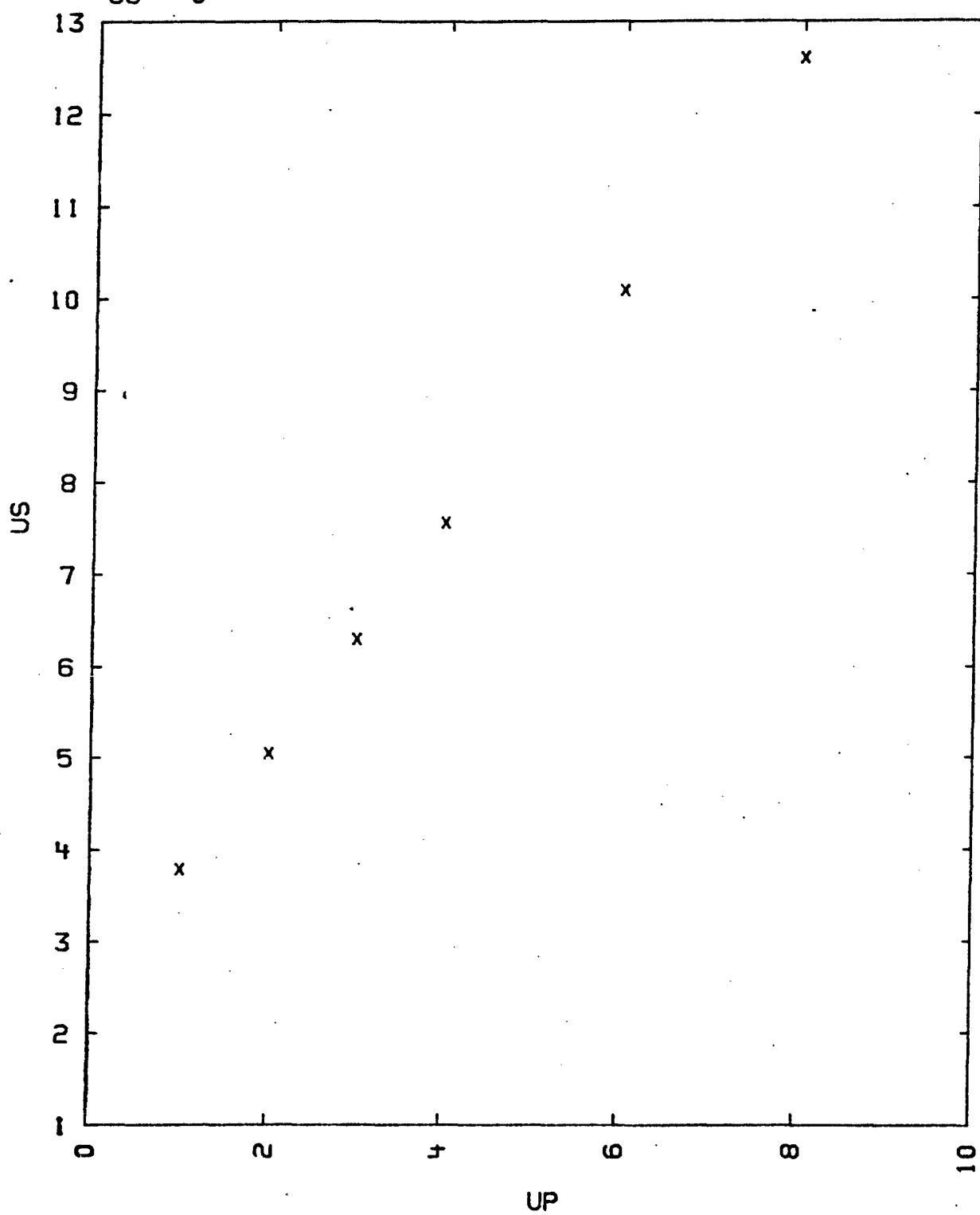
- 1) SOURCE: COMPILER
- 1A) THE DATA OF 99---1 AND 2 WERE USED FOR THIS FIT.
- 2) V01 AND C0 CALCULATED AS IN 99---1
- 3) SEE ALSO THE ZERO PRESSURE PROPERTIES IN COMMENT 4 PAGE 99---2

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TABLE I

SODIUM SUMMARY

99---0



99---1
SODIUM

NA

$V_0 = 1.033 \text{ CC/G}$
 $V_{01} = 1.034 \text{ CC/G}$.

 $C_0 = 2.33 \text{ KM/SEC.}$

IN THE TABLE BELOW, VELOCITIES ARE GIVE IN MM/MICROSEC.
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0
0.968	4.336	1.417	59.5	0.6732
-	4.229	1.335	54.7	0.6843
-	4.238	1.347	55.3	0.6822
-	4.918	1.842	87.7	0.6255
-	4.883	1.869	88.3	0.6172
-	4.914	1.849	88.0	0.6237
-	5.521	2.407	128.6	0.5640
-	5.529	2.404	128.7	0.5652
-	5.540	2.419	129.7	0.5634
-	5.505	2.375	126.6	0.5686
-	5.561	2.423	130.4	0.5643
-	6.245	2.998	181.2	0.5199
-	6.225	2.996	180.5	0.5187
-	6.925	3.518	235.8	0.4920
-	6.911	3.547	237.3	0.4868
-	7.942	4.300	330.6	0.4586
-	8.036	4.392	341.6	0.4535
-	8.076	4.392	343.8	0.4562

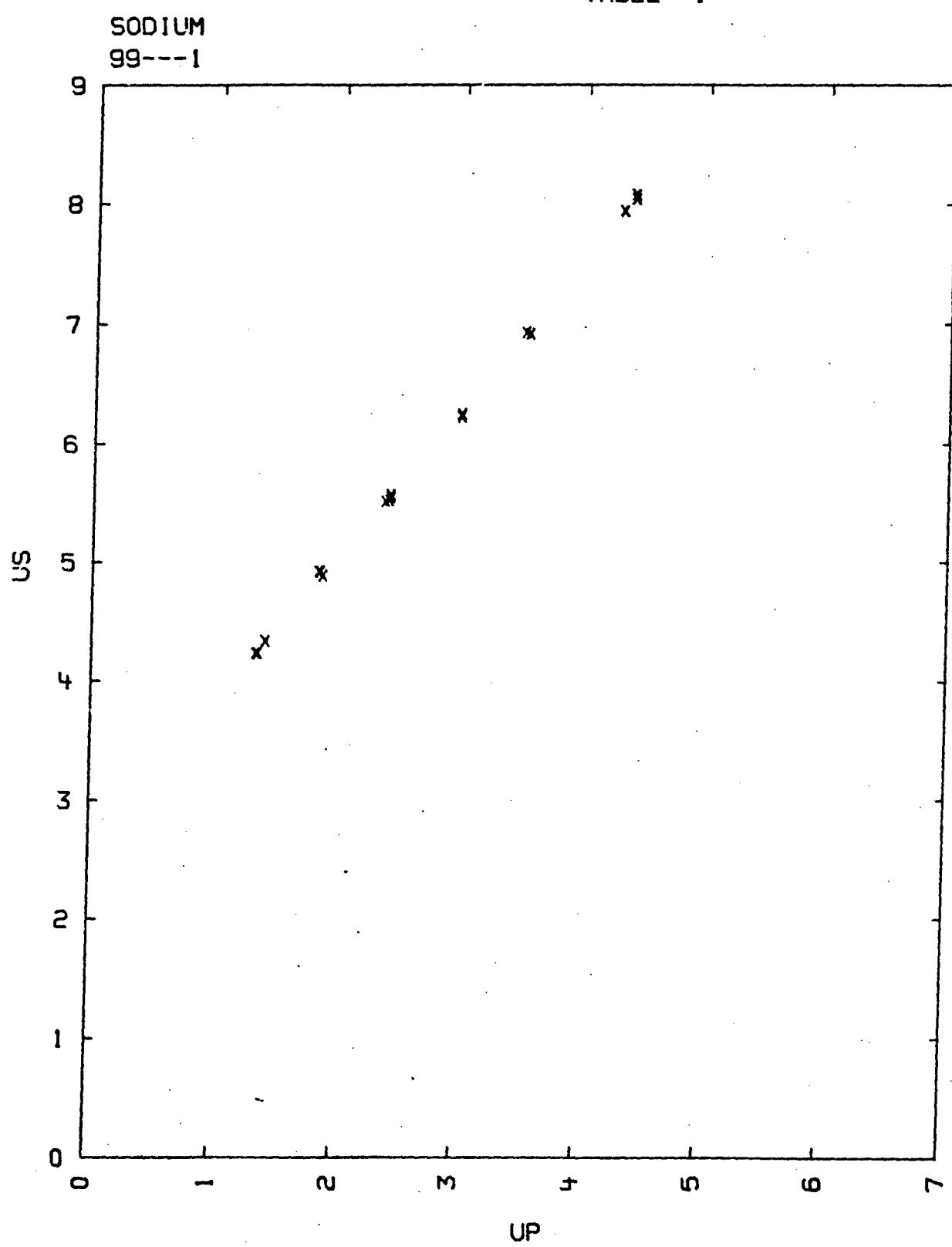
 $US = 2.563 + 1.242 UP \text{ KM/SEC.}$

COMMENTS:

- 1) SOURCE: RICE, M. H.
J. PHYS. CHEM. SOLIDS, VOL. 26, P. 483 (1965)
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 2024 ALUMINUM
- 3) C_0 IS CALCULATED BY USING A BULK MODULUS OF 5.92×10^{10} DYNES/CM 2 .
AMERICAN INSTITUTE OF PHYSICS HANDBOOK, P. 3-81 (MCGRAW-HILL BOOK COMPANY, INC., N.Y., 1957).
- 4) V_{01} IS ALSO OBTAINED FROM THE ABOVE REFERENCE

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TABLE I



99---2
SODIUM

NA

$V_0 = 1.03 \text{ CC/G}$
 $V_{01} = 1.035 \text{ CC/G}$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC,
AND PRESSURE IN KILOBARS.

TABLE

SAMPLE				BASE PLATE		
RHO0	US	UP	P	V/V0	MATERIAL	UP
0.97	3.91	1.10	42	0.719	AL	0.69
-	4.72	1.77	81	0.625	AL	1.14
-	5.53	2.43	132	0.565	AL	1.60
-	7.70	4.09	305	0.467	AL	2.82
-	9.26	5.25	471	0.433	AL	3.70
-	10.57	6.32	648	0.402	PLEXIGLAS	5.85
-	12.52	7.95	966	0.364	PLEXIGLAS	7.38

US = $2.50 + 1.27 \text{ UP KM/SEC}$
SIGMA US = 0.067 KM/SEC

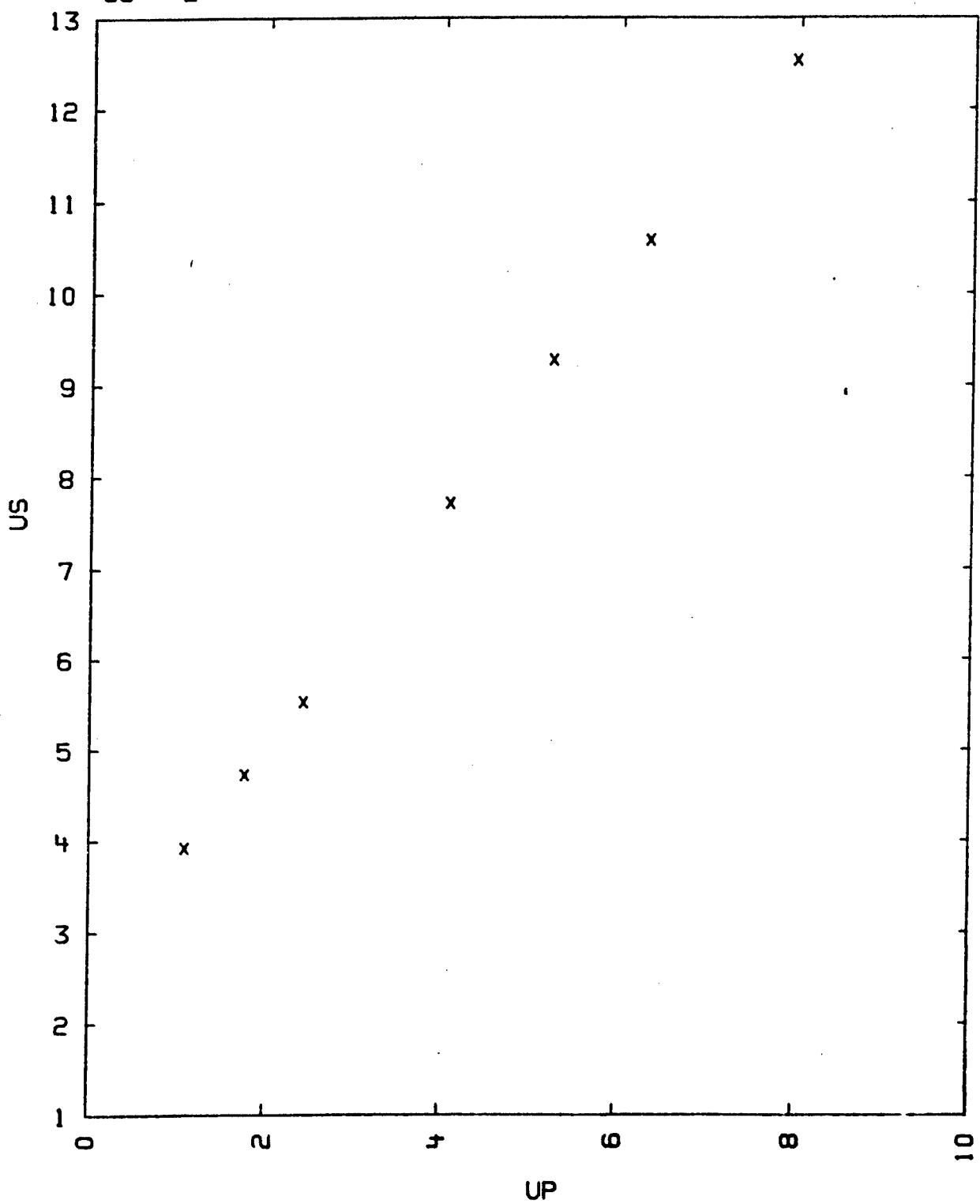
COMMENTS:

- 1) SOURCE: BAKANOVA, A. A., DUODOLADOV, I. P., TRUNIN, R. F.
FIZIKA TVERDOGO TELA, VOL. 7, NO. 6, PP. 1616 - 1622 (1965).
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) THE FOLLOWING HUGONIOT RELATIONSHIPS FOR THE BASE PLATE WERE USED:
AL STANDARD US = $5.25 + 1.39 \text{ UP KM/SEC}$ RHO0 = 2.71 G/CC .
PLEXIGLAS US = $3.10 + 1.32 \text{ UP KM/SEC}$ RHO0 = 1.18 G/CC (FOR US EQUAL OR GREATER THAN 6.0 KM/SEC).
- 4) OTHER LISTED NA PARAMETERS:
AT 300.0 DEG. K
RHO0 0.97 G/CC
COMPRESSIBILITY 16.3 PER MEGABAR
GRUNEISEN COEF. 1.30
AT 0.0 DEG. K
RHO0 1.013 G/CC
COMPRESSIBILITY 14.3 PER MEGABAR
ELECTRONIC HEAT CAPACITY 596 ERG/G DEG. SQUARE
THE VALUE OF V01 WAS OBTAINED FROM A. TAYLOR AND BRENDA J. KAGLE,
CRYSTALLOGRAPHIC DATA ON METAL AND ALLOY STRUCTURES
(DOVER PUBLICATIONS, INC., NEW YORK, N.Y., 1963).

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TABLE I

SODIUM
99---2



100---0
POTASSIUM SUMMARY

K

$$V_0 = 1.16 \text{ CC/G}$$

$$V_{01} = 1.175 \text{ CC/G}$$

$$C_0 = 1.98 \text{ KM/SEC}$$

THE TABLE LISTS THE HUGONIOT POINTS CALCULATED FROM THE FIT GIVEN BELOW.
 UNITS ARE: G/CC, KM/SEC, KBAR AND KBAR.CC/G FOR THE ENERGY DIFFERENCE.

TABLE

RHO0	US	UP	P	V/V0	E-E0
.860	3.145	1.	27.0	0.682	5.00
-	4.321	2.	74.3	0.537	20.0
-	5.497	3.	142.	0.454	45.0
-	6.673	4.	230.	0.400	80.0
-	7.849	5.	338.	0.363	125.
-	10.201	7.	614.	0.314	245.
-	12.553	9.	972.	0.283	405.

$$US = 1.969 + 1.176 \cdot UP, \quad \text{SIG.US} = 0.04 \text{ KM/SEC}$$

FOR UP BETWEEN 1.1 AND 8.4 KM/SEC.

COMMENTS

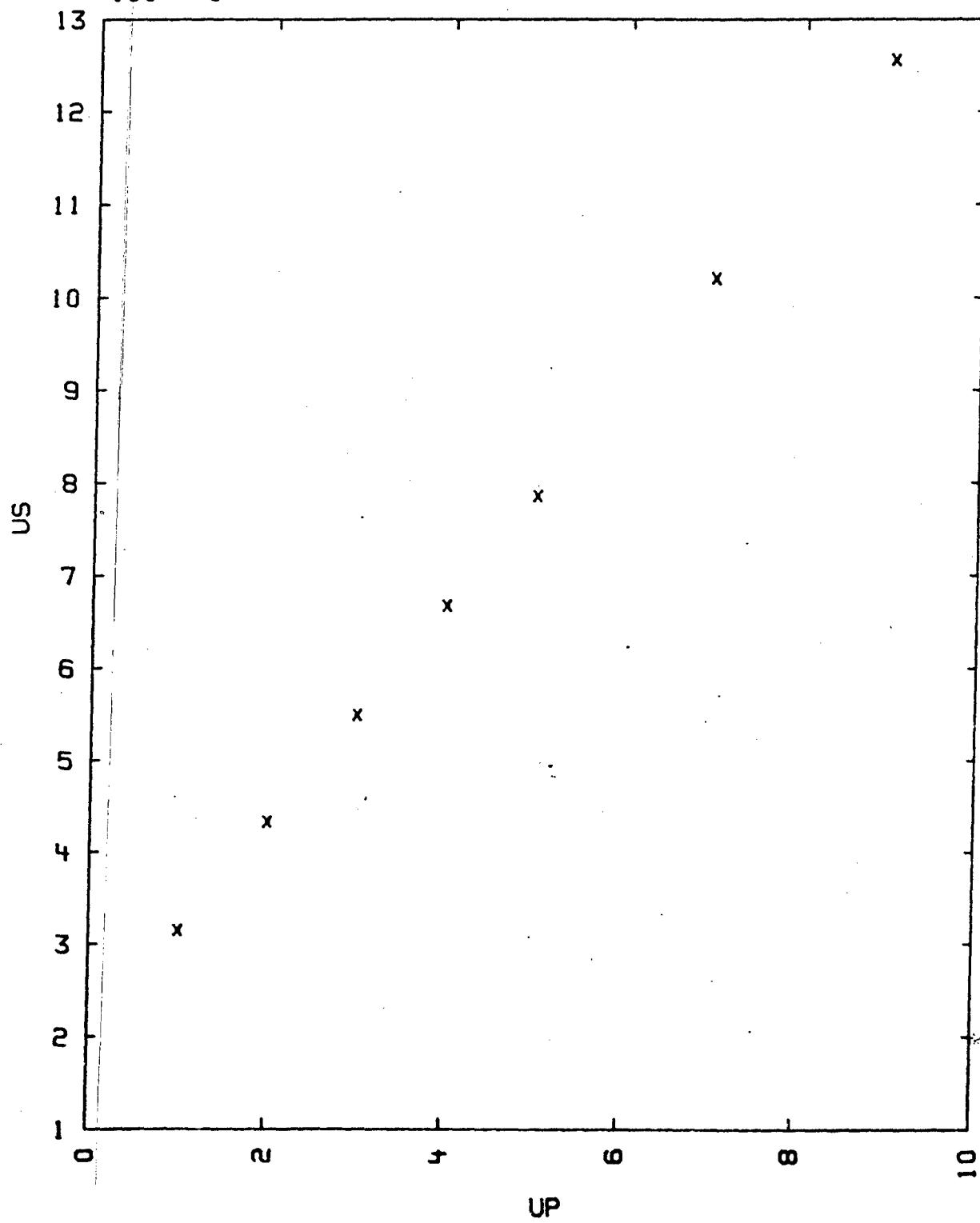
- 1) SOURCE: COMPILER
- 1A) THE DATA FROM 100---1 AND 2 WERE USED FOR THIS SUMMARY.
- 2) ALL POINTS WERE WEIGHTED EQUALLY
- 3) VOI WAS CALCULATED FROM THE CUBIC LATTICE CONSTANT A = 5.344 ANGSTROM
 AM. INST. OF PHYS. HANDBOOK. (MCGRAW-HILL BOOK CO. N. Y., 1963) 2ND.
 ED.
- 4) CO WAS OBTAINED FROM THE CUBIC LATTICE CONSTANTS:
 $C_{11} = 37.1 \text{ KBAR}, \quad C_{12} = 31.5 \text{ KBAR}, \quad C_{44} = 18.8 \text{ KBAR}$
 P. A. SMITH AND C. S. SMITH. J. PHYS. CHEM. SOLIDS., V. 26, P. 279
 (1965)

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TABLE I

POTASSIUM SUMMARY

100---0



100---1
POTASSIUM

K

$$V_0 = 1.163 \text{ CC/G.}$$

$$V_{01} = 1.175 \text{ CC/G.}$$

$$C_0 = 2.06 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0
0.860	3.641	1.391	43.6	0.6180
-	3.633	1.402	43.8	0.6141
-	4.198	1.882	68.2	0.5500
-	4.187	1.928	69.4	0.5395
-	4.258	1.917	70.2	0.5498
-	4.864	2.513	105.1	0.4833
-	4.874	2.532	106.1	0.4805
-	4.921	2.522	106.7	0.4875
-	4.943	2.561	108.9	0.4819
-	4.949	2.561	109.0	0.4825
-	5.489	2.993	141.3	0.4547
-	5.641	3.149	152.8	0.4418
-	5.683	3.174	155.1	0.4415
-	5.747	3.183	157.3	0.4461
-	5.845	3.318	166.8	0.4323
-	7.108	4.324	264.3	0.3917
-	7.258	4.468	278.9	0.3844
-	7.392	4.573	290.7	0.3814

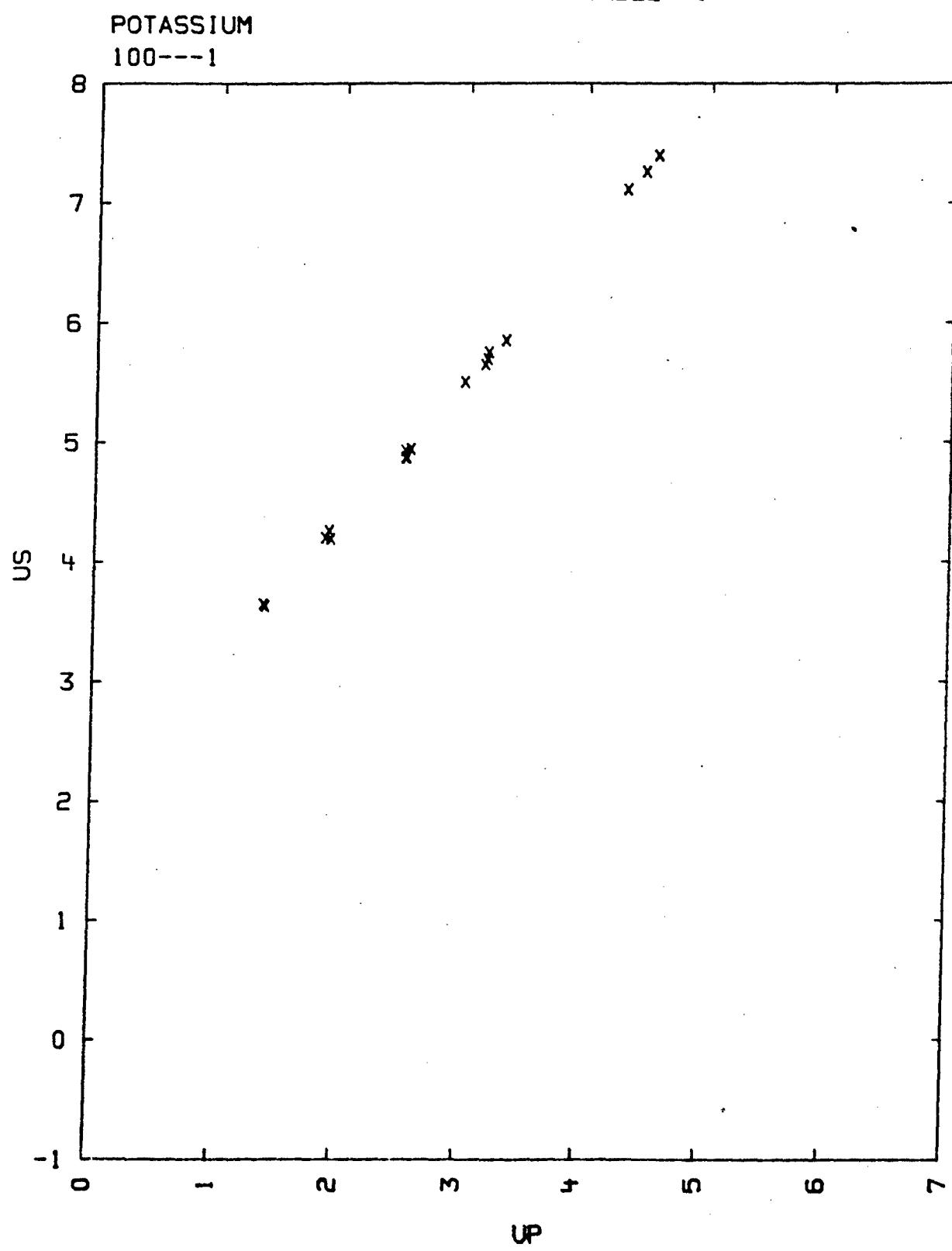
$$US = 1.930 + 1.188 UP, \quad SIC.US = 0.039 \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: RICE, M. H.
J. PHYS. CHEM. SOLIDS, VOL. 26, P. 483 (1965)
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 2024 ALUMINUM
- 3) CO IS A CALCULATED VALUE OBTAINED BY USING A BULK MODULUS OF
3.61 E 11 DYNE/CM².
AMERICAN INSTITUTE OF PHYSICS HANDBOOK, P. 3-81 (MCGRAW-HILL BOOK
COMPANY, INC., N.Y., 1957).
- 4) V01 IS CALCULATED USING A LATTICE CONSTANT OF 5.344 ANGSTROMS.
AMERICAN INSTITUTE OF PHYSICS HANDBOOK, (MCGRAW-HILL BOOK COMPANY,
INC., NEW YORK, 1963) 2ND ED.

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TABLE I



100---2
POTASSIUM

K

$V_0 = 1.16$
 $\rho_{01} = 1.099 \text{ CC/G}$

IN THE TABLE BELOW, DENSITY IS GIVEN IN G/CC, VELOCITIES IN KM/SEC.
AND PRESSURE IN KILOBARS

TABLE

SAMPLE					BASE PLATE	
RHO0	US	UP	P	V/V0	MATERIAL	UP
0.86	3.37	1.15	33	0.658	AL	0.69
-	4.86	2.38	100	0.510	AL	1.50
-	6.99	4.29	257	0.385	AL	2.82
-	9.39	6.32	510	0.327	PLEXIGLAS	6.51
-	11.87	8.41	860	0.292	PLEXIGLAS	7.38

$US = 2.04 + 1.17 UP \text{ KM/SEC}$
 $SIGMA US = 0.044 \text{ KM/SEC}$

COMMENTS:

- 1) SOURCE: BAKANOVA, A. A., DUDOLADOV, I. P., TRUNIN, R. F.
FIZIKA TVERDOGOGO TELA, VOL. 7, NO. 6, PP. 1616 - 1622 (1965)
- 2) EXPERIMENTAL TECHNIQUE A.
DATA REDUCTION TECHNIQUE B.
- 3) THE FOLLOWING HUGONIOT RELATIONSHIPS FOR THE BASE PLATE WERE USED:
AL STANDARD $US = 5.25 + 1.39 UP \text{ KM/SEC}$ $\rho_{00} = 2.71 \text{ G/CC}$
PLEXIGLAS $US = 3.10 + 1.32 UP \text{ KM/SEC}$ $\rho_{00} = 1.18 \text{ G/CC}$ (FOR US EQUAL OR GREATER THAN 6.0 KM/SEC).
- 4) OTHER PARAMETERS LISTED FOR K WERE:
AT 300.0 DEG. K

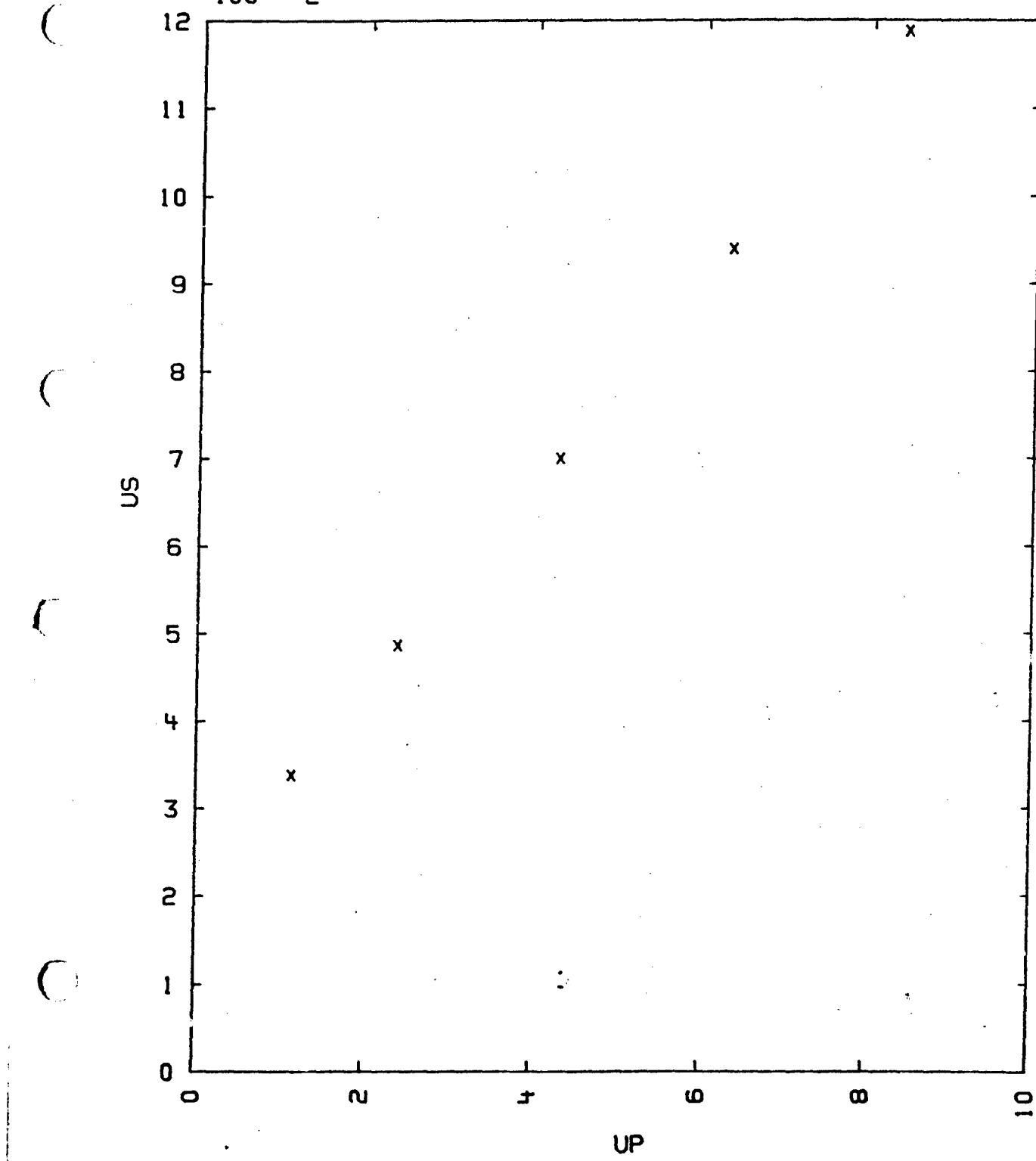
RHO0	0.86 G/CC
COMPRESSIBILITY	35.6 PER MEGABAR
GRUNEISEN COEF.	1.30

AT 0.0 DEG. K

RHO0	0.909 G/CC
COMPRESSIBILITY	18.5 PER MEGABAR
ELECTRONIC HEAT CAPACITY	560 ERG/G DEG. SQUARE

TABLE I

POTASSIUM
100---2



101---1
RUBIDIUM

RB

$$V_0 = 0.6536 \text{ CC/G}$$

$$V_{01} = 0.6536 \text{ CC/G}$$

$$C_0 = 1.13 \text{ KM/SEC.}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

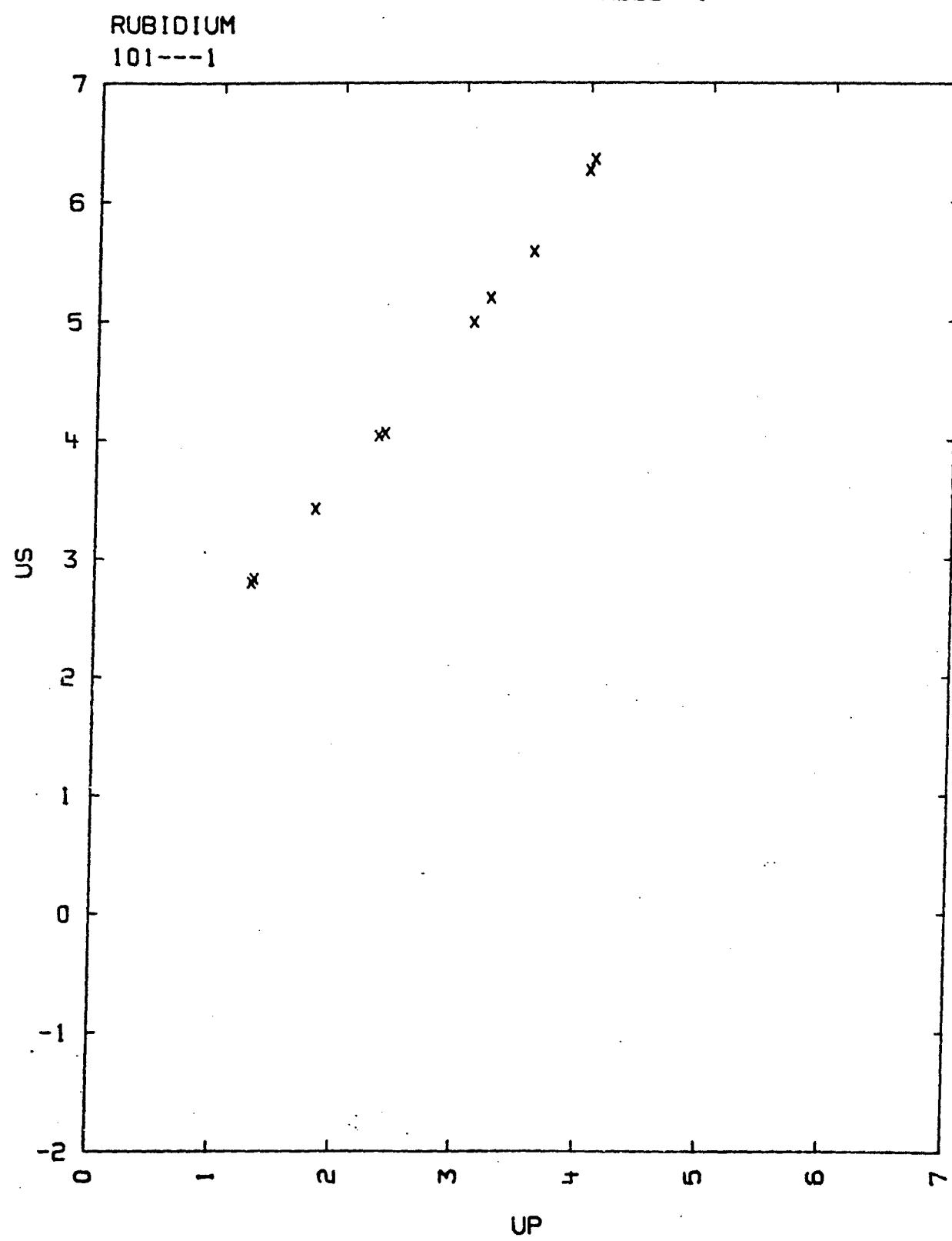
RHO0	US	UP	P	V/V0
1.530	2.786	1.289	54.9	0.5373
-	2.820	1.312	56.6	0.5348
-	3.412	1.814	94.7	0.4683
-	4.032	2.320	143.3	0.4253
-	4.050	2.373	147.0	0.4141
-	4.988	3.085	235.4	0.3815
-	5.187	3.220	255.5	0.3792
-	5.574	3.563	303.9	0.3608
-	6.256	4.001	383.0	0.3604
-	6.349	4.043	392.7	0.3632

$$US = 1.184 + 1.232 UP, \text{ SIG US} = 0.030 \text{ KM/SEC.}$$

COMMENTS:

- 1) SOURCE: RICE, M. H.
J. PHYS. CHEM. SOLIDS, VOL. 26, P. 483, (1965)
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 2024 ALUMINUM
- 3) C0 IS CALCULATED BY USING A BULK MODULUS OF 1.962 E 10 DYNE/CM**2.
C. KITTEL, INTRODUCTION TO SOLID STATE PHYSICS,
(JOHN WILEY AND SONS, INC., NEW YORK, 1956) 2ND ED.

TABLE I



102---1
CESIUM

CS

$$V_0 = 0.5476 \text{ CC/G}$$

$$V_{01} = 0.5219 \text{ CC/G}$$

$$C_0 = 0.87 \text{ KM/SEC}$$

IN THE TABLE BELOW, VELOCITIES ARE GIVEN IN MM/MICROSEC.,
PRESSURE IN KILOBARS AND DENSITY IN G/CC.

TABLE

RHO0	US	UP	P	V/V0
1.826	2.489	1.402	63.7	0.4367
-	2.493	1.395	63.5	0.4404
-	2.994	1.752	96.1	0.4128
-	2.998	1.770	96.9	0.4098
-	3.695	2.239	151.1	0.3940
-	3.719	2.268	154.0	0.3902
-	4.953	3.202	289.6	0.3535
-	5.822	3.894	403.6	0.3312
-	5.847	3.919	418.4	0.3297

$$US = 0.34 + 1.600 \cdot UP + 0.050 \cdot UP^{*2} \text{ KM/SEC}, \text{ SIG.US}=0.013 \text{ KM/SEC}$$

COMMENTS:

- 1) SOURCE: RICE, M. H.
J. PHYS. CHEM. SOLIDS, VOL. 26, P. 483, (1965)
- 2) EXPERIMENTAL TECHNIQUE B
DATA REDUCTION TECHNIQUE B
STANDARD MATERIAL 2024 ALUMINUM.
- 3) CO IS CALCULATED BY USING A BULK MODULUS OF 1.457 E 10 DYNE/CM**2.
C. KITTEL, INTRODUCTION TO SOLID STATE PHYSICS, 2D ED.,
JOHN WILEY AND SONS, INC., NEW YORK, 1957.
- 4) V01 IS CALCULATED USING THE LATTICE CONSTANT OF 6.13 ANGSTROMS.
W. P. PEARSON, INTERNATIONAL SERIES OF MONOGRAPHS ON METAL PHYSICS
AND PHYSICAL METALLURGY, (PERGAMON PRESS, NEW YORK, 1958).

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TABLE I

CESIUM
102---1

